

Q 210.51

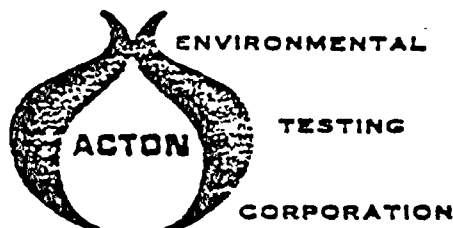
Test Report No. 14801-1

No. of Pages 38

Rev 1

Report of Test on

SEISMIC VIBRATION TESTING
OF LIMITORQUE SMB-1/H3BC ACTUATORS
FOR LIMITORQUE CORPORATION
UNDER PURCHASE ORDER NO. TL516



Date June 11, 1979

	Prepared	Checked	Approved
By	B. Esposito	R. Gilfoy	M. L. Tolf
Signed	<i>Bruce Esposito</i>	<i>R. Gilfoy</i>	<i>M. L. Tolf</i>
Date	<i>June 13, 1979</i>	<i>13 June 79</i>	<i>6/13/79</i>

BE/hmf

31251

SEISMIC TEST SMB-1-60/H3BC

TEST REPORT #80048

SUBJECT: Project 678037 - Pre Seismic Torque Switch Calibration and Monitoring performed during Seismic Testing.

REFERENCE: Limitorque P/O 678037
Acton Report #14801-1

Unit Identification and Description

SMB-1-60, I.D. #678037-C mounted on an H3BC; I.D. #678037-C1.

Pre-Test Torque Switch Calibration

Procedure: Measured torque output of SMB-1-60/H3BC using 10,000' # torque stand.

Result: An actual torque of 5,740' # was obtained at a torque switch setting of 2 3/4.

Seismic Test Monitoring

Procedure: Both limit switch rotors and "open" torque switch were monitored for chatter by a 1 millisecond electronic chatter circuit.

The actuator was run from an open position (controlled by a limit switch), to a closed "torqued out" position (controlled by the torque switch) back to the open limit position during each of the seismic dwells and at the end of the seismic qualification.

Result and Conclusion

Refer to the general discussion of seismic testing for test reports 80047 and 80048 inserted in front of these reports.

Paul T. Young
Paul T. Young
Assistant Chief Test Engineer

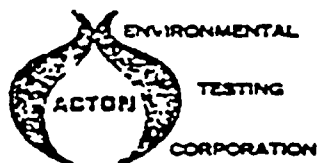
1/20/79

wcd

REVISION RECORD

DATE	REVISION NUMBER	PAGE NUMBER	PARAGRAPH NUMBER	CHANGES OR ADDITIONS	APPROVED BY
6/11/79	0	-----FIRST ISSUE-----			
7/25/79	1	1	4.0	Changed 1/17/79 to 4/9/79	Bruce Caputo July 25, 1979
		4	3.0	"	
		6	3.2	Reworded sentence	
		9	3.3	Added word "Independently"	
		15	3.7	Changed seismic aging test to "Plant Vibration Aging Test" in 4 places	
		17	4.2	Added 6.0g Reworded last sentence	
		19	4.4	Changed seismic Aging Test to "Plant Vibration Aging Test" in 2 places	
		22	Figure 1	Removed OBE curve	

Report No. 14801-1



2/14/51

Administrative Data

1.0 Purpose of Test: To subject actuators to seismic vibration testing.

2.0 Manufacturer: Limitorque Corporation
Lynchburg, Virginia

3.0 Manufacturer's Type or Model No: SMB-1/H3BC

4.0 Drawing, Specification or Exhibit: Limitorque Seismic Test of
Limitorque SMB-3/H52C and
SMB-1/H3BC R&D Project 678037
dated April 9, 1979

5.0 Quantity of Items Tested: Two (2) units marked #3 and #4

6.0 Security Classification of Items: Unclassified

7.0 Date Test Completed: May 10, 1979

8.0 Test Conducted By: C. Pilotte
D. McLaughlin

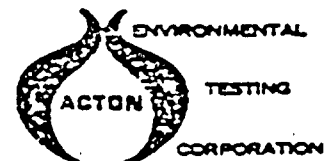
9.0 Disposition of Specimens: Returned to Limitorque Corporation.

10.0 Abstract:

Refer to the results section herein.

Report No. 14801 -1
Rev 1

Page 1



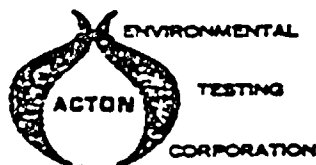


3/10/81

1.0 TEST ITEMS

Two (2) Limatorque SMB-1/H3BC Actuators were submitted by Limatorque Corporation for seismic vibration testing at Acton Environmental Testing Corporation (AETC). The items were marked #3 and #4.

Report No. 14801 -1

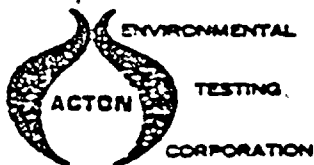


Page 2

2.0 TEST REQUIREMENTS

The purpose of this test was to subject the actuators specified in section 1.0 above to the seismic vibration tests specified in section 3.0 below to determine their ability to withstand such vibration without evidence of mechanical damage, deterioration, loss of their ability to operate properly, or contact chatter in excess of one (1) millisecond where applicable.

Report No. 14801-1



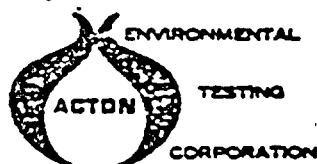
Page 3



3.0 TEST PROCEDURES

This test was performed under the direction of Mr. Paul Young of Limatorque Corporation in accordance with Limatorque Seismic Test of Limatorque SMB-3/H5BC and SMB-1/H3BC, Research and Development Project 678037 dated April 9, 1979.

Report No. 14801-1
Rev 1

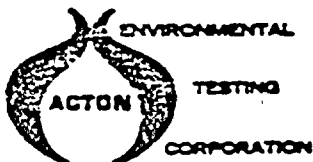




3.1 Test Mounting

The actuator under test was bolted to a test fixture. The test fixture was then secured to the appropriate vibration table. Refer to the individual test descriptions for specifics.

Report No. 14801-1





2000

3.2 Test Conditions

All tests were performed at room temperature and pressure.

During the resonance survey specified in section 3.4 below, the actuator under test was not operated.

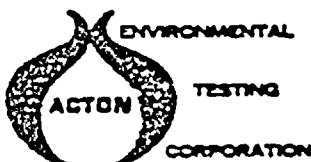
During the qualification test specified in section 3.5, the actuator under test was cycled open and closed once/dwell.

During the multiple frequency test specified in section 3.6 below, the actuator under test was cycled open and closed once/run.

During the aging test specified in section 3.7-below, the actuator under test was cycled at the beginning of each test and then once every fifteen (15) minutes during the test.

The actuator under test was subjected to some tests with the seismic support bracket and some tests without the bracket. Refer to the individual test descriptions for specific details.

Report No. 14801-1
Rev 1



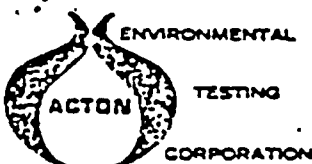
3.3 Test Monitoring

The actuator under test was visually monitored for any evidence of mechanical damage, deterioration or loss of its ability to operate properly where applicable.

The actuator under test was monitored with accelerometers by AETC personnel to determine its mechanical response. Data from these accelerometers through appropriate signal conditioning was recorded on visicorder recording paper included with this report for all tests except the multiple frequency tests during which data was recorded onto magnetic tape. During the multiple frequency tests, the output of the control accelerometer was also analyzed on-line and the X-Y plots of the test response spectra are included as part of this test report.

The accelerometers were located as follows:

Report No. 14891-1



Page 7

ACCELEROMETER NO. AXIS SENSING ORIENTATION* LOCATION**

1	H ₁	on H3BC Housing
2	H ₂	
3	V	
4	H ₁	on top of H3BC housing or on bracket when used
5	H ₂	
6	V	
7	H ₁	on SMB housing
8	H ₂	
9	V	
10	H ₁	on end cap of motor
11	H ₂	
12	V	
13	H ₁	on top of handwheel
14	H ₂	
15	V	
16	H ₁	on base of table
17	H ₂	
18	V	Control is #18

For tests 1A-9A, 16A-18A, the triaxial group of accelerometers 16, 17 and 18 was reoriented as follows: (#18 remained the control accelerometer).

* H₁ is the horizontal axis parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis.

