



SUPPLEMENTARY REPORT FOR THE
UNRESOLVED ISSUES OF R. E. GINNA
STORAGE TANKS SEISMIC QUALIFICATION

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SUPPLEMENTARY REPORT FOR THE
UNRESOLVED ISSUES OF R. E. GINNA
STORAGE TANKS SEISMIC QUALIFICATION

Prepared for
ROCHESTER GAS & ELECTRIC CORPORATION
89 East Avenue
Rochester, New York

September 1984

Prepared by
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458 Boston Street
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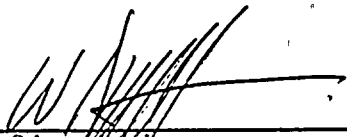
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CERTIFICATION

The undersigned, a registered Professional Engineer, competent in the field of component stress analysis, certifies that to the best of his knowledge and belief the analysis calculations for the subject tanks as presented in this seismic stress report comply with the provisions of the applicable portions of the ASME Boiler and Pressure Vessel Code, Section III, Nuclear Power Plant Components and standard acceptable engineering practice.

Components: Refueling Water Storage Tank and Horizontal Waste Hold-up Tank
Plant: R. E. Ginna




Walter Djordjevic
Commonwealth of Massachusetts
No. 30495

SUPPLEMENTARY REPORT FOR THE
UNRESOLVED ISSUES OF R. E. GINNA
STORAGE TANKS SEISMIC QUALIFICATION
Revision 0, September 1984

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1. INTRODUCTION

The seismic qualification for the storage tanks at the R. E. Ginna nuclear power plant has been performed by Stevenson & Associates in References 1 to 3. The storage tanks analyzed are the Refueling Water Storage Tank (RWST), Vertical Hold-up Tanks, and the Waste Hold-up Tank. A review of the Stevenson & Associates qualification reports has been conducted by Structural Mechanics Associates, Inc. (Ref. 4) in support of EG&G Idaho, Inc. The solutions to some of the issues raised in the review and the analysis of RWST under external nozzle loads are presented in this report.

Section 2 of this report contains the summary of the results. Section 3 presents the analysis procedure and results for the nozzle load analysis. Section 4 shows that the shear cone pull-out capacity is greater than the yielding capacity of the anchor bolts. Section 5 provides an alternative analysis for the Waste Hold-up Tank for which peak of the response spectra is used for the transverse direction seismic response.

2. SUMMARY OF RESULTS

The results of the nozzle analysis indicate that the 10-inch nozzle near the base of the RWST resist the combination of the deadweight, injection phase loading, and the SSE loading within the limits of the acceptance criteria. The minimum factor of safety is 1.35.

The analysis of the shear cone pull-out capacity indicates that the previous analysis based on yielding failure of the anchor bolts is satisfactory. The strength of concrete shear cone pull-out is much greater than the yield strength of the anchor bolt.

The re-analysis for the Waste Hold-up Tank based on the peak of the spectra shows that the stresses are still within the acceptance criteria. This is because the transverse direction is less critical for the stresses concerned. The effect of non-full tank conditions are also considered. The results show that full tank is the most critical condition due to the low frequencies associated with the sloshing modes.

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3. NOZZLE LOAD ANALYSIS FOR RWST

The nozzle considered is the 10-inch butt welding nozzle near the base of the RWST. Both the tank shell and the reinforcing plate have a thickness of 5/16-inch.

Loads

External piping loads due to deadweight, injection phase, and safe shutdown earthquake (SSE) are considered. The magnitude of the loads is provided by RG&E in Ref. 5 and is tabulated in Table I.

The stresses in the tank wall due to hydrostatic pressure and SSE seismic loads are also considered and combined with the piping loads.

Acceptance Criteria

The acceptance criteria for the analysis is based on ASME Pressure Vessel & Piping code, Section III, Division I, Subsection NC-3867. The stresses calculated are:

- Primary membrane stresses due to internal pressure
- Primary membrane stresses due to tank seismic response
- Local membrane stresses due to external piping loads

Peak stresses due to stress concentration at the corners and the local bending stresses due to external piping loads are not calculated since fatigue need not be considered in seismic qualification.

The allowable stress is 2.4S according to the code. From the value of S given in Ref. 1, the allowable stress is

$$\text{Allowable stress} = 2.4 \times 17.8 = 42.7 \text{ ksi}$$

Analysis Procedure and Results

Stresses in the tank wall due to hydrostatic pressure and tank SSE response are obtained from Ref. 1.

The local membrane stresses due to external piping loads are calculated using the Bijlaard's method (Ref. 6). The stresses at eight response points are calculated (see the figure in Table II). For locations immediately adjacent the nozzle, a tank wall thickness of 5/8-inch is used in the analysis. For locations at the junction of the reinforcing plate and the vessel, the nozzle diameter is taken to equal the dimensions of the reinforcing plate while using 5/16-inch as the tank wall thickness. This procedure is suggested by Bijlaard in Ref. 7 and 8 for cases with reinforcing pads and has been shown to yield conservative results.

