

UNITED STATES OF AMERICA  
NUCLEAR REGULATORY COMMISSION

BEFORE THE ATOMIC SAFETY AND LICENSING BOARD

In the Matter of	)	
	)	
TEXAS UTILITIES ELECTRIC	)	Docket Nos. 50-445 and
COMPANY, <u>et al.</u>	)	50-446
	)	
(Comanche Peak Steam Electric	)	(Application for
Station, Units 1 and 2)	)	Operating Licenses)

APPLICANTS' STATEMENT OF MATERIAL FACTS  
AS TO WHICH THERE IS NO GENUINE ISSUE

1. A 1/16 inch gap was designed into each U-bolt restraint on a rigid frame. As a first support design effort, it was viewed that this gap would accommodate the thermal and seismic movement of piping. (The movement due to a seismic event was preliminarily calculated to be very small, i.e., less than 1/32 inch, for almost all piping.) Accordingly, in the initial pipe support design (prior to as-built conditions), all such U-bolts had been considered as only one-way restraints (because the lateral gap was present). Affidavit at 3.
2. As the as-built design review and corresponding pipe support reanalyses were being conducted, Applicants determined that the thermal movement of piping associated with some U-bolt supports would exceed the 1/16 inch gap provided and some seismic movement may exceed 1/32 inch. Id.

3 As part of the as-built review program Gibbs & Hill reran the thermal piping analyses at all locations where U-bolts were initially considered as one-way restraints and where the piping thermal movement was equal to or exceeded 1/16 inch. Those reanalyses indicated that even assuming the U-bolt acted as a two-way restraint, the piping stresses would remain well within allowable values. Id. at 4.

4. Applicants decided to replace all U-bolts on rigid frames initially considered as one-way restraints where piping thermal movements were computed to equal or exceed 1/16 inch in the original analysis. Id.

5. There are currently seventy Unit 1 and common U-bolts on rigid frames which were originally modeled as one-way restraints. While the maximum thermal pipe movement associated with each of the seventy U-bolts would not exceed the 1/16 inch design gap, the maximum thermal plus seismic movement of eight would. Id. at 6.

6. To assess the impact on piping analyses of a U-bolt installed in the plant acting as a two-way restraint, Applicants conservatively reanalyzed stress problems associated with the two worst case U-bolts (i.e., those with the maximum combined thermal and seismic movement, CC-X-013-012-A43R and CC-1-007-040-A63R) and a representative sample of other U-bolts initially considered as providing one-way restraints. Id. at 6-7.



The reanalyses reflected that any effects of the U-bolts acting as two-way restraints on piping stresses and associated loads (e.g., nozzle and anchor loads) would be small or negligible and would not result in exceeding allowable stresses or manufacturers' allowable values. Further, the analyses reflected that the effects on other associated piping supports are generally decreases in the loads; where there are increases, they are well within allowables. Id. at 8-10.

7. If maximum thermal and seismic movement were assumed to occur simultaneously, there would be a lateral load (in addition to the load in the normal direction) acting on eight of the 70 U-bolts, noted above. Id. at 10. CASE acknowledges that this lateral load will be small when compared to the load in the normal direction (CASE Findings of Fact at II-3).

8. Applicants commissioned ITT Grinnell to carry out a series of tests on U-bolt capability to carry both normal and lateral loads. The tests reflected that even for lateral displacements exceeding the maximum that could occur, the lateral load would not impair the capability of the U-bolt to carry its load in the normal direction. For example, the tests reflect that even if the maximum seismic plus thermal lateral displacement were to induce a lateral load equal to fifty percent of the rated normal load, the U-bolt would still have more than a factor of 2.5 margin of safety in its normal direction. Id. at 10-11.

9. Using conservative assumptions, it was determined that all of the U-bolts at issue here were well within the manufacturer's interaction formula limits. Id. at 12-15.

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- Table 3 Same as Table 2, but for stress problem AB-1-62E; as-built loads dated 5-21-82 and reanalysis loads dated 5-10-84.
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- Table 8 Results of seismic reanalysis of stress problem AB-1-62E, modeling U-bolts on rigid frames as two-way restraints.
- Tables 9a and 9b Results of seismic reanalysis of stress problem AB-1-62F, modeling U-bolts on rigid frames as two-way restraints.
- Tables 10a and 10b Results of seismic reanalysis of stress problem AB-1-65, modeling U-bolts on rigid frames as two-way restraints.

Table 1

Unit 1 and Common Building 1

ALL U-BOLTS ON RIGID FRAMES MODELED AS ONE-WAY RESTRAINTS

	MAXIMUM THERMAL MOVEMENT (IN)	MAXIMUM SEISMIC MOVEMENT (IN)	TOTAL MOVEMENT (IN)	LINE SIZE (IN)	DISTANCE TO NEAREST RIGID SUPPORT IN OPPOSITE DIRECTION
AF-1-038-004-S33R	.002	.0006	.0026	3	1' - 4 1/4"
AF-1-052-035-S62R	.005	.004	.009	4	2' - 9"
AF-1-059-001-S33R	.007	.005	.012	10	1' - 10 11/16"
AF-1-059-003-S33R	.020	.017	.037	8	5' - 2 5/8"
AF-1-067-001-S33R	0.000	.016	.016	3	1' - 9 1/4"
AF-1-098-022-S63R	.031	.004	.035	4	4' - 2 3/16"
AF-1-098-032-S33R	.006	.008	.014	4	3' - 1 1/2"
AF-1-098-700-S33R	.019	.037	.056	4	4' - 0"
AF-1-100-005-S43A	.003	.033	.036	4	1' - 4 1/2"
AF-1-100-007-S53R	0.000	.009	.009	4	0' - 6 1/4"
AF-1-100-009-S53R	.002	.017	.019	4	0' - 11 3/4"
AF-1-100-025-S53R	.003	.031	.034	4	0' - 5"
AF-1-101-003-S33R	.018	.029	.047	4	1' - 5 7/8"
AF-1-101-014-S53R	.001	.022	.023	4	0' - 6 3/4"
BR-X-056-713-A53R	.003	.003	.006	4	0' - 3"
BR-X-162-700-A53R	.029	.025	.054	3	5' - 3 1/4"
CC-1-003-001-A73R	.002	.016	.018	3	4' - 1"
CC-1-005-001-A73R	.019	.022	.041	3	4' - 1"
*CC-1-007-040-A63R	.028	.113	.141	6	5' - 7 1/2"
CC-1-011-023-A53R	.005	.0442	.0492	6	1' - 9"
*CC-1-011-029-A53R	.050	.0668	.1168	6	2' - 0 3/4"
CC-1-065-001-S33R	.013	.003	.016	18	2' - 2 7/16"
CC-1-137-709-E63R	.008	.001	.009	3	5' - 1"
CC-1-146-006-S43R	.004	.0374	.0414	6	4' - 0"
CC-1-146-007-S43R	.008	.0275	.0355	6	2' - 11"
CC-1-148-003-S43R	.013	.0032	.0162	4	1' - 5 7/8"
CC-1-166-005-S43R	.009	.004	.013	4	1' - 4"
CC-2-005-703-A73R	.003	.005	.008	3	3' - 3 1/8"
*CC-2-007-006-A53R	.034	.033	.067	6	6' - 11"
*CC-2-007-007-A53R	.006	.085	.091	6	6' - 11 1/4"
CC-2-020-007-A33R	.001	.0001	.0011	24	1' - 0 1/4"
CC-2-035-712-E33R	.010	.020	.030	6	3' - 8"
CC-2-126-005-F43R	.008	.0261	.0341	12	1' - 3 1/2"
CC-2-126-006-F43R	.006	.039	.045	12	2' - 3 5/8"
CC-2-126-011-F43R	0.000	0.000	0.000	12	0' - 7"
CC-2-164-401-A53R	.001	.001	.002	10	0' - 5 15/16"
*CC-X-013-012-A43R	.038	.1277	.1657	6	7' - 3"
CC-X-021-002-A43R	.014	.0252	.0392	6	3' - 3"
CC-X-023-014-A43R	.003	.0082	.0112	6	1' - 3"
CC-X-025-005-A43R	.014	.0257	.0397	6	11' - 6 1/8"
CC-X-027-008-A43R	.022	.0079	.0299	4	2' - 7"

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Table 1

Unit 1 and Common Building 1

ALL U-BOLTS ON RIGID FRAMES MODELED AS ONE-WAY RESTRAINTS

	MAXIMUM THERMAL MOVEMENT (IN)	MAXIMUM SEISMIC MOVEMENT (IN)	TOTAL MOVEMENT (IN)	LINE SIZE (IN)	DISTANCE TO NEAREST RIGID SUPPORT IN OPPOSITE DIRECTION
AF-1-038-004-S33R	.002	.0006	.0026	3	1' - 4 1/4"
AF-1-052-035-S62R	.005	.004	.009	4	2' - 9"
AF-1-059-001-S33R	.007	.005	.012	10	1' - 10 11/16"
AF-1-059-003-S33R	.020	.017	.037	8	5' - 2 5/8"
AF-1-067-001-S33R	0.000	.016	.016	3	1' - 9 1/4"
AF-1-098-022-S63R	.031	.004	.035	4	4' - 2 3/16"
AF-1-098-032-S33R	.006	.008	.014	4	3' - 1 1/2"
AF-1-098-700-S33R	.019	.037	.056	4	4' - 0"
AF-1-100-005-S43A	.003	.033	.036	4	1' - 4 1/2"
AF-1-100-007-S53R	0.000	.009	.009	4	0' - 6 1/4"
AF-1-100-009-S53R	.002	.017	.019	4	0' - 11 3/4"
AF-1-100-025-S53R	.003	.031	.034	4	0' - 5"
AF-1-101-003-S33R	.018	.029	.047	4	1' - 5 7/8"
AF-1-101-014-S53R	.001	.022	.023	4	0' - 6 3/4"
BR-X-056-713-A53R	.003	.003	.006	4	0' - 3"
BR-X-162-700-A53R	.029	.025	.054	3	5' - 3 1/4"
CC-1-003-001-A73R	.002	.016	.018	3	4' - 1"
CC-1-005-001-A73R	.019	.022	.041	3	4' - 1"
*CC-1-007-040-A63R	.028	.113	.141	6	5' - 7 1/2"
CC-1-011-023-A53R	.005	.0442	.0492	6	1' - 9"
*CC-1-011-029-A53R	.050	.0668	.1168	6	2' - 0 3/4"
CC-1-065-001-S33R	.013	.003	.016	18	2' - 2 7/16"
CC-1-137-709-E63R	.008	.001	.009	3	5' - 1"
CC-1-146-006-S43R	.004	.0374	.0414	6	4' - 0"
CC-1-146-007-S43R	.008	.0275	.0355	6	2' - 11"
CC-1-148-003-S43R	.013	.0032	.0162	4	1' - 5 7/8"
CC-1-166-005-S43R	.009	.004	.013	4	1' - 4"
CC-2-005-703-A73R	.003	.005	.008	3	3' - 3 1/8"
CC-2-007-006-A53R	.034	.033	.067	6	6' - 11"
*CC-2-007-007-A53R	.006	.085	.091	6	6' - 11 1/4"
CC-2-020-007-A33R	.001	.0001	.0011	24	1' - 0 1/4"
CC-2-035-712-E33R	.010	.020	.030	6	3' - 8"
CC-2-126-005-F43R	.008	.0261	.0341	12	1' - 3 1/2"
CC-2-126-006-F43R	.006	.039	.045	12	2' - 3 5/8"
CC-2-126-011-F43R	0.000	0.000	0.000	12	0' - 7"
CC-2-164-401-A53R	.001	.001	.002	10	0' - 5 15/16"
*CC-X-013-012-A43R	.038	.1277	.1657	6	7' - 3"
CC-X-021-002-A43R	.014	.0252	.0392	6	3' - 3"
CC-X-023-014-A43R	.003	.0082	.0112	6	1' - 3"
CC-X-025-005-A43R	.014	.0257	.0397	6	11' - 6 1/8"
CC-X-027-008-A43R	.022	.0074	.0299	4	2' - 7"

	MAXIMUM THERMAL MOVEMENT (IN)	MAXIMUM SEISMIC MOVEMENT (IN)	TOTAL MOVEMENT (IN)	LINE SIZE (IN)	DISTANCE TO NEAREST RIGID SUPPORT IN OPPOSITE DIRECTION
CC-X-031-002-A43R	.034	.0019	.0359	4	5' - 2"
*CC-X-034-005-A43R	.032	.0337	.0675	4	4' - 9 13/16"
CC-X-039-005-F43R	.010	.0031	.0131	12	3' - 7"
CC-X-082-700-A33R	.003	.0159	.0189	24	11' - 8 1/4"
CH-1-240-717-A33R	.002	0.000	.002	3	1' - 3 1/2"
CH-1-249-005-S43R	.001	.018	.019	3	1' - 7"
CH-1-249-010-S33R	.001	.018	.019	3	0' - 8 1/2"
CS-1-015-700-S42R	.017	.0086	.0256	3	1' - 5 1/4"
CS-1-076-005-A42R	.039	.002	.041	4	0' - 3"
*CS-1-155-019-S42R	.018	.051	.069	3	3' - 1/2"
CS-1-155-020-S42R	.020	.018	.038	3	3' - 5/8"
CS-1-251-022-S53R	.031	.0272	.0582	3	3' - 3"
CS-1-319-008-S53R	.005	.0099	.0149	3	1' - 1 1/4"
CS-1-454-008-S52R	.004	.0139	.0179	3	0' - 5 3/8"
CH-1-005-008-C42R	.001	.001	.002	3	0' - 5 1/2"
SF-X-005-031-F53R	0.000	.0046	.0046	10	3' - 10"
SF-X-006-010-F53R	0.000	.0039	.0039	10	3' - 10"
SF-X-024-702-A43R	.030	.0194	.0494	4	2' - 9 5/8"
SF-X-034-012-F53R	0.000	.0021	.0021	10	3' - 10"
SI-1-029-055-S32R	0.000	.0273	.0273	24	2' - 11 1/4"
SI-1-031-004-A32R	.003	.0028	.0058	12	0' - 8 9/16"
SI-1-039-016-S22R	.002	.005	.007	4	0' - 4 7/8"
SI-1-039-017-S22R	0.000	.018	.018	4	1' - 2 5/8"
SI-1-039-022-S22R	0.000	.016	.016	4	0' - 5 7/8"
SW-2-010-004-A33R	.019	.010	.029	24	1' - 4 5/8"
*SW-2-102-018-A43R	.044	.0347	.0787	10	5' - 6"
SW-2-129-022-A43R	.004	.014	.018	10	0' - 6"
SW-2-129-027-A43R	.007	.045	.052	10	1' - 9 1/8"
WP-1-049-700-S43R	.005	.002	.007	3	1' - 1"

\* Indicates two-way action

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Table 2

SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX
		FX	FY	FZ	FX	FY	FZ	%CHANGE
CC-X-004-001-A43R	A		579			519		.2
	B		951			713		.3
	C		1017			1008		.3
CC-X-004-002-A43R	A		235			255		0
	B		353			353		0
	C		394			394		0
CC-X-004-003-A43R	A		357			357		0
	B		602			602		0
	C		639			639		0
CC-X-004-004-A43R	A		421			421		0
	B		807			807		0
	C		972			972		0
CC-X-004-005-A43R	A		340			340		0
	B		551			551		.2
	C		621			621		0
CC-X-004-006-A43R	A		435	25		435	25	0
	B		679	603		679	592	-2
	C		761	755		760	743	-1.6
CC-X-004-007-A43R	A		243			243		0
	B		413			413		0
	C		458			458		0
CC-X-004-008-A43R	A		166	85		166	85	0
	B		475	636		475	626	-1.6
	C		562	734		562	772	-1.6
CC-X-004-009-A43R	A	46			46			0
	B	813			311			-2
	C	1127			1126			-0.09

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SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX
		FX	FY	FZ	FX	FY	FZ	%CHANGE
CC-X-004-010-A43R	A			120			120	0
	B			635			613	-3.5
	C			752			727	-3.4
CC-X-004-011-A43R	A			238			238	0
	B			682			674	-1.1
	C			794			784	-1.2
CC-X-004-012-A43R	A	390			390			0
	B	699			693			-0.9
	C	736			719			-2.3
CC-X-005-001-A43R	A		165			165		0
	B		203			203		0
	C		214			214		0
CC-X-005-002-A43S	A		119			119		0
	B							
	C							
CC-X-005-003-A43K	A							
	B			232			227	-2.2
	C			292			287	-1.7
CC-X-005-004-A43R	A	285			285			0
	B	461			459			-0.4
	C	519			518			-0.2
CC-X-006-001-A43S	A		199			199		0
	B							
	C							
CC-X-006-002-A43S	A		360			360		0
	B							
	C							

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SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX
		FX	FY	FZ	FX	FY	FZ	%CHANGE
CC-X-006-003-A43R	A	87			87			0
	B	199			196			-1.5
	C	228			226			-0.9
CC-X-013-001-A43R	A		909			909		
	B		1660			1693		2.1
	C		1807			1856		2.8
CC-X-013-002-A43R	A		413			413		0
	B		905			896		-0.9
	C		998			992		-0.6
CC-X-013-003-A43R	A		507	99		507	99	0
	B		1006	196		997	190	-1.2
	C		1141	222		1131	216	-1.0
CC-X-013-004-A43R	A		413			413		0
	B		780			778		-0.2
	C		882			883		+0.1
CC-X-013-006-A43R	A		388			388		0
	B		614			615		+0.16
	C		659			662		+0.45
CC-X-013-009-A43S	A		545			545		0
	B							
	C							
CC-X-013-010-A43R	A	101			101			0
	B	365			384			+5.2
	C	443			485			+9.5
CC-X-013-011-A43R	A		221	186		221	186	0
	B		310	390		299	313	-4.0
	C		334	439		316	419	-4.8



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LOADS 8-16-82

LOADS 5-10-84

MAX

SUPPORT NUMBER		FX	FY	FZ	FX	FY	FZ	%CHANGE
* CC-X-013-012-A43R	A		161			161	0	
	B		211			213	199	
	C		597			232	271	
CC-X-013-013-A43R	A	200	409		200	409		0
	B	432	741		401	733		-2.0
	C	433	320		452	317		-1.6
CC-X-013-014-A43R	A	52			52			0
	B	423			417			-2.6
	C	513			517			-1.2
CC-X-013-015-A43R	A	306			306			0
	B	1143			1131			-1.0
	C	1355			1369			+1.0
CC-X-013-016-A43R	A		72	162		72	162	0
	B		321	720		309	693	-3.7
	C		379	850		366	821	-3.4
CC-X-013-017-A43R	A		14	32		14	32	0
	B		128	288		118	266	-7.6
	C		160	360		149	336	-6.7
CC-X-013-018-A43R	A		168	726		163	726	0
	B		351	1520		353	1526	+1.7
	C		388	1680		397	1718	+2.2
CC-X-014-753-A43R	A		272			272		0
	B							
	C							
CC-X-014-751-A43R	A	320			320			0
	B	893			744			-16.7
	C	1045			924			-11.6

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SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX
		FX	FY	FZ	FX	FY	FZ	%CHANGE
<u>CC-X-023-001-A43R</u>	A		598			593		0
	B		520			527		-1.6
	C		684			2557		-1.6
<u>CC-X-023-002-A43R</u>	A		478			478		0
	B		952			855		-10.2
	C		1055			932		-11.6
<u>CC-X-023-003-A43R</u>	A		650			222		0
	B		1301			274		-2.1
	C		1413			1380		-2.3
<u>CC-X-023-005-A43R</u>	A		480			480		0
	B		939			931		-5.9
	C		1143			1066		-6.7
<u>CC-X-023-006-A43R</u>	A		402			402		0
	B		834			714		-14.4
	C		941			800		-15.0
<u>CC-X-023-007-A43R</u>	A	2758	452		2758	452		0
	B	5095	939		5105	920	7135	-1
	C	5573	1122		5586	1040		-1.05
<u>CC-X-023-008-A43R</u>	A		207	426		207	426	0
	B		776	1593		751	1542	-3.2
	C		875	1795		839	1722	-4.1
<u>CC-X-023-009-A43R</u>	A		56			56		0
	B		471			413		-12.3
	C		563			497		-11.7
<u>CC-X-023-010-A43R</u>	A		69	142		69	142	0
	B		332	632		319	656	-3.8
	C		387	793		369	753	-4.5

AB-1-65

SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX %CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC-023-011-A43R	A		160	237		160	237	0
	B		1493	1925	1332	1235	120	-1.2
	C		1542	2237	1398	1520	2253	-1.4
CC-X-023-012-A43R Note Pt 19921	A		155			155		0
	B		335			312		-6.9
	C		332			359		+7.5
CC-X-023-012-A43R Note Pt 792	A	2578			2513			0
	B	4359			4341			-.4
	C	4739			4717			-.5
CC-X-023-013-A43K	A							
	B			279			159	-43.
	C			350			211	-39.7
* CC-X-023-014-A43R	A			326		-	326	
	B			636		71	613	
	C			700		91	681	
* CC-X-023-015-A43R	A			434		-	434	
	B			835		88	834	
	C			907		107	907	
CC-X-024-002-A43R	A	44			44			0
	B	131			154			-14.9
	C	219			195			-10.9
CC-X-024-003-A43S Note Pt 9956	A		328			328		0
	B							
	C							
CC-X-025-003-A43S Note Pt 9941	A		548			548		0
	B							
	C							

AB-1-65

SUPPORT NUMBER		LOADS 8-16-82			LOADS 5-10-84			MAX
		FX	FY	FZ	FX	FY	FZ	%CHANGE
<u>CC-X-025-004-A43R</u>	A	56			56			0
	B	333			222			-33.3
	C	392			284			-27.5
<u>CC-X-025-005-A43R</u>	A		456			456	-	
	B		568			540	203	
	C		596			568	203	
<u>CC-1-106-001-A43R</u>	A		2165			2165		0
	B		2546			2552		+1.2
	C		2635			2644		+1.3
<u>CC-1-106-002-A43R</u>	A	3251		236	3251		236	0
	B	3378		788	3378		792	~0
	C	3413		905	3413		913	~0
<u>CC-1-120-001-A43R</u>	A		4244			4244		0
	B		4984			4999		+1.3
	C		5135			5158		+1.4
<u>CC-1-120-002-A43R</u>	A			1119			1119	0
	B			1392			1369	-1.6
	C			1452			1427	-1.7
<u>CC-1-123-001-A43R</u>	A		1992			1992		0
	B		2773			2771		~0
	C		2938			2936		~0
<u>CC-1-123-002-A43R</u>	A			1164			1164	0
	B			2424			2418	-1.2
	C			2689			2632	-1.3
<u>CC-X-024-004-A43R</u>	A		198			198		0
	B		280			267		-4.6
	C		304			270		-11.2



Seismic Displacements

	$\Delta X$	$\Delta Y$	$\Delta Z$
CC-X-013-012-A43R Node Pt. 9850	+0.058 -0.057	0.0 0.0	+0.028 -0.030
CC-X-025-005-A43R Node Pt. 9932	+0.027 -0.026	0.0 -0.001	+0.019 -0.019
CC-X-023-014-A43R Node Pt. 6	+0.011 -0.012	+0.004 -0.010	+0.001 -0.001
CC-X-023-015-A43R Node Pt. 77	+0.014 -0.014	+0.006 -0.011	+0.001 -0.001



Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPSES

Subject As-Built Stress Verification Analysis Results-Maximum Stress

Calculation Number Problem AB-1-065 Sheet No. 20

Revision	Original Date	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
1										
Prepared	J.J. P. 7/27/82									
Checked	J.B.W. 7/28/82									

AB-1-65 WITHOUT U-BOLT LATERAL STIFFNESS PER QIA BOOK

### MAXIMUM STRESS RESULTS

Plant Condition	Equation	(1) Allowable Stress	Calculated Stress	Node	S.I.F.	Description
Normal and Upset	8	$S_h$ 15000	2457	937	3.26	Unreinforced BRANCH CONNEX
Normal and Upset	9	$1.2 S_h$ 18000	14698	989	1.699	Elbow
Emergency	9	$1.8 S_h$ 27000	17582	989	1.699	Elbow
Faulted	9	$2.4 S_h$ 36000	Faulted EQUATION 9 IS LIMITED BY EMERGENCY EQUATION 9			
Normal and Upset	10 (2)	$S_A$ 22500	15812	914	7.44	45° Lateral tee CONNEX
Normal and Upset	11	$S_A + S_h$ 37500	17514	914	7.44	45° Lateral tee CONNEX.

(1) Piping Material: SA-106 Gr. B  
Design Temperature: 200°F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.

Checking Method #

1. Use by the drafting
2. Auditing Calculation file is compared
3. Computer Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes

F-166, 4-81

Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPSES

Subject Is-Built Stress Verification-Analysis, Results-Maximum Stress

Calculation Number Problem AB-1-65

Sheet No. 1 TEST R1

Revision	Date	# Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Design									
Prepared									
Checked									

AB-1-65 WITH U-BOLT LATERAL STIFFNESS; PER COMPUTER ANALYSIS J219 L.H. MAY 8, 1984; OUTPUT TAPE T15176/021  
MAXIMUM STRESS RESULTS

Plant Condition	Equation	(1) Allowable Stress	Calculated Stress	Node	S.I.F.	Description
Normal and Upset	8	$S_h$ 15,000.0	2457.0	937	3.26	Unreinforced Branch Connection
Normal and Upset	9	$1.2 S_h$ 18,000.0	14686.0	989	1.699*	ELBOW
Emergency	9	$1.8 S_h$ 27,000.0	17566.0	989	1.699*	ELBOW
Faulted	9	$2.4 S_h$ 36000.0	FAULTED CONDITION EQUATION 9 IS BOUNDED BY EMERGENCY CONDITION EQUATION 9			
Normal and Upset	10 (2)	$S_A$ 22500.0	15812.0	914	7.44	45° Lateral TEE CONNECTION
Normal and Upset	11	$S_A + S_h$ 37500.0	17515.0	914	7.44	45° Lateral TEE CONNECTION

(1) Piping Material: SA-106 Gr. B  
Design Temperature: 200°F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.

\* S.I.F. = 0.756 = (0.75)(2.266) = 1.699 FOR EQUATION 9

Checking Method #

1. Verify the drawing.  
2. Verify the calculation results.  
3. Compare the calculation results with the design code.  
4. Compare the results of the calculation with the results of the stress analysis.

F-166, 4-81

Gibbs & Hill, Inc. Job No. 2323 Client TUSI-CPSES  
 Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
 Calculation Number PROBLEM AB-1-65 Sheet No. ORIGINAL RUN RESULTS

Revision	Drawn	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
1			1	5/17/84						
2			2	5/17/84						
3			3	5/17/84						
4			4	5/17/84						

BRP/OTW ER CC-X-AB-012A - NODE: 9945 (NOZZLE) IDENT: EVAP. CONDENSER TAX - WAEVND-02

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	42		133		153		BOUNDED BY	
$F_{AXIAL} Y$	480		520		529			
$F_z$	157		240		264		EMERG.	
PRY Shear	157		274		305			
$M_x$	225		317		344			
$M_{TAX} Y$	86		178		209			
$M_z$	320		428		452			
WRT Bend	341		533		568			

BRP/OTW ER CC-X-AB-012B - NODE: 9962 4" NOZZLE IDENT: DIST. COOLER TAX - WAEVND-02

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	53		119		145		BOUNDED BY	
$F_{AXIAL} Y$	24		78		97			
$F_z$	65		101		113		EMERG.	
PRY Shear	73		156		184			
$M_x$	303		430		468			
$M_{TAX} Y$	154		214		225			
$M_z$	13		45		56			
WRT Bend	303		432		471			

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see

(3) For anchors, complimentary problem results must be tabulated.

Checking Method #

See the drawing for details. The analysis was performed using the following software: FEA 100, 4-81

F-100, 4-81

AB-1-65 WITHOUT U-BOLT LATERAL STIFFNESS - PER Q.A. BOOK



Gibbs &amp; Hill, Inc.

Job No.

2323

Client

TUSI-CPSES

Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS - ANCHOR/NOZZLE STRESS.

Calculation Number PROBLEM AB-1-065

Sheet No. ORIGINAL RUN RESULTS

Revision	Original	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method										
Prepared	BKG		JSA	5/17/81						
Checked			JSA	5/17/81						

BRP/OTHER CC-X-AB-008 NODE: 9864 (6" NOZZLE) IDENT: EVAP. CONDENSER TBX-WPEVND-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$P_x$	44		132		159		BOUNDED	
$P_y$	388		691		746		BY	
$P_z$	34		441		516		EMERGENCY	
FR/Shear	56		460		540			
$M_x$	107		638		735			
$M_y$	143		731		838			
$M_z$	237		576		643			
MR/Bend	260		860		977			

BRP/OTHER CC-X-AB-008 NODE: 9873 (4" NOZZLE) IDENT: DIST. COOLER TBX-WPEVND-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$P_x$	78		279		333			
$P_y$	99		162		176		BOUNDED	
$P_z$	44		106		123		BY	
FR/Shear	90		298		355		EMERGENCY	
$M_x$	161		371		429			
$M_y$	64		104		112			
$M_z$	37		121		144			
MR/Bend	165		390		453			

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see

(3) For anchors, complimentary problem results must be tabulated

Checking Method #

Under the guidance of the American Institute of Steel Construction, Inc. (AISC) and the American Institute of Welding (AIW), the following information is provided for the design of welded connections. This information is based on the latest available data and is subject to change without notice. The design of welded connections should be based on the latest available data and is subject to change without notice. The design of welded connections should be based on the latest available data and is subject to change without notice.

F-186, 4-81

AB-1-65 WITHOUT U-BOLT LATERAL STIFFNESS PER QA BOOK

Gibbs & Hill, Inc. Job No. 2323 Client TUSI-CPSES  
 Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
 Calculation Number PROBLEM AB-1-65 Sheet No. ORIGINAL RUN RESULTS

Revision	Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Design										
Prep										
Check										

BRP/OTHER CC-X-AB-004-NODE: 9804 (4" NOZZLE) IDENT: DIST. COOLER TRX-BREVE

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL.	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>	8		21		24		BOUNDED BY	
F <sub>y</sub>	16.5		176		179			
F <sub>z</sub>	20		52		60		EMERGENCY	
PRY/Shear	22		58		65			
M <sub>x</sub>	125		228		254			
M <sub>y</sub>	16		20		21			
M <sub>z</sub>	141		188		200			
MAX Bend	188		296		323			

BRP/OTHER CC-X-AB-004-NODE: 9817 (6" NOZZLE) IDENT: EVAP. CONDENSER TRX-BREVE-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL.	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>	60		99		109			
F <sub>y</sub>	464		623		668		BOUNDED BY	
F <sub>z</sub>	78		110		119			
PRY/Shear	98		148		161		EMERGENCY	
M <sub>x</sub>	230		414		467			
M <sub>y</sub>	107		126		131			
M <sub>z</sub>	237		363		401			
MAX Bend	328		551		616			

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see (3) For anchors, complimentary problem results must be tabulated F166, 4-51

Checking Method #

1) AS-BUILT  
 2) AS-BUILT  
 3) AS-BUILT

AB-1-65 WITHOUT U-BOLT LATERAL STIFFNESS  
 AS PER QA BOOK



Gibbs & Hill, Inc. Job No. 2323 Client TUSI-CPSES  
 Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
 Calculation Number PROBLEM AB-1-65 Sheet No. \*TEST RUN

Revision	Drawn	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checked										
Prepared										
Checked										

BRP/OTHER CC-X-AB-12A-NODE: 9445 (6" NOZZLE) IDENT: EVAP CONDENSER  
 TBX-WPEVND-02

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>	48.		106.		127.		BOUNDED BY	
F <sub>y</sub>	453.		464.		468.			
F <sub>z</sub>	124.		200.		226.		EMERGENCY	
FR/Shear	133.		226.4		259.2			
M <sub>x</sub>	145.		283.		313.			
M <sub>y</sub>	69.		189.		227.			
M <sub>z</sub>	310.		592.		423.			
MR/Bend	366.2		483.5		526.2			

BRP/OTHER CC-X-AB-12B-NODE: 9442 (4" NOZZLE) IDENT: DIST. COOLER  
 TBX-WPEVND-02

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>	32.		110.		138.			
F <sub>y</sub>	19.		76.		95.		BOUNDED BY	
F <sub>z</sub>	47.		76.		88.			
FR/Shear	56.4		133.7		163.7		EMERGENCY	
M <sub>x</sub>	241.		343.		383.			
M <sub>y</sub>	120.		176.		190.			
M <sub>z</sub>	12.		43.		54.			
MR/Bend	241.3		345.7		386.8			

Notes: 1) Forces are in lbs, moments in ft-lbs  
 2) For allowable loads see

Checking Method #

Use the following  
 American Institute of Steel Construction, Inc.  
 Allowable Stress Design and Plastic Design  
 Specification for Structural Steel Buildings

(3) For anchors, complimentary  
 problem results must be  
 tabulated F-166, 4-81

AB-1-65 WITH U-BOLT LATERAL STIFFNESS  
 Job# 219 dtd. 5/8/84 OUTPUT TAPE T15176/021

Gibbs & Hill Inc. Job No. 2323 Client TUSI-CPSES  
 Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
 Calculation Number PROBLEM AB-1-65 Sheet No. \*TEST RUN

Revision	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Design									
Preparer									
Checker									

BRP/OTHER CC-X AB-008 NODE: 9864 -6" NOZZLE IDENT: EVAP-CONDENSE  
 TBX-SKAFVND-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL.	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	-52.		147		-180.		BOUNDED BY	
$F_{Y_{ALL}}$	454		494		508.			
$F_z$	-57.		139.		-168.		EMERGENCY	
PR/Shear	77.2		202.3		246.2			
$M_x$	223.		344.		387.			
$M_{Y_{ALL}}$	52.		237.		304.			
$M_z$	181.		313.		442.			
MR/Bend	287.2		507.4		587.5			

BRP/OTHER CC-Y AB-008 NODE: 9873 4" NOZZLE IDENT: DIST COOLER  
 TBX-NRGEVND-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL.	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	88.		236.		300.			
$F_{Y_{ALL}}$	102.		159.		174.		BOUNDED BY	
$F_z$	45.		114.		137.			
PR/Shear	98.8		262.1		329.8		EMERGENCY	
$M_x$	163.		398.		478.			
$M_{Y_{TENSION}}$	54.		65.		69.			
$M_z$	43.		105.		132.			
MR/Bend	168.6		411.6		485.9			

Notes: 1) Forces are in lbs, moments in Ft-lbs  
 2) For allowable loads see (3) For anchors, complimentary problem results must be tabulated F-155, 4-81  
 Checking Method #

AB-1-65 WITH U-BOLT LATERAL STIFFNESS  
 Job # 219 04d 5/8/84: OUTPUT TAPE T15176/021

Gibbs & Hill, Inc. Job No. 2323 Client TUST-CPSES  
 Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
 Calculation Number PROBLEM AB-1-65 Sheet No. \* TEST RUN

Revision	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #									
Prepared									
Checked									

BRP/OTHER CC-X-AB-04 - NODE: 9804 (4" NOZZLE) IDENT: DIST. COOLER TBX-BREVRE-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	8		-20		-23		BOUNDED	
$F_y$	165		176		179		BY	
$F_z$	20		51		58		EMERGENCY	
PR/Shear	21.5		54.8		62.4			
$M_x$	125		222		247			
$M_y$	16		20		21			
$M_z$	-141		-185		-197			
MR/Bend	188.4		289.0		316.0			

BRP/OTHER CC-X-AB-04 - NODE: 9817 (6" NOZZLE) IDENT: EVAP. CONDENSER TBX-BREVRE-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	-60		-97		-108			
$F_y$	464		623		668		BOUNDED	
$F_z$	78		109		118		BY	
PR/Shear	98.4		146.0		160.0		EMERGENCY	
$M_x$	230		414		467			
$M_y$	107		125		130			
$M_z$	-234		-362		-399			
MR/Bend	328.1		550.0		614.2			

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see

(3) For anchors, complimentary problem results must be tabulated

Checking Method #

Use the following  
 American Institute of Steel Construction, Inc.  
 Allowable Stress Design and Plastic Design  
 Specification for Structural Steel Buildings

F-166, 4-81

AB-1-65 WITH U-BOLT LATERAL STIFFNESS  
 106# 219 dtd. 5/8/84 OUTPUT TABS T15176/021



AB-1-62E

Table 3

SUPPORT NUMBER		LOADS 5-21-82			LOADS 5-10-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC-1-007-007-A43S	A		337			337		0
	B							
	C							
CC-1-007-013-A43R	A		450			450		0
	B		693			690		-0.4
	C		731			728		-0.4
CC-1-007-014-A43S	A		322			322		0
	B							
	C							
CC-1-007-015-A43K	A							
	B	1015			1013			-0.1
	C	1254			1252			-0.1
CC-1-007-016-A43R	A			78			78	0
	B			1370			1340	-2.1
	C			1667			1634	-1.9
CC-1-007-017-A43R	A		622			622		
	B		1601			1469		-8.2
	C		1795			1648		-8.1
CC-1-007-018-A43R	A			4			4	
	B			343			365	6
	C			416			444	6
CC-1-007-019-A43R	A		667			667		
	B		1736			1483		-17
	C		1882			1615		-16
CC-1-007-020-A43R	A			4			4	
	B			683			735	+8
	C			845			908	+8

AB-1-62E

SUPPORT NUMBER		LOADS 5-21-82			LOADS 5-10-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC-1-007-021-A43R	A		558			558		
	B		1304			1284		-1
	C		1459			1458		-0.07
CC-1-007-022-A43R	A		381			381		
	B		626			624		-0.3
	C		677			690		+0.4
CC-1-007-023-A43R	A			170			170	
	B			580			576	-1
	C			656			647	-2
CC-1-007-024-A43R	A		276			276		
	B		428			425		-0.5
	C		458			455		-0.4
CC-1-007-025-A43R	A	49			49			
	B	563			721			+29
	C	763			944			+25
CC-1-007-026-A43K	A							
	B			604			614	+2
	C			884			896	+1
CC-1-007-027-A43K	A							
	B	492			540			+10
	C	731			787			+8
CC-1-007-028-A43K	A							
	B		330			327		-0.9
	C		441			438		-0.7
CC-1-007-029-A43R	A			49			49	
	B			1068			1023	-4.2
	C			1236			1191	-3.6



AB-1-62E

SUPPORT NUMBER		LOADS			LOADS			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC-1-007-030-A53R	A	159			159			
	B	789			652			-17
	C	941			790			-16
<u>CC-1-007-031-A53S</u>	A		1166			1166		
	B							
	C							
<u>CC-1-007-032-A53R</u>	A			61			61	
	B			969			979	+1
	C			1183			1207	+2
<u>CC-1-007-033-A63R</u>	A	92			92			
	B	878			730			-17
	C	1013			855			-15
<u>CC-1-007-034-A63K</u>	A							
	B			962			889	-8
	C			1158			1090	-6
* <u>CC-1-007-035-A63R</u>	A		526			526		
	B		3396		291	2828	291	
	C		3852		362	3258	362	
<u>CC-1-007-036-A63R</u>	A	50			50			
	B	729			577			-20
	C	835			701			-16
<u>CC-1-007-037-A63R</u>	A		630			630		
	B		1566			1481		-5
	C		1710			1636		-4
<u>CC-1-007-038-A63R</u>	A	11			11			
	B	813			690			-15
	C	988			858			-13

AB-1-62E

SUPPORT NUMBER		LOADS 5-21-82			LOADS 5-10-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC-1-007-039-A63R	A		691			691		
	B		1652			2006		+21
	C		1741			2284		+31
* CC-1-007-040-A63R	A	7			7			
	B	1021			812	806		
	C	1228			998	957		
CC-1-007-041-A63R	A		704			704		
	B		1762			1691		-4
	C		1833			1854		+1
CC-1-007-042-A63R	A	20			20			
	B	715			526			-3
	C	848			642			-2
CC-1-007-043-A63R	A			68			68	
	B			2672			2230	-16
	C			2980			2551	-14
CC-1-007-044-A73R	A		1237			1237		
	B		2884			2620		-9
	C		2975			2714		-9
CC-1-007-045-A73R	A	30			30			
	B	1566			1284			-21
	C	1941			1583			-22
CC-1-007-048-A43R	A			50			50	
	B			948			934	-1
	C			1164			1149	-1
CC-1-948-700-A73R	A			44			44	
	B			1014			832	-18
	C			1145			962	-16

AB-1-62E

LOADS 5-21-82      LOADS 5-10-84

SUPPORT NUMBER		FX	FY	FZ	FX	FY	FZ	%CHANGE
CC-1-007-704-A43R	A	101			101			
	B	2222			2856			+28
	C	2941			3677			+25
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							

## SEISMIC DISPLACEMENT

	$\Delta X$	$\Delta Y$	$\Delta Z$
CC-1-007-035-A63R	+ .06	+ .002	+ .038
	- .06	- .002	- .038
CC-1-007-040-A63R	+ .001	+ .067	+ .044
	- .001	- .073	- .044



# AB-162E WITHOUT U-BOLT LATERAL STIFFNESS PER Q.A. BAK

Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPSES

Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS - ANCHOR/NOZZLE/PENE.

Calculation Number PROBLEM AB-1-62E Sheet No. 23

Revision	Original	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking										
Preparer	J. B. RIV	3/27/82								
Checker	R. D. W.	3/29/82								

BRP: -CC-1-AB-046

NODE: 1359

IDENT: C01-CCATST-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	26.0		522.0		645.0			
$F_y$ (max)	159.0	975.0	749.0	1950.0	786.0	2930.0		
$F_z$	7.0		46.0		56.0			
PR/Shear	27.0	975.0	524.0	1950.0	647.0	2930.0		
$M_x$	152.0		760.0		830.0			
$M_y$ (max)	66.0	2479.0	258.0	4958.0	273.0	7438.0		
$M_z$	106.0		2309.0		2518.0			
MR/Bend	240.0	1240.0	2431.0	2479.0	2657.0	3719.0		

BRP: N/A

NODE: N/A

IDENT: N/A

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
$F_x$	SINCE THERE IS NO COMPLIMENTARY PROBLEM, TO FILL THIS PART OF THE TABLE IS NOT REQUIRED!							
$F_y$								
$F_z$								
PR/Shear								
$M_x$								
$M_y$								
$M_z$								
MR/Bend								

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see 1774.1

(3) For anchors, complimentary problem results must be



AB-1-62E WITH U-BOLT LATERAL STIFFNESS PER  
COMPUTER ANALYSIS JS25 1st May 7, 1984  
OUTPUT TAPE T15176/020

Gibbs & Hill Inc. Job No. 2323 Client TUST-CPSES  
Subject AS-BUILT STRESS VERIFICATION - ANALYSIS RESULTS-ANCHOR/NOZZLE/PENE.  
Calculation Number PROBLEM AB-1-62E Sheet No. \* TEST RUN

Revision	Original	Date	Rev	Date	Rev	Date	Rev	Date	Rev	Date
Design			1							
Prepared			JSA	5/17/84						
Checked			WDE	5/17/84						

BRP/OTHER CC-1-AB-046-NODE: 1359- IDENT: CPI-CCATST-01

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>	15 -23		457. -465		554. -561		BOUNDED BY EMERGENCY	
F <sub>y</sub>	154 67	-975	627. -407	1950.	661. -441	2930.		
F <sub>z</sub>	8 -2		42. -36		49. -48			
PR/ Shear	243	-975	466.8	1950.	563.1	2930.		
M <sub>x</sub>	155 -102		620. -548		675. -623			
M <sub>y</sub>	34 -63	2479.	183. -212	4938.	195. -224	7438.		
M <sub>z</sub>	192 31		141. -132		2145. -1915			
MR/ Bend	246.7	1240.	2056.7	2479.	2248.7	3719.		

BRP/OTHER NODE: IDENT:

	NORMAL		UPSET		EMERGENCY		FAULTED	
	AS-BU.	ALL	AS-BU.	ALLO.	AS-BU.	ALLO.	AS-BU.	ALLO.
F <sub>x</sub>								
F <sub>y</sub>								
F <sub>z</sub>								
PR/ Shear								
M <sub>x</sub>								
M <sub>y</sub>								
M <sub>z</sub>								
MR/ Bend								

Notes: 1) Forces are in lbs, moments in Ft-lbs

2) For allowable loads see

(3) For anchors, complimentary  
problem results must be  
tabulated

Checking Method #

Use the following  
allowable stresses for steel anchors  
and bolts in concrete  
and masonry  
and masonry with appropriate steel and masonry  
and masonry with appropriate steel and masonry

F-186, 4-81

Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPEES  
 Subject As-Built Stress Verification Analysis Results-Maximum Stress  
 Calculation Number Problem AB-1-G2E Sheet No. 22

Revision	Original	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method										
Prepared	W. J. Hill	03/27/88								
Checked	W. J. Hill	3/28/88								

# AB-1-G2E WITHOUT U-BOLT LATERAL STIFFNESS

## MAXIMUM STRESS RESULTS

PER Q.A. BOOK

Plant Condition	Equation	(1) Allowable Stress	Calculated Stress	Node	S.I.F.	Description
Normal and Upset	8	$S_h$				
		15000	6094.0	325	1.0	CRUN
Normal and Upset	9	$1.2 S_h$				
		18000	8822.0	1349	1.9	CRUN
Emergency	9	$1.8 S_h$				
		27000	10090.0	1349	1.9	CRUN
Faulted	9	$2.4 S_h$	FAULTED EQUATION 9 IS BOUNDED BY EMERGENCY EQUATION 9			
Normal and Upset	10 (2)	$S_A$				
		22500	3724.0	1322	2.266	ELBOW
Normal and Upset	11	$S_A + S_h$				
		37500	8026.0	325	1.0	CRUN

(1) Piping Material: SA-106 Gr. B

Design Temperature: 225°F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.

Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPSES

Subject As-built stress Verification Analysis Results-Maximum Stress

Calculation Number Problem AB-1-62E Sheet No. TEST RUN

Revision	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Design									
Preparer									
Checker									

AB-1-62E WITH U-BOLT LATERAL STIFFNESS PER  
COMPUTER ANALYSIS 7525 dtd May 7, 1984  
MAXIMUM STRESS RESULTS

OUTPUT TAPE TIS176/020

Plant Condition	Equation	(1) Allowable Stress	Calculated Stress	Node	S.I.F.	Description
Normal and Upset	8	$S_h$ 15000.0	2413.0	2315 (END)	1.0	CRUN
Normal and Upset	9	$1.2 S_h$ 18000.0	7641.0	1331 (ABG)	2.266	ELBOW
Emergency	9	$1.8 S_T$ 27000.0	9424.0	309 (END)	1.0	CRUN
Faulted	9	$2.4 S_h$ 36000.0	FAULTED CONDITION EQUATION 9 IS BOUNDED BY EMERGENCY CONDITION EQUATION 9			
Normal and Upset	10 (2)	$S_A$ 22500.0	4782.0	1336 (END)	2.266	ELBOW
Normal and Upset	11	$S_A + S_h$ 37500.0	5922.0	1336 (END)	2.266	ELBOW

(1) Piping Material: SA-106 Gr. B

Design Temperature: 225°F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.



AB-2-6313

Table 4

SUPPORT NUMBER		LOADS 1-17-83			LOADS 5-11-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC 2-133-401-A43R	A	3570			3570			
	B	11279			11406			+1
	C	12587			12845			+2
CC 2-158-405-A43R	A		3392			3392		
	B		6512			6696		+3
	C		7165			7466		+4
CC 2-158-410-A43R	A			321			321	
	B			3777			3592	-5
	C			4949			4671	-6
CC 2-158-406-A43R	A		4398			4398		
	B		8846			9670		+9
	C		10090			11310		+11
CC 2-158-411-A43R	A			565			565	
	B			4514			4631	+3
	C			5846			5932	+1
CC 2-158-403-A43R	A		5953			5953		
	B		13028			13234		+2
	C		14267			14560		+2
CC 2-158-409-A43R	A			3022			3022	
	B			6567			6642	+1
	C			7522			7558	+5
CC 2-158-407-A43K	A							
	B		4388			4786		+8
	C		5400			5847		+8
CC 2-158-408-A43K	A							
	B	5508			5654			+3
	C	7044			7307			+4

AB-2-63B

(2)

SUPPORT NUMBER		LOADS 1-17-83			LOADS 5-11-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC2-126-401-A43R	A		2625			2625		
	B		5641			5981		+6
	C		6572			7074		+7
CC2-126-402-A43R	A	152			152			
	B	2844			2847			+0.4
	C	3425			3443			+1.5
CC2-126-403-A43R	A	116	59		116	59		
	B	1914	975		1835	935		-4.4
	C	2520	1284		2392	1219		-5.4
CC2-126-403-A43R	A		1986			1986		
	B		4250			4512		+6
	C		5212			5568		+6
CC2-126-404-A43K	A							
	B			4672			4993	+6
	C			6158			6706	+8
CC2-126-001-F43R	A	1225			1225			
	B	4157			4321			+4
	C	5008			5237			+4
CC2-126-002-F43R	A		2213			2213		
	B		4901			5333		+9
	C		5818			6396		+9
CC2-126-003-F43R	A		1231			1231		
	B		2856			2617		-8
	C		2969			3000		+1
CC2-126-009-F43R	A			239			239	
	B			1518			1579	+4
	C			1826			1902	+4



AB-2-63B

(3)

LOADS 1-17-83

LOADS 5-11-84

SUPPORT NUMBER		FX	FY	FZ	FX	FY	FZ	%CHANGE
* CC2-126-010-F43R	A	624			624			
	B	1998			2545			+21
	C	2353			3180			+26
* CC2-126-011-F43R	A		2242			2242		
	B		3388		61	3553		
	C		3683		73	3957		
CC2-126-004-F43K	A							
	B			2664			2685	+8
	C			3675			3748	+2
CC2-126-007-F43R	A		2173			2173		
	B		3361			4741		+41
	C		3608			5804		+61
* CC2-126-005-F43R	A	615			615			
	B	2904			3434	952		
	C	3757			4542	1484		
* CC2-126-006-F43R	A	46			46			
	B	2235			2707	318		
	C	2521			3130	387		
CC2-126-008-F43R	A		2821			2821		
	B		5702			6085		+7
	C		6538			7082		+8
CC2-164-403-A53S	A		941			941		
	B							
	C							
CC2-164-401-A53R	A	1034			1034			
	B	1974			1991			+8
	C	2441			2455			+6

SUPPORT NUMBER		LOADS 1-17-83			LOADS 5-11-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC2-164-402-A53R	A			836			836	
	B			1704			1763	+3
	C			2148			2208	+3
CC2-164-404-A63K	A							
	B	464			575			+19
	C	643			772			+17
CC2-164-405-A63R	A			413			413	
	B			953			1047	+9
	C			1163			1246	+7
CC2-164-406-A63R	A		6073			6073		
	B		8920			9415		+5
	C		9871			10351		+5
CC2-164-407-A63K	A							
	B	2189			3014			+27
	C	2659			3508			+24
CC2-164-408-A63R	A			177			177	
	B			1064			1170	+9
	C			1444			1561	+7
CC2-164-409-A63K	A		1418			1418		
	B		2082			2239		+7
	C		2267			2420		+6
CC2-164-410-A63R	A			356			356	
	B			1637			1875	+13
	C			2064			2296	+10
CC2-164-411-A63R	A		1545			1545		
	B		2401			2465		+3
	C		2597			2695		+4

AB-2-63B

5

SUPPORT NUMBER		LOADS 1-17-83			LOADS 5-11-84			%CHANGE
		FX	FY	FZ	FX	FY	FZ	
CC2-164-412-A63R	A		924	833		924	833	
	B		1653	1491		1782	1606	+8
	C		1969	1775		2158	1916	+10
CC2-164-414-A63R	A		35	69		35	69	
	B		590	1151		660	1289	+12
	C		823	1635		944	1242	+13
CC2-164-415-A63R	A		1746			1746		
	B		3115			3102		-.4
	C		3718			3693		-.7
CC2-164-416-A63R	A			495			495	
	B			1495			1547	+3
	C			1779			1825	+3
CC2-164-417-A63R	A	5168			5168			
	B	7504			7825			+4
	C	8202			8508			+4
Node Point 1853	A		2284			2284		
	B		3144			3159		+1.5
	C		3460			3483		+1.7
	A							
	B							
	C							
	A							
	B							
	C							
	A							
	B							
	C							

# SEISMIC DISPLACEMENT

SUPPORT NUMBER	$\Delta X$	$\Delta Y$	$\Delta Z$
<del>CC-2-126-010-F43R</del> <del>N.P. 730</del>	<del>+ .113</del> <del>- .113</del>	<del>+ .213</del> <del>- .214</del>	<del>+ .139</del> <del>- .140</del>
CC-2-126-011-F43R N.P. (X) 6	+ .113 - .113	+ .218 - .217	+ .147 - .147
N.P. (Y) 4	+ .123 - .122	+ .212 - .212	+ .139 - .139
CC-2-126-005-F43R N.P. (X) 14	+ .113 - .113	+ .027 - .028	+ .173 - .174
N.P. (Y) 10	+ .118 - .118	+ .212 - .212	+ .166 - .166
CC-2-126-006-F43R N.P. (X) 22	+ .113 - .113	+ .235 - .235	+ .166 - .165
N.P. (Y) 18	+ .117 - .117	+ .212 - .212	+ .172 - .172



Gibbs & Hill, Inc. Job No. 11-2321-030 Client TUST-CPSES  
 Subject As-Built Stress Verification Analysis Results-Maximum Stress  
 Calculation Number Problem AB-2-63B Sheet No. 14

Revision	Date	Rev	Date	Rev	Date	Rev	Date	Rev	Date
1	11/3/83								
Prepared	BKG								
Checked	BKG								

AB-2-63B WITHOUT U-BOLT LATERAL STIFFNESS  
 MAXIMUM STRESS RESULTS

PER Q.A BOOK OF AB-2-63B

Plant Condition	Equation	(1) Allowable Stress	Calculated Stress	Node	S.I.F.	Description
Normal and Upset	8	$S_h$ 15000	4763	710 (BEG)	2.954	TEE
Normal and Upset	9	$1.2 S_h$ 18000	13697	812 (END)	3.07	WELDOLET
Emergency	9	$1.8 S_h$ 27000	15417	812 (END)	3.07	??
Faulted	9	$2.4 S_h$ 36000	BOUNDED BY EMERGENCY CONDITION			
Normal and Upset	10 (2)	$S_A$ 22500	14485	852 (BEG)	2.605	ELBOW
Normal and Upset	11	$S_A + S_h$ 37500	15796	??	??	??

(1) Piping Material: SA-106 GR. B  
 Design Temperature: 225 °F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.

Gibbs & Hill, Inc. Job No. 11-2323-030 Client TUSI-CPSES

Subject As-Built Stress Verification Analysis Results-Maximum Stress

Calculation Number Problem AB-2-63B Sheet No. \*TEST RUN

Revision	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
1	5/17/84								
2	5/17/84								
3									
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100									

AB-2-63B WITH U-BOLT LATERAL STIFFNESS PER COMPUTER ANALYSIS J 746 dtd May 9, 1984 OUTPUT TAPE T15176/022 MAXIMUM STRESS RESULTS

Plant Condition	Equation	Allowable Stress (1)	Calculated Stress	Node	S.I.P.	Description
Normal and Upset	8	$S_h$ 15,000.0	4763.0	710 HAG	2.954	TEE
Normal and Upset	9	$1.2 S_h$ 18,000.0	13,900.0	812 BNO	3.07	WELDOLET
Emergency	9	$1.8 S_h$ 27,000.0	15670.0	812 CND	3.07	WELDOLET
Faulted	9	$2.4 S_h$ 36,000.0	FAULTED CONDITION EQUATION 9 IS BOUNDED BY EMERGENCY CONDITION EQUATION 9			
Normal and Upset	10 (2)	$S_A$ 22500.0	14485.0	852 BAG	2.605	ELBOW
Normal and Upset	11	$S_A + S_h$ 37500.0	15796.0	852 HAG	2.605	ELBOW

(1) Piping Material: SA-106 Gr. B

Design Temperature: 225°F

Note: A separate maximum stress sheet should be used for each piping material within a problem.

(2) Based upon reference 11, equation 10 may be exceeded if equation 11 is satisfied.

# ANCHOR SUPPORTS

## AB-2-63B

SUPPORT NUMBER		LOADS 1-17-83			LOADS 5-11-84			%CHAN
		A	B	C	A	B	C	
<u>CC-2-133-700-A43A</u>	FX	3062	8189	9000	3062	8217	9051	
	FY	14744	19822	20677	14744	20061	21022	
	FZ	1166	5428	6446	1166	5467	6500	
	MX	82092	286248	322596	82092	298380	343476	
	MY	106500	216540	234096	106500	220464	239688	
	MZ	102636	210108	230028	102636	212304	233676	
<u>CC-2-159-700-A43A</u>	FX	2713	6603	10230	2713	6620	10253	
	FY	4177	6963	8986	4177	7201	9214	
	FZ	3184	6060	7733	3184	6135	7820	
	MX	202584	327828	408624	202584	34988	422196	
	MY	283020	414732	531612	283020	415680	532920	
	MZ	46968	181884	307836	46968	183132	309144	
<u>CC-X-022-001-F43A</u>	FX	1330	3803	4252	1330	3760	4197	
	FY	3244	5074	5661	3244	5287	5965	
	FZ	2457	8330	9378	2457	8576	9759	
	MX	64380	391752	444900	64380	409200	471252	
	MY	27456	98100	111492	27456	98304	112680	
	MZ	13416	148560	169008	13416	154836	180780	
<u>CC-X-070-700-A75A</u>	FX	15835	18052	18629	15835	18116	18698	
	FY	5914	8313	8910	5914	8313	8910	
	FZ	166	6816	8634	166	6788	8567	
	MX	22488	324852	394044	22488	325284	394356	
	MY	48540	163260	198492	48540	162744	195852	
	MZ	53436	90912	103296	53436	90972	103332	



Gibbs &amp; Hill, Inc. Job No.

Client

Subject

Calculation Number

Sheet No.

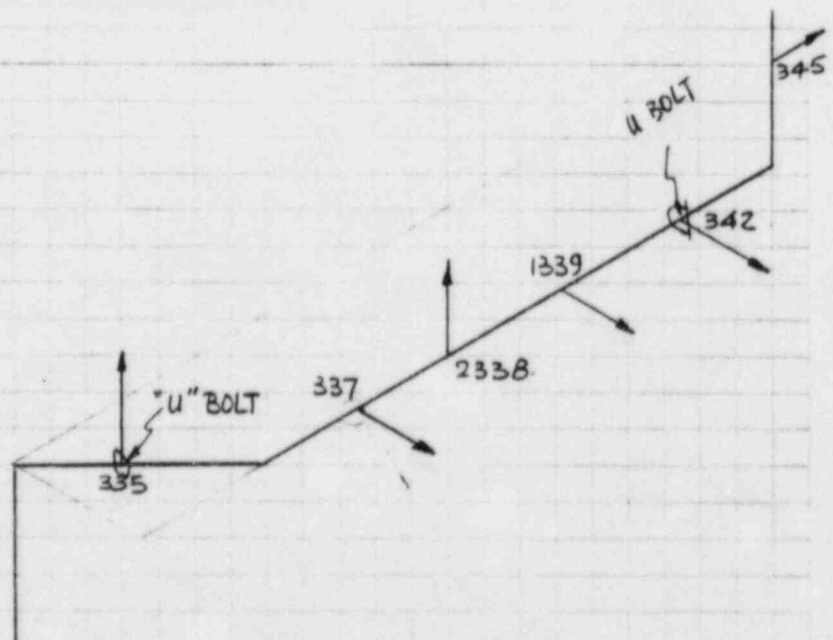
Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #										
Preparer										
Checker										

AB-1-62E

BRH	NODE	w/o U-BOLTS			w. U-BOLTS.		
		FX	FY	FZ	FX	FY	FZ
CC-1-007-035-A63R	335		29/-52		890/-500	37/-65	890/-500
CC-1-007-040-A63R	342	6/-4			20/-35	40/-72	

BRH	NODE	THERMAL DISP w/o U-BOLT		
		DX	DY	DZ
* CC-1-007-035-A63R	335	.001/-0.002		.143/-0.256
* CC-1-007-040-A63R	342		.011/-0.092	

STRESSES w/o		STRESSES w/o	
1322	3724	1336	4782
1303	3698	337	4432
1323	3627	337	4432
1421	3566	1323	4219



Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

F-166, 7-82



Prob. 62E 6"-CC-1-07-152-3

"U" size =  $\frac{5}{8}" \times 6"$  ,  $K_u = 1.31 \times 10^4$  , Generic  $K = 1 \times 10^6$

"U" bolt locations:

Node	Mark No.	Direction	load Max. Pos.		load Max. Neg.		STRESS	COMMENTS
			x	z	x	z		
335	CC-1-007-035-A63R	skew horizontal	890	890	-500	-500	1023	
342	CC-1-007-040-A63R	Y	40		-72		423	

Eq. 10 (thermal stress), 4 highest values:

w/o 'U' bolts

Node	stress
1322	3724 psi
1303	3698
1323	3627
1421	3566

with 'U' bolts

Node	stress
1336	4782 psi
337	4432
337	4432
1323	4219

Anchor loads.

Thermal max. positive

Node No.	F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub>	M <sub>y</sub>	M <sub>z</sub>
510 (with)	0	24	0	0	4	73
510 (w/o)	0	24	0	0	4	72
1359 (with)	14	32	6	164	35	99
1359 (w/o)	15	33	6	159	37	104

Thermal max negative

F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub>	M <sub>y</sub>	M <sub>z</sub>
-130	-0	-167	-228	0	0
-130	-0	-167	-228	0	0
-24	-56	-3	-93	-62	-56
-28	-58	-3	-90	-65	-59

Table 6

**Gibbs & Hill, Inc.**

Job No.

Client

Subject

Calculation Number

Sheet No.

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #										
Preparer										
Checker										

AB-1-62F

BRH	NODE	W/O U-BOLT			W. U-BOLT		
		FX	FY	FZ	FX	FY	FZ
CC-1-011-006-A63R	267		255/-453		267/-477	309/-549	
-029-	260			12/-21		408/-228	17/-30
-026-	250	44/-22			45/-23	3/-	
-024-	245			5/-4		-1/-9	6/-6
-023-	2242			17/-40		-1/-3	16/-40
-021	1239			40/-18		27/-	40/18

BRH	NODE	THERMAL DISP W/O U-BOLT			W/O		W.	
		DX	DY	DZ	NODE	STRESSES	NODE	STRESSES
x CC-1-011-006-A63R	267	.054/-0.03	-		1271	6632	1270	6845
-029-	260		.021/-0.037		1270	6128	1271	6676
-026-	250		-.001		5272	6128	5272	5812
-024-	245		.002/-		5273	5381	5273	5456
-023-	2242		.001/-		267	1664	267	1513
-021-	1239		-.005		260	669	260	1198

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

F-166, 7-82

**Gibbs & Hill, Inc.** Job No. Client  
 Subject  
 Calculation Number Sheet No.

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #										
Preparer										
Checker										

AB-1-65

BRH	NODE	w/o u BOLT			w. u BOLT		
		F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>	F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>
CC-X-013-012-A43R	9850		1/-5			65/-115	140/-80
CC-X-023-015-A43R	77			246/-434		1/0	254/-449
CC-X-023-014-A43R	6			321/-181		1/0	350/-197
CC-X-025-005-A43R	9932	141/-251				114/-202	42/-75

w/o "u" Bolt w. u BOLT.

BRH	Th. DISP w/o u BOLT		
	D <sub>x</sub>	D <sub>y</sub>	D <sub>z</sub>
CC-X-013-012-A43R	.019/-032	-	.024/-043
CC-X-023-015-A43R	.037/-066	-	-
CC-X-023-014-A23R	.011/-019	-	-
CC-X-025-005-A43R	.009/-016	-	.018/-01

NODE	STRESS	NODE	STRESS
914	15812	914	15812
913	13702	913	13703
9917	12465	9917	12541
1414	10352	1414	10352
77	1884	77	1944
6	1191	6	1413

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

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Gibbs &amp; Hill, Inc. Job No. 002323-030 Client TUSI

Subject. STUDY OF THE LOADS, MUTS AND STRESSES

Calculation Number PROB 1-62E Sheet No. 12 OF 6

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #	1									
Preparer	DT	5.10.84								
Checker	CSV	5.10.84								

~~SPG. HGR. SUMMARY~~  $\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
335		2870 3326		0.071 0.088	0.003 0.003	0.055 0.063
342	1014. 1221.			0.001 0.001	0.099 0.113	0.052 0.059
335	291 362	2302 2732	291 362	0.049 0.060	0.002 0.003	0.032 0.033
342	805 991	806 958		0.001 0.001	0.062 0.073	0.038 0.044

w/o U-bolt

w U-bolt

## ANCHOR LOAD REPORT

$\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub> (ft-lb)	M <sub>y</sub>	M <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
510	451. 673	207. 299.	519 767.	1083 1572	823 1215	1031 1544	0.003 0.005	0.017 0.025	0.026 0.038
1359	497. 619.	590. 627.	39. 48	607. 677.	193. 207.	2123. 2332.	-	-	-
510	486 713	208 320	531 781	1095 1589	840 1236	1101 1625	0.003 0.005	0.017 0.025	0.026 0.038
1359	442 539	473 507	34 48	465 521	149 161	1769 1953	-	-	-

w U-bolt

w U-bolt

NODE	σ (N.4PST)	σ (EMG)	NODE	σ (N.4PST)	σ (EMG)
1331	7641	8793	1349	8822	10090
309	7502	9424	1331	8679	9715
1349	7228	8249	1450	8627	9419
1450	7129	7828	1351	7526	7882
335	3330	3755	1324	7336	8784
342	3205	3701	342	4968	5741

Highest stress (psi)

w/o U-bolt

w U-bolt

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes



Gibbs &amp; Hill, Inc. Job No. 002323-030 Client TUSI

Subject STUDY OF THE LOADS, MUTS AND STRESSES

Calculation Number PROB 1-62F Sheet No. 3 OF 6

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #	1									
Preparer	AT	5.10.84								
Checker	CSV	5.10.84								

W/O 'U' BOLT

SPG. HGR. SUMMARY  $\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
267		1853. 1964.		0.066 0.104	0.002 0.002	0.017 0.029
260			399. 523.	0.115 0.124	0.053 0.067	0.0 0.001
250	1010. 1443.			0.001 0.001	0.178 0.237	0.048 0.065
245			1808. 2535.	0.033 0.045	0.112 0.144	0.002 0.003
2242			600. 786.	0.032 0.044	0.036 0.044	0.001 0.001
1239			694. 1068.	0.032 0.045	0.079 0.092	0.001 0.001

ANCHOR LOAD REPORT  $\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub> (FT-LB)	M <sub>y</sub>	M <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
541	1073. 1525.	317. 419.	575. 862.	1179. 1790.	711. 1212.	2322. 3335.	0.002 0.003	0.001 0.002	0.007 0.012
1296	95. 160.	51. 72.	98. 128.	123. 210.	112. 143.	279. 448.	-	-	-

HIGHEST STRESSES:  
(PSI)

NODE	$\sigma$ (N. UPST)	$\sigma$ (EMG)
1250	9063	10423
1265	8982	10624
1275	8716	10198
1251	8488	10033
250	5348	6640
2242	4605	5267

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

Gibbs &amp; Hill, Inc. Job No. 002323-030 Client TUSI

Subject STUDY OF THE LOADS, MUTS AND STRESSES

Calculation Number PROB 1-62F Sheet No. 4 OF 6

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #	1									
Preparer	WT	5.10.84								
Checker	CSU	5.10.84								

WITH 'U' BOLTSPG. HGR. SUMMARY  $\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	$F_x$ (lbs)	$F_y$	$F_z$	$D_x$ (in.)	$D_y$	$D_z$
267	449. 549.	1591. 1949.	- -	0.034 0.05	1.002 0.002	0.012 0.019
260		644. 768.	391. 507.	0.109 0.127	0.049 0.059	- 0.001
250	720. 1162.	1134. 1348.		0.001 0.001	0.087 0.103	0.038 0.053
245		805. 972.	1599. 2196.	0.025 0.033	0.061 0.074	0.002 0.002
2242		472. 579.	596. 769.	0.025 0.033	0.036 0.044	0.001 0.001
1239		767. 922.	649. 976.	0.025 0.033	0.059 0.07	0.001 0.001

ANCHOR LOAD REPORT $\frac{1}{2}$  SSE + SAM  
SSE + FSAM

NODE	$F_x$ (lbs)	$F_y$	$F_z$	$M_x$ (ft-lb)	$M_y$	$M_z$	$D_x$ (in.)	$D_y$	$D_z$
541	1140. 1585.	317. 420.	581. 828.	1191. 1714.	717. 1014.	2387. 3397.	0.002 0.003	0.001 0.002	0.007 0.012
1293	88. 143.	51. 70.	87. 115.	120. 183.	111. 141.	259. 409.	-	-	-

HIGHEST STRESSES:  
(psi)

NODE	$\sigma$ (N.4PST)	$\sigma$ (EMG)
1248	9097	10639.
1265	8529	9840
1275	8503	9937
5284	8083	10608
250	5293	6332
1239	4196	5333.

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

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Gibbs &amp; Hill, Inc. Job No. 002323-030 Client TUSI

Subject STUDY OF THE LOADS, MUTS AND STRESSES

Calculation Number PROB 1-65

Sheet No. 5 OF 6

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #										
Preparer	RV	5.10.84								
Checker	CSV	5.10.84								

W/0 'U' BOLTSPG. HGR. SUMMARY  $\frac{1}{2}$  SSE + ~~SAM~~  
SSE + ~~SAM~~

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
9850		368. 436.		0.064 0.077	0.0 0.0	0.108 0.128
77			401. 473.	0.01 0.014	0.0 0.012	0.0 0.0
6			310. 374.	0.01 0.012	0.007 0.008	0.0 0.0
9932		112. 140.		0.027 0.034	0.0 0.0	0.021 0.026

ANCHOR LOAD REPORT

 $\frac{1}{2}$  SSE + ~~SAM~~  
SSE + ~~SAM~~

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub> (ft-lb)	M <sub>y</sub>	M <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
708	858. 1039.	851. 1030.	669. 806.	617. 746.	3382. 4067.	1873. 2267.	-	-	-
937	1525. 1846.	97. 117.	431. 520.	1123. 1345.	2638. 4379.	379. 459.	-	-	-
9804	12. 15.	11. 14.	31. 39.	99. 125.	5. 6.	46. 57.	-	-	-
9817	38. 48.	159. 204.	32. 41.	184. 237.	19. 24.	128. 166.	-	-	-
9864	38. 115.	503. 358.	406. 482.	531. 628.	588. 694.	340. 406.	-	-	-
9873	201. 255.	53. 76.	62. 79.	210. 268.	39. 47.	84. 107.	-	-	-
9945	91. 111.	40. 50.	89. 112.	91. 119.	91. 123.	108. 182.	-	-	-

HIGHEST STRESSES :  
(PSI)

NODE	$\sigma$ (N.40ST)	$\sigma$ (EMG)
989	14698	17582
988	13620	16276
986	12748	15228
990	11644	13874
985	11622	13878
987	11503	13719
77	3402	3829

Checking Method #

1. Line-by-line checking  
 2. Alternative Calculation Results compared  
 3. Identical Calculation Results compared



Gibbs &amp; Hill, Inc. Job No. 002323-030 Client TUSI

Subject STUDY OF THE LOADS, MUTS AND STRESSES

Calculation Number PROB 1-65 Sheet No. 6 OF 6

Revision	Original Issue	Date	Rev.	Date	Rev.	Date	Rev.	Date	Rev.	Date
Checking Method #	1									
Preparer	PJ	5.10.84								
Checker	CSV	5.10.84								

WITH 'L' BOLTSPG. HGR. SUMMARY  $\frac{1}{2}$  SSE ~~FSAM~~  
SSE ~~FSAM~~

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	D <sub>x</sub> (in.)	D <sub>y</sub>	D <sub>z</sub>
9850		52. 70	199. 271	0.042 .057	0.0 0.0	0.015 .021
77		88. 107	400. 473	0.011 0.014	0.007 0.008	0.0 0.0
6		72. 91	288. 346	0.009 0.011	0.005 0.007	0.0 0.0
9932		84. 112	144. 189	0.02 0.026	0.0 0.0	0.011 0.014

ANCHOR LOAD REPORT $\frac{1}{2}$  SSE ~~FSAM~~  
SSE ~~FSAM~~

NODE	F <sub>x</sub> (lbs)	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub> (ft.-lb)	M <sub>y</sub>	M <sub>z</sub>	D <sub>x</sub> (in)	D <sub>y</sub>	D <sub>z</sub>
708	858 1039	851 1030	668 807	619. 753	3376. 4075	1873. 2267	-	-	-
937	1528. 1852	97. 117	432 522	1127. 1357	3641 4396	379. 459	-	-	-
9804	12. 15	11. 14	31. 39	97. 122	4. 6	45 56	-	-	-
9817	37 47	159. 204	31. 41	184 237	18. 24	128. 165	-	-	-
9864	94. 128	41. 54	81 111	120 163	185 252	193. 261	-	-	-
9873	148 212	57. 72	69 92	235. 314	11 15	62. 89	-	-	-
9945	57. 79	11 15	76 101	88 117	119. 157	82. 113	-	-	-

HIGHEST STRESSES:  
(PSI)

NODE	$\sigma$ (N.40ST)	$\sigma$ (EMG)
989	14686.	17566.
988	13607.	16259
986	12724.	15196.
990	11635	13861
985	11596.	13842
987	11488.	13679.
77	3416.	3843.

Checking Method #

1. Line-by-line checking
2. Alternative Calculation Results compared
3. Identical Calculation Results compared
4. Compare inputs and results of computer with corresponding inputs and results of similar codes.

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TESTING OF FIGURE 137N U-BOLTS  
FOR THE  
COMANCHE PEAK STEAM ELECTRIC STATION  
TEXAS UTILITIES SERVICES INCORPORATED  
GLEN ROSE, TEXAS

ITT GRINNELL CORPORATION  
PIPE HANGER DIVISION  
PIPING/STRUCTURAL ANALYSIS SECTION  
PROVIDENCE, RHODE ISLAND

APRIL 16, 1984

Product Engineering Report No. 562

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Side Loading : 4" pipe	
24" pipe	
Normal Loading: 4" pipe	
24" pipe	
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## TEST OBJECTIVE

The purpose of the present tests is the evaluation of the behavior of the ITT Grinnell Figure 137N or equivalent U-bolts used on the Comanche Peak Project, under different loading conditions anticipated during the life of the support.

These loading conditions are summarized below:

### Loading Condition 1 (See Figure 1)

The U-bolt is subjected to an initial normal load  $P_n$  per ITT Grinnell or NPSI Load Capacity Data Sheets (LCDS) and to a simultaneous uniformly increasing axial load  $P_a$ . The axial load will be increased in a controlled manner and the deflection of the U-bolt in the axial direction and change in normal load will be recorded. Failure of the U-bolt to perform its function and sustain the applied loads in either direction, or the presence of excessive displacements will result in the termination of the test.

### Loading Condition 2 (See Figure 2)

The U-bolt is subjected to an initial normal load  $P_n$  per ITT Grinnell or NPSI Load Capacity Data Sheets (LCDS) and to a simultaneous uniformly increasing side or lateral load  $P_s$ . The side load will be increased in a controlled manner, and the deflection of the U-bolt in the lateral direction and change in normal load will be recorded. Failure of the U-bolt to perform its function and sustain the applied loads in either direction, or the presence of excessive displacements will result in the termination of the test.

### Loading Condition 3 (See Figure 3)

The U-bolt is subjected to a normal load  $P_n$ . This normal load will be increased in a controlled manner until failure of the U-bolt occurs in the tensile mode. Elongation of the U-bolt will be recorded.

## TEST SET-UP

The RHIELE testing machine located in the Research, Development and Engineering Department of ITT Grinnell in Providence, R.I. will be used for testing the Figure 137N 4" and 24" pipe U-bolts.

The test fixture consists of the lower test stand shown in sketch No. SA-4995-1 (see Appendix III) to support the U-bolt and pipe section. The lower test stand has two different designs, one for the 4" and the other for the 24" pipe U-bolt.

Sketch No. SA-4995-3 (see Appendix III) shows the upper test fixture which consists of a length of structural tubing attached to the upper fixed end of the machine.

The structural tubing supports the pipe and U-bolt under testing, through rods, eye-nuts, clevises and hex nuts (see Sketch No. SA-4995-4 & 5 of Appendix III). This upper test fixture is the same for both the 4" and 24" pipe U-bolts.

In order to generate axial and side loads at the centerline of the pipe, a lateral load base support is used shown in sketch No. SA-4995-2 (see Appendix III). This base support is used for holding the cylinder end of a hydraulic pump that in turn pushes axially or laterally on the pipe through load-inducing columns or rods. The location of this lateral load base support is shown in sketch No. SA-4995-4 & 5 (see Appendix III). It is secured on to the RHIELE testing machine.

In order to allow axial and lateral movements of the pipe, the introduction of pinned-joints was necessary in order to allow rotation (see Detail "A" on Sketch SA-4995-4, Appendix III).

In order to obtain conservative results, the pipe spools used were very stiff in comparison to actual pipes, in the case of the 24" being heavily stiffened with reinforcing circular plates. In utilizing this configuration, minimal energy is dissipated in deforming the pipe spool pieces, the major expansion of energy taking place in the U-bolt deformation.



LOADING CONDITION 1

LCD NORMAL LOAD  $P_N$  AND AXIAL LOAD  $P_A$

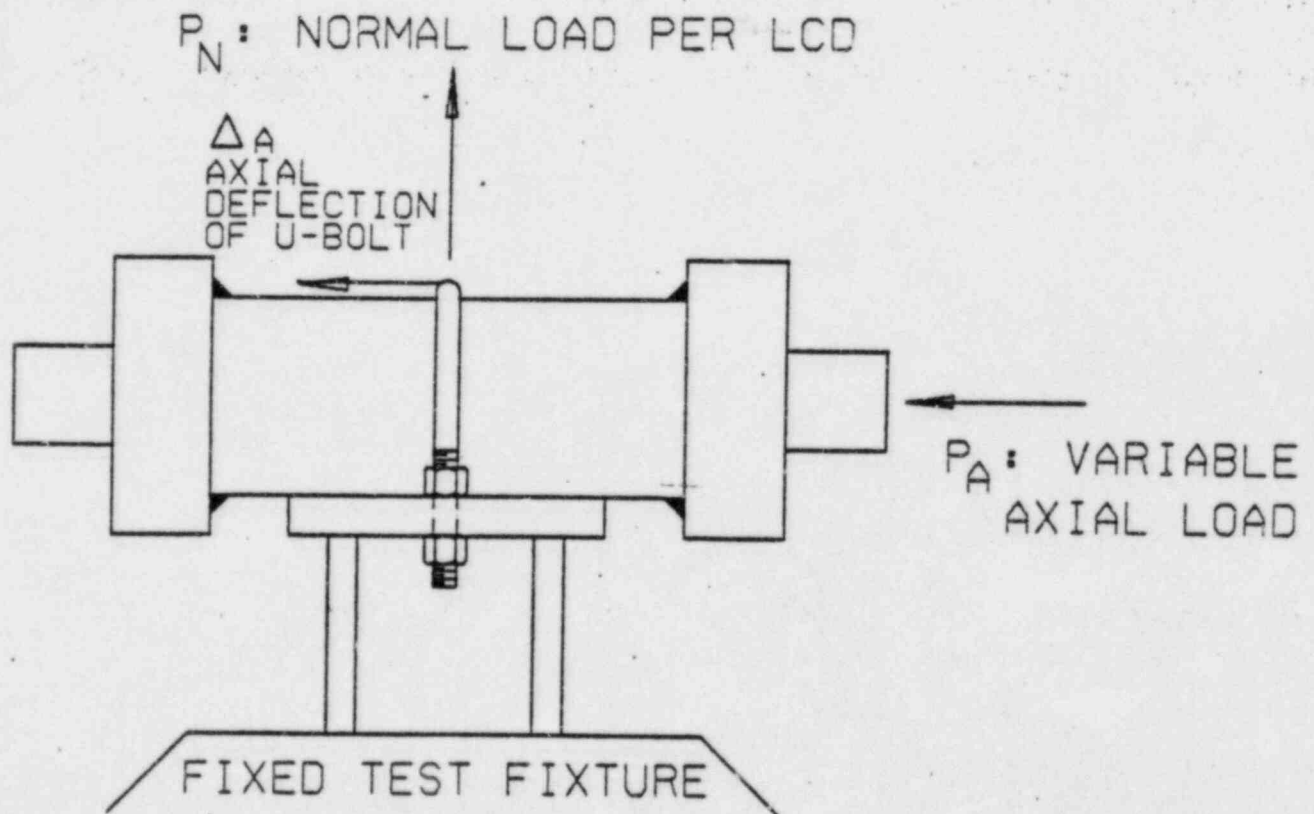


FIGURE 1

LOADING CONDITION 2

LCD NORMAL LOAD  $P_N$  AND SIDE LOAD  $P_S$

$P_N$ : NORMAL LOAD PER LCDS

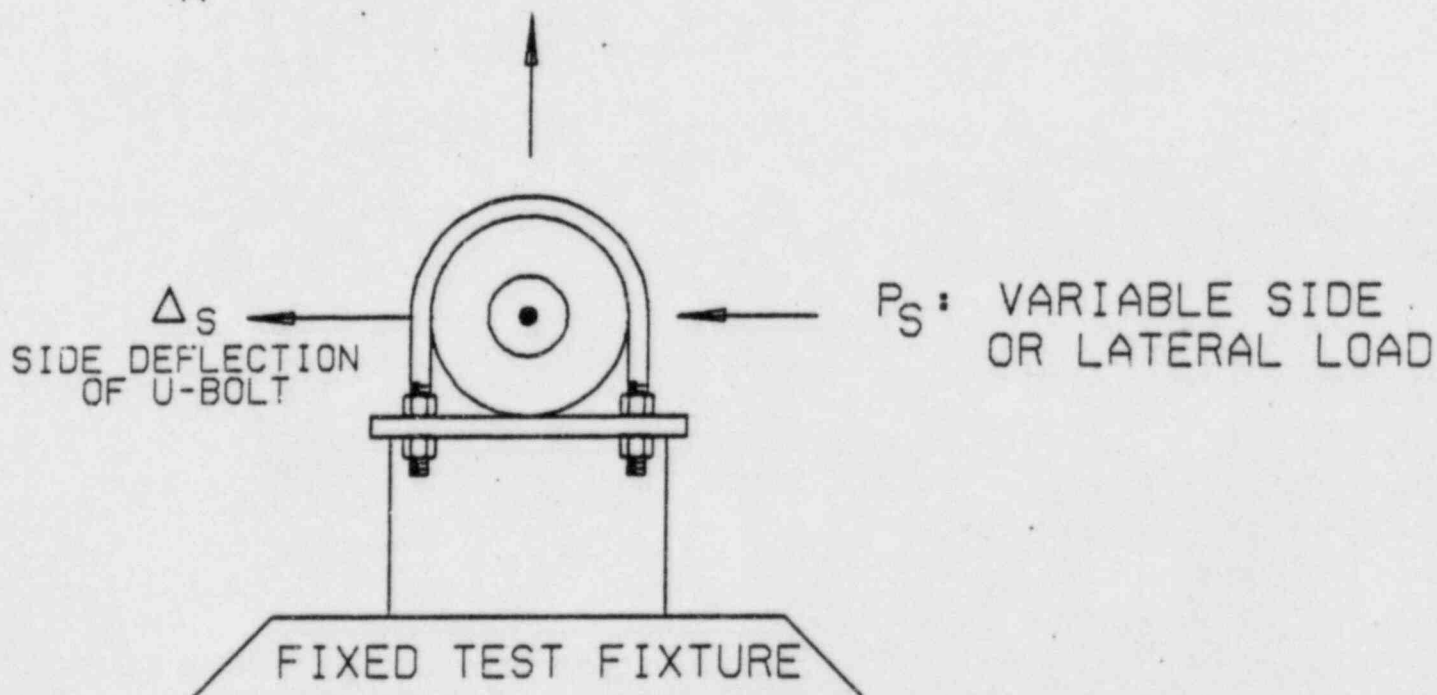


FIGURE 2

LOADING CONDITION 3

NORMAL LOAD  $P_N$  TO U-BOLT FAILURE

$P_N$  : NORMAL LOAD TO FAILURE

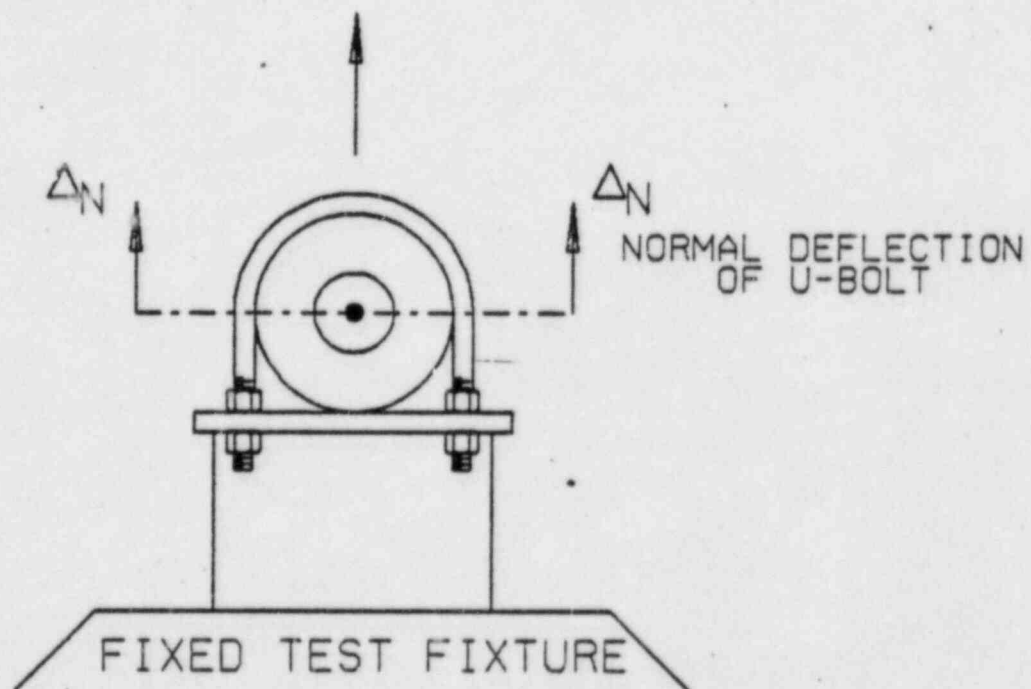


FIGURE 3

## APPLICATION OF LOADS

### Normal Loads

Normal loads will be applied through the testing machine. For the 4" pipe U-bolt this normal load will be 2260 lb and for the 24" pipe U-bolt the normal load will be 9920 lb (per ITTG DRS 137 Rev. 3; NPSI LCD PUS-040,240 Revision 1).

### Axial Loads

Axial loads are the loads acting along the axis parallel to the length of the pipe. These loads will be applied through the "ENERPAC S177" hydraulic pump and cylinder set. The cylinder will be suitably installed on the testing machine and its height adjusted to act along the center line of the pipe.

### Side or Lateral Loads

Side or lateral loads are the loads acting perpendicular to the axis of the pipe in the plane of the U-bolt. The "ENERPAC S177" hydraulic pump and cylinder set will be used to generate these loads. The cylinder will be installed in the same manner as for axial load and its height adjusted to act perpendicular to the center line of the pipe.

## MEASUREMENTS

All normal loads ( $P_n$ ) are measured on the Rhiele testing machine dial.

Side and axial loads are recorded through strain gages. These gages are installed on the load-inducing rods and connected to a strain measuring device (see appendix IV). At any time during testing and for any U-bolt axial or lateral deflection the strains may be recorded and printed out.

Strains obtained are then averaged and converted into pounds force (see Appendix V).

## TESTING PROCEDURE

Three (3) tests will be performed for each of the loading conditions 1, 2 and 3 for each of two U-bolt sizes. The sizes involved are:

- (a) 4" pipe, 1/2" rod size U-bolt Figure 137N, LCDS rated at 2260 lb normal load  $P_n$ , and 184 lb side load  $P_s$ .
- (b) 24" pipe, 1" rod size U-bolt Figure 137N, LCDS rated at 9920 lb normal load  $P_n$ . No LCDS rating for side loading exists as of this time.



4" Pipe, 1/2" Rod U-Bolt Testing

Axial load plus rated normal load (Condition 1)

Test A  
Test B  
Test C

Side load plus rated normal load (Condition 2)

Test D  
Test E  
Test F

Normal load to failure (Condition 3)

Test G  
Test H  
Test I

## LOADING CONDITION 1

### Rated Normal Load Plus Variable Axial Load

U-bolt size : 4" pipe  
1/2" rod.

Three tests A,B and C were performed. In each test, the U-bolt was loaded with the LCDS rated normal load of 2260 lb. Axial load was induced through the hydraulic pump. The magnitude of the axial load was determined through strain gages placed on the load-inducing columns and connected to the cylinder end of the pump.

As the hydraulic pump was activated, strains and deflections were recorded as well as any change in the original normal load.

### Test 'A'

Test 'A' was performed as two separate parts. The test started using a pressure gage on the hydraulic pump. Halfway along the test it became evident that the pressure gage could not accommodate the axial load needed to produce significant axial deflections.

It was then decided that axial loads would be determined using strain gages mounted on the load-inducing columns. The gages were properly placed and the test continued.

TEST 'A'

Axial Deflection (IN)	Gage Pressure (PSIG)	Axial Load (LB)	Normal Load (LB)
.0	13.	0.	2260.
.002	46.	37.	2260.
.005	57.	61.	2260.
.010	67.	83.	2260.
.021	77.	104.	2260.
.027	85.	119.	2260.
.035	95.	139.	2260.
.044	102.	154.	2260.
.051	112.	175.	2260.
.058	120.	190.	2260.
.063	126.	202.	2260.
.069	135.	221.	2260.
.075	140.	232.	2260.
.080	146.	245.	2260.
.085	152.	258.	2260.
.094	162.	277.	2260.
.122	192.	338.	2270.
.151	212.	380.	2305.
.176	227.	413.	2335.
.203	240.	440.	2375.
.251	260.	479.	2490.
.312	290.	542.	2650.