Report No. 14801-1

**Refer to included photographs



Test Monitoring (continued)

ACCELEROMETER NO.	AXIS SENSING ORIENTATION*		V TESTS
	H ₁ TESTS	H ₂ TESTS	
16	V	V	H ₁
17	H ₂	H ₁	H ₂
18	H ₁	H ₂	V

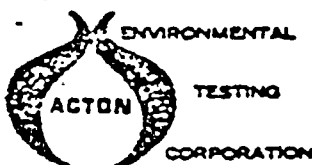
During the tests specified in sections 3.5, 3.6, and 3.7 below, one (1) closed set of contacts of the limit switch and one (1) closed set of contacts of the torque switch were independently monitored for contact chatter in excess of one (1) millisecond.

* H₁ is the horizontal axis parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis.

Report No. 14801-1
Rev 1





2/10/51

3.4 Resonance Survey

The resonance survey consisted of a sinusoidal input with peak horizontal or vertical accelerations of 0.75g at frequencies from 5 through 60 Hz and back to 5 Hz. The resonance survey was performed at a sweep rate of 1/2 octave/minute.

The resonance survey was performed both with and without the seismic support bracket using actuator #4. The input was applied using the AETC Ling A249 electrodynamic exciter as follows:

TEST NO.	AXIS OF EXCITATION*	SEISMIC SUPPORT BRACKET
1A	V	No
2A	V	Yes
3A	H ₂	Yes
4A	H ₂	No
5A	H ₁	No
6A	H ₁	Yes

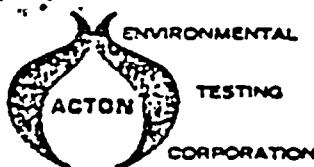
*Refer to the included photographs

H₁ is the horizontal axis parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis

Report No. 14801-1





2/13, 5!

3.5 Qualification Test with Seismic Support Bracket

The qualification test with the seismic support bracket consisted of a continuous steady state sinusoidal dwell input with peak horizontal or vertical accelerations of 6.0g except where limited by AETC shaker table capabilities as specified below. The sinusoidal dwell test was performed at 1/3 octave intervals of frequency from 5.0 through 32 Hz (5.0, 6.3, 8.0, 10.0, 12.5, 16.0, 20.0, 25.0, 32.0) as well as at 35 Hz. Each dwell was for a duration of fifteen (15) seconds.

This qualification test was performed with the seismic support bracket using actuators #3 and #4.

The input was applied using the AETC Ling A249 electrodynamic exciter as follows:

TEST NO.	AXIS OF EXCITATION*	ACTUATOR NO.
7A	H ₁	4
8A	H ₂	3
9A	V	3

The level of input applied was as follows:

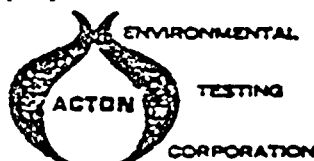
**Refer to the included photographs*

H₁ is the horizontal axis parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis

Report No. 14801-1



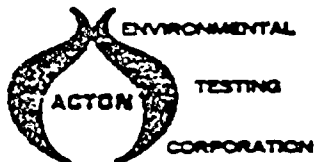


24-51

Qualification Test with Seismic Support Bracket (continued)

<u>FREQUENCY (Hz)</u>	<u>INPUT LEVEL (g)</u>
5.0	1.0
6.3	1.5
8.0	3.0
10.0	5.0
12.5	5.5
16.0	6.0
20.0	6.0
25.0	6.0
32.0	6.0
35.0	6.0

Report No. 14801-1



3.6 Multiple Frequency Test

A biaxial multiple frequency excitation was applied. The test input was recorded on a 14-channel tape recorder, each track having discrete frequency sine beats recorded at a different frequency and delay between beats. All frequencies were recorded at maximum levels.

The input was played back through a 14-channel tape recorder. The outputs of the 14-channels were then combined in a 14-channel mixer which resulted in a multiple frequency output.

The individual mixer channels had gain controls so that the level of each output tape channel passing through the mixer could be controlled. In this manner, the required test spectrum was shaped by controlling the level of individual frequencies.

Qualification tests, consisting of biaxial periodic pseudo-random excitation were performed.

The test duration for each input was thirty (30) seconds.

The level was such that the Test Response Spectra (TRS) from the control accelerometer computed at $Q=10$ (5% damping) would envelop the SSE Required Response Spectra (RRS) shown in Figure 1.

The test response spectra were also analyzed at $Q=20$ (2.5% damping) and $Q=50$ (1.0% damping).

Report No. 14801 -1

2017

Multiple Frequency Test (continued)

The multiple frequency test was performed with the seismic support bracket using actuator #3. The input was applied using the AETC hydraulic seismic simulator as follows:

<u>TEST NO.</u>	<u>BIAXIAL DIRECTION OF EXCITATION*</u>
10A-11A**	-H ₁ & V
12A	-H ₁ & V
13A	-H ₂ & V
14A	H ₂ & V
15A	H ₁ & V

**Refer to the included photographs*

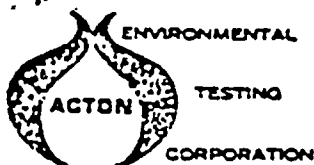
H₁ is the horizontal axis parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis

***Shaker table equalization runs.*

Report No. 14861-1





200.51

3.7 Plant Vibration Aging Test

The plant vibration aging test consisted of a sinusoidal input with peak horizontal or vertical acceleration of $0.75g$ at frequencies from 10 thru 100 Hz and back to 10 Hz. The plant vibration aging test was performed at a sweep rate of two (2) octaves/minute. The input was applied for 90 minutes/axis. The plant vibration aging test was performed with the seismic support bracket using actuator #3. The input was applied using the AETC Ling A249 electrodynamic exciter as follows:

TEST NO.	AXIS OF EXCITATION*
16A	H_1
17A	H_2
18A	V

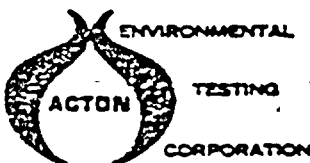
*Refer to the included photographs.

H_1 is the horizontal axis parallel to the motor axis

H_2 is the horizontal axis perpendicular to the motor axis

V is the vertical axis

Report No. 14801-1
Rev 1





4.0 TEST RESULTS

4.1 Resonance Survey Test Results

The following resonances were detected during the resonance survey specified in section 3.4 above:

TEST NO.	AXIS OF EXCITATION*	SEISMIC SUPPORT BRACKET	RESONANCES (Hz)
1A	V	No	None detected
2A	V	Yes	None detected
3A	H ₂	Yes	49.0, 58.0
4A	H ₂	No	41.0, 46.0, 58.0
5A	H ₁	No	34.0, 41.0, 47.0
6A	H ₁	Yes	56.0

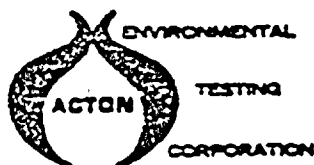
*Refer to the included photographs

H₁ is the horizontal parallel to the motor axis

H₂ is the horizontal axis perpendicular to the motor axis

V is the vertical axis

Report No. 14801-1



2/2-51

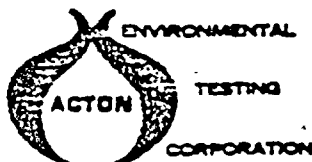
4.2 Qualification Test Results with Seismic Support Bracket

During the 35 Hz portion of test 7A, (6.0g), the interface gasket between the clutch housing and the main housing vibrated out and the three (3) 3/4" bolts of the seismic support bracket loosened up.

At this time, actuator #4 was replaced with actuator #3.

There was no evidence of mechanical damage, deterioration or contact chatter as a result of the qualification test specified in section 3.5 above other than that specified above.

Report No. 14801-1
Rev 1



Page 17

THE UNIVERSITY OF CHICAGO

DEPARTMENT OF THE HISTORY OF ARTS

THE UNIVERSITY OF CHICAGO

DEPARTMENT OF THE HISTORY OF ARTS

THE UNIVERSITY OF CHICAGO

DEPARTMENT OF THE HISTORY OF ARTS

THE UNIVERSITY OF CHICAGO

DEPARTMENT OF THE HISTORY OF ARTS

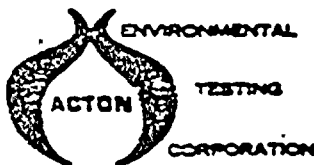
THE UNIVERSITY OF CHICAGO

218.51

4.3 Multiple Frequency Test Results

There was no evidence of mechanical damage, deterioration or contact chatter as a result of the multiple frequency test specified in section 3.6 above.

