

Design Analysis

Ginna Station

EVALUATION OF GINNA RCS COOLANT TEMPERATURE  
TO SUPPORT LTOPS REQUIREMENTS

Rochester Gas and Electric Corporation  
89 East Avenue  
Rochester, New York 14649

DA-ME-97-031

Revision 0


April 14, 1997

Prepared by:

  
Responsible Engineer

4/15/97  
Date

Reviewed by:

  
Independent Reviewer or Group Leader

4/23/97  
Date

TECHNICAL INPUT FORM				
EIN	RRC01			
KEYWORDS	Brittle Fracture, 10CFR50 Appendix G			
CROSS REF	LTOPS			
PSSL 01	EWR/ OTHER	PROPRIETARY	YES	NO X
COMMENT				
SUPERSEDES				



Revision Status Sheet

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All	0

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1.0

OBJECTIVE

This evaluation will calculate the RCS coolant temperature such that corresponding metal temperature at a distance one-fourth of the RV section thickness from the inside surface will not be less than RTndt + 50 F. This is needed to protect the reactor vessel from being exposed to conditions of fast propagating brittle fracture per requirements of 10CFR50 Appendix G.

2.0

CONCLUSIONS

This analysis showed that for a vessel beltline weld material RTndt of 232 F, an instrumentation uncertainty of 21.1 F, and consideration of conservative heat transfer parameters, the temperature difference between coolant and 1/4 thickness is 3.2 F and corresponding RCS coolant temperature is 306.3 F.

3.0

DESIGN INPUT

3.1

10 CFR Part 50 Appendix G

3.2

ASME Code Case N-514, "Low Temperature Overpressure Protection", Section XI, Division 1

3.3

WCAP-14864, "R. E. Ginna Heatup and Cooldown Limit Curves For Normal Operation", June 1996.

3.4

USNRC RG 199, Rev. 2, "Radiation Embrittlement of Reactor Vessel Material", May 1988

4.0

REFERENCES

4.1

RG&E Analysis, DA-EE-93-154, "Uncertainty Of Cold Leg Temperature Recording", EWR 10106

4.2

CEO, P. Bamford to A. Rochino, "RV Annulus Temperatures", 4/2/97.

4.3

Marks' Standard Handbook For Mechanical Engineers, 9th Edition, by E. A. Avallone, and T. Baumeister

4.4

V. M. Faires, "Thermodynamics", 4th Edition

4.5

Westinghouse Dwg No. 117804E, Rev. 7, "Material List Reactor Vessel."

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DESIGN INPUT

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10 CFR Part 50 Appendix G

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CEO, P. Bamford to A. Rochino, "RV Annulus Temperatures", 4/2/97.

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- 4.6 Westinghouse Dwg No. 117849E, Rev. 2, "General Outline  
Of Reactor Vessel" . . .
- 4.7 TRANSCO, Inc. Dwg No. TP 3609-1, "Elev. of Reactor  
Vessel (Typical Panel Sections).



## 5.0 ASSUMPTIONS

- 5.1 Assumptions, if used, to simplify analysis and/or provide conservative results are fully documented in the body of this design analysis and do not require verification at a later date.
- 5.2 Since the results of this calculation will be utilized to justify Ginna LTOP Enable Temperature, conservative values of heat transfer parameters will be used in the calculation so as to give a higher RCS coolant temperature.

## 6.0 COMPUTER CODES

- 6.1 None

## 7.0 ANALYSIS

The temperature of the RCS coolant will be determined based on a known temperature at a point 1/4 of the RV section thickness from the inside surface of the vessel. Steady state heat transfer methodology is then employed since the air annulus temperature outside of vessel is also known. The film coefficients at the inside and outside surfaces are conservatively taken to represent typical conditions as listed in heat transfer textbooks.

### 7.1 Heat Transfer From 1/4 Thickness To Outside

By definition, resistance to heat transfer in cylindrical solid walls is given by the equation, (Reference 4.4)

$$R_w = - \frac{DT}{Q} = \frac{\ln(D_o/D_i)}{2\pi ZK} \quad (1)$$

Where: DT = Temperature difference between the wall surfaces, deg F

Q = Heat Transfer, Btu/hr

D<sub>o</sub> = Outside surface diameter, ft

D<sub>i</sub> = Inside surface diameter, ft

Z = Length of cylinder, ft

K = thermal conductivity of cylindrical wall,  
Btu ft/hr-sq ft-F

Resistance of films at surfaces of the cylindrical walls is expressed by the relation,

$$R_i = 1/A_i H_i \quad ; \quad \text{or} \quad (2)$$

$$R_o = 1/A_o H_o$$

Where:  $R_i$  = Resistance of film inside cylinders,  
(deg F-hr)/Btu

$R_o$  = Resistance of film outside cylinders,  
(deg F-hr)/Btu

$A_i$  = Inside surface area, sq ft

$A_o$  = Outside surface area, sq ft

$H_o$  = Outside film coefficient, Btu/(hr sq ft F)

$H_i$  = Inside film coefficient, Btu/(hr sq ft F)

The heat transfer path from the 1/4 thickness section to the outside is made up of composite cylinders representing the remaining part of the reactor vessel and the insulation, each cylinder having its own resistance. This is shown in Figure 1 below. In addition, there will be a film resistance at the outside surface.

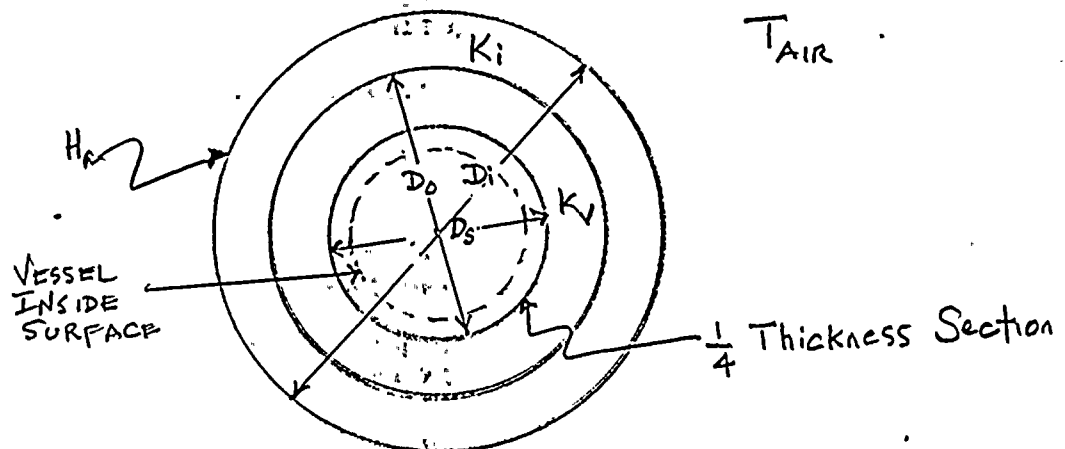


Figure 1.  
Composite Cylinders From 1/4 Thickness To Outside

The heat transfer through the series of resistances is expressed as follows:

$$Q = DT / (\text{Sum of Resistances in Path}) \quad (3)$$

Where DT is the temperature difference between 1/4 thickness section and the outside annulus temperature. The resistances in series consist of the remaining vessel thickness, the insulation, and the outside film. Details of Equation 3 is shown below.

$$Q = \frac{(T_s - T_{air})}{\frac{\ln(D_o/D_s)}{2\pi K_v} + \frac{\ln(D_i/D_o)}{2\pi K_i} + \frac{1}{A_i h_a}} \quad (4)$$

Where:

- $T_s$  = Temperature at 1/4 thickness section  
= 282 F (Design Input 3.3)
- $T_{air}$  = Annulus temperature surrounding the vessel, 50F (Reference 4.2).
- $D_o$  = Outside diameter of the vessel  
= 145" or 12.0833' (Reference 4.6)
- $D_s$  = Diameter of 1/4 thickness section  
= 135.25" or 11.2708' (Reference 4.6)
- $D_i$  = Outside diameter of insulation, 3" thick per Reference 4.7  
= 151" or 12.5833'
- $A_i$  =  $\pi D_i$  = Insulation Surface area, sq ft  
= 12.5833  $\pi$  sq ft
- $K_v$  = Thermal conductivity of vessel material which is ASTM A508 (Ref. 4.5)  
Btu ft/hr-sq ft-F;  
From Table 4.4.1 of Reference 4.3, conductivity of steel is given as 26.2 Btu ft/hr-sq ft-F

Ki = Thermal conductivity of vessel insulation, which is Kaowool (Ref.4.7)  
 = 0.059 Btu ft/hr-sq ft-F from Table 4.4.3 of Reference 4.3.