TEST 'A' (CONTINUED)

Axial Deflection (IN)	Axial Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.312	53	-119	-33.0	554.	2825.
.325	53	-123	-35.0	587.	2860.
.340	53	-125	-36.0	604.	2920.
.355	54	-129	-37.5	629.	2990.
.400	55	-137	-41.0	688.	3180.
.450	58	-149	-45.5	763.	3410.
.500	61	-162	-50.5	847.	3620.
.550	65	-175	-55.0	923.	3830.
.600	66	-184	-59.0	990.	4030.
.650	68	-202	-67.0	1124.	4510.
.700	68	-218	-75.0	1259.	4990.
.750	71	-240	-84.5	1418.	5480.



TEST 'B'

Axial Deflection (IN)	Axial Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0.	2260.
.015	1	-10	-4.5	75.	2260.
.030	3	-16	-6.5	109.	2250.
.045	6	-23	-8.5	143.	2245.
.060	10	-30	-10.0	168.	2245.
.075	11	-36	-12.5	210.	2245.
.090	12	-41	-14.5	243	2245.
.110	14	-48	-17.0	285.	2260.
.125	15	-53	-19.0	319.	2275.
.150	16	-60	-22.0	369.	2320.
.175	16	-66	-25.0	420.	2365.
.200	14	-68	-27.0	453.	2415.
.225	14	-69	-27.5	461.	2475.
.250	11	-70	-29.5	495.	2540.
.275	10	-71	-30.5	512.	2600.
.300	9	-74	-32.5	545.	2685.
.325	8	-77	-34.5	579.	2770.
.350	9	-81	-36.0	604.	2860.
.375	7	-84	-38.5	646.	2965.
.400	7	-88	-40.5	680.	3080.
.425	6	-92	-43.0	722.	3200.
.450	6	-97	-45.5	763.	3330.
.475	7	-103	-48.0	805.	3460.
.500	8	-109	-50.5	847.	3600.
.550	11	-123	-56.0	940.	3850.
.600	12	-136	-62.0	1040.	4170.
.650	6	-151	-72.5	1217.	4770.
.700	2	-168	-83.0	1393.	5400.
.750	-1	-184	-92.5	1552.	5860.
.800	-6	-198	-102.0	1712.	6350.
.850	-13	-227	-120.0	2014.	6800.

TEST 'C'

Axial Deflection (IN)	Axial Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	2280.
.015	2	-12	-5.0	84.	2280.
.030	3	-17	-7.0	117.	2280.
.045	3	-20	-8.5	143.	2290.
.060	3	-22	-9.5	159.	2310.
.075	2	-24	-11.0	185.	2330.
.090	2	-26	-12.0	201.	2345.
.110	1	-25	-12.0	201.	2365.
.125	4	-32	-14.0	235.	2400.
.150	4	-35	-15.5	260.	2440.
.175	5	-40	-17.5	294.	2495.
.200	6	-44	-19.0	319.	2545.
.225	7	-49	-21.0	352.	2585.
.250	9	-54	-22.5	378.	2630.
.275	10	-59	-24.5	411.	2685.
.300	11	-64	-26.5	445.	2750.
.325	12	-69	-28.5	478.	2805.
.350	13	-74	-30.5	512.	2865.
.375	15	-82	-33.5	562.	2930.
.400	17	-88	-35.5	595.	3015.
.425	21	-97	-38.0	638.	3090.
.450	23	-103	-40.0	671.	3175.
.475	25	-109	-42.0	705.	3275.
.500	24	-114	-45.0	755.	3375.
.550	24	-124	-50.0	839.	3620.
.600	25	-134	-54.5	915.	3835.
.650	25	-147	-61.0	1024.	4190.
.700	23	-163	-70.0	1175.	4640.
.750	22	-182	-80.0	1342.	5250.
.800	26	-210	-92.0	1544.	5860.
.850	33	-263	-115.0	1930.	6275.

LOADING CONDITION 2

Rated Normal Load Plus Variable Side Load

U-bolt size : 4" pipe  
              1/2" rod

Three tests D, E and F were performed. In each test the U-bolt was loaded with the LCDS rated normal load of 2260. lb. Side load was induced through the hydraulic pump. As in the case of the axial load, the magnitude of the side load was determined through strain gages placed on the load-inducing columns. The load inducing columns are inserted in the cylinder end of the pump and bear upon the pipe.

Deflections and strains were recorded as the hydraulic pump was activated. Also recorded was any change in the normal load.

TEST 'D'

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0.	2270.
.015	4	-15	-5.5	92.	2225.
.030	9	-25	-8.0	134.	2180.
.045	13	-36	-11.5	193.	2125.
.060	22	-54	-16.0	268.	2125.
.075	30	-69	-19.5	327.	2125.
.090	37	-92	-27.5	461.	2160.
.110	42	-119	-38.5	646.	2185.
.125	42	-136	-47.0	789.	2200.
.150	34	-157	-61.5	1032.	2245.
.175	27	-173	-73.0	1225.	2300.
.200	21	-182	-80.5	1351.	2360.
.225	15	-190	-87.5	1468.	2436.
.250	10	-196	-93.0	1561.	2500.
.275	5	-201	-98.0	1644.	2585.
.300	0	-207	-103.5	1737.	2675.
.325	-8	-211	-109.5	1837.	2775.
.350	-16	-213	-114.5	1921.	2880.
.375	-24	-216	-120.0	2014.	2990.
.400	-33	-219	-126.0	2114.	3100.
.425	-42	-224	-133.0	2232.	3225.
.450	-48	-232	-140.0	2349.	3310.
.475	-52	-242	-147.0	2467.	3425.
.500	-61	-248	-154.5	2593.	3550.
.550	-81	-259	-170.0	2853.	3835.
.600	-101	-267	-184.0	3088.	4110.
.650	-129	-276	-202.5	3398.	4425.
.700	-156	-284	-220.0	3692.	4675.
.750	-183	-295	-239.0	4010.	4875.
.800	-221	-299	-260.0	4363.	5325.
.850	-261	-302	-281.5	4724.	5880.
.900	-311	-307	-309.0	5185.	6375.



TEST 'E'

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	2260.
.015	18	-35	-8.5	143.	2145.
.030	37	-68	-15.5	260.	2070.
.045	51	-100	-24.5	411.	2030.
.060	56	-122	-33.0	554.	2020.
.075	62	-144	-41.0	688.	2010.
.090	67	-167	-50.0	839.	2010.
.110	73	-194	-60.5	1015.	2075.
.125	76	-214	-69.0	1158.	2140.
.150	76	-236	-80.0	1342.	2235.
.175	75	-252	-88.5	1485.	2330.
.200	73	-263	-95.0	1594.	2430.
.225	71	-272	100.5	1686.	2540.
.250	68	-278	-105.0	1762.	2630.
.275	64	-283	109.5	1837.	2720.
.300	57	-288	-115.5	1938.	2830.
.325	50	-291	-120.5	2022.	2930.
.350	42	-294	-126.0	2114.	3020.
.375	34	-298	-132.0	2215.	3110.
.400	24	-302	-139.0	2332.	3220.
.425	17	-305	-144.0	2416.	3280.
.450	5	-307	-151.0	2534.	3420.
.475	-1	-310	-155.5	2609.	3510.
.500	-5	-320	-162.5	2727.	3590.
.550	-21	-331	-176.0	2953.	3810.
.600	-44	-336	-190.0	3188.	4030.
.650	-69	-341	-205.0	3440.	4270.
.700	-100	-346	-223.0	3742.	4540.
.750	-133	-349	-241.0	4044.	4790.
.800	-174	-346	-260.0	4363.	5190.
.850	-214	-348	-281.0	4715.	5590.

TEST 'F'

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	2260.
.015	-3	-10	-6.5	109.	2210.
.030	0	-19	-9.5	159.	2160.
.045	0	-28	-14.0	235.	2120.
.060	1	-37	-18.0	302.	2120.
.075	0	-45	-22.5	378.	2130.
.090	-2	-53	-27.5	461.	2150.
.110	-6	-64	-35.0	587.	2200.
.125	-11	-73	-42.0	705.	2240.
.150	-19	-87	-53.0	889.	2300.
.175	-29	-98	-63.5	1066.	2380.
.200	-38	-104	-71.0	1191.	2440.
.225	-48	-108	-78.0	1309.	2480.
.250	-54	-111	-82.5	1384	2500.
.275	-59	-116	-87.5	1468	2550.
.300	-65	-120	-92.5	1552.	2610.
.325	-70	-123	-96.5	1619.	2680.
.350	-75	-127	-101.0	1695.	2750.
.375	-81	-131	-106.0	1779.	2830.
.400	-87	-136	-111.5	1871.	2940.
.425	-96	-139	-117.5	1972.	3050.
.450	-105	-141	-123.0	2064.	3120.
.475	-116	-142	-129.0	2165.	3250.
.500	-127	-145	-136.0	2282.	3290.
.550	-150	-150	-150.0	2517.	3550.
.600	-171	-155	-163.0	2735.	3790.
.650	-196	-162	-179.0	3004.	4080.
.700	-221	-171	-196.0	3289.	4340.
.750	-242	-178	-210.0	3524.	4500.
.800	-269	-187	-228.0	3826.	4800.
.850	-299	-195	-247.0	4145.	5230.
.900	-332	-204	-268.0	4497.	5640.
.950	-366	-213	-289.5	4858.	6100.

### LOADING CONDITION 3

#### Normal Load to Failure

U-Bolt size : 4" pipe  
1/2" rod

Three tests G,H and I were performed. In each test the U-bolt was loaded normally by means of the tensile machine. The normal load was increased gradually until the U-bolt failed. The charting capability of the machine was utilized to plot the normal load against the normal deflection or elongation of the U-bolt. The location of failure for each U-bolt was observed.

24" Pipe, 1" Rod U-Bolt Testing

Axial load plus rated normal load (Condition 1)

Test K  
Test L  
Test M

Side load plus rated normal load (Condition 2)

Test N  
Test O  
Test P

Normal load to failure (Condition 3)

Test Q  
Test R  
Test S



LOADING CONDITION 1

Rated Normal Load Plus Variable Axial Load

U-bolt size : 24" pipe  
1" rod

Three tests K,L and M were performed. In each test the U-bolt was loaded with the LCDS rated normal load of 9920 lb. Axial load was induced through the hydraulic pump.

The testing procedure is the same as for tests A,B and C.

TEST 'K'

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	9920.
.025	-18	-5	-11.5	193	9850.
.050	-31	-8	-19.5	327	9750.
.075	-52	-1	-26.5	445	9600.
.100	-63	-2	-32.5	545	9500.
.125	-70	-5	-37.5	629	9400.
.150	-76	-8	-42.0	705	9350.
.175	-84	-10	-47.0	789	9350.
.200	-117	16	-50.5	847	9300.
.225	-127	14	-56.5	948	9300.
.250	-135	15	-60.0	1007	9300.
.275	-144	17	-63.5	1066	9300.
.300	-151	18	-66.5	1116	9300.
.325	-160	18	-71.0	1191	9400.
.350	-168	20	-74.0	1242	9500.
.375	-177	22	-77.5	1300	9500.
.400	-185	23	-81.0	1359	9550.
.425	-194	23	-85.5	1435	9700.
.450	-201	22	-89.5	1502	9800.
.475	-207	21	-93.0	1561	9900.
.500	-211	19	-96.0	1611	9950.
.525	-217	19	-99.0	1661	10050.
.550	-222	18	-102.0	1712	10200.
.575	-227	16	-105.5	1770	10300.
.600	-232	13	-109.5	1837	10400.
.625	-233	11	-111.0	1863	10550.
.650	-237	11	-113.0	1896	10700.
.675	-241	9	-116.0	1946	10800.
.700	-245	7	-119.0	1997	11000.
.725	-251	8	-121.5	2039	11150.
.750	-256	8	-124.0	2081	11300.

TEST 'K' (CONTINUED)

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-263	10	-126.5	2122	11500
.800	-266	8	-129.0	2165	11700
.825	-271	8	-131.5	2207	11900
.850	-278	7	-135.5	2274	12100
.875	-282	8	-137.0	2299	12350
.900	-288	7	-140.5	2358	12550
.925	-296	8	-144.0	2416	12800
.950	-301	7	-147.0	2467	13050
.975	-308	6	-151.0	2534	13350
1.000	-314	6	-154.0	2584	13600
1.025	-320	5	-157.5	2643	13850
1.050	-326	4	-161.0	2702	14150
1.075	-331	2	-164.5	2760	14450
1.100	-336	0	-168.0	2819	14700
1.125	-342	-1	-171.5	2878	15000
1.150	-345	-4	-174.5	2928	15250
1.175	-351	-7	-179.0	3004	15600
1.200	-354	-10	-182.0	3054	15950
1.225	-358	-14	-186.0	3121	16250
1.250	-364	-19	-191.5	3213	16600
1.275	-368	-22	-195.0	3272	16850
1.300	-374	-26	-200.0	3356	17200
1.325	-377	-30	-203.5	3415	17650
1.350	-383	-35	-209.0	3507	18000
1.375	-388	-39	-213.5	3583	18400
1.400	-393	-44	-218.5	3666	18700
1.425	-395	-50	-222.5	3734	19050
1.450	-400	-56	-228.0	3826	19500
1.475	-404	-64	-234.0	3927	19850
1.500	-405	-72	-238.5	4002	20200

TEST 'L'

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	9950
.025	-19	-3	-11.0	185	9900
.050	-25	-7	-16.0	268	9850
.075	-29	-10	-19.5	327	9850
.100	-32	-14	-23.0	386	9850
.125	-33	-20	-26.5	445	9800
.150	-38	-23	-30.5	512	9800
.175	-42	-27	-34.5	579	9800
.200	-47	-31	-39.0	654	9800
.225	-51	-35	-43.0	722	9850
.250	-56	-38	-47.0	789	9900
.275	-59	-43	-51.0	856	9950
.300	-64	-47	-55.5	931	10000
.325	-67	-52	-59.5	998	10050
.350	-70	-57	-63.5	1066	10150
.375	-72	-63	-67.5	1133	10250
.400	-74	-69	-71.5	1200	10350
.425	-75	-76	-75.5	1267	10450
.450	-78	-81	-79.5	1334	10500
.475	-81	-33	-57.0	956	10600
.500	-130	-44	-87.0	1460	10700
.525	-132	-51	-91.5	1535	10800
.550	-135	-57	-96.0	1611	10850
.575	-136	-63	-99.5	1670	10950
.600	-138	-68	-103.0	1728	11000
.625	-140	-74	-107.0	1795	11100
.650	-141	-80	-110.5	1854	11200
.675	-141	-87	-114.0	1913	11300
.700	-141	-92	-116.5	1955	11400
.725	-141	-100	-120.5	2022	11500
.750	-141	-107	-124.5	2081	11650



TEST 'L' (CONTINUED)

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-141	-114	-127.5	2139	11800
.800	-142	-119	-130.5	2190	12000
.825	-142	-127	-134.5	2257	12150
.850	-142	-136	-139.0	2332	12400
.875	-142	-143	-142.5	2391	12650
.900	-142	-152	-147.0	2467	12900
.925	-140	-162	-151.0	2534	13100
.950	-140	-170	-155.0	2601	13400
.975	-140	-179	-159.5	2676	13700
1.000	-139	-186	-162.5	2727	13950
1.025	-138	-198	-168.0	2819	14250
1.050	-136	-209	-172.5	2895	14600
1.075	-135	-220	-177.5	2978	14900
1.100	-134	-230	-182.0	3054	15200
1.125	-131	-243	-187.0	3138	15500
1.150	-132	-250	-191.0	3205	15900
1.175	-129	-264	-196.5	3297	16250
1.200	-126	-277	-201.5	3381	16650
1.225	-125	-287	-206.0	3457	16950
1.250	-141	-303	-222.0	3725	17300

TEST 'M'

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	9950
.025	-30	2	-14.0	235	9950
.050	-44	4	-20.0	336	9950
.075	-42	0	-21.0	352	9700
.100	-49	-5	-27.0	453	9800
.125	-54	-11	-32.5	545	9850
.150	-58	-16	-37.0	621	9850
.175	-62	-21	-41.5	696	9900
.200	-66	-25	-45.5	763	9950
.225	-95	-4	-49.5	831	9950
.250	-94	-16	-55.0	923	10000
.275	-94	-25	-59.5	998	10050
.300	-97	-33	-65.0	1091	10150
.325	-99	-41	-70.0	1171	10200
.350	-106	-44	-75.0	1259	10250
.375	-115	-37	-76.0	1275	10350
.400	-118	-48	-83.0	1393	10450
.425	-130	-38	-84.0	1410	10550
.450	-132	-50	-91.0	1527	10700
.475	-141	-42	-91.5	1535	10750
.500	-143	-55	-99.0	1661	10900
.525	-154	-46	-100.0	1678	10900
.550	-157	-51	-104.0	1745	11100
.575	-161	-55	-108.0	1812	11200
.600	No readings available due to U-bolt "slip"				
.625	"	"	"	"	"
.650	-186	30	-78.0	1309	11100
.675	-211	32	-89.5	1502	11450
.700	-227	34	-96.5	1619	11700
.725	-238	32	-103.0	1728	12000
.750	-240	23	-108.5	1821	12250

TEST 'M' (CONTINUED)

Axial Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Axial Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-241	15	-113.0	1896	12500
.800	-244	7	-118.5	1988	12700
.825	-246	1	-122.5	2056	12950
.850	-247	-5	-126.0	2114	13200
.875	-249	-12	-130.5	2190	13350
.900	-252	-17	-134.5	2257	13600
.925	-250	-26	-138.0	2316	13850
.950	-254	-33	-143.5	2408	14100
.975	-256	-39	-147.5	2475	14350
1.000	-258	-44	-151.0	2534	14550
1.025	-259	-52	-155.5	2609	14800
1.050	-261	-58	-159.5	2676	15100
1.075	-263	-64	-163.5	2744	15300
1.100	-268	-66	-167.0	2802	15550
1.125	-268	-72	-170.0	2853	15800
1.150	-273	-77	-175.0	2937	16050
1.175	-275	-84	-179.5	3012	16350
1.200	-279	-90	-184.5	3096	16650
1.225	-281	-96	-188.5	3163	16950
1.250	-286	-104	-195.0	3272	17250
1.275	-286	-112	-199.0	3339	17550
1.300	-286	-123	-204.5	3432	17900
1.325	-290	-128	-209.0	3507	18250
1.350	-292	-136	-214.0	3591	18550
1.375	-294	-143	-218.5	3666	18850
1.400	-289	-154	-221.5	3717	19100
1.425	-295	-160	-227.5	3815	19500
1.450	-297	-169	-233.0	3910	19850
1.475	-299	-178	-238.5	4002	20200
1.500	-299	-191	-245.0	4111	20650

LOADING CONDITION 2

Rated Normal Load Plus Variable Side Load

U-bolt size : 24" pipe  
1" rod

Three tests N,O and P were performed. In each test the U-bolt was loaded with the LCDS rated normal load of 9920 lb. Side load was induced through the hydraulic pump.

The testing procedure is the same as for tests D,E and F.



TEST 'N'

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	9920
.025	-38	24	-7.0	117	9900
.050	-76	49	-13.5	227	9900
.075	-107	68	-19.5	327	9950
.100	-130	80	-25.0	420	10050
.125	-158	97	-30.5	512	10250
.150	-187	114	-36.5	612	10400
.175	-222	134	-44.0	738	10550
.200	-256	155	-50.5	847	10700
.225	-295	178	-58.5	982	10850
.250	-330	199	-65.5	1099	11000
.275	-365	220	-72.5	1217	11200
.300	-398	239	-79.5	1334	11300
.325	-433	260	-86.5	1451	11550
.350	-468	279	-94.5	1586	11700
.375	-496	295	-100.5	1686	11950
.400	523	309	-107.0	1795	12150
.425	-555	326	-114.5	1921	12350
.450	-584	341	-121.5	2039	12550
.475	-612	355	-128.5	2156	12800
.500	634	365	-134.5	2257	13000
.525	-660	377	-141.5	2374	13300
.550	-683	387	-148.0	2483	13500
.575	-711	402	-154.5	2593	13750
.600	-737	416	-160.5	2693	14000
.625	-766	432	-167.0	2802	14250
.650	-790	446	-172.0	2886	14500
.675	-818	462	-178.0	2987	14750
.700	-843	477	-183.0	3071	15000
.725	-868	492	-188.0	3155	15250
.750	-891	507	-192.0	3222	15500

TEST 'N' (CONTINUED)

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-908	519	-194.5	3264	15800
.800	-941	538	-201.5	3381	16050
.825	-962	553	-204.5	3432	16350
.850	-989	570	-209.5	3515	16650
.875	-1010	588	-215.0	3608	16950
.900	-1039	603	-218.0	3658	17250
.925	-1066	621	-222.5	3734	17550
.950	-1095	640	-227.5	3817	17850
.975	-1122	657	-232.5	3901	18150
1.000	-1146	674	-236.0	3960	18450
1.025	-1171	692	-239.5	4019	18800
1.050	-1202	711	-245.5	4119	19150
1.075	-1231	730	-250.5	4203	19500
1.100	-1260	750	-255.0	4279	20000
1.125	-1293	772	-260.5	4371	20150
1.150	-1325	793	-266.0	4463	20550
1.175	-1354	814	-270.0	4531	20900
1.200	-1391	838	-276.5	4640	21300
1.225	-1424	860	-282.0	4732	21600
1.250	-1458	885	-286.5	4807	22050
1.275	-1496	910	-293.0	4917	22400
1.300	-1538	938	-300.0	5034	22800
1.325	-1576	964	-306.0	5135	23250
1.350	-1616	993	-311.5	5227	23650
1.375	-1661	1022	-319.5	5361	24100
1.400	-1710	1053	-328.5	5512	24500
1.425	-1750	1081	-334.5	5613	24900
1.450	-1797	1113	-342.0	5739	25350
1.475	-1844	1143	-350.5	5881	25700
1.500	-1888	1172	-358.0	6007	26100

TEST '0'

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	-28	14	-7.0	117	9950
.025	-59	29	-15.0	252	9950
.050	-86	44	-21.0	352	9950
.075	-104	52	-26.0	436	9950
.100	-123	61	-31.0	520	9950
.125	-145	72	-36.5	612	9950
.150	-168	85	-41.5	696	9950
.175	-192	99	-46.5	780	9950
.200	-211	108	-51.5	864	10000
.225	-242	128	-57.0	956	10000
.250	-270	145	-62.5	1049	10100
.275	-295	160	-67.5	1133	10100
.300	-316	174	-71.0	1191	10100
.325	-342	189	-76.5	1284	10100
.350	-363	202	-80.5	1351	10100
.375	-385	215	-85.0	1426	10150
.400	-402	225	-88.5	1485	10200
.425	-421	238	-91.5	1535	10250
.450	-447	250	-98.5	1653	10300
.475	-466	261	-102.5	1720	10350
.500	-484	272	-106.0	1779	10450
.525	-503	282	-110.5	1854	10550
.550	-520	292	-114.0	1913	10700
.575	-542	303	-119.5	2005	10800
.600	565	316	-124.5	2089	11000
.625	590	328	-131.0	2198	11200
.650	-614	340	-137.0	2299	11350
.675	-636	353	-141.5	2374	11550
.700	-665	368	-148.5	2492	11700
.725	-691	382	-154.5	2593	11900
.750	-720	398	-161.0	2702	12100

TEST 'O'(CONTINUED)

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-750	414	-168.0	2819	12300
.800	-774	430	-172.0	2886	12500
.825	-803	447	-178.0	2987	12750
.850	-833	464	-184.5	3096	12950
.875	-859	480	-189.5	3180	13150
.900	-887	497	-195.0	3272	13400
.925	-919	516	-201.5	3381	13650
.950	-942	533	-204.5	3432	13900
.975	-969	549	-210.0	3524	14150
1.000	-992	566	-213.0	3574	14400
1.025	-1026	587	-219.5	3683	14700
1.050	-1046	602	-222.0	3725	15000
1.075	-1069	618	-225.5	3784	15300
1.100	-1102	638	-232.0	3893	15600
1.125	-1123	654	-234.5	3935	15950
1.150	-1147	671	-238.0	3994	16300
1.175	-1171	687	-242.0	4061	16600
1.200	-1202	707	-247.5	4153	16950
1.225	-1231	725	-253.0	4245	17300
1.250	-1262	747	-257.5	4321	17650
1.275	-1286	764	-261.0	4380	18000
1.300	-1305	780	-262.5	4405	18200
1.325	-1347	806	-270.5	4539	18650
1.350	-1387	832	-277.5	4656	19050
1.375	-1415	853	-281.0	4715	19450
1.400	-1449	877	-286.0	4799	19800
1.425	-1486	901	-292.5	4908	20200
1.450	-1525	928	-298.5	5009	20600
1.475	-1557	952	-302.5	5076	21000
1.500	-1596	978	-309.0	5185	21350



TEST 'P

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.0	0	0	0	0	9920
.025	-58	35	-11.5	193	9920
.050	-96	57	-19.5	327	9850
.075	-132	80	-26.0	436	9850
.100	-176	108	-34.0	571	9800
.125	-215	134	-40.5	680	9750
.150	-253	160	-46.5	780	9700
.175	-287	182	-52.5	881	9700
.200	-322	206	-58.0	973	9650
.225	-353	227	-63.0	1057	9650
.250	-385	247	-69.0	1158	9650
.275	-412	265	-73.5	1233	9650
.300	-438	282	-78.0	1309	9650
.325	-460	294	-83.0	1393	9650
.350	-482	305	-88.5	1485	9750
.375	-501	314	-93.5	1569	9800
.400	-522	324	-99.0	1661	9850
.425	-541	333	-104.0	1745	9950
.450	-560	342	-109.0	1829	10000
.475	-581	352	-114.5	1921	10100
.500	-599	360	-119.5	2005	10250
.525	-620	369	-125.5	2106	10350
.550	-637	378	-129.5	2173	10450
.575	-656	389	-133.5	2240	10600
.600	-682	403	-139.5	2341	10800
.625	-699	414	-142.5	2391	10900
.650	-719	425	-147.0	2467	11100
.675	-742	438	-152.0	2551	11250
.700	-763	450	-156.5	2626	11400
.725	-789	465	-162.0	2718	11600
.750	-816	481	-167.5	2810	11750

TEST 'P (CONTINUED)

Side Deflection (IN)	Strains (in/in x 10 <sup>-6</sup> )			Side Load (LB)	Normal Load (LB)
	Gage #1	Gage #4	Average		
.775	-841	496	-172.5	2895	12000
.800	-867	512	-177.5	2978	12200
.825	-893	528	-182.5	3062	12450
.850	-919	544	-187.5	3146	12700
.875	-944	560	-192.0	3222	12950
.900	-966	575	-195.5	3280	13250
.925	-991	591	-200.0	3356	13550
.950	-1011	606	-202.5	3398	13850
.975	-1035	622	-206.5	3465	14150
1.000	-1060	638	-211.0	3541	14450
1.025	-1085	655	-215.0	3608	14800
1.050	-1110	672	-219.0	3675	15150
1.075	-1133	688	-222.5	3734	15450
1.100	-1164	708	-228.0	3826	15850
1.125	-1200	730	-235.0	3943	16150
1.150	-1220	747	-236.5	3968	16550
1.175	-1254	769	-242.5	4069	17000
1.200	-1286	790	-248.0	4161	17400
1.225	-1322	813	-254.5	4271	17800
1.250	-1352	835	-258.5	4338	18200
1.275	-1386	857	-264.5	4438	18600
1.300	-1413	877	-268.0	4497	18950
1.325	-1452	903	-274.5	4606	19450
1.350	-1482	924	-279.0	4682	19800
1.375	-1519	949	-285.0	4782	20250
1.400	-1559	975	-292.0	4900	20650
1.425	-1600	1002	-299.0	5017	21100
1.450	-1632	1025	-303.5	5093	21500
1.475	-1671	1050	-310.5	5210	22000
1.500	-1709	1076	-316.5	5311	22400

LOADING CONDITION 3  
Normal Load to Failure

U-bolt size : 24" pipe  
1" rod

Three tests Q, R and S were performed. In each test the U-bolt was loaded normally by means of the tensile machine. The normal load was increased gradually until the U-bolt failed. The charting capability of the machine was utilized to plot the normal load against the normal deflection or elongation of the U-bolt. The location of failure for each U-bolt was observed.

LOAD RATING ACCORDING TO ASME NF-3260

SIZES	MATERIAL PROPERTIES - Stress in KSI Values from C.M.T.R.'s at 70° F				
	$S_u$	$S_y$	$.5S_u$	$.6S_y$	$F_{ALL}/S_u$
4" PIPE 1/2" ROD	63.08	45.13	31.54	27.08	0.43
24" PIPE 1" ROD	73.4	51.6	36.7	30.96	0.42

SIZES	MATERIAL PROPERTIES - Stress in KSI Values from ASME Appx. 1 at 100° F				
	$S_u$	$S_y$	$.5S_u$	$.6S_y$	$F_{ALL}/S_u$
4" PIPE 1/2" ROD	58	36	29	21.6	0.37
24" PIPE 1" ROD	58	36	29	21.6	0.37

SIZES	MATERIAL PROPERTIES - Stress in KSI Values from ASME Appx. 1 at 650° F				
	$S_u$	$S_y$	$.5S_u$	$.6S_y$	$F_{ALL}/S_u$
4" PIPE 1/2" ROD	58	26.1	29	15.66	0.27
24" PIPE 1" ROD	58	26.1	29	15.66	0.27

$F_{ALL}$  : Smallest of  $.5S_u$  and  $.6S_y$

Per NF-3262.3, Winter 1974 Addenda,

$$\text{LOAD RATING} = T.L. \times (F_{ALL}/S_u)$$

where T.L. = TEST LOAD

-the TEST LOAD for normal loading is the final load that produces failure of the U-bolt,

-for axial and lateral loading, the test load may be taken as the load at maximum observed axial or lateral deflection, since at this deflection the support was still sustaining the indicated loads.



# LOADS AND DEFLECTIONS FOR USE IN LOAD RATING OF 4" & 24" PIPE U-BOLTS

TEST	LOAD	DEFLECTION	TEST	LOAD	DEFLECTION
A	1418	.75	K	4002	1.50
B	2014	.85	L	3725	1.25
C	1930	.85	M	4111	1.50
D	5185	.90	N	6007	1.50
E	4715	.85	O	5185	1.50
F	4858	.95	P	5311	1.50
G	19825	FAILURE	Q	89500	FAILURE
H	18675	FAILURE	R	90000	FAILURE
I	19850	FAILURE	S	90750	FAILURE

Based on the minimum values of the above table,

TEST LOAD for 4" Axial Load Test = 1418 LBS.  
 TEST LOAD for 4" Lateral Load Test = 4715 LBS.  
 TEST LOAD for 4" Normal Load Test = 18675 LBS.

TEST LOAD for 24" Axial Load Test = 3725 LBS.  
 TEST LOAD for 24" Lateral Load Test = 5185 LBS.  
 TEST LOAD for 24" Normal Load Test = 89500 LBS.

For material properties at 70° (per C.M.T.R.'s),

NF Load Rating for 4" Axial Load = 1418 (0.43) = 610 LBS.  
 NF Load Rating for 4" Lateral Load = 4715 (0.43) = 2027 LBS.  
 NF Load Rating for 4" Normal Load = 18675 (0.43) = 8030 LBS.  
 NF Load Rating for 24" Axial Load = 3725 (0.42) = 1565 LBS.  
 NF Load Rating for 24" Lateral Load = 5185 (0.42) = 2178 LBS.  
 NF Load Rating for 24" Normal Load = 89500 (0.42) = 37590 LBS.

For material properties at 100° (per ASME Appex. I),

NF Load Rating for 4" Axial Load = 1418 (0.37) = 525 LBS.  
 NF Load Rating for 4" Lateral Load = 4715 (0.37) = 1745 LBS.  
 NF Load Rating for 4" Normal Load = 18675 (0.37) = 6910 LBS.  
 NF Load Rating for 24" Axial Load = 3725 (0.37) = 1378 LBS.  
 NF Load Rating for 24" Lateral Load = 5185 (0.37) = 1918 LBS.  
 NF Load Rating for 24" Normal Load = 89500 (0.37) = 33115 LBS.

For material properties at 650° (per ASME Appex. I),

NF Load Rating for 4" Axial Load = 1418 (0.27) = 383 LBS.  
 NF Load Rating for 4" Lateral Load = 4715 (0.27) = 1273 LBS.  
 NF Load Rating for 4" Normal Load = 18675 (0.27) = 5042 LBS.  
 NF Load Rating for 24" Axial Load = 3725 (0.27) = 1006 LBS.  
 NF Load Rating for 24" Lateral Load = 5185 (0.27) = 1400 LBS.  
 NF Load Rating for 24" Normal Load = 89500 (0.27) = 24165 LBS.

## TEST RESULTS

### AXIAL LOADING OF U-BOLTS

#### 4" PIPE

1/2" rod U-bolts

Tests A,B,C

Figures A-1 to A-6 of Appendix I

Axial loads of between 1342 and 1552 lbs were required to produce 0.75" of deflection in the axial direction for three different specimens. Load-deflection paths were comparatively similar for all U-bolts.

Normal load increased from the LCDS rated load of 2260. lbs at the beginning of the test, to a load of between 5250 and 5860 lbs at 0.75" of axial deflection.

Upon removal of the axial load, the axial deflection went down to approximately 0.51" and the normal load was reduced to 2500 lbs (test C data only). No measurements were taken upon removal of normal load.

#### 24" PIPE

1" rod U-bolts

Tests K,L,M

Figures A-16 to A-21 of Appendix I

Axial loads of between 3213 and 3725 lbs were required to produce 1.25" of deflection in the axial direction for three different specimens. Load-deflection paths were comparatively similar except for test M when the U-bolt "slipped" on the pipe at about 0.6" of axial deflection.

Normal load increased from the LCDS rated load of 9920 lbs at the beginning of the test to a load of between 16600 and 17300 lbs at 1.25" of axial deflection.

Upon removal of the axial load, the axial deflection went down to between 0.602" and 0.742" and the normal load was reduced to between 7500 lbs and 9300 lbs. No measurements were taken upon removal of normal load.

## TEST RESULTS (CONTINUED)

### LATERAL LOADING OF U-BOLTS

#### 4" PIPE

1/2" rod U-bolts

Tests D,E,F

Figures A-7 to A-12 of Appendix I

Lateral loads of between 3524 and 4044 lbs were required to produce 0.75" of deflection in the lateral direction, for three different specimens. Load-deflection paths were comparatively similar for all U-bolts.

Normal load increased from the LCDS rated load of 2260 lbs at the beginning of the test, to a load of between 4500 and 4875 lbs at 0.75" of lateral deflection.

Upon removal of the lateral load, the axial deflection was reduced to between 0.507" and 0.627" and the normal load was reduced to between 2450 lbs and 2570 lbs (tests E and F only). No measurements were taken upon removal of normal load.

#### 24" PIPE

1" rod U-bolts

Tests N,O,P

Figures A-22 to A-27 of Appendix I

Lateral loads of between 4321 and 4807 lbs were required to produce 1.25" of deflection in the lateral direction for three different specimens. Load-deflection paths were comparatively similar for all U-bolts.

Normal load increased from the LCDS rated load of 9920 lbs at the beginning of the test, to a load of between 18200 and 22050 lbs at 1.25" of lateral deflection.

Upon removal of the lateral load, the axial deflection was reduced to between 0.638" and 0.683" and the normal load was reduced to between 7450 lbs and 8350 lbs. No measurements were taken upon removal of normal load.

## TEST RESULTS (CONTINUED)

### NORMAL LOADING OF U-BOLTS

#### 4" PIPE

1/2" rod U-bolts

Tests G,H,I

Figures A-13 to A-15 of Appendix I

Normal loads of 19825, 18675 and 19850 lbs were required to break the three different U-bolts.

Based on theoretical calculations involving the tensile stress area of the threaded section of the U-bolts and the ultimate strength of the U-bolt material, failure should occur at 17890 lbs.

Failure occurred, as expected, in the threaded section of the U-bolt, in the region between the two hex nuts securing the U-bolt on the base plate. Excessive necking is evident.

#### 24" PIPE

1" rod U-bolts

Tests Q,R,S

Figures A-28 to A-30 of Appendix I

Normal loads of 89500, 90000 and 90750 lbs were required to break the three different U-bolts.

Based on theoretical calculations involving the tensile stress area of the threaded section of the U-bolts and the ultimate strength of the U-bolt material, failure should occur at 88475 lbs.

Failure occurred, as expected, in the threaded section of the U-bolt, in the region between the two hex nuts securing the U-bolt on the base plate. Excessive necking is evident.



# SUMMARY OF RESULTS

PIPE SIZE	LOAD DIRECTION	TEST	MAXIMUM TEST LOAD	NF-RATED* LOAD	LCD LOAD	REMARKS
4"	AXIAL	A	1418	383	N/A	no failure $\Delta = 0.75"$
		B	2014			no failure $\Delta = 0.85"$
		C	1930			no failure $\Delta = 0.85"$
4"	LATERAL	D	5185	1273	184	no failure $\Delta = 0.90"$
		E	4715			no failure $\Delta = 0.85"$
		F	4858			no failure $\Delta = 0.95"$
4"	NORMAL	G	19825	5042	2260	failure
		H	18675			failure
		I	19850			failure
24"	AXIAL	K	4002	1006	N/A	no failure $\Delta = 1.50"$
		L	3725			no failure $\Delta = 1.25"$
		M	4111			no failure $\Delta = 1.50"$
24"	LATERAL	N	6007	1400	N/A	no failure $\Delta = 1.50"$
		O	5185			no failure $\Delta = 1.50"$
		P	5311			no failure $\Delta = 1.50"$
24"	NORMAL	Q	89500	24165	9920	failure
		R	90000			failure
		S	90750			failure

All loads in pounds

- \* These NF rated loads are those loads which have been extracted from the test results of the U-bolt specimens covered in this report. These loads are determined in accordance with the ASME Boiler and Pressure Vessel Code, Section III, Subsection NF, Article NF-3262.3, Winter 1974 Addenda. They reflect the maximum design load similar U-bolts installed at the plant may be qualified to. These values are presented to emphasize the fact that the items under investigation poses considerable reserve strength in comparison to their published design load.

## CONCLUSIONS

During all tests, the U-bolt specimens sustained permanent deformation with no loss in the load-carrying capacity for either the axial or lateral loading direction. The load carrying capability only ceased at the actual U-bolt failure for the pullout test. For both the axial and lateral load tests, the U-bolts sustained the indicated loads.

Determination of the yield point of the U-bolt cannot be made from the plots of the tests included in the Appendix I. The nature of the applied normal load causes a stress stiffening effect where the state of stress in the member adds significantly to the overall stiffness. Hence with the U-bolts being loaded both normally and in bending, the transition between elastic and inelastic behavior is difficult to detect. It was observed during the course of testing that under the simultaneous loading, axial displacements in excess of one-eighth inch caused no noticeable permanent deformation upon removal of load.

The increase in normal load as axial or lateral deflection of the U-bolt increased may be attributed to the following. As the U-bolt is displaced, the bending resistance of the U-bolt predominates at smaller values of deflection. As yielding proceeds through the cross-section, this bending resistance diminishes. To account for this condition, the U-bolt resists load in the normal direction along the U-bolt axis. Throughout the course of each test, a space or "gap" existed between the pipe spool and base fixture as the pipe was forced upward against the U-bolt. This "gap" never closed. Stretching of the U-bolt caused the normal load to increase. This points out the fact that there is a tremendous amount of reserve strength available in the U-bolt to accommodate very large load increases. Also to be concluded is the fact that under the action of U-bolt tension and bending forces at low values of displacement (below one-eighth inch), no permanent deformation occurs in U-bolts which undergo such loading.

To summarize, it must be stated that failure of the specimens due to either axial or lateral load was not accomplished, even at the excessive maximum values of displacement for which the testing was done. It appears that after the elastic limit of the material was reached, yielding occurred over a wide span of deflection due to the nature of the load. Even under cyclic loading with displacements in the order of one-eighth inch, failure of the bolt due to fatigue is nonexistent, since no yielding was evident within this range of deflections.

In light of these observations it may be concluded that the U-bolt assemblies will perform satisfactorily at postulated displacement values under operating conditions. Added to the safe performance of the U-bolts is the evidence contained herein of their substantial reserve strength.

## APPENDIX I

The purpose of this appendix is to graphically represent the results obtained through testing the 4" and 24" pipe U-bolts. The graphs are self-explanatory, depicting deflection of the pipe and U-bolt (axial or lateral) and load (axial or lateral) required to produce that deflection.

Each test represents a separate U-bolt, and each curve represents a separate test. Three curves of the same test (axial load on 4" pipe U-bolt for instance) are also plotted together to show comparative behavior over the entire deflection range. A combined graph is also included to show behavior over the first one-eighth inch of deflection.