Report No. 14801-1

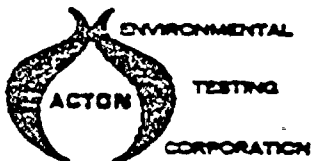


Page 18

4.4 Plant Vibration Aging Test Results

There was no evidence of mechanical damage, deterioration, or contact chatter as a result of the plant vibration aging test specified in section 3.7 above.

Report No. 14801-1
Rev 1



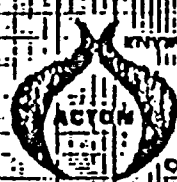
5.0 TEST EQUIPMENT LIST

NAME	MFGR.	MODEL	SER. NO.	RANGE	ACCURACY	INV. #	CAL. FREQ.
Accelerometer	B&K	4343	472687	1-8000 Hz	+5%	AC307	3 months
"	"	"	472688	"	"	AC309	" "
"	"	8302	344778	1 Hz - 5 KHz	"	AC310	" "
"	"	"	344783	"	"	AC318	" "
"	"	"	344781	"	"	AC322	" "
"	"	"	450710	"	"	AC323	" "
"	"	"	472976	"	"	AC329	" "
"	"	"	472989	"	"	AC336	" "
"	"	"	473010	"	"	AC338	" "
"	PCB	302A	2834	"	"	AC367	" "
"	"	"	2843	"	"	AC371	" "
"	"	"	2844	"	"	AC374	" "
"	"	"	666	0.25 - 5 KHz	"	AC375	" "
"	"	"	667	"	"	AC376	" "
"	"	"	670	"	"	AC379	" "
"	"	"	671	"	"	AC380	" "
"	"	"	672	"	"	AC381	" "
"	"	"	673	"	"	AC382	" "
"	"	"	2845	1 Hz - 5 KHz	"	AC383	" "
"	"	"	694	0.25 Hz - 5 KHz	"	AC384	" "
"	"	"	2851	1 Hz - 5 KHz	"	AC392	" "
"	"	"	2853	"	"	AC395	" "
"	"	"	2854	"	"	AC396	" "
"	"	"	2855	"	"	AC410	" "



TEST EQUIPMENT LIST

NAME	MFGR.	MODEL	SER. NO.	RANGE	ACCURACY	INV.#	CAL. FREQ.
Accelerometer	PCD	302A	2856	1 Hz - 5 KHz	+5%	AC414	3 months
"	"	"	1772	"	"	AC415	" "
"	"	"	2857	"	"	AC444	" "
"	"	303A03	740	1 Hz - 70 KHz	"	AC464	" "
"	"	"	741	"	"	AC465	" "
"	"	"	743	"	"	AC467	" "
"	"	"	745	"	"	AC469	" "
Timer Seconds	SE	S60		0-60 seconds 1 rpm	+0.1sec	FM311	6 months
Scope, Storage	Tektronix	564-	11582	DC - 10 MHz	+3%	OS309	3 months
"	"	"	9027	"	"	OS311	" "
Filter-Dual	Ithaco	4302	35207	10 - 1 MHz	+3%	AM346	6 months
VTVH	Dallintine	310A	5580	10 Hz - 2 MHz, 0-100 volts 6 ranges	+5%	HV305	3 months
Sweep Oscillator	SDY	SD-104-5	21A	0.005 Hz - 50 KHz	+1%	SG315	6 months
Audio Generator	HFA	1G72		10 Hz - 100 KHz	+5%	SG327	" "
Visicorder	Honeywell	906B	8687	DC - 2 KHz 12 channel	+1db	RE301	3 months
"	"	906C2	99334	"	"	RE311	" "
Recorder Tape	Amp	FR1300	7430293	14 channel FM 1" Tape 0-2.5 KHz	+0.5%	RE331	" "
"	Honeywell	906C	99078	DC - 2 KHz 12 channel	+1db	RE335	" "
Recorder X-Y	MFE	715E	42167	Input: 1-10-100 MV 1-10V Both channels	+0.5%	RE340	" "
Visicorder	Honeywell	1508	15-1670	DC-2 KHz 12 channel 8" paper	+1db	RE344	" "



ACTON
ENGINEERING
CORPORATION

Test No. 100
Date: APRIL 1, 1970
Customer: LITTON 5-1490
Test Item: N
Test Item: S/N
Type: Test Blank
Spec. No. 100-385, DRIFT 8
Part No. 100-385, DRIFT 12
Conditions: OPERATING
Temperature: ROOM
Period: 30 SECONDS
Control: VIBRATION
Pick-up No. CONTROL
Pick-up Axis: VERTICAL
Operator: SKEGG
Test Engineer: SKEGG

GRMS

REQUIRED RESPONSE SPECTRA (RRS)
Q=10 (5% DAMPING)
TO BE ANALYZED @ Q=10, 20, 50

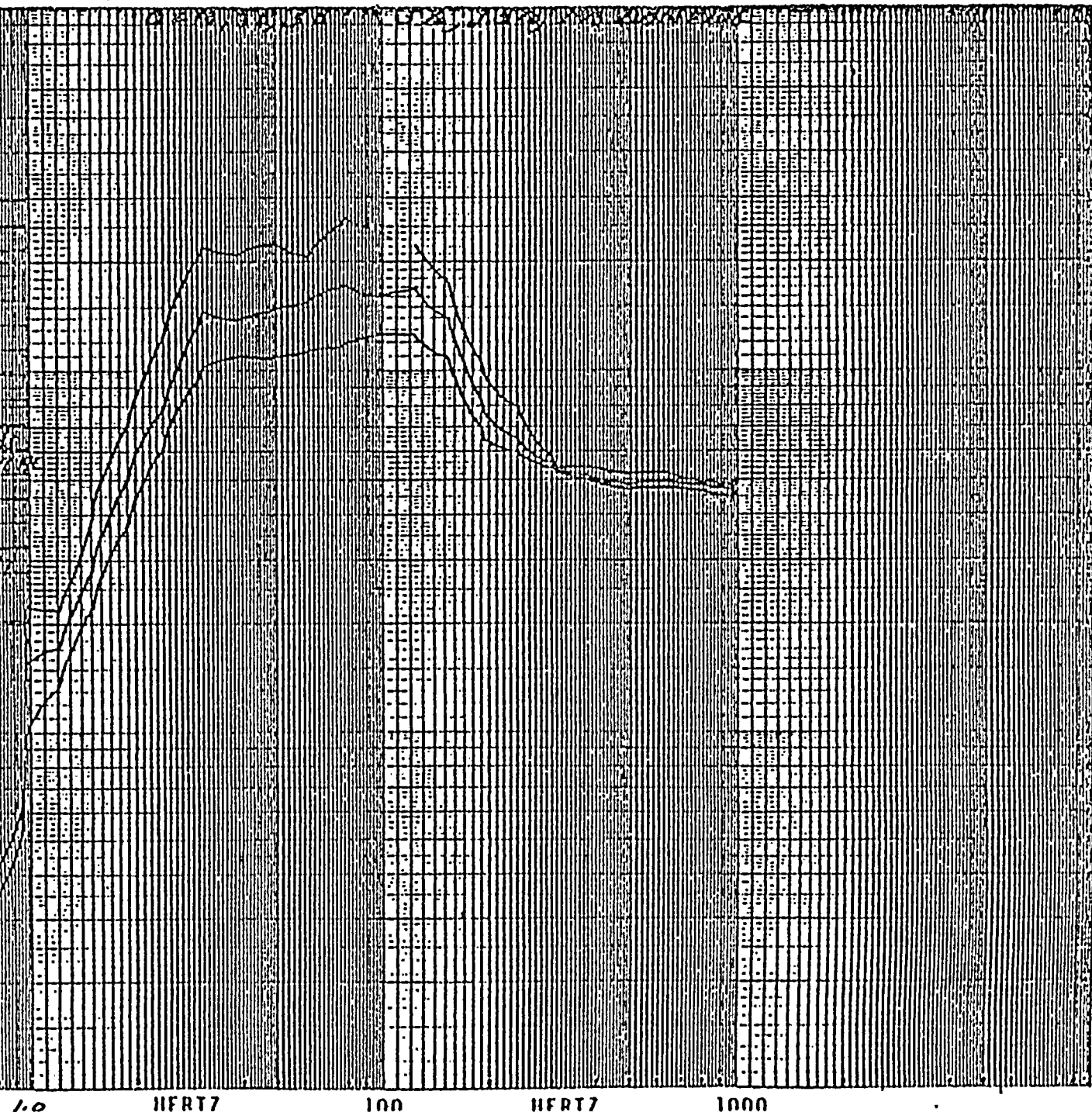
FIGURE





Form No. 1
Date 8/2/57
Customer L. H. W. W. W. W.
Test No. 1000
Type of Test Static
Spec. No. 1000
Test No. 1000
Conditions of Test 1000
Temperature 1000
Period of Test 1000
Cobalt No. 1000
Fiducial No. 1000
Operator 1000
Test No. 1000