The resulting maximum principal stresses and the safety margins are summarized in Table II.

Types of Loading	Forces (kips)			Moments (in-kips)		
	FX	FY	FZ	MX	MY	MZ
Deadweight	0.01	0.15	0.02	-0.2	2.0	-1.7
Injection	-0.32	2.54	-0.11	-3.1	-9.6	-60.7
Phase	1.54	0.44	2.24	16.2	113.0	10.5
SSE						

TABLE I External Piping Loads

Location	Maximum Principal Stress (ksi)	Factors of Safety
A	14.7	2.9
B	14.4	3.0
C	12.6	3.4
D	12.3	3.5
A'	31.6	1.35
B'	30.7	1.39
C'	26.0	1.64
D'	25.5	1.67

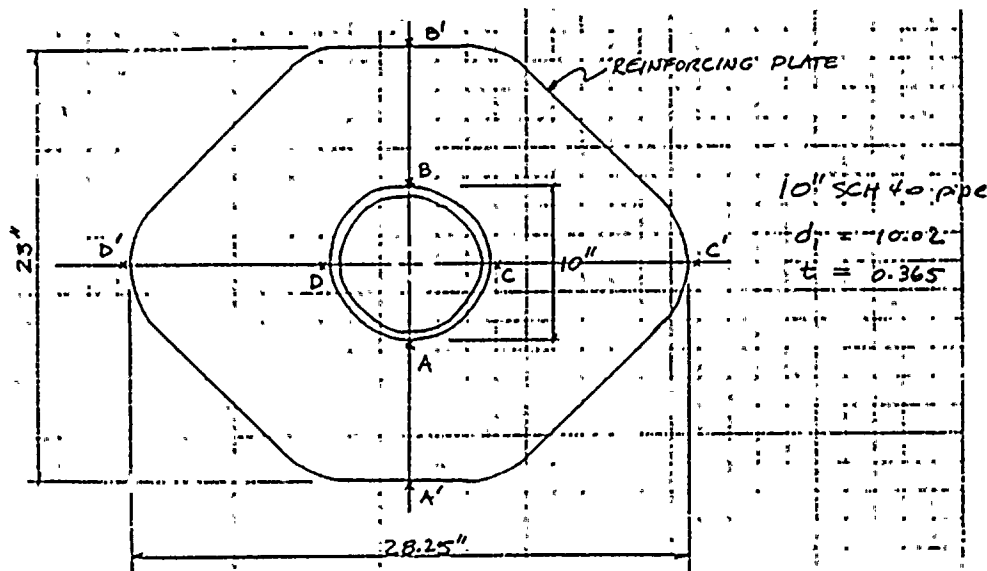


TABLE II Maximum Principal Stresses and Safety Margins

4. ANCHOR BOLT SHEAR CONE PULL-OUT CAPACITY

In the previous analysis report for RWST (Ref. 1), The capacity of the base anchor bolts is assumed to be governed by the yielding of the anchor bolts. The allowable load for one bolt is equal to 132 kips. There has been questions as to whether the capacity may be governed by the pull-out of the concrete shear cone.

Based on ACI 349-77, Appendix B, and taking into account the reduction of capacity due to overlapping of adjacent shear cones, the allowable load for one bolt is 312 kips. This verifies the validity of the previous analysis.



5. ALTERNATIVE ANALYSIS FOR THE WASTE HOLDUP TANK

In the previous qualification report for the Waste Holdup Tank (Ref. 3), the fundamental frequency for the tank in the transverse direction had been calculated and was shown to be rigid. The review by SMA (Ref. 4) questioned the validity of the calculation and suggested that the fundamental frequency would occur at a lower frequency.

Due to the complex geometry of the structure, the basic assumptions in the approximate analytical analysis cannot be easily justified save for a full scale finite element analysis. An alternative analysis procedure, using 1.5 times the peak of the response spectra at 7% damping as the applied lateral load, is performed for the transverse direction seismic response. The resulting stresses are summarized in Table III, where all the stresses are the same as in Ref. 3 except that the stresses due to transverse earthquake are increased by the new spectral acceleration. The results show that the stresses are still well within the allowable level even with the conservative spectral acceleration assumed. This is because the transverse direction is by far the less critical direction for the stresses concerned.

The effects of sloshing for partially full tanks are also evaluated. The frequencies for the Waste Holdup Tank with water levels of 1/4, 1/2, and 3/4 depth are calculated based on the figures in Ref. 9. The frequencies are 0.40, 0.44, and 0.54 Hz in the transverse direction and 0.12, 0.18, and 0.23 Hz in the longitudinal direction. The corresponding spectral accelerations at 0.5% damping are 0.109, 0.116, and 0.134 g in the transverse direction. Since these spectral accelerations are far less than the accelerations considered for the full tank case, it is concluded that the full tank is the most critical condition for stress evaluation.

