

ADVANCE COPY

ISOLATION/PURGE VALVE ANALYSIS FOR

6"-1200 BUTTERFLY VALVE

Project Site Susquehanna Steam Electric Station
Berwick, Pennsylvania
Customer Pennsylvania Power & Light
Engineer Bechtel Power Corporation
Specification No. 8856
Original Purchase Order 8856-P-31-AC
Original Pratt Job No. D-0026-3
Valve Tag Nos. HBB-BF-AO-5721

General Arrangement Drawings C-2600 Rev. 5

Cross Section Drawing C-2988 Rev. 2

Prepared by: Rac N. Kaza

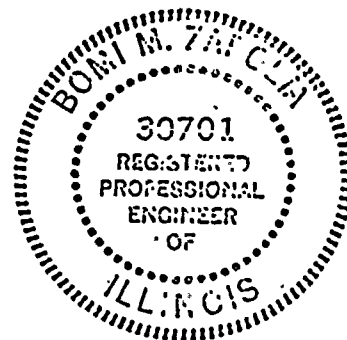
Date: 3/12/82

Reviewed by: T. J. Whina

Date: 3-12-82

Certified by: Ben M. Faraba

Date: 3/12/82



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12. 11. 1962

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I. Introduction

This investigation has been made in response to a request by the customer/engineer for evaluation of containment isolation/purge valves during a faulted condition arising from a loss of coolant accident (LOCA).

The analysis of the structural and operational adequacy of the valve assembly under such conditions is based principally upon containment pressure vs. time data, system response (delay) time, piping geometry upstream of the valve, back pressure due to ventilation components downstream of the valve, valve orientation and direction of valve closure.

The above data as furnished by the customer/engineer forms the basis for the analysis. Worst case conditions have been applied in the absence of definitive input.

II. Considerations

The NRC guidelines for demonstration of operability of purge and vent valves, dated 9/27/79, have been incorporated in this evaluation as follows:

- A.1. Valve closure time during a LOCA will be less than or equal to the no-flow time demonstrated during shop tests, since fluid dynamic effects tend to close a butterfly valve. Valve closure rate vs. time is based on a sinusoidal function.
2. Flow direction through valve contributing to highest torque; namely, flow toward the hub side of disc if asymmetric, is used in this analysis. Pressure on upstream side of valve as furnished by customer/engineer is utilized in calculations. Downstream pressure vs. LOCA time is assumed to be worst case.
3. Worst case is determined as a single valve closure of the inside containment valve, with the outside containment valve fixed at the fully open position.
4. Containment back pressure will have no effect on cylinder operation since the same back pressure will also be present at the inlet side of the cylinder and differential pressure will be the same during operation.
5. Purge valves supplied by Henry Pratt Company do not normally include accumulators. Accumulators, when used, are for opening the valve rather than closing.
6. Torque limiting devices apply only to electric motor operators which were not furnished with purge valves evaluated in this report.

7&8. Drawings or written description of valve orientation with respect to piping immediately upstream, as well as direction of valve closure, are furnished by customer/engineer. In this report, worst case conditions have been applied to the analysis; namely, 90° elbow (upstream) oriented 90° out-of-plane with respect to valve shaft, and leading edge of disc closing toward outer wall of elbow. Effects of downstream piping on system back pressure have been covered in paragraph A.2. (above).

B. This analysis consists of a static analysis of the valve components indicating if the stress levels under combined seismic and LOCA conditions are less than 90% of yield strength of the materials used.

A valve operator evaluation is presented based on the operators ability to resist the reaction of LOCA-induced fluid dynamic torques.

C. Sealing integrity can be evaluated as follows:

Decontamination chemicals have very little effect on EPT and stainless steel seats. Molded EPT seats are generically known to have a cumulative reiradiation resistance of 1×10^6 rads at a maximum incidence temperature of 350°F. It is recommended that seats be visually inspected every 18 months and be replaced periodically as required.

Valves at outside ambient temperatures below 0°F, if not properly adjusted, may have leakage due to thermal contraction of the elastomer, however, during a LOCA, the valve internal temperature would be expected to be higher than ambient which tends to increase sealing capability after valve closure. The presence of debris or damage to the seats would obviously impair sealing.

III. Method of Analysis

Determination of the structural and operational adequacy of the valve assembly is based on the calculation of LOCA-induced torque, valve stress analysis and operator evaluation.

A. Torque calculation

The torque of any open butterfly valve is the summation of fluid dynamic torque and bearing friction torque at any given disc angle.

Bearing friction torque is calculated from the following equation:

$$T_B = P \times A \times U \times \frac{d}{2}$$

where

P = pressure differential, psi

A = projected disc area normal to flow, in²

U = bearing coefficient of friction

d = shaft diameter, in.

Fluid dynamic torque is calculated from the following equations:

For subsonic flow

$$\left[R_{CR} \geq \frac{P_1}{P_2} > 1.07 \text{ (approx.)} \right]$$

$$T_D = D^3 \times C_{T1} \times P_2 \times \sqrt{\frac{K}{1.4}} \times F_{RE}$$

For sonic flow

$$\left[\frac{P_1}{P_2} \geq R_{CR} \right]$$

$$T_D = D^3 \times C_{T2} \times P_2 \times \sqrt{\frac{K}{1.4}} \times F_{RE} \quad (F_{RE} \geq 1)$$

Where

T_D = fluid dynamic torque, in-lbs.

F_{RE} = Reynold number factor

R_{CR} = critical pressure ratio, (f (α))

P_1 = upstream static pressure at flow condition, psia

P_2 = downstream static pressure at flow condition, psia

D = disc diameter, in.

C_{T1} = subsonic torque coefficient

C_{T2} = sonic torque coefficient

K = isentropic gas exponent (≈ 1.2 for air/steam mix)

α = disc angle, such that 90° = fully open; 0° = fully closed

Note that C_{T1} and C_{T2} are a function of disc angle, an exponential function of pressure ratio, and are adjusted to a 5" test model using a function of Reynolds number.

Torque coefficients and exponential factors are derived from analysis of experimental test data and correlated with analytically predicted behavior of airfoils in compressible media.

Empirical and analytical findings confirm that subsonic and sonic flow conditions across the valve disc have an unequal and opposite effect on dynamic torque. Specifically, increases in upstream pressure in the subsonic range result in higher torque values, while increasing P_1 in the sonic range results in lower torques. Therefore, the point of greatest concern is the condition of initial sonic flow, which occurs at a critical pressure ratio.

The effect of valve closure during the transition from subsonic to sonic flow is to greatly amplify the resulting torques. In fact, the maximum dynamic torque occurs when initial sonic flow occurs coincident with a disc angle of 72° (symmetric) or 68° (asymmetric) from the fully closed position.



The following computer output summarizes calculation data and torque results for valve opening angles of 90° to 0° .

JOB:SUSQUEHANA/BECHTEL

SAT.STEAM/AIR MIXTURE WITH 1.4 LBS STEAM PER 1-LBS AIR.

SPEC.GR.= .738255 MOL.WT.= 21.3872 KAPA(ISENT.EXP.)= 1.19775 R= 72.1972

GAS CONSTANT-CALC.

SONIC SPEED(MOVING MIXTR.)= 1354.57 FEET/SEC AT 265 DEG.

ABSOL.MAX.TORQUE(FIRST SONIC)AT 72-68 DG.VLV.ANG.= 478 IN-LBS @ 68 DEG.

MAX.TORQUE INCLUDES SIZE EFFECT(REYNOLDS NO.ETC)APPX. X .996595 FOR 6 IN
CH BASIC LINE I.D.ALL PRESSURES USED:STATIC(TAP)PRESS.-ABSOLUTE;P2 INCL.RECOVERY PRESS.
(TORQUE)CALC'S VALIDITY:P1/P2>1.07;

VALVE TYPE: 6"-1200 CLASS 150
 DISC SIZE: 5.2 INCHES OFFSET ASYMMETRIC DISC
 SHAFT DIA.: 1 INCHES
 BEARING TYPE: BRONZE
 SEATING FACTOR: 15
 INLET PRESS.VAR.MAX.: 48.2 PSIA
 OUTLET PRESSURE(P6): 17.2 PSIA (72 DEG. ACTUAL PRESS.ONLY(VAR.))
 MAX.ANG.FLOW RATE: 6658.94 CFM; 7698.92 SCFM; 423.231 LB/MIN
 CRIT.SONIC FLOW-90DG: 518.539 LB/MIN AT 19.651 INLET PSIA
 VALVE INLET DENSITY: 6.35583E-02 LB/FT^3-MIN. .129262 LB/FT^3-MAX.
 FULL OPEN DELTA P: 7.74903 PSI
 SYSTEM CONDITIONS:

PIPE IN-PIPE-OUT -AND- AIR/STEAM MIXTURE SERVICE @ 265 DEG.F
 MINIMUM 0.75 DIAH. PIPE DOWNSTREAM FROM CENT.LINE SHAFT.

P1 ABS. PRESSURE(ADJ.)FOLLOWS TIME/PRESS.TRANSIENT CURVE.

--5 IN.MODEL EQUIV.VALUES-----ACTUAL SIZE VALUES-----

| ANGLE | P1 | P2 | DELP | PRESS. | FLOW | TD | TB+TH | TIME(LOCA) |
|------------|-------|-------|-------|---------|----------|-------------------|----------|-------------|
| APPRX.PSIA | PSIA | PSI | RATIO | (SCFM) | (LB/MIN) | -----INCHLBS----- | TD-TB-TH | SEC. |
| 90 | 23.70 | 15.50 | 8.20 | .654 CR | 7698 | 423 | 140 | 128 0.20 |
| 85 | 27.98 | 15.80 | 12.18 | .565 | 11888 | 653 | 169 | 154 0.44 |
| 80 | 31.32 | 15.99 | 15.32 | .511 | 12896 | 708 | 209 | 190 0.68 |
| 75 | 34.23 | 16.07 | 18.16 | .469 CR | 13344 | 733 | 370 | 337 0.90 |
| 72 | 35.78 | 15.96 | 19.82 | .446 CR | 12888 | 708 | 489 | 446 1.02 |
| 70 | 36.74 | 15.94 | 20.80 | .434 CR | 12553 | 690 | 457 | 417 1.10 |
| 65 | 38.84 | 15.73 | 23.11 | .405 CR | 11543 | 634 | 439 | 401 1.27 |
| 60 | 40.51 | 15.50 | 25.00 | .383 CR | 10054 | 552 | 345 | 308 1.41 |
| 55 | 41.72 | 15.26 | 26.46 | .366 CR | 8558 | 470 | 302 | 256 1.52 |
| 50 | 42.45 | 15.10 | 27.36 | .356 | 7045 | 387 | 225 | 172 1.58 |
| 45 | 42.70 | 14.97 | 27.73 | .351 | 6983 | 383 | 187 | 128 1.60 |
| 40 | 42.80 | 14.89 | 27.91 | .348 | 4807 | 264 | 130 | 65 1.62 |
| 35 | 43.11 | 14.81 | 28.30 | .343 | 3674 | 201 | 82 | 12 1.68 |
| 30 | 43.60 | 14.76 | 28.84 | .339 | 2779 | 152 | 49 | 73 -24 1.79 |
| 25 | 44.24 | 14.73 | 29.51 | .333 | 1983 | 109 | 35 | 78 -43 1.93 |
| 20 | 45.01 | 14.71 | 30.30 | .327 | 1241 | 68 | 28 | 83 -55 2.10 |
| 15 | 45.86 | 14.70 | 31.16 | .321 | 705 | 38 | 12 | 88 -76 2.30 |
| 10 | 46.74 | 14.70 | 32.04 | .315 | 338 | 18 | 7 | 92 -85 2.52 |
| 5 | 47.57 | 14.70 | 32.87 | .309 | 108 | 5 | 5 | 96 -90 2.76 |
| 0 | 48.20 | 14.70 | 33.50 | .305 | 0 | 0 | 628 | 88 539 3.00 |

SEATING + BEARING + HUB SEAL TORQUE (H/M)=

628 IN-LBS @ 0 DEG.

MAX.DYN. - BEARING - HUB SEAL TORQUE (H/M) =

489 IN-LBS @ 70 DEG.

B. Valve Stress Analysis

The Pratt butterfly valve furnished was specifically designed for the requirements of the original order which did not include specific LOCA conditions.

The valve stress analysis consists of two major sections: 1) the body analysis, and 2) all other components.

The body is analyzed per rules and equations given in paragraph NB 3545 of Section III of the ASME Boiler and Pressure Vessel Code. The other components are analyzed per a basic strength of materials type of approach. For each component of interest, tensile and shear stress levels are calculated. They are then combined using the formula:

$$S_{\max} = \frac{1}{2}(T_1+T_2) + \frac{1}{2} \sqrt{(T_1+T_2)^2 + 4(S_1+S_2)^2}$$

where

S_{\max} = maximum combined stress, psi

T_1 = direct tensile stress, psi

T_2 = tensile stress due to bending, psi

S_1 = direct shear stress, psi

S_2 = shear stress due to torsion, psi

The calculated maximum valve torque resulting from LOCA conditions is used in the seismic stress analysis, attachment #2, along with "G" loads per design specification. The calculated stress values are compared to code allowables if possible, or LOCA allowables of 90% of the yield strength of the material used.

C. OPERATOR EVALUATION

Model: Bettis 521C-SR 60

Rating: 7,000 in-lbs. at full open and closed positions only.

4620 in-lbs at 68°

4000 in-lbs at 45° (minimum rating)

Max. Valve Torque: 628 in-lbs.

The maximum torque generated during a LOCA induces reactive forces in the load carrying components of the actuator.

Since the LOCA induced torque derived in this analysis is less than the maximum absorption rating of the operator, it is concluded that the Bettis models furnished are structurally suitable to withstand combined LOCA and seismic loads.

IV. Conclusion

It is concluded that the valve structure and the valve actuator are both capable of withstanding combined seismic and LOCA-induced loads based on the calculated torques developed in this analysis.

ATTACHMENT 1A

PRATT PROPOSAL LETTER

PRATT**HENRY PRATT COMPANY**

creative engineering for fluid systems

401 SOUTH HIGHLAND AVENUE • AURORA, ILLINOIS 60507

April 16, 1981

Bechtel Power Corporation
P.O. Box 3965
San Francisco, CA 94119

Attention: Mr. E.B. Poser
Project Engineer

SUBJECT: Susquehanna Steam Electric Station
Containment Isolation/Purge Valve Analysis

Gentlemen:

With reference to your recent inquiry regarding suitability of the valves and actuators to withstand aerodynamic LOCA conditions, please note the following:

1. Torque calculations will be performed for aerodynamic torque generated as a result of LOCA. These calculations will be performed using the following data to be furnished by you.

- A. Containment Pressure - Time Curves
- B. Containment Temperature - Time Curves
- C. The combined resistance coefficient for all ventilation system components downstream of the valve (one for each valve size) or

A graph of back pressure vs. LOCA time at a distance 10-12 diameters downstream of the valve. Consider also the capacity of the piping, filter and duct work to resist increases in back pressure.

- D. Maximum and minimum delay times from LOCA to initiation of valve rotation.

- E. Drawings or written description of valve orientation with respect to elbow immediately upstream of valve (within 6 diameters), as well as direction of valve closure (clockwise or counterclockwise) as viewed from operator end.

April 16, 1981

PRATT

In the absence of the above information, the following assumptions will apply to the purge valve analysis.

1. Back pressure of 19.7 psia throughout valve closing cycle. Higher back pressure increases maximum dynamic torque and valve stresses.
 2. Delay time from LOCA to initiation of valve rotation shall be chosen to permit initial sonic flow condition and critical valve disc angle to coincide, resulting in maximum possible dynamic torque.
 3. 90° elbow immediately upstream, oriented 90° out-of-plane with respect to valve shaft, with leading edge of disc closing away from outside radius of elbow. Such orientation and closure will increase torque values by 20% or more.
 2. Based on the above results, a static load stress analysis will be provided for valve components affected by the dynamic torque loadings in combination with pressure and seismic loads.
- The actuator supplier will be asked to verify the suitability of the actuator for the reaction or back drive force resulting from aerodynamic torque conditions.
3. The cost of performing the evaluation of the valve components will be \$12,800 each size for 6", 18" and 24" valves.
 4. The completion of this analysis is projected to be twenty-six (26) weeks after receipt of purchase order and data requested above based on availability of engineering schedule.
 5. Our response to NRC's criteria for demonstrating operability of purge valves is included in the analysis.

This proposal is for investigative analysis only and is not intended to guarantee the adequacy of the equipment as furnished when subjected to LOCA loads currently being defined.

The proposal is valid for thirty (30) days. The terms of payment will be Net 30 Days.

We hope you will find the proposal responsive to your needs. If we can be of any additional assistance in this matter, please advise.

Very truly yours,

HENRY PRATT COMPANY


Glenn L. Beane

Manager-Application Engineering

GLB/tl

ATTACHMENT 1B

CUSTOMER/ENGINEER RESPONSE TO
REQUEST FOR INFORMATION

APPLICATION FORM

JUN 19 1981

Bechtel Power Corporation

Engineers—Constructors

Henry Pratt Company
401 South Highland Avenue
Aurora, Illinois 60507

Fifty Beale Street

San Francisco, California

Mail Address: P.O. Box 3965, San Francisco, CA 94119



Attention: Mr. G. L. Beane

Subject: Susquehanna Steam Electric Station
Units 1 and 2 Job 8856
P.O. 8856-P-31-AC, Containment
Isolation/Purge Valve Analysis

JUN 15 '81 141808

Pratt 50*
34933

Gentlemen:

In order to perform the analysis Henry Pratt requested certain information. The following is our reply:

- A. Containment pressure time curve is attached.
- B. Containment temperature time curve is attached.
- C. A back pressure of 19.7 psia should be used in this analysis. This back pressure is per the assumptions in your letter of April 16, 1981.
- D. Minimum delay time is 0.1 seconds. Maximum delay time is 5 seconds.
- E. Isometric drawings for both units are attached. We believe that Henry Pratt is in a better position to determine the direction of valve closure as viewed from the operator end. This information is not apparent on the drawings you submitted to Bechtel.

In addition, if Henry Pratt's 16 week analysis report shows the valves to be unqualified, Henry Pratt will state at what angle the valves must be blocked open in order to meet the NRC's interim position. Henry Pratt will also make recommendations on how to block the valves and to provide a detailed drawing of the stop.

We trust that the foregoing information is satisfactory and will enable you to complete the qualification of the subject valves. If you have any questions, please contact Al Daily at (415) 768-9235 or A. Tiongson at (415) 768-7770.

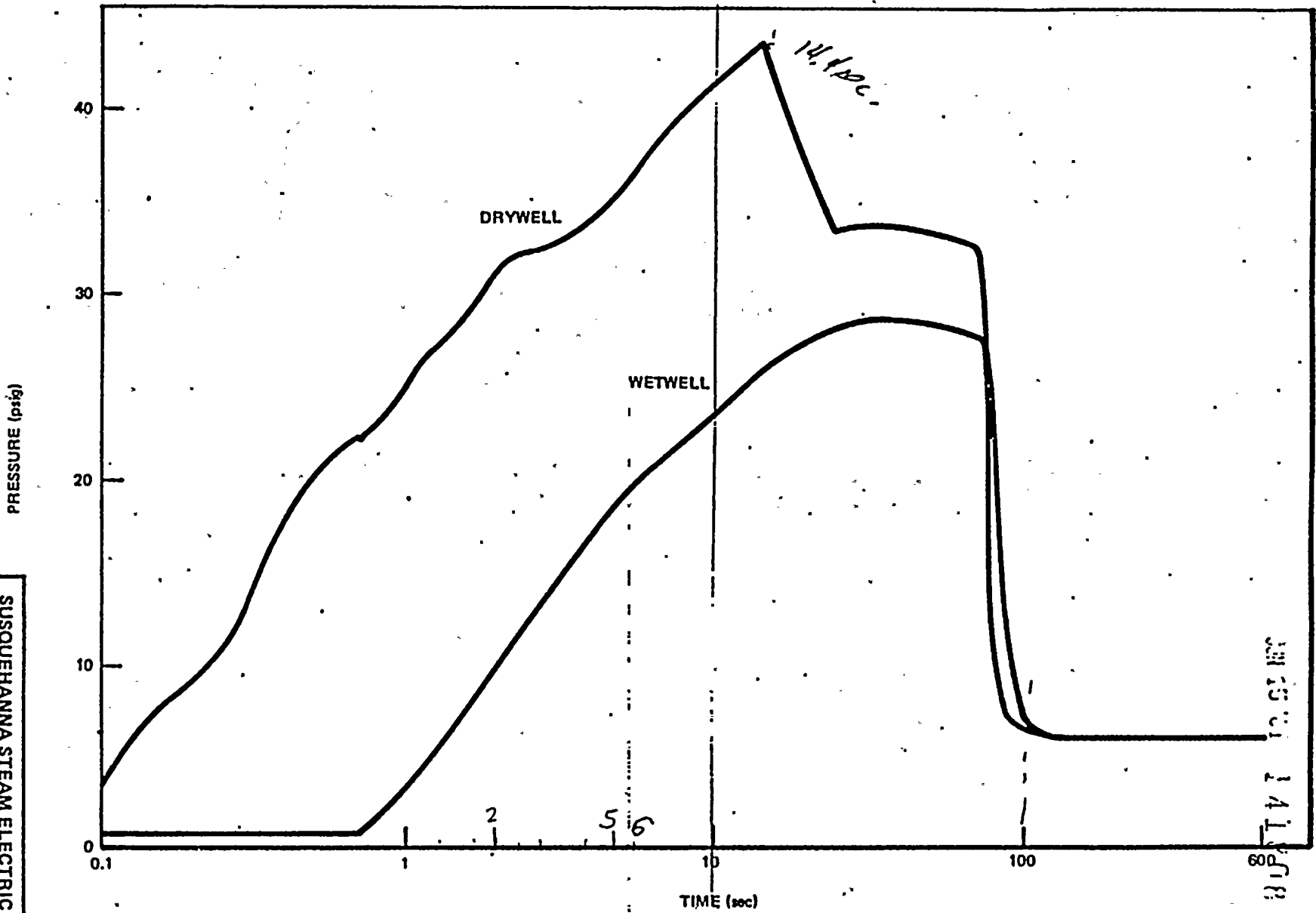
Very truly yours,

E. B. Poser
Project Engineer

Written Response Req'd: No
Design Document Changes: No
CHN/APT/cgs
WP30/3-1

cc: Mr. T. M. Crimmins, Jr. (PL) w/a

cc T. Wro NA Pratt Engr.