TEST "A" AXIAL LOAD - 4" PIPE

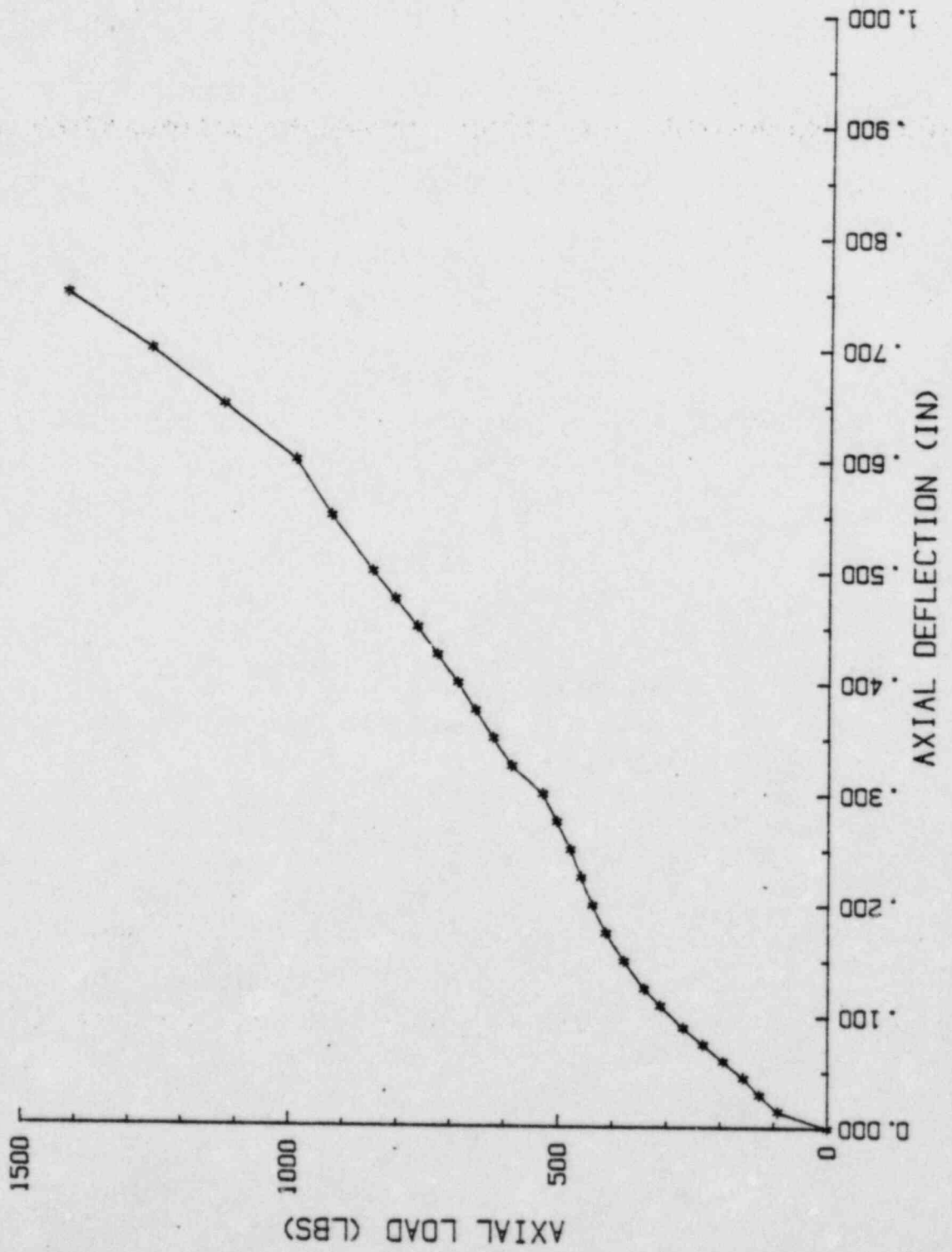


FIGURE A-1 TEST "A" : AXIAL DEFLECTION VS. AXIAL LOAD



TEST "B" AXIAL LOAD - 4" PIPE

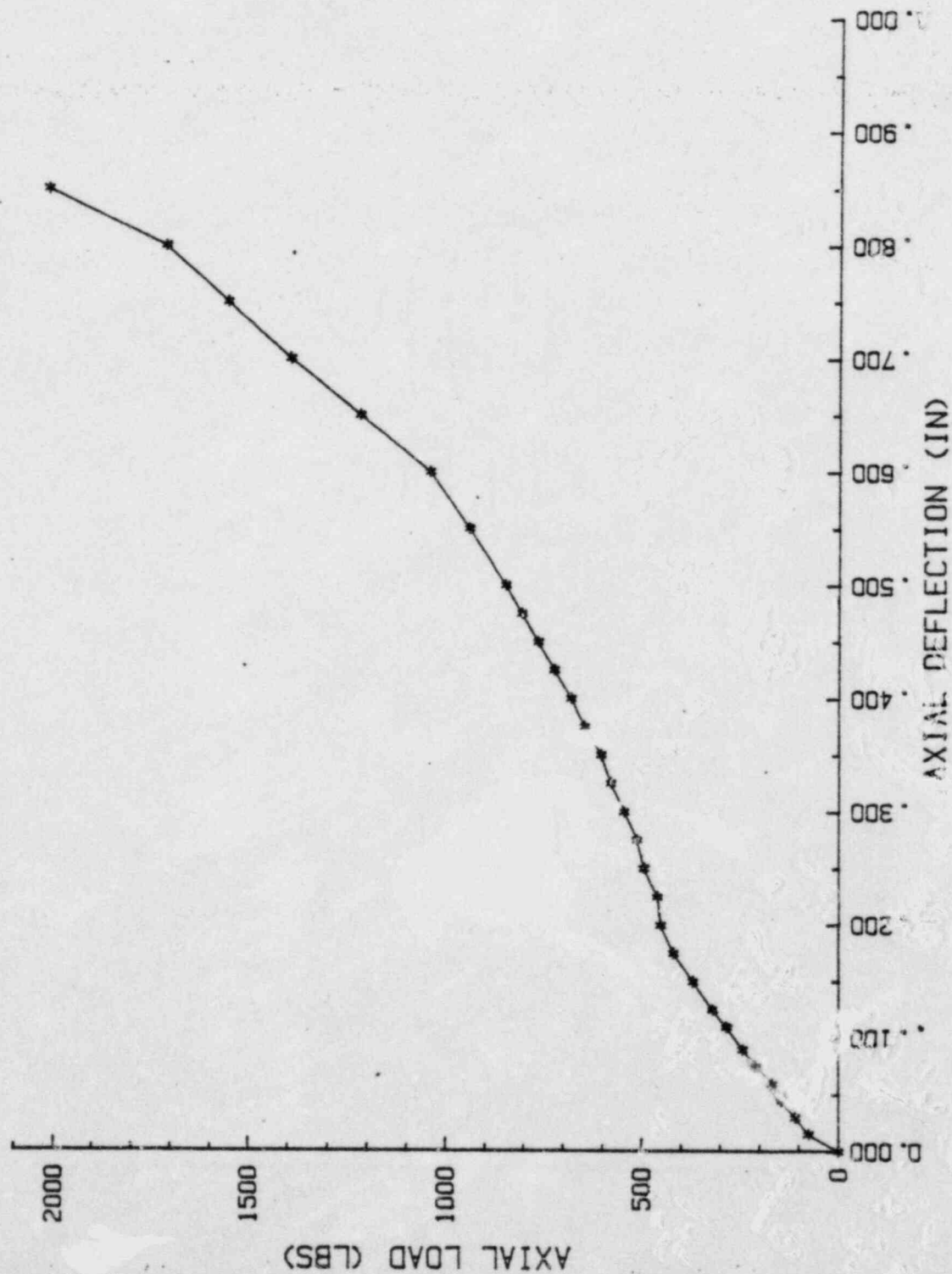


FIGURE A-2 TEST "B": AXIAL DEFLECTION VS. AXIAL LOAD

TEST "C" AXIAL LOAD - 4" PIPE

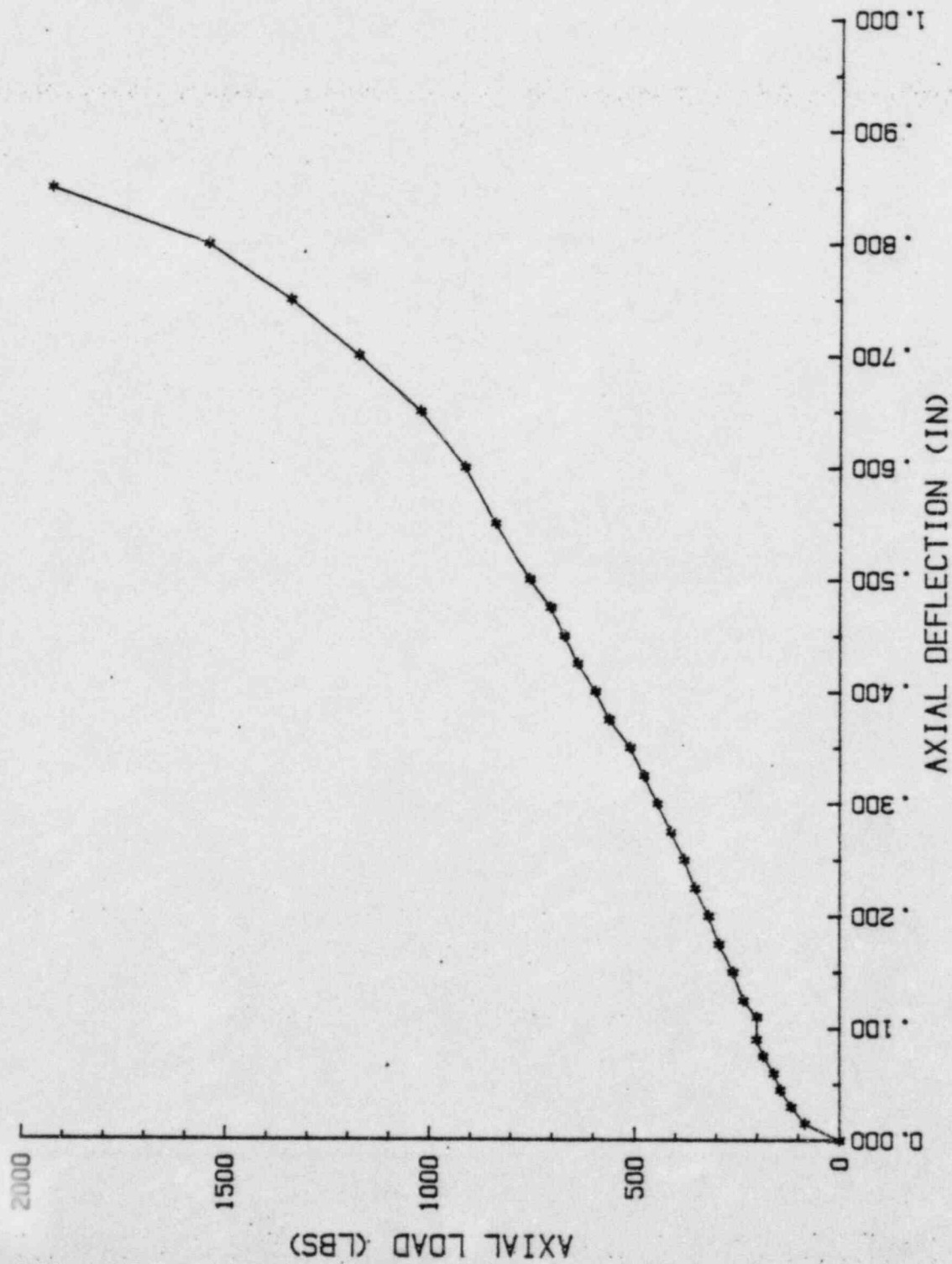


FIGURE A-3 TEST "C": AXIAL DEFLECTION VS. AXIAL LOAD

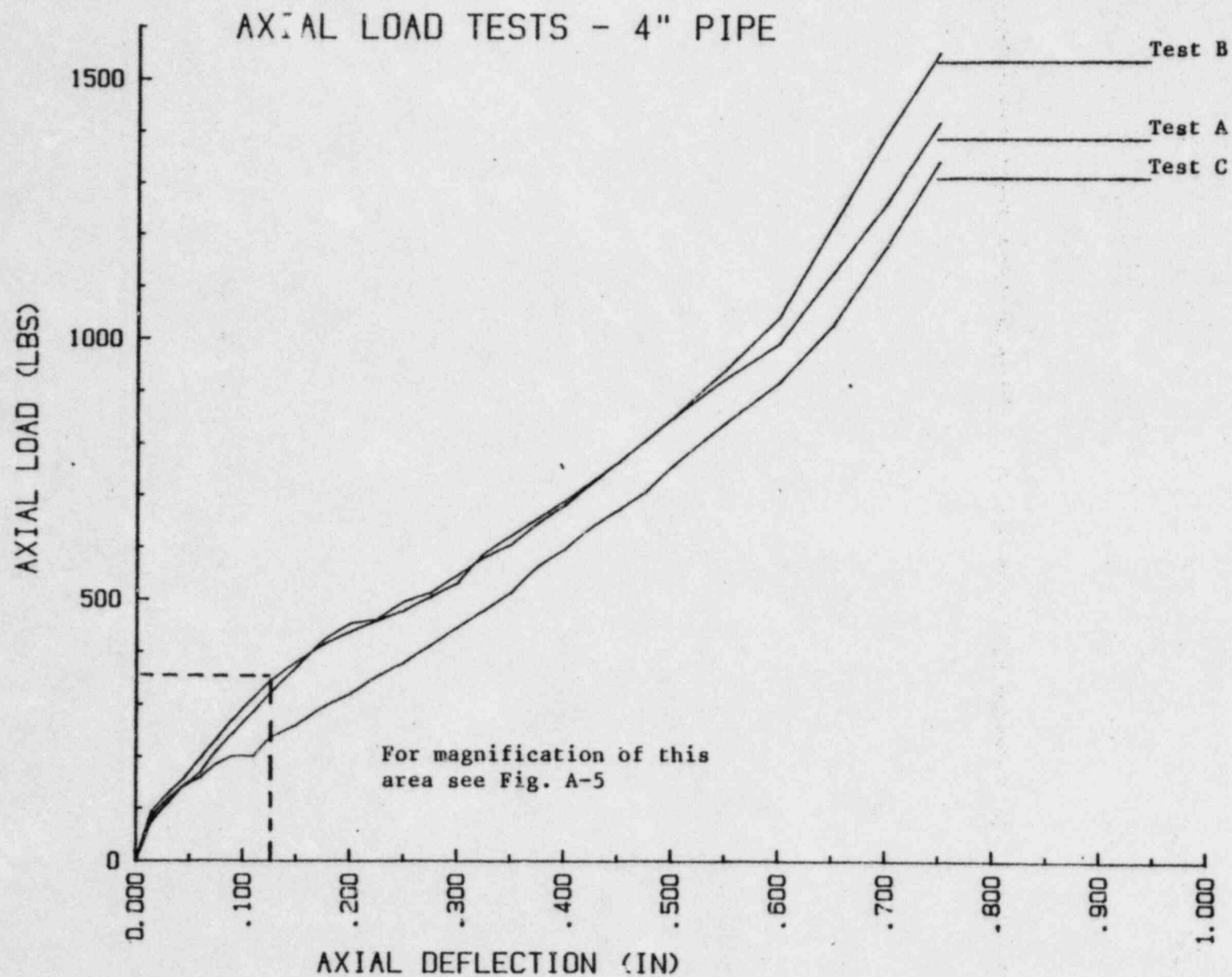


FIGURE A-4 TEST COMPARISONS: AXIAL DEFLECTION VS. AXIAL LOAD

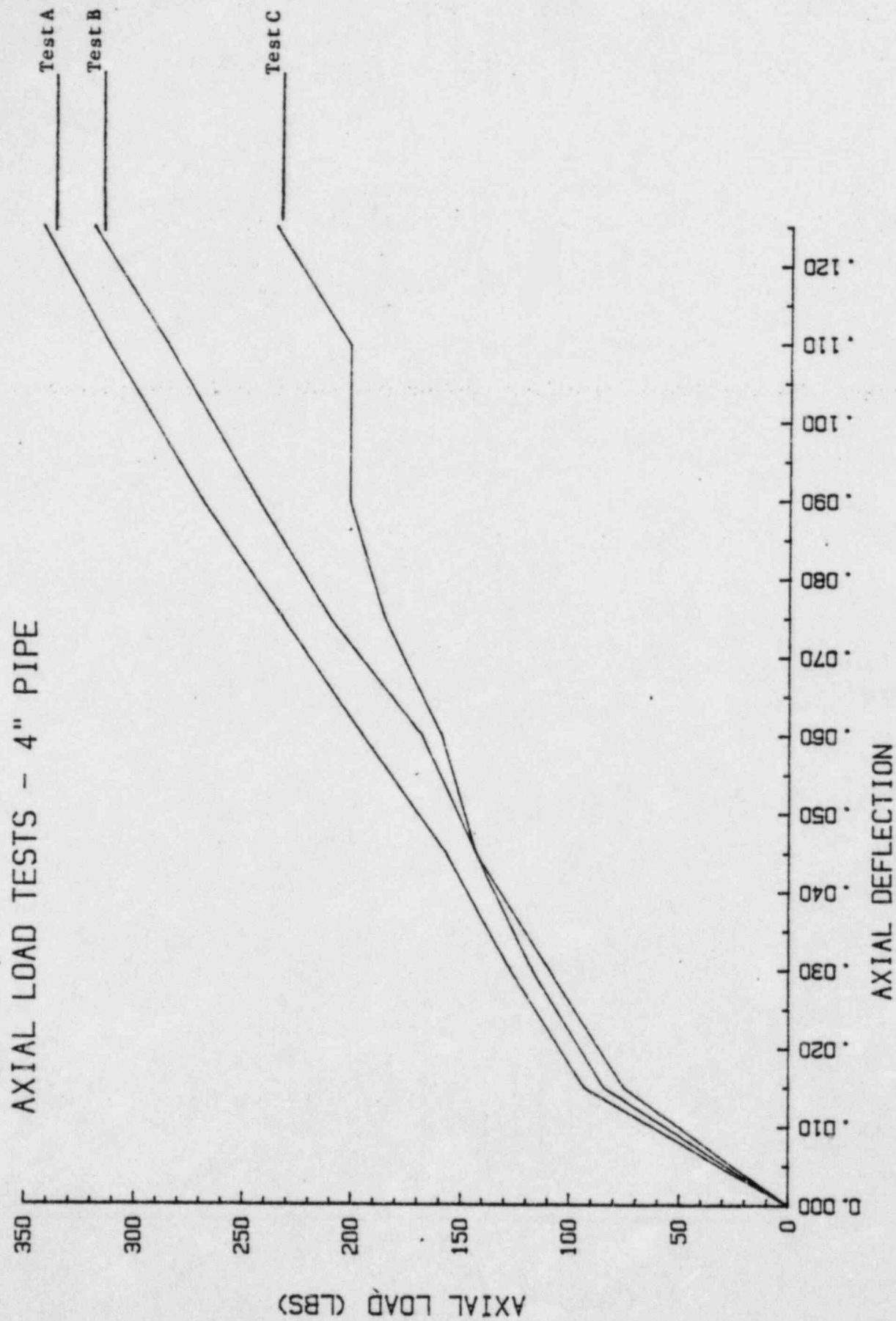


FIGURE A-5 TEST COMPARISONS: AXIAL DEFLECTION VS. AXIAL LOAD  
SCALE MAGNIFICATION- 0 IN. TO .125 IN.



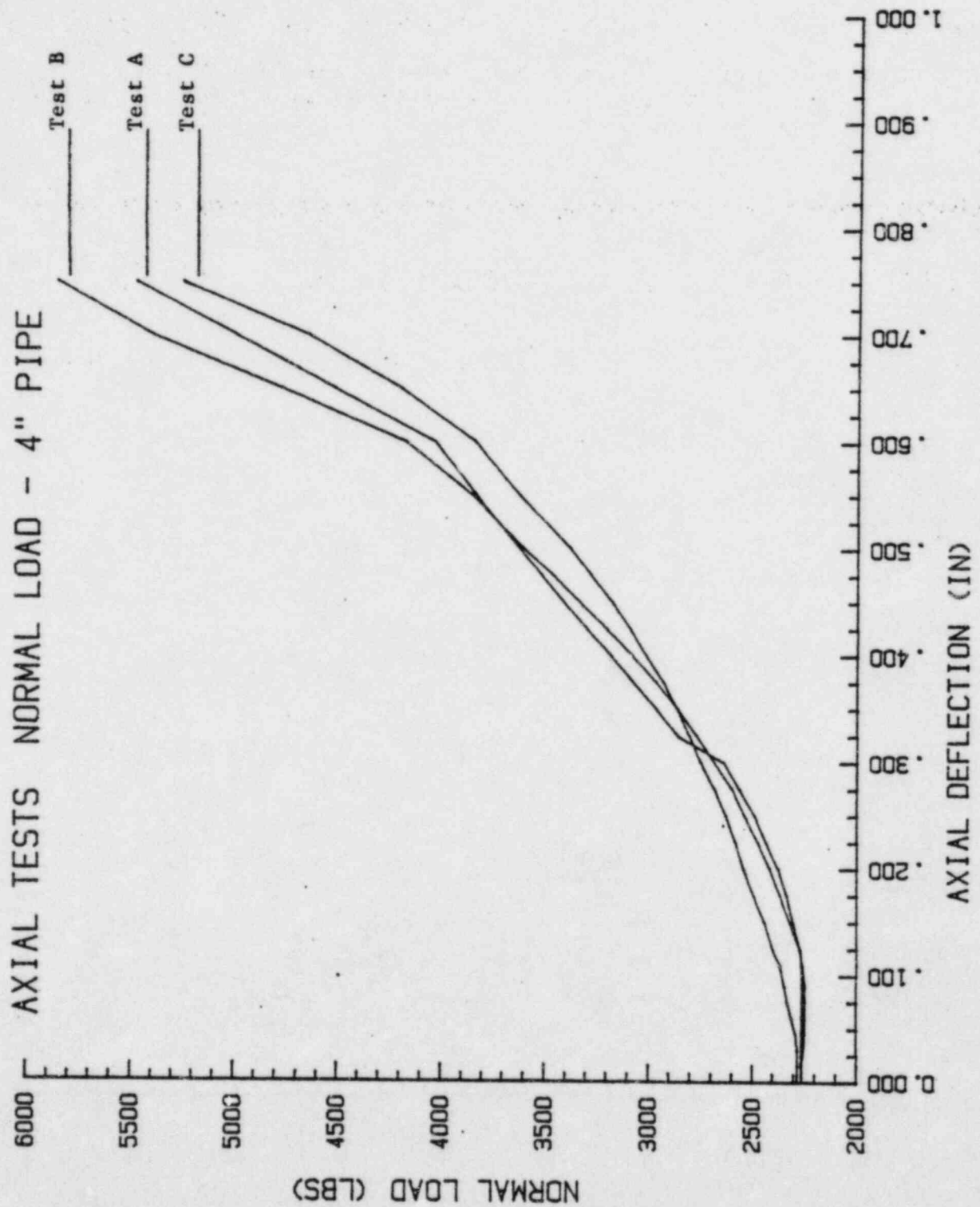


FIGURE A-6 TEST COMPARISONS: AXIAL DEFLECTION VS. NORMAL LOAD

# TEST "D" LATERAL LOAD - 4" PIPE

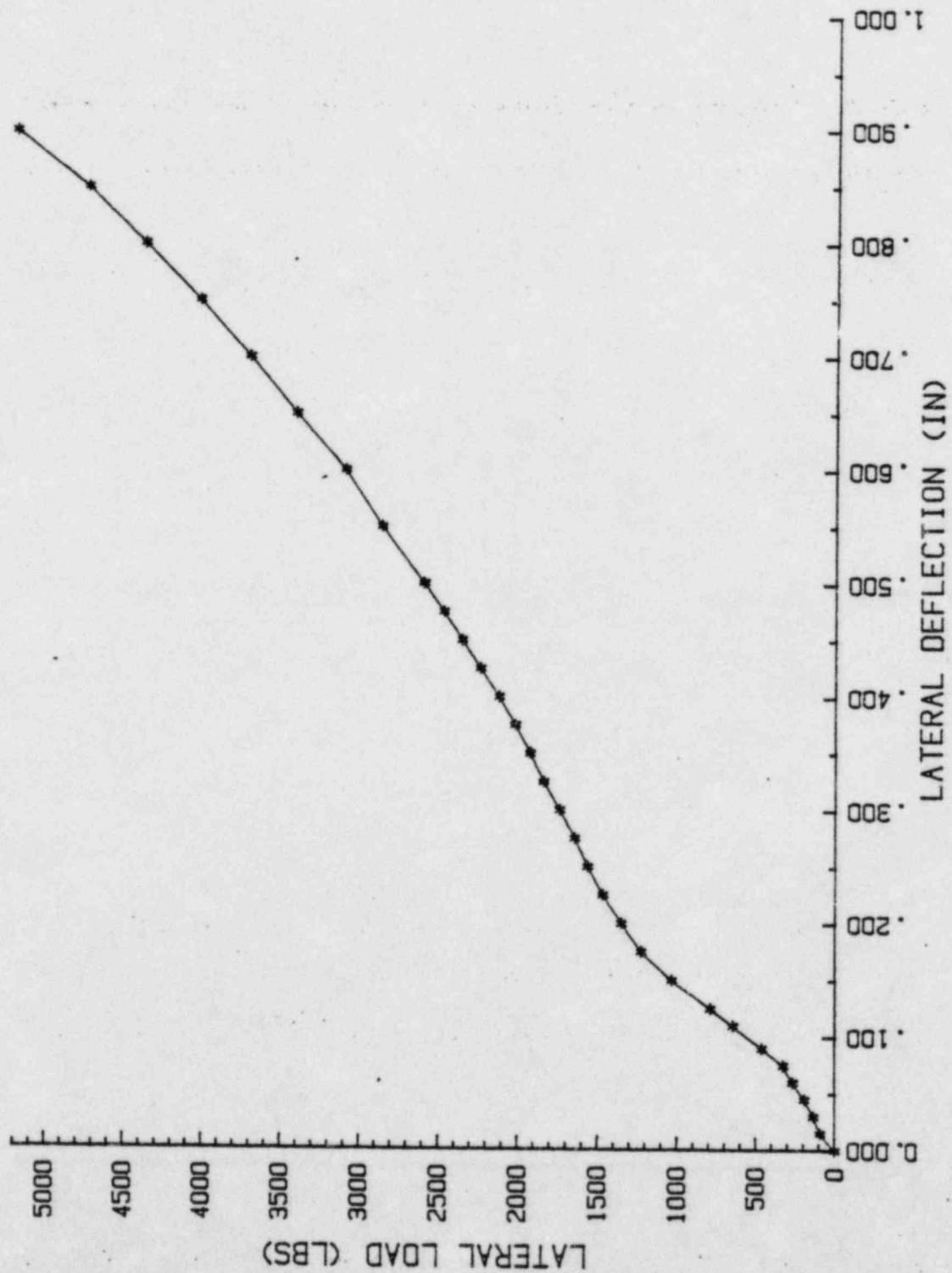


FIGURE A-7 TEST "D": LATERAL DEFLECTION VS. LATERAL LOAD

# TEST "E" LATERAL LOAD - 4" PIPE

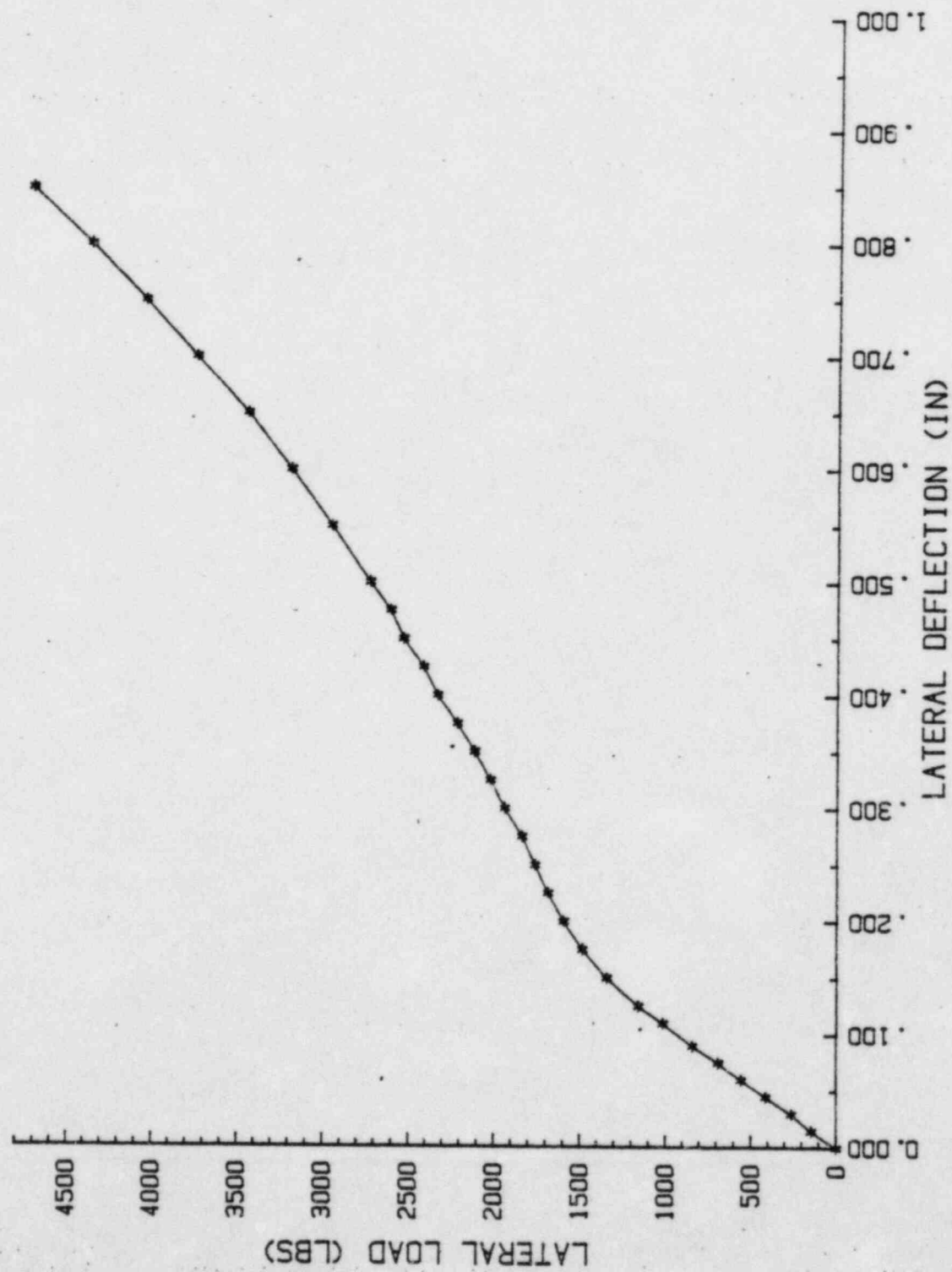


FIGURE A-8 TEST "E": LATERAL DEFLECTION VS. LATERAL LOAD

TEST "F" LATERAL LOAD - 4" PIPE

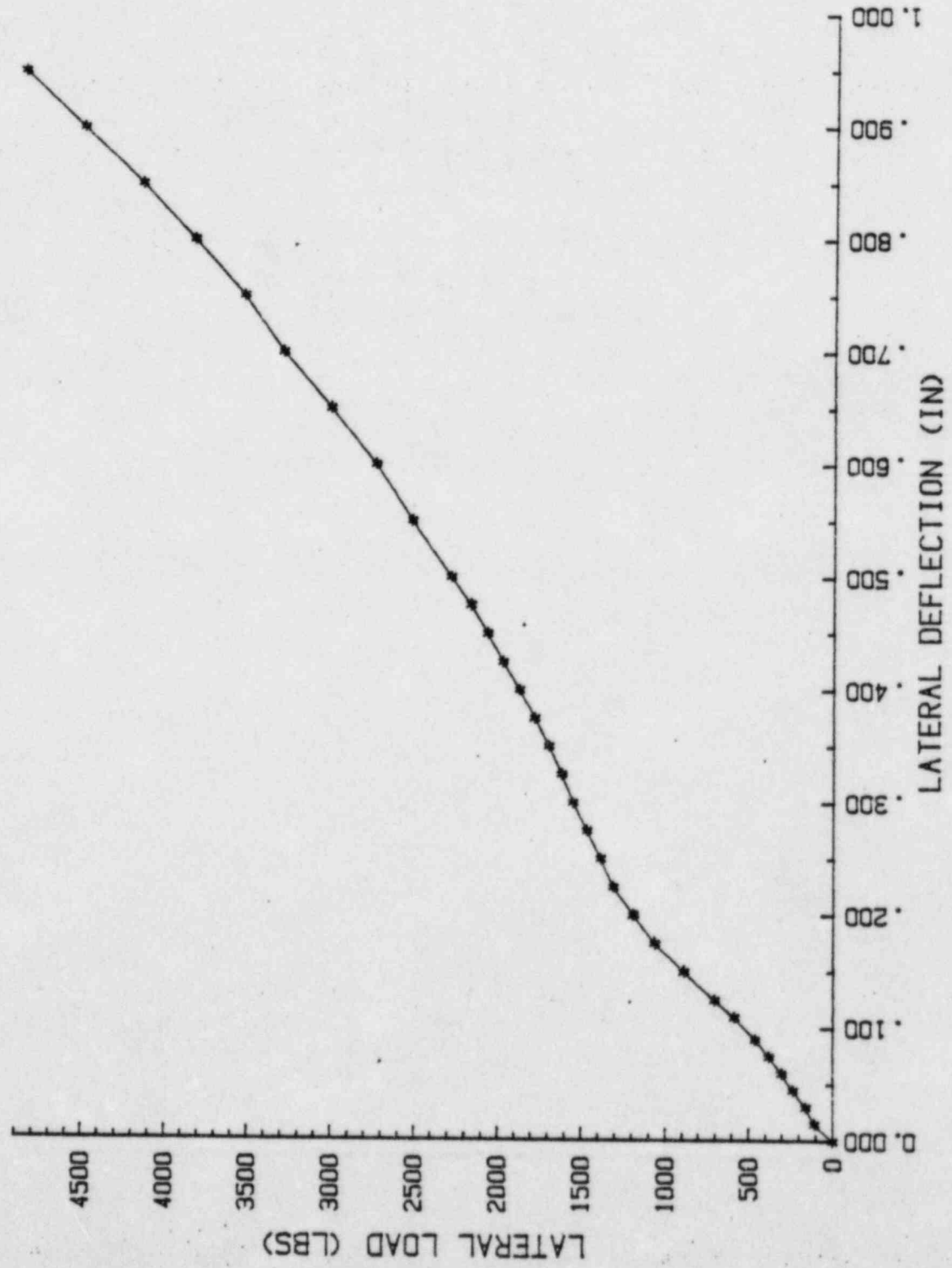


FIGURE A-9 TEST "F": LATERAL DEFLECTION VS. LATERAL LOAD



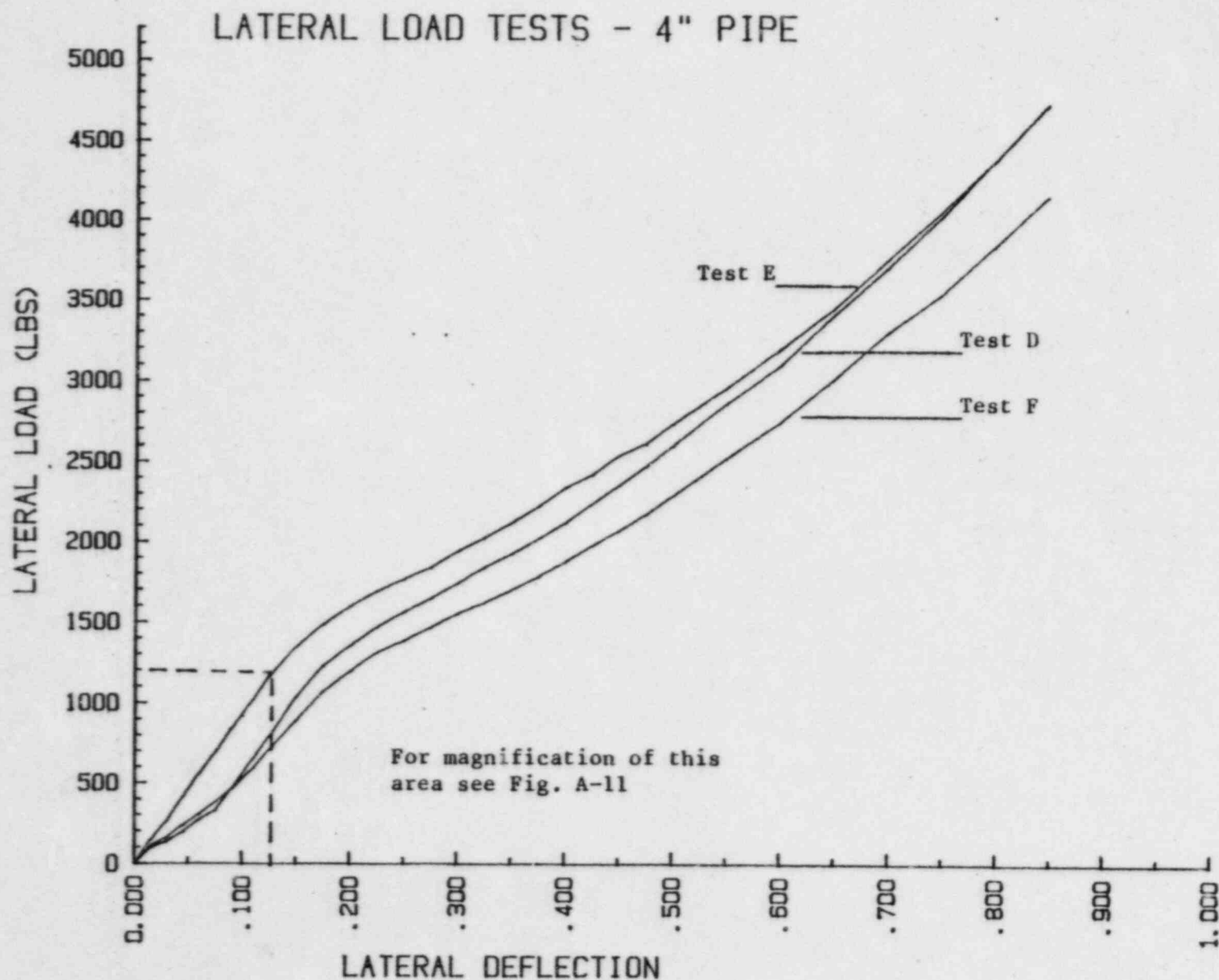
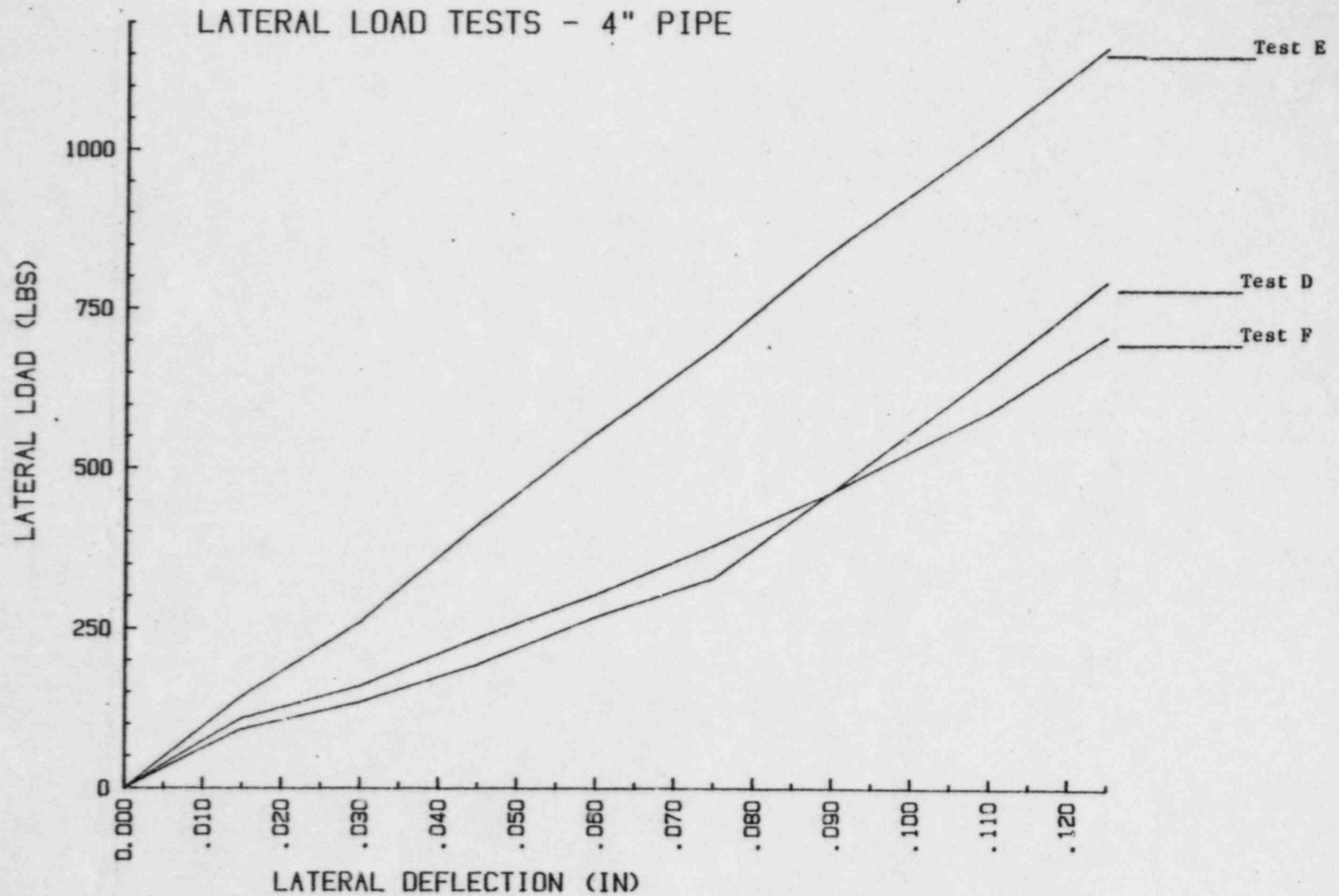


FIGURE A-10 TEST COMPARISONS: LATERAL DEFLECTION VS. LATERAL LOAD



**FIGURE A-11** TEST COMPARISONS: LATERAL DEFLECTION VS. LATERAL LOAD  
SCALE MAGNIFICATION- 0 IN. TO .125 IN.

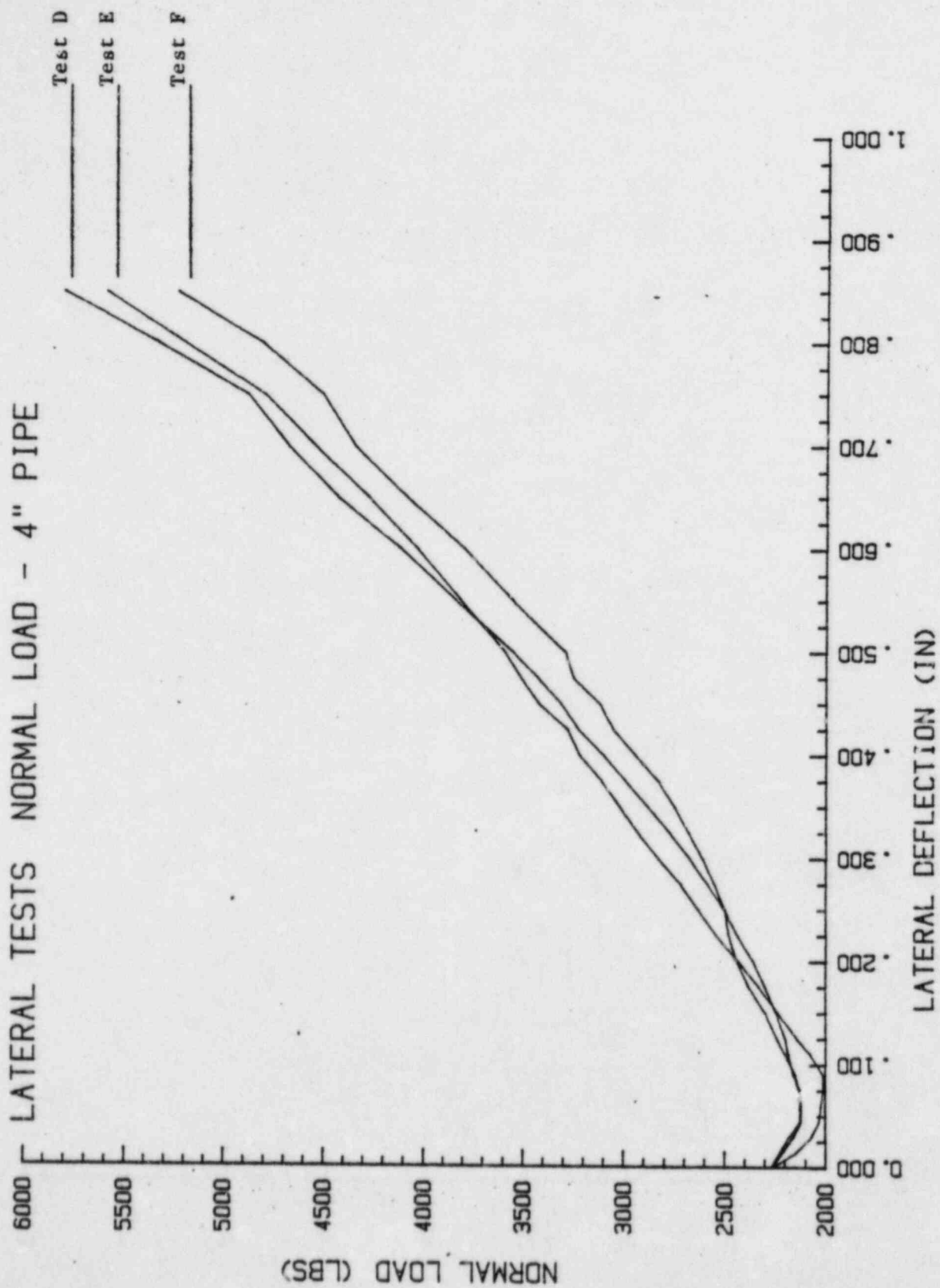
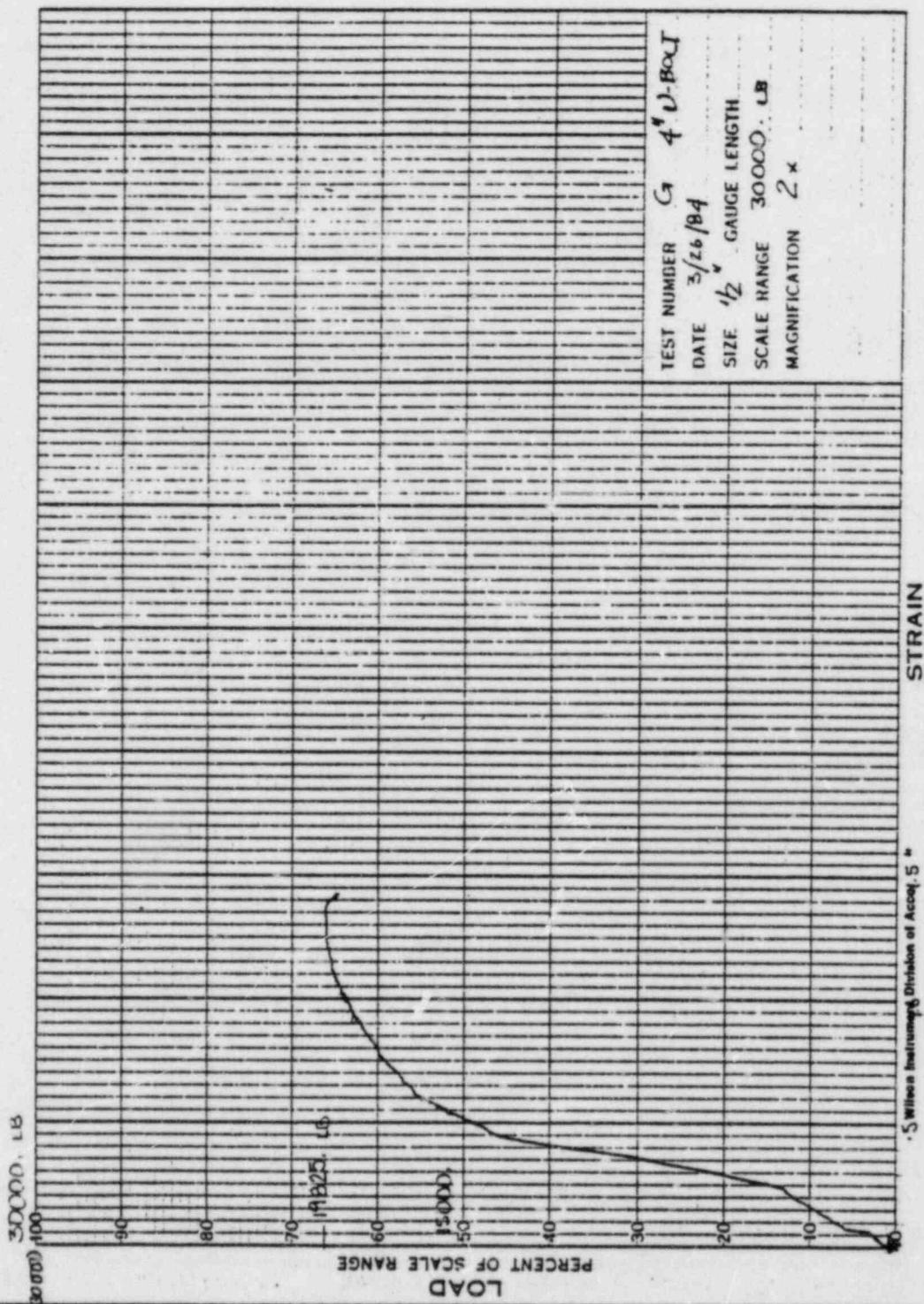