Component	Maximum Stress (ksi)	Allowable Stress (ksi)	Safety Factor
TANK WALL			
Membrane Stress	5.9	35.6	6.0
Local Bending Stress	65	80	1.23
Ring Compression	11.9	35.6	2.99
SPHERICAL HEAD			
Membrane Tension	1.2	35.6	30
ANCHOR BOLTS			
Tensile Stress	23.5	40.6	1.73
SADDLE SUPPORT			
Tension in Flange	1.0	24	24
Flange Bending	35.5	39.6	1.12
Web Compression	1.6	45	28
Center Stiffener	4.0	36	9.0
End Stiffener	5.2	36	6.9
WELDING			
Tank-Saddle	10.5	36	3.4

TABLE III Calculated Maximum Stresses and Safe Margins

6. REFERENCES

1. Stevenson & Associates, "Seismic Qualification Report for the Refueling Water Storage Tank at the R. E. Ginna Plant," prepared for Rochester Gas & Electric Corporation, August 1983.
2. Stevenson & Associates, "Seismic Qualification Report for Vertical Hold-up Tanks at the R. E. Ginna Plant," prepared for Rochester Gas & Electric Corporation, September 1983.
3. Stevenson & Associates, "Seismic Qualification Report for the Waste Hold-up Tank at the R. E. Ginna Plant," prepared for Rochester Gas & Electric Corporation, September 1983.
4. D. A. Wesley, "Review of the Seismic Qualification of the R. E. Ginna Storage Tank Seismic Qualification for the Systematic Evaluation Program (SEP)," prepared for EG&G Idaho, Inc., June 1984.
5. Stevenson & Associates document 83C2209-LR-001.
6. K. R. Wichman, A. G. Hopper, and J. L. Mershon, "Local Stresses in Spherical and Cylindrical Shells due to External Loadings," Welding Research Council Bulletin 107, August 1965, (Revised March 1979).
7. P. P. Bijlaard, "Stress from Radial Loads in Cylindrical Pressure Vessels," Welding Research Supplement, Vol. 33, No. 12, 1954, pp. 615s-623s.
8. P. P. Bijlaard, "Additional Data on Stresses in Cylindrical Shells Under Local Loading," Welding Research Council Bulletin 50, May 1959, pp. 10-50.
9. H. N. Abramson, "The Dynamic Behavior of Liquids in Moving Containers," NASA SP-106, 1966.

7. APPENDIX -- Analytical Calculations

SYSTEM GINNA Water Tanks

COMPONENT NAME RWST Waste Holdup Tank COMPONENT NO N/A

LOCATION Aux. Building ELEVATION 236 ft

COMPONENT SAFETY FUNCTION: ACTIVE ☐ PASSIVE 1 ☐ 2 ☒

S-LIST PAGE NO N/A

METHOD OF ANALYSIS: Analytical stress Analysis

SPECTRAL CURVES USED: R.E. Ginna Site Specific Ground Spectra contained in B3C2209-DR-005, File 3

DAMPING VALUE ASSUMED: 7 %

ACCEPTANCE BEHAVIOR CRITERIA USED: ASME B&PV Code, Section III, 1980, ACI 349-77, Appendix B,

COMPUTER CODE USED: N/A

REMARKS:

S&A

DESIGN
REPORT
COVER
SHEET

FIGURE 9.0

CONTRACT NO

83C2209

REV. NO	0
BY	TMT
DATE	9/4/84
CHK'D	WD
DATE	9/11/84
APPR.	WD
DATE	9/11/84



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JOB No. 83C2209 SHEET 1 OF 27

GINNA UNSETTLED ISSUES

REVISIONS	0	TMT 974-SV
		WD 9/13/84

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27

GINNA

UNSETTLED ISSUES

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WJ 9/13/84

INTRODUCTION

The objective of this calculation is to settle the unresolved issues of S&A project 83C2209 - Qualification of RWST, Vertical holdup tanks, and waste holdup tank at R.E. GINNA power plant. The issues discussed are based on the conference phone conversation on August 1, 1984.

Three goals are to be achieved in this calc.

1. Qualify RWST subject to external piping nozzle loads.
2. Verify that the concrete pull-out capacity of the anchor bolts at the base of the RWST is greater than the allowable yield stress. This ensures the factor of safety calculated previously is the minimum.
3. Provide an alternative analysis for the transverse response of the waste holdup tank using plate of the spectra, and thus skirt the calculation of fundamental frequency.