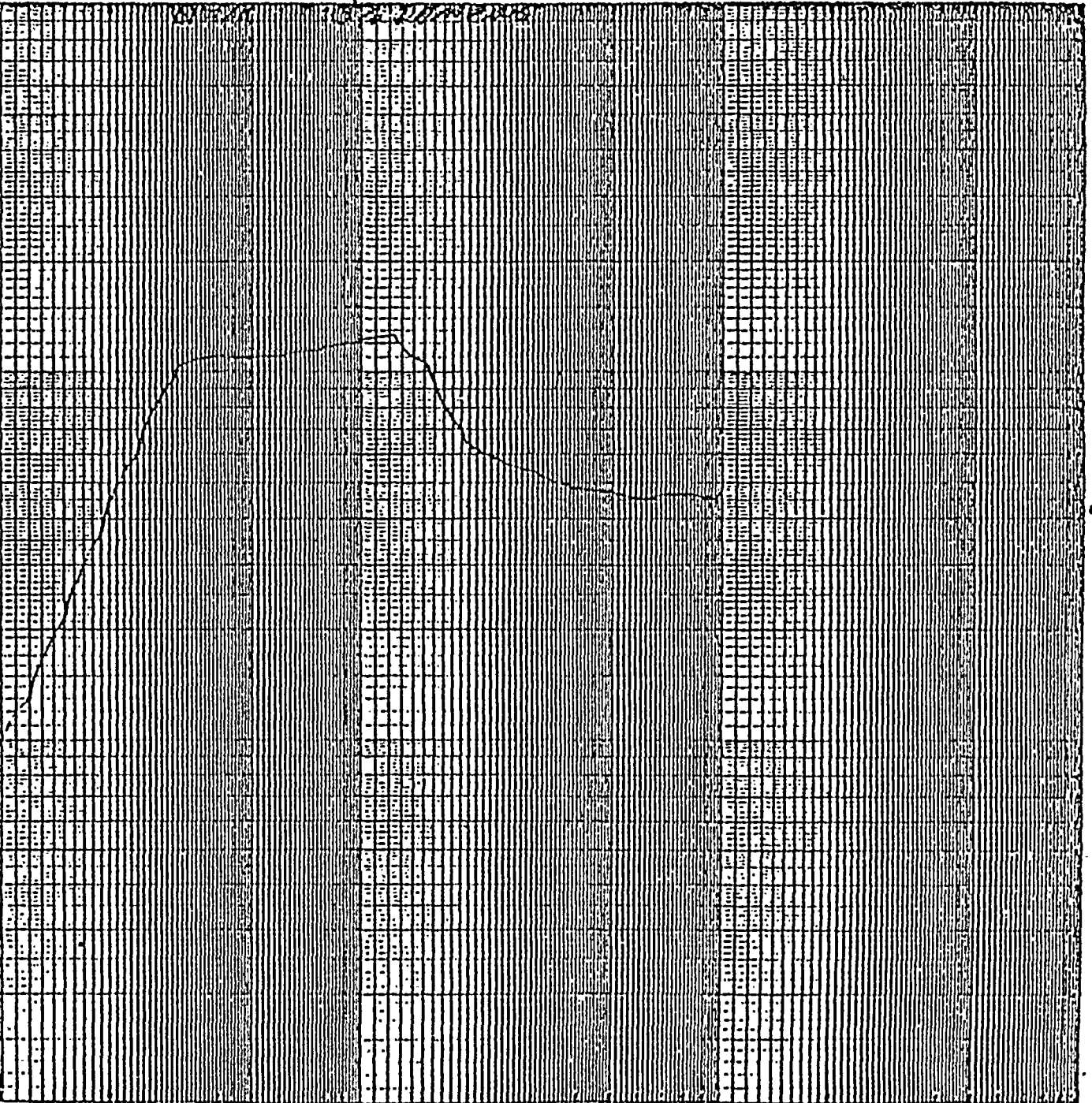
GRMS





Title: *1000*
 Date: *10/22/51*
 Location: *1000*
 Test No.: *1000*
 Spec. No.: *1000*
 Condition: *1000*
 Period of Test: *1000*
 Control: *1000*
 Operator: *1000*
 Test No.: *1000*

CHART



HERTZ

100

HERTZ

1000

10.

100

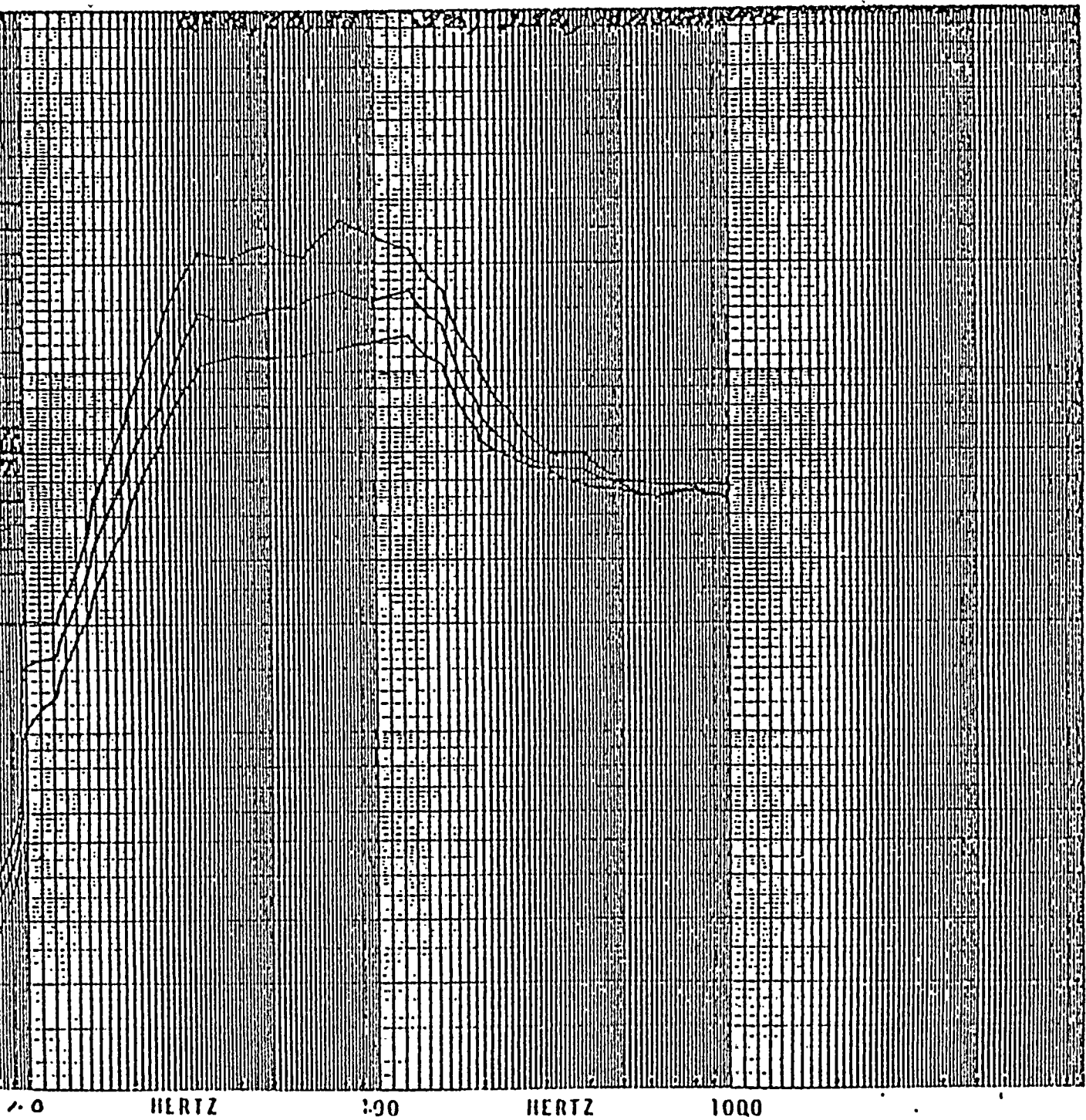
10

10.51



Test No. 100
 Date 5/1/50
 Customer General Electric
 Test Unit 100
 Test Item 100
 Type of Test 100
 Spec. No. 100
 Test No. 100
 Conditions 100
 Remarks 100
 Period of Test 100
 Control 100
 Pickup No. 100
 Pickup Amp. 100
 Operator 100
 Test Room 100