FIGURE A-12 TEST COMPARISONS: LATERAL DEFLECTION VS. NORMAL LOAD



DIVIDE BY MAGNIFICATION RATIO

Riddle® Testing Machines

FIGURE A-13 TEST "G" : NORMAL LOAD VS. ELONGATION

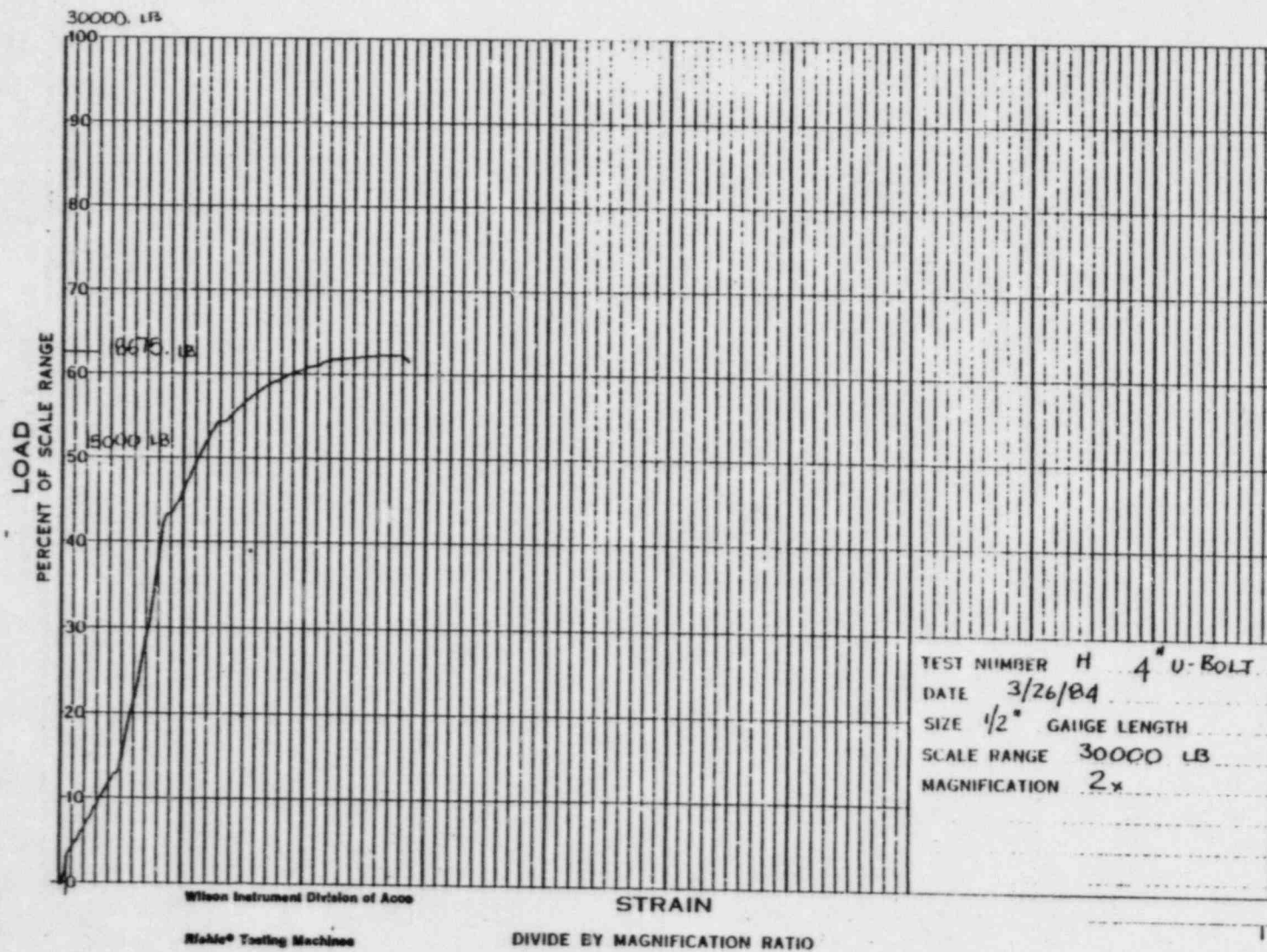


FIGURE A-14 TEST "H" : NORMAL LOAD VS. ELONGATION



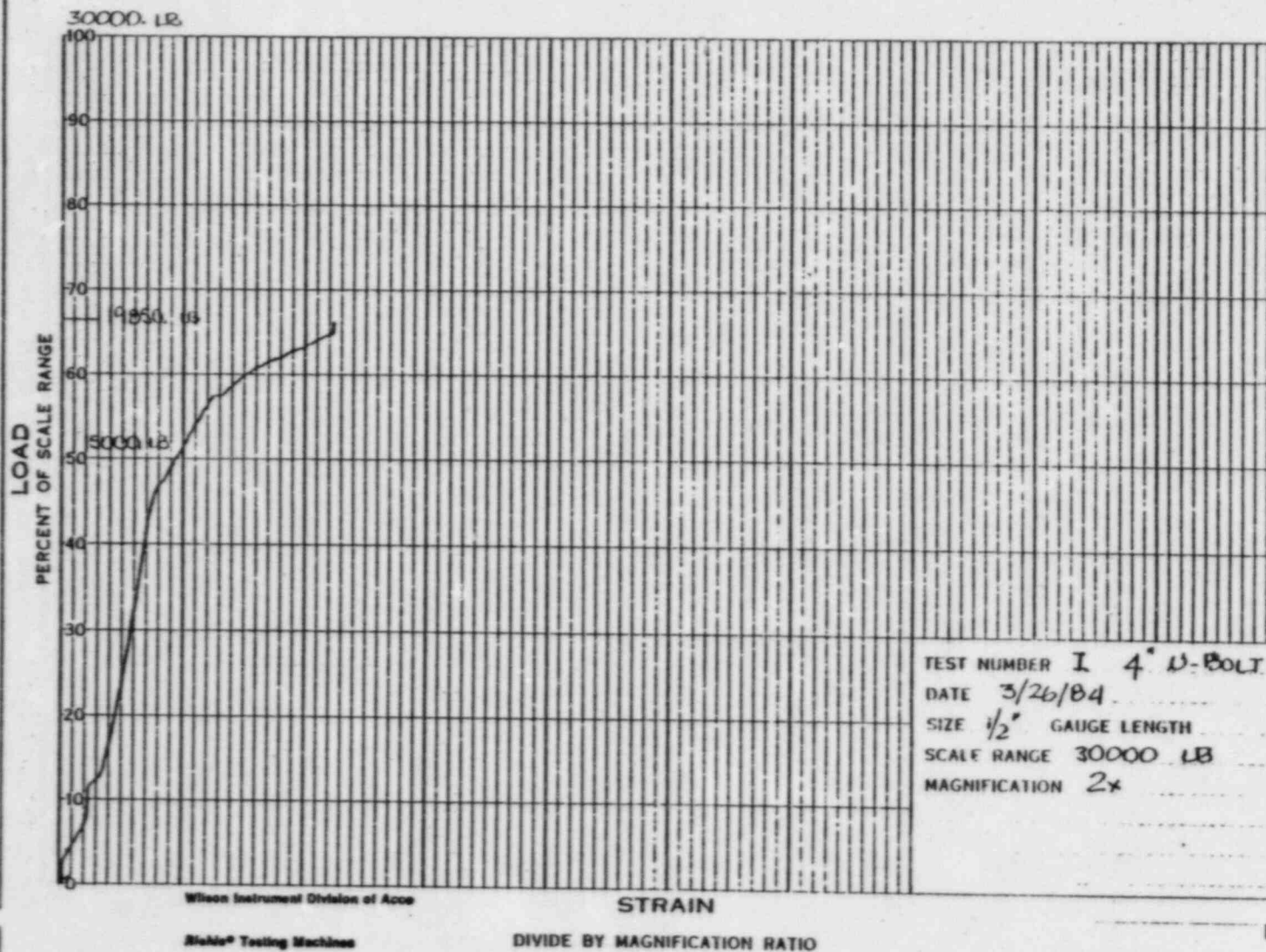


FIGURE A-15 TEST "I" : NORMAL LOAD VS. ELONGATION

TEST "K" AXIAL LOAD - 24" PIPE

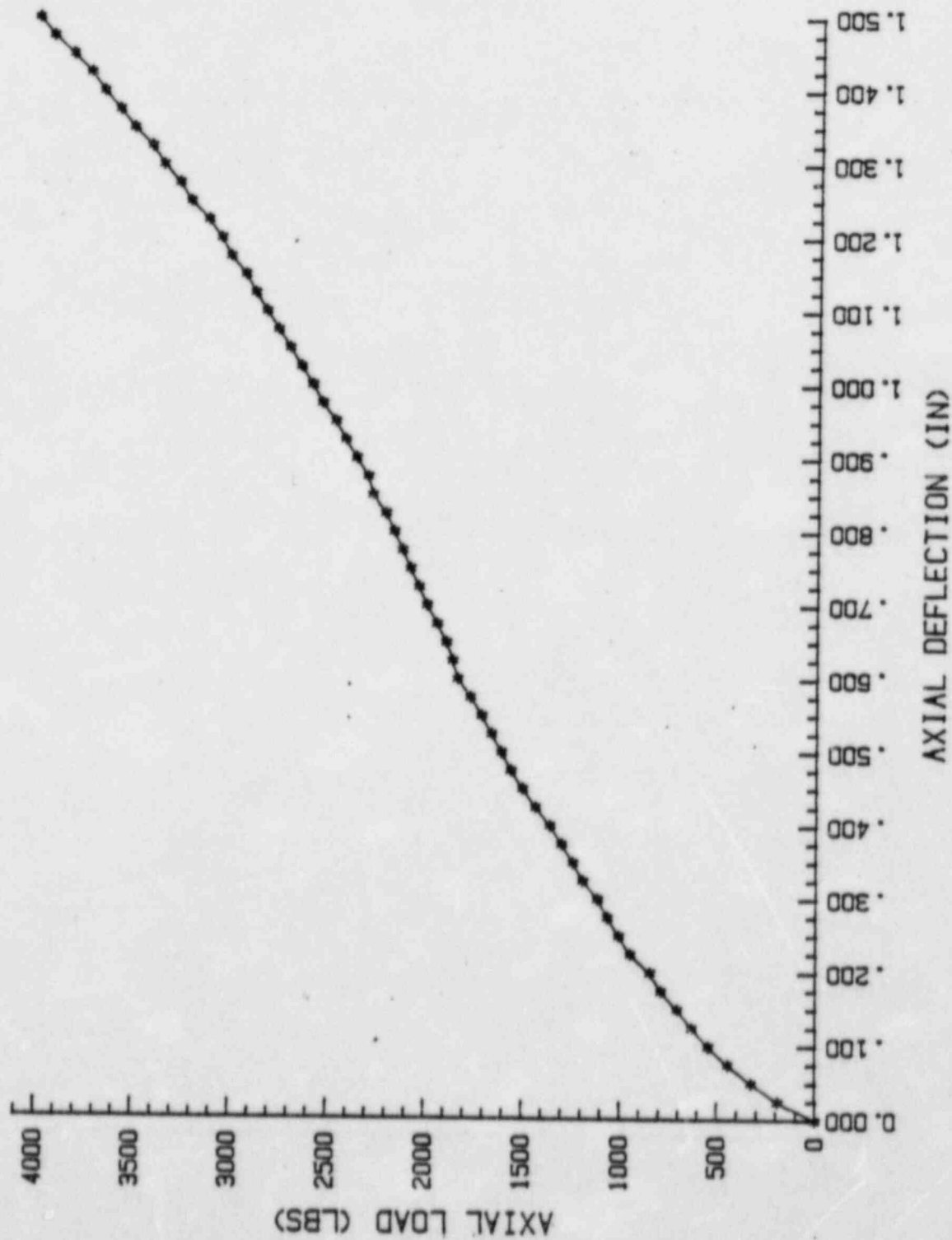


FIGURE A-16 TEST "K": AXIAL DEFLECTION VS. AXIAL LOAD

TEST "L" AXIAL LOAD - 24" PIPE

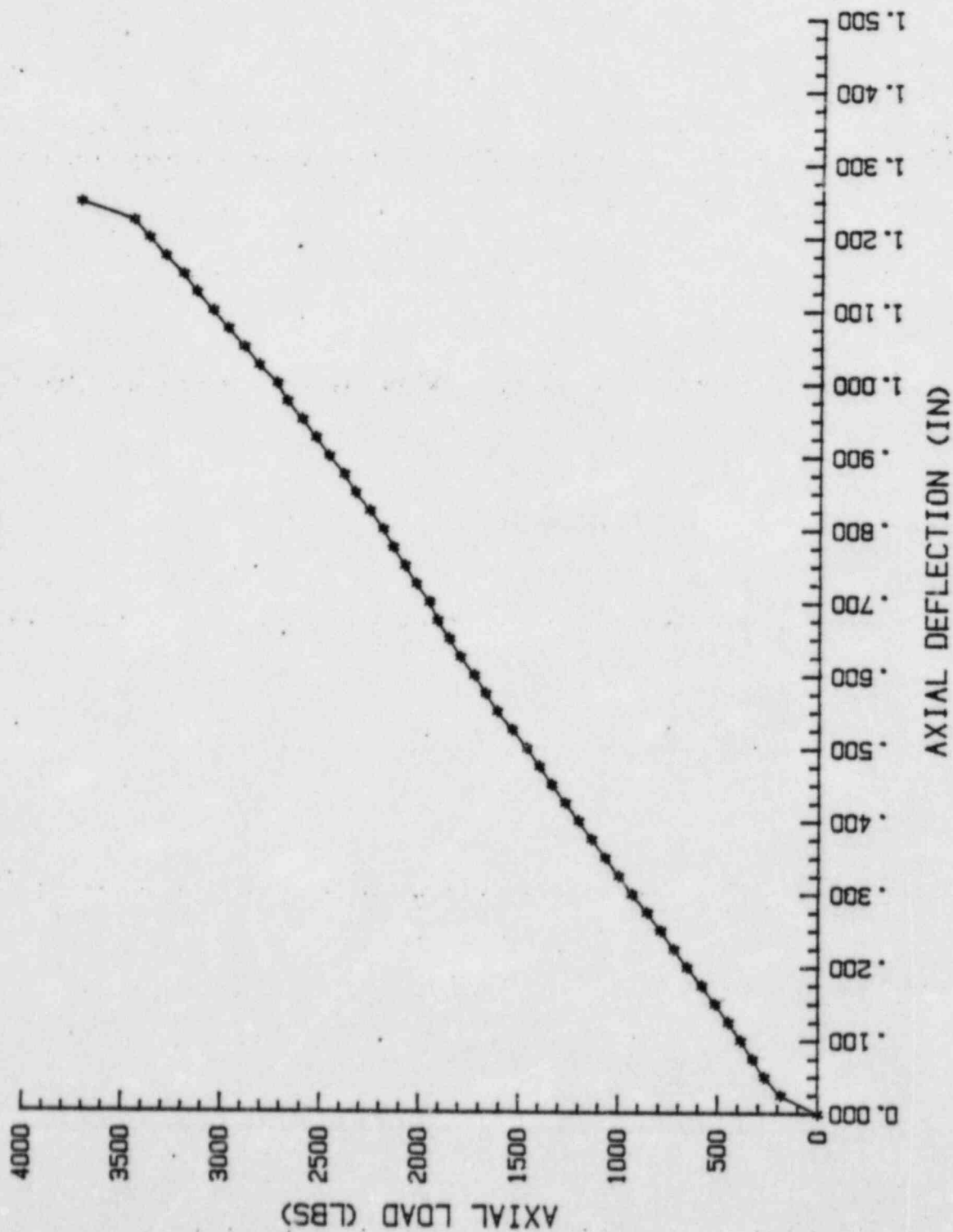


FIGURE A-17 TEST "L": AXIAL DEFLECTION VS. AXIAL LOAD

# TEST "M" AXIAL LOAD - 24" PIPE

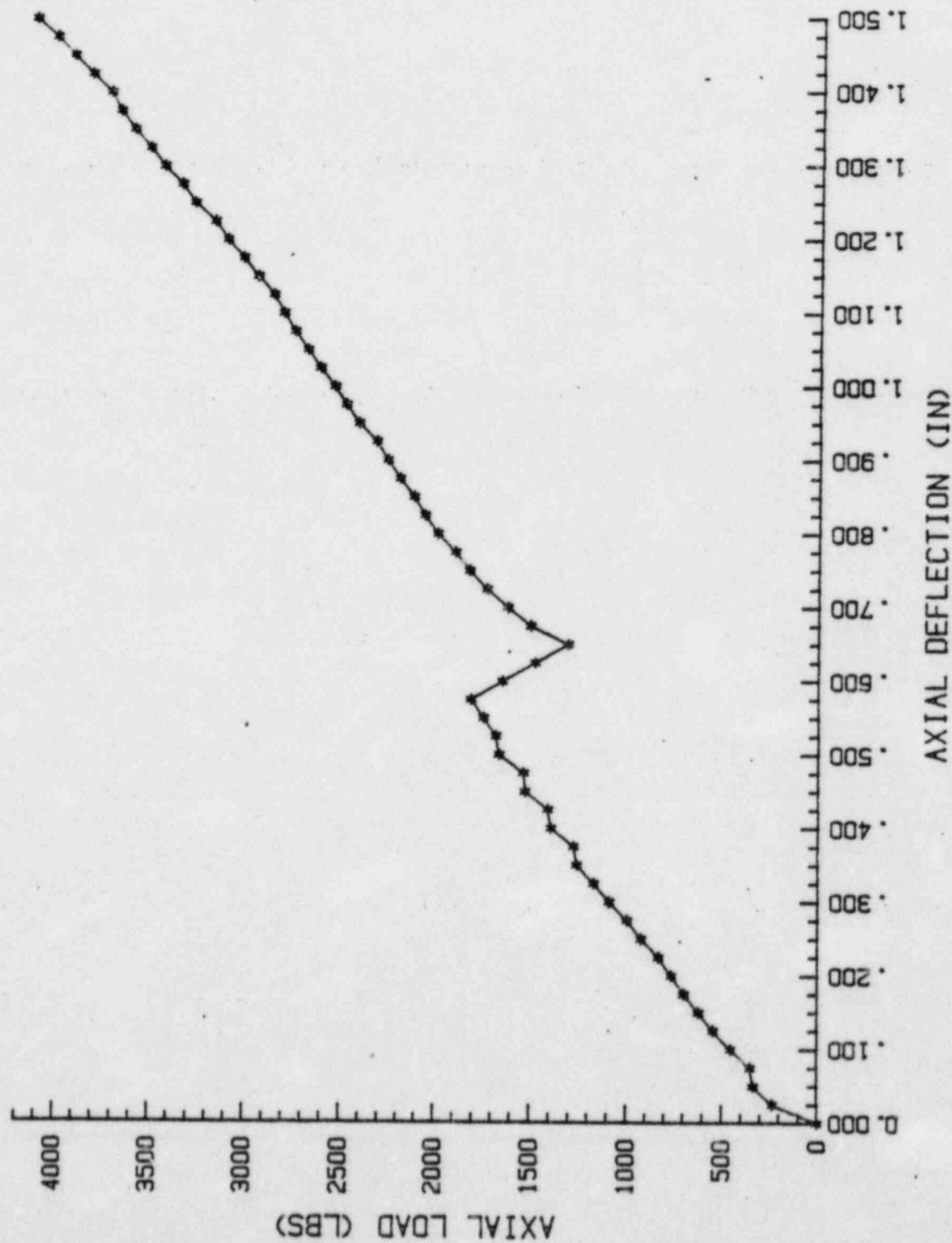
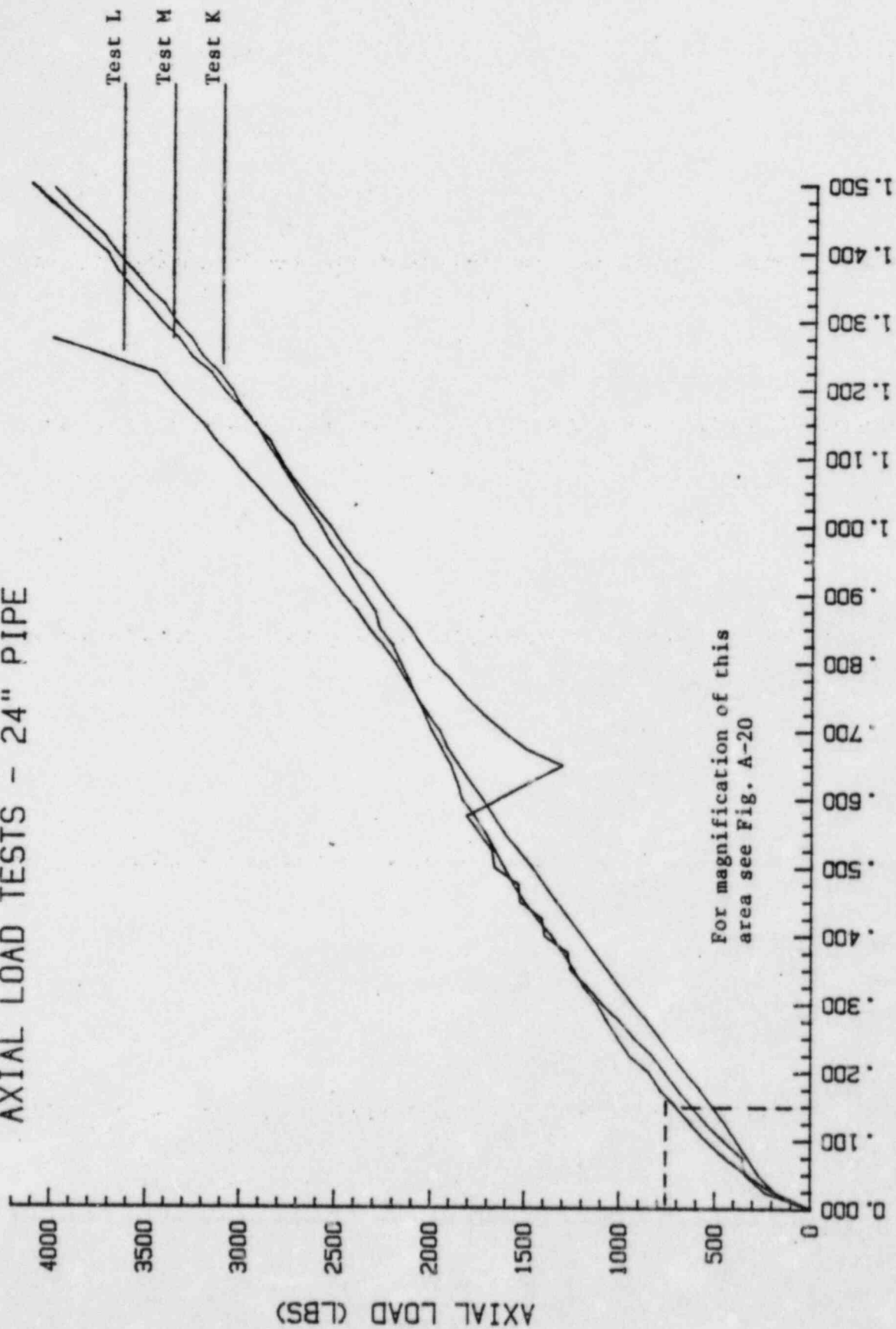


FIGURE A-18 TEST "M": AXIAL DEFLECTION VS. AXIAL LOAD

# AXIAL LOAD TESTS - 24" PIPE

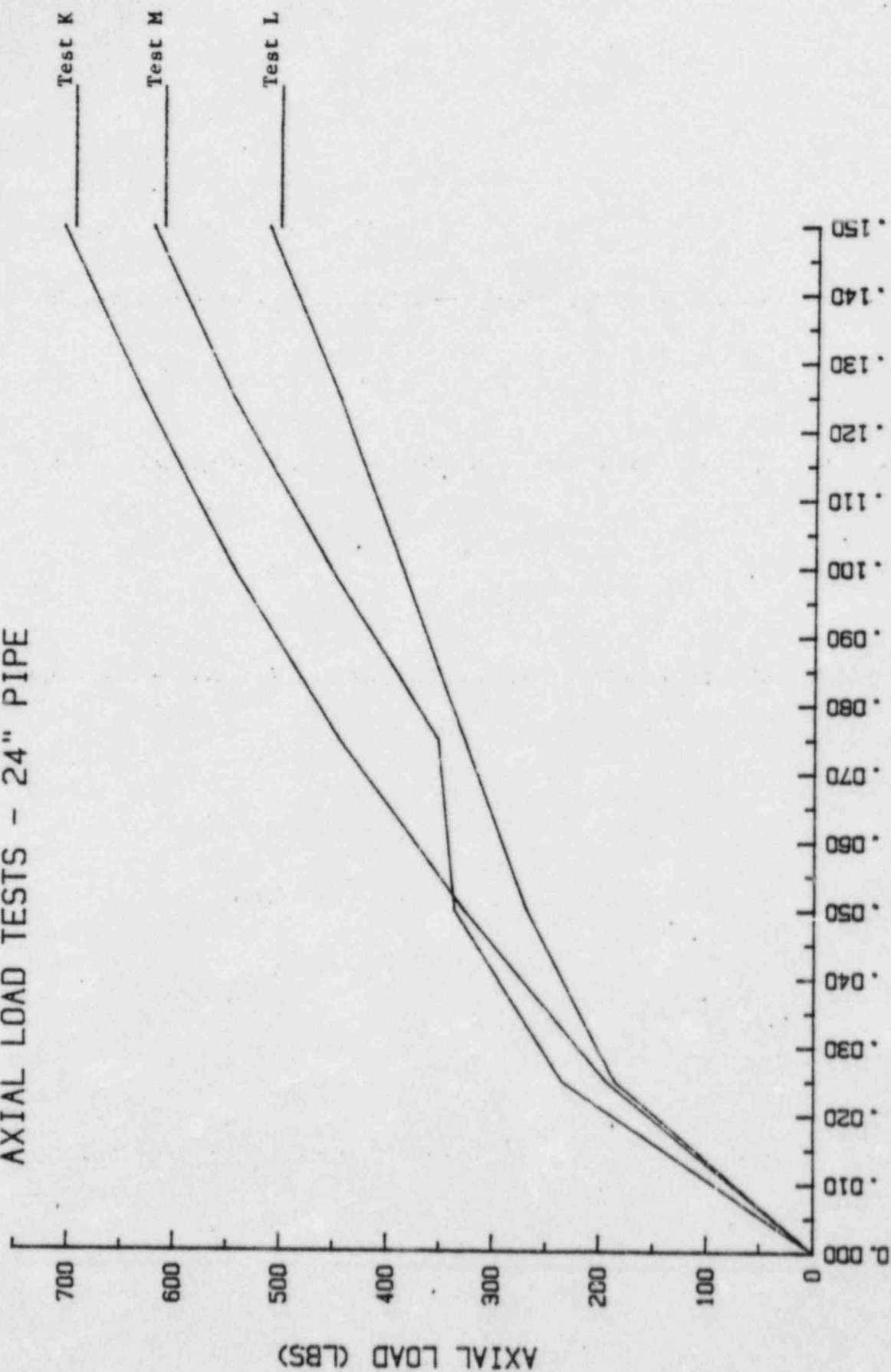


AXIAL DEFLECTION (IN)

FIGURE A-19 TEST COMPARISONS: AXIAL DEFLECTION VS. AXIAL LOAD



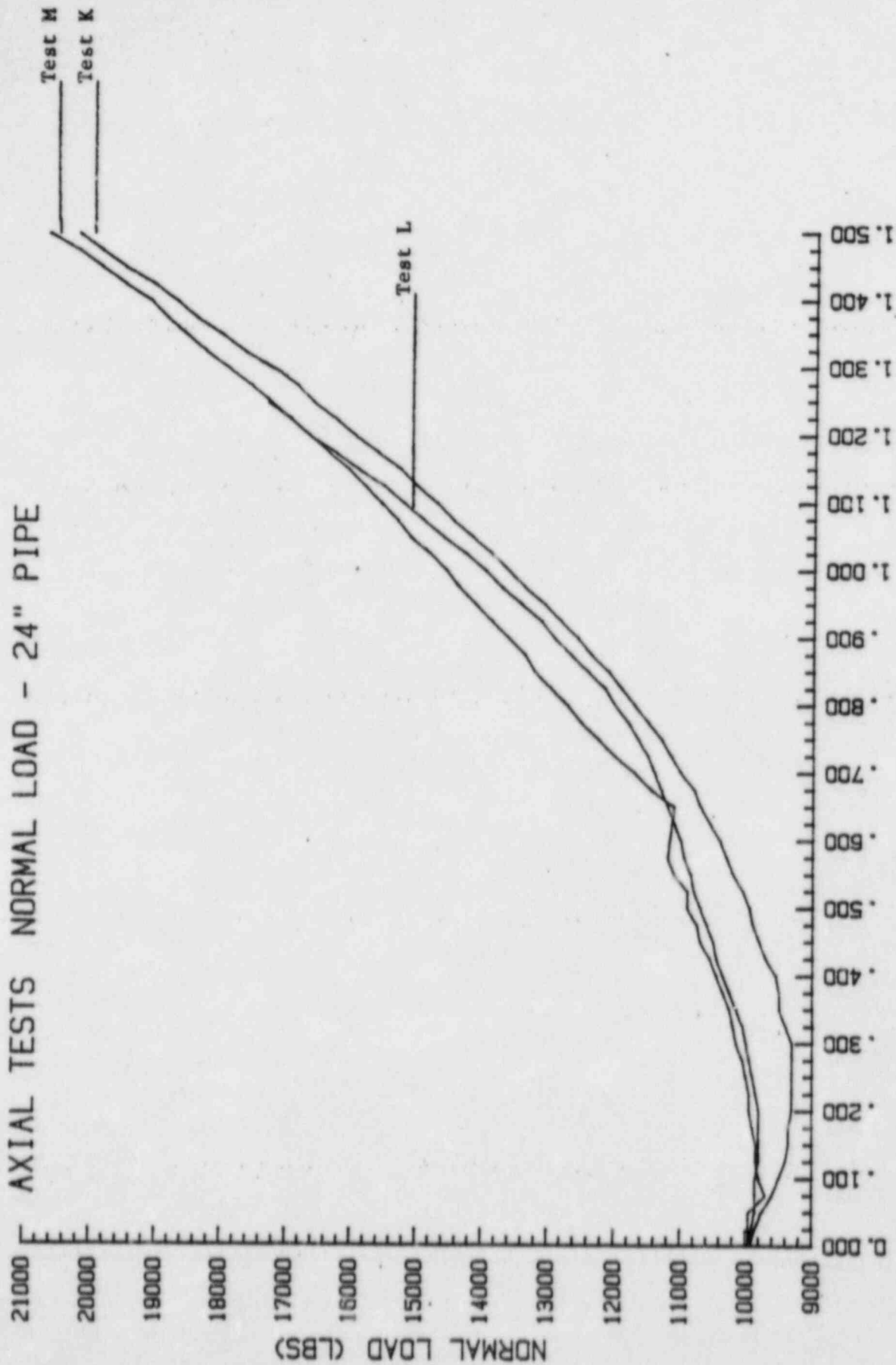
# AXIAL LOAD TESTS - 24" PIPE



AXIAL DEFLECTION (IN)

FIGURE A-20 TEST COMPARISONS: AXIAL DEFLECTIONS VS. AXIAL LOAD  
SCALE MAGNIFICATION- 0 IN. TO .15 IN.

# AXIAL TESTS NORMAL LOAD - 24" PIPE



AXIAL DEFLECTION (IN)  
 FIGURE A-21 TEST COMPARISONS: AXIAL DEFLECTION VS. NORMAL LOAD

# TEST "N" LATERAL LOAD - 24" PIPE

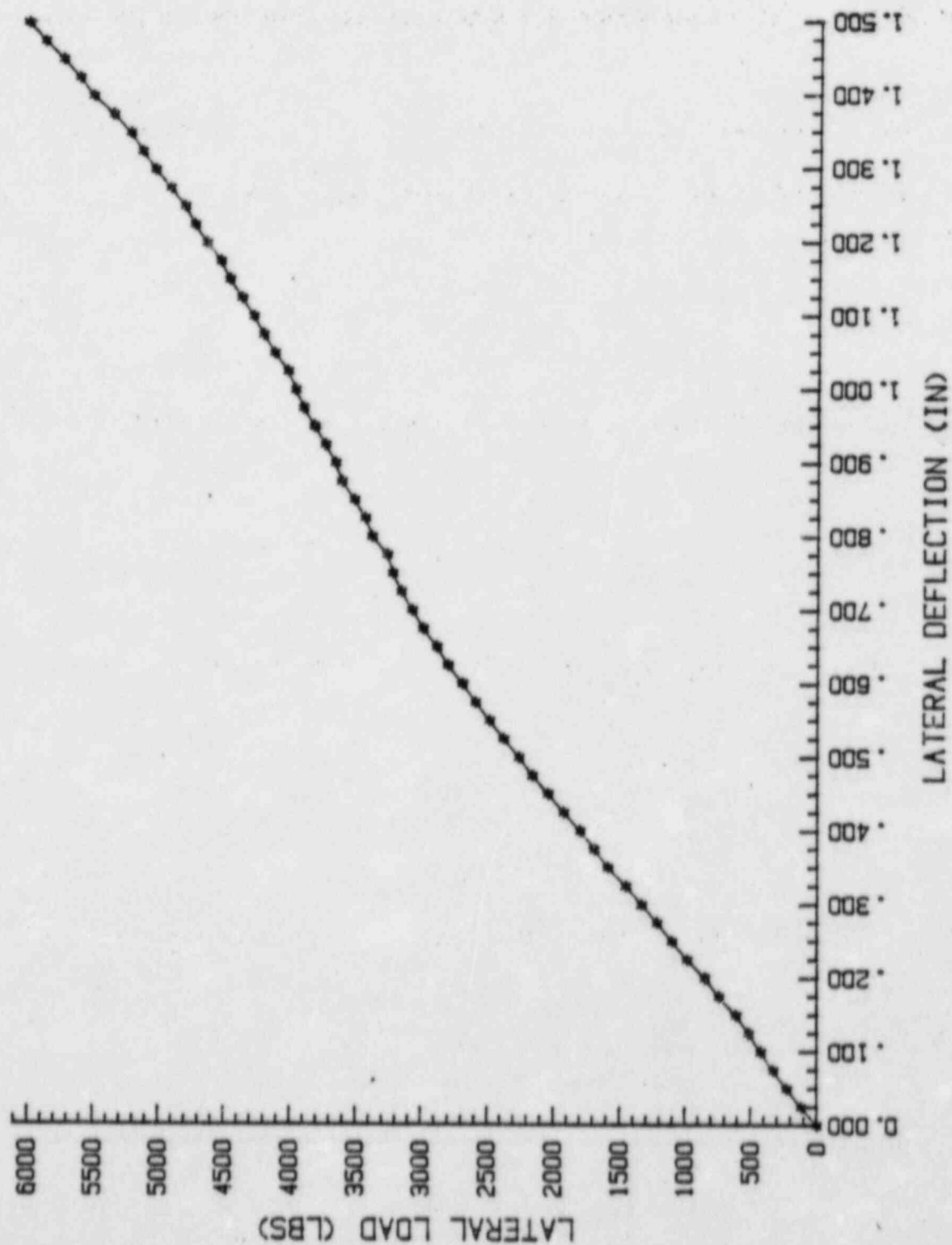


FIGURE A-22 TEST "N": LATERAL DEFLECTION VS. LATERAL LOAD

TEST "0" LATERAL LOAD - 24" PIPE

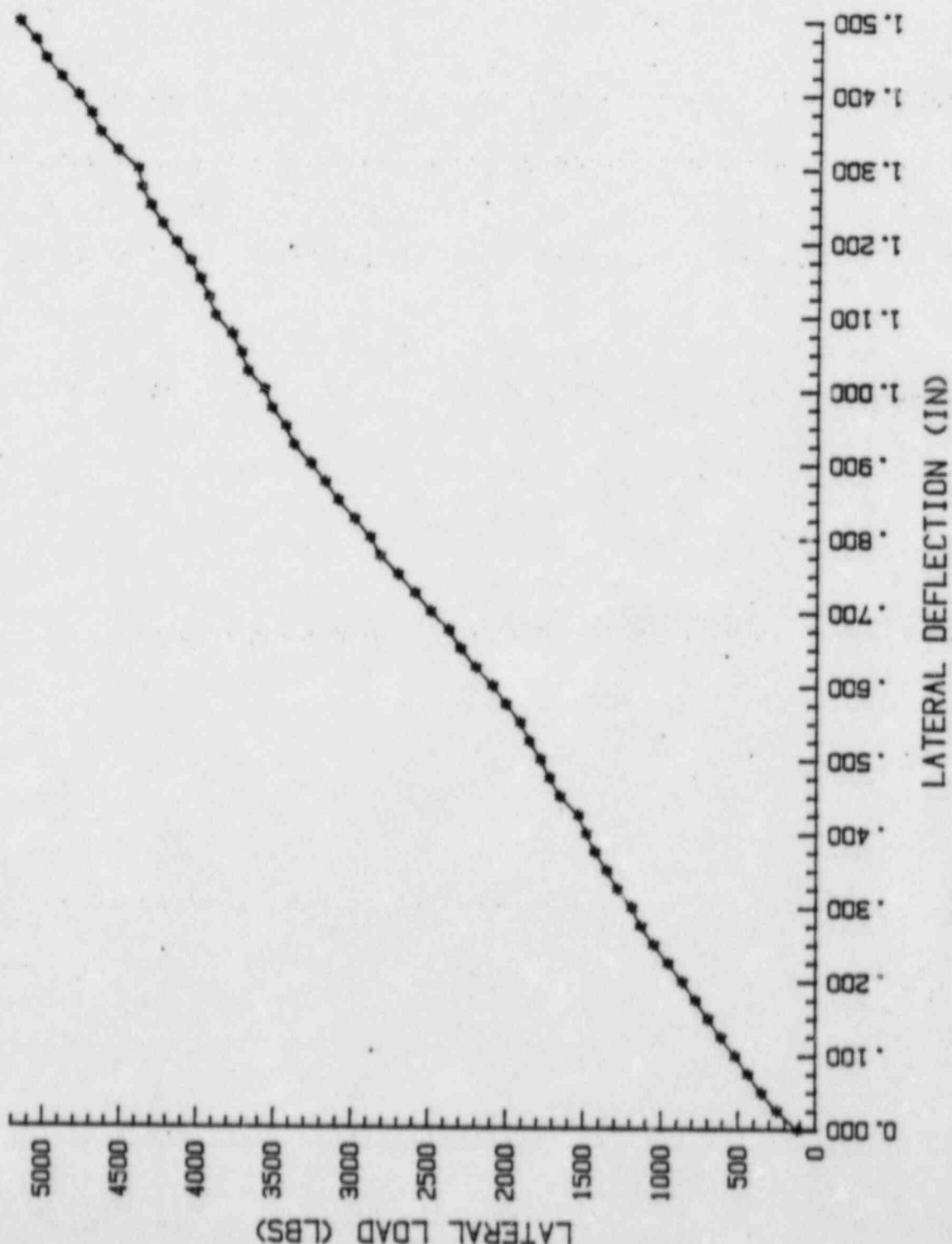


FIGURE A-23 TEST "0": LATERAL DEFLECTION VS. LATERAL LOAD

TEST "P" LATERAL LOAD - 24" PIPE

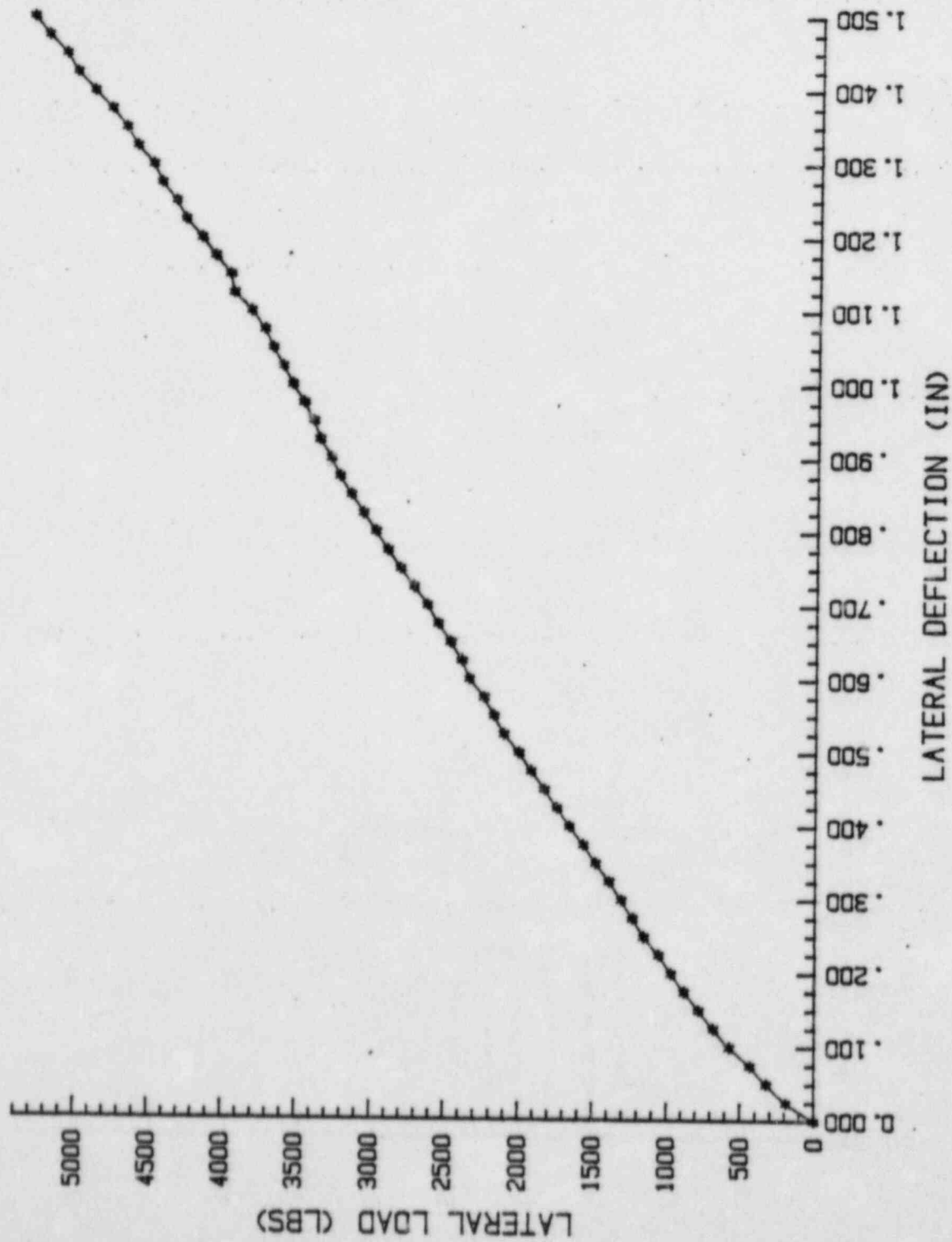


FIGURE A-24 TEST "P": LATERAL DEFLECTION VS. LATERAL LOAD



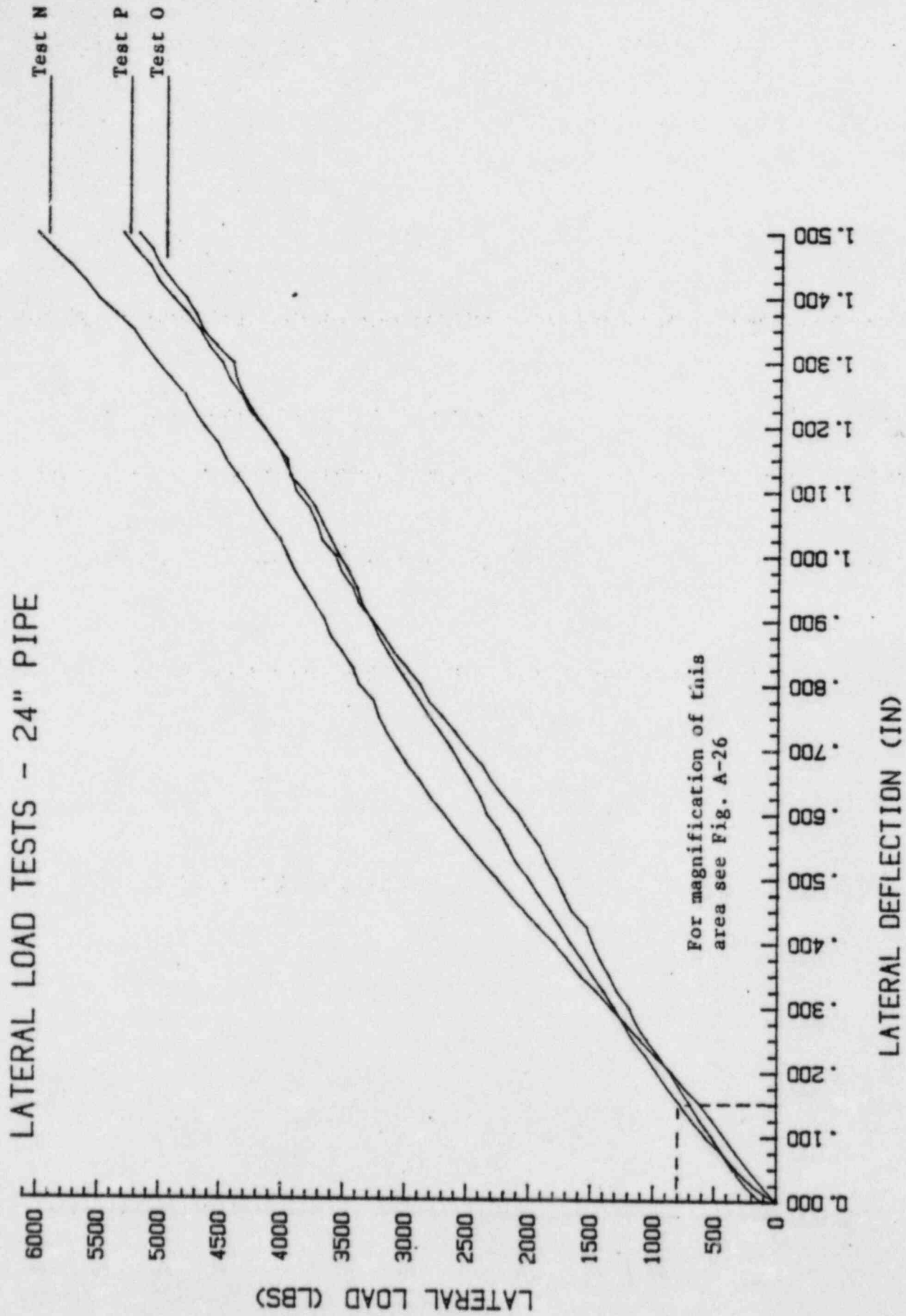
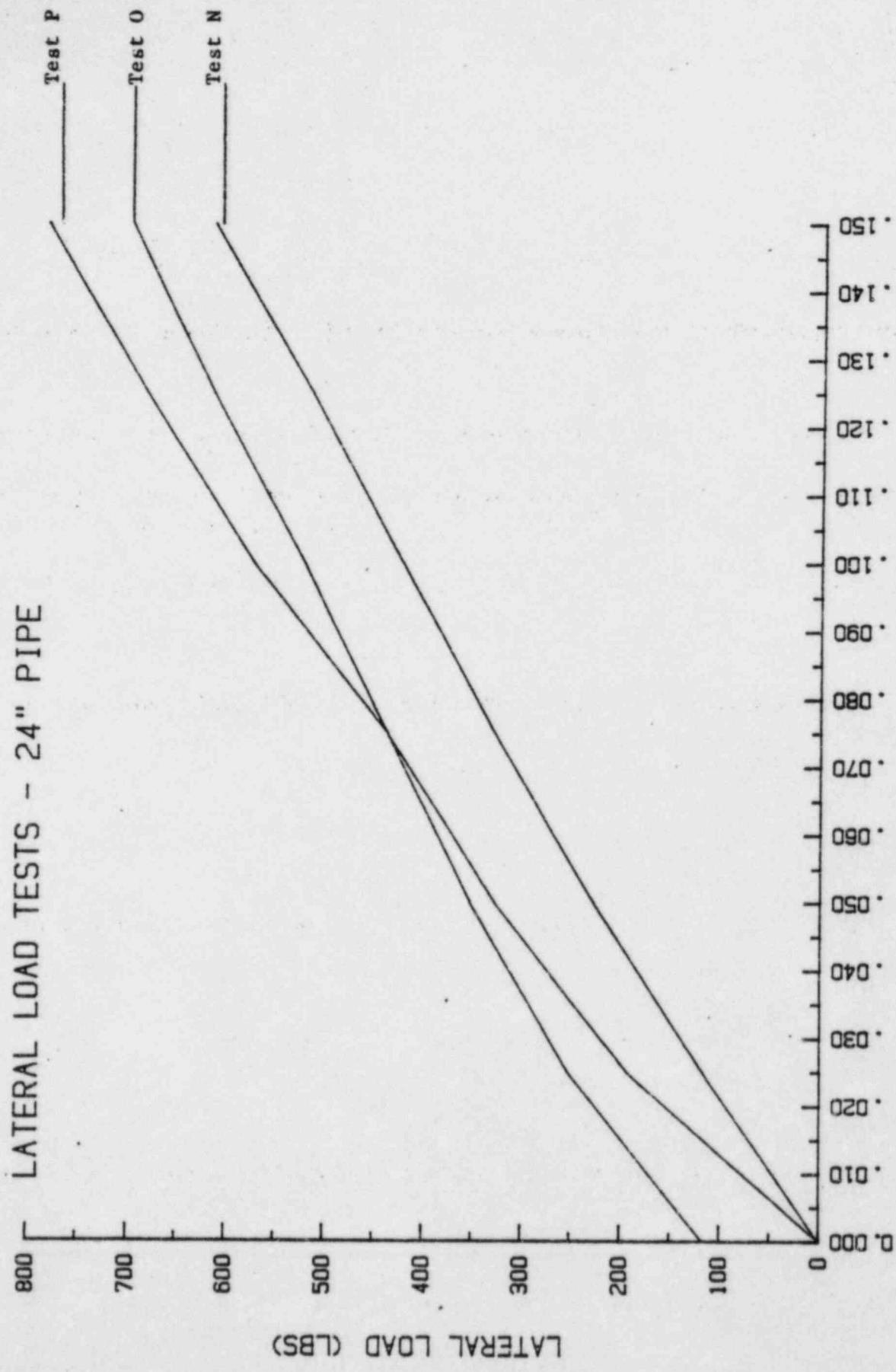


FIGURE A-25 TEST COMPARISONS: LATERAL DEFLECTION VS. LATERAL LOAD

# LATERAL LOAD TESTS - 24" PIPE



LATERAL DEFLECTION (IN)

FIGURE A-26 TEST COMPARISONS: LATERAL DEFLECTION VS. LATERAL LOAD  
SCALE MAGNIFICATION- 0 IN. TO .15 IN.