SUSQUEHANNA STEAM ELECTRIC STATION
UNITS 1 AND 2
FINAL SAFETY ANALYSIS REPORT

PRESSURE RESPONSE FOR
RECIRCULATION LINE BREAK

FIGURE 6.2-2

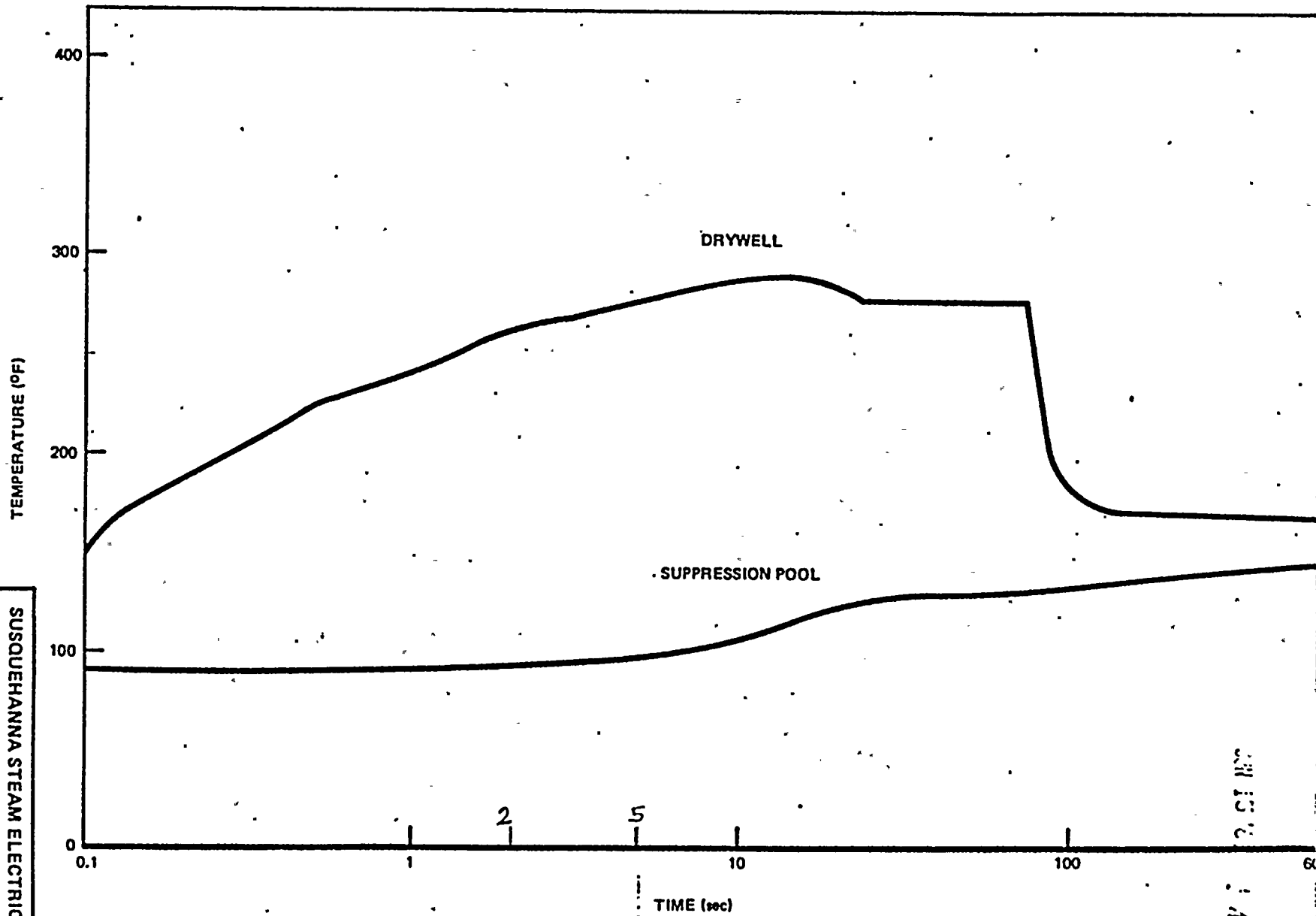
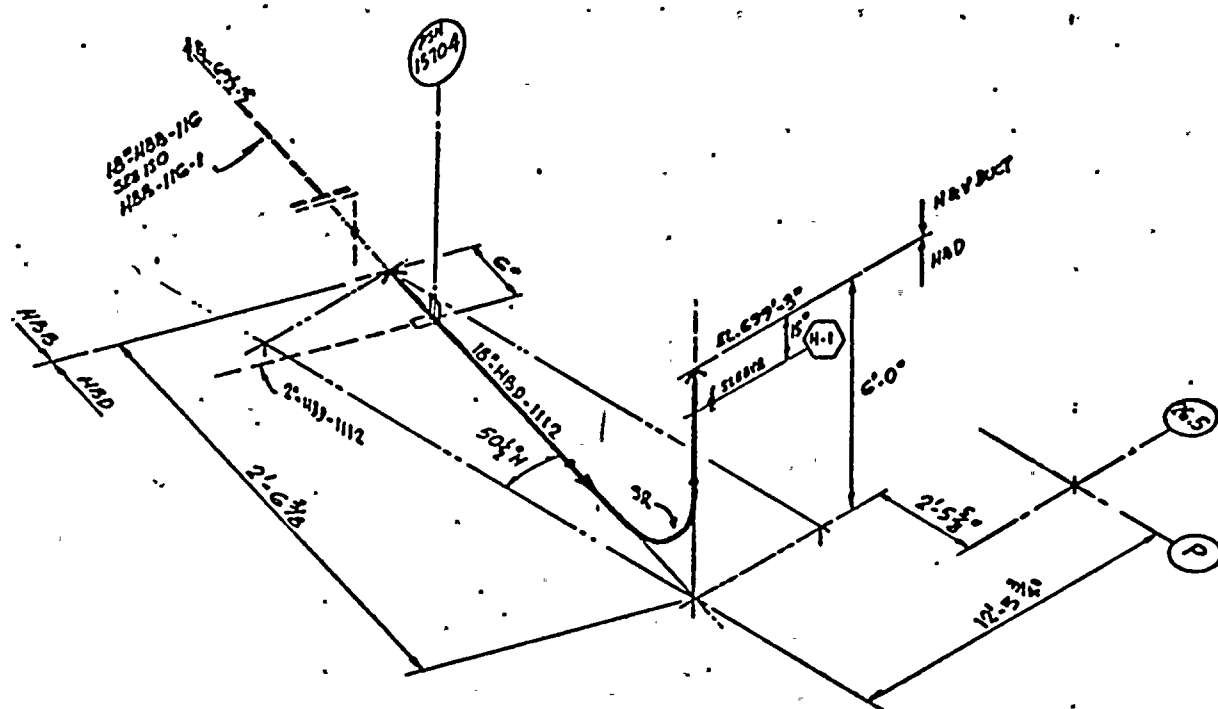
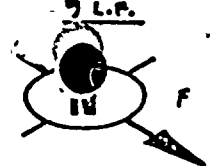


FIGURE 6.2-3

SUSQUEHANNA STEAM ELECTRIC STATION
UNITS 1 AND 2
FINAL SAFETY ANALYSIS REPORT
TEMPERATURE RESPONSE FOR
RECIRCULATION LINE BREAK



REFERENCE DRAWINGS

M- 157-01 REV. 8 P.A.I.D.
M- 27-3 15\"/>

ORIGINALLY ISSUED AS PART OF
150 HBB-11G-1

| | | | | | | |
|--|------|-------------------|----|------|-------------|------|
| DESIGNED BY | DATE | REVISIONS | BY | DATE | APPROVED BY | DATE |
| PENNSYLVANIA POWER & LIGHT COMPANY ALLENTOWN, PENNSYLVANIA BUCKLEHAMPTON STEAM ELECTRIC STATION - UNIT 1, UNIT 2 BECHTEL - SAN FRANCISCO | | | | | | |
| ISOMETRIC - REACTOR BLDG. CONTAINMENT ATM. CONTROL - UNIT 1 | | | | | | |
| | | 157-3 DRAWING NO. | | 8858 | | 2 |

150 REAC 20 CTMT ATMOS CON



NO PIPE SUPPORTS REQ'D


**COMPONENTS ON
THIS DWG. ARE Q-LISTED
SPEC. 8856-G-9 APPLIES**

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[illegible]

PENNSYLVANIA POWER & LIGHT COMPANY
ALLEGANY, PENNSYLVANIA
BRIDGEHARNA STEAM ELECTRIC STATION - UNIT 2, UNIT 3
SCHTEL - SAN FRANCISCO

ISOMETRIC - REACTOR BLDG.
CONTAINMENT ATMOS. CONTROL - UNIT 1

| | | | | |
|---|---------|-----------|---------------|----|
|  | 700 No. | 57.9 | Operating No. | 01 |
| | 8858 | HBB-117-1 | | G |


REVISED SPOOL No. 3. DIM. 5' 6 1/2" WAS
2' 9 1/2", 3' 3 1/2" WAS 7' 3 1/2" AND ADDED
OFF-SET.

M-157 SH-1 REV. 4 P.E.I.D.
M-29-B REV. 28 PIPING PLAN AREA-29

[illegible]

| | | | |
|--|------------|-----------|------|
| SCALE ~ | DESIGNED ~ | DRAWN DKB | DATE |
| PENNSYLVANIA POWER & LIGHT COMPANY ALLENTOWN, PENNSYLVANIA SCRUBHURST STEAM ELECTRIC STATION - UNIT 2, UNIT 3 | | | |
| BENTLEY - SAN FRANCISCO | | | |

ISOMETRIC - REACTOR BLDG.
CONTAINMENT ATMOS. CONTROL UNIT-1

| | | | | |
|---|-------|-----------|----------|---------|
|  | NO. 1 | 57-5 | SEARCHED | INDEXED |
| | 8858 | HBB-118-3 | | 5 |




| M-157 SN.1 | REV.4 | P.F.I.D. | | | |
|------------|-------|----------|------|--------|----|
| M-27-1 | - 13 | P-ING | PLAN | - AREA | 27 |
| M-29-1 | - 12 | " | " | " | 29 |
| M-29-2 | - 11 | " | " | " | " |
| M-29-3 | - 9 | " | " | " | " |

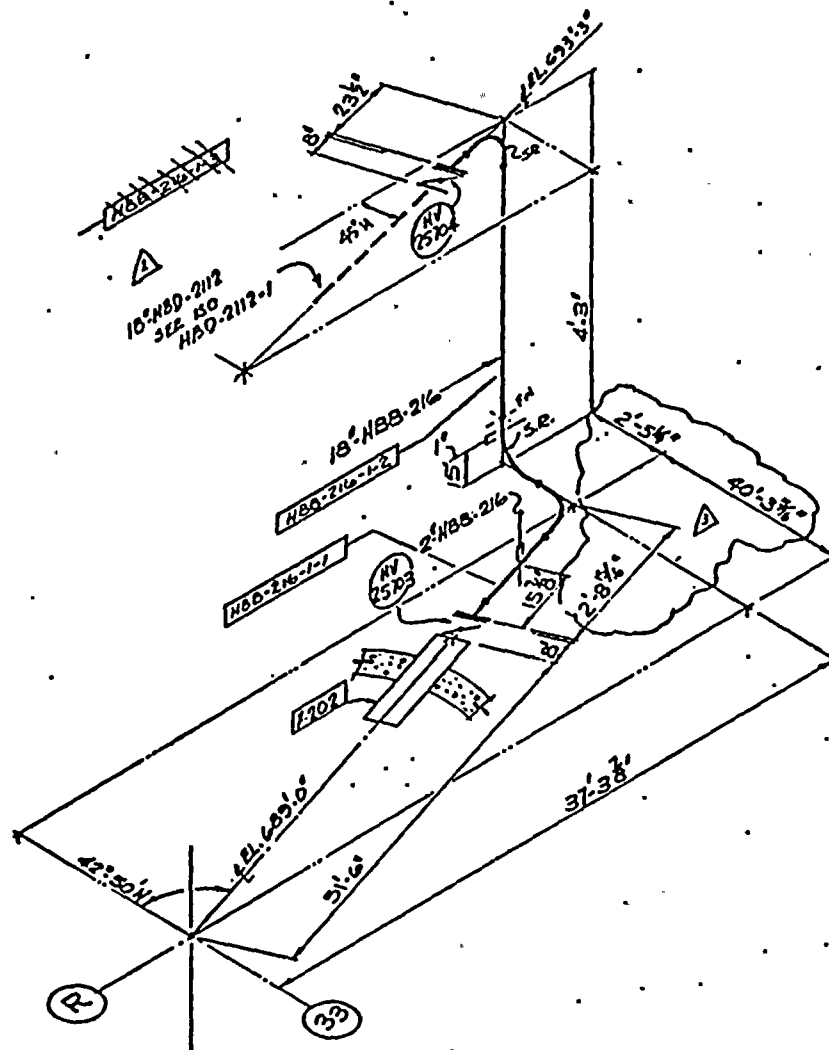
| | | | | | | | |
|---|------|--|---|----|----|----|----|
| Q | 8-17 | Wrote 2 letters and sent to the FBI. | Q | 10 | 11 | 12 | 13 |
| A | 8-17 | 1cc John & MARY MAE 1917 | Q | 11 | 12 | 13 | 14 |
| A | 8-17 | John & MARY MAE 1917 1918 and 1919 | Q | 12 | 13 | 14 | 15 |
| A | 8-17 | 1cc Gino SUMMERS | Q | 13 | 14 | 15 | 16 |
| A | 8-17 | Letter from mother about the construction | Q | 14 | 15 | 16 | 17 |
| A | 8-17 | Letter for FABRICATION | Q | 15 | 16 | 17 | 18 |
| Q | 8-17 | 8-17 | Q | 16 | 17 | 18 | 19 |

PENNSYLVANIA POWER & LIGHT COMPANY
ALLERTOWN, PENNSYLVANIA
BROOKHAVEN STEAM ELECTRIC STATION - UNIT 1, UNIT 2
BECHTEL - SAN FRANCISCO


ISOMETRIC - REACTOR BLDG.
CONTAINMENT ATMOS. CONTROL - UNIT 1

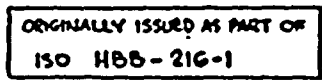
| | | | | |
|---|---------|------|------------|-----|
|  | FIG No. | 57.5 | DESIGN No. | 001 |
| | 8858 | | H20-122-1 | 5 |





M-2157541 REK 0 P/10
M-30.3 11 PIPING PLAN AREA 80

| | | |
|---|-------------------|-----------|
|  | 8858 HBB-214-1 | 57-4 3 |
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M-2157 44-1 REV.0 P.21.D.
M-30-3 " FILING PLAN - AREA 30.

02 150 REFNO DOCTMT ATMOS CON M

17 =



REV.  NOTE
ROTATE VALVE NY-25713

REFERENCE DRAWINGS

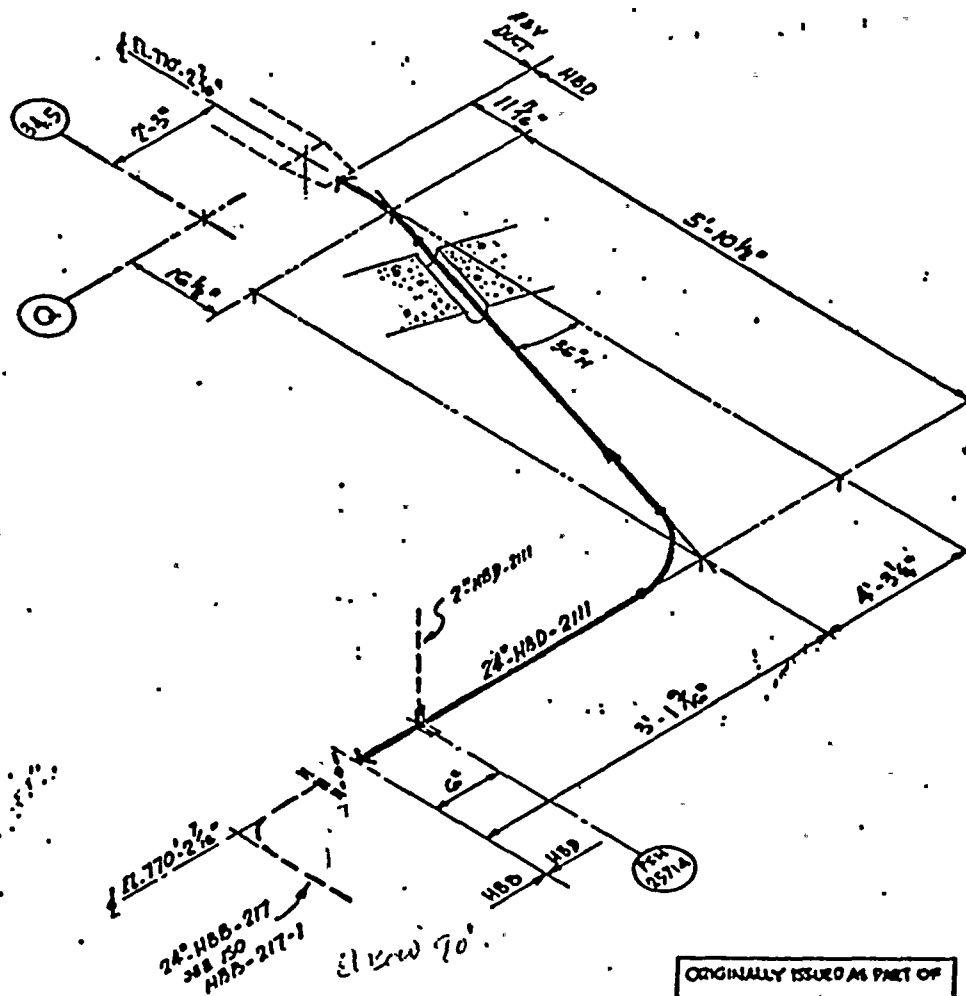
M-2157SK1 REV 3 P11D
M-32.5 (20) PIPING PLAN-AREA 32

**COMPONENTS ON
THIS DWG. ARE Q-LISTED
SPEC. 8856-G-9 APPLIES**

[illegible]

| | | |
|-----|----------------------------|---|
| 121 | ISO REAC AD CINT ATMOS CON | W |
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REFERENCE DRAWINGS

M-2157 and REV. O P.B.D.
M-72-9 PIPING PLAN - AREA 92

ORIGINALLY ISSUED AS PART OF
ISO HBB-217-1

| | | | | | |
|--|------------------------|-------------------|----|----------|------|
| △ | REVISION | DATE | BY | CHKD | APPD |
| 1 | ISSUED FOR FABRICATION | 2/1 | | | |
| <p>DESIGNED BY: R. J. [Signature]</p> <p>CHECKED BY: [Signature]</p> <p>APPROVED BY: [Signature]</p> | | | | | |
| <p>PENNSYLVANIA POWER & LIGHT COMPANY</p> <p>ALLENTOWN, PENNSYLVANIA</p> <p>BUCKLEUPHANA STEEL ELECTRIC STATION - UNIT 2, UNIT 3</p> <p>BECHTEL - SAN FRANCISCO</p> | | | | | |
| <p>ISOMETRIC - REACTOR BLDG</p> <p>CONTAINMENT ATM. CONTROL - UNIT 2</p> | | | | | |
| <p>8150</p> | | <p>HBD-2111-1</p> | | <p>1</p> | |