GRMS



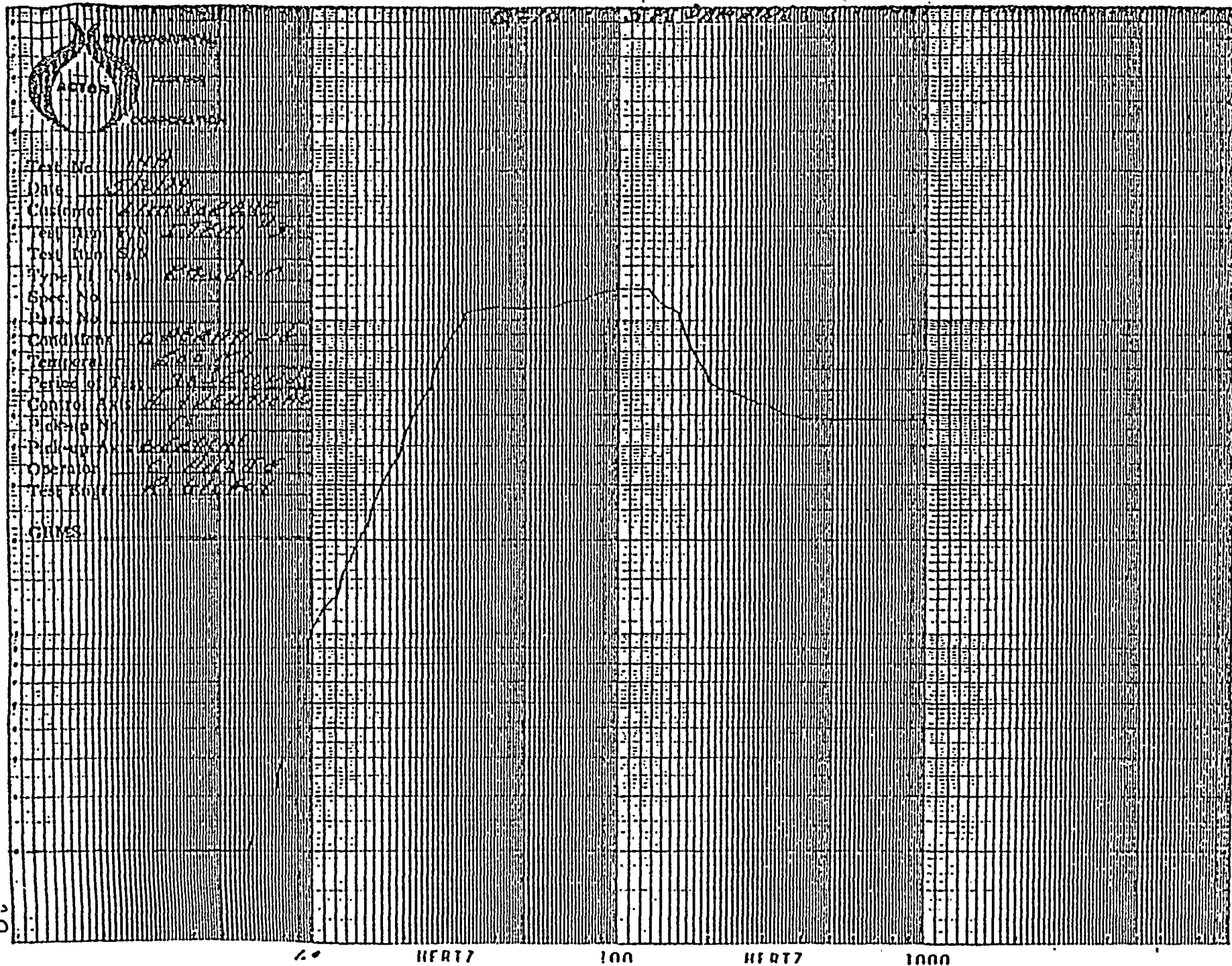
10

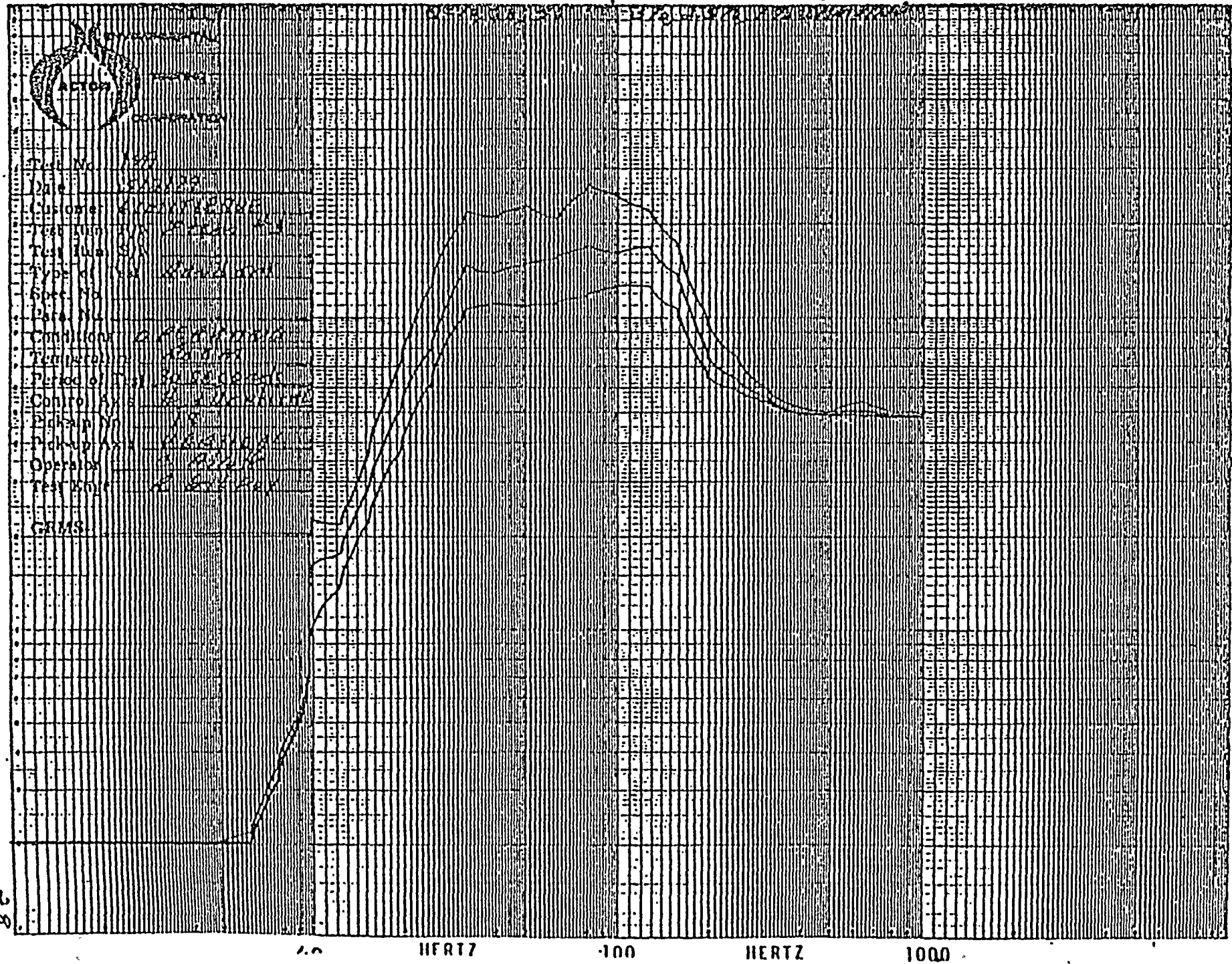
20

10

210.51

26





10

10

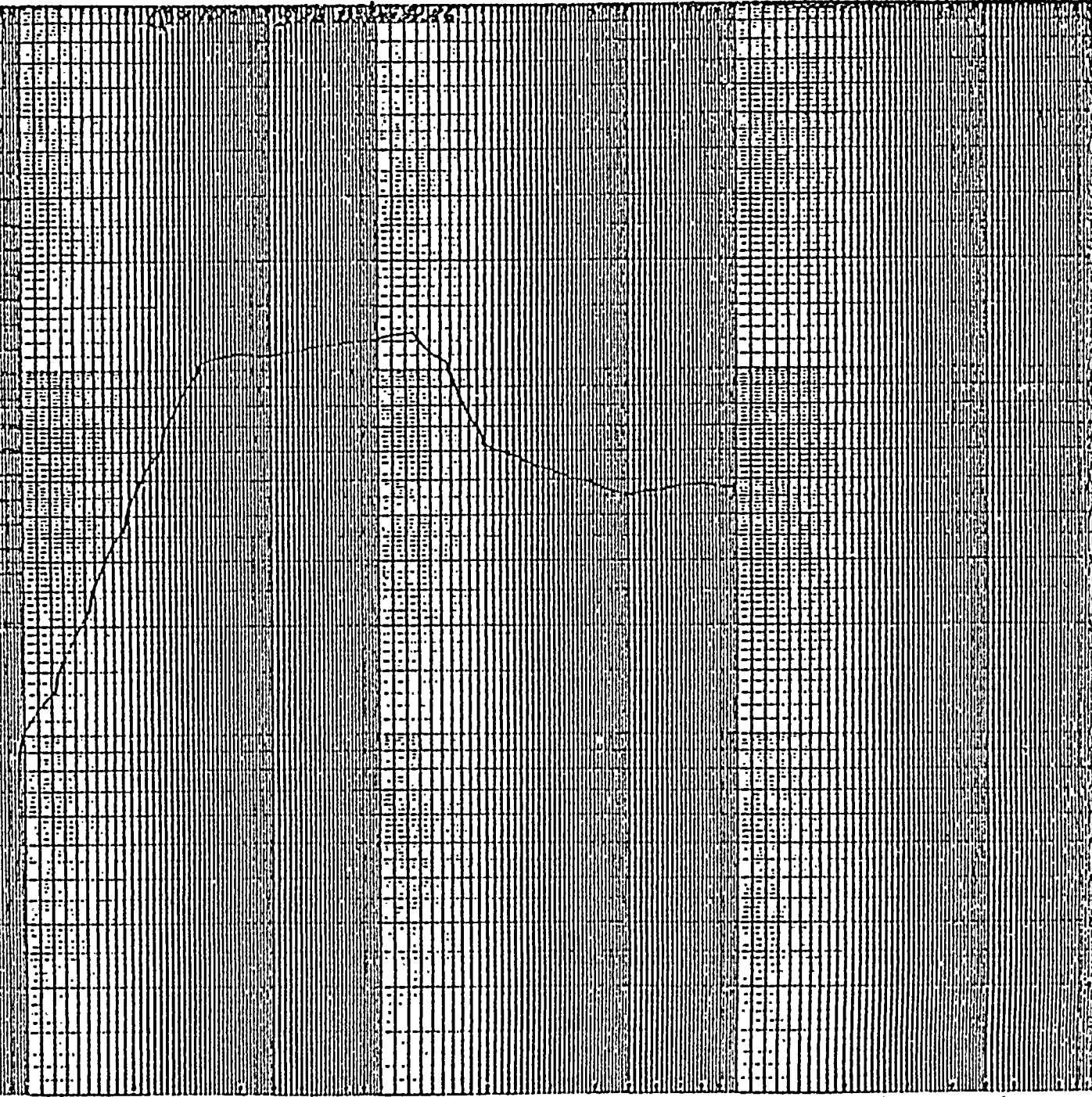
2/10/51





Date *12/27/51*
 Customer *Lockheed*
 Test Item *Engine*
 Test Plan *SI*