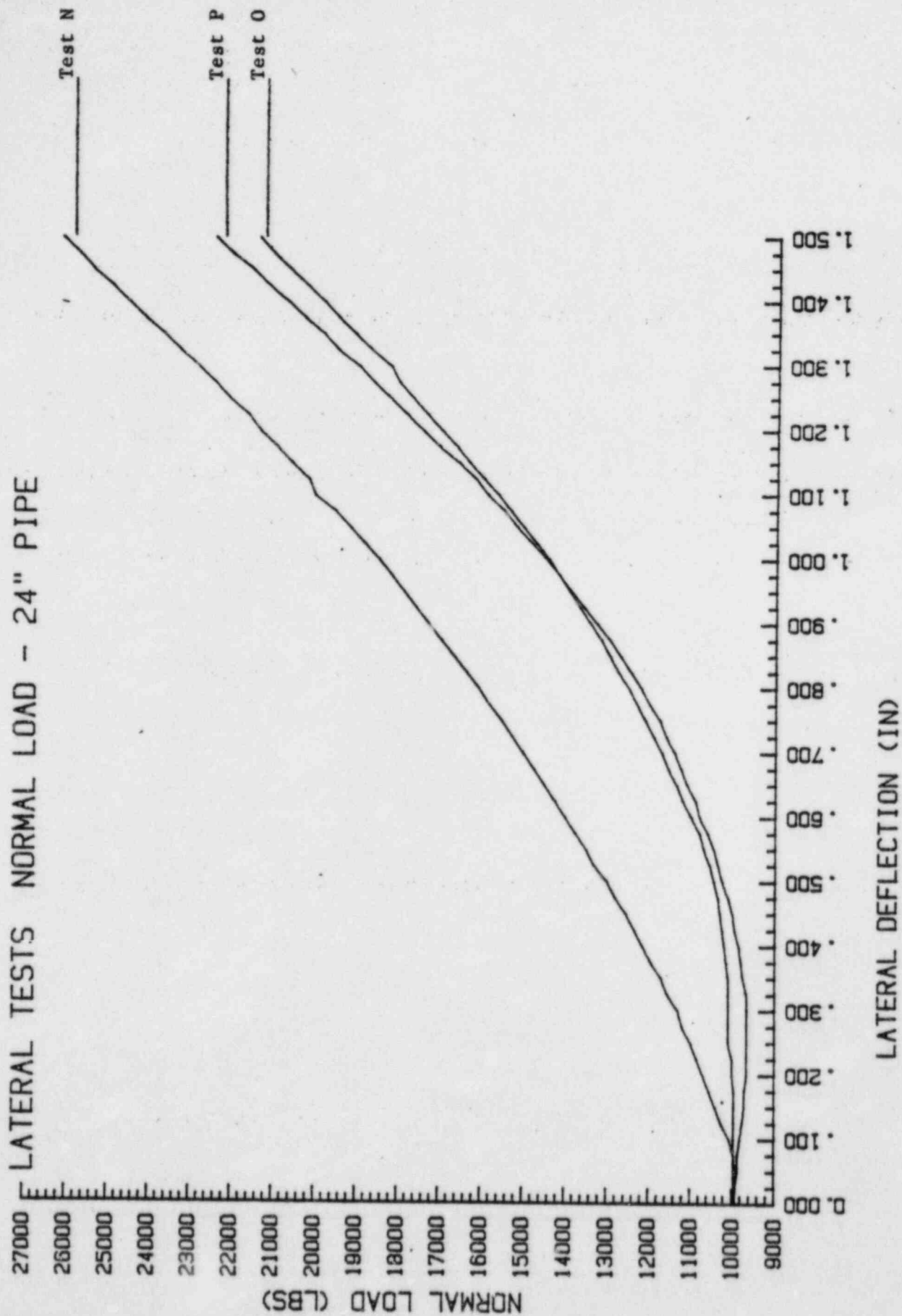


FIGURE A-27 TEST COMPARISONS: LATERAL DEFLECTION VS. NORMAL LOAD

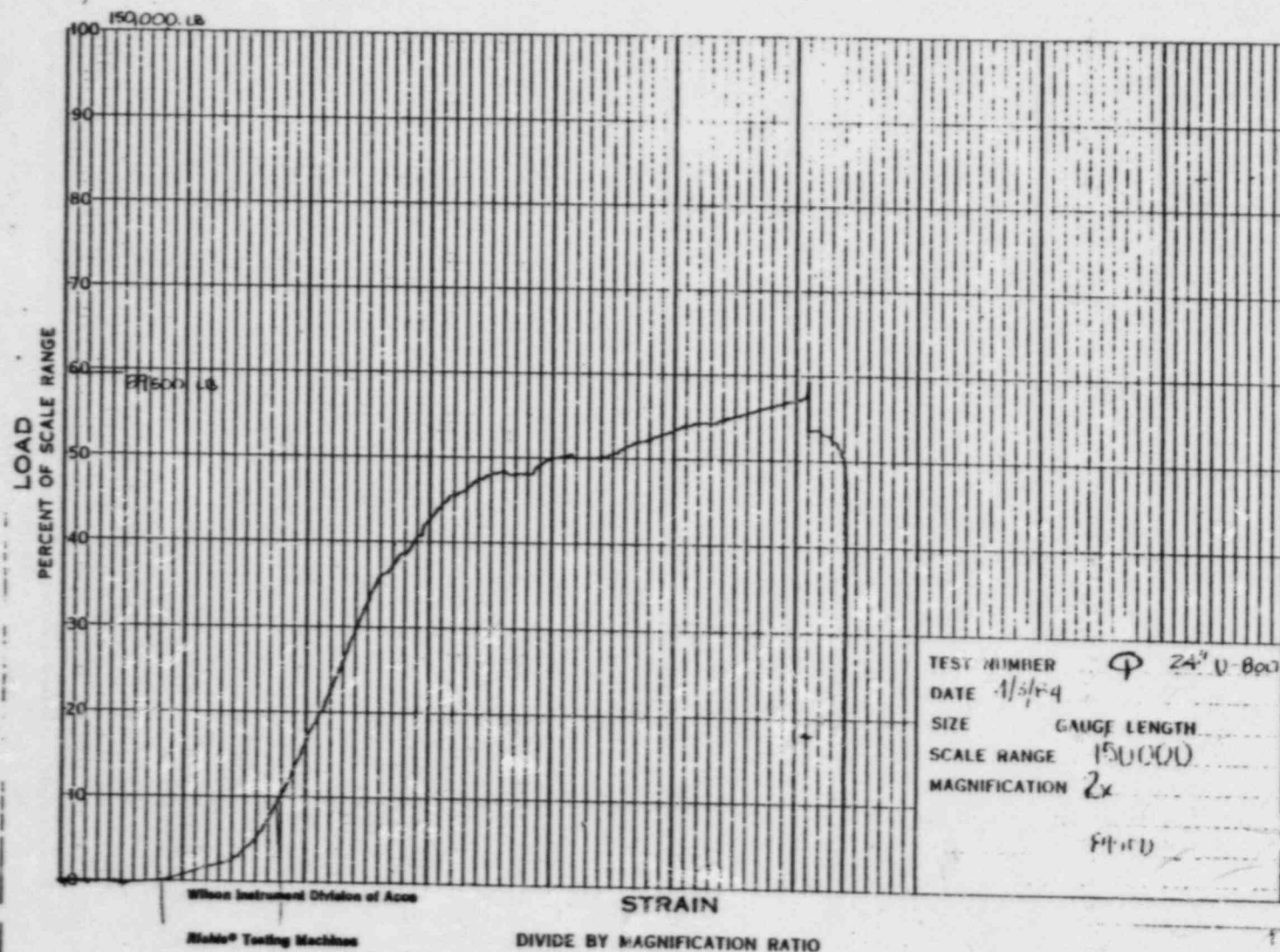


FIGURE A-28 TEST "Q" : NORMAL LOAD VS. ELONGATION

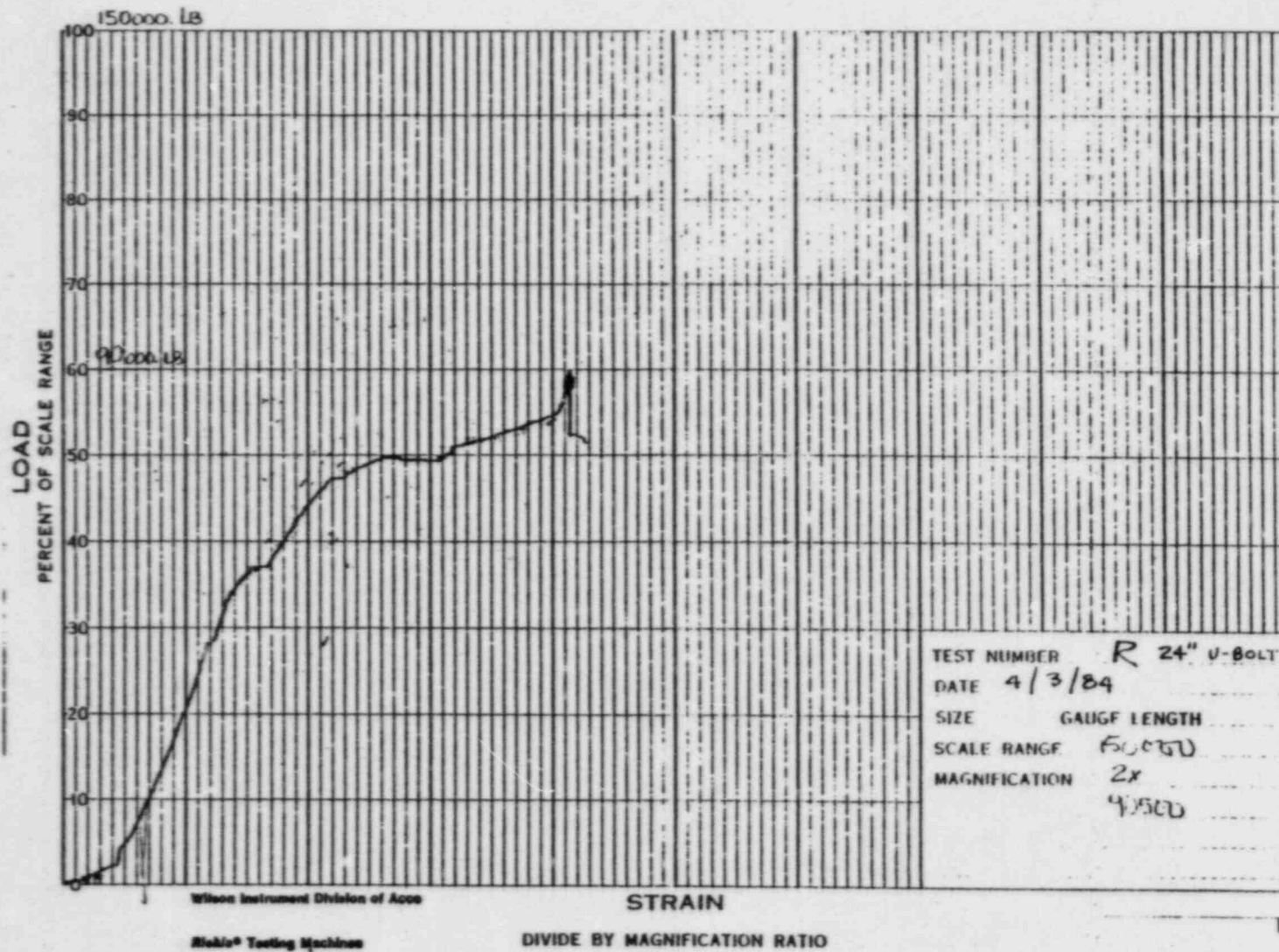


FIGURE A-29 TEST "R" : NORMAL LOAD VS. ELONGATION



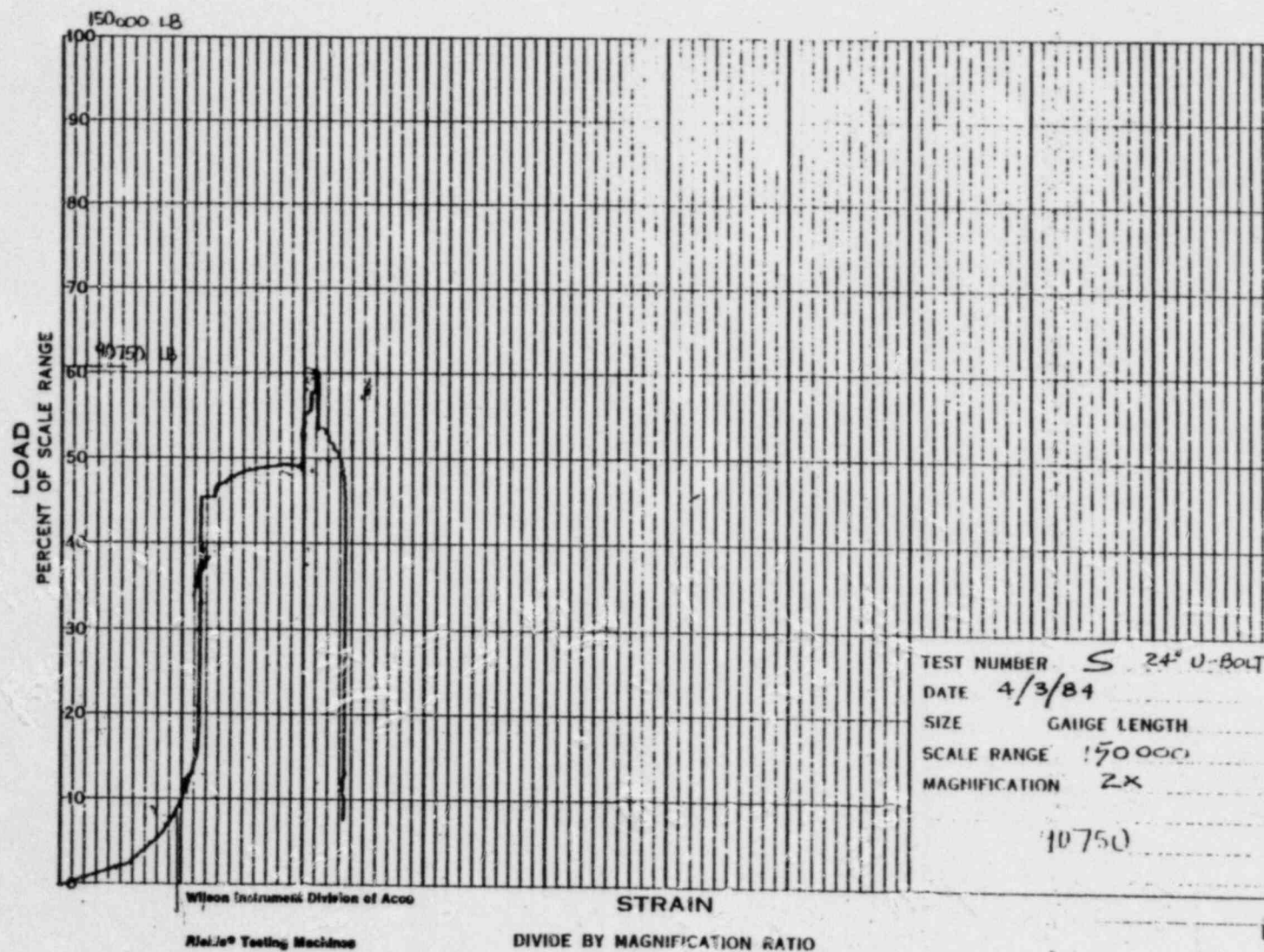
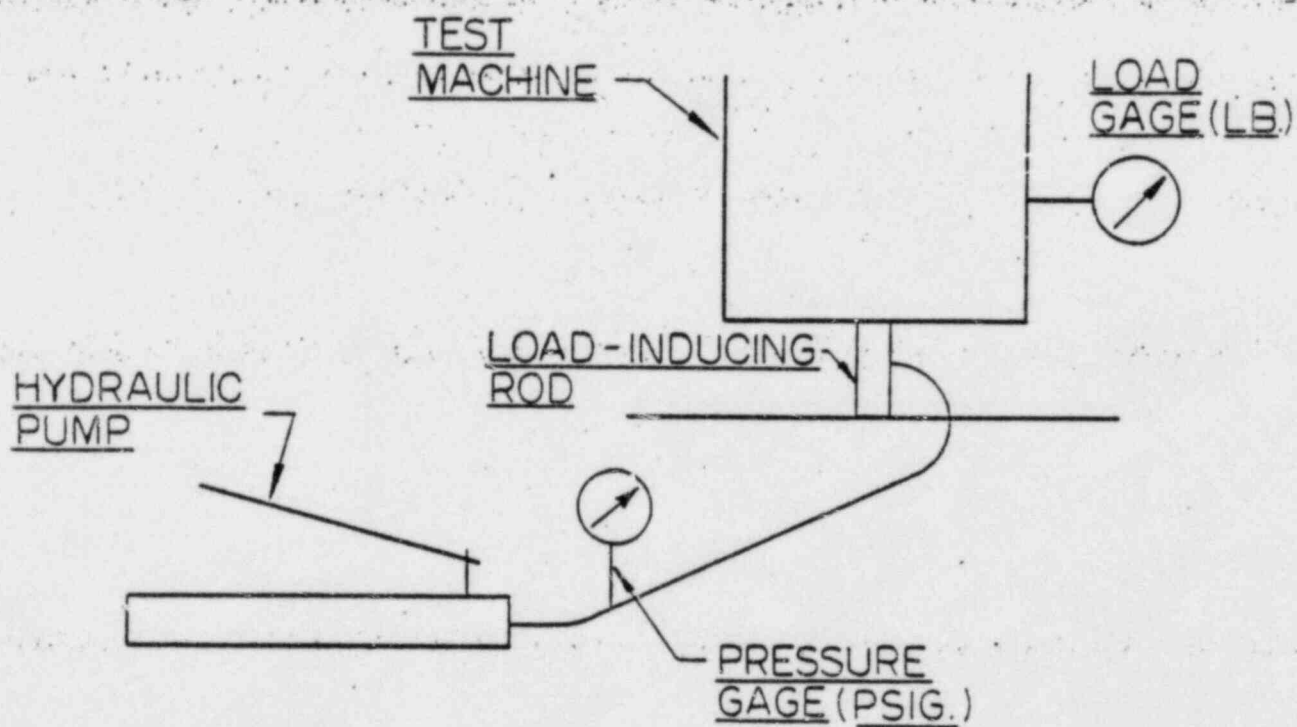


FIGURE A-30 TEST "S" : NORMAL LOAD VS. ELONGATION

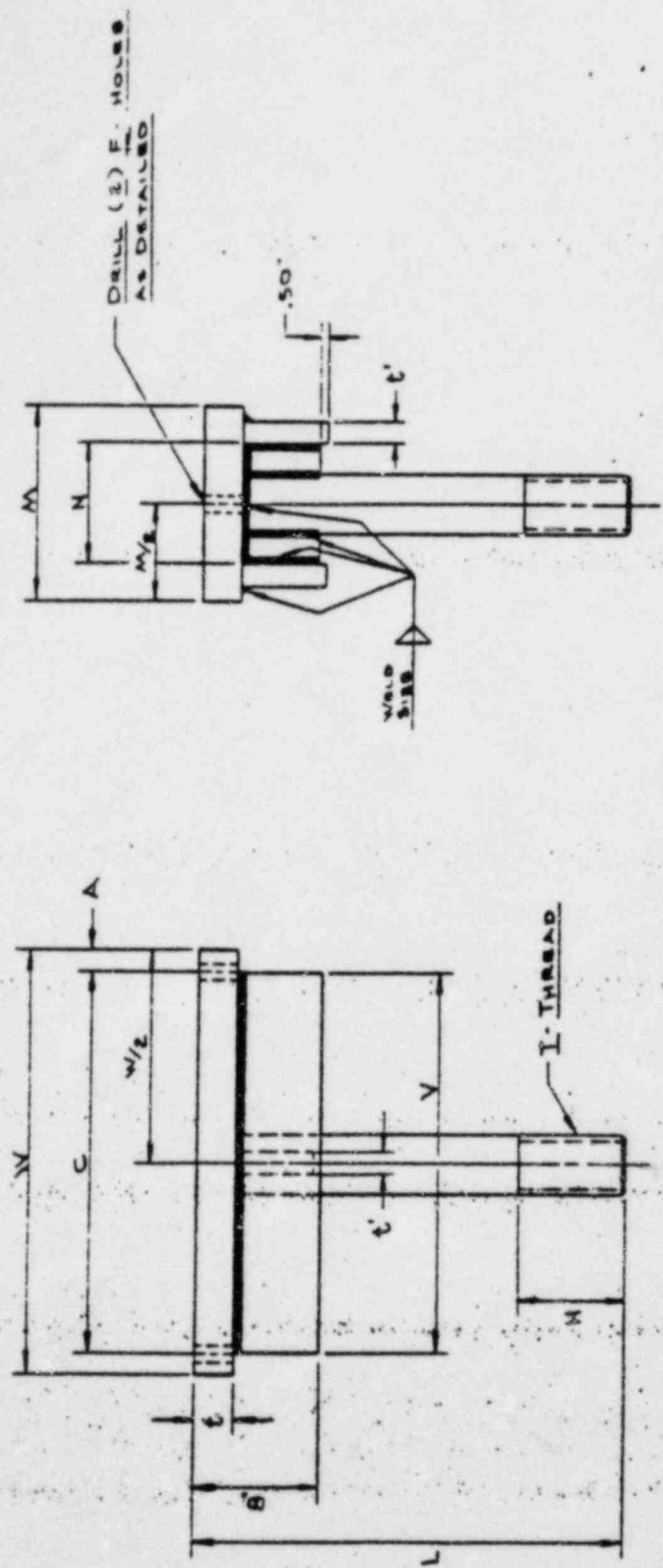
APPENDIX II

CALIBRATION OF 300 PSIG PRESSURE GAGE



APPENDIX III

TEST FIXTURE SKETCHES



PIPE SIZE	A	C	F-HOLE	H	L	M	N	t	t'	V	T-THREAD	WELD SIZE	W
4"	5.47	9.06	.96	4.00	24.00	8.00	4.50	1.00	.75	8.00	2'-8UN-2A	5/16	12.00
24"	1.44	29.12	1.06	7.00	29.00	19.00	6.00	2.00	1.50	29.00	4'-8UN-2A	3/4	28.00

SCALE: NOM. E.P.L. ~

MAT'L SPEC. ASTM A-36 OR EQUIVALENT  
 FAB. SPEC. N/A  
 WELD SPEC. N/A  
 WELD PROCEDURE N/A  
 FINISH, EXCEPT AS NOTED N/A  
 Q.A. REVIEW N/A

LIST OF PARTS  
 SUPPLEMENT

TOLERANCE EXCEPT AS NOTED OR A.S.M.  
 DATE 3-13-84  
 DRAWN E.J.S.  
 CHECKED  
 APPD

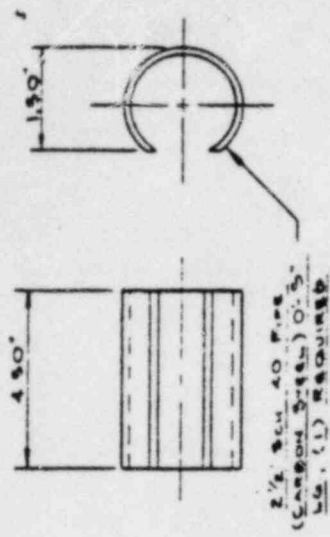
ITT Goodell Corporation  
 PIPE HANGER DIVISION  
 RESEARCH, DEVELOPMENT & ENGINEERING  
 PROVIDENCE, R.I.

B FIG. 137 SIDE LOAD TEST STAND

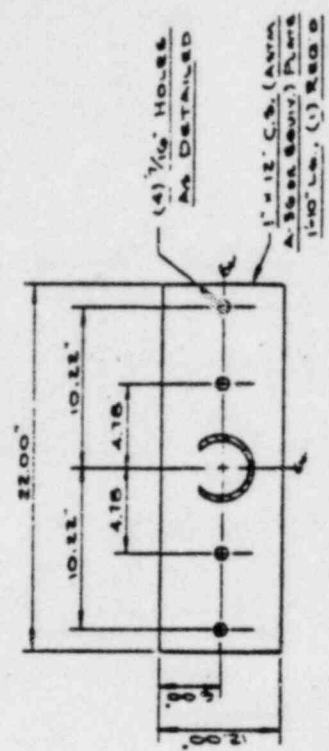
SA-4995-1

REV. -





DETAIL "A"



DETAIL "B"

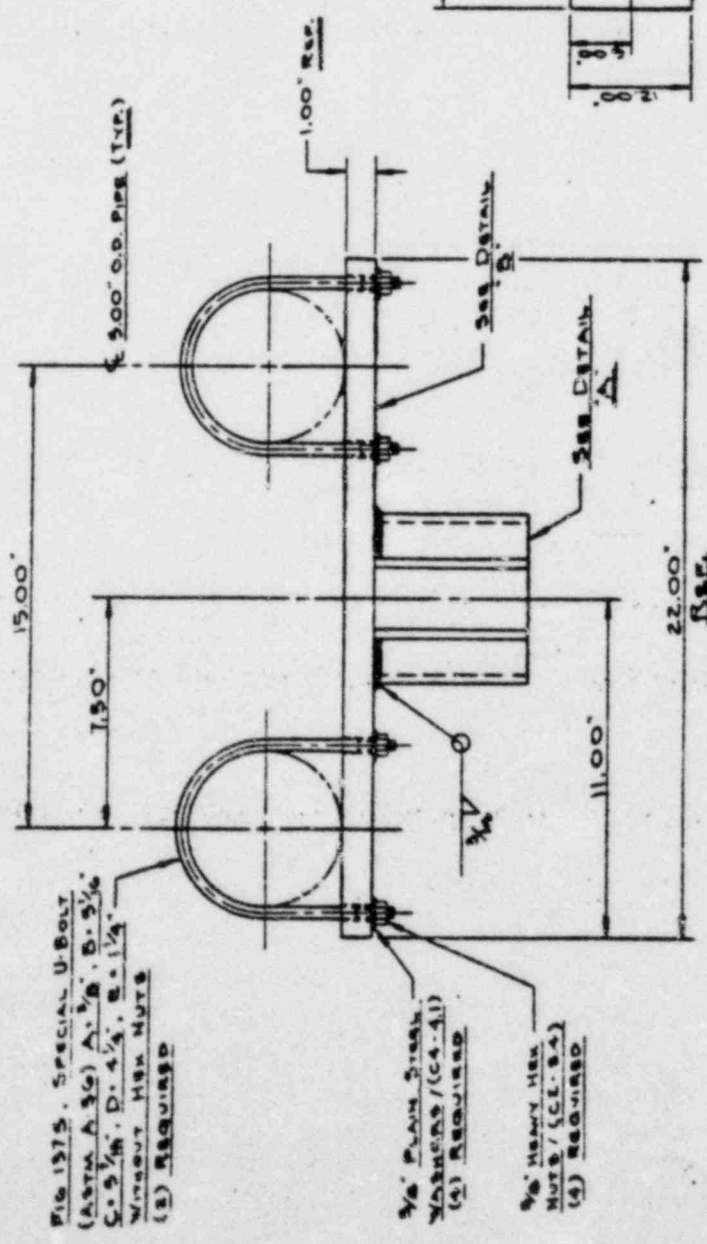


FIG. 1375. SPECIAL U-BOLT  
(ASTM A 36) A. 5/8" B. 5/16"  
C. 5/16" D. 4/16" E. 1/4"  
WITHOUT HEX NUTS  
(3) REQUIRED

SCALE: New EPL ~		LIST OF PARTS		ITT Grinnell Corporation PIPE HANGER DIVISION RESEARCH, DEVELOPMENT & ENGINEERING PROVIDENCE, R.I.	
MAT'L SPEC.	AS SHOWN	SUPERSEDES		REV. -	
FAB. SPEC.	N/A	TOLERANCE EXCEPT AS NOTED		REV. -	
WELD SPEC.	N/A	DATE 3-14-84		REV. -	
NOE PROCEDURE	N/A	DRAWN BY		REV. -	
FINISH, EXCEPT AS NOTED	N/A	CHKD		REV. -	
Q.A. REVIEW	N/A	APPD		REV. -	
		B		SA-4995-2	
		LATERAL LOAD BASE SUPPORT			

REV.

DESCRIPTION

DCN. BY APPD.

DATE

REV.

DESCRIPTION

DCN. BY APPD.

DATE

REV.

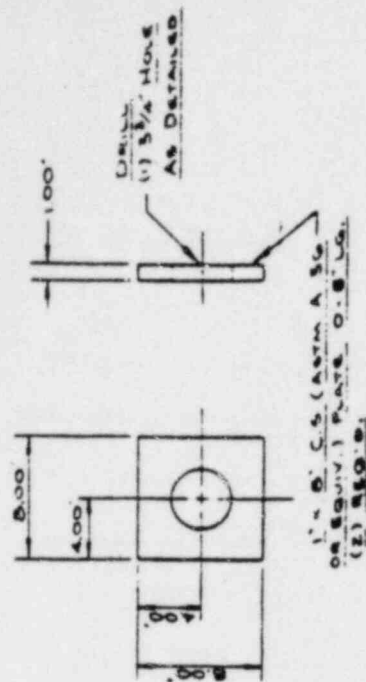
DESCRIPTION

DCN. BY APPD.

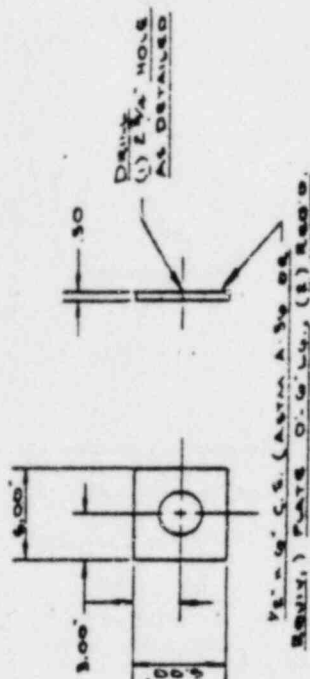
DATE

REV.

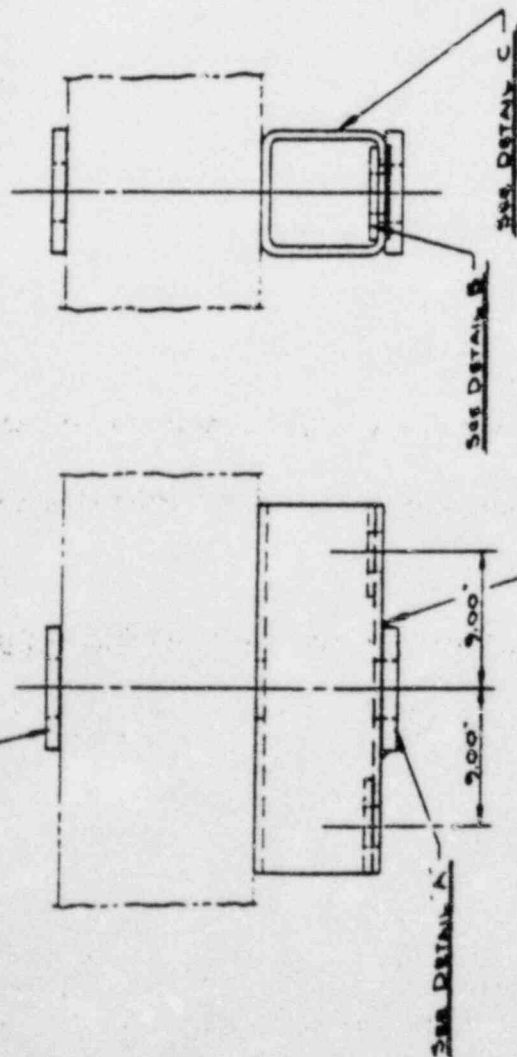
DESCRIPTION



DETAIL A



DETAIL B

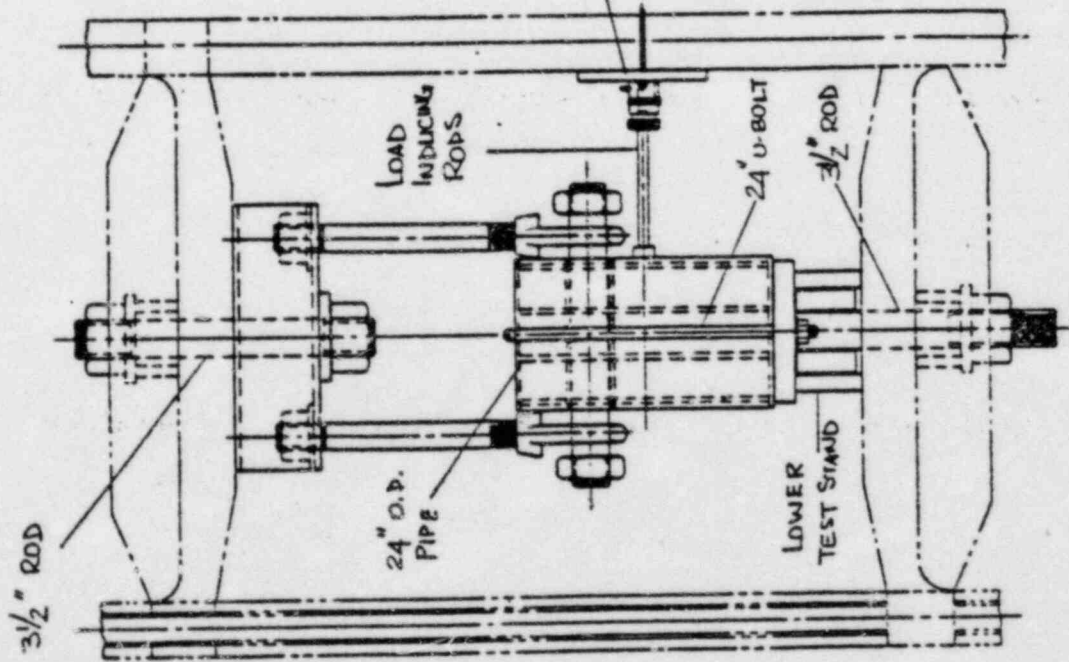


DETAIL C

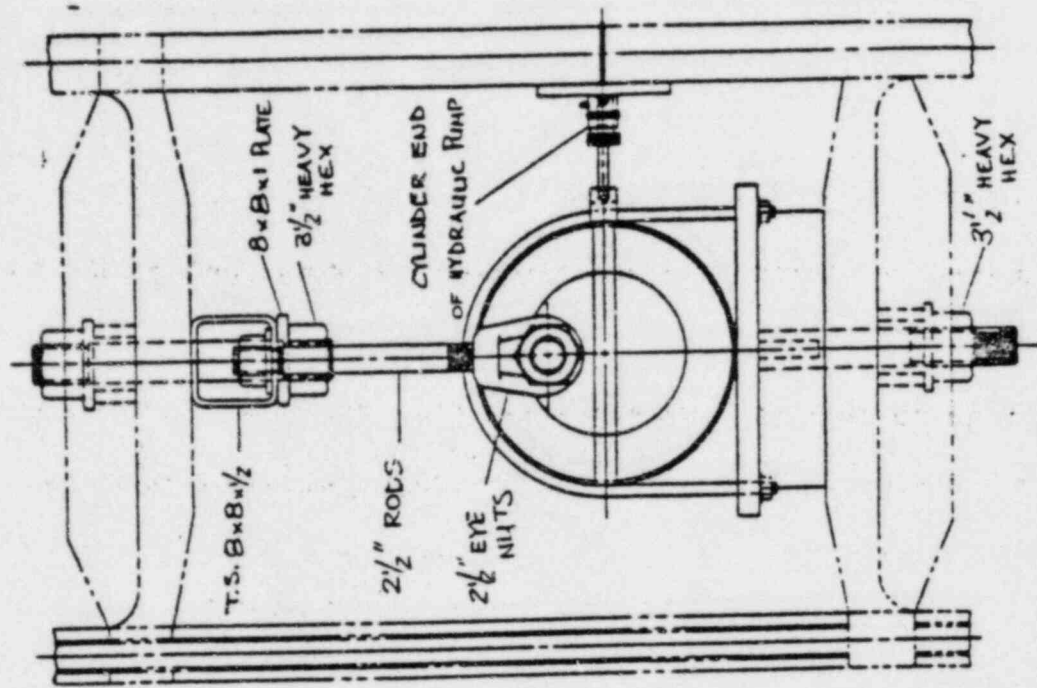
SCALE: NONE E.P.L. ~		LIST OF PARTS		ITT Grinnell Corporation PIPE HANGER DIVISION RESEARCH, DEVELOPMENT & ENGINEERING PROVIDENCE, R.I.	
		SUPERIMPOSE			
MAT'L SPEC.	AS SHOWN	TOLERANCE EXCEPT AS NOTED	DATE 3-14-84	B	UPPER TEST FUTURE
FAB. SPEC.	N/A				
WELD SPEC.	N/A	PRAC. CALCDN DEC. ANG.	DRAWN: EIS CHK'D:		
NOE PROCEDURE	N/A				
FINISH, EXCEPT AS NOTED	N/A				
O.A. REVIEW	N/A			SA-4995-3	
				REV. -	



DETAIL 'A'



AXIAL LOADING

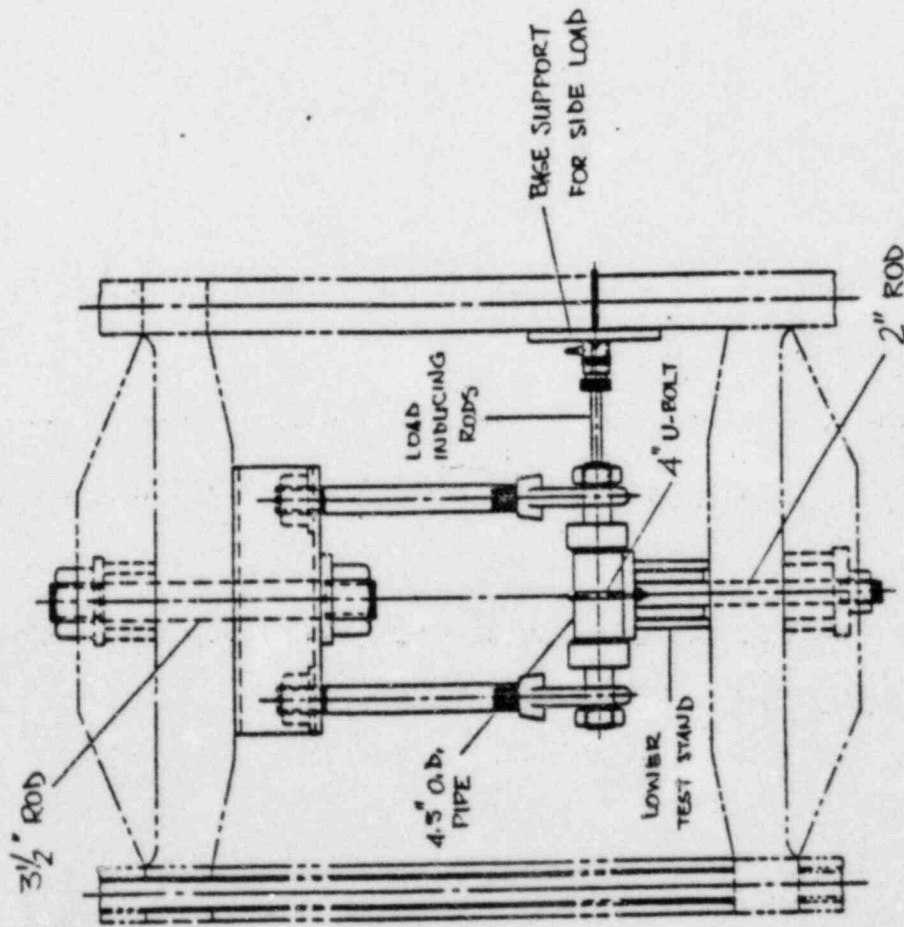


SIDE LOADING

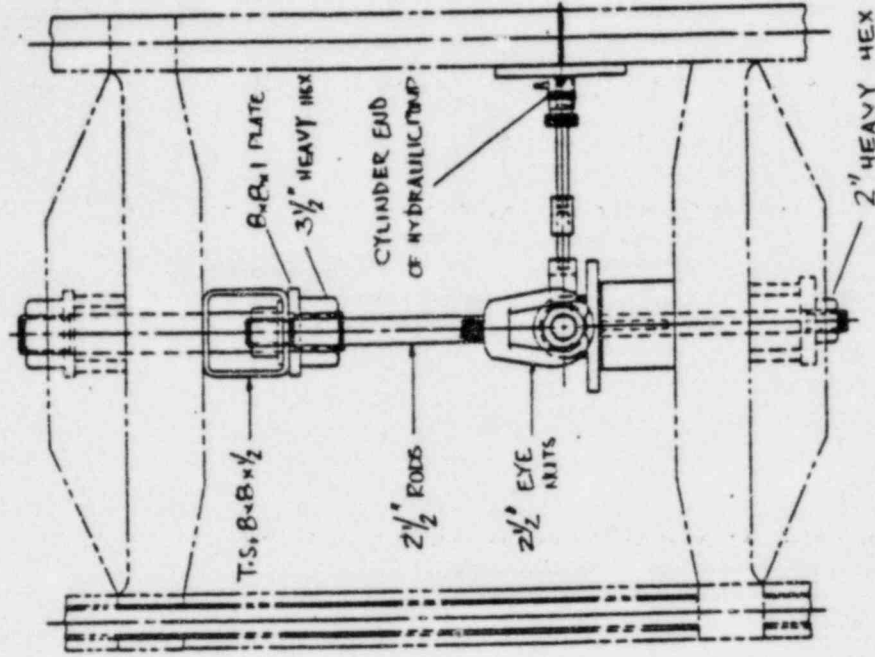
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MATERIAL SPEC.		DESIGN SPEC.	DATE: 11/1/64	
FAB. SPEC.		WELD SPEC.	DATE: 11/1/64	
TEST PROCEDURE		FINISH SPEC.	DATE: 11/1/64	
G.A. REVIEW		DATE: 11/1/64		