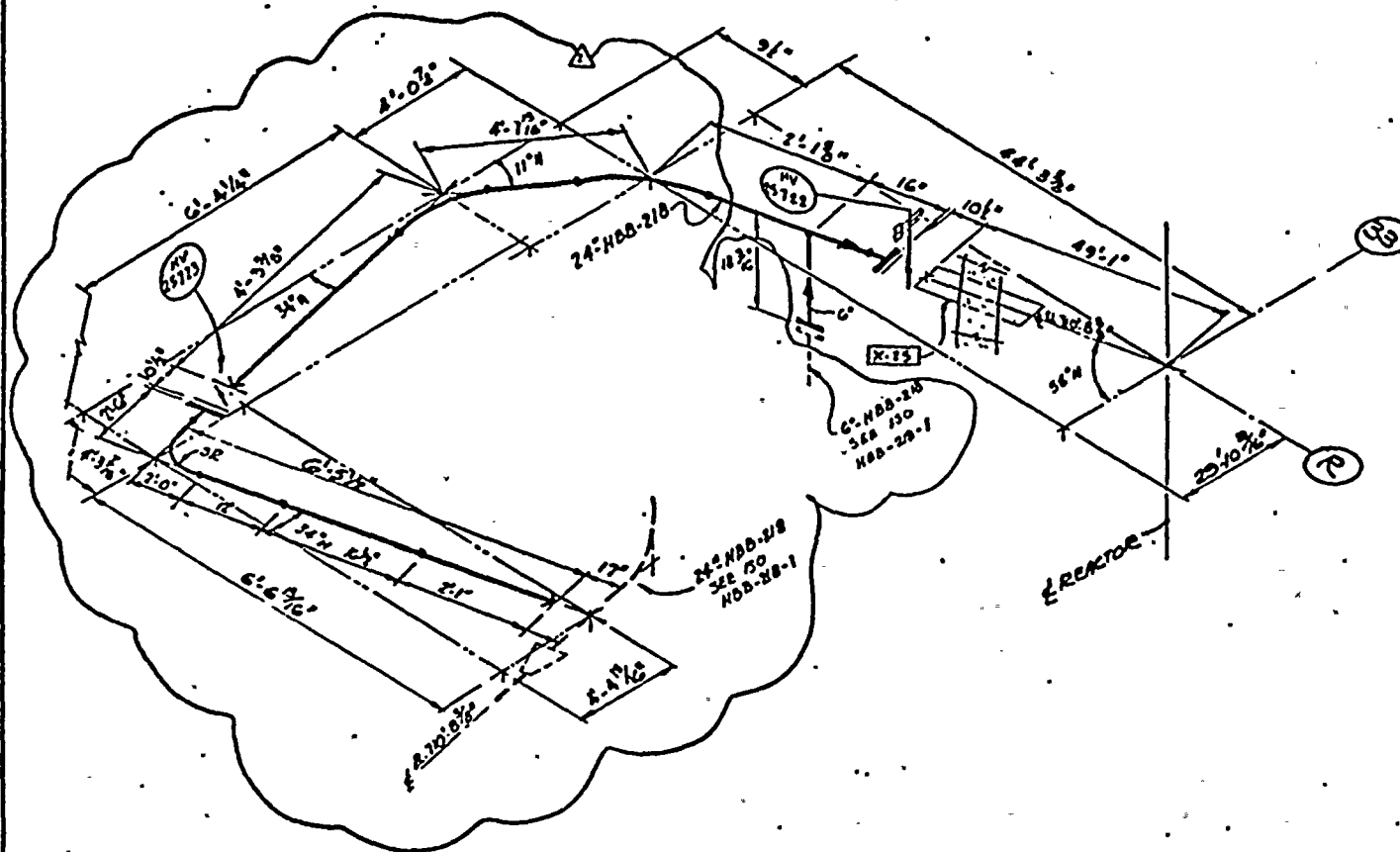
ISO HBB-217-1



M-2157 SH.1 REY. PLID
M-94-3 " II PIPING PLAN-AREASH

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REFERENCE DRAWINGS

M-2151-3H.1
M-34-3



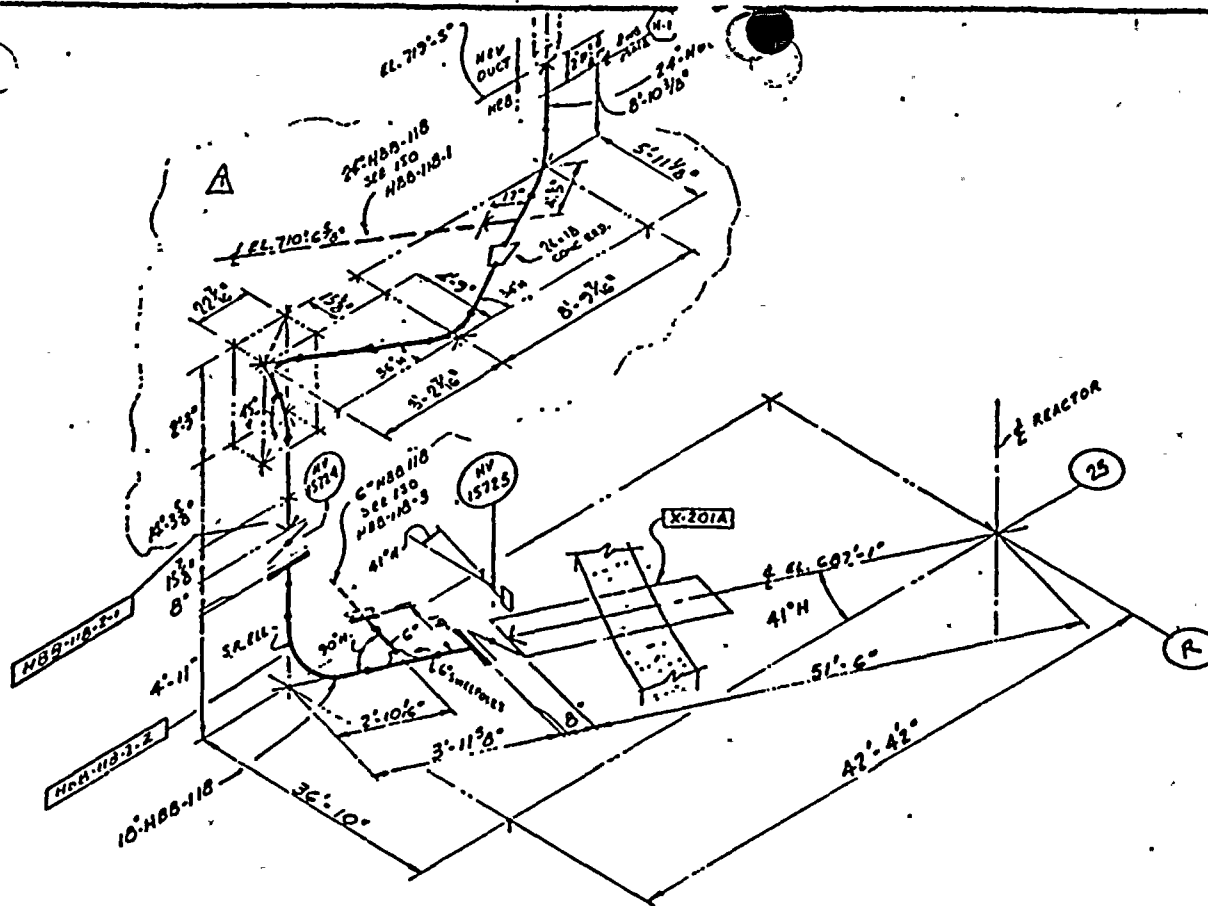
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PIPING PLAN-ARM 34

COMPONENTS ON
THIS DWG. ARE Q-LISTED
SPEC. 8858-G-9 APPLIES

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| ITEM | REVISION | DATE | BY | CHKD | APPD |
| 1 | ISSUED FOR FABRICATION | 10/1/58 | W. J. H. | W. J. H. | W. J. H. |
| <p>PENNSYLVANIA POWER & LIGHT COMPANY ALLENTOWN, PENNSYLVANIA BUCKLEHAMMA STEAM ELECTRIC STATION - UNIT 2, UNIT 3 BECHTEL - SAN FRANCISCO</p> | | | | | |
| <p>ISOMETRIC - REACTOR BLDG. CONTAINMENT ATMOS. CONTROL - UNIT 2</p> | | | | | |
| <p>8858</p> | | <p>57.3</p> | | <p>2</p> | |


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REV. Δ NOTE: SPOOL-11 24' OF WIRE USED 710.6 $\frac{1}{2}$ WAS 710.9.
REV. DMS TO UNIT.

REV.  NOTE: ROUTES COME TO NEW DFT. BLOC.
COMM. FOR ISO-1 FOR R-242

REFERENCE DRAWINGS

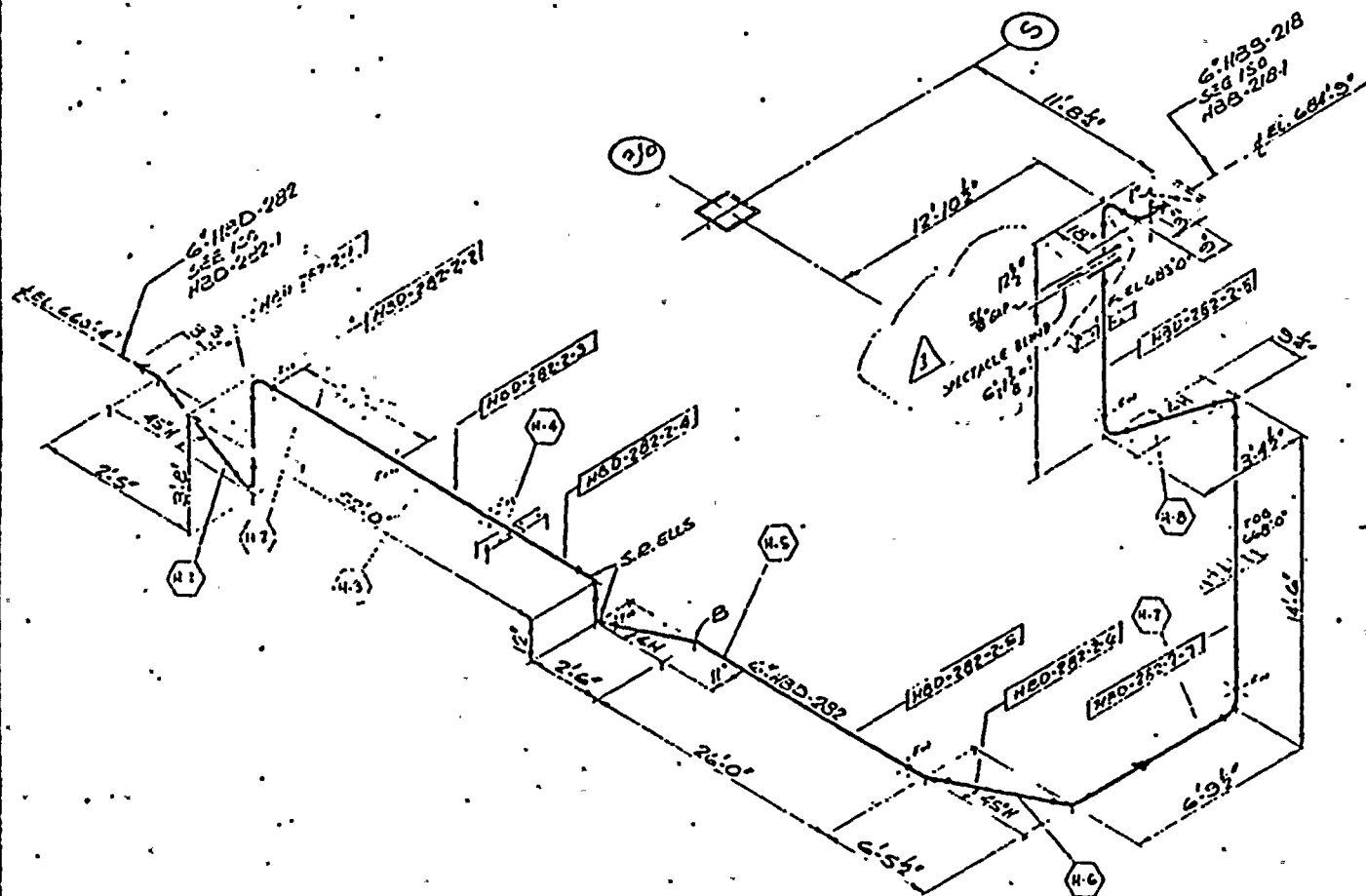
M-157 34.1 REV. 4 P.C.I.D.
M-29-3 - 11 PIPING PLAN-AREA 29

**COMPONENTS ON
THIS DWG. ARE Q-LISTED
SPEC. 8856-G-9 APPLIES**

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REFERENCE DRAWINGS

| NO. | REV. | DESCRIPTION |
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| M-2157.54.1 | REV. | PID |
| M-32.1 | 11 | PIPING PLAN AREA 32 |
| M-34.1 | 11 | 32 |
| M-34.2 | 11 | 32 |
| M-34.3 | 11 | 32 |

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| PENNSYLVANIA POWER & LIGHT COMPANY ALLENTOWN, PENNSYLVANIA BUCKLEPPA RESEARCH & DEVELOPMENT UNIT BECHTEL - SAN FRANCISCO | | | |
| ISOMETRIC - REACTOR BLDG. CONTAINMENT ATMOS. CONTROL - UNIT 2 | | | |
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M-157-1 REV 0 P. N. I. D.
M-15-8 " PIPING PLAN - AREA 88

ORIGINALLY ISSUED AS PART OF
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M-157 SN.1 REV. 4 P. & I.D.
M-29-3 - 11 PIPING PLAN - AREA 29

**COMPONENTS ON
THIS DWG. ARE Q-LISTED
SPEC. 8856-G-9 APPLIES**

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ATTACHMENT 2

Nuclear

Purge Valve

Stress

Analysis

SEISMIC ANALYSIS
FOR 6 INCH
NUCLEAR PURGE VALVE

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NOMENCLATURE

The nomenclature for this analysis is based upon the nomenclature established in paragraph NB-3534 of Section III of the ASME Boiler and Pressure Vessel Code. Where the nomenclature comes directly from the code, the reference paragraph or figure for that symbol is given with the definition. For symbols not defined in the code, the definition is that assigned by Henry Pratt Company for use in this analysis.

ANALYSIS NOMENCLATURE

| | |
|-------|--|
| A_f | Effective fluid pressure area based on fully corroded interior contour for calculating crotch primary membrane stress (NB-3545.1(a)), in ² |
| A_m | Metal area based on fully corroded interior contour effective in resisting fluid force on A_f (NB-3545.1(a)), in ² |
| A_3 | Tensile area of cover cap bolt, in ² |
| A_4 | Shear area of cover cap bolt, in ² |
| A_5 | Tensile area of trunnion bolt, in ² |
| A_6 | Shear area of trunnion bolt, in ² |
| A_7 | Tensile area of operator bolt, in ² |
| A_8 | Shear area of operator bolt, in ² |
| B_1 | Unsupported shaft length, in. |
| B_2 | Bearing bore diameter, in. |
| B_3 | Bonnet bolt tensile area, in ² |
| B_4 | Bonnet bolt shear area, in ² |
| B_5 | Bonnet body cross-sectional area, in ² |
| B_6 | Top bonnet weld size, in. |
| B_7 | Bottom bonnet weld size, in. |
| B_8 | Distance to outer fiber of bonnet from shaft on y axis, in. |
| B_9 | Distance to outer fiber of bonnet from shaft on x axis, in. |
| C | A factor depending upon the method of attachment of head, shell dimensions, and other items as listed in NC-3225.2, dimensionless (Fig. NC-3225.1 thru Fig. NC-3225.3) |
| C_b | Stress index for body bending secondary stress resulting from moment in connected pipe (NB-3545.2(b)) |
| C_p | Stress index for body primary plus secondary stress, inside surface, resulting from internal pressure (NB-3545.2(a)) |

ANALYSIS NOMENCLATURE

| | |
|----------------|--|
| C ₂ | Stress index for thermal secondary membrane stress resulting from structural discontinuity |
| C ₃ | Stress index for maximum secondary membrane plus bending stress resulting from structural discontinuity |
| C ₆ | Product of Young's modulus and coefficient of linear thermal expansion, at 500°F, psi/°F (NB-3550) |
| C ₇ | Distance to outer fiber of disc for bending along the shaft, in. |
| C ₈ | Distance to outer fiber of disc for bending about the shaft, in. |
| C ₉ | Distance to outer fiber of flat plate of disc for bending of unsupported flat plate, in. |
| d | Inside diameter of body neck at crotch region. (NB-3545.1(a)), in. |
| d _m | Inside diameter used as basis for determining body minimum wall thickness, (NB-3541), in. |
| D ₁ | Valve nominal diameter, in. |
| D ₂ | Shaft diameter, in. |
| D ₃ | Disc pin diameter, in. |
| D ₄ | Thrust collar outside diameter, in. |
| D ₅ | Spring pin diameter, in. |
| D ₆ | Cover cap bolt diameter, in. |
| D ₇ | Trunnion bolt diameter, in. |
| D ₈ | Operator bolt diameter, in. |
| D ₉ | Bonnet bolt diameter, in. |
| E | Modulus of elasticity, psi |
| F _b | Bending modulus of standard connecting pipe, as given by Figures NB-3545.2-4 and NB-3545.2-5, in ³ |
| F _d | 1/2 x cross-sectional area of standard connected pipe, as given by Figures NB-3545.2-2 and NB-3545.2-3, in. ² |
| F _N | Natural frequency of respective assembly, hertz |

ANALYSIS NOMENCLATURE

| | |
|-------|--|
| F_x | W_3g_x --Seismic force along x axis due to seismic acceleration acting on operator extended mass, pounds |
| F_y | W_3g_y --Seismic force along y axis due to seismic acceleration acting on operator extended mass, pounds |
| F_z | W_3g_z --Seismic force along z axis due to seismic acceleration acting on operator extended mass, pounds |
| g | Gravitational acceleration constant, inch-per-second ² |
| G_b | Valve body section bending modulus at crotch region (NB-3545.2(b)), in ³ |
| G_d | Valve body section area at crotch region (NB-3545.2(b)), in ² |
| G_t | Valve body section torsional modulus at crotch region (NB-3545.2(b)), in ³ |
| g_x | Seismic acceleration constant along x axis |
| g_y | Seismic acceleration constant along y axis |
| g_z | Seismic acceleration constant along z axis |
| h_g | Gasket moment arm, equal to the radial distance from the centerline of the bolts to the line of the gasket reaction (NC-3225), in. |
| H_2 | Top trunnion bolt square, in. |
| H_3 | Bottom trunnion bolt square, in. |
| H_4 | Bonnet bolt square, in. |
| H_5 | Operator bolt square, in. |
| H_6 | Bonnet bolt circle, in. |
| H_7 | Operator bolt circle, in. |
| H_8 | Bonnet height, in. |
| H_9 | Actual body wall thickness, in. |
| I_1 | Bonnet body moment of inertia about x axis, in ⁴ |
| I_2 | Bonnet body moment of inertia about y axis, in ⁴ |
| I_3 | Disc area moment of inertia for bending about the shaft in ⁴ |

ANALYSIS NOMENCLATURE

| | |
|----------------|--|
| I ₄ | Disc area moment of inertia for bending along the shaft; in ⁴ |
| I ₅ | Moment of inertia of valve body, in ⁴ |
| I ₆ | Moment of inertia of shaft, in ⁴ |
| I ₇ | Disc area moment of inertia for bending of unsupported flat plate, in ⁴ |
| J ₁ | Distance to neutral bending axis for top trunnion bolt pattern along x axis, in. |
| J ₂ | Distance to neutral bending axis for top trunnion bolt pattern along y axis, in. |
| J ₃ | Distance to neutral bending axis for bonnet bolt pattern along x axis, in. |
| J ₄ | Distance to neutral bending axis for bonnet bolt pattern along y axis, in. |
| J ₅ | Distance to neutral bending axis for operator bolt pattern along x axis, in. |
| J ₆ | Distance to neutral bending axis for operator bolt pattern along y axis, in. |
| K | Spring constant |
| K ₁ | Distance of bonnet leg from shaft centerline, in. |
| K ₂ | Thickness of disc above shaft, in. |
| K ₃ | Length along z axis to cg of bonnet plus adapter plate assembly, in. |
| K ₄ | Top trunnion width, in. |
| K ₅ | Top trunnion depth, in. |
| K ₆ | Height of top trunnion, in. |
| L ₁ | Valve body face-to-face dimension, in. |
| L ₂ | Thickness of operator housing under trunnion bolt, in. |
| L ₃ | Length of engagement of cover cap bolts in bottom trunnion, in. |
| L ₄ | Length of engagement of trunnion bolts in top trunnion, in. |