Ha = film coefficient at the surface of the insulation with the annulus room.  
 From Table 4.4.9 of Reference 4.3 a typical value for air outside of tubes is given as 7.5 Btu/hr-sq ft-F. This will be increased to 10 Btu/hr-sq ft-F for conservative value.  
 = 10 Btu/hr-sq ft-F

Z = Length of Cylinder, which will be assumed equal to 1 ft.

Substituting the above values into Equation 4, we have:

$$\begin{aligned}
 Q &= \frac{(282 - 50)}{\frac{\ln(12.0833/11.2708)}{2\pi(1)(26.2)} + \frac{\ln(12.5833/12.0833)}{2\pi(1)(0.059)} + \frac{1}{\pi(12.5833)(10)}} \\
 &= \frac{232}{0.000422848 + 0.1093822 + 0.0025296} \\
 &= 232/0.1123347 \\
 &= 2065.2573 \text{ Btu/hr for 1 ft vessel length} \quad (5)
 \end{aligned}$$

## 7.2 Heat Transfer From 1/4 Thickness To RCS Coolant

The heat transfer path from the 1/4 thickness section to the inside part of the vessel, containing the RCS coolant is also made up of composite cylinders representing the cladding, and the 1/4 section of the vessel. In addition, there is also a film resistance at the inside surface. This is shown in Figure 2, which is depicted in the next page.



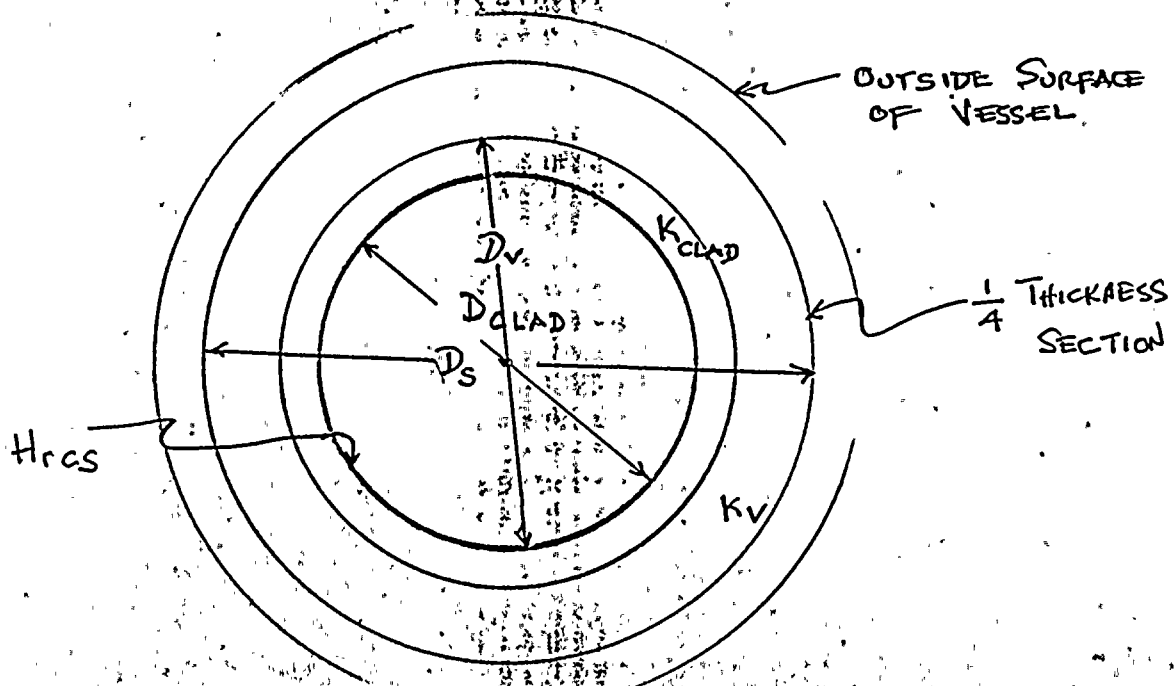


Figure 2  
Composite Cylinders From 1/4 Thickness To Inside

The heat transfer through the series of resistances in Figure 2 is expressed below.

$$Q = DT / (\text{Sum of Resistances in Path}) \quad (3)$$

Where DT is the temperature difference between coolant temperature, Trcs and the metal temperature at 1/4 thickness, Ts which is equal to 282 F (Design Input 3.3). Considering the resistance formulations in Equations (1) and (2), the heat transfer equation in (3) becomes,

$$Q = \frac{(Trcs - Ts)}{\frac{\ln(Ds/Dv)}{2\pi ZKv} + \frac{\ln(Dv/Dclad)}{2\pi ZKclad} + \frac{1}{Aclad Hrcs}} \quad (6)$$

Where:  $T_s$  = Temperature at 1/4 thickness section  
 = 282 F (Design Input 3.3)

$T_{rcs}$  = Temperature of coolant inside the vessel, F. This will be calculated.

$D_s$  = Diameter of 1/4 thickness section  
 = 135.25" or 11.2708'

$D_v$  = Inside diameter of the vessel  
 = 132" or 11.0'

$K_v$  = Thermal conductivity of vessel material which is ASTM A508 (Ref. 4.5), Btu ft/hr-sq ft-F.  
 From Table 4.4.1 of Reference 4.3, conductivity of steel is given as 26.2 Btu ft/hr-sq ft-F.

$K_{clad}$  = Thermal conductivity of cladding material which is Type 304 SS, Btu ft/hr-sq ft-F. From Table 4.4.1 of Ref. 4.3, conductivity of stainless steel is also equal to 26.2 Btu ft/hr-sq ft-F.

$D_{clad}$  = Inside diameter of cladding  
 = 131.688" or 10.974'

$A_{clad} = \pi Z D_{clad} =$  Surface area inside clad  
 = 10.974  $\pi$  Z sq ft

$h_{clad}$  = Film coefficient at the inside surface of the vessel clad. From Table 4.4.9 of Reference 4.3, a typical value for water inside pipe is 1260 Btu/hr-sq ft-F. For conservative results, we will use a film coefficient value of 1000 Btu/hr-sq ft-F.

$Z$  = Length of vessel, which will be assumed equal to 1 ft.

Substituting the above values into Equation (6), we obtain,

$$Q = \frac{(Trcs - 282)}{\frac{\ln(11.2708/11.0)}{2\pi(1)(26.2)} + \frac{\ln(11.0/10.974)}{2\pi(1)(26.2)} + \frac{1}{\pi(10.974)(1000)}}$$

$$Q = \frac{Trcs - 282}{0.000147735 + 0.000014375 + 0.000029006} \quad (7)$$

$$Q = \frac{Trcs - 282}{0.000191116} \quad (7')$$

The heat flow in Equation 7' is equal to that in Equation 5. Consequently, we have

$$Q = 2065.2573(0.000191116) = Trcs - 282$$

Solving for Trcs,

$$Trcs = 282 + 0.395 = 282.395 \text{ F}$$

A conservative assumption will be now be considered where the vessel insulation resistance decreases by an order of magnitude, i.e., by a factor of 10. In Equation (4) this will increase the heat flow to,

$$\begin{aligned} Q &= 232/(0.000422848 + 0.01093822 + 0.0025296) \\ &= 232/0.013890668 \\ &= 16701.86 \text{ Btu/hr per ft length of vessel} \end{aligned}$$

Substituting this value into Equation (7'), we can solve for the coolant temperature,

$$Trcs = 282 + 16701.86(0.000191116) = 285.2 \text{ F} \quad (8)$$

SUMMARY OF RESULTS

For a vessel beltline weld material RTndt of 232 F, and a corresponding metal temperature at a distance one-fourth of the vessel thickness from the inside surface of 282 F, and an instrumentation uncertainty of 21.1 F (Reference 4.1), this calculation has shown the following results:

Temperature Difference Between Coolant  
and 1/4 Thickness = 3.2 F

RCS Coolant Temperature = 285.2 + 21.1  
= 306.3 F



Design Analysis  
Ginna Station  
Uncertainty of RCS Cold Leg Temperature Recording

Rochester Gas and Electric Corporation  
89 East Avenue  
Rochester, New York 14649

DA-EE-93-154

Revision 0

EWR 10106

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11-22-93

Date

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Discipline Reviewer

11-22-93

Date

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Nuclear Safety & Licensing  
"When Required Otherwise "N/A" and Initials"

Date

Approved by:

Charles A. Tinkell  
Discipline Manager

11-24-93

Date



Document Control Sheet

NUCLEAR SAFETY & LICENSING  
INQUIRY DATA BLOCK

Changed or New Equipment/System Information Requires copy to Ginna if any box is checked below.	Safety Class From GMEDB	Review by NS&L
<u>Requires Copy to Ginna. (Check applicable box)</u>	<u>See #1</u>	<u>(Y/N)</u>
<input type="checkbox"/> Setpoints (Instrument, Relief Valve, Time Delay, Other)	_____	<u>See (#2)</u>
<input type="checkbox"/> Operating Parameters (Flow, Pressure, Temperature, Volume, Other)	_____	<u>See (#2)</u>
<input type="checkbox"/> Operational Restrictions	_____	<u>See (#3)</u>
<input type="checkbox"/> UFSAR changes are required Section(s) _____	_____	<u>See (#4)</u>

NOTES:

- (#1) If any box is checked, consult the GMEDB records to determine the component safety class, then enter "SR" if Safety Related, or "SS" if Safety Significant or "NSR" if Non-Safety Related.  
 (#2) If Safety Class is "SR" or "SS" review by NS&L is required.  
 (#3) If box is checked, review by NS&L is required.  
 (#4) Responsible NES Engineer shall complete the UFSAR section.  
 If UFSAR changes are required, review by NS&L is required.