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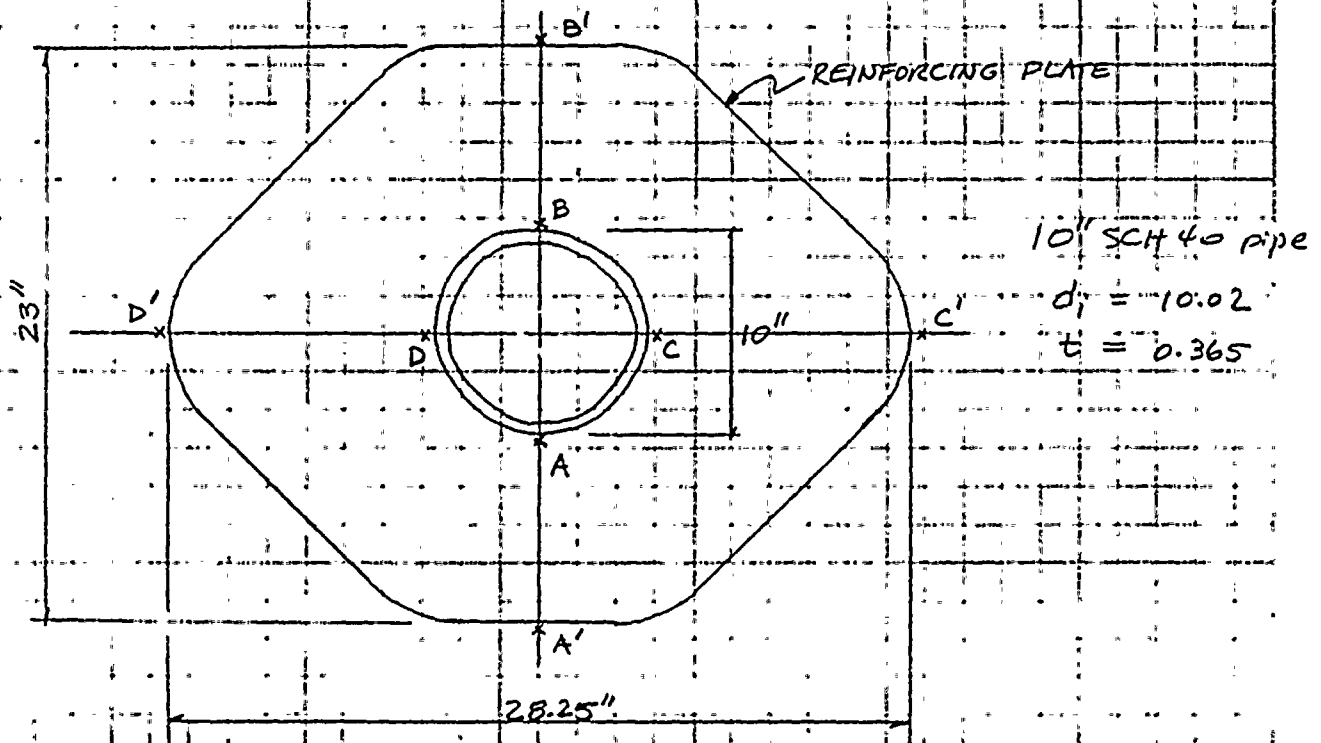
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RWST NOZZLE LOADS

REVISIONS	0	TMT	8/31/84
		KBM	9/19/84

RWST NOZZLE LOAD ANALYSIS

The nozzle considered is the 10" butt welding nozzle near the bottom of the RWST. The details of the pipe-tank connection is shown on the next page. The tank wall thickness is 5/16 inch at the location, where the nozzle opening is reinforced by a 5/16" plate of the shape as shown in the figure below:



Bijlaard analysis is performed based on the graphs in Ref. 1.
Eight stress points are checked:

Points A, B, C, D at junction of nozzle and reinforcing plate

Points A', B', C', D' at junction of reinforcing plate and tank wall.