 Type of Test *Endurance*
 Spec. No.
 Para. No.
 Conditions *2800 RPM*
 Temperature *220 F*
 Period of Test *10 hours*
 Control Axis *Engine*
 Checkup No.
 Checkup *Engine*
 Operator *Engine*
 Test Point *Engine*

GMS



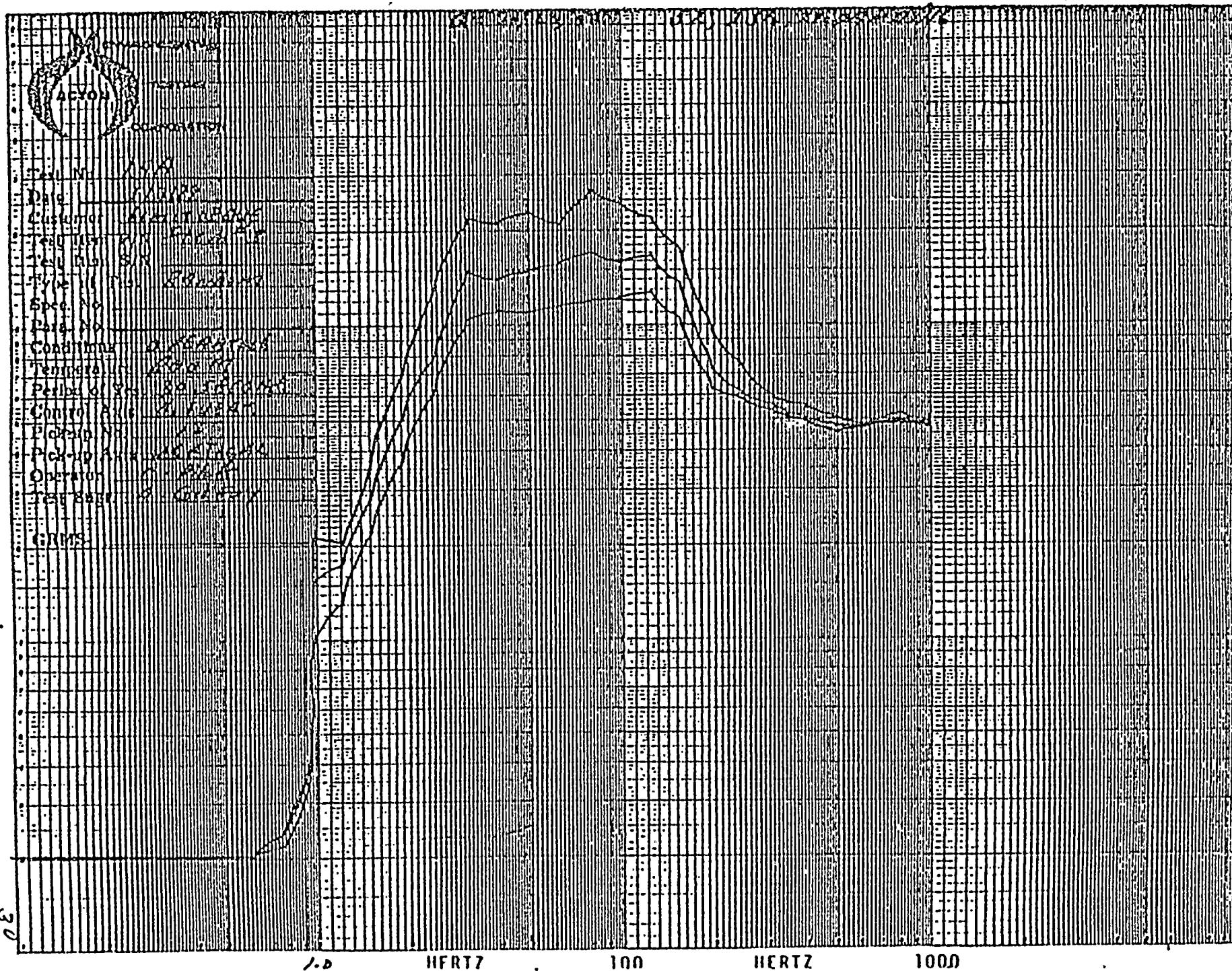
be

10.