DOCUMENT CONTROL DATA FORM

PLANT SYSTEMS AND STRUCTURES LIST  
(Ref. 2.3; PSSSL Numeric Identifiers)

KEY WORDS:  
RCS Cold Leg

CROSS REFERENCED TO:  
N/A

SUPERSEDED REFERENCE DATA:  
N/A

EIN DESIGNATORS (S):  
RK-3

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2	0				
Attach. A	0				
Attach. B	0				

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DESIGN ANALYSIS1.0 Objective

Determine the uncertainty of RCS A&B cold leg temperature indication on RK-3 as a result of this modification. Then perform a comparison of the new uncertainty to the existing uncertainty of RCS A&B cold leg temperature currently indicated on RK-3 and show that the new uncertainty is less than or equal to that which currently exists.

2.0 Conclusion

The resultant loop uncertainty of the proposed modification is less than that of the existing configuration, therefore, the new configuration does not impact existing setpoints or operational requirements.

3.0 Design Inputs

- 3.1 Design Criteria, "RCS Cold Leg Temperature Recorder," EWR 10106, Rev. 0, October 14, 1993.
- 3.2 Safety Evaluation, SEV 1006, "RCS Cold Leg Temperature Recorder," EWR 10106, Rev. 0.

4.0 Referenced Documents

- 4.1 "Ginna Instrument Channel Uncertainty Report," RCS Temperature (Cold Leg Recorder), Channel TR-450A, Volian Enterprises, July 89.
- 4.2 RG&E Drawing 33013-2662
- 4.3 Calibration Procedure CP-I-RVLS-INTR-15.1, "Calibration of Reactor Vessel Level Monitoring System Train A Instrumentation," Rev. 3.
- 4.4 Westinghouse Specification IM 4D2B1-01EW for UR100 Micro Recorder
- 4.5 Foxboro Specification MI 2AO-130, "Voltage to Current Converter," dated May 1978
- 4.6 Foxboro Specification TI 2AI-180, August 1977.

JCS  
DA-EE-93-154  
EWR 10106

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5.0 Assumptions

None

6.0 Computer Codes

RCST409B (Refer to Attachment B)

7.0 Analysis

The basis uncertainty of the existing RCS A&B cold leg temperature indication on MCB chart recorder RK-3 is  $\pm 21.1$  °F (Ref. 4.1). The instrument uncertainty for the proposed modification as calculated in Attachment A to this analysis is  $\pm 20.25$  °F which is less than the existing design basis uncertainty.

8.0 Results

A comparison of the existing basis uncertainty to the uncertainty of the proposed modification shows that the proposed modification uncertainty is less than the that of the existing basis. Therefore, there is no impact on existing setpoints or operational requirements as a result of this modification.



## Proposed Modification Instrument Uncertainty Impact

## Description:

The proposed instrument modification is shown diagrammatically in Figure 1.

## Existing Instrument Basis Uncertainty:

$\pm 21.1$  °F

The basis for the existing cold leg recorder uncertainty is the "Ginna Instrument Channel Uncertainty Report", RCS Temperature (Cold Leg Recorder), Channel TR-450A, Volian Enterprises, July 89. The existing loop is not qualified for accident environments.

## Proposed Instrument Modification Uncertainty:

$\pm 2.893527$  % span

The instrument span used in the proposed modification is 700 °F. This yields a temperature uncertainty of,

$\pm 20.25469$  °F

Since the total loop uncertainty for the proposed modification is less than the basis uncertainty of the existing configuration there is no impact on existing setpoints, or operating requirements.





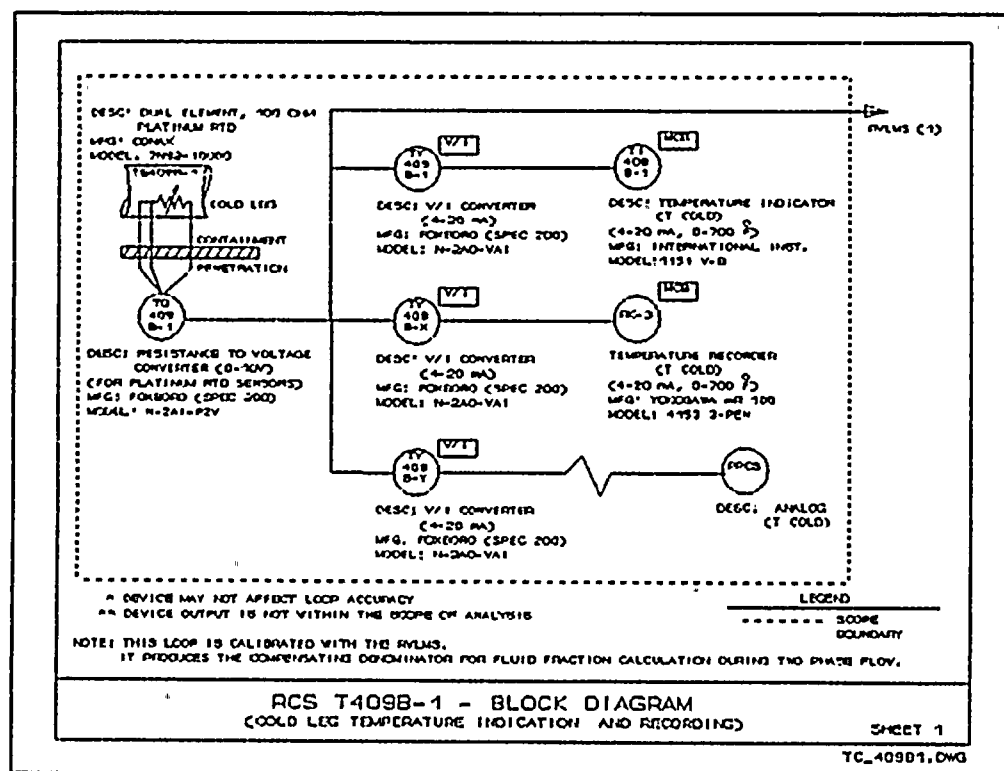


Figure 1



01209.939

LOOP T409B-1

	PMU	SU	ReU	DU	M&TEU	TU	AEU	CLU+AB	SU
TE409B-1	0.1	0	0	0	0	0	0	0	0
TQ409B-1	0	0.75	0	0	0.1414214	1	0	0	0
TY409B-1	0	0	0.6	0	0.2	1	0	0	0
RK_3	0	0	0.6	1	0.1414214	2	0	0	0
RESULTA	0.01	0.5625	0.72	1	0.08	8	0	0	0

TEMPERATURE(DEG F) 700 COUNTER N/A

RND ERR 2.8935273 BIAS 0

SUOU N/A +/- N/A

TLU 0 +/- 2.8935273



01209.940

LOOP T436B-1  
MODULE TE406B-1

MFG CONAX  
MODEL 7782-10000

PRESSURE: IVA

EFFECT	PROCESS			SENSOR (NORMAL)						
	Pma	Psa	Se	Sspe	Ste	Spsa	Sme	Sd	St	See
E1		0.1								
E2										
E3										
E4										
E5										
RESULTANT	0	0.1	0	0	0	0	0	0	0	0

EFFECT	PACK A AND PANEL MOUNTED					
	Rsa	Rta	Rpsa	Rme	Rtd	Rte
E1						
E2						
E3						
E4						
E5						
RESULTANT	0	0	0	0	0	0

EFFECT	ACCIDENT RELATED										SEISMIC
	Cra	Re	Te	Pe	S/Ce	AB	CI	SI	PI	TBI	CSI
E1	0						0				
E2											
E3											
E4											
E5											
RESULTANT	0	0	0	0	0	0	0	0	0	0	0