ANALYSIS NOMENCLATURE

| | |
|-----------------------------|---|
| L_5 | Bearing length, in. |
| L_6 | Length of structural disc hub welds, in. |
| L_7 | Length of engagement of bonnet bolts in adapter plate, in. |
| L_8 | Length of engagement of bonnet bolts in bonnet, in. |
| L_9 | Length of engagement of stub shaft in disc, in. |
| m | Reciprocal of Poisson's ratio |
| M | Mass of component |
| M_x | $W_3(g_y Z_0 + g_z Y_0)$, operator extended mass seismic bending moment about the x axis, acting at the base of the operator, in-lbs. |
| M_y | $W_3(g_x Z_0 + g_z X_0)$, operator extended mass seismic bending moment about the y axis, acting at the base of the operator, in-lbs. |
| M_z | $W_3(g_x Y_0 + g_y X_0)$, operator extended mass seismic bending moment about the z axis, in-lbs. |
| $\overline{M_x}$ | $M_x + F_y T_5$, operator extended mass seismic bending moment about the x axis, acting at the bottom of the adapter plate, in-lbs. |
| $\overline{M_y}$ | $M_y + F_x T_5$, operator extended mass seismic bending moment about the y axis, acting at the bottom of the adapter plate, in-lbs. |
| $\overline{\overline{M_x}}$ | $M_x + F_y (T_5 + H_8) + g_y W_4 K_3$, operator extended mass seismic bending moment about the x axis, acting at the base of the bonnet, in-lbs. |
| $\overline{\overline{M_y}}$ | $M_y + F_x (T_5 + H_8) + g_x W_4 K_3$, operator extended mass seismic bending moment about the y axis, acting at the base of the bonnet, in-lbs. |
| M_8 | Bending moment at joint of flat plate to disc hub, in-lbs. |
| N_a | Permissible number of complete start-up/shut-down cycles at hr/100°F/hr/hr fluid temperature change rate (NB-3545.3) |
| NA | Not applicable to the analysis of the system |
| N_1 | Number of top disc pins |

ANALYSIS NOMENCLATURE

| | |
|----------|--|
| N_2 | Number of operator bolts |
| N_3 | Number of trunnion bolts |
| P_d | Design pressure, psi |
| P_r | Primary pressure rating, pounds |
| P_s | Standard calculation pressure from Figure NB-3545.1-1, psi |
| P_e | Largest value among P_{eb} , P_{ed} , P_{et} , psi |
| P_{eb} | Secondary stress in crotch region of valve body caused by bending of connected standard pipe, calculated according to NB-3545.2(b), psi |
| P_{ed} | Secondary stress in crotch region of valve body caused by direct or axial load imposed by connected standard piping, calculated according to NB-3545.2(b), psi |
| P_{et} | Secondary stress in crotch region of valve body caused by twisting of connected standard pipe, calculated according to NB-3545.2(b), psi |
| P_m | General primary membrane stress intensity at crotch region, calculated according to NB-3545.1(a), psi |
| P_m' | Primary membrane stress intensity in body wall, psi |
| Q_p | Sum of primary plus secondary stresses at crotch resulting from internal pressure, (NB-3545.2(a)), psi |
| Q_T | Thermal stress in crotch region resulting from 100°F/hr fluid temperature change rate, psi |
| Q_{T1} | Maximum thermal stress component caused by through wall temperature gradient associated with 100°F/hr fluid temperature change rate (NB-3545.2(c)), psi |
| Q_{T2} | Maximum thermal secondary membrane stress resulting from 100°F/hr fluid temperature change rate, psi |
| Q_{T3} | Maximum thermal secondary membrane plus bending stress resulting from structural discontinuity and 100°F/hr fluid temperature change rate, psi |
| r | Mean radius of body wall at crotch region (NB-3545.2(c)-1), in. |
| r_i | Inside radius of body at crotch region for calculating Q_p (NB-3545.2(a)), in. |



ANALYSIS NOMENCLATURE

| | |
|---|---|
| r_2 | Fillet radius of external surface at crotch (NB-3545.2 (a)), in. |
| R_4 | Disc radius, in. |
| R_5 | Shaft radius, in. |
| R_m | Mean radius of body wall, in. |
| R_6 | Radius to O-ring in cover cap, in. |
| S | Assumed maximum stress in connected pipe for calculating P_e (NB-3545.2(b)), 30,000 psi |
| S_m | Design stress intensity, (NB-3533), psi |
| S_n | Sum of primary plus secondary stress intensities at crotch region resulting from 100°F/hr temperature change rate (NB-3545.2), psi |
| S_{p1} | Fatigue stress intensity at inside surface in crotch region resulting from 100°F/hr fluid temperature change rate (NB-3545.3), psi |
| S_{p2} | Fatigue stress intensity at outside surface in crotch region resulting from 100°F/hr fluid temperature change rate (NB-3545.3), psi |
| S(1) through S(71) are listed after the alphabetical section. | |
| t_e | Minimum body wall thickness adjacent to crotch for calculating thermal stresses (NB-3545.2(c)-1), in. |
| t_m | Minimum body wall thickness as determined by NB-3541, in. |
| T_e | Maximum effective metal thickness in crotch region for calculating thermal stresses, (NB-3545.2(c)-1), in. |
| ΔT_2 | Maximum magnitude of the difference in average wall temperatures for walls of thicknesses t_e , T_e , resulting from 100°F/hr fluid temperature change rate, °F |
| T_1 | Thickness of cover cap behind bolt head, in. |
| T_2 | Thickness of shaft behind spring pin, in. |
| T_3 | Thrust collar thickness, in. |
| T_4 | Cover cap thickness, in. |
| T_5 | Adapter plate thickness, in. |

ANALYSIS NOMENCLATURE

| | |
|----------------|---|
| T ₆ | Thickness of bottom bonnet plate, in. |
| T ₇ | Thickness of top bonnet plate, in. |
| T ₈ | Maximum required operating torque for valve, in-lbs. |
| U ₁ | Area of bottom bonnet weld, in ² |
| U ₂ | Area of top bonnet weld, in ² |
| U ₃ | Shaft bearing coefficient of friction |
| U ₄ | Bearing friction torque due to pressure loading (shaft journal bearings) |
| U ₅ | Bearing friction torque due to pressure loading plus seismic loading (shaft journal bearings) |
| U ₆ | Thrust bearing friction torque |
| V ₁ | Distances to bolts in bolt pattern on adapter plate, in. |
| V ₂ | Distances to bolts in bolt pattern on adapter plate, in. |
| V ₃ | Distances to bolts in bolt pattern on adapter plate, in. |
| V ₄ | Distances to bolts in bolt pattern on adapter plate, in. |
| V ₅ | Distance to bolts in bolt pattern on bonnet, in. |
| V ₆ | Distance to bolts in bolt pattern on bonnet, in. |
| V ₇ | Distance to bolts in bolt pattern on bonnet, in. |
| V ₈ | Distance to bolts in bolt pattern on bonnet, in. |
| W | Total bolt load, pounds |
| W ₁ | Valve weight, pounds |
| W ₂ | Banjo weight, pounds |
| W ₃ | Operator weight, pounds |
| W ₄ | Bonnet and adapter plate assembly weight, pounds |
| W ₆ | Weld size of disc structural welds, in. |

ANALYSIS NOMENCLATURE

| | |
|------------|---|
| Z_1 | Bending section modulus of bonnet welds along x-axis, in. ³ |
| Z_2 | Bending section modulus of bonnet welds along y-axis, in. ³ |
| Z_3 | Torsional section modulus of bottom bonnet welds, in. ³ |
| Z_4 | Torsional section modulus of top bonnet welds, in. ³ |
| Z_7 | Distance to edge of disc hub, inches |
| Δy | Maximum static deflection of component, inches |
| U_3 | Shaft bearing coefficient of friction |
| U_4 | Bearing friction torque due to pressure loading (shaft journal bearings) |
| U_5 | Bearing friction torque due to pressure loading plus seismic loading (shaft journal bearings) |
| U_6 | Thrust bearing friction torque |

ANALYSIS NO ENCLATURE

- S(1) Combined bending stress in disc, psi
- S(2) Bending stress in disc due to bending along the shaft, psi
- S(3) Bending stress in disc due to bending about the shaft, psi
- S(4) Shear tear out of shaft through disc, psi
- S(5) Combined stress in shaft, psi
- S(6) Combined bending stress in shaft, psi
- S(7) Combined shear stress in shaft, psi
- S(8) Bending stress in shaft due to seismic and pressure loads along x axis, psi
- S(9) Bending stress in shaft due to seismic load along y axis, psi
- S(10) Torsional shear stress in shaft due to operating loads, psi
- S(11) Direct shear stress in shaft due to pressure and seismic loads, psi
- S(12) Torsional shear stress at reduced pin cross-section, psi
- S(13) Combined shear stress in pin, psi
- S(14) Direct shear stress in pin due to seismic load, psi
- S(15) Shear stress in pin due to torsional load, psi
- S(16) Bearing stress on pin, psi
- S(17) Compressive stress on shaft bearing due to seismic and pressure loads, psi
- S(18) Shear tear out of cover cap bolt through tapped hole in bottom trunnion.
- S(19) Shear tear out of cover cap bolt through cover cap, psi

ANALYSIS NOMENCLATURE

- S(20) Combined stress in cover cap bolts, psi
- S(21) Shear stress in cover cap bolts due to torsional loading, psi
- S(22) Direct tensile stress in cover cap bolts due to seismic and pressure loads, psi
- S(23) Combined stress in cover cap, psi
- S(24) Radial stress in cover cap, psi
- S(25) Tangential stress in cover cap, psi
- S(26) Shear stress in cover cap, psi
- S(27) Bearing stress on thrust collar, psi
- S(28) Shear load on thrust collar spring pin, pounds
- S(29) Bearing stress of spring pin on thrust collar, psi
- S(30) Shear tear out of spring pin through thrust collar, psi
- S(31) Shear tear out of spring pin through bottom of the shaft, psi



ANALYSIS NOMENCLATURE

- S(32) Shear tear out of trunnion bolt through tapped hole in trunnion, psi
- S(33) Bearing stress of trunnion bolt on tapped hole in trunnion, psi
- S(34) Bearing stress of trunnion bolt on through hole in bonnet plate, psi
- S(35) Shear tear out of trunnion bolt head through bonnet plate, psi
- S(36) Combined stress in trunnion bolt, psi
- S(37) Direct tensile stress in trunnion bolt, psi
- S(38) Tensile stress in trunnion bolt due to bending moment, psi
- S(39) Direct shear stress in trunnion bolt, psi
- S(40) Shear stress in trunnion bolt due to torsional load, psi
- S(41) Shear tear out of operator bolt head through hole in bonnet, psi
- S(42) Bearing stress of operator bolt on through hole in bonnet, psi
- S(43) Combined stress in operator bolts, psi
- S(44) Direct tensile stress in operator bolts, psi
- S(45) Tensile stress in operator bolts due to bending moment, psi
- S(46) Direct shear stress in operator bolts, psi

ANALYSIS NOMENCLATURE

- S(47) Shear stress in operator bolts due to torsional loads, psi
- S(48) Combined stress in bonnet body, psi
- S(49) Direct tensile stress in bonnet body, psi
- S(50) Tensile stress in bonnet body due to bending moment, psi
- S(51) Direct shear stress in bonnet body, psi
- S(52) Shear stress in bonnet body due to torsional load, psi
- S(53) Combined shear stress in bottom bonnet weld, psi
- S(54) Total tensile stress in bottom bonnet weld, psi
- S(55) Direct tensile stress in bottom bonnet weld, psi
- S(56) Tensile stress in bottom bonnet weld due to bending moment, psi
- S(57) Total shear stress in bottom bonnet weld, psi
- S(58) Direct shear stress in bottom bonnet weld, psi
- S(59) Shear stress in bottom bonnet weld due to torsional load, psi
- S(60) Combined shear stress in top bonnet weld, psi
- S(61) Total tensile stress in top bonnet weld, psi
- S(62) Direct tensile stress in top bonnet weld, psi
- S(63) Tensile stress in top bonnet weld due to bending moment, psi
- S(64) Total shear stress in top bonnet weld, psi
- S(65) Direct shear stress in top bonnet weld, psi
- S(66) Shear stress in top bonnet weld due to torsional load, psi



ANALYSIS NOMENCLATURE

- S(67) Combined stress in trunnion body, psi
- S(68) Direct tensile stress in trunnion body, psi
- S(69) Tensile stress in trunnion body due to bending moment, psi
- S(70) Direct shear stress in trunnion body, psi
- S(71) Shear stress in trunnion body due to torsional load, psi

SUMMARY TABLE INTRODUCTION

In the following pages, the pertinent data for the butterfly valve stress analysis is tabulated in three categories:

1. Stress Levels for Valve Components
2. Natural Frequencies of Components
3. Valve Dimensional Data

In Table 1, Stress Levels for Valve Components, the following data is tabulated:

Component Name

Code Reference (when applicable)

Stress Level Name and Symbol

Analysis Reference Page

Material Specification

Actual Stress Level

Allowable Stress Level

The material specifications are taken from Section II of the code when applicable. Allowable stress levels are S_m for tensile stresses and $.6 S_m$ for shear stresses. The allowable levels are the same whether the calculated stress is a combined stress or results from a single load condition. S_m is the design stress intensity value as defined in Appendix I, Tables I-7.1 of Section III of the code.

In Table 2, Natural Frequencies of Valve Components, the following data is tabulated:

Summary Table Introduction

Component Name

Natural Frequency Symbol

Analysis Reference Page

Component Material

Natural Frequency

In Table 3, Valve Dimensional Data, the values for the pertinent valve dimensions and parameters are given.

TABLE 1

STRESS LEVELS FOR VALVE COMPONENTS

| COMPONENT | CODE REF. PARAGRAPH | SYMBOL & NAME | | REF. PAGE | MATERIAL | STRESS LEVEL, PSI | ALLOWABLE STRESS LEVEL PSI |
|-----------|------------------------|--|----------------------------------|----------------|-------------------|----------------------|----------------------------------|
| Body | NB-3545.1 | Primary membrane stress in crotch | P_m | 35 | ASME SA-516 GR.55 | 564 | S_m 13700 |
| | | Primary membrane stress in body | P'_m | 36 | ASME SA-516 GR.55 | 941 | S_m 13700 |
| | NB-3545.2 | Primary plus secondary stress due to internal pressure | Q_p | 36 | ASME SA-516 GR.55 | 2800 | S_m 13700 |
| | NB-3545.2 | Pipe Reaction stresses Axial Load Bending Load Torsional Load | P_{ed} P_{eb} P_{et} | 36 36 36 | ASME SA-516 GR.55 | 2655 4100 4100 | 1.5 S_m 20550 |
| | NB-3545.2 | Thermal Secondary stress | Q_t | 38 | ASME SA-516 Gr.55 | 1105 | S_m 13700 |
| | NB-3545.2 | Primary plus secondary stress | S_n | 38 | ASME SA-516 Gr.55 | 7109 | 3 S_m 41100 |
| | NB-3545.3 | Normal duty fatigue stress $N_a \geq 2000$ | S_p | 38 | ASME SA-516 Gr.55 | 5469 | S_m 13700 |
| Disc | NB-3546.2 | Combined bending stress in disc | $S(1)$ | 39 | ASME SA-516 Gr.55 | 1591 | 1.5 S_m 20550 |
| | NB-3546.2 | Shear tear out of shaft thru disc | $S(4)$ | 41 | ASME SA-516 Gr.55 | 916 | .6 S_m 8220 |
| | | | | | | | |

TABLE 1

STRESS LEVELS FOR VALVE COMPONENTS

| COMPONENT | CODE REF. PARAGRAPH | SYMBOL & NAME | | REF. PAGE | MATERIAL | STRESS LEVEL, PSI | ALLOWABLE STRESS LEVEL PSI |
|---------------|------------------------|--|-------|--------------|-----------------------------------|----------------------|----------------------------------|
| Shaft. | NB-3546.3 | Combined stress in shaft | S(5) | 42 | ASME SA-564 Type 630 Cond. H-1150 | 12745 | Sm 33700 |
| | NB-3546.3 | Torsional shear stress at reduced pin cross-section | S(12) | 43 | ASME SA-564 Type 630 Cond. H-1150 | 5702 | .6Sm 20220 |
| Disc Pin | NB-3546.3 | Combined shear stress in top disc pin | S(13) | 44 | ASME SA-320 Gr. B8M | 3928 | .6Sm 8160 |
| | NB-3546.3 | Bearing stress on top pin | S(16) | 44 | ASME SA-320 Gr. B8M | 1787 | Sm 13600 |
| Shaft Bearing | | Comprehensive stress on shaft bearing | S(17) | 45 | ASTM B-438 Gr. 1 Type II | 2964 | Sm 4000 |
| Cover Cap | NB-3546.1 | Shear tear out of cover cap bolt through tapped holes in bottom trunnion | S(18) | 46 | ASME SA-516 Gr.55 | 938 | .6Sm 8220 |
| | NB-3546.1 | Shear tear out of cover cap bolt head thru cover cap | S(19) | 46 | ASME SA-516 Gr.70 | 201 | .6Sm 10500 |
| | NB-3546.1 | Combined stress in cover cap bolts | S(20) | 46 | ASME SA-193 Gr.B7 | 5973 | Sm 25000 |
| | | Combined stress in cover cap | S(23) | 46 | ASME SA-516 Gr.70 | 2698 | Sm 17500 |

TABLE 1

STRESS LEVELS FOR VALVE COMPONENTS

| COMPONENT | CODE REF. PARAGRAPH | SYMBOL & NAME | | REF. PAGE | MATERIAL | STRESS LEVEL, PSI | ALLOWABLE STRESS LEVEL PSI |
|-------------------|------------------------|--|-------|--------------|--------------------------------------|----------------------|----------------------------------|
| Thrust Bearing | | Bearing stress on thrust collar | S(27) | 49 | SAE-660 | 174 | Sm 8800 |
| | | Shear load on thrust collar spring pin | S(28) | 49 | AISI 420 | 291 | Pm 1540# |
| | | Bearing stress of spring pin on thrust collar | S(29) | 49 | SAE-660 | 1147 | Sm 8800 |
| | | Shear tear out of spring pin thru bottom of shaft. | S(31) | 49 | ASME SA-564 Type 630 Cond. H-1150 | 832 | .6Sm 20220 |
| | | | | | | | |

TABLE 1

STRESS LEVELS FOR VALVE COMPONENTS

| COMPONENT | CODE REF. PARAGRAPH | SYMBOL & NAME | | REF. PAGE | MATERIAL | STRESS LEVEL, PSI | ALLOWABLE STRESS LEVEL PSI |
|----------------------|------------------------|---|-------|--------------|-------------------|----------------------|----------------------------------|
| Operator Mounting | | Shear tear out of trunnion bolt thru tapped hole in trunnion | S(32) | 50 | ASME SA-516 Gr.55 | 3333 | .6Sm 8220 |
| | | Bearing stress of trunnion bolt on tapped hole | S(33) | 50 | ASME SA-516 Gr.55 | 3604 | Sm 13700 |
| | | Bearing stress of trunnion bolt on thru hole in bonnet | S(34) | 50 | ASME SA-36 | 4505 | Sm 12600 |
| | | Shear tear out of trunnion bolt head, thru bonnet | S(35) | 52 | ASME SA-36 | 2244 | .6Sm 7560 |
| | | Combined stress in trunnion bolt | S(36) | 52 | ASTM A-193 Gr.B7 | 23917 | Sm 25000 |
| | | Shear tear out of operator bolt head thru hole in bonnet | S(41) | 52 | ASME SA-36 | 999 | .6Sm 7560 |
| | | Bearing stress of operator bolt on thru hole in bonnet | S(42) | 52 | ASME SA-36 | 3537 | Sm 12600 |
| | | Combined stress in operator bolt | S(43) | 54 | ASTM A-193 Gr.B7 | 12972 | Sm 25000 |
| | | Combined stress in bonnet body | S(48) | 54 | ASME SA-36 | 2705 | Sm 12600 |

TABLE 1

STRESS LEVELS FOR VALVE COMPONENTS

| COMPONENT | CODE REF. PARAGRAPH | SYMBOL & NAME | | REF. PAGE | MATERIAL | STRESS LEVEL, PSI | ALLOWABLE STRESS LEVEL PSI |
|--------------------------------|------------------------|--|-------|--------------|----------------------|----------------------|----------------------------------|
| Operator Mounting Cont'd | | Combined shear stress in bottom bonnet welds | S(53) | 56 | | 953 | .6Sm 7200 |
| | | Combined shear stress in top bonnet welds | S(60) | 57 | | 537 | .6Sm 7200 |
| | | Combined stress in trunnion body | S(67) | 58 | ASME SA-516 Gr.55 | 2786 | Sm 13700 |
| | | | | | | | |

Table 2 NATURAL FREQUENCIES OF VALVE COMPONENTS

| Component Name | Natural Frequency Symbol | Ref. Page | Material | Natural Frequency (Hertz) |
|----------------|--------------------------|-----------|---|---------------------------|
| Body | F_{N1} | 59 | ASME SA-516 Gr. 55 | 13825 |
| Banjo | F_{N2} | 60 | ASME SA-564 Type 630 Cond. H-1150 | 8765 |
| Cover Cap | F_{N3} | 60 | ASME SA-516 Gr. 70 | 4212 |
| Bonnet | F_{N4} | 61 | ASME SA-36 | 383 |