ITEM: 4995-4		REV. 1	
DATE: 11/1/64		DATE: 11/1/64	
BY: J. J. JONES		BY: J. J. JONES	
CHECKED: J. J. JONES		CHECKED: J. J. JONES	
APPROVED: J. J. JONES		APPROVED: J. J. JONES	



AXIAL LOADING



SIDE LOADING

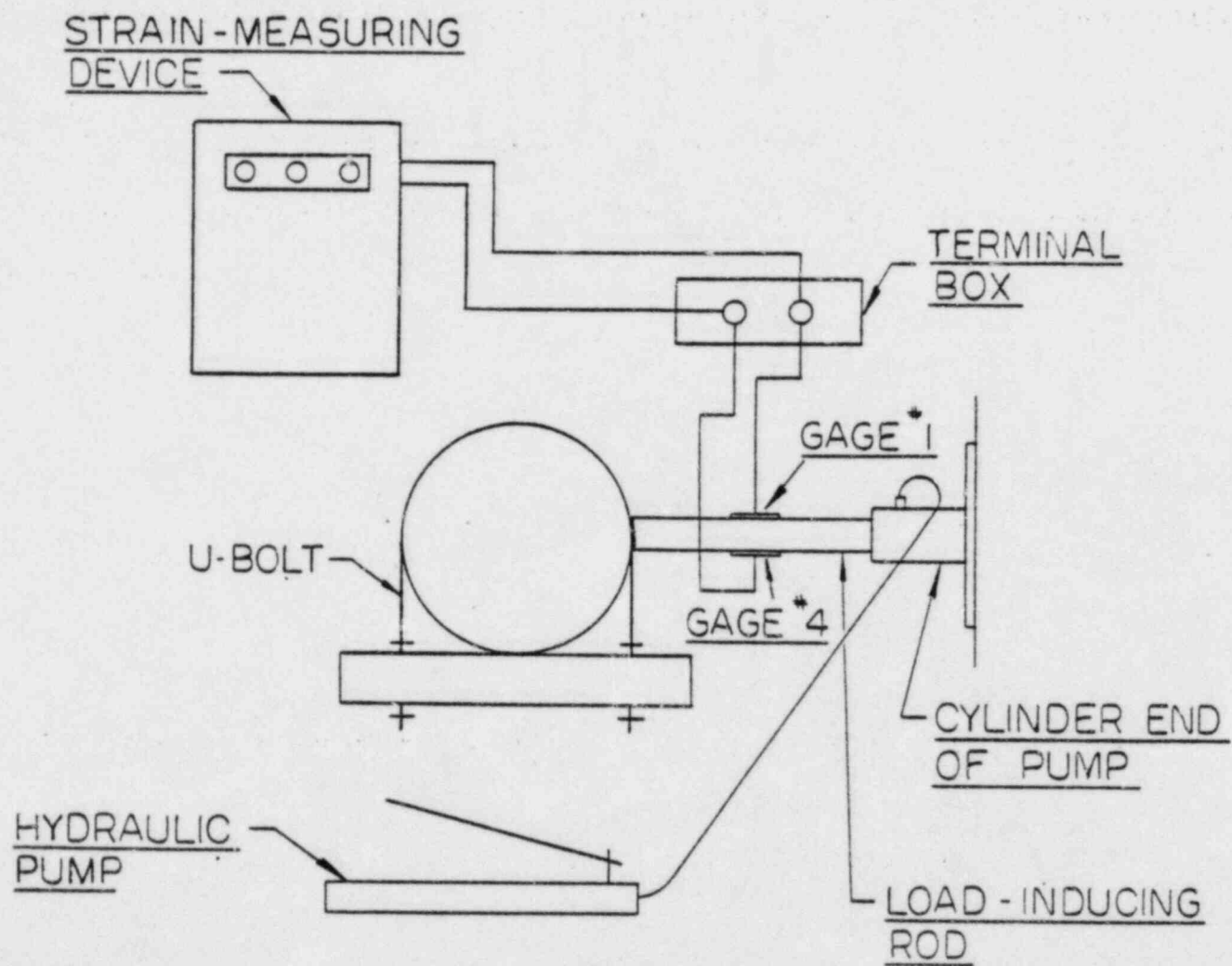
SCALE: 1/4" = 1' - 0"		LIST OF PARTS		ITC Standard Equipment	
MATERIAL SPEC.		EXPLANATION		NOT SUBJECT TO CHANGE	
FAB. SPEC.		DATE		REVISIONS	
WELD SPEC.		5-20-84		C 330' OF U.S.A. 7397	
NOT PROCEDURE		FACILITY		SA-4995-5	
FINISH EXCEPT AS NOTED		DRAWN BY		REV	
QA REVIEW		DATE		BY	



APPENDIX IV

MEASUREMENT OF STRAINS IN LOAD-INDUCING RODS





APPENDIX V

STRAIN TO LOAD CONVERSION

The stress-strain relationship within a material's elastic range is given by;

$$\sigma = E \epsilon$$

where  $\sigma$  = stress in PSI

$\epsilon$  = strain in IN/IN

E = Young's Modulus in PSI

Since the gages are placed diametrically opposite on the 0.875 in. diameter rod whose material Young's modulus is 27900000 PSI, then,

$$\sigma = (27900000) \epsilon$$

$$\text{but } \sigma = \frac{P}{A}$$

$$\text{therefore } \frac{P}{A} = (27900000) \epsilon$$

$$\text{or } P = (27900000) A \epsilon$$

$$P = (27900000) \frac{\pi}{4} (.875)^2$$

$$P = (16780000) \epsilon$$

$$P = 16.78 \epsilon' \text{ LBS}$$

$$\text{where } \epsilon' = \text{strain in IN/IN} \times 10^{-6}$$

APPENDIX VI

PHOTOGRAPHS OF TEST

LIST OF PHOTOGRAPHS

- Photo 1 : General Test Set-up for Axial Load on 4" Pipe U-Bolt  
2 : Hydraulic Pump and Cylinder Arrangement  
3 : Strain Gages on Load-Inducing Rods  
4 : Typical Axial Deflection and Dial Indicator  
5 : Permanently Deflected Shape
- Photo 6 : General Test Set-up for Lateral Load on 4" Pipe U-Bolt  
7 : Load-Inducing Rods and Dial Indicator  
8 : Lift-up of Hex Nut due to Excessive Deformation  
9 : Pronounced Deformation of U-Bolt  
10 : Permanently Deflected Shape
- Photo 11 : Normal Load Set-up for 4" Pipe U-Bolt  
12 : Lift-up of Hex Nut due to Excessive Deformation  
13 : Failure of U-Bolt in the Tensile Mode  
14 : 'Necking' of Threaded Section of U-Bolt
- Photo 15 : General Test Set-up for Axial Load on 24" Pipe U-Bolt  
16 : Axially Deflected U-Bolt  
17 : Axially Deflected U-Bolt  
18 : Axially Deflected U-Bolt
- Photo 19 : Lateral Load on 24" Pipe U-Bolt General Deflected Shape  
20 : Lateral Load on 24" Pipe U-Bolt General Deflected Shape  
21 : Lateral Load on 24" Pipe U-Bolt General Deflected Shape  
22 : Detail of U-Bolt and Hex Nut at Plate Interface
- Photo 23 : Eminent Failure of U-Bolt under Normal Load  
24 : Failed U-Bolt at Threaded Section  
25 : Failed U-Bolt at Threaded Section  
26 : 'Necking' of Threaded Section of U-Bolt  
and Damaged Washer due to Bearing



4" AXIAL LOAD

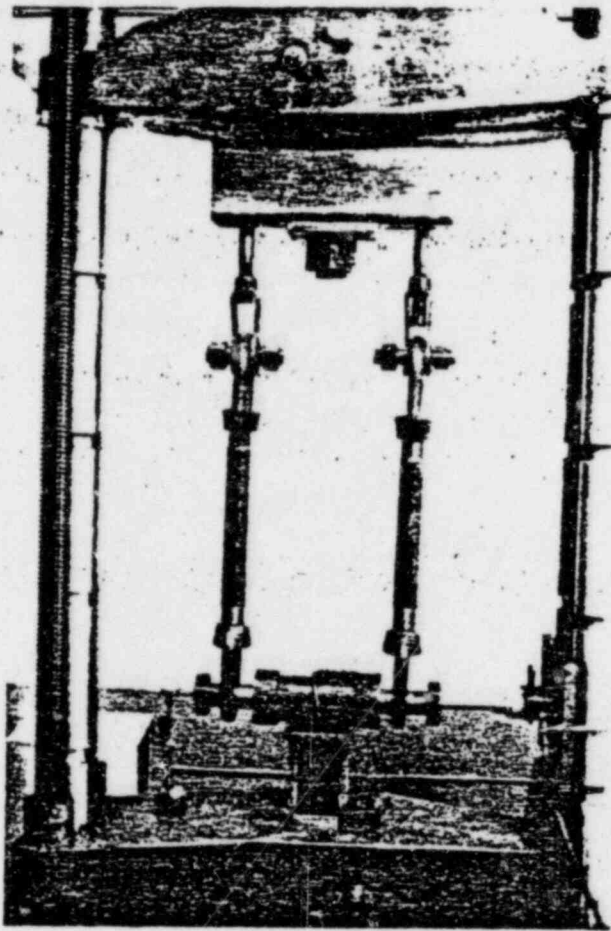


PHOTO 1  
General Test Set-up for  
Axial Load on 4" Pipe U-Bolt

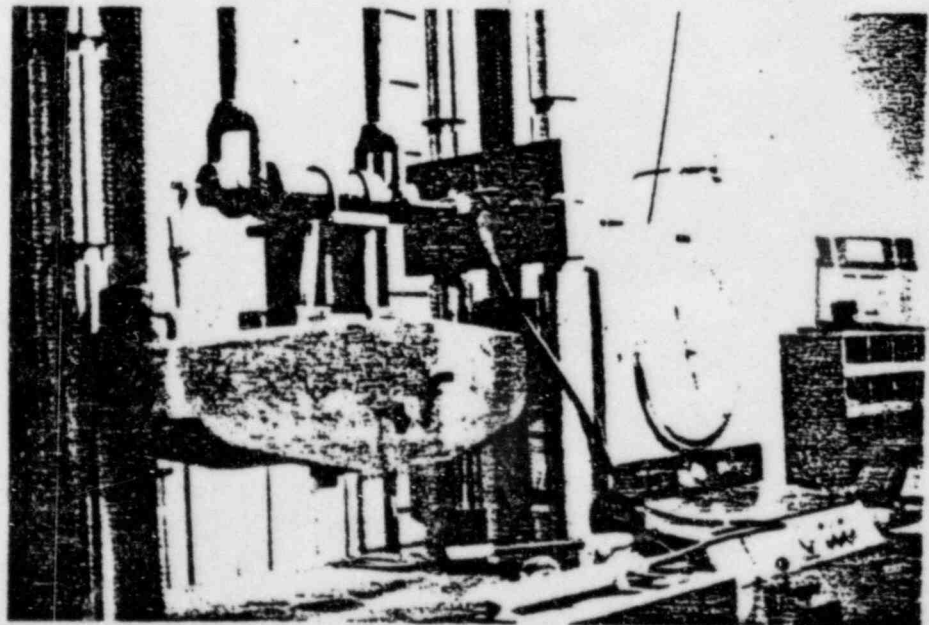


PHOTO 2  
Hydraulic Pump  
and Cylinder  
Arrangement

4" AXIAL LOAD

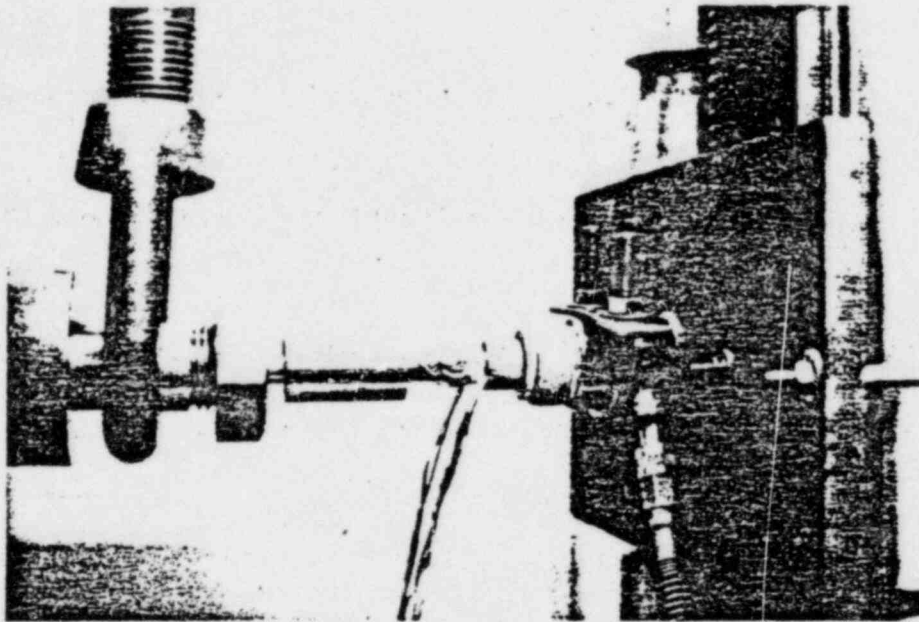


PHOTO 3 : Strain-Gages on Load-Inducing Rods

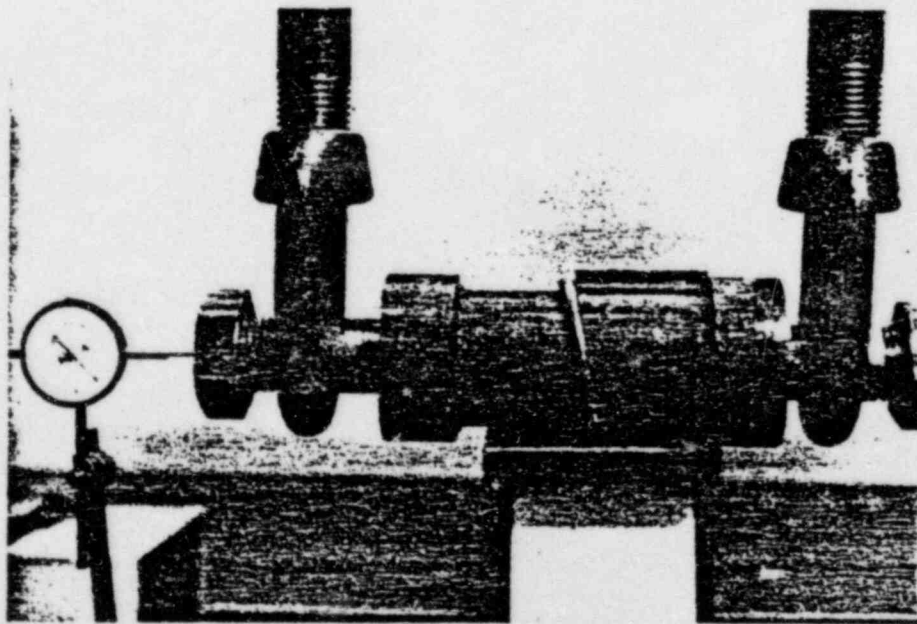


PHOTO 4 : Typical Axial Deflection and Dial Indicator

4" AXIAL LOAD

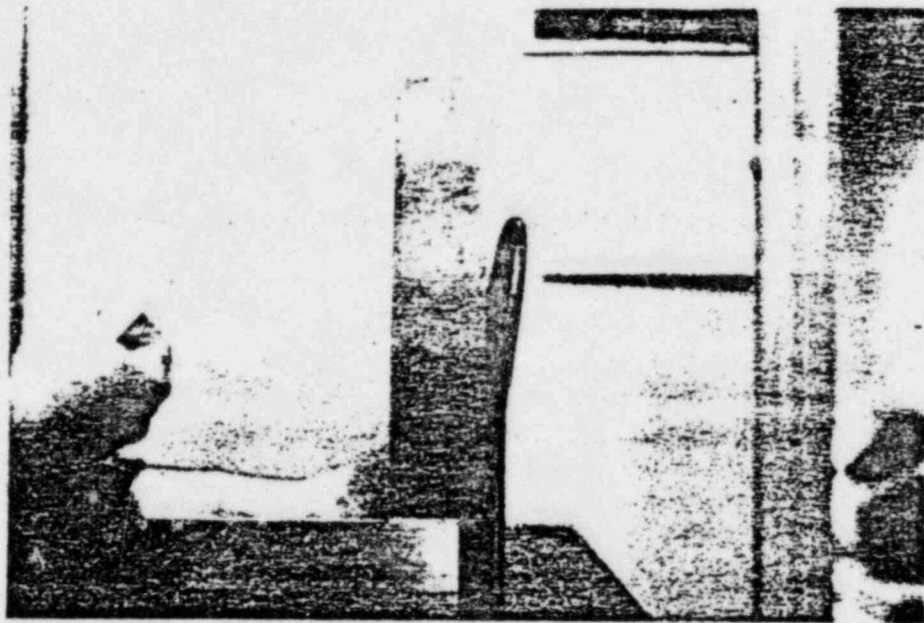


PHOTO 5 : Permanently Deflected Shape

4" LATERAL LOAD

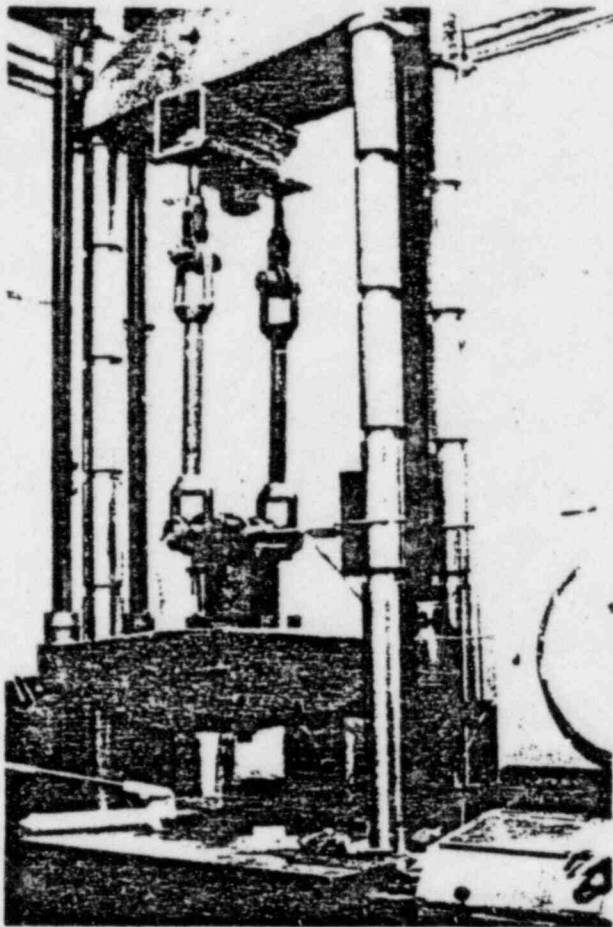
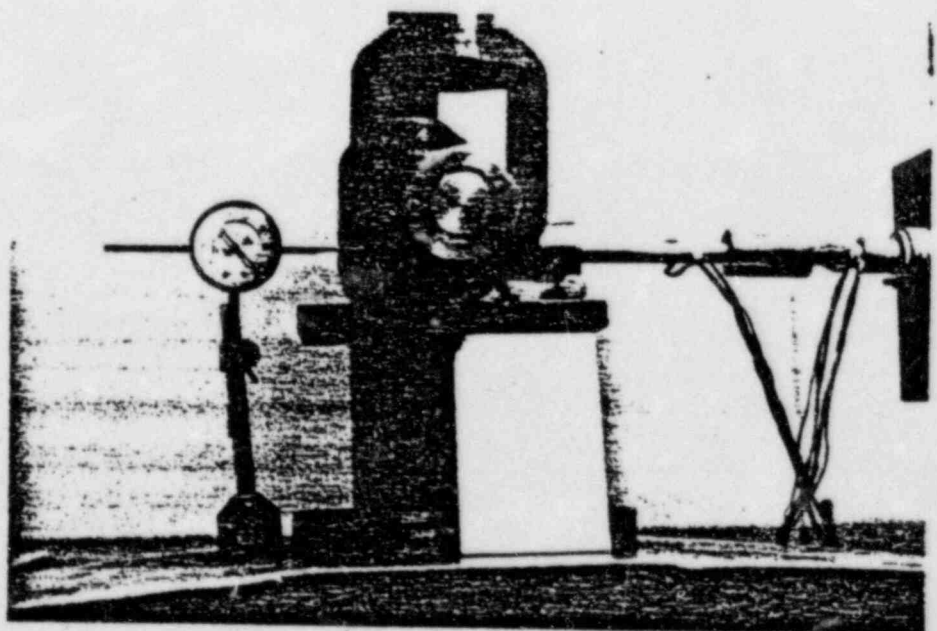


PHOTO 6 :  
General Test Set-up for Side  
Load on 4" Pipe U-Bolt

PHOTO 7 :  
Load-Inducing Rods  
and Dial Indicator





4" LATERAL LOAD

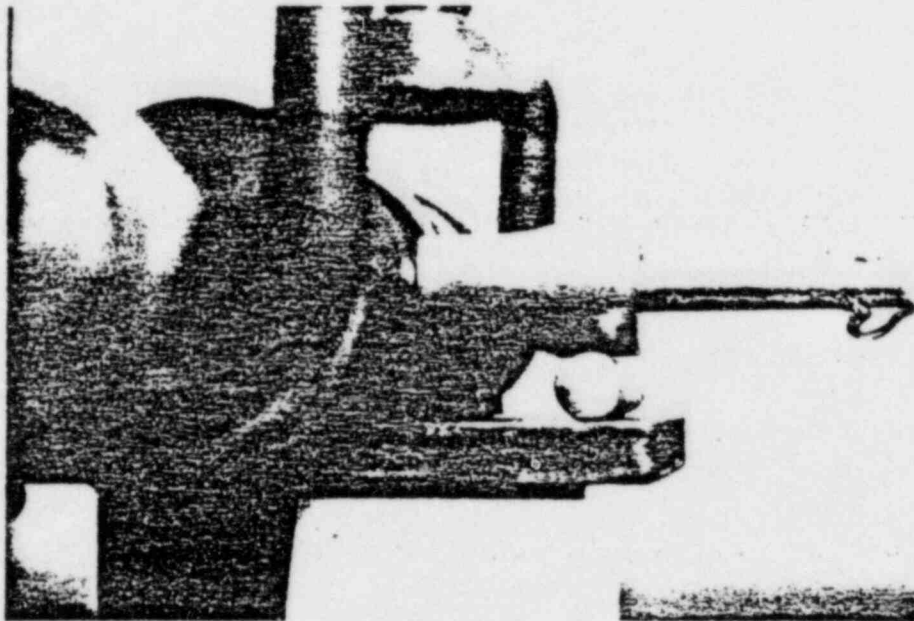


PHOTO 8 : Lift-up of Hex Nut due to Excessive Deformation

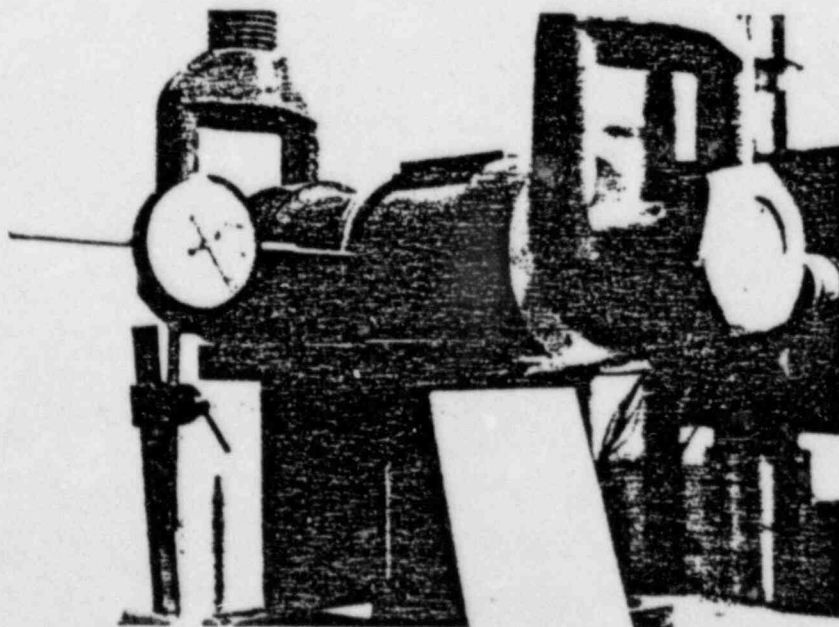


PHOTO 9 : Pronounced Deformation of U-Bolt



4" LATERAL LOAD

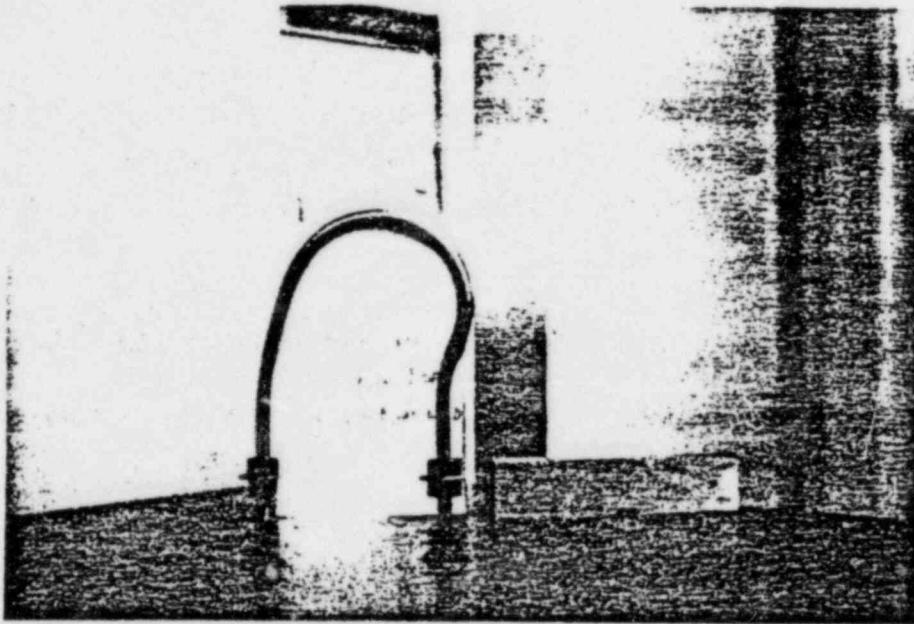


PHOTO 10 : Permanently Deflected Shape

4" NORMAL LOAD

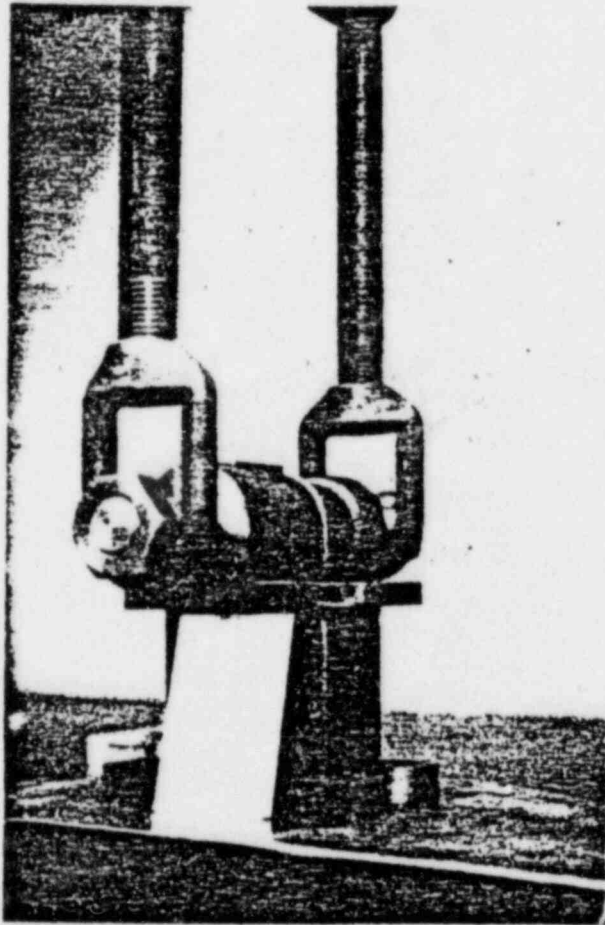


PHOTO 11 :  
Normal Load Set-up for 4"  
Pipe U-Bolt



PHOTO 12 :  
Lift-up of Hex  
Nut due to  
Excessive  
Deformation

4" NORMAL LOAD

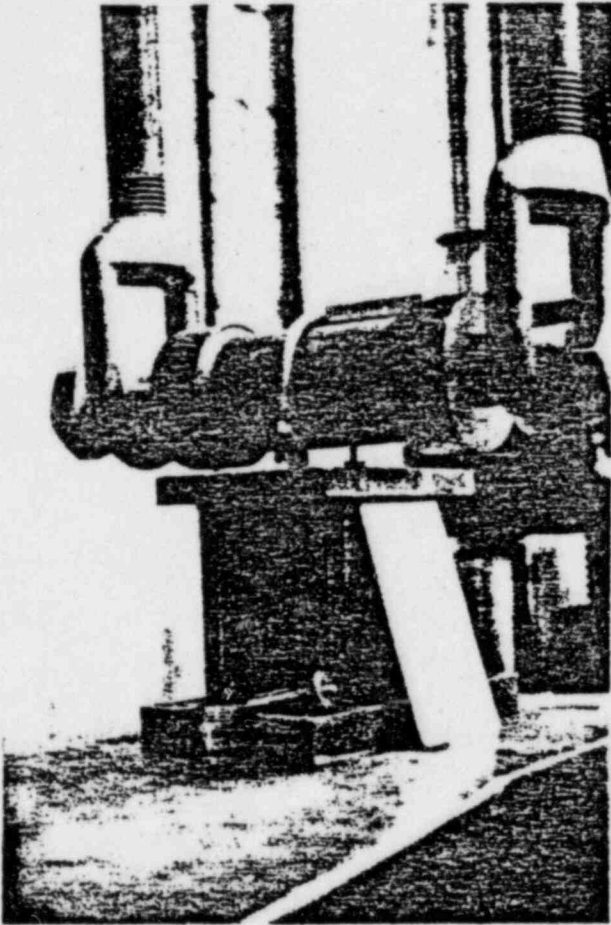
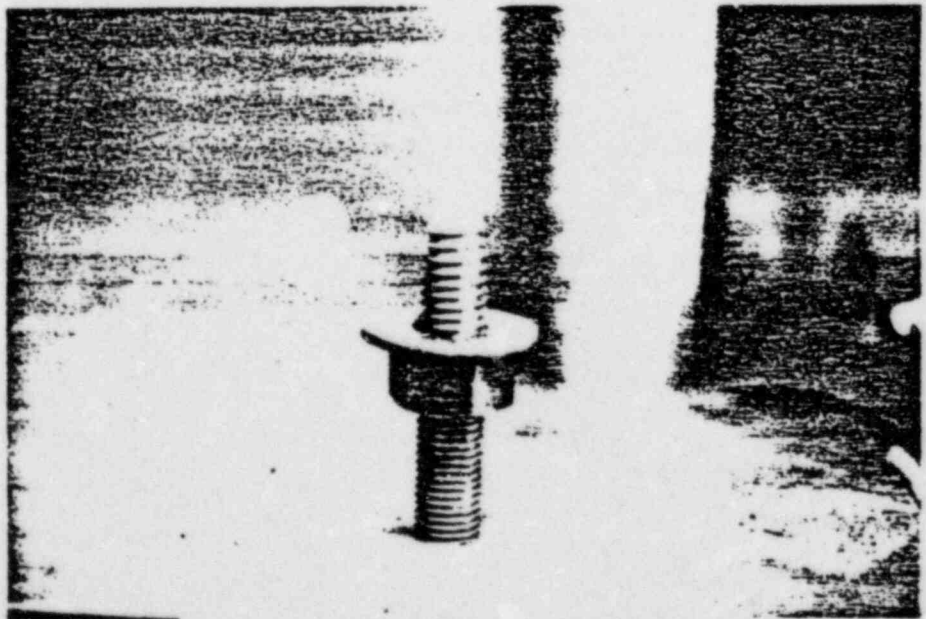


PHOTO 13 :  
Failure of U-Bolt in the  
Tensile Mode

PHOTO 14 :  
'Necking' of  
Threaded Section  
of U-Bolt



24" AXIAL LOAD

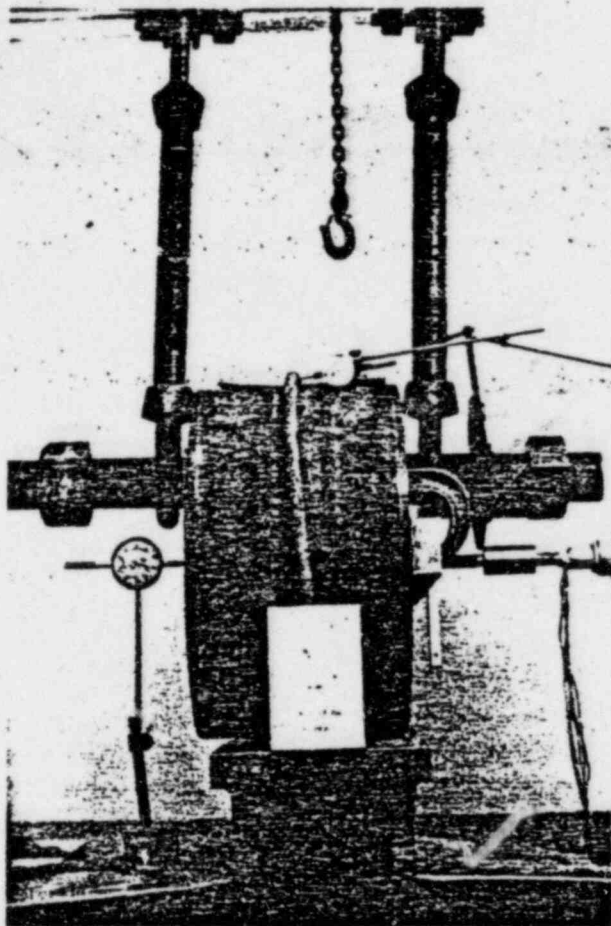
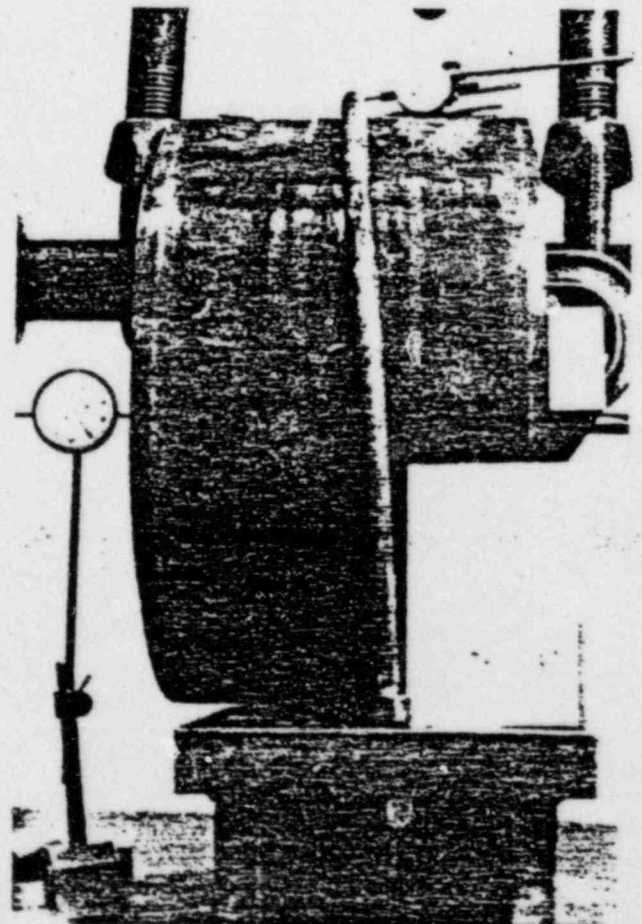