GROUP	PMU	SU	RaU	DU	M&TEU	TU	AEU	AB+CLU	SU
	0.1	0	0	0	0	0	0	0	0

MODULE ERROR
0 +/- 0.1



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11

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01209.941

 LOOP T402B-1  
 MODULE TQ402B-1

 MFG FOXBORO  
 MODEL N-2ALP2V

PRESSURE N/A

EFFECT	PROCESS		SENSOR (NORMAL)							
	Pma	Pea	Se	3tps	Ste	Spse	Sme	Sd	St	See
E1			0.75		0	0		0	1	0.1
E2			0							0.1
E3										
E4										
E5										
RESULTAN	0	0	0.75	0	0	0	0	0	1	0.1442136

EFFECT	PACK A AND PANEL MOUNTED					
	Pra	Pra	Pra	Pra	Pra	Pra
E1						
E2						
E3						
E4						
E5						
RESULTAN	0	0	0	0	0	0

EFFECT	ACCIDENT RELATED											SEISMIC
	Cra	Pra	Ta	Pa	S/Ca	AB	CI	SI	PI	TBI	CSI	Se
E1												0
E2												
E3												
E4												
E5												
RESULTAN	0	0	0	0	0	0	0	0	0	0	0	0

GROUP	PMU	SU	PraU	OU	MATEU	TU	AEU	AB+CLU	SU
	0	0.75	0	0	0.1442136	1	0	0	0

MODULE ERROR		
0	+	1.25797456





01209.942

LOOP T409B-1  
MODULE TY40WB-1MFG FOXBORO  
MODEL N-2AO-VAJ

PRESSURE N/A

EFFECT	PROCESS		SENSOR (NORMAL)							
	Pma	Pra	Pa	Spa	Sta	Spa	Sma	Sd	St	Sce
E1										
E2										
E3										
E4										
E5										
RESULTANT	0	0	0	0	0	0	0	0	0	0

EFFECT	PACK A AND PANEL MOUNTED						
	Pra	Pra	Rpa	Pma	Pra	Pra	Pra
E1	0.6				0	1	0.2
E2							
E3							
E4							
E5							
RESULTANT	0.6	0	0	0	0	1	0.2

EFFECT	ACCIDENT RELATED											SEISMIC
	Cra	Ra	Te	Pa	S/Ca	AB	Cl	Sl	Pi	TDi	CSI	Se
E1												0
E2												
E3												
E4												
E5												
RESULTANT	0	0	0	0	0	0	0	0	0	0	0	0

GROUP	PMU	SU	ReU	DU	M&TEU	TU	AEU	AB+CLU	SU
	0	0	0.6	0	0.2	1	0	0	0

MODULE ERROR	
0	+/- 1.18321596



01209.943

LOOP T407B-1  
MODULE PK-3MFG YOKOGAWA  
MODEL mR 100 4153 J PEN

PRESSURE N/A

EFFECT	PROCESS		SENSOR (NORMAL)							
	Pma	Pea	Se	Spa	Ste	Spsa	Sma	Sd	St	Sce
E1										
E2										
E3										
E4										
E5										
RESULTANT	0	0	0	0	0	0	0	0	0	0

EFFECT	PACK A AND PANEL MOUNTED						
	Pma	Pea	Pspa	Pma	Psa	Pst	Pce
E1	0.6			0	1	2	0.1
E2							0.1
E3							
E4							
E5							
RESULTANT	0.6	0	0	0	1	2	0.14142136

EFFECT	ACCIDENT RELATED											SEISMIC
	Cra	Re	Te	Pe	S/Ce	AB	CI	SI	PI	TEI	CSI	Se
E1												0
E2												
E3												
E4												
E5												
RESULTANT	0	0	0	0	0	0	0	0	0	0	0	0

GROUP	PMU	SU	RU	DU	MATEU	TU	AEU	AB+CLU	SU
	0	0	0.6	1	0.14142136	2	0	0	0

MODULE ERROR		
0	±	2.3194927



01209.944

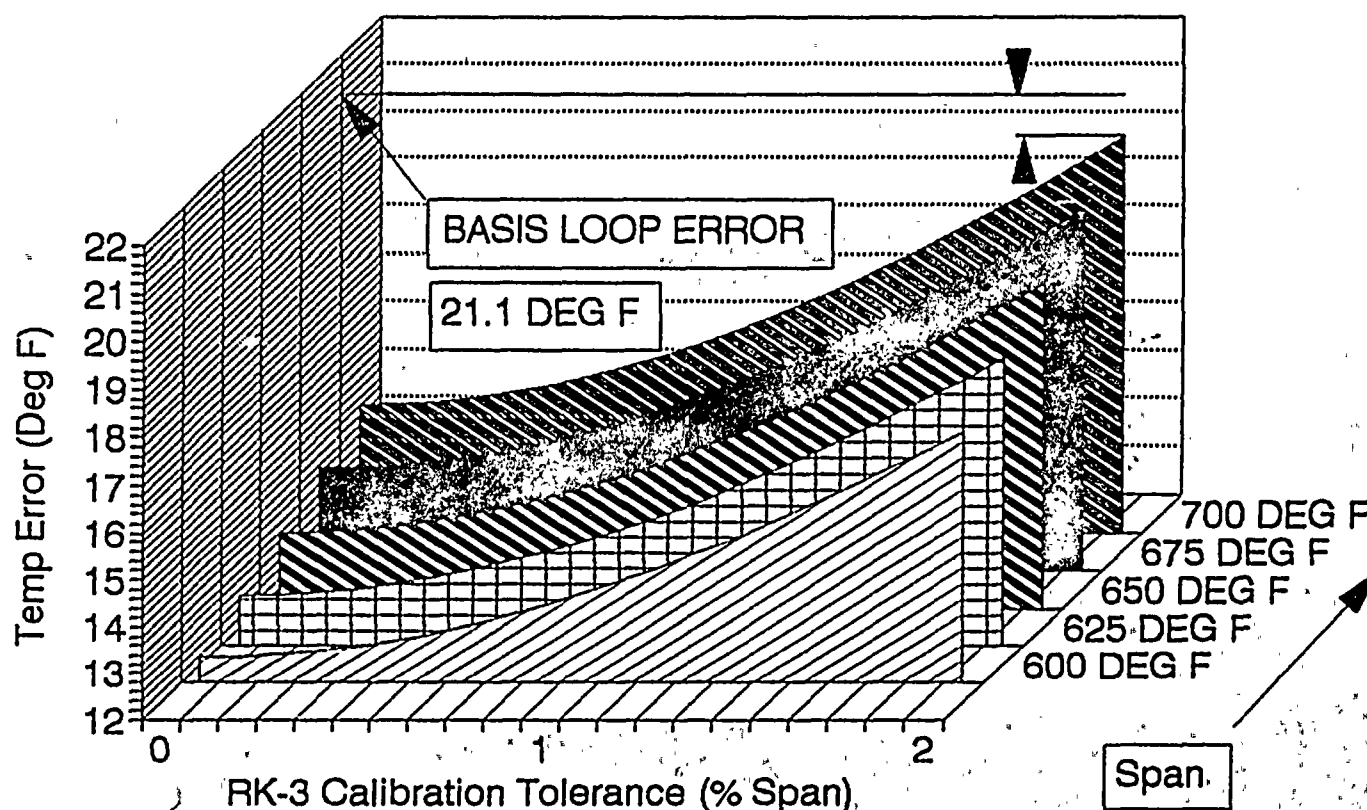
SPAN(?F)	ERROR (?F)				
	600	625	650	675	700
12.54631	13.06908	13.59184	14.1146	14.63737	
12.56065	13.08401	13.60737	14.13073	14.65409	
12.60357	13.12872	13.65387	14.17902	14.70417	
12.67478	13.20289	13.73101	14.25913	14.78724	
12.7738	13.30604	13.83828	14.37053	14.90277	
12.9	13.4375	13.975	14.5125	15.05	
13.05259	13.59644	14.1403	14.68416	15.22802	
13.23065	13.78192	14.3332	14.88448	15.43575	
13.43317	13.99288	14.5526	15.11231	15.67203	
13.65906	14.22819	14.79732	15.36645	15.93557	
13.90719	14.48666	15.06613	15.64559	16.22506	
14.17639	14.76707	15.35775	15.94844	16.53912	
14.46548	15.0682	15.67093	16.27366	16.87639	
14.77329	15.38884	16.00439	16.61995	17.2355	
15.09868	15.72779	16.3569	16.98601	17.61512	
15.44053	16.08389	16.72724	17.3706	18.01395	
15.79778	16.45603	17.11427	17.77251	18.43075	
16.16942	16.84314	17.51687	18.19059	18.86432	
16.55446	17.24422	17.93399	18.62376	19.31353	
16.95199	17.65832	18.36466	19.07099	19.77732	
17.36116	18.08455	18.80793	19.53131	20.25469	

RK-3 CAL TOL (% SPAN)
0
0.1
0.2
0.3
0.4
0.5
0.6
0.7
0.8
0.9
1
1.1
1.2
1.3
1.4
1.5
1.6
1.7
1.8
1.9
2



1. 1925, 1926, 1927, 1928

## Temperature Error (Deg F) vs RK-3 Calibration Tolerance (% Span)





## Computer Software Documentation

The computer programs utilized in this analysis are classified as type 4 in accordance with Appendix B of QE 330, Rev. 0.