MACHINE 1.5 TO
10.42" ϕ

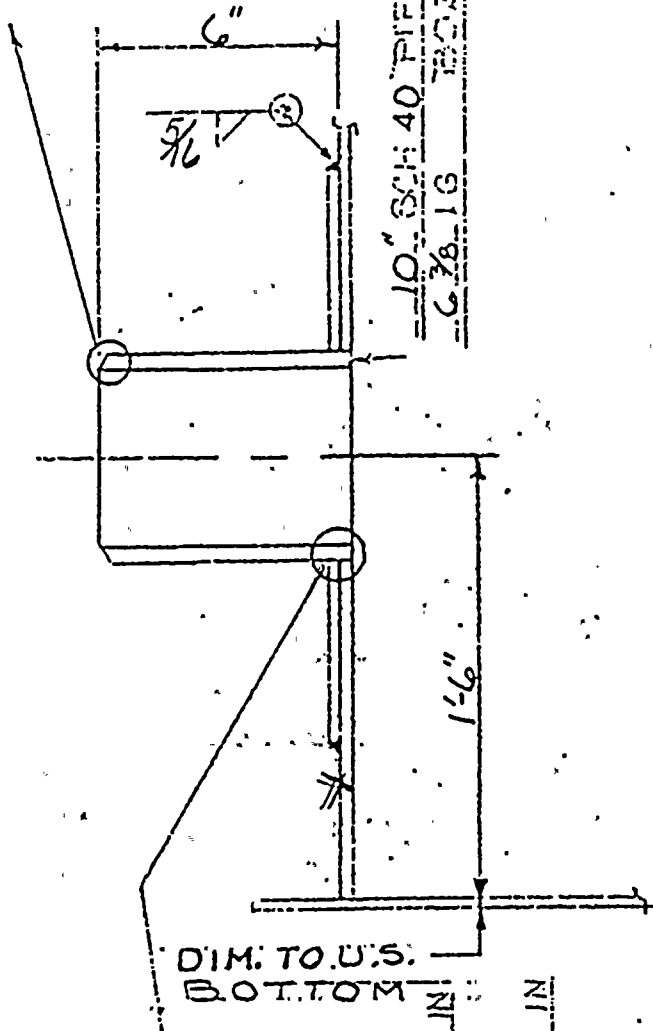
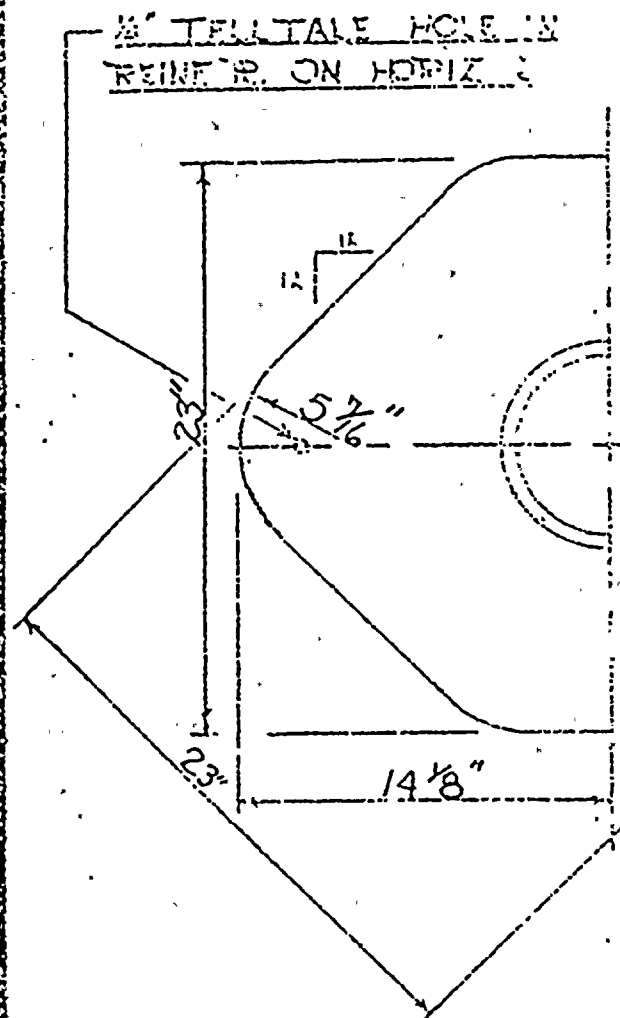
37.5°

BEV. OS.

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FOR INFORMATION ONLY

KBM 9/19/84



10" BUTT WELDING NOZZLE

STAINLESS STEEL

OUTLET & SUCTION CONN

2 FEED PER TX
1 TANKER FEED

ITEM NO. MB-3

PITTSBURGH-DES MOINES STEEL CO. (PART)
ENGINEERS - MANUFACTURERS - SUPPLIERS

GILBERT ASSOCIATES, INC

DATE 12-14-66 216020

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RWST NOZZLE LOADS

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LOADS

- External loads from piping include:

- Deadweight
- Injection phase
- SSE

The magnitude of the loads is provided in 54A QA file 83C2209-LR-001

	FX	FY	FZ	MX	MY	MZ
Deadweight	0.01 k	0.15	0.02	-0.2 k-m	2.0	-1.7
Inj. phase	-0.32	2.54	-0.11	-3.1	-9.6	-60.7
SSE	1.54	0.44	2.24	16.2	113.0	10.5

Since the directions of the deadweight and the injection phase load are fixed, the two loads are summed algebraically. The induced stresses are calculated on standard Bijlaard analysis forms.

The stresses induced by SSE are also calculated on standard Bijlaard analysis forms, however, the stresses due to different component of loads are combined by the SRSS rule. Since the seismic load can act in both directions, the final stress values are the sum of the absolute values of the deadweight + injection phase loads and the SSE loads.

- Besides the above external loads, stresses in the RWST tank wall due to internal pressure and SSE response will also be considered.



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RWST NOZZLE LOADS

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KRM 9/19/84

ACCEPTANCE CRITERIA

The acceptance criteria will be based upon ASME Pressure Vessel and Piping code, Section III, Division I, Section NC-3867 (b). The stresses calculated are:

- primary membrane stresses due to internal pressure
- primary membrane stresses due to tank seismic response
- local primary membrane stresses due to external piping loads

Peak stresses due to stress concentration at the corners and the local bending stresses (secondary stresser) due to external piping loads will not be considered since fatigue is not of concern in seismic analysis.

The allowable stress, according to NC-3867 (b), is $2.4S$. S has been calculated in the previous RWST report, $S = 17.8$ ksi, thus

$$\text{Allowable stress} = 2.4 \times 17.8 = \underline{\underline{42.7}} \text{ ksi}$$



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RWST NOZZLE LOADS

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STRESS CALCULATIONS

The maximum membrane stresses in the unreinforced tank wall are from the previous report,

	Dead load	Seismic
Hoop stress σ_θ	17.7 ksi	4.7
Axial stress σ_x	0.2	11.7
Shear stress $\tau_{\theta x}$	0	3.2

At the reinforcing plate, these stresses are 1/2 of the above.