Job Number: D-34933(D-0026-3) Valve Size: 6"-1200Operator Mounting: TEE BONNET Operator: 521C-SR60

| | | |
|----------------------------|-----------------------------|------------------------------|
| A _f <u>9.74</u> | C ₃ <u>.50</u> | g <u>32.2</u> |
| A _m <u>6.59</u> | C ₆ <u>2.49</u> | G _b <u>48.29</u> |
| A ₃ <u>.078</u> | C ₇ <u>1.40</u> | G _d <u>30.51</u> |
| A ₄ <u>.068</u> | C ₈ <u>1.50</u> | G _T <u>96.58</u> |
| A ₅ <u>.142</u> | C ₉ <u>.50</u> | g _x <u>5</u> |
| A ₆ <u>.126</u> | d <u>6.065</u> | g _y <u>5</u> |
| A ₇ <u>.142</u> | d _m <u>6.065</u> | g _z <u>5</u> |
| A ₈ <u>.126</u> | D ₁ <u>6</u> | H ₂ <u>2.5</u> |
| B ₁ <u>.63</u> | D ₂ <u>1.0</u> | H ₃ <u>2.5</u> |
| B ₂ <u>1.25</u> | D ₃ <u>.435</u> | H ₄ <u>NIA</u> |
| B ₃ <u>NIA</u> | D ₄ <u>2.6</u> | H ₅ <u>3.188</u> |
| B ₄ <u>NIA</u> | D ₅ <u>.25</u> | H ₆ <u>NIA</u> |
| B ₅ <u>6.5</u> | D ₆ <u>.375</u> | H ₇ <u>NIA</u> |
| B ₆ <u>.25</u> | D ₇ <u>.50</u> | H ₈ <u>7</u> |
| B ₇ <u>.25</u> | D ₈ <u>.50</u> | H ₉ <u>1.317</u> |
| B ₈ <u>3.5</u> | D ₉ <u>NIA</u> | I ₁ <u>60.79</u> |
| B ₉ <u>2.5</u> | E <u>30E6</u> | I ₂ <u>10.45</u> |
| C <u>.3</u> | F _b <u>6.6</u> | I ₃ <u>4.94</u> |
| C _b <u>1</u> | F _d <u>2.7</u> | I ₄ <u>5.17</u> |
| C _p <u>3</u> | F _x <u>560</u> | I ₅ <u>209.92</u> |
| C _o <u>1.02</u> | F _y <u>560</u> | I ₆ <u>.049</u> |
| C ₂ <u>.42</u> | F _z <u>560</u> | I ₇ <u>NIA</u> |

| | | |
|----------------------------|-----------------------------|-----------------------------|
| J ₁ <u>.75</u> | M _z <u>5561</u> | ΔT ₂ <u>1</u> |
| J ₂ <u>.50</u> | M _x <u>10532</u> | T ₁ <u>1.188</u> |
| J ₃ <u>N/A</u> | M _y <u>7066</u> | T ₂ <u>.172</u> |
| J ₄ <u>N/A</u> | M _x <u>10532</u> | T ₃ <u>.75</u> |
| J ₅ <u>.50</u> | M _y <u>7066</u> | T ₄ <u>.374</u> |
| J ₆ <u>.50</u> | M ₈ <u>N/A</u> | T ₅ <u>N/A</u> |
| K ₀ <u>.86</u> | N _a <u>2000</u> | T ₆ <u>.50</u> |
| K ₁ <u>N/A</u> | N ₁ <u>N/A</u> | T ₇ <u>.50</u> |
| K ₂ <u>.50</u> | N ₂ <u>4</u> | T ₈ <u>62.8</u> |
| K ₃ <u>3.5</u> | N ₃ <u>4</u> | U ₁ <u>7.5</u> |
| K ₄ <u>3.5</u> | P _d <u>275</u> | U ₂ <u>7.5</u> |
| K ₅ <u>4.5</u> | P _r <u>150</u> | U ₃ <u>.25</u> |
| K ₆ <u>1.0</u> | P _s <u>285</u> | V ₁ <u>N/A</u> |
| L ₁ <u>5</u> | Q _{T1} <u>1000</u> | V ₂ <u>N/A</u> |
| L ₂ <u>N/A</u> | r <u>1.203</u> | V ₃ <u>N/A</u> |
| L ₃ <u>.468</u> | r _i <u>3.03</u> | V ₄ <u>N/A</u> |
| L ₄ <u>.625</u> | r ₂ <u>1.0</u> | V ₅ <u>N/A</u> |
| L ₅ <u>1.0</u> | R ₄ <u>2.6</u> | V ₆ <u>N/A</u> |
| L ₆ <u>N/A</u> | R ₅ <u>.5</u> | V ₇ <u>N/A</u> |
| L ₇ <u>N/A</u> | R _m <u>3.69</u> | V ₈ <u>N/A</u> |
| L ₈ <u>.50</u> | R ₆ <u>1.4</u> | W ₁ <u>12.4</u> |
| L ₉ <u>5.2</u> | S <u>30,000</u> | W ₂ <u>12.5</u> |
| m <u>3.5</u> | t _e <u>1.092</u> | W ₃ <u>112</u> |
| M _x <u>6087</u> | t _m <u>.28</u> | W ₄ <u>30</u> |
| M _y <u>2621</u> | T _e <u>1.317</u> | W ₆ <u>N/A</u> |

W₇ 9
W₈ NIA
X₀ 1.87
Y₀ 8.06
Z₀ 2.81
Z₁ 21.46
Z₂ 6.14
Z₃ 23.76
Z₄ 23.76
Z₇ NIA

Standard Stress Report

for

NRS Butterfly Valve

with

Bonnet Mounted

Cylinder Operator

ANALYSIS INTRODUCTION

Described in the following pages is the analysis used in verifying the structural adequacy of the main elements of the NRS butterfly valve. The analysis is structured to comply with Paragraph NB-3550 of Section III of the ASME Boiler and Pressure Vessel Code (hereafter referred to as the code). In the analysis, the design rules for Class 1 valves are used, since the requirements for this class of valve is much more explicit than for either Class 2 or 3 design rules. The design rules for Class 2 and 3 are exceeded by the rules for Class 1 valves.

Valve components are analyzed under the assumption that the valve is either at maximum fluid dynamic torque or seating against the maximum design pressure. Analysis temperature is 300°F. Seismic accelerations are simultaneously applied in each of three mutually perpendicular directions.

Seismic loads are made an integral part of the analysis by the inclusion of the acceleration constants g_x , g_y , g_z . The symbols g_x , g_y , g_z represent accelerations in the x, y and z directions respectively. These directions are defined with respect to the valve body centered co-ordinate system as illustrated in Figure 1. Specifically, the x axis is along the pipe axis, the z axis is along the shaft axis, and the y axis is mutually perpendicular to the x and z axes, forming a right hand triad with them.

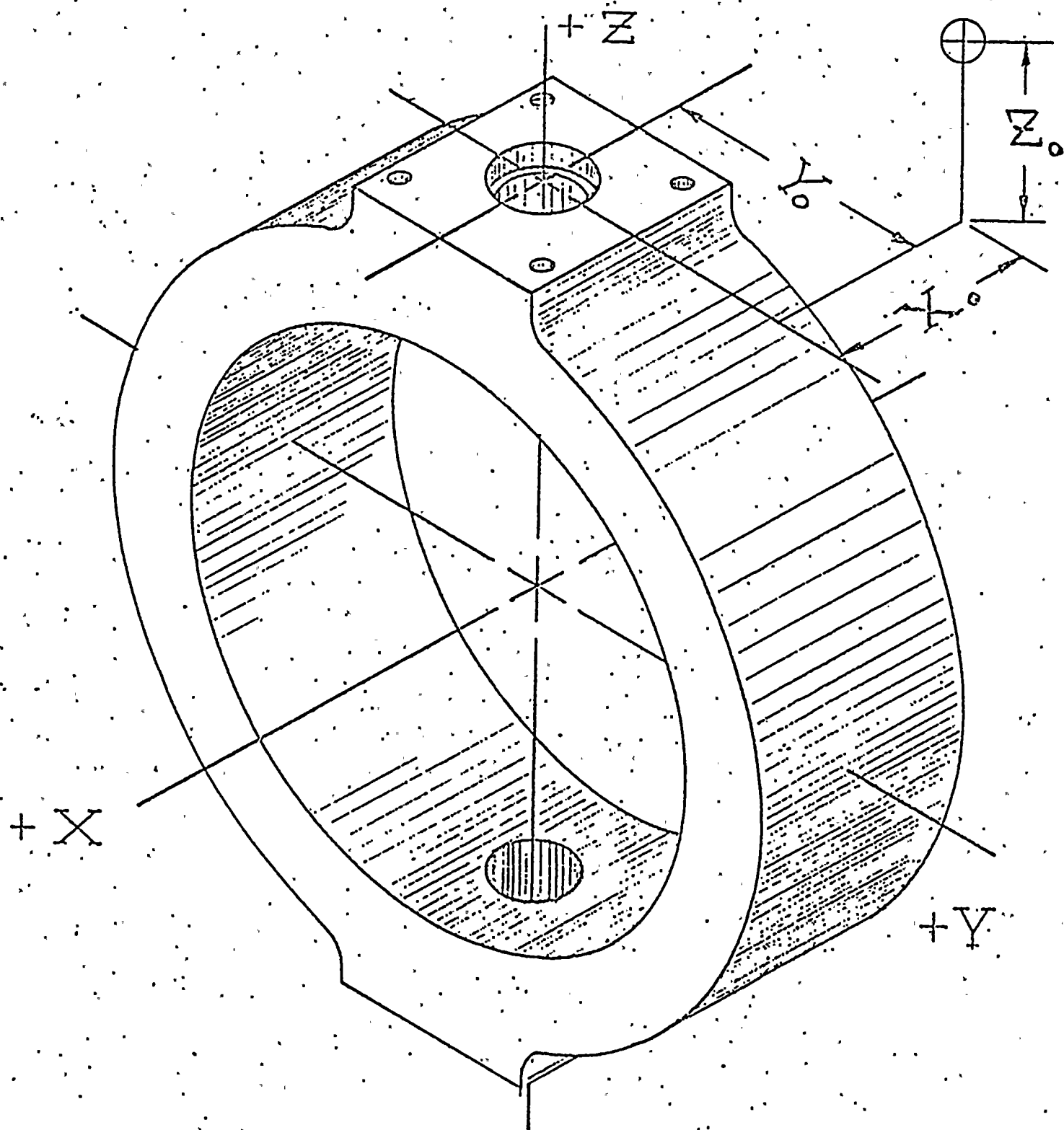


Figure 1. VALVE BODY SPATIAL ORIENTATION

Analysis Introduction

Valve orientation with respect to gravity is taken into account by adding the appropriate quantity to the seismic loads. The justification for doing this is that a gravitational load is completely equivalent to a 1g seismic load.

The analysis of each main element or sub-assembly of the butterfly valve is described separately in an appropriately titled section. In addition to containing sketches where appropriate, each section contains an explanation of the basis for each calculation. Where applicable, it also contains an interpretation of code requirements as they apply to the analysis.

Figure 2 is a cross-section view of the butterfly valve, and its associated components. Detailed sketches are provided throughout the report to clearly define the geometry.

Disc Analysis

Shear Tear Out of Shaft

The disc is designed so the minimum thickness of material surrounding the shaft extension in the disc is above the shaft on the arch side. The loading is due to both seismic and pressure loads.

$$S(4) = \frac{\pi P_s R_4^2 + W_2 \sqrt{g_x^2 + g_y^2 + g_z^2}}{2L_9(K_2 + D_2(1 - \sin 45^\circ))} = \text{Shear tear out shaft through disc, psi.}$$

SHAFT ANALYSIS

The shaft is analyzed in accordance with Paragraph NB-3546.3 of Section III of the Code. The shaft loading is a combination of seismic, pressure and operating loads. Maximum torsional loading is either a combination of seating and bearing torque or bearing and dynamic torque. Columnar stress is not considered in the shaft loading due to its negligible effect on the stress levels. Figure 2 shows the banjo assembly with the through shaft.

Shaft stresses due to pressure, seismic and operating loads:

$$S(5) = \frac{S(6)}{2} + \frac{(S(6)^2 + 4 S(7)^2)^{\frac{1}{2}}}{2}$$

where

$$S(6) = (S(8)^2 + S(9)^2)^{\frac{1}{2}} \quad = \text{Combined bending stress, psi}$$

$$S(8) = \frac{(\pi R_4^2 P_s + W_2 g_x) \cdot 25 B_1 R_5}{\pi \cdot 25 R_5^4} \quad = \text{Bending tensile stress due to pressure and seismic loads along x axis, psi}$$

$$S(9) = \frac{.25 W_2 g_y B_1 R_5}{.25 \pi R_5^4} \quad = \text{Bending tensile stress due to seismic loads along y axis, psi}$$

$$S(7) = (S(10)^2 + S(11)^2)^{\frac{1}{2}} \quad = \text{Combined shear stress, psi}$$

$$S(10) = \frac{T_8 R_5}{.5 \pi R_5^4} \quad = \text{Torsional shear stress, psi}$$

$$S(11) = 1.333 \left[\frac{.5 \pi R_4^2 P_s + .5 W_2 (g_x^2 + g_y^2)^{\frac{1}{2}}}{\pi R_5^2} \right] \quad = \text{Direct shear stress, psi}$$

Also worthy of attention is the torsional shear stress at the reduced cross-section where the pin passes through the shaft.

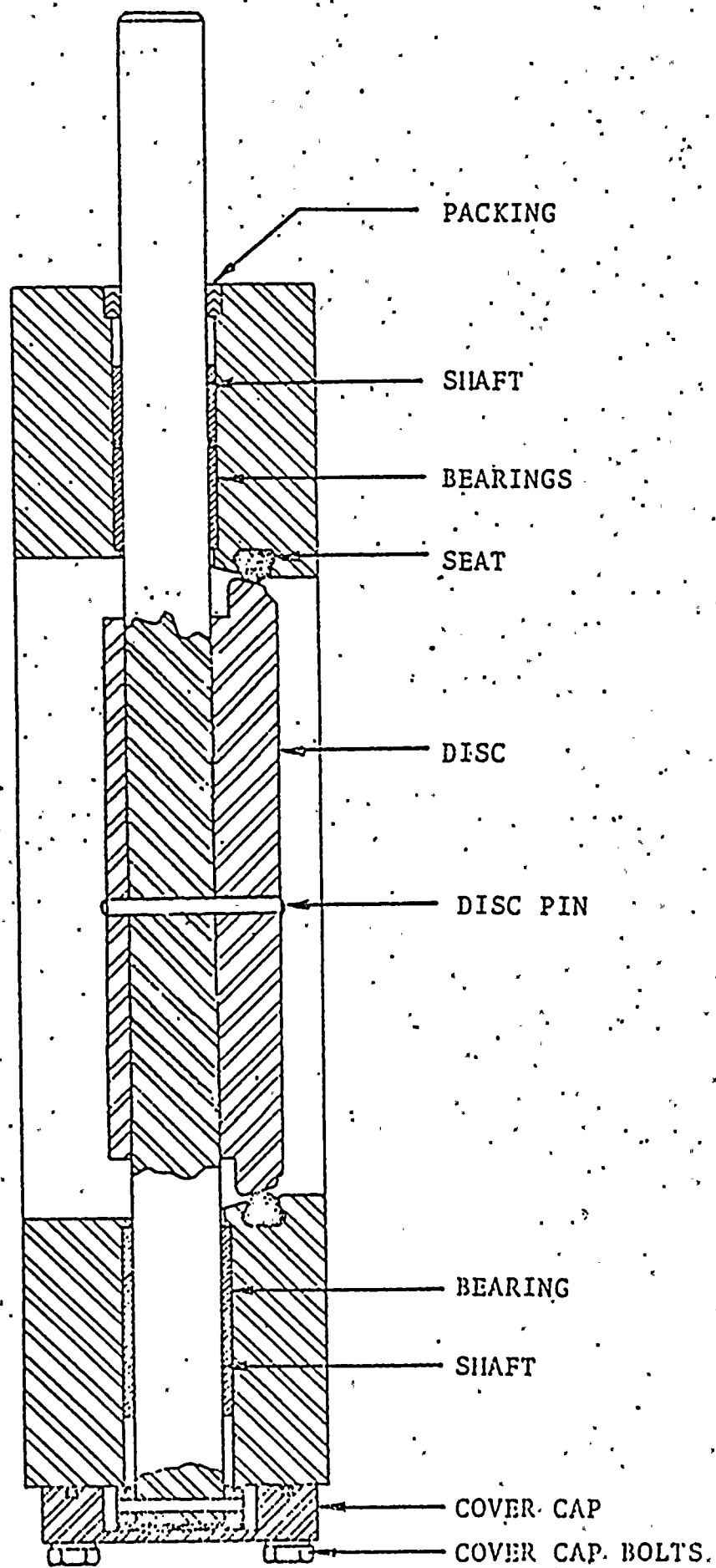


FIGURE 2 VALVE CROSS-SECTION

END CONNECTION ANALYSIS

The NRS butterfly valve is a uniflange design. Rather than having flanges that are external to and distinct from the body, the body shell is fabricated so that the end connections are machined directly into the body shell. The outside and inside diameter of the body shell conform to the requirements of the American National Standards Institute (ANSI) standard B16.5. The end connections, either flanged or weld end, also conform to this standard.

BODY ANALYSIS

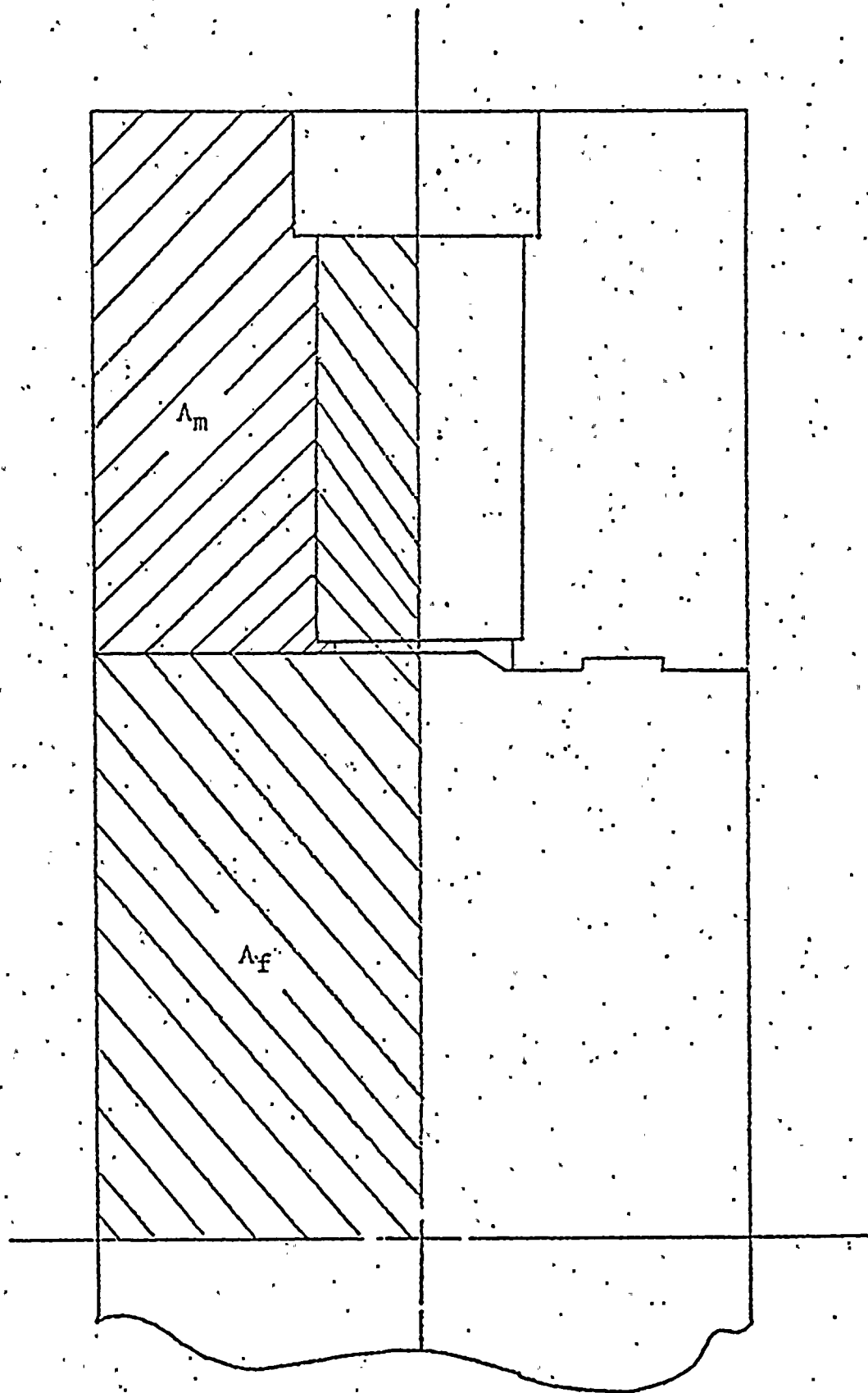
The body analysis consists of calculations as detailed in Paragraph NB-3540 of Section III of the Code. Paragraph NB-3540 is not highly oriented to butterfly valves as related to various design and shape rules. Therefore, certain of the design equations cannot be directly applied for butterfly valves. Where interpretation unique to the calculation is necessary, it is explained in the subsection containing that calculation description.