## Type 4: Computer Software Package Documentation Summary

1. Title of Report or Analysis RCS-T409B-1, DA-EE-93-154
2. Author George Daniels, et al
3. Verification of Program

(a) (1) Name of Program: RCST409B

(2) Description (include source code location)

The principal program consists of linked spreadsheet files written using QUATTRO PRO 3.0 (Borland International, Inc.) Cell algorithms and code are attached.

(3) Algorithm Bases

☐ Found in text, page N/A

☐ Reference N/A

(4) Numerical Methods Used

(i) Description N/A

(ii) Reference N/A

(iii) Other N/A

### Instrument Uncertainty Data Processing System

In order to accommodate instrument uncertainties with significant variability related to process state parameters, a computer program using an advanced spreadsheet environment (QUATTRO PRO) is utilized. A typical file structure is shown in Figure 1 below.

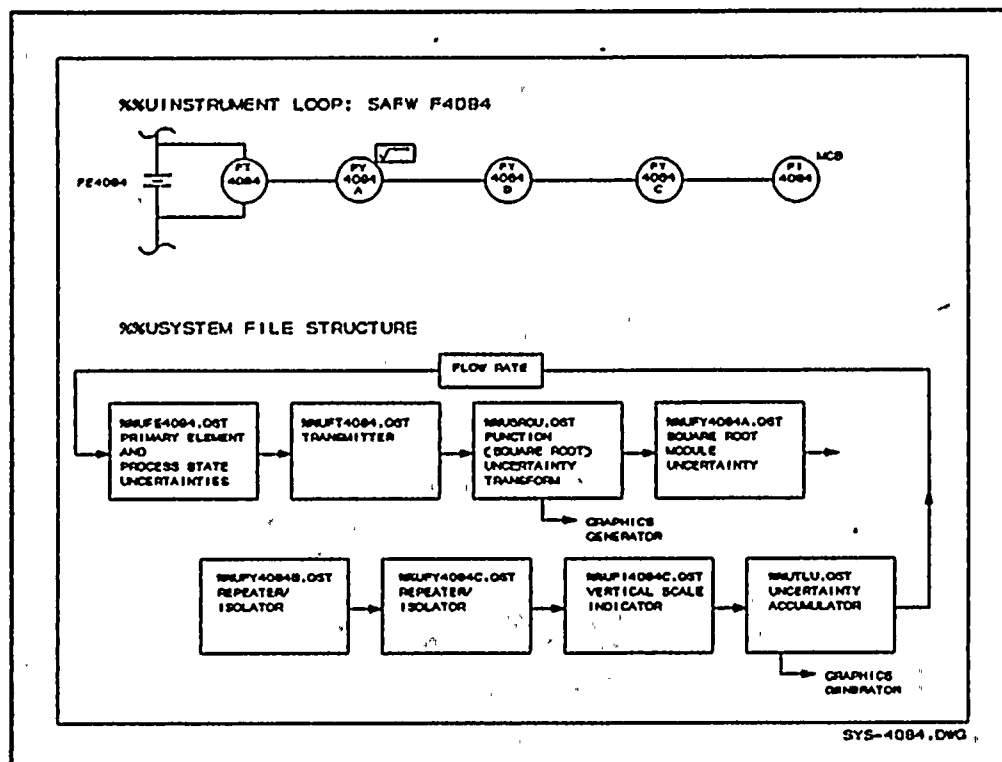


Figure 1

A complete listing of files used for this analysis is provided in the following pages.



14-12-88



14-12-88



### Graphics Generation

The graphics documented in the Analysis text are intended to provide sufficient insight into complex parametric uncertainty effects to verify conformity with performance requirements. However the software system provides the capability for generating additional graphics, either to check existing calculations, or for providing additional data related to instrument performance during normal or emergency conditions.

Each uncertainty accumulator file uses cells C30-I30 as the graphics variable block. The macro "GRAPHMAC" is used to generate text files for the selected parameter set. GRAPHMAC is located in the LIB.QST file. It must be configured for the limits, range, and desired granularity of the parameters being graphed. Text files produced by GRAPHMAC are then imported to a QUATTRO PRO spreadsheet for use by the graphics utility. An example of this capability is shown in Figure 2 below.

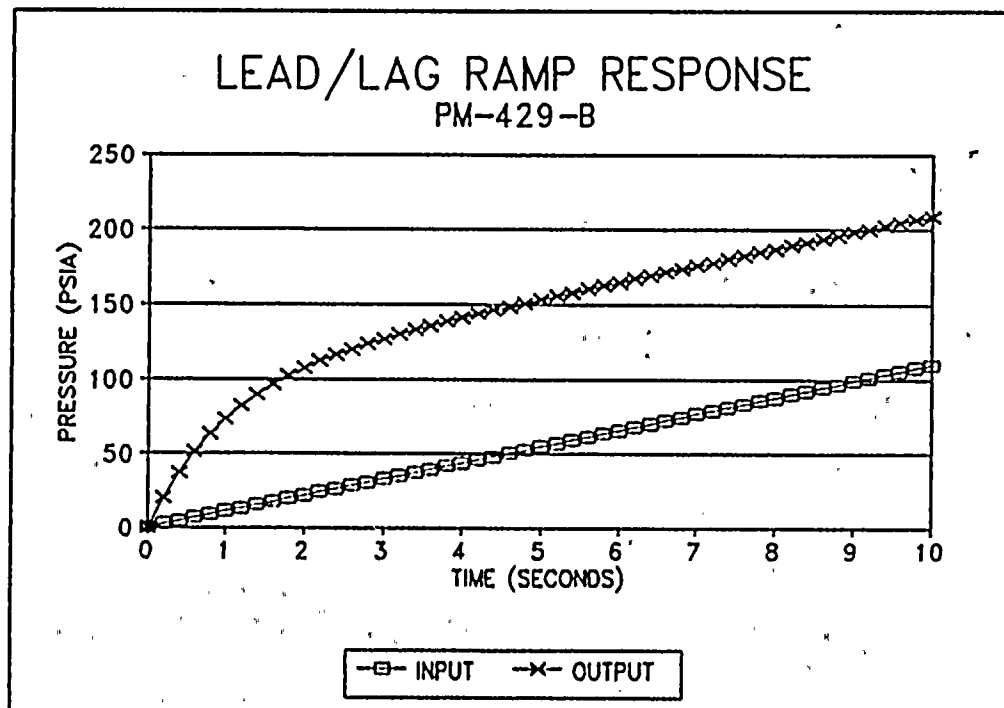


Figure 2

01209.949

COLD LEG TEMPERATURE - INDICATOR LOOP



## LIB.QST

```
B3: 20
F3: 1
J3: 5
N3: 1
R3: 20
B4: 980
F4: 100
J4: 200
N4: 100
R4: 1000
B5: 20
F5: 1
J5: 5
N5: 1
R5: 20
B8: '{FOR E10,B3,B4,B5,B10}
F8: '{FOR E10,F3,F4,F5,F10}
J8: '{FOR E10,J3,J4,J5,J10}
N8: '{FOR E10,N3,N4,N5,N10}
R8: '{FOR E10,R3,R4,R5,R10}
B10: '{/ Print;Block}C20..E20~
E10: 1020
F10: '{/ Print;Block}C22..E22~
J10: '{/ Print;Block}C24..E24~
N10: '{/ Print;Block}C26..E26~
R10: '{/ Print;Block}C30..G30~
B11: '{/ Print;Go}
F11: '{/ Print;Go}
J11: '{/ Print;Go}
N11: '{/ Print;Go}
R11: '{/ Print;Go}
```