5

10

210.51



ACVON

Test No. 1000

Date 10/1/55

Customer ACVON

Test No. 1000

Test No. 1000

Spec. No. 1000

Part No. 1000

Condition 1000

Temperature 1000

Pressure 1000

Control Axis 1000

Power No. 1000

Pick-up Axis 1000

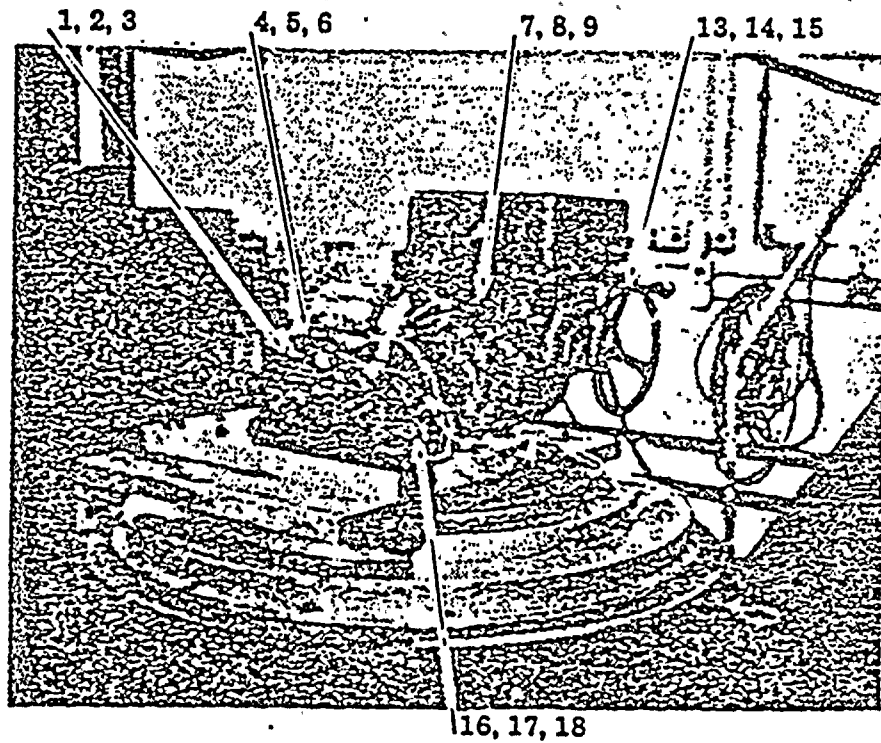
Operator 1000

Test Date 10/1/55

CHMS

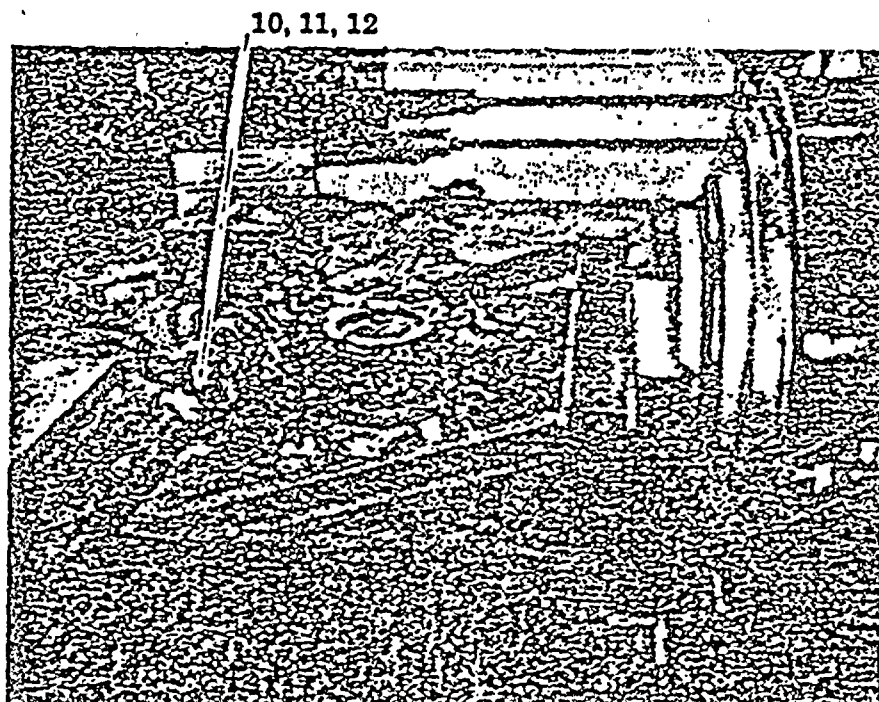
010.51





ACCELEROMETER LOCATIONS
VERTICAL AXIS
ACTUATOR #4
TESTS 1A & 2A
ACTUATOR #3 SIMILAR FOR TESTS 9A & 18A

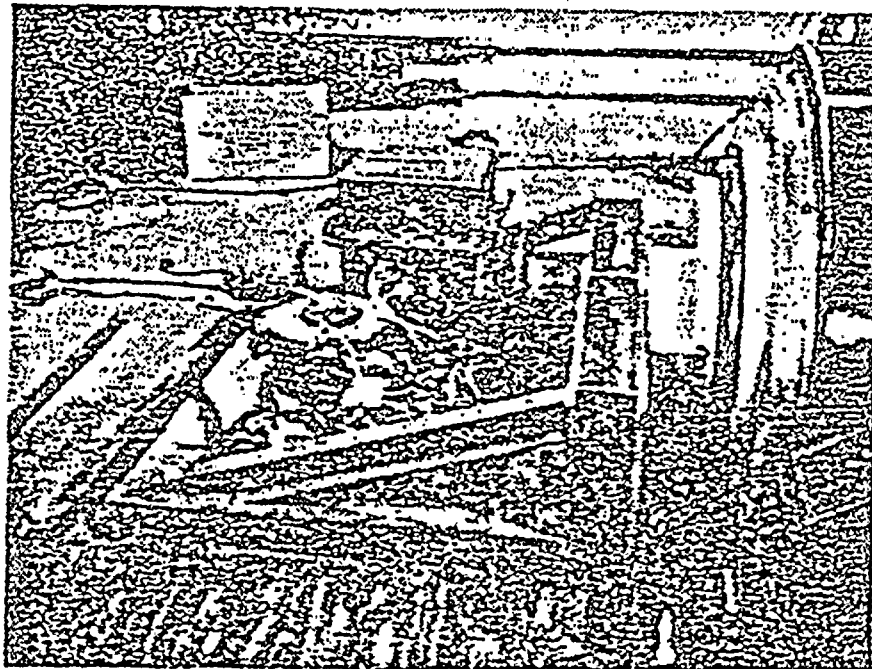
Report No. 14801-1



ACCELEROMETER LOCATIONS
H₂ AXIS
ACTUATOR #4
TESTS 3A & 4A

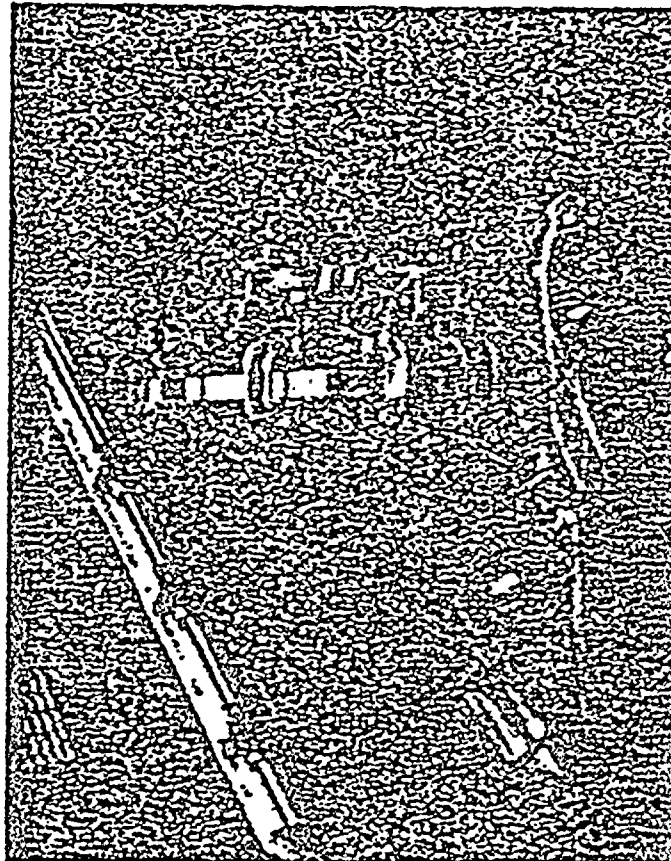
Report No. 14801-1





H₁ AXIS
ACTUATOR #4
TESTS 5A, 6A, 7A
(ACTUATOR #3 SIMILAR FOR TEST 16A)