Figure 3 illustrates the essential features of the body geometry through the trunnion area of the valve. The symbols used to define specific dimensions are consistent with those used in the analysis and with the nomenclature used in the Code.

1. Minimum Body Wall Thickness

Paragraph NB-3542 gives minimum body wall thickness requirements for standard pressure rated valves.

The actual minimum wall thickness in the NRS valve occurs between the flange bolt holes and body bore.



PRESSURE-AREA ANALYSIS

BODY CROSS-SECTION

Figure 3

Body Analysis

2. Body Shape Rules

The NRS valve meets the requirements of Paragraph NB-3544 of the code for body shape rules. The external fillet at trunnion to body intersection must be greater than thirty percent of the minimum body wall thickness.

3. Primary Membrane Stress Due to Internal Pressure

Paragraph NB-3545.1 defines the maximum allowable stress in the neck to flow passage junction. In a butterfly valve, this corresponds with the trunnion to body shell junction. Figure 3 shows the geometry through this section.

The code defines the stresses in this area using the pressure area method. As seen in Figure 3, certain code-defined dimensions are not applicable to this style of butterfly valve. For example, there is no radius at the crotch when seen in a view along the flow pattern, as the neck extends to the face of the body. To comply with the intent of the code, the areas A_f and A_m are interpreted as shown in the cross-section (Figure 3). Using these areas, the primary membrane stress is then calculated.

$$P_m = (A_f/A_m + .5) p_s$$

Body Analysis

As an alternate method of determining the primary membrane stress, an equivalent analysis for primary membrane stress is applied to an area away from the trunnions. In these areas, the metal area and fluid area are as shown in Figure 4. Since the depth of the metal area is equal to the depth of the fluid area, the ratio A_f/A_m is equivalent to the mean radius of the body over the thickness of the body shell; R_m/H_g . The primary membrane stress through this section is then:

$$P_m' = (R_m/H_g + .5) p_s$$

4. Secondary Stresses

A. Body Primary plus secondary stress due to internal pressure: Paragraph NB-3545.2(a) of Section III of the code defines the formulas used in calculating this stress.

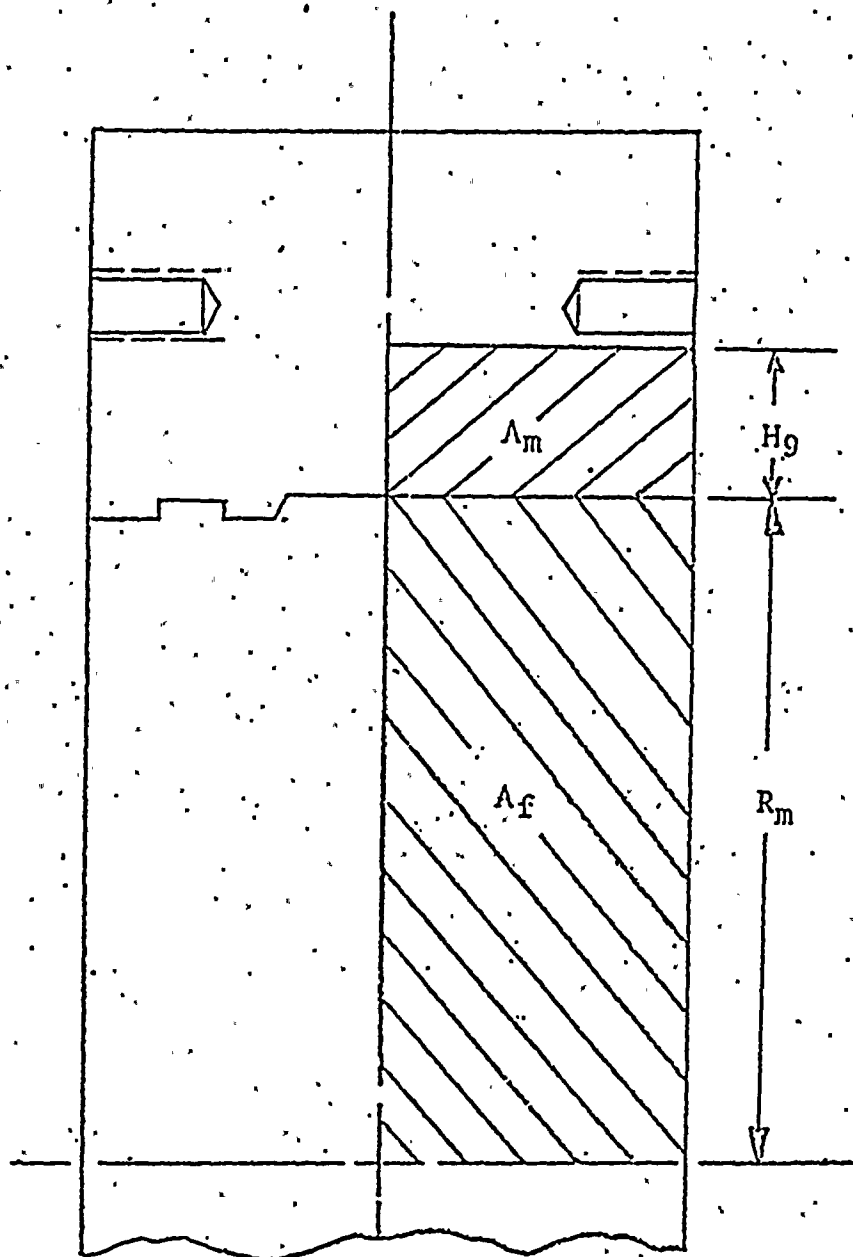
$$Q_p = C_p \left[\frac{r_i}{t_e} + .5 \right] p_s$$

B. Secondary stress due to pipe reaction: Paragraph NB-3545.2(b) gives the formulas for finding stress due to pipe reaction.

$$P_{cd} = \frac{F_d S}{G_d} \quad (\text{Direct or Axial Load Effect})$$

$$P_{eb} = \frac{C_b F_b S}{G_b} \quad (\text{Bending Load Effect})$$

$$P_{et} = \frac{2 F_b S}{G_t} \quad (\text{Torsional Load Effect})$$



PRESSURE AREA ANALYSIS

CROSS-SECTION IN BODY

Figure 4

Body Analysis

C. Thermal secondary stress: Paragraph NB-3545.2(c) of Section III of the code gives formulas for determining the thermal secondary stresses in the pipe.

$$Q_T = Q_{T1} + Q_{T2}$$

Where

$$Q_{T2} = C_6 C_2 \Delta T_2$$

D. Primary plus secondary stresses: This calculation is per Paragraph NB-3545.2 and is simply the sum of the three previous secondary stresses.

$$S_n = Q_p + P_e + 2Q_{t2} \leq 3S_m$$

5. Valve Fatigue Requirements

Paragraph NB-3543.3 of Section III of the code defines requirements for normal duty valve fatigue.

The allowable stress level is found from Figure I-9.0. Since the number of cycles is unknown, a maximum value of 2,000 is assumed. The allowable stress can then be found from Figure I-9.1 for carbon steel. This then gives an allowable stress of 65,000 psi.

$$S_{p1} = 2/3 Q_p + P_{eb}/2 + Q_{T3} + 1.3Q_{T1}$$

$$S_{p2} = .4 Q_p + P_{eb} + 2Q_{T3}$$

Where:

$$Q_{T3} = C_6 C_3 \Delta T_2$$

DISC ANALYSIS

Section NB-3546.2 defines the design requirements of the valve disc. Both primary bending and primary membrane stress are mentioned in this section. For a flat plate such as the butterfly valve disc, membrane stress is not defined until the deflection of the disc reaches one-half the disc thickness. Since total deflection of the disc is much less than one-half the thickness, membrane stresses are not applicable to the analysis.

Figure 5 shows the disc for the NRS butterfly valves. The disc is designed to provide a structurally sound pressure retaining component while providing the least interference to the flow.

Primary Bending Stress

Due to the manner in which the disc is supported, the disc experiences bending both along the shaft axis and about the shaft axis. The combined bending stress is maximized at the disc center where the maximum moment occurs. The moment is a result of a uniform pressure load.

Combined bending stress in disc:

$$S(1) = (S(2)^2 + S(3)^2)^{1/2}$$

Where:

$$S(2) = \frac{.90413 P_S R_d^3 C_7}{I_4} \quad = \text{Bending stress due to moment along shaft axis, psi}$$

$$S(3) = \frac{.6666 P_S R_d^3 C_8}{I_3} \quad = \text{Bending stress due to moment about shaft axis, psi}$$

SHAFT ANALYSIS

The shaft is analyzed in accordance with Paragraph NB-6.3 of Section III of the Code. The shaft loading is a combination of seismic, pressure and operating loads. Maximum torsional loading is either a combination of seating and bearing torque or bearing and dynamic torque. Columnar stress is not considered in the shaft loading due to its negligible effect on the stress levels. Figure 2 shows the banjo assembly with the rough shaft.

Shaft stresses due to pressure, seismic and operating loads:

$$S(5) = \frac{S(6)}{2} + \frac{(S(6)^2 + 4 S(7)^2)^{1/2}}{2}$$

where

$$S(6) = (S(8)^2 + S(9)^2)^{1/2} \quad = \text{Combined bending stress, psi}$$

$$S(8) = \frac{(\pi R_4^2 P_s + W_2 g_x) \cdot 2S B_1 R_5}{\pi \cdot .25 R_5^4} \quad = \text{Bending tensile stress due to pressure and seismic loads along x axis, psi}$$

$$S(9) = \frac{.25 W_2 g_y B_1 R_5}{.25 \pi R_5^4} \quad = \text{Bending tensile stress due to seismic loads along y axis, psi}$$

$$S(7) = (S(10)^2 + S(11)^2)^{1/2} \quad = \text{Combined shear stress, psi}$$

$$S(10) = \frac{T_8 R_5}{.5 \pi R_5^4} \quad = \text{Torsional shear stress, psi}$$

$$S(11) = 1.333 \left[\frac{.5 \pi R_4^2 P_s + .5 W_2 (g_x^2 + g_y^2)^{1/2}}{\pi R_5^2} \right] \quad = \text{Direct shear stress, psi}$$

Also worthy of attention is the torsional shear stress at the reduced cross-section where the pin passes through the shaft.

Shaft Analysis

$$S(12) = S(10) \left[\begin{array}{c} \frac{\pi R_5^4}{2} \\ \frac{\pi R_5^4}{2} - \frac{D_2 D_3^3}{12} - \frac{D_3 D_2^3}{12} \end{array} \right]$$

DISC PIN ANALYSIS

As seen in Figure 2, there is one through shaft and one disc pin. The pin is subject to seismic and torsional loads.

Combined shear stress in top disc pin:

$$S(13) = (S(14)^2 + S(15)^2)^{1/2}$$

Direct stress on disc pin due to seismic loads:

$$S(14) = \frac{W_7 g_z}{2N_1 (.785) D_3^2}$$

Torsional shear stress in disc pin:

$$S(15) = \frac{T_8 - .5U_5}{2N_1 R_5 .785 D_3^2}$$

Bearing stress on disc pin:

$$S(16) = \frac{T_8 - .5U_5}{2R_5 K_2 D_3 N_1}$$

Where:

$$U_4 = .785 (2R_4)^2 P_0 U_3 R_5$$

$$U_5 = U_4 + W_2 g_x U_3 R_5$$

$$P_0 = \text{Actual Shut-Off Pressure}$$

SHAFT BEARING ANALYSIS.

The sleeve bearings in the trunnion (Figure 2) are subjected to both seismic and pressure loads.

$$S(17) = \frac{P_d R_4^2 + W_2 (g_x^2 + g_y^2)^{\frac{1}{2}}}{2 \cdot L_5 D_2} \quad \text{= Compressive stress on shaft bearing, psi}$$

COVER CAP ANALYSIS

Figure 6 shows the bottom trunnion assembly, including the cover cap and cover cap bolts.

1. Cover cap bolt stresses:

The cover cap experiences loading from the weight of the banjo and from pressure loads. In determining stress levels, the bolts are assumed to share torsional and tensile loading equally.

Shear tear out of bolts through tapped holes in trunnion:

$$S(18) = \frac{W_2 \sqrt{g_x^2 + g_y^2 + g_z^2} + \pi P_s R_6^2}{4 L_3 2.83 D_6}$$

Shear tear out of bolt heads through cover cap, psi:

$$S(19) = \frac{W_2 \sqrt{g_x^2 + g_y^2 + g_z^2} + \pi P_s R_6^2}{4 T_1 5.2 D_6}$$

Combined stress in bolts, psi:

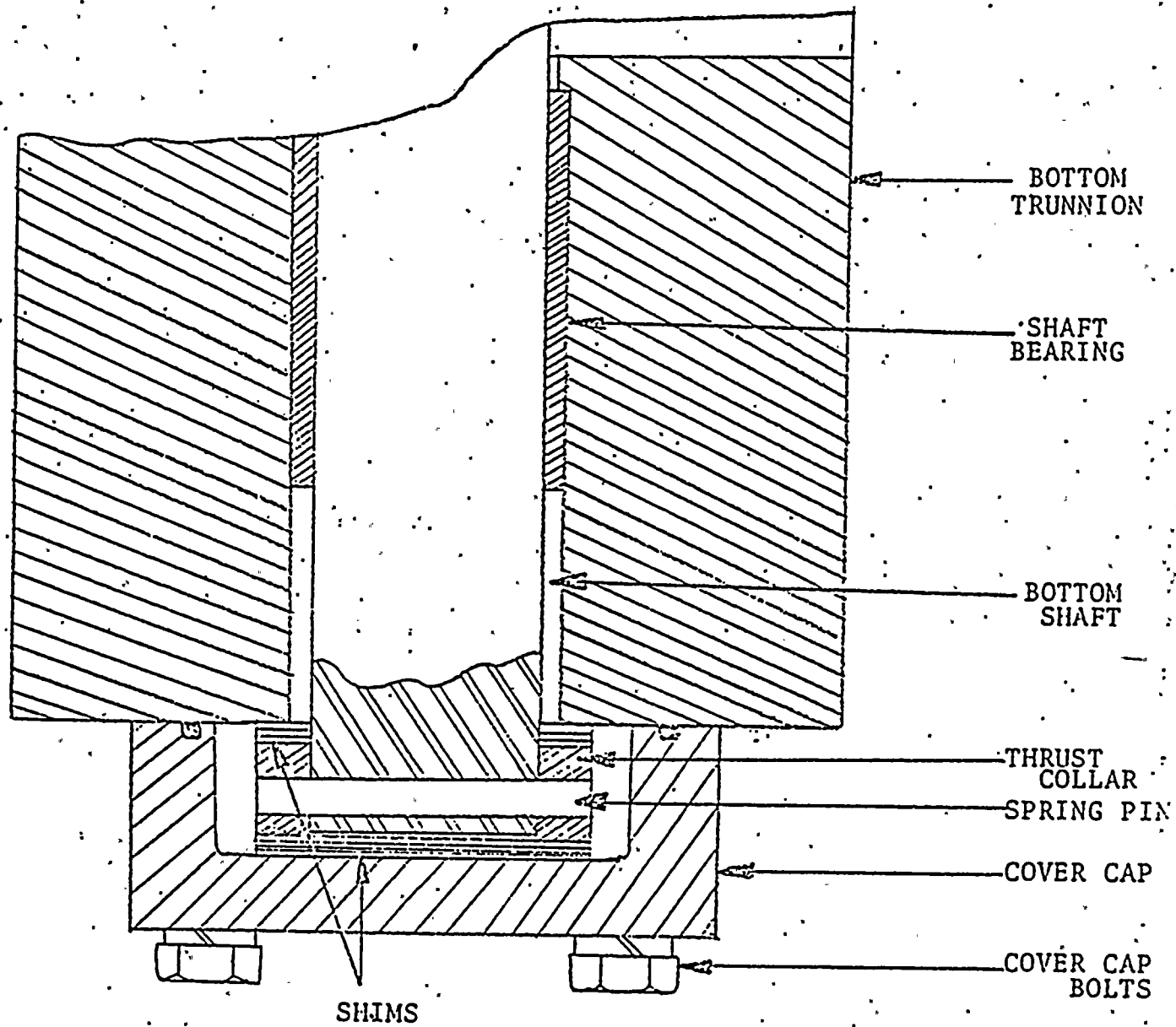
$$S(20) = \frac{S(22)}{2} + \frac{(S(22)^2 + 4S(21)^2)^{1/2}}{2}$$

Where:

$$S(21) = \frac{.25 W_2 \sqrt{g_x^2 + g_y^2 + g_z^2} (D_2 + .66 (D_4 - D_2))}{.707 H_3 4 A_4}$$

Shear Stress in Bolts Due to Torsional Load.





BOTTOM TRUNNION AND THRUST BEARING ASSEMBLY

Figure 6



Cover Cap Analysis

$$S(22) = \frac{W_2 \sqrt{g_x^2 + g_y^2 + g_z^2} + \pi P_s R_6^2}{4 A_3} = \text{Tensile Stress in Bolts Due to Seismic And Pressure Loads, psi}$$

2. Cover cap stresses:

The combined stress in the covercap is calculated using the following formulas:

$$S(23) = \frac{S(24) + S(25)}{2} + \frac{((S(24) + S(25))^2 + 4S(26)^2)^{1/2}}{2}$$

Where:

$$S(24) = \frac{3(.785 (D_4 + .25)^2 P_s + W_2 g_z)}{4 \pi T_4^2} = \text{Radial Stress}$$

$$S(25) = \frac{3(.785 (D_4 + .25)^2 P_s + W_2 g_z)}{4 \pi T_4^2 m} = \text{Tangential Stress}$$

$$S(26) = \frac{.785 (D_4 + .25)^2 P_s + W_2 g_z}{\pi (D_4 + .25) T_4} = \text{Shear Stress}$$

THRUST BEARING ANALYSIS

As seen in figure 6, the thrust bearing assembly is located in the bottom trunnion. It provides restraint for the banjo in the z direction, assuring that the disc edge remains correctly positioned to maintain optimum sealing. Formulas used to analyze the assembly are given below:

1. Bearing stress on thrust collar due to seismic and pressure loads:

$$S(27) = \frac{W_2 \sqrt{g_x^2 + g_y^2 + g_z^2} + \pi P_s R_5^2}{.785 (D_4^2 - (D_2 + .25)^2)}$$

2. Shear load on thrust collar spring pin due to seismic, pressure and torsional loads:

$$S(28) = \left[(W_2 g_z + \pi P_s R_5^2)^2 + \left(\frac{.25 W_2 g_z (D_2 + .0833 + .66 (D_4 - D_2))}{R_5} \right)^2 \right]^{1/2}$$

3. Bearing stress of spring pin on thrust collar:

$$S(29) = \frac{((W_2 g_z + \pi P_s R_5^2)^2 + (.25 W_2 g_z)^2)^{1/2}}{D_5 (D_4 - D_2)}$$

4. Shear tear out of spring pin through bottom of shaft:

$$S(31) = \frac{W_2 g_z + \pi P_s R_5^2}{.2 D_2 T_2}$$

OPERATOR MOUNTING ANALYSIS

The operator mounting consists of the top trunnion, the bonnet, the operator housing, and the bolt connections. The elements of the assembly are shown in Figure 7.

1. Bolt stresses and localized stress due to bolt loads. The following assumptions are used in the development of the equations:

A. Torsional, direct shear, and direct tensile loads are shared equally by all bolts in the pattern.

B. Moments across the bolt pattern are opposed in such a way that the load in each bolt is proportional to its distance from the neutral bending axis.