PHOTO 15 :  
General Test Set-up for  
Axial Load on 24" Pipe U-Bolt

PHOTO 16 :  
Axially Deflected  
U-Bolt





24" AXIAL LOAD

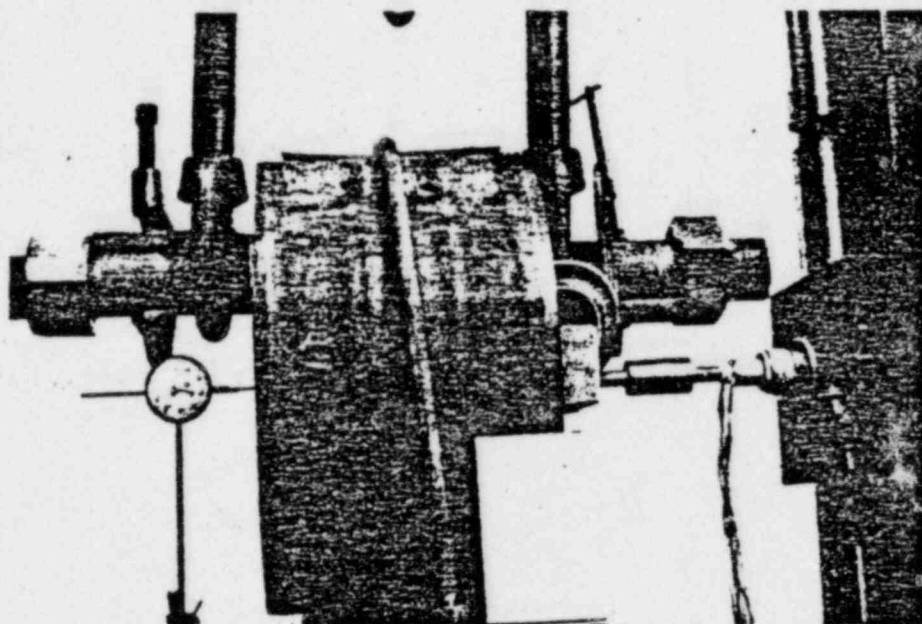


PHOTO 17 : Axially Deflected U-Bolt

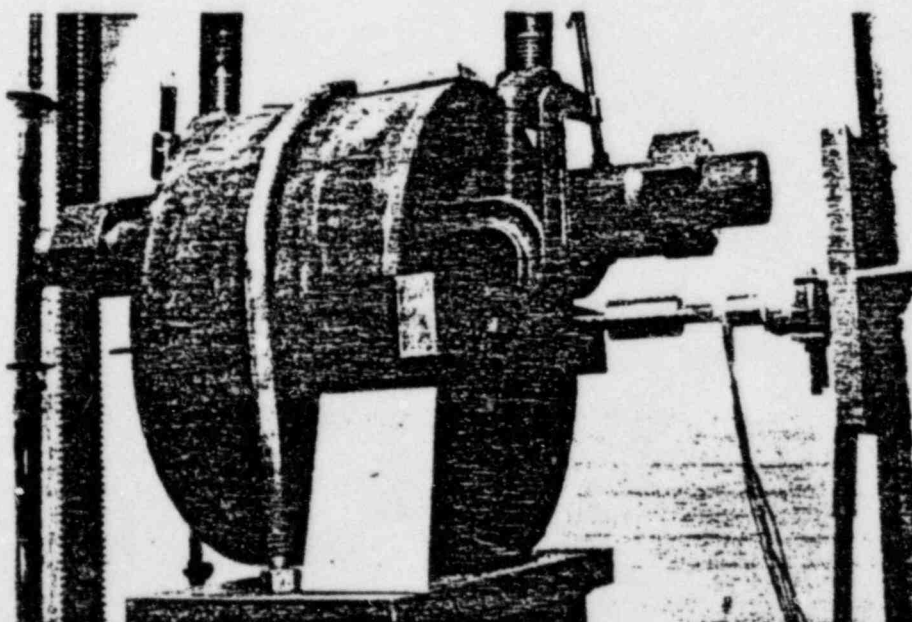


PHOTO 18 : Axially Deflected U-Bolt



24" LATERAL LOAD

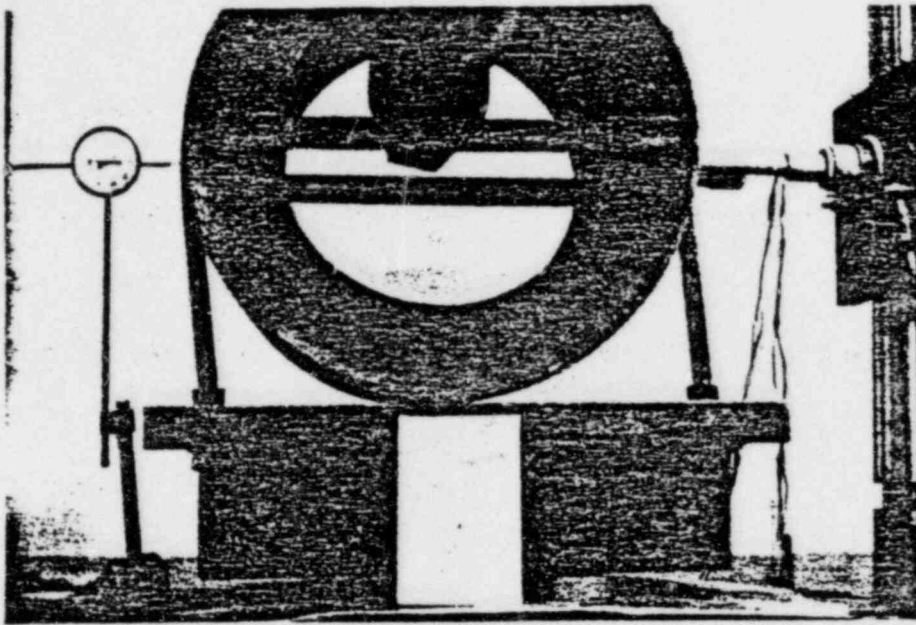


PHOTO 19 : Side Load on 24" Pipe U-Bolt General Deflected Shape

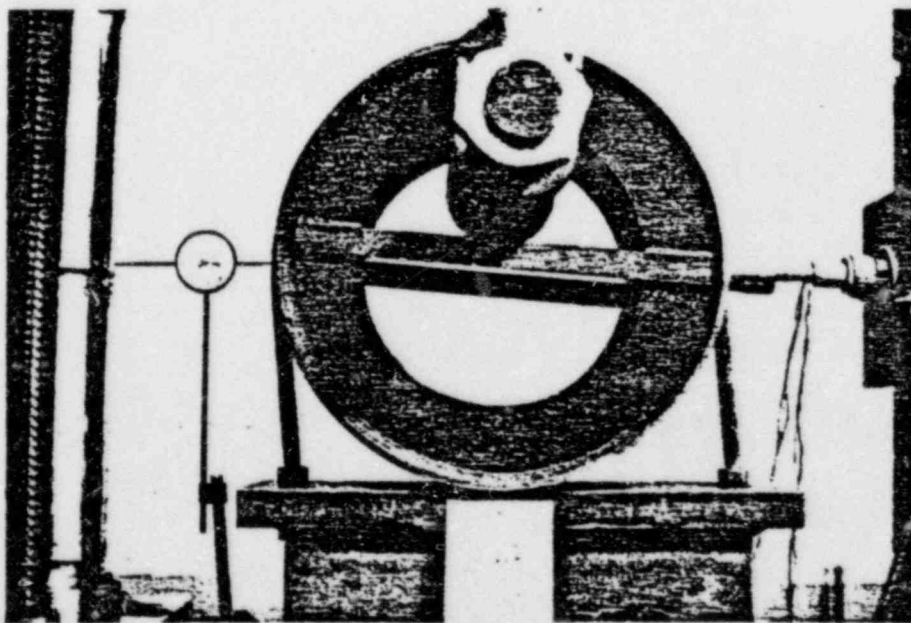


PHOTO 20 : Side Load on 24" Pipe U-Bolt General Deflected Shape

24" LATERAL LOAD

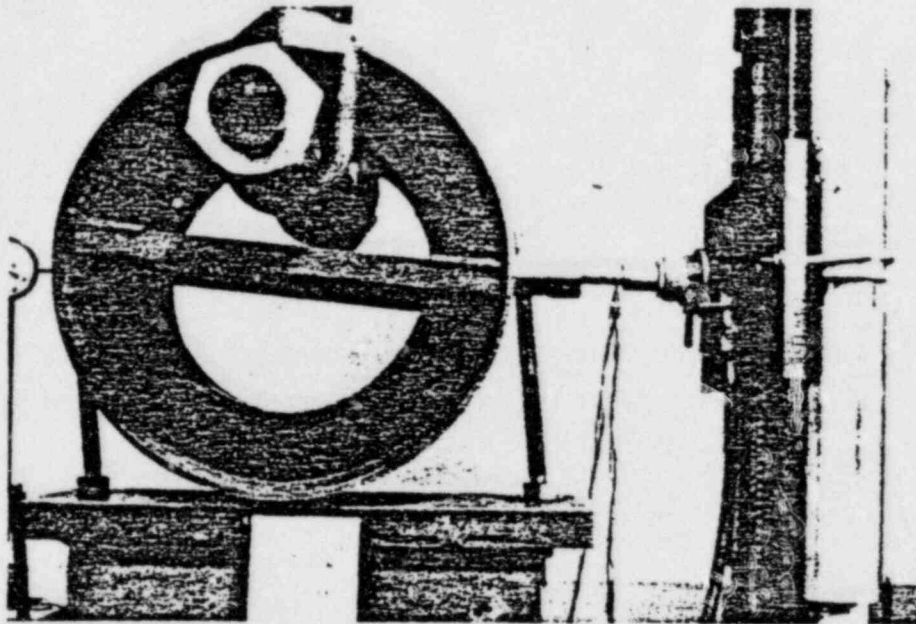


PHOTO 21 : Side Load on 24" Pipe U-Bolt General Deflected Shape

PHOTO 22 :  
Detail of U-Bolt  
and Hex Nut at  
Plate Interface



24" NORMAL LOAD

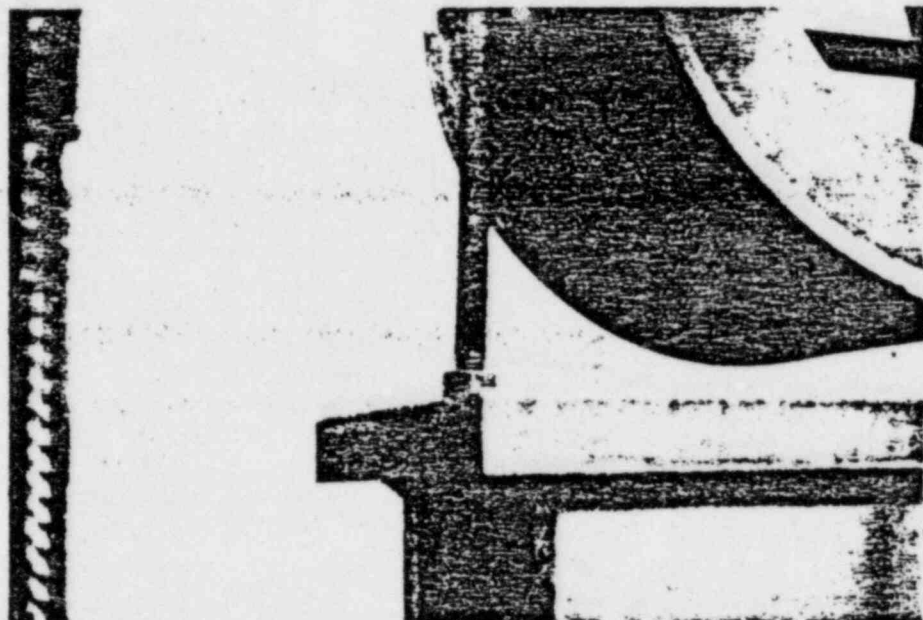


PHOTO 23 : Eminent Failure of U-Bolt under Normal Load

PHOTO 24 :  
Failed U-Bolt  
at Threaded Section



24" NORMAL LOAD

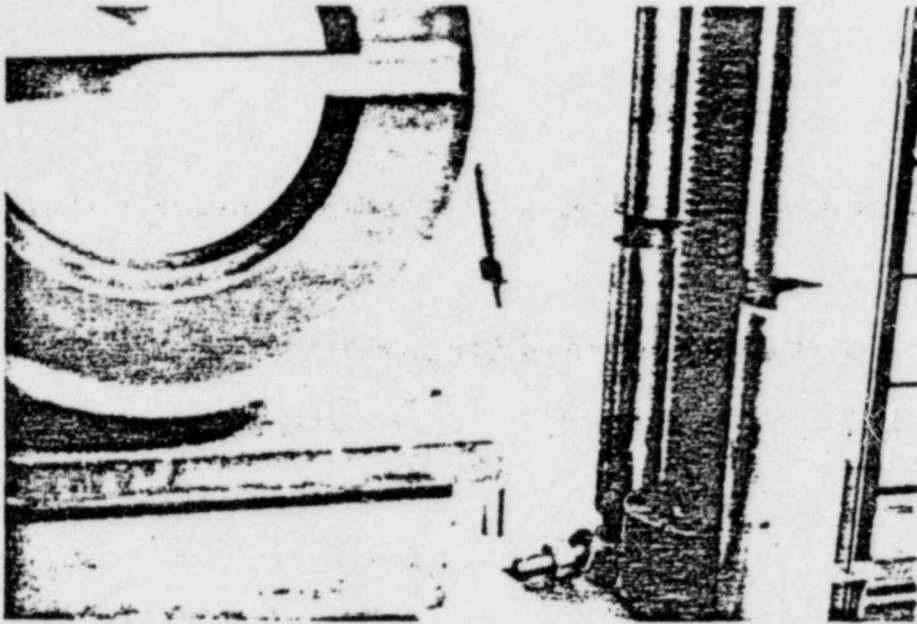


PHOTO 25 : Failed U-Bolt at Threaded Section

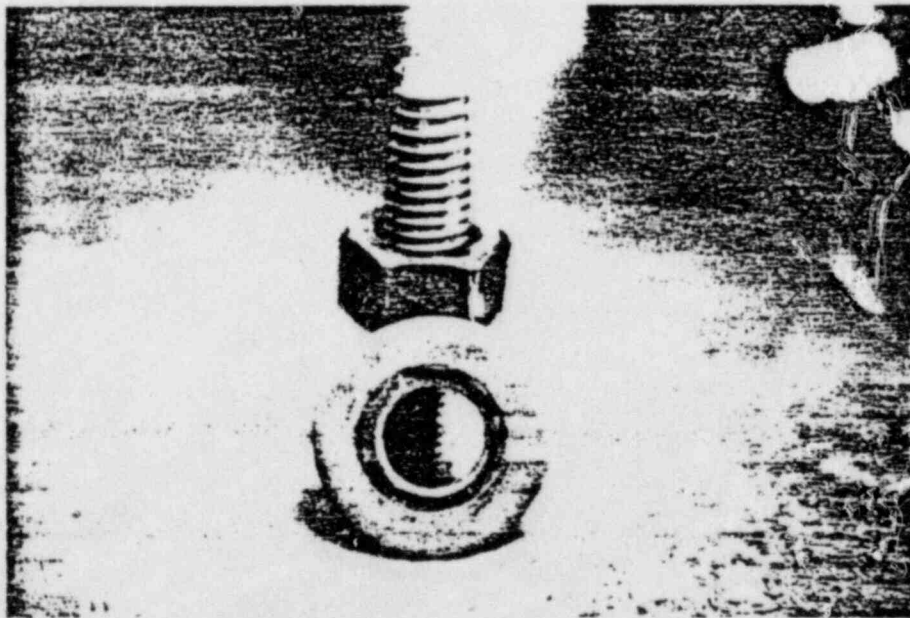


PHOTO 26 : 'Necking' of Threaded Section of U-Bolt  
and Damaged Washer due to Bearing



APPENDIX VII

CERTIFIED MATERIAL TEST REPORTS



CERTIFIED MATERIAL TEST REPORT

1/2 INCH ROD

# AFCO STEEL

### MATERIAL CERTIFICATION

YOUR ORDER # CPF-1310-S

OUR ORDER J 50543

The material listed below is supplied in accordance with the following specifications based on the review of the material manufacturer's certified mill test report and the requirements of your p.o.

- ☐ ASTM Specification \_\_\_\_\_
- ☒ ASME Specification SA36
- ☒ ASME Section III, Subsection NP Class I Component  
Supports, 1974 Edition W'74 Addenda.
- ☐ Paragraphs \_\_\_\_\_
- ☐ Other \_\_\_\_\_

This material was procured and controlled by our Quality Assurance Program, Rev. 5, ASME-NCA-3800 Supplement, Rev. 0, dated 1/28/82  
Approved by your BRV-12913 dated 2/3/82.

[illegible]

Prepared by/date QW 2/25/82 Checked by/date Q70 3/3/82

- ☒ CMTR (QSC Number w/exp. date)  
☐ CMTR (Approved Vendor)

- ☐ Ultrasonic report  
☐ Other



Page 19422

FOR INFORMATION ONLY

CERTIFICATE OF COMPLIANCE

Material Furnished To: <i>AFCO Steel Little Rock, Arkansas</i>	No. <u>107</u> Date: <u>1/18/82</u>
Customer Order No. <u>11408</u> Delivery Copy No. <u>15728</u>	ASME Quality System Cert. No. <u>QSC-327</u> Expiration Date: <u>March 31, 1984</u>

TYPE	DESCRIPTION	TESTS PERFORMED
ASME SA-36	<p>750 Pieces 1 1/2" x 20' Rounds Heat Number 16134</p> <p><b>AFCO STEEL QUALITY ASSURANCE</b></p> <p>Current Approved Mill <u>✓</u> Warehouse _____ Service _____ Other _____</p> <p>Approved by Quality Control Date <u>1-28-82</u> By <u>[Signature]</u></p> <p>"ASME QUALITY SYSTEM CERTIFICATE (MATERIALS) QSC-327 MARCH 31, 1984"</p>	<p>Chemical and Physical (See attached CNTR from Auburn Steel Company)</p> <p>CPF 1310 S Item 24</p> <p>TESTED ON RIR NO. <u>18477</u></p>

This is to certify that this material has been supplied in accordance with the Quality System Program approved by the ASME as required by Section NCA-3800 of the Code.

This is to certify that the material described above and tested or examined as shown, meets or exceeds specification requirements noted

☐ ASTM A-36

☒ ASME SA-36

☐ \_\_\_\_\_

(1980 Edition thru Summer 1981 Addenda, Subsection NF-2000, Class 3, Section 111 NCA-3800, 10 CFR 211)

[Signature] 1/16/82  
QA Manager Date

Page 20 of 22

AUBURN STEEL COMPANY INC. • AUBURN, N. Y. 13001

Address only for P.O. Order

FOR INFORMATION ONLY

DATE 12-18-81

CUSTOMER P.O. #674911-N

PHYSICAL AND CHEMICAL TESTS

1308

SHIPMENT NO. 12-1-179

Test No.	Description	Yield Point	Tensile Strength	Elong. %	Red. %	Bonds	CHEMICAL ANALYSIS					Specification or Remark
							C	Mn	P	S		
	ROUND			8" GL		180						
16134	1" RD	45,130	63,080	24.5		1/2 D	14	52	0.17	0.38		
		45,390	63,590	27.0		1/2 D						

MATERIAL CONFORMS TO THE FOLLOWING SPECIFICATIONS:

ASTM A36-77  
ASME SA36

CODE

I certify the above results to be correct as contained in the records of the company

G. J. Guoga  
Senior Metallurgist

10452 RCP 21

THIS MATERIAL PRODUCED UNDER QUALITY ASSURANCE PROGRAM IN ACCORDANCE WITH SECTION 3 OF NCA-3800 AND ALSO CONFORMS WITH 10-CFR-21.

THIS PRODUCT WAS PRODUCED IN ACCORDANCE WITH Q.A. PROGRAM REVISION 1 DATED 1-2-81.

CPF 1805  
Item 24

DUBOSE STEEL	
QA REVIEW	
SATISFACTORY	
INITIAL	DATE 1/18/87

AFCO Steel  
PO# 11408

FOR INFORMATION ONLY

AFCO STEEL  
QUALITY ASSURANCE

Current Approved  
Mill ☒ Warehouse ☐  
Service ☐  
Other ☐  
Approved by Quality Control  
Date 1-28-82 By [Signature]

THIS INSPECTION  
LIMITED ON

INT. NO. 18477

1/24/22



FOR INFORMATION ONLY

CERTIFICATE OF COMPLIANCE

THIS MATERIAL PRODUCED BY AUBURN STEEL COMPANY, INC.  
AND SOLD TO DUBOSE STEEL INC. ON PURCHASE ORDER 6749-11  
WAS PRODUCED IN ACCORDANCE WITH AUBURN STEEL COMPANY, INC.  
QUALITY ASSURANCE MANUAL REVISION 1 DATED 1-1-81  
AND AUDITED BY DUBOSE STEEL INC. ON OCTOBER 10, 1980.

AUBURN STEEL CO., INC.

*Richard A. Buchan*

RICHARD A. BUCHAN  
MANAGER QUALITY ASSURANCE

AFCO Steel  
PO# 11408

CF 13105  
Hew 24

AFCO STEEL  
QUALITY ASSURANCE

Mill ☒ Current Approved  
Service ☐ Warehouse  
Other ☐  
Approved by Quality Control  
Date 1-28-82 By 7421

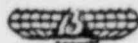
INSPECTION  
TESTED ON  
P.N. NO. 18477

Page 22/22



CERTIFIED MATERIAL TEST REPORT

1 INCH ROD



BROWN & ROOT  
QUALITY ASSURANCE DEPARTMENT  
CONFORMANCE CERTIFICATE

PROJECT: TUGCO

JOB NO.:

UNIT

PAGE 1 OF 1

(1) VENDOR <b>AFCO STEEL</b>		(2) ADDRESS OF VENDOR FACILITY <b>LITTLE ROCK, AR</b>																																					
(3) PURCHASE ORDER NO. <b>CPF-1074-S</b>	(4) SPECIFICATION NO. ---	REV. ---	(5) DRAWING NO. ---																																				
(6) ITEM DESCRIPTION <b>QA MATERIAL</b> <b>SPEC. ASME SEC. III SUB SEC. NF</b> <b>(CLASS 1) 1974-W1974</b> <b>SA 36</b> <b>AFCO REF: 50382 LOAD 4</b>		(7) NO. OF ITEMS <b>42</b>	(8) NO. RELEASED <b>5</b>																																				
(9) COMPLETE <input type="checkbox"/> YES <input checked="" type="checkbox"/> NO																																							
(10) SERIAL OR IDENTIFICATION NO(S) <b>ITEM # 38 - 25 ps 1" x 40 - (A22020)</b> <b>39 - 25 ps 1 1/8" x 20 - (1-8926)</b> <b>40 - 12 ps 1 1/2" x 40 - (M04231)</b> <b>40 - ONE ps 1 1/2" x 20 - (M04231)</b> <b>41 - 25 ps 2" x 20 - (M00827)</b> <b>42 - 10 ps 2 1/2" x 20 - (E09111)</b>																																							
Records checked (✓) below are being transmitted with this certificate as required by the Brown & Root Document Data Sheet.																																							
<table border="0"><tr><td>(11) RECORDS TRANSMITTAL LIST</td><td><input checked="" type="checkbox"/> Material Certifications</td><td><input type="checkbox"/> Operating Elec. Test Records</td><td><input type="checkbox"/> Visual Inspection Records</td></tr><tr><td></td><td><input type="checkbox"/> Heat Treat Records</td><td><input type="checkbox"/> Non-Operating Elec. Test Records</td><td><input type="checkbox"/> Dimensional Inspect. Records</td></tr><tr><td></td><td><input type="checkbox"/> RT Film &amp; Records</td><td><input type="checkbox"/> Pressure Test Records</td><td><input type="checkbox"/> Cleanliness Records</td></tr><tr><td></td><td><input type="checkbox"/> PT/MT Records</td><td><input type="checkbox"/> Seat Tightness Test Records</td><td><input type="checkbox"/> Training Records</td></tr><tr><td></td><td><input type="checkbox"/> UT Records</td><td><input type="checkbox"/> Performance Test Records</td><td><input type="checkbox"/> Packaging Records</td></tr><tr><td></td><td><input type="checkbox"/> Approved Stress Report</td><td><input type="checkbox"/> Wt's Personnel Qual. Certs.</td><td><input type="checkbox"/> Spare Parts List</td></tr><tr><td></td><td><input type="checkbox"/> Approved Design Analyses</td><td><input type="checkbox"/> NDE Personnel Qual. Certs.</td><td></td></tr><tr><td></td><td><input type="checkbox"/> Special Handling Instruc.</td><td><input type="checkbox"/> Nameplate Facsimile</td><td></td></tr><tr><td></td><td><input type="checkbox"/> Operating Manuals</td><td><input type="checkbox"/> Code Data Reports</td><td></td></tr></table>				(11) RECORDS TRANSMITTAL LIST	<input checked="" type="checkbox"/> Material Certifications	<input type="checkbox"/> Operating Elec. Test Records	<input type="checkbox"/> Visual Inspection Records		<input type="checkbox"/> Heat Treat Records	<input type="checkbox"/> Non-Operating Elec. Test Records	<input type="checkbox"/> Dimensional Inspect. Records		<input type="checkbox"/> RT Film & Records	<input type="checkbox"/> Pressure Test Records	<input type="checkbox"/> Cleanliness Records		<input type="checkbox"/> PT/MT Records	<input type="checkbox"/> Seat Tightness Test Records	<input type="checkbox"/> Training Records		<input type="checkbox"/> UT Records	<input type="checkbox"/> Performance Test Records	<input type="checkbox"/> Packaging Records		<input type="checkbox"/> Approved Stress Report	<input type="checkbox"/> Wt's Personnel Qual. Certs.	<input type="checkbox"/> Spare Parts List		<input type="checkbox"/> Approved Design Analyses	<input type="checkbox"/> NDE Personnel Qual. Certs.			<input type="checkbox"/> Special Handling Instruc.	<input type="checkbox"/> Nameplate Facsimile			<input type="checkbox"/> Operating Manuals	<input type="checkbox"/> Code Data Reports	
(11) RECORDS TRANSMITTAL LIST	<input checked="" type="checkbox"/> Material Certifications	<input type="checkbox"/> Operating Elec. Test Records	<input type="checkbox"/> Visual Inspection Records																																				
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	<input type="checkbox"/> Approved Design Analyses	<input type="checkbox"/> NDE Personnel Qual. Certs.																																					
	<input type="checkbox"/> Special Handling Instruc.	<input type="checkbox"/> Nameplate Facsimile																																					
	<input type="checkbox"/> Operating Manuals	<input type="checkbox"/> Code Data Reports																																					
Deviations: <input checked="" type="checkbox"/> None <input type="checkbox"/> Listed Below																																							
(12) REMARKS  <b>FOR INFORMATION ONLY</b>																																							
(13) VENDOR CERTIFICATION <p>THE VENDOR CERTIFIES that the item(s) described above are in conformance with the requirements of the Brown &amp; Root Specification with the approved deviations listed above; are suitable for the purpose intended; are free from defects in design, workmanship, and materials; and are new and of specified quality. A copy of this completed Conformance Certificate will be included with the bill of lading and shipped with the item(s) to Brown &amp; Root, Inc. at the address designated in the procurement documents.</p>																																							
VENDOR AUTHORIZED SIGNATURE: <i>James C. Williams</i>		TITLE: <b>QA SPECIALIST</b>	DATE: <b>9/25/81</b>																																				
The Vendor has certified that the items above meet all contractual requirements. Brown & Root has reviewed evidence supporting this Certificate and, except as noted under "Remarks" above, has verified conformance to requirements. This Certificate does not waive any rights Brown & Root may have under the Purchase Order including the right to reject the item(s) upon discovery of deficiencies during or after arrival at destination.																																							
Final Surveillances: <input type="checkbox"/> Performed <input checked="" type="checkbox"/> Waived																																							
SIGNATURE OF BAR SURVEILLANCE SPECIALIST:  / /	DATE:  / /	TELECOPY NUMBER IF WAIVED (COPY ATTACHED): <b>SW 1971</b>	DATE: <b>9/25/81</b>																																				
(14) RELEASE STATEMENT The item(s) described above are hereby released by Brown & Root Quality Assurance. Shipment may be made subject to authorization by																																							







# MATERIAL CERTIFICATION

to BROWN & ROOT INC.

Your Order # CPF-1074S-

TEXAS UTILITIES

Our Order # 50382

GLEN ROSE, TEXAS

The material listed below is supplied in accordance with the following specifications based on the review of the material manufacturer's certified mill test report and the requirements of your P.O.

☒ ASTM Specification A36

☒ ASME Specification SA36

☒ ASME Section III, Subsection NF Class I Component Support, 1974 Edition W 1974 Addenda.

☐ Paragraphs \_\_\_\_\_

☐ Other \_\_\_\_\_

This material was procured and controlled by our Quality Assurance Program, revision 5, approved by your BRV12616 dated 5-21-81.

Item	Pieces	Description	Heat #	Other
38	25	1" x 40 ROD	A22020	
39	25	1 1/2" x 20	L-3926	
40	12	1 1/2" x 40	M04231	
40	ONE	1 1/2" x 20	M04231	
41	25	2" x 20	M00827	
42	10	2 1/2" x 20	E09111	

Prepared by James C. Williams Date 9-25-81

Checked by Robert F. Oatis Date 9/25/81

## Attachments

- ☒ CMTR (Manufacturer's)
- ☒ CMTR (Approved Vendor)
- ☐ Ultrasonic Report
- ☐ Other \_\_\_\_\_

FOR INFORMATION ONLY





CERTIFIED MATERIAL TEST REPORT

1/2 INCH HEX NUT



nps industries, inc.

10420 metric boulevard  
austin, texas 78758  
telephone 512-836-4161

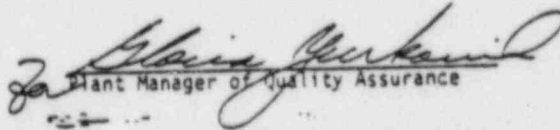
FOR INFORMATION ONLY

DATE: 2/15/83

CERTIFICATE OF CONFORMANCE

REFERENCE: Texas Utilities Services, Inc.  
P. O. Number CP-0046A.1/CPPA 25899  
NPS Order No.: NA-1336  
Item Number(s): 11, 12, 13, 16, 17

We certify that the materials supplied on NPSI Shipping Notice AUS- 15312109  
conform to the referenced purchase order and to the applicable requirements  
of ASME Section III, Subsection NF, Class 1, 1974 Edition, Winter 1974  
Addenda.

  
Plant Manager of Quality Assurance

A.S.M.E. Certificate of Authorization Number N-2323-2 expires on July 13, 1985.

R1A 21249

a subsidiary of nuclear power services inc.

Page 185 of 192



10420 merik boulevard  
austin, texas 78758  
telephone 512-836-4161

Production Release: 602 37  
Code: ASME III/1

ITEM NO.	DESCRIPTION	MIC.NO.
11	SMPP-06	5923 NR SA1936B7
12	SMPP-08	7933 NR SA1936B7
13	SMPP-10	5362 NR SA1936B7
14	FHN-04	7696 NB SA3076B
17	FHN-D3	7695 NB SA3076B

FOR INFORMATION ONLY

RTR 21241

Q.A. REVIEW: S. Dahl  
DATE: 2/15/83



CERTIFIED MATERIAL TEST REPORT

1 INCH HEX NUT





nps industries, inc.

10420 metric boulevard  
austin, texas 78758  
telephone 512-438-4161

QAA-G040  
10/22/81

DOC. REVIEWED NPSI	
QA 2/1	DATE 1/27/82

MATERIAL TRACER RECORD

MARK NO. 7 FAN-1104-300 REV. \_\_\_\_\_ PROD. RELEASE 15100 CODE ASME

ITEM NO.	01	02	03	04	05	06	07	08	09	10
MIC. NO.							641518			
ITEM NO.	11	12	13	14	15	16	17	18	19	20
MIC. NO.										
ITEM NO.	21	22	23	24	25	26	27	28	29	30
MIC. NO.										

SUBASSEMBLIES

DRAWING/ITEM NO.	
MATERIAL	MIC. NO.

DRAWING/ITEM NO.	
MATERIAL	MIC. NO.

DRAWING/ITEM NO.	
MATERIAL	MIC. NO.

DRAWING/ITEM NO.	
MATERIAL	MIC. NO.

DRAWING/ITEM NO.	
MATERIAL	MIC. NO.

FOR INFORMATION ONLY

THIS INSPECTION  
DOCUMENTED ON

RIR NO. 17697 pg 65 of 72

a subsidiary of nuclear power services inc.

BETHLEHEM STEEL CORPORATION  
Industrial Fast. Div. - Western Region

320-3082  
JJ34-4865-00

331  
971525-00 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 32 33 34 35 36 37 38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80 81 82 83 84 85 86 87 88 89 90 91 92 93 94 95 96 97 98 99 100  
COLLECT

WINCO NUCLEAR FASTENERS INC  
8700 SW 26TH SUITE A  
PORTLAND OR 97219

WINCO NUCLEAR FASTENERS INC  
8700 SW 26TH SUITE A  
PORTLAND OR 97219

11/20/81 SEATTLE WA  
RETL REDGAWAY TFR

13 CARTONS MATERIAL MEETS OR EXCEEDS  
THE REQUIREMENTS OF

SA 307 GB  
Q.A. DOCUMENTATION REVIEW  
INTL 12/1 DATE 12/26/82

729

I HAVE COUNTED THE CONTAINERS ON  
THIS LOAD AND CERTIFY THAT THE  
COUNT AGREES WITH THIS BILL OF LADING

1 PALLET - 35#  
FURN A/M CONV

3370.8 BN NUT HVY HX DC BLKN SA 307-768 GR B  
(2575 009 463)

1	300	1	128
2	200	1-1/8	118
3	320	1-1/2	416
4	20	1-3/4	41

FOR INFORMATION ONLY

703

This is to certify that the  
material listed has been tested  
and inspected in accordance  
with the methods prescribed in  
the applicable specification and  
with respect to properties or  
characteristics for which no such  
methods are prescribed. In accord  
ance with standard mill testing  
and inspection practices. Based  
upon such tests and inspection  
practices the undersigned has  
approved said material as ful-  
filling the requirements of  
said specification.

REP ACT TESTS

NPSI REC'D INSPECTION CODE ACCEPTED			
NPT	<input type="checkbox"/>	Q-1	<input type="checkbox"/>
QCC	<input type="checkbox"/>	DATE	1/1/83

Chief Metallurgist

QUALITY CONTROL  
APPROVED  
BY CSR DATE 11-23-81

11-25-81

NPSI AUSTIN	
MIC NO.	645 NR
PO NO.	32-523
SHEET	1 OF 2

FOR INSPECTION

DOCUMENTED ON

RIR NO. 17497

SHIPPING LIST

pg 66 of 72



## DEPARTMENT OF CIVIL ENGINEERING AND ENGINEERING MECHANICS

Carlson Materials Laboratory  
(212) 280-3522 280-3145

Engineering Terrace Building  
Amsterdam Avenue at 120th Street

DATE Aug. 27, 29 & Sept. 2, 3, 1975  
MADE FOR NPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED \_\_\_\_\_  
TESTED BY G. Anderson  
WITNESS \_\_\_\_\_

## REPORT OF TESTS

Page 1 of 1

Lateral Load Tests of Heavy Duty U-Bolts

Test Specimens: Two 4" and two 6" Heavy Duty U-Bolts (7/8" stock, 7/8 - 9 thread) were selected from a group of 3 each submitted by NPS Industries.

How Tested: The test setup is shown on Page 2 of this report. The 4 x 4 x 3/8 angle was bolted to a steel pedestal (14 WF 370 with 1 1/2" pl. top and bottom) bolted to the floor. The U-Bolt was attached to the angle through 15/16 diameter holes.

In all tests, due to the interference fit, a gap, 1/8" - 3/16", was found between the top of the U-Bolt and the pipe. No attempt was made to reduce this gap, but, rather the nuts on both sides of the U-Bolt were tightened so that the pipe slid freely in the U-Bolt.

Each load P/2 indicated in the test setup was monitored by a strain gage load cell calibrated before use. The latest calibration of the testing machine used for the calibration is attached.

All deflections were measured by new standard dial indicators with 1" travel and a 0.001" least reading.

In tests after Test No. 81071 an additional deflection "B" was measured. For one test in each size the deflection "U" was measured relative to the pedestal and in the other test relative to the free end of the angle.

The loading increment of 50 lbs. was found to be too small and was increased as reflected in the data. The tests were stopped at an approximate permanent set of 1/4" in deflection "U".

Results of Tests: The data recorded for each test will be found on Pages 3 through 6 and plotted deflection "U" versus load "P" on Pages 7 and 8.

Aside from the permanent lateral distortion of the U-Bolts it should be noted that at the conclusion of each test the pipe moved very freely in the U-Bolt. No visual distortions were found in the angles except in Test No. 81078. After this test the corner of the angle above the U-Bolt was bent back approximately 0.1" in 2".

Attachments: Calibration of Testing Machine  
Photograph of Test No. 81077, no load.

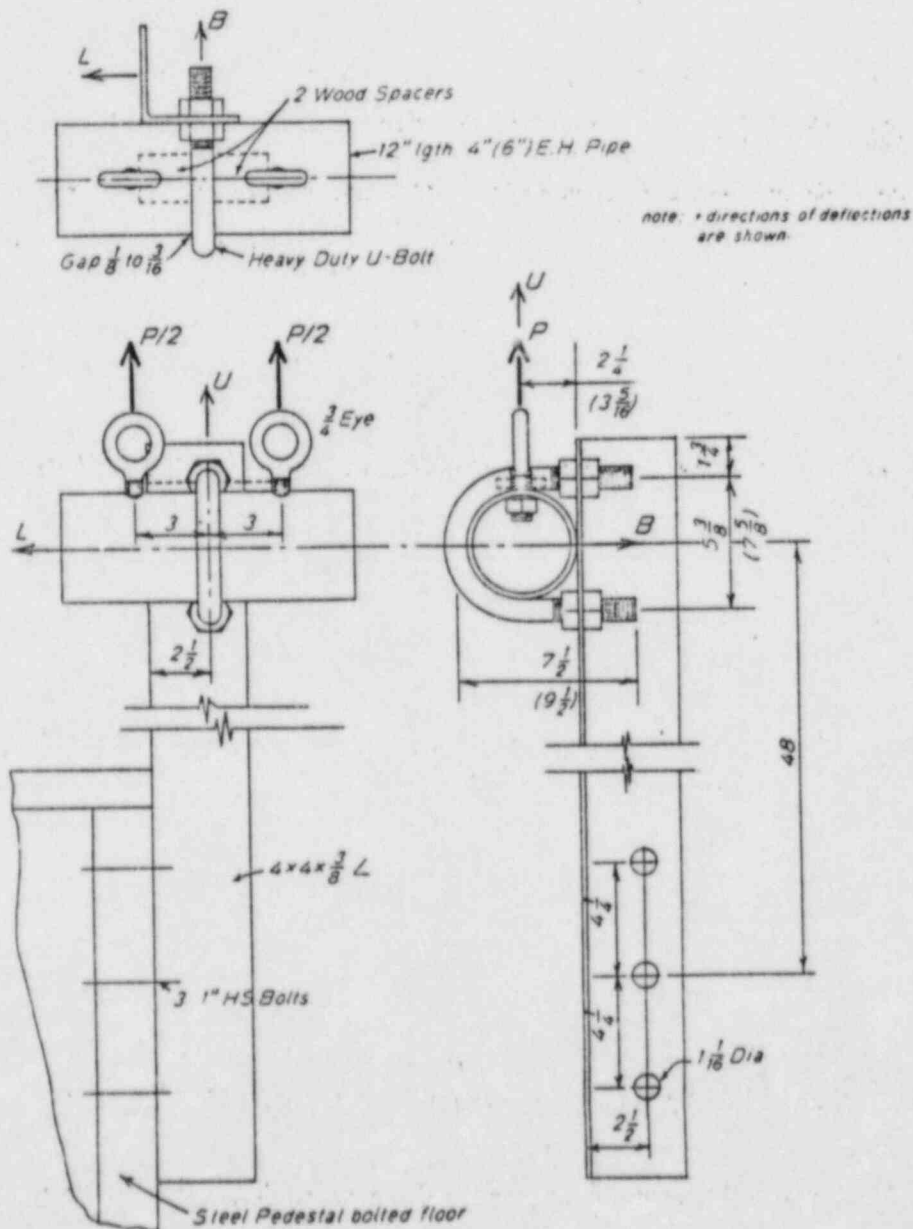


DATE Aug. 27, 29 & Sept. 2, 3, 1975  
FOR NPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED \_\_\_\_\_  
TESTED BY G. Anderson  
WITNESS \_\_\_\_\_

REPORT OF TESTS

Page 2 of 8



TEST SETUP



DATE Aug. 27, 29 & Sept. 2, 3, 1975  
 MADE FOR NPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED \_\_\_\_\_  
 TESTED BY G. Anderson  
 WITNESS \_\_\_\_\_

REPORT OF TESTS

Page 3 of 8

Lab Test No. 81071

6" Heavy-Duty U-Bolt, Deflection U measured relative to Supporting Pedestal

Load, P kips	Deflections, ins.			Set, ins.		
	U	L	B	U	L	B
0.31	005	002				
0.41	007	003		001	0	
0.5	009	003		—	—	
0.6	011	004		0	0	
0.71	013	001		—	—	
0.8	014	001		0	0	
0.9	018	001		—	—	
1.0	010	001		0	0	
1.1	015	001		—	—	
1.2	015	002		001	001	
1.31	018	002		—	—	
1.42	034	002		001	001	
1.5	033	004		—	—	
1.6	035	004		002	001	
1.8	038	003		002	001	
2.0	045	003		004	001	
2.2	050	003		005	002	
2.4	058	004		007	002	
2.64	066	004		009	002	
2.8	072	005		—	—	
3.0	078	006		012	002	
3.2	086	007		017	001	
3.6	109	007		032	002	
4.0	181	007		093	003	
4.4	215	009		118	005	
4.8	272	015		160	003	
5.0	310	016		192	003	
5.2	358	019		235	004	
5.4	399	020		271	002	

DEPARTMENT OF CIVIL ENGINEERING AND ENGINEERING MECHANICS

Coleman Materials Laboratory  
(212) 280-3522, 280-3145

Engineering Terrace Building  
Amsterdam Avenue at 125th Street

DATE Aug. 27, 29 & Sept. 2, 3, 1975  
MADE FOR NPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED \_\_\_\_\_  
TESTED BY G. Anderson  
WITNESS \_\_\_\_\_

REPORT OF TESTS

Page 4 of 8

Lab Test No 51077

6" Heavy Duty U-Bolt, Deflection U measured relative to Free End of Angle

Load, P kips	Deflections, ins.			Set, ins.		
	U	L	B	U	L	B
0.3	003	—	—	—	—	—
0.6	006	004	012	0	-003	002
0.9	010	—	—	—	—	—
1.2	014	003	058	0	-003	002
1.5	017	—	—	—	—	—
1.8	022	006	081	001	-003	003
2.1	027	008	092	—	—	—
2.4	032	008	106	004	-003	004
2.8	048	009	122	012	-002	006
3.0	054	014	130	016	-002	007
3.2	060	014	135	020	-002	006
3.4	070	014	145	028	-002	007
3.6	081	017	151	035	-002	007
3.8	091	018	160	045	-002	009
4.0	104	020	166	055	-002	008
4.4	125	023	179	073	-001	009
4.8	154	027	192	100	-001	011
5.2	200	031	206	146	0	014
5.6	265	037	218	206	001	018
6.0	335	042	233	272	001	026

DATE Aug. 27, 29 & Sept. 2, 3, 1975  
 MADE FOR MPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED \_\_\_\_\_  
 TESTED BY G. Anderson  
 WITNESS \_\_\_\_\_

REPORT OF TESTS

Page 5 of 8

Lab Test No 81078

4" Heavy Duty U-Bolt, Deflection U measured relative to Free End of Angle

Load, P kips	Deflections, ins.			Set, ins.		
	U	L	B	U	L	B
0.4	.001	.007	.009	0	.002	0
0.8	.004	.009	.022	0	-.001	.001
1.2	.006	.009	.035	0	-.001	.002
1.4	.007	.009	.044	0	-.002	.003
1.6	.007	.015	.042	0	-.002	.002
1.8	.009	.009	.056	0	-.004	.004
2.0	.011	.010	.063	0	-.004	.004
2.2	.011	.017	.059	.001	-.002	.005
2.4	.012	.018	.064	.001	-.002	.005
2.6	.014	.020	.070	.001	-.001	.006
2.8	.015	.022	.075	.002	-.001	.007
3.0	.017	.023	.080	.002	-.001	.008
3.2	.018	.024	.085	.003	-.001	.009
3.4	.020	.024	.094	.004	-.001	.011
3.6	.022	.026	.097	.005	-.001	.014
3.8	.024	.029	.103	.006	-.001	.015
4.0	.027	.030	.109	.008	-.001	.017
4.2	.030	.033	.116	.010	0	.020
4.4	.034	.034	.128	.012	0	.024
4.6	.038	.036	.135	.015	0	.028
4.8	.042	.039	.140	.018	0	.032
5.0	.046	.040	.157	.022	.002	.040
5.2	.051	.042	.166	.025	.002	.046
5.6	.060	.043	.188	.031	.002	.057
6.0	.071	.047	.225	.040	.003	.085
6.4	.083	.051	.255	.050	.005	.112
7.2	.118	.067	.289	.083	.015	.143
8.0	.166	.078	.370	—	—	—
9.6	.286	.135	.490	.244	.026	.362
10.0	.330	.197	.530	.289	.021	.407

## DEPARTMENT OF CIVIL ENGINEERING

Material Laboratory  
121 250-3522, 280-3145

Test Room Building  
Austrian Avenue at 120th Street

Aug. 27, 29 & Sept. 2, 3, 1975  
for NPS Industries, Inc.  
235 Moore St.  
Hackensack, N.J. 07601

MACHINE USED G. Anderson

TESTED BY

WITNESS

## REPORT OF TESTS

Page 6 of 8

Lab Test No. 81079

4" Heavy Duty U-Bolt, Deflection U measured relative to Supporting Pedestal

Load, P kips	Deflections, ins.			Set, ins.		
	U	L	B	U	L	B
0.4	.003	.007	.006	0	0	.001
0.8	.007	.008	.016	.001	0	.001
1.2	.013	.008	.025	.002	0	.001
1.6	.019	.010	.033	.002	0	.001
2.0	.024	.011	.042	.002	0	.001
2.4	.029	.013	.050	.003	0	.002
2.8	.033	.017	.059	.003	0	.003
3.0	.036	.019	.063	.004	.001	.003
3.2	.039	.021	.068	.005	.001	.003
3.4	.042	.022	.072	.006	.002	.004
3.6	.045	.024	.077	.007	.002	.005
3.8	.049	.024	.082	.009	.002	.006
4.0	.054	.027	.088	.011	.002	.007
4.2	.059	.027	.093	.014	.002	.008
4.4	.065	.028	.100	.017	.002	.009
4.6	.069	.029	.104	.019	.001	.011
4.8	.075	.029	.110	.023	.001	.012
5.0	.081	.031	.116	.025	0	.014
5.5	.099	.038	.119	.036	.001	.017
6.0	.119	.039	.129	.051	0	.020
6.5	.156	.043	.143	.079	0	.024
7.0	.209	.047	.163	.118	0	.028
8.0	.298	.051	.179	.190	0	.035
10.0	.470	.063	.199	.347	.003	.050

Carlson Materials Lab. story  
212 280-1522, 280-2145

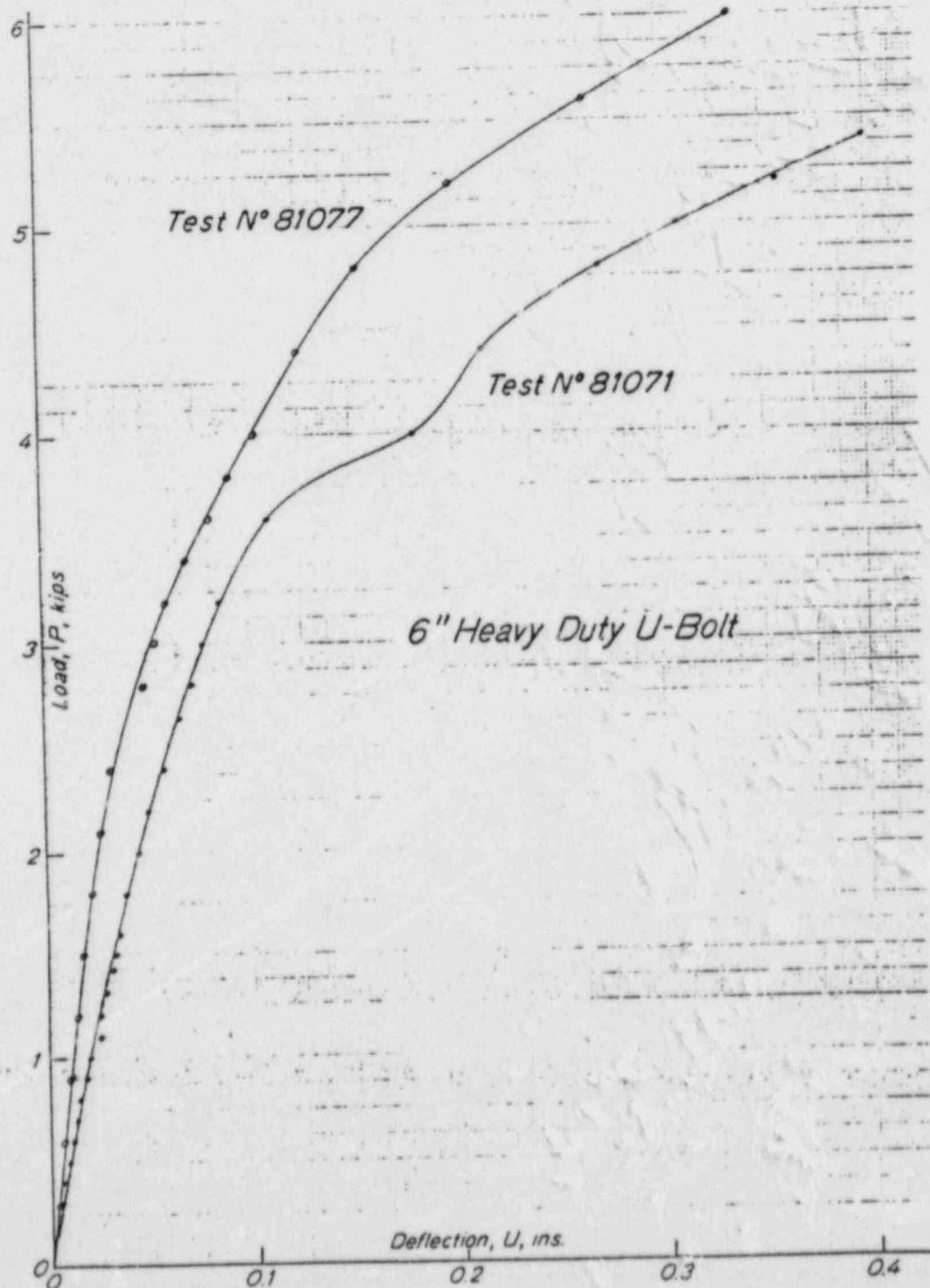
Engineering Testing Building  
American Avenue at 150th Street

DATE Aug. 27, 29 & Sept. 2, 3, 1975  
TEST FOR NPS Industries, Inc.  
215 Moore St.  
Hackensack, N.J. 07601

MACHINE USED G. Anderson  
TESTED BY   
WITNESSES

REPORT OF TESTS

Page 7 of 8





Carlson Materials Laboratory  
(212) 280-3572, 280-3145

Engineering Terrace Building  
Amsterdam Avenue at 120th Street

DATE Aug. 27, 29 & Sept. 2, 3, 1975

MADE FOR NPS Industries, Inc.

235 Moore St.

Hackensack, N.J. 07601

MACHINE USED G. Anderson

TESTED BY

WITNESS

REPORT OF TESTS

Page 8 of 8