The local membrane stresses due to piping loads are calculated by Bijlaard's method in pages 8-15.

For locations just next to the nozzle, i.e., points A, B,

C, D, the analysis is made as if the tank wall having a thickness of 5/8 inch. For locations at the junction of the vessel and the reinforcing plate, (Points A', B', C', D')

the nozzle diameter is taken as equal to the dimension of the reinforcing plate, while original tank thickness 5/8 inch is used. This procedure is recommended by Bijlaard in Ref. 2 and 3 and has been shown to be conservative.

Sign of

For convenience, the deadweight + injection phase loads have been changed. The parameter γ for points A', B', C', and D' is 509 which is outside the charts provided in Reference 1. The data was extrapolated on the log-log scale.



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RWST NOZZLE LOADS

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REVISIONS

DEAD WEIGHT + INJECTION PHASE

LOCATIONS : A, B, C, D

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads*

Radial load, $P = 310$ lb.
Circ. Moment, $M_c = 7600$ n. lb.
Long. Moment, $M_L = 62400$ n. lb.
Torsion Moment, $M_T = 3300$ n. lb.
Shear Load, $V_c = 70$ lb.
Shear Load, $V_L = 2650$ lb.

2. Geometry

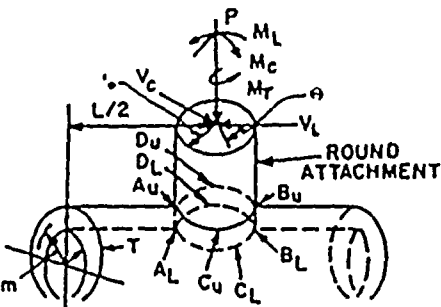
Vessel thickness, $T = 0.625$ in.
Attachment radius, $R_a = 5.38$ in.
Vessel radius, $R_m = 159$ in.

3. Geometric Parameters

$\gamma = \frac{R_m}{T} = 254$
 $\beta = (0.875) \frac{R_a}{R_m} = 0.03$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = 1.0$

*NOTE: Enter all force values in accordance with sign convention



CYLINDRICAL SHELL

From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES - if load is opposite that shown, reverse signs shown							
			Au	AL	Bu	BL	Cu	CL	Du	DL
3C or 4C	$\frac{N\psi}{P/R_m} = 41$ $\frac{N\psi}{P/R_m} = 47$	$K_n \left(\frac{N\psi}{P/R_m} \right) \cdot \frac{P}{R_m T} =$	130 150	-150	-150	-150	-130	-130	-130	-130
1C or 2C-1	$\frac{M\psi}{P} =$	$K_b \left(\frac{M\psi}{P} \right) \cdot \frac{6P}{T^2} =$		+		+		+		+
3A	$\frac{M\psi}{M_c/R_m \beta} = 6$	$K_n \left(\frac{M\psi}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} =$	100				-100	-100	+100	+100
1A	$\frac{M\psi}{M_c/R_m \beta} =$	$K_b \left(\frac{M\psi}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$						+		+
3B	$\frac{M\psi}{M_L/R_m \beta} = 21$	$K_n \left(\frac{M\psi}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	2760	-2760	-2760	+2760	+2760			
1B or 1B-1	$\frac{M\psi}{M_L/R_m \beta} =$	$K_b \left(\frac{M\psi}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$								
Add algebraically for summation of ϕ stresses, σ_ϕ			-2910	-2910	+2610	+2610	-230	-230	-30	-30
3C or 4C	$\frac{N_a}{P/R_m} = 41$ $\frac{N_a}{P/R_m} = 47$	$K_n \left(\frac{N_a}{P/R_m} \right) \cdot \frac{P}{R_m T} =$	130 150	-130	-130	-130	-130	-150	-150	-150
1C-1 or 2C	$\frac{M_a}{P} =$	$K_b \left(\frac{M_a}{P} \right) \cdot \frac{6P}{T^2} =$		+		+		+		+
4A	$\frac{M_a}{M_c/R_m \beta} = 3.5$	$K_n \left(\frac{M_a}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} =$	120				-120	-120	+120	+120
2A	$\frac{M_a}{M_c/R_m \beta} =$	$K_b \left(\frac{M_a}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$						+		+
4B	$\frac{M_a}{M_L/R_m \beta} = 15.8$	$K_n \left(\frac{M_a}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	760	-760	-760	+760	+760			
2B or 2B-1	$\frac{M_a}{M_L/R_m \beta} =$	$K_b \left(\frac{M_a}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$								
Add algebraically for summation of X stresses, σ_x			-890	-890	630	630	-270	-270	-30	-30
Shear stress due to Torsion, M_T	$\tau_{xy} = \tau_{yx} = \frac{M_T}{2\pi R_m^2 T} = 30$		+30	+30	+30	+30	+30	+30	+30	+30
Shear stress due to load, V_c	$\tau_{xy} = \tau_{yx} = \frac{V_c}{\pi R_m T} = 10$		+10	+10	-10	-10				
Shear stress due to load, V_L	$\tau_{xy} = \tau_{yx} = \frac{V_L}{\pi R_m T} = 250$						+250	+250	+250	+250
Add Algebraically for summation of shear stresses, τ_{xy}			40	40	20	20	280	280	-220	-220