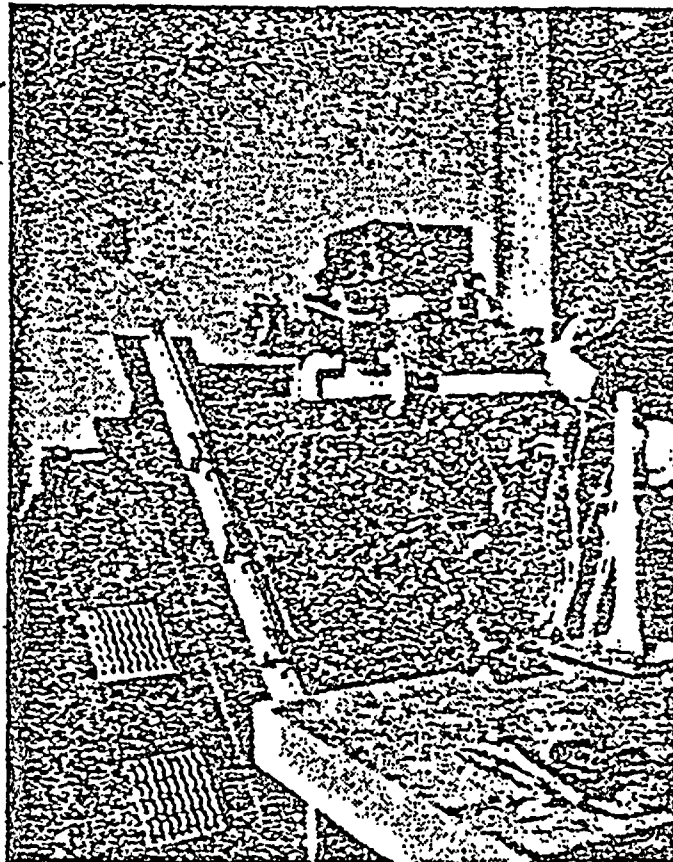
Report No. 14801-1



-H₁ & V
BIAXIAL DIRECTION OF EXCITATION
ACTUATOR #3
TESTS 10A-12A

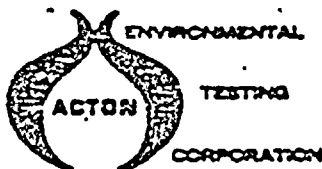
Report No. 14801-1

210.51



-H₂ & VERTICAL
BIAXIAL DIRECTION OF EXCITATION
ACTUATOR #3
TEST 13A

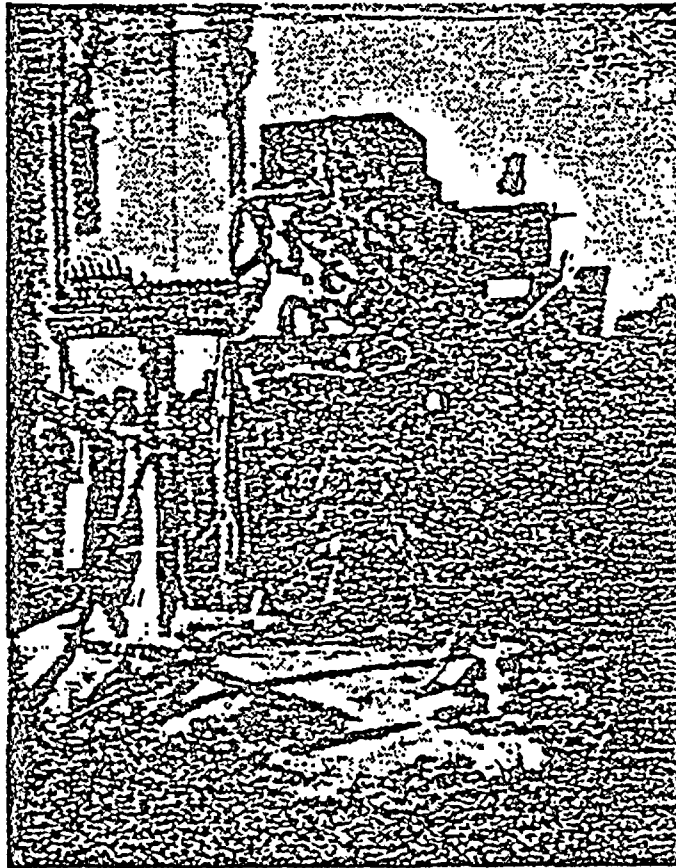
Report No. 14801-1



Page 35

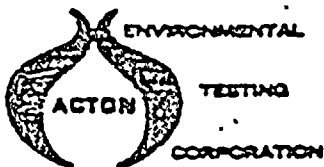


2/0, 5/



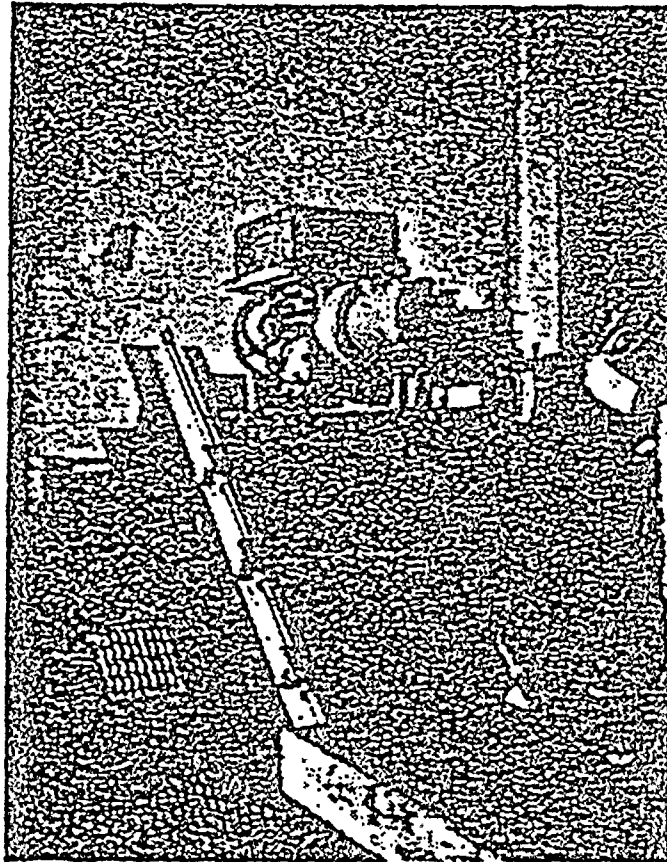
H₂ & V
BIAXIAL DIRECTION OF EXCITATION
ACTUATOR #3
TEST 14A

Report No. 14801-1



Page 36

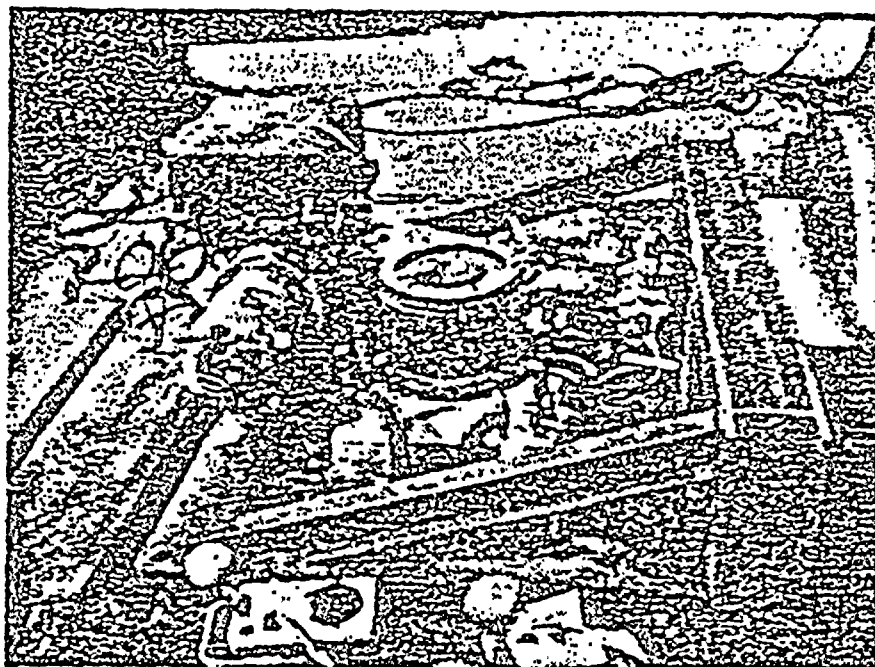




H₁ & V
BIAXIAL DIRECTION OF EXCITATION
ACTUATOR #3
TEST 15A

Report No. 14801-1





H₂ AXIS
ACTUATOR #3
TESTS 8A & 17A

Report No. 14801-1