(a) Shear tear out of trunnion bolt through tapped hole in top trunnion.

$$S(32) = \frac{F_z + W_4 \sqrt{g_x^2 + g_y^2 + g_z^2}}{4} + \frac{\overline{M}_x (J_2 + H_2)}{2J_2^2 + 2(J_2 + H_2)^2} + \frac{\overline{M}_y (J_1 + H_2)}{2J_1^2 + 2(J_1 + H_2)^2}$$

$$\frac{\quad}{.9\pi L_4 D_7}$$

(b) Bearing stress on tapped holes in trunnion.

$$S(33) = \frac{M_z + T_8}{4(.707 H_2)} + \frac{(F_x^2 + F_y^2)^{\frac{1}{2}}}{4} + \frac{W_4 (g_x^2 + g_y^2)^{\frac{1}{2}}}{4}$$

$$\frac{\quad}{D_7 L_4}$$

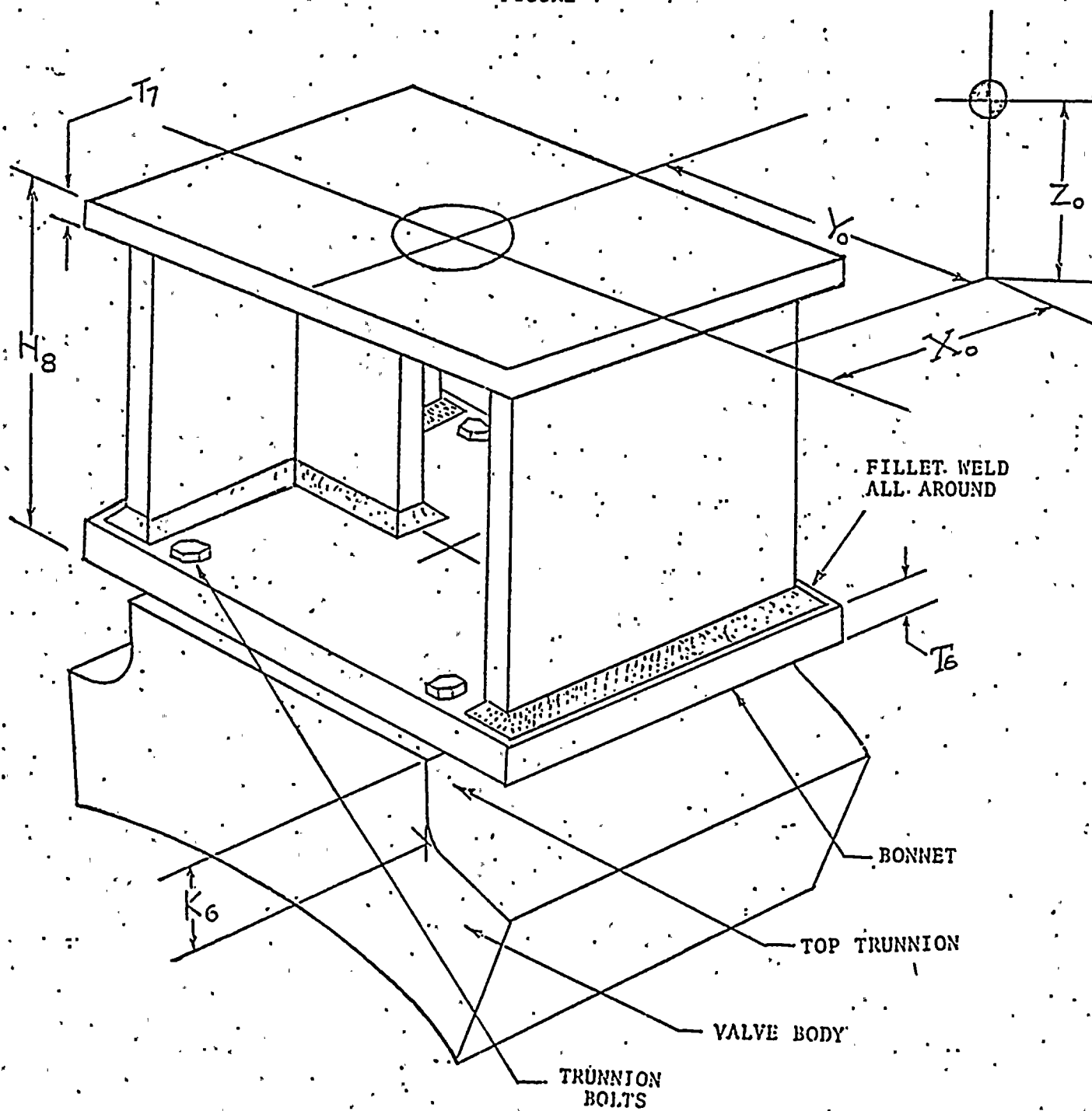
(c) Bearing stress on through hole in bonnet.

$$S(34) = \frac{M_z + T_8}{4(.707 H_2)} + \frac{(F_x^2 + F_y^2)^{\frac{1}{2}}}{4} + \frac{W_4 (g_x^2 + g_y^2)^{\frac{1}{2}}}{4}$$

$$\frac{\quad}{D_7 T_6}$$

TOP TRUNNION MOUNTING

FIGURE 7



Operator Mounting Analysis

- d. Shear tear out of trunnion bolt heads through bonnet.

$$S(35) = \frac{F_z + W_4 g_z}{4} + \frac{\overline{M}_x (J_2 + H_2)}{2J_2^2 + 2(J_2 + H_2)^2} + \frac{\overline{M}_y (J_1 + H_2)}{2J_1^2 + 2(J_1 + H_2)^2}$$

5.2 D7T6

- e. Combined stress in trunnion bolts (See Fig: 8)

$$S(36) = \frac{S(37) + S(38)}{2} + \frac{((S(37) + S(38))^2 + 4(S(39) + S(40))^2)^{\frac{1}{2}}}{2}$$

Where

$$S(37) = \frac{F_z + W_4 g_z}{4 A_5} = \text{Direct Tensile Stress, psi}$$

$$S(38) = \frac{\overline{M}_x (J_2 + H_2)}{2J_2^2 + 2(J_2 + H_2)^2} + \frac{\overline{M}_y (J_1 + H_2)}{2J_1^2 + 2(J_1 + H_2)^2} = \text{Tensile stress due to extended mass bending moment, psi}$$

A₅

$$S(39) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}} + W_4 (g_x^2 + g_y^2)^{\frac{1}{2}}}{4 A_6} = \text{Direct shear stress, psi}$$

$$S(40) = \frac{(M_z + T_8)}{(.707 H_2) 4 A_6} = \text{Shear stress due to torsional load, psi}$$

- f. Shear tear out of operator bolt head through hole in bonnet.

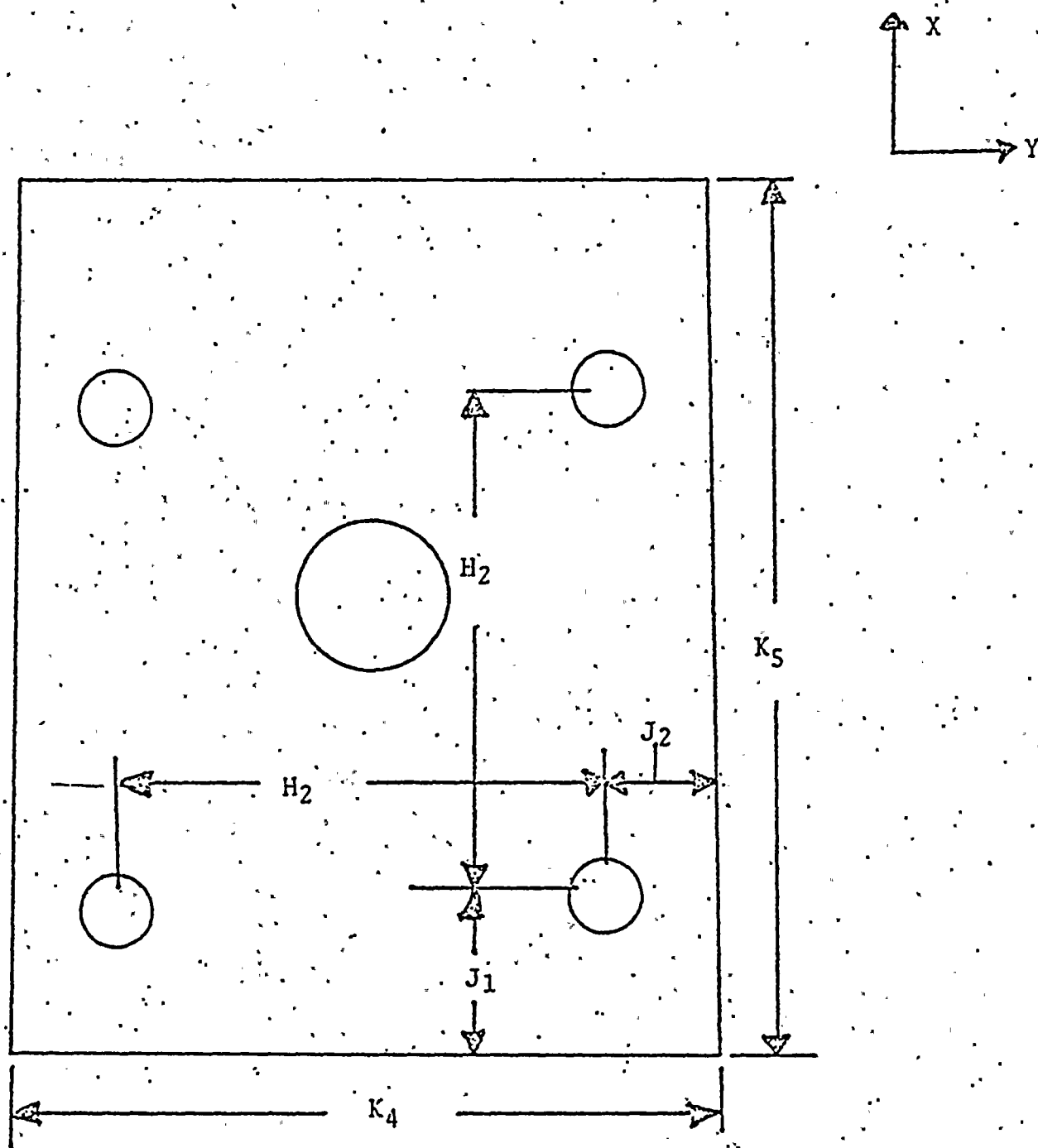
$$S(41) = \frac{F_z}{N_2} + \frac{M_x (J_4 + H_4)}{2J_4^2 + 2(J_4 + H_4)^2} + \frac{M_y (J_3 + H_4)}{2J_3^2 + 2(J_3 + H_4)^2}$$

5.2 D8T7

- g. Bearing stress on tapped holes in bonnet.

$$S(42) = \frac{M_z + T_8}{(.707 H_4) N_2} + \frac{(F_x^2 + F_y^2)^{\frac{1}{2}}}{N_2}$$

D8T7



TOP TRUNNION BOLTING

Figure 8

Operator Mounting Analysis

h. Combined stress in operator bolts (See Fig. 9)

$$S(43) = \frac{S(44)+S(45)}{2} + \frac{((S(44)+S(45))^2 + 4(S(46)+S(47))^2)^{\frac{1}{2}}}{2}$$

Where.

$$S(44) = \frac{F_z}{N_2 A_7} = \text{Direct tensile stress, psi}$$

$$S(45) = \frac{M_x(J_4+H_4)}{2J_4^2+2(J_4+H_4)^2} + \frac{M_y(J_3+H_4)}{2J_3^2+2(J_3+H_4)^2} = \text{Tensile stress due to bending, psi}$$

A_7

$$S(46) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}}}{N_2 A_8} = \text{Direct shear stress}$$

$$S(47) = \frac{M_z + T_8}{(.707H_4)N_2 A_8} = \text{Shear stress due to torsion, psi}$$

2. Bonnet Stresses

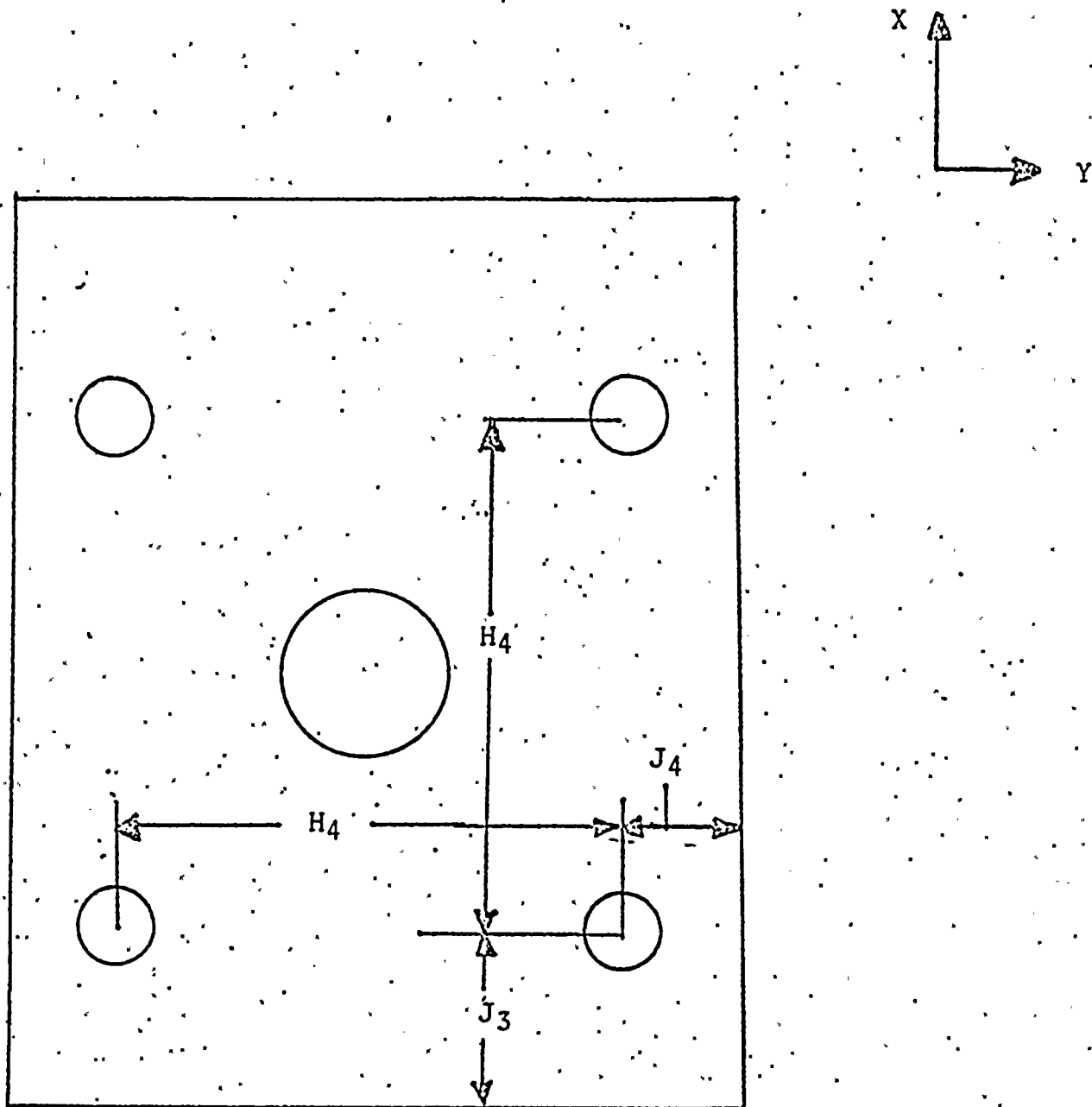
The bonnet stresses are calculated with the assumption that loading is through the bolt connections as previously defined.

a. The maximum combined stress in the bonnet was calculated using the following formulas:

$$S(48) = \frac{S(49)+S(50)}{2} + \frac{((S(49)+S(50))^2 + 4(S(51)+S(52))^2)^{\frac{1}{2}}}{2}$$

= Combined stress in bonnet legs

$$S(49) = \frac{F_z + W_4 g_z}{B_5} = \text{Direct tensile stress, psi}$$



BONNET BOLT PATTERN

Figure 9

Operator Mounting Analysis

$$S(50) = \frac{\overline{M}_x B_8}{I_1} + \frac{\overline{M}_y B_9}{I_2} = \text{Tensile stress due to bending moment, psi}$$

Where

$$S(51) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}} + W_4(g_x^2 + g_y^2)^{\frac{1}{2}}}{B_5} = \text{Direct shear stress, psi}$$

$$S(52) = \frac{T}{K_0} C_0 = \text{Shear stress in bonnet body due to torsional load, psi}$$

Where

T = Torque, in-lbs.

C₀ = Torsional constant for non-circular cross section

K₀ = Function of cross-section, in.⁴

- b. The maximum combined shear stress in the bonnet mounting plate to body welds was calculated using the following formulas:

Bottom Bonnet Weld

$$S(53) = \frac{(S(54)^2 + 4S(55)^2)^{\frac{1}{2}}}{2} = \text{Combined shear stress in bottom weld, psi}$$

Where

$$S(54) = S(56) + S(57) = \text{Total tensile stress, psi}$$

$$S(56) = \frac{F_z + W_4 g_z}{U_1} = \text{Direct tensile stress, psi.}$$

$$S(57) = \frac{\overline{M}_x}{Z_1} + \frac{\overline{M}_y}{Z_2} = \text{Bending tensile stress}$$

$$S(55) = S(58) + S(59) = \text{Total shear stress}$$

$$S(58) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}} + W_4(g_x^2 + g_y^2)^{\frac{1}{2}}}{U_1} = \text{Direct shear stress, psi}$$

Operator Mounting Analysis

$$S(59) = \frac{M_z + T_8}{Z_3} = \text{Torsional shear stress, psi}$$

Top Bonnet Weld

$$S(60) = \frac{(S(61)^2 + 4S(62)^2)^{\frac{1}{2}}}{2} = \text{Combined shear stress in top bonnet weld}$$

Where

$$S(61) = S(63) + S(64) = \text{Total tensile stress, psi}$$

$$S(63) = \frac{F_z}{U_2} = \text{Direct tensile stress, psi}$$

$$S(64) = \frac{M_x}{Z_1} + \frac{M_y}{Z_2} = \text{Bending tensile stress, psi}$$

$$S(62) = S(65) + S(66) = \text{Total shear stress, psi}$$

$$S(65) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}}}{U_2} = \text{Direct shear stress, psi}$$

$$S(66) = \frac{M_z + T_8}{Z_4} = \text{Torsional shear stress, psi}$$

c. Trunnion Body Stress

The trunnion body stresses are calculated using the following assumptions:

1. Operator loading is through the bolt connections.
2. There is an equal and opposite reaction to the bolt loads at the body.

Operator Mounting Analysis

The combined stress in the trunnion body was calculated using the following formulas:

$$S(67) = \frac{S(68)+S(69)}{2} + \frac{((S(68)+S(69))^2 + 4(S(70)+S(71))^2)^{\frac{1}{2}}}{2}$$

Where

$$S(68) = \frac{F_z + W_4 g_z}{K_4 K_5 - .785 B_2^2} = \text{Direct tensile stress, psi}$$

$$S(69) = \frac{(M_x + F_y K_6) \cdot 5 K_4}{.0833 K_5 K_4^3 - \pi B_2^4} + \frac{(M_y + F_x K_6) \cdot 5 K_5}{.0833 K_4 K_5^3 - \pi B_2^4} = \text{Bending tensile stress, psi}$$

$$S(70) = \frac{(F_x^2 + F_y^2)^{\frac{1}{2}} + W_4 (g_x^2 + g_y^2)^{\frac{1}{2}}}{K_4 K_5 - .785 B_2^2} = \text{Direct shear stress, psi}$$

$$S(71) = \frac{(M_z + T_8) \cdot 5 (K_4^2 + K_5^2)^{\frac{1}{2}}}{.0833 (K_4 K_5^3 + K_5 K_4^3) - \pi B_2^4} = \text{Torsional shear stress, psi}$$

FREQUENCY ANALYSIS

A. Introduction

To calculate the natural frequency of the various components of the NRS valve, a model system with a single degree of freedom is constructed. The individual components and groups of components are modeled and analyzed as restoring spring forces which act to oppose the respective weight forces they are subjected to. The static deflection of the component is calculated and is related to natural frequency as:

$$F_n = \frac{1}{2\pi} \sqrt{\frac{K}{M}}$$

or

$$F_n = \frac{1}{2\pi} \sqrt{\frac{g}{\Delta y}}$$

or

$$F_n = \left(\frac{9.8}{\Delta y} \right)^{1/2}$$

The analysis details the equations and assumptions used in determining the natural frequencies listed in the summary table. Sketches are provided where appropriate.

B. Valve Body Assembly

The body shell, as seen in Figure 1, is assumed to experience loading due to the entire valve weight.

Natural Frequency of the body shell:

$$F_{N1} = \left(\frac{9.8}{\Delta y_1} \right)^{1/2}$$

Frequency Analysis

Where

$$\Delta y_1 = \frac{W_1 L_1^3}{48 E I_5}$$

= Maximum deflection of body shell due to valve weight, in.