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RWST NOZZLE LOADS

TMT 8/31/84

KSM 9/10/84

REVISIONS

SSE

LOCATIONS: A, B, C, D

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads*

Radial load,
Circ. Moment,
Long. Moments,
Torsion Moment,
Shear Load,
Shear Load,

$$P = 1540 \text{ lb.}$$

$$M_c = 113000 \text{ in. lb.}$$

$$M_L = 10500 \text{ in. lb.}$$

$$M_T = 16200 \text{ in. lb.}$$

$$V_c = 2240 \text{ lb.}$$

$$V_L = 440 \text{ lb.}$$

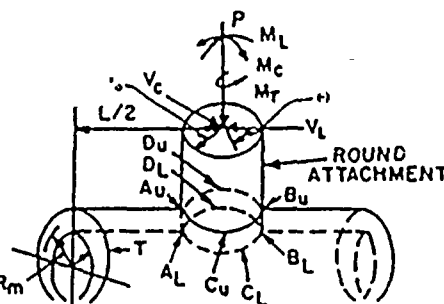
2. Geometric Parameters

$$\gamma = \frac{R_m}{T} = 254$$

$$B = (0.875) \frac{t_o}{R_m} = 0.03$$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = \frac{1.0}{R_m}$

*NOTE: Enter all force values in accordance with sign convention



CYLINDRICAL SHELL

From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES - if load is opposite that shown, reverse signs shown							
			A _u	A _L	B _u	B _L	C _u	C _L	D _u	D _L
3C or 4C	$\frac{M_b}{P/R_m} = 41$	$K_n \left(\frac{M_b}{P/R_m} \right) \cdot \frac{P}{R_m T} = 640$	-730	-730	-730	-730	-640	-640	-640	-640
1C or 2C-1	$\frac{M_b}{P} =$	$K_b \left(\frac{M_b}{P} \right) \cdot \frac{6P}{T} =$								
3A	$\frac{M_c}{M_c/R_m \beta} = 6$	$K_n \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} = 1430$					-1430	-1430	+1430	+1430
1A	$\frac{M_c}{M_c/R_m \beta} =$	$K_b \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T} =$								
3B	$\frac{M_L}{M_L/R_m \beta} = 21$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} = 470$	-470	-470	+470	+470				
1B or 2B-1	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T} =$								
SRSS P, M _c , M _L on			870	870	870	870	1570	1570	1570	1570
3C or 4C	$\frac{M_b}{P/R_m} = 41$	$K_n \left(\frac{M_b}{P/R_m} \right) \cdot \frac{P}{R_m T} = 640$	-640	-640	-640	-640	-730	-730	-730	-730
1C-1 or 2C	$\frac{M_b}{P} =$	$K_b \left(\frac{M_b}{P} \right) \cdot \frac{6P}{T} =$								
4A	$\frac{M_c}{M_c/R_m \beta} = 3.5$	$K_n \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} = 1790$					-1790	-1790	+1790	+1790
2A	$\frac{M_c}{M_c/R_m \beta} =$	$K_b \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T} =$								
4B	$\frac{M_L}{M_L/R_m \beta} = 5.8$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} = 130$	-130	-130	+130	+130				
2B or 2B-1	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T} =$								
SRSS P, M _c , M _L on			650	650	650	650	1930	1930	1930	1930
Add algebraically for combined stress, σ _x										
Shear stress due to Torsion, M _T	$\tau_{xy} = \tau_{yx} = \frac{M_T}{2\pi r_o^2 T} = 140$		+140	+140	+140	+140	+140	+140	+140	+140
Shear stress due to load, V _c	$\tau_{xy} = \tau_{yx} = \frac{V_c}{\pi r_o T} = 210$		+210	+210	-210	-210				
Shear stress due to load, V _L	$\tau_{xy} = \tau_{yx} = \frac{V_L}{\pi r_o T} = 40$						-40	-40	+40	+40
SRSS M _T , V _c , V _L on			250	250	250	250	150	150	150	150
Add algebraically for combined stress, τ _{xy}										



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SUBJECT _____ JOB No. 83C2209 SHEET 10 OF 27

R.W.S.T. NOzzle LOADS

REVISIONS	0	TMT 8/31/84
	1	KBM 9/19/84

DEAD WEIGHT + INJECTION PHASE

LOCATIONS: "A", "B"

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads*

Radial load, $P = 310$ lb.
Circ. Moment, $M_c = 2600$ in. lb.
Long. Moments, $M_L = 62400$ in. lb.
Torsion Moment, $M_t = 3300$ in. lb.
Shear Load, $V_c = 90$ lb.
Shear Load, $V_L = 2690$ lb.