## TE409B\_1.QST

A1: 'LOOP  
B1: 'T409B-1  
D1: 'MFG  
E1: 'CONAX  
A2: 'MODULE  
B2: 'TE409B-1  
D2: 'MODEL  
E2: '7N92-10000  
A4: 'PRESSURE (PERCENT)  
B4: 'N/A  
C5: 'PROCESS  
I5: 'SENSOR (NORMAL)  
A6: '  
B6: 'EFFECT  
C6: 'Pma  
D6: 'Pea  
E6: 'Sa  
F6: 'Sspe  
G6: 'Ste  
H6: 'Spse  
I6: 'Sme  
J6: 'Sd  
K6: 'St  
L6: 'Sce  
A7: '  
B7: 'E1  
C7: '  
D7: 0.1  
E7: '  
G7: '  
H7: '  
J7: '  
K7: '  
L7: '  
A8: '  
B8: 'E2  
C8: '  
E8: '  
A9: '  
B9: 'E3  
A10: '  
B10: 'E4  
A11: '  
B11: 'E5  
A12: '  
B12: 'RESULTANT  
C12: @SQRT(C7^2+C8^2+C9^2+C10^2+C11^2)  
D12: @SQRT(D7^2+D8^2+D9^2+D10^2+D11^2)



E12: @SQRT(E7^2+E8^2+E9^2+E10^2+E11^2)  
 F12: @SQRT(F7^2+F8^2+F9^2+F10^2+F11^2)  
 G12: @SQRT(G7^2+G8^2+G9^2+G10^2+G11^2)  
 H12: @SQRT(H7^2+H8^2+H9^2+H10^2+H11^2)  
 I12: @SQRT(I7^2+I8^2+I9^2+I10^2+I11^2)  
 J12: @SQRT(J7^2+J8^2+J9^2+J10^2+J11^2)  
 K12: @SQRT(K7^2+K8^2+K9^2+K10^2+K11^2)  
 L12: @SQRT(L7^2+L8^2+L9^2+L10^2+L11^2)

D14: ' RACK AND PANEL  
 E14: 'AND PANEL  
 F14: ' MOUNTED  
 B15: 'EFFECT  
 C15: ' Rea  
 D15: ' Rte  
 E15: ' Rpse  
 F15: ' Rme  
 G15: ' Red  
 H15: ' Ret  
 I15: ' Rce  
 B16: 'E1  
 B17: 'E2  
 B18: 'E3  
 B19: 'E4  
 B20: 'E5  
 B21: 'RESULTANT  
 C21: @SQRT(C16^2+C17^2+C18^2+C19^2+C20^2)  
 D21: @SQRT(D16^2+D17^2+D18^2+D19^2+D20^2)  
 E21: @SQRT(E16^2+E17^2+E18^2+E19^2+E20^2)  
 F21: @SQRT(F16^2+F17^2+F18^2+F19^2+F20^2)  
 G21: @SQRT(G16^2+G17^2+G18^2+G19^2+G20^2)  
 H21: @SQRT(H16^2+H17^2+H18^2+H19^2+H20^2)  
 I21: @SQRT(I16^2+I17^2+I18^2+I19^2+I20^2)  
 F23: 'ACCIDENT RELATED  
 N23: 'SEISMIC  
 B24: 'EFFECT  
 C24: ' Crae  
 D24: ' Re  
 E24: ' Te  
 F24: ' Pe  
 G24: ' S/Ce  
 H24: ' AB



```

I24: ' C1
J24: ' S1
K24: ' P1
L24: ' TB1
M24: ' CS1
N24: ' Se
B25: 'E1
C25: 0
I25: 0
B26: 'E2
B27: 'E3
B28: 'E4
B29: 'E5
B30: 'RESULTANT
C30: @SQRT(C25^2+C26^2+C27^2+C28^2+C29^2)
D30: @SQRT(D25^2+D26^2+D27^2+D28^2+D29^2)
E30: @SQRT(E25^2+E26^2+E27^2+E28^2+E29^2)
F30: @SQRT(F25^2+F26^2+F27^2+F28^2+F29^2)
G30: @SQRT(G25^2+G26^2+G27^2+G28^2+G29^2)
H30: @SUM(H25..H29)
I30: @SUM(I25..I29)
J30: @SUM(J25..J29)
K30: @SUM(K25..K29)

```

```

L30: @SUM(L25..L29)
M30: @SUM(M25..M29)
N30: @SQRT(N25^2+N26^2+N27^2+N28^2+N29^2)
B34: 'GROUP
C34: ' PMU
D34: ' SU
E34: ' ReU
F34: ' DU
G34: ' M&TEU
H34: ' TU
I34: ' AEU
J34: ' AB+CLU
K34: ' SU
C35: @SQRT(C12^2+D12^2)
D35: @SQRT(E12^2+F12^2+G12^2+H12^2+I12^2)
E35: @SQRT(C21^2+D21^2+E21^2+F21^2)
F35: @SQRT(J12^2+G21^2)
G35: @SQRT(L12^2+I21^2)

```

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.H35: @SQRT(K12^2+H21^2)  
I35: @SQRT(C30^2+D30^2+E30^2+F30^2+G30^2)  
J35: @SUM(H30..M30)  
K35: @SQRT(N30^2)



TI409B\_1.QST

A1: 'LOOP  
 B1: 'T409B-1  
 D1: 'MFG  
 E1: 'INTL INST  
 A2: 'MODULE  
 B2: 'TI409B-1  
 D2: 'MODEL  
 E2: '1151 V-B  
 A4: 'PRESSURE (PERCENT)  
 B4: 'N/A  
 C5: ' PROCESS  
 I5: 'SENSOR (NORMAL)  
 A6: '  
 B6: 'EFFECT  
 C6: ' Pma  
 D6: ' Pea  
 E6: ' Sa  
 F6: ' Sspe  
 G6: ' Ste  
 H6: ' Spse  
 I6: ' Sme  
 J6: ' Sd  
 K6: ' St  
 L6: ' Sce  
 A7: '  
 B7: 'E1  
 C7: '  
 E7: '  
 G7: '  
 H7: '  
 J7: '  
 K7: '  
 L7: '  
 A8: '  
 B8: 'E2  
 C8: '  
 E8: '  
 A9: '  
 B9: 'E3  
 A10: '  
 B10: 'E4  
 A11: '  
 B11: 'E5  
 A12: '  
 B12: 'RESULTANT  
 C12: @SQRT(C7^2+C8^2+C9^2+C10^2+C11^2)  
 D12: @SQRT(D7^2+D8^2+D9^2+D10^2+D11^2)  
 E12: @SQRT(E7^2+E8^2+E9^2+E10^2+E11^2)





F12: @SQRT(F7^2+F8^2+F9^2+F10^2+F11^2)  
 G12: @SQRT(G7^2+G8^2+G9^2+G10^2+G11^2)  
 H12: @SQRT(H7^2+H8^2+H9^2+H10^2+H11^2)  
 I12: @SQRT(I7^2+I8^2+I9^2+I10^2+I11^2)  
 J12: @SQRT(J7^2+J8^2+J9^2+J10^2+J11^2)  
 K12: @SQRT(K7^2+K8^2+K9^2+K10^2+K11^2)  
 L12: @SQRT(L7^2+L8^2+L9^2+L10^2+L11^2)  
 D14: ' RACK AND PANEL

E14: 'AND PANEL  
 F14: ' MOUNTED  
 B15: 'EFFECT  
 C15: ' Rea  
 D15: ' Rte  
 E15: ' Rpse  
 F15: ' Rme  
 G15: ' Red  
 H15: ' Ret  
 I15: ' Rce  
 B16: 'E1  
 C16: 1.5  
 F16: 0.7  
 G16: 1  
 H16: 1  
 I16: 0.1  
 B17: 'E2  
 B18: 'E3  
 B19: 'E4  
 B20: 'E5  
 B21: 'RESULTANT  
 C21: @SQRT(C16^2+C17^2+C18^2+C19^2+C20^2)  
 D21: @SQRT(D16^2+D17^2+D18^2+D19^2+D20^2)  
 E21: @SQRT(E16^2+E17^2+E18^2+E19^2+E20^2)  
 F21: @SQRT(F16^2+F17^2+F18^2+F19^2+F20^2)  
 G21: @SQRT(G16^2+G17^2+G18^2+G19^2+G20^2)  
 H21: @SQRT(H16^2+H17^2+H18^2+H19^2+H20^2)  
 I21: @SQRT(I16^2+I17^2+I18^2+I19^2+I20^2)  
 F23: 'ACCIDENT RELATED  
 N23: 'SEISMIC  
 B24: 'EFFECT  
 C24: ' Crae  
 D24: ' Re