C. Banjo Assembly

Figure 2 shows the banjo assembly in the body. The natural frequency of the banjo assembly is calculated using the following:

$$F_{N2} = \left(\frac{9.8}{\Delta y_2} \right)^{\frac{1}{2}}$$

Where

$$\Delta y_2 = \frac{W_7 B_1^3}{12 E I_6}$$

= Maximum deflection of shaft, inches

D. Cover Cap Assembly

As seen in Figure 6, the cover cap supports the banjo. The natural frequency of the cover cap is calculated as follows:

$$F_{N3} = \left(\frac{9.8}{\Delta y_3} \right)^{\frac{1}{2}}$$

Where

$$\Delta y_3 = \frac{3(m^2 - 1) W_2 (.5D_4 + .125)^2}{16\pi E T_4^3 m^2}$$

= Maximum deflection of cover cap

E. Bonnet Assembly

Figure 7 shows the top trunnion assembly. The following assumptions are made in calculating the bonnet natural frequency:

Frequency Analysis

1. The worst valve assembly mounting position is where the bending moment is predominant in producing deflection.
2. The bonnet is assumed fixed at the top trunnion.
3. The adapter plate is assumed to be integral with and have a cross-section the same as the component it mounts to.

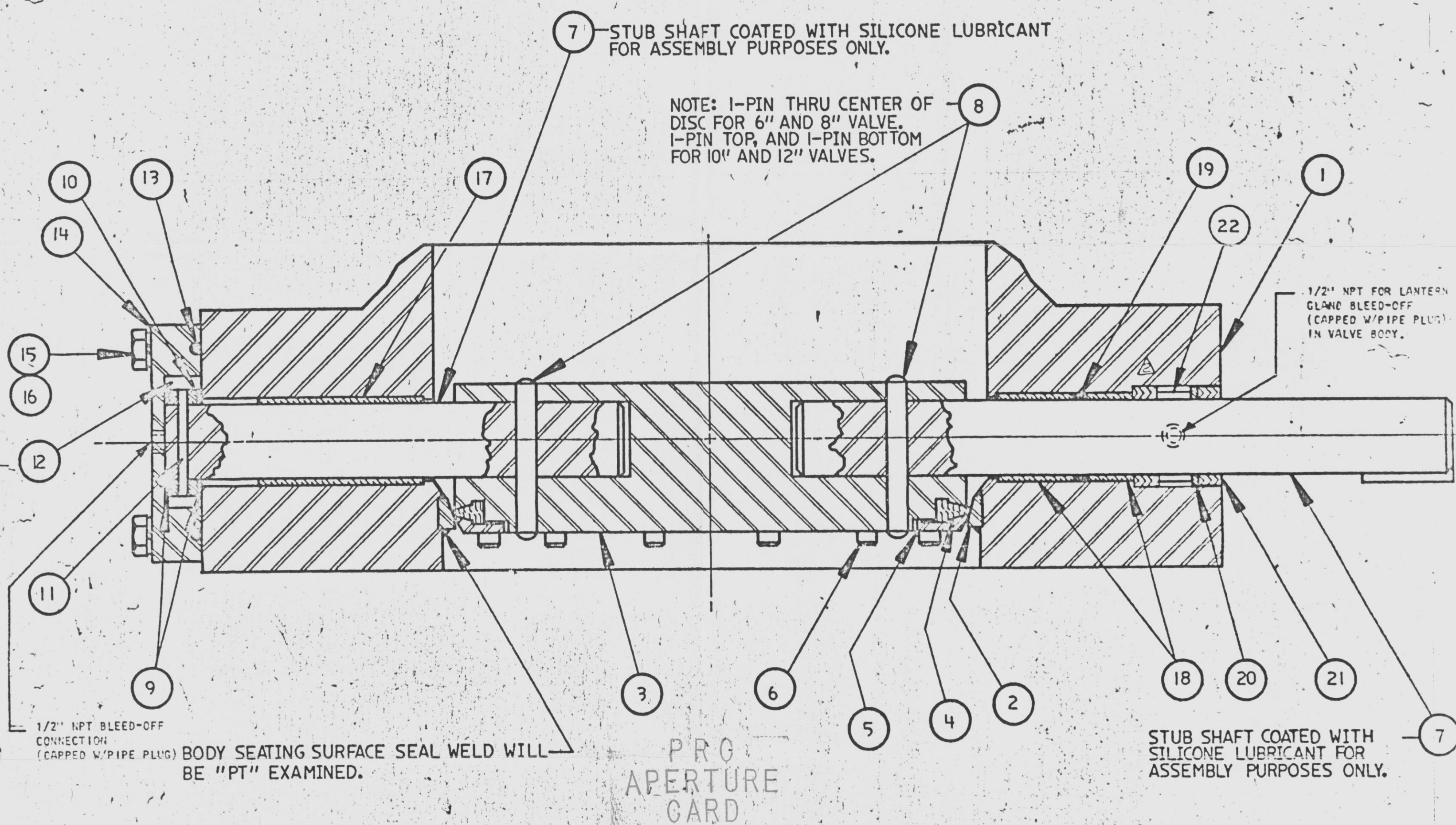
Natural frequency of bonnet:

$$F_{n4} = \left(\frac{9.8}{\Delta y_4} \right)^{\frac{1}{2}}$$

Where

$$\Delta y_4 = \frac{W_3 H_8^3 + W_4 K_3^3}{3EI_1} + \frac{W_3 Z_0 H_8^2}{2EI_1}$$

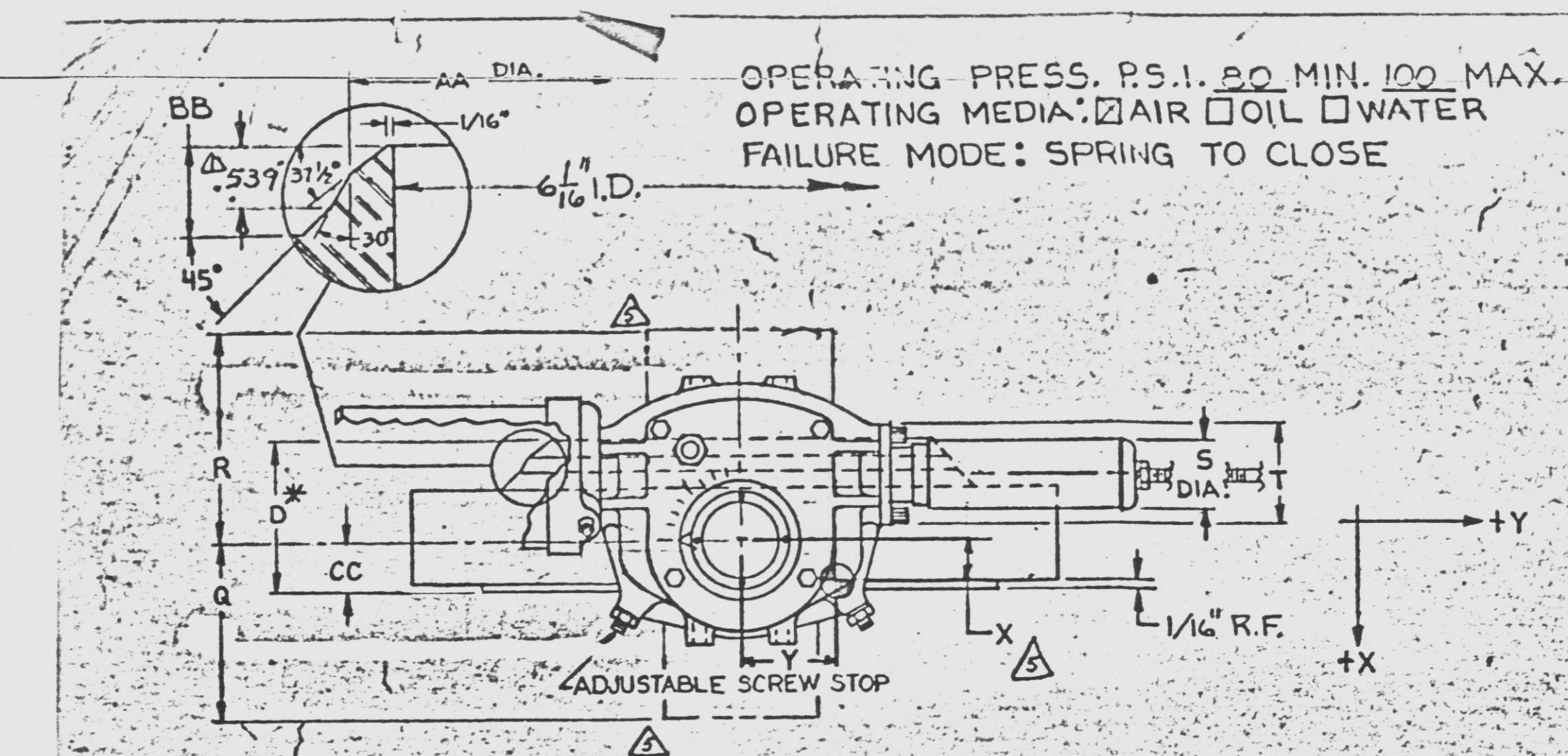
| PARTS AND MATERIALS OF CONSTRUCTION | | PARTS AND MATERIALS OF CONSTRUCTION | |
|-------------------------------------|-----------------------------|-------------------------------------|------------------------------|
| 1. BODY: MAT'L.; | SA-516 GR. 55 | 12. GREASE: DOW CORNING III | |
| 2. SEATING SURFACE: MAT'L.; | SA-479 TYPE 304 | 13. O-RING: MAT'L.; E.P.T. | |
| 3. DISC: MAT'L.; | SA-516 GR. 55 | 14. BOTTOM COVER: MAT'L.; | SA-516 GR. 70 |
| 4. SEAT: MAT'L.; E.P.T. | | 15. COVER BOLTS: MAT'L.; | SA-193 GR. B-7 |
| 5. CLAMP SEGMENT RING: MAT'L.; | SA-285 GR. C | 16. LOCKWASHER: MAT'L.; | CARBON STEEL |
| 6. CLAMP SEGMENT SCREWS: MAT'L.; | SA-193 GR. B-7 | 17. BOTTOM BEARING: MAT'L.; | ASTM B-438 GR. 1 TYPE 2 BRZ. |
| 7. SHAFT: MAT'L.; | SA-564 TYPE 630 COND. H1150 | 18. TOP BEARINGS: MAT'L.; | ASTM B-438 GR. 1 TYPE 2 BRZ. |
| 8. PINS: MAT'L.; | SA-320 GR. B8M | 19. SHAFT SEAL: MAT'L.; | E.P.T. |
| 9. THRUST COLLAR SHIMS: MAT'L.; | HARD BRASS | 20. PACKING RETAINER RING: MAT'L.; | SB-144 ALLOY 3B |
| 10. THRUST COLLAR: MAT'L.; | SAE 660 BRONZE | 21. PACKING: MAT'L.; | E.P.T. V-RINGS |
| 11. THRUST COLLAR PIN: MAT'L.; | AISI 420 STN. STL. | 22. LANTERN GLAND RING: MAT'L.; | ASTM A-269 |



* MATERIAL AND NDE STANDARDS SHALL BE IN ACCORDANCE WITH ASME SECTION III CLASS 2 REQUIREMENTS.

| | | | | | |
|---|----|--|-------|--|-------|
| SCALE: NONE DATE: 6-18-75 DRAWN BY: [Signature] CHECKED BY: CYK APPROVED: [Signature] DWG. NO.: C-2988 | | 6" THRU 12" ASME SECTION III CLASS 2 VALVE CROSS SECT. AND MATERIALS LIST FLANGE X WELD END | | CUSTOMER: BECHTEL POWER CORP. CUSTOMER P. O. P8856-P-31-AC PRATT ORDER NO.: D-0026-3 PROJECT: PENNSYLVANIA POWER & LIGHT CO., SUSQUEHANNA ITEM NO. 1.17 UNIT-1 6"-HBB-BF-A0-5721 1.18 UNIT-2 6"-HBB-BF-A0-5721 | |
| REV. DATE | BY | CHK'D | APP'D | APP'D | APP'D |
| 1. 6-18-75 | RA | RA | RA | RA | RA |
| HENRY PRATT COMPANY AUBURN, ILL. | | | | | |

8206160247-01



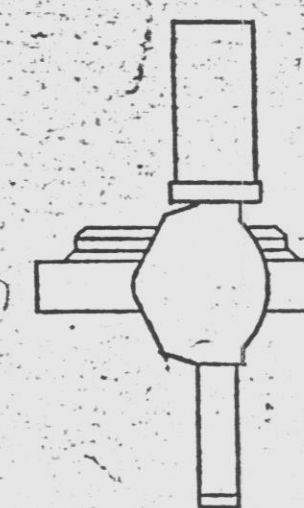
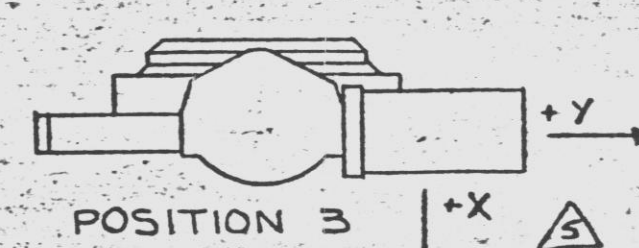
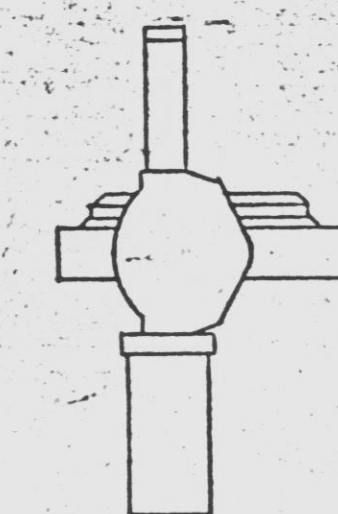
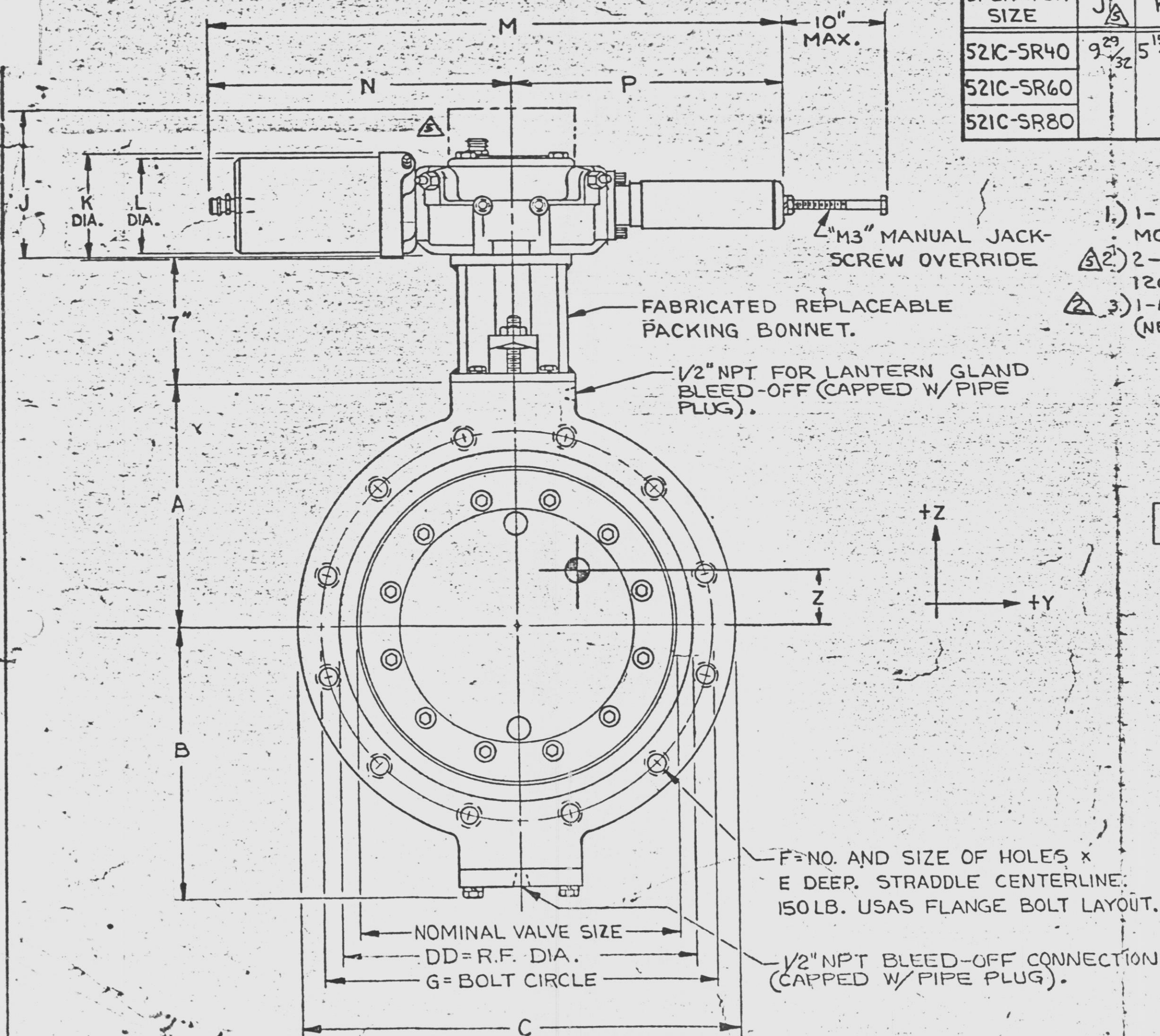
| VALVE SIZE | A | B | C | D* | E | F | G | AA | BB | CC | DD |
|------------|-------|---|----|----|--------|-------|-------|---------|-------|-------|-------|
| 6 | 6 1/2 | 8 | 11 | 6 | 1 3/16 | 8 3/4 | 9 1/2 | 6 29/32 | 1 1/2 | 2 1/2 | 8 1/2 |
| 8 | | | | | | | | | | | |
| 10 | | | | | | | | | | | |
| 12 | | | | | | | | | | | |
| 14 | | | | | | | | | | | |
| 16 | | | | | | | | | | | |
| 18 | | | | | | | | | | | |
| 20 | | | | | | | | | | | |
| 24 | | | | | | | | | | | |

NOTE: ALL DIMENSIONS ARE SHOWN IN INCHES.
*D ± 1/16" THRU 10" VALVES.
*D ± 1/8" FOR 12" VALVES AND LARGER.
OVERALL DIMENSIONS ARE MAXIMUM
OTHERS ARE NOMINAL

| OPERATOR SIZE | J | K | L | M | N | P | Q | R | S | T |
|---------------|---------|--------|-------|--------|--------|---------|--------|---------|-------|-------|
| 521C-SR40 | 9 29/32 | 5 5/16 | 5 3/8 | 43 1/2 | 28 3/4 | 14 3/16 | 10 7/8 | 10 1/32 | 2 1/4 | 3 3/4 |
| 521C-SR60 | | | | | | | | | | |
| 521C-SR80 | | | | | | | | | | |

ACCESSORIES (MOUNTED ON OPERATOR)

- 1-CIRCLE SEAL SINGLE SOLENOID VALVE
MODEL S.V. 315 9101-1 120V.A.C. 60 HZ.
- 2-NAMCO EA-740-20000 LIMIT SWITCHES
120 V.A.C. 10 AMP D.P.D.T.
- 1-HOFFMAN JUNCTION BOX # A-606 CHNF
(NEMA IV).



PRC
APERTURE
CARD

STOWER: BECHTEL POWER CORP.
CUSTOMER P.O. P8856-P-31-AC
PART ORDER NO. D-0026-3
PROJECT: PENNSYLVANIA POWER & LIGHT CO., SUSQUEHANNA
ITEM QUAN. SIZE OPER. OPERATOR VALVE CROSS VALVE OPER. TAG
1 2 6" A3 521C-SR60-M3 C-2988A 165# 112# * A: 0.1 3.58 7.85
NOTE: FOR BONNET CROSS-SECT. & PARTS + MAT'L. LIST SEE DWGS A-8721 & A-8722.
TAG: UNIT 1-GHBB-BF-AO-5721 1.17
UNIT 2-GHBB-BF-AO-5721 1.18
PAINT: EXTERIOR OF VALVE & OPERATOR
1 COAT OF CARBID 4-E-2 PRIMER GOOD FOR 600°F.
NOTE: FOR BONNET CROSS-SECT. & PARTS & MAT'L. LIST SEE DWGS. A-8721 & A-8722.

| FLANGE x WELD END | | | |
|--|---------|---------|-----------|
| A 17-75 | B 17-75 | C 17-75 | D 17-75 |
| A 18-75 | B 18-75 | C 18-75 | D 18-75 |
| A 19-75 | B 19-75 | C 19-75 | D 19-75 |
| REV. DATE | BY | APP | REV. DATE |
| 11-21-71 | | | 11-21-71 |
| HENRY PRATT COMPANY AURORA, ILL. | | | |
| GENERAL ARRANGEMENT NUCLEAR N.R.S. VALVE W/BON. & BETTIS SPRING-RETURN OPER. | | | |
| SCALE: NONE DATE: 5-23-71 | | | |
| DRAWN BY: G. DAVIS CHECKED BY: C. H. DAVIS | | | |
| APPROVED: 1 | | | |
| PART NO. D-0026-3 | | | |
| DWG. NO. C-2600 | | | |

8206160247-02

D