E24: ' Te  
 F24: ' Pe  
 G24: ' S/Ce  
 H24: ' AB  
 I24: ' Cl  
 J24: ' Sl  
 K24: ' Pl  
 L24: ' TB1  
 M24: ' CS1  
 N24: ' Se  
 B25: 'E1  
 N25: 0  
 B26: 'E2  
 B27: 'E3  
 B28: 'E4  
 B29: 'E5  
 B30: 'RESULTANT  
 C30: @SQRT(C25^2+C26^2+C27^2+C28^2+C29^2)  
 D30: @SQRT(D25^2+D26^2+D27^2+D28^2+D29^2)  
 E30: @SQRT(E25^2+E26^2+E27^2+E28^2+E29^2)  
 F30: @SQRT(F25^2+F26^2+F27^2+F28^2+F29^2)  
 G30: @SQRT(G25^2+G26^2+G27^2+G28^2+G29^2)  
 H30: @SUM(H25..H29)

I30: @SUM(I25..I29)  
 J30: @SUM(J25..J29)  
 K30: @SUM(K25..K29)  
 L30: @SUM(L25..L29)  
 M30: @SUM(M25..M29)  
 N30: @SQRT(N25^2+N26^2+N27^2+N28^2+N29^2)  
 B34: 'GROUP  
 C34: ' PMU  
 D34: ' SU  
 E34: ' ReU  
 F34: ' DU  
 G34: ' M&TEU  
 H34: ' TU  
 I34: ' AEU  
 J34: ' AB+CLU  
 K34: ' SU  
 C35: @SQRT(C12^2+D12^2)  
 D35: @SQRT(E12^2+F12^2+G12^2+H12^2+I12^2)

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E35: @SQRT(C21^2+D21^2+E21^2+F21^2)  
F35: @SQRT(J12^2+G21^2)  
G35: @SQRT(L12^2+I21^2)  
H35: @SQRT(K12^2+H21^2)  
I35: @SQRT(C30^2+D30^2+E30^2+F30^2+G30^2)  
J35: @SUM(H30..M30)  
K35: @SQRT(N30^2)

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## TLU.QST

A1: 'LOOP  
B1: 'T409B-1  
A3: '  
C3: 'PMU  
D3: 'SU  
E3: 'ReU  
F3: 'DU  
G3: 'M&TEU  
H3: 'TU  
I3: 'AEU  
J3: 'CLU+AB  
K3: 'SU  
B4: 'TE409B-1  
C4: +[TE409B\_1]C35  
D4: +[TE409B\_1]D35  
E4: +[TE409B\_1]E35  
F4: +[TE409B\_1]F35  
G4: +[TE409B\_1]G35  
H4: +[TE409B\_1]H35  
I4: +[TE409B\_1]I35  
J4: +[TE409B\_1]J35  
K4: +[TE409B\_1]K35  
B5: 'TQ409B-1  
C5: +[TQ409B\_1]C35  
D5: +[TQ409B\_1]D35  
E5: +[TQ409B\_1]E35  
F5: +[TQ409B\_1]F35  
G5: +[TQ409B\_1]G35  
H5: +[TQ409B\_1]H35  
I5: +[TQ409B\_1]I35  
J5: +[TQ409B\_1]J35  
K5: +[TQ409B\_1]K35  
B6: 'TY409B-1  
C6: +[TY409B\_1]C35  
D6: +[TY409B\_1]D35  
E6: +[TY409B\_1]E35  
F6: +[TY409B\_1]F35  
G6: +[TY409B\_1]G35  
H6: +[TY409B\_1]H35  
I6: +[TY409B\_1]I35  
J6: +[TY409B\_1]J35  
K6: +[TY409B\_1]K35  
B7: 'TI409B-1  
C7: +[TI409B\_1]C35  
D7: +[TI409B\_1]D35  
E7: +[TI409B\_1]E35  
F7: +[TI409B\_1]F35  
G7: +[TI409B\_1]G35

H7: +[TI409B\_1]H35  
 I7: +[TI409B\_1]I35  
 J7: +[TI409B\_1]J35  
 K7: +[TI409B\_1]K35  
 A8: '  
 B8: 'RESULTANT  
 C8: @IF(ROC=1,0,+C4^2+C6^2+C7^2+C5^2)  
 D8: +D4^2+D6^2+D7^2+D5^2

E8: +E4^2+E6^2+E7^2+E5^2  
 F8: @IF(ROC=1,0,+F4^2+F6^2+F7^2+F5^2)  
 G8: @IF(ROC=1,0,+G4^2+G6^2+G7^2+G5^2)  
 H8: @IF(ROC=1,0,+H4^2+H6^2+H7^2+H5^2)  
 I8: +I4^2+I6^2+I7^2+I5^2  
 J8: +J4+J6+J7+J5  
 K8: +K4^2+K6^2+K7^2+K5^2  
 E9: '  
 A10: 'PRESSURE (PERCENT)  
 C10: ' N/A  
 D10: ' COUNTER  
 E10: ' N/A  
 G10: 'RATE OF CHANGE  
 I10: 1  
 O10: '  
 N11: 1  
 A12: '  
 B12: 'RND ERR  
 C12: @SQRT(@SUM(C8..I8)+K8)  
 D12: ' BIAS  
 E12: +J8  
 N12: 100  
 K13: '  
 N13: 1  
 A14: '  
 B14: 'SUOU  
 C14: ' N/A  
 D14: ' +/-  
 E14: ' N/A  
 B15: '  
 G15: +BIAS+RANDOM  
 A16: '  
 B16: 'TLU





```

C16: +E12
D16: ' +/-
E16: 3SQRT(C12^2)
G17: +BIAS-RANDOM
B18: 'SYSTEM STATE SPECIFICATION BLOCK
N18: '
B19: 'NORM OPS
D19: 1
E19: '
F19: +NORM#AND#(#NOT#ACCHI)#AND#(#NOT#ACCLO)
N19: '{FOR E10,N11,N12,N13,N21}
D20: '
E20: '
L20: '
N20: '
B21: 'ACC HI
C21: '
D21: 0
E21: '
F21: +ACCHI#AND#(#NOT#NORM)#AND#(#NOT#ACCLO)
K21: '
N21: '{/ Print;Block}E14..E14~
A22: '

```

```

D22: '
E22: '
N22: '{/ Print;Go}
B23: 'ACC LO
D23: 0
F23: +ACCLO#AND#(#NOT#NORM)#AND#(#NOT#ACCHI)
      B      2      4      :
@IF((#NOT#ACCHI)#AND#(#NOT#ACCLO)#AND#(#NOT#NORM)=1,"SYSTEM STATE
SPECIFICATION ERROR"," ")
      B      2      5      :
@IF((NORM#AND#ACCHI)#OR#(ACCHI#AND#ACCLO)#OR#(ACCLO#AND#NORM),"SY
STEM STATE SPECIFICATION ERROR",B24)

```



## TQ409B\_1.QST

A1: 'LOOP  
B1: 'T409B-1  
D1: 'MFG  
E1: 'FOXBORO  
A2: 'MODULE  
B2: 'TQ409B-1  
D2: 'MODEL  
E2: 'N-2AI-P2V  
A4: 'PRESSURE (PERCENT)  
B4: 'N/A  
C5: 'PROCESS  
I5: 'SENSOR (NORMAL)  
A6: '  
B6: 'EFFECT  
C6: 'Pma  
D6: 'Pea  
E6: 'Sa  
F6: 'Sspe  
G6: 'Ste  
H6: 'Spse  
I6: 'Sme  
J6: 'Sd  
K6: 'St  
L6: 'Sce  
A7: '  
B7: 'E1  
C7: '  
E7: 0.75  
G7: 0  
H7: 0  
J7: 0  
K7: 1  
L7: 0.1  
A8: '  
B8: 'E2  
C8: '  
E8: 0  
L8: 0.1  
A9: '  
B9: 'E3  
A10: '  
B10: 'E4  
A11: '  
B11: 'E5  
A12: '  
B12: 'RESULTANT  
C12: @SQRT(C7^2+C8^2+C9^2+C10^2+C11^2)  
D12: @SQRT(D7^2+D8^2+D9^2+D10^2+D11^2)

E12: @SQRT(E7^2+E8^2+E9^2+E10^2+E11^2)  
 F12: @SQRT(F7^2+F8^2+F9^2+F10^2+F11^2)  
 G12: @SQRT(G7^2+G8^2+G9^2+G10^2+G11^2)  
 H12: @SQRT(H7^2+H8^2+H9^2+H10^2+H11^2)  
 I12: @SQRT(I7^2+I8^2+I9^2+I10^2+I11^2)  
 J12: @SQRT(J7^2+J8^2+J9^2+J10^2+J11^2)  
 K12: @SQRT(K7^2+K8^2+K9^2+K10^2+K11^2)  
 L12: @SQRT(L7^2+L8^2+L9^2+L10^2+L11^2)

D14: ' RACK AND PANEL  
 E14: 'AND PANEL  
 F14: ' MOUNTED  
 B15: 'EFFECT  
 C15: ' Rea  
 D15: ' Rte  
 E15: ' Rpse  
 F15: ' Rme  
 G15: ' Red  
 H15: ' Ret  
 I15: ' Rce  
 B16: 'E1  
 B17: 'E2  
 B18: 'E3  
 B19: 'E4  
 B20: 'E5  
 B21: 'RESULTANT  
 C21: @SQRT(C16^2+C17^2+C18^2+C19^2+C20^2)  
 D21: @SQRT(D16^2+D17^2+D18^2+D19^2+D20^2)  
 E21: @SQRT(E16^2+E17^2+E18^2+E19^2+E20^2)  
 F21: @SQRT(F16^2+F17^2+F18^2+F19^2+F20^2)  
 G21: @SQRT(G16^2+G17^2+G18^2+G19^2+G20^2)  
 H21: @SQRT(H16^2+H17^2+H18^2+H19^2+H20^2)  
 I21: @SQRT(I16^2+I17^2+I18^2+I19^2+I20^2)  
 F23: 'ACCIDENT RELATED  
 N23: 'SEISMIC  
 B24: 'EFFECT  
 C24: ' Crae  
 D24: ' Re  
 E24: ' Te  
 F24: ' Pe  
 G24: ' S/Ce  
 H24: ' AB

```

I24: ' C1
J24: ' S1
K24: ' P1
L24: ' TB1
M24: ' CS1
N24: ' Se
B25: ' E1
C25: @IF([TLU]NORM=0,0.56,0)
I25: @IF([TLU]ACCLO=1,-0.43,0)
N25: 0
B26: ' E2
B27: ' E3
B28: ' E4
B29: ' E5
B30: ' RESULTANT
C30: @SQRT(C25^2+C26^2+C27^2+C28^2+C29^2)
D30: @SQRT(D25^2+D26^2+D27^2+D28^2+D29^2)
E30: @SQRT(E25^2+E26^2+E27^2+E28^2+E29^2)
F30: @SQRT(F25^2+F26^2+F27^2+F28^2+F29^2)
G30: @SQRT(G25^2+G26^2+G27^2+G28^2+G29^2)
H30: @SUM(H25..H29)
I30: @SUM(I25..I29)
J30: @SUM(J25..J29)

```

```

K30: @SUM(K25..K29)
L30: @SUM(L25..L29)
M30: @SUM(M25..M29)
N30: @SQRT(N25^2+N26^2+N27^2+N28^2+N29^2)
B34: ' GROUP
C34: ' PMU
D34: ' SU
E34: ' ReU
F34: ' DU
G34: ' M&TEU
H34: ' TU
I34: ' AEU
J34: ' AB+CLU
K34: ' SU
C35: @SQRT(C12^2+D12^2)
D35: @SQRT(E12^2+F12^2+G12^2+H12^2+I12^2)
E35: @SQRT(C21^2+D21^2+E21^2+F21^2)
F35: @SQRT(J12^2+G21^2)

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G35: @SQRT(L12^2+I21^2)  
H35: @SQRT(K12^2+H21^2)  
I35: @SQRT(C30^2+D30^2+E30^2+F30^2+G30^2)  
J35: @SUM(H30..M30)  
K35: @SQRT(N30^2)





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TY409B\_1.QST

A1: 'LOOP  
B1: 'T409B-1  
D1: 'MFG  
E1: 'FOXBORO  
A2: 'MODULE  
B2: 'TY409B-1  
D2: 'MODEL  
E2: 'N-2AO-VAI  
A4: 'PRESSURE (PERCENT)  
B4: 'N/A  
C5: ' PROCESS  
I5: 'SENSOR (NORMAL)  
A6: '  
B6: 'EFFECT  
C6: ' Pma  
D6: ' Pea  
E6: ' Sa  
F6: ' Sspe  
G6: ' Ste  
H6: ' Spse  
I6: ' Sme  
J6: ' Sd  
K6: ' St  
L6: ' Sce  
A7: '  
B7: 'E1  
C7: '  
E7: '  
G7: '  
H7: '  
J7: '  
K7: '  
L7: '  
A8: '  
B8: 'E2  
C8: '  
E8: '  
A9: '  
B9: 'E3  
A10: '  
B10: 'E4  
A11: '  
B11: 'E5  
A12: '  
B12: 'RESULTANT  
C12: @SQRT(C7^2+C8^2+C9^2+C10^2+C11^2)  
D12: @SQRT(D7^2+D8^2+D9^2+D10^2+D11^2)  
E12: @SQRT(E7^2+E8^2+E9^2+E10^2+E11^2)

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F12: @SQRT(F7^2+F8^2+F9^2+F10^2+F11^2)  
 G12: @SQRT(G7^2+G8^2+G9^2+G10^2+G11^2)  
 H12: @SQRT(H7^2+H8^2+H9^2+H10^2+H11^2)  
 I12: @SQRT(I7^2+I8^2+I9^2+I10^2+I11^2)  
 J12: @SQRT(J7^2+J8^2+J9^2+J10^2+J11^2)  
 K12: @SQRT(K7^2+K8^2+K9^2+K10^2+K11^2)  
 L12: @SQRT(L7^2+L8^2+L9^2+L10^2+L11^2)  
 D14: ' RACK AND PANEL

E14: 'AND PANEL  
 F14: ' MOUNTED  
 B15: 'EFFECT  
 C15: ' Rea  
 D15: ' Rte  
 E15: ' Rpse  
 F15: ' Rme  
 G15: ' Red  
 H15: ' Ret  
 I15: ' Rce  
 B16: 'E1  
 C16: 0.6  
 G16: 0  
 H16: 1  
 I16: 0.2  
 B17: 'E2  
 B18: 'E3  
 B19: 'E4  
 B20: 'E5  
 B21: 'RESULTANT  
 C21: @SQRT(C16^2+C17^2+C18^2+C19^2+C20^2)  
 D21: @SQRT(D16^2+D17^2+D18^2+D19^2+D20^2)  
 E21: @SQRT(E16^2+E17^2+E18^2+E19^2+E20^2)  
 F21: @SQRT(F16^2+F17^2+F18^2+F19^2+F20^2)  
 G21: @SQRT(G16^2+G17^2+G18^2+G19^2+G20^2)  
 H21: @SQRT(H16^2+H17^2+H18^2+H19^2+H20^2)  
 I21: @SQRT(I16^2+I17^2+I18^2+I19^2+I20^2)  
 F23: 'ACCIDENT RELATED  
 N23: 'SEISMIC  
 B24: 'EFFECT  
 C24: ' Crae  
 D24: ' Re  
 E24: ' Te

F24: ' Pe  
 G24: ' S/Ce  
 H24: ' AB  
 I24: ' Cl  
 J24: ' Sl  
 K24: ' Pl  
 L24: ' TB1  
 M24: ' CS1  
 N24: ' Se  
 B25: ' E1  
 N25: 0  
 B26: ' E2  
 B27: ' E3  
 B28: ' E4  
 B29: ' E5  
 B30: ' RESULTANT  
 C30: @SQRT(C25^2+C26^2+C27^2+C28^2+C29^2)  
 D30: @SQRT(D25^2+D26^2+D27^2+D28^2+D29^2)  
 E30: @SQRT(E25^2+E26^2+E27^2+E28^2+E29^2)  
 F30: @SQRT(F25^2+F26^2+F27^2+F28^2+F29^2)  
 G30: @SQRT(G25^2+G26^2+G27^2+G28^2+G29^2)  
 H30: @SUM(H25..H29)  
 I30: @SUM(I25..I29)

J30: @SUM(J25..J29)  
 K30: @SUM(K25..K29)  
 L30: @SUM(L25..L29)  
 M30: @SUM(M25..M29)  
 N30: @SQRT(N25^2+N26^2+N27^2+N28^2+N29^2)  
 B34: ' GROUP  
 C34: ' PMU  
 D34: ' SU  
 E34: ' ReU  
 F34: ' DU  
 G34: ' M&TEU  
 H34: ' TU  
 I34: ' AEU  
 J34: ' AB+CLU  
 K34: ' SU  
 C35: @SQRT(C12^2+D12^2)  
 D35: @SQRT(E12^2+F12^2+G12^2+H12^2+I12^2)  
 E35: @SQRT(C21^2+D21^2+E21^2+F21^2)

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F35: @SQRT(J12^2+G21^2)  
G35: @SQRT(L12^2+I21^2)  
H35: @SQRT(K12^2+H21^2)  
I35: @SQRT(C30^2+D30^2+E30^2+F30^2+G30^2)  
J35: @SUM(H30..M30)  
K35: @SQRT(N30^2)



Attachment VIII

WCAP-14684

(First use of P/T limit methodology. No change from that provided  
in September 13, 1996 RG&E letter to NRC.)