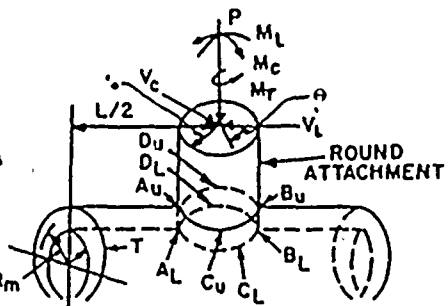
2. Geometry

Vessel thickness, $T = 0.2125$ in.
Attachment radius, $r_o = 16.5$ in.
Vessel radius, $R = 15.9$ in.

3. Geometric Parameters

$\gamma = \frac{R_m}{T} = \frac{509}{0.2125} = 2395$
 $\beta = (0.875) \frac{r_o}{R_m} = 0.063$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = \frac{1.0}{R_m}$
*NOTE: Enter all force values in accordance with sign convention



CYLINDRICAL SHELL

From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES - if load is opposite that shown, reverse signs shown							
			A _u	A _L	B _u	B _L	C _u	C _L	D _u	D _L
3C or 4C	$\frac{M_c}{P/R_m} = 71$	$K_n \left(\frac{M_c}{P/R_m} \right) \cdot \frac{P}{R_m T} = 440$	-440	-440	-440	-440	-	-	-	-
1C or 2C-1	$\frac{M_c}{P} =$	$K_b \left(\frac{M_c}{P} \right) \cdot \frac{6P}{T} =$	-	-	-	-	-	+	-	+
3A	$\frac{M_c}{M_c/R_m \beta} =$	$K_n \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} =$					-	-	+	+
1A	$\frac{M_c}{M_c/R_m \beta} =$	$K_b \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T} =$					-	+	+	-
3B	$\frac{M_L}{M_L/R_m \beta} = 60$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} = 7520$	-7520	-7520	+7520	+7520				
1B or 1B-1	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T} =$								
Add algebraically for summation of ϕ stresses, ϕ_ϕ			-7960	-7960	7080	7080				
3C or 4C	$\frac{M_L}{P/R_m} = 46$	$K_n \left(\frac{M_L}{P/R_m} \right) \cdot \frac{P}{R_m T} = 290$	-290	-290	-290	-290	-	-	-	-
1C-1 or 2C	$\frac{M_L}{P} =$	$K_b \left(\frac{M_L}{P} \right) \cdot \frac{6P}{T} =$	-	-	-	-	-	+	-	+
4A	$\frac{M_L}{M_L/R_m \beta} =$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$					-	-	+	+
2A	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T} =$					-	+	+	-
4B	$\frac{M_t}{M_t/R_m \beta} = 22$	$K_n \left(\frac{M_t}{M_t/R_m \beta} \right) \cdot \frac{M_t}{R_m \beta T} = 2760$	-2760	-2760	+2760	+2760				
2B or 2B-1	$\frac{M_t}{M_t/R_m \beta} =$	$K_b \left(\frac{M_t}{M_t/R_m \beta} \right) \cdot \frac{6M_t}{R_m \beta T} =$								
Add algebraically for summation of χ stresses, χ_χ			-3050	-3050	2470	2470				
Shear stress due to Torsion, M_t		$\tau_\phi = \tau_\chi = \frac{M_t}{2\pi r_o^2 T} = 13$	+13	+13	+13	+13	+	+	+	+
Shear stress due to load, V_c		$\tau_\phi = \frac{V_c}{\pi r_o T} = 8$	+8	+8	-8	-8				
Shear stress due to load, V_L		$\tau_\phi = \frac{V_L}{\pi r_o T} =$					-	-	+	+
Add Algebraically for summation of shear stresses, τ_ϕ			20	20	5	5				



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SUBJECT _____ JOB No. 83C2209 SHEET 11 OF 27

RWST. NOZZLE LAPS

REVISIONS	0	TMT	8/31/80
		KGM	9/19/84

SSE

LOCATIONS : A', B'

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads

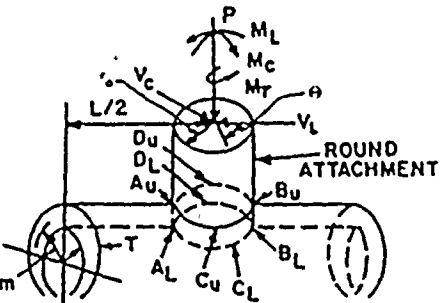
Radial load, $P = 1540$ lb.
Circ. Moment, $M_c = 113000$ in. lb.
Long. Moment, $M_L = 10500$ in. lb.
Torsion Moment, $M_T = 16200$ in. lb.
Shear Load, $V_c = 2240$ lb.
Shear Load, $V_L = 440$ lb.

2. Geometric Parameters

$\gamma = \frac{R_n}{T} = 509$
 $\beta = (0.875) \frac{t_o}{R_m} = 0.063$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = 1.0$

*NOTE: Enter all force values in accordance with sign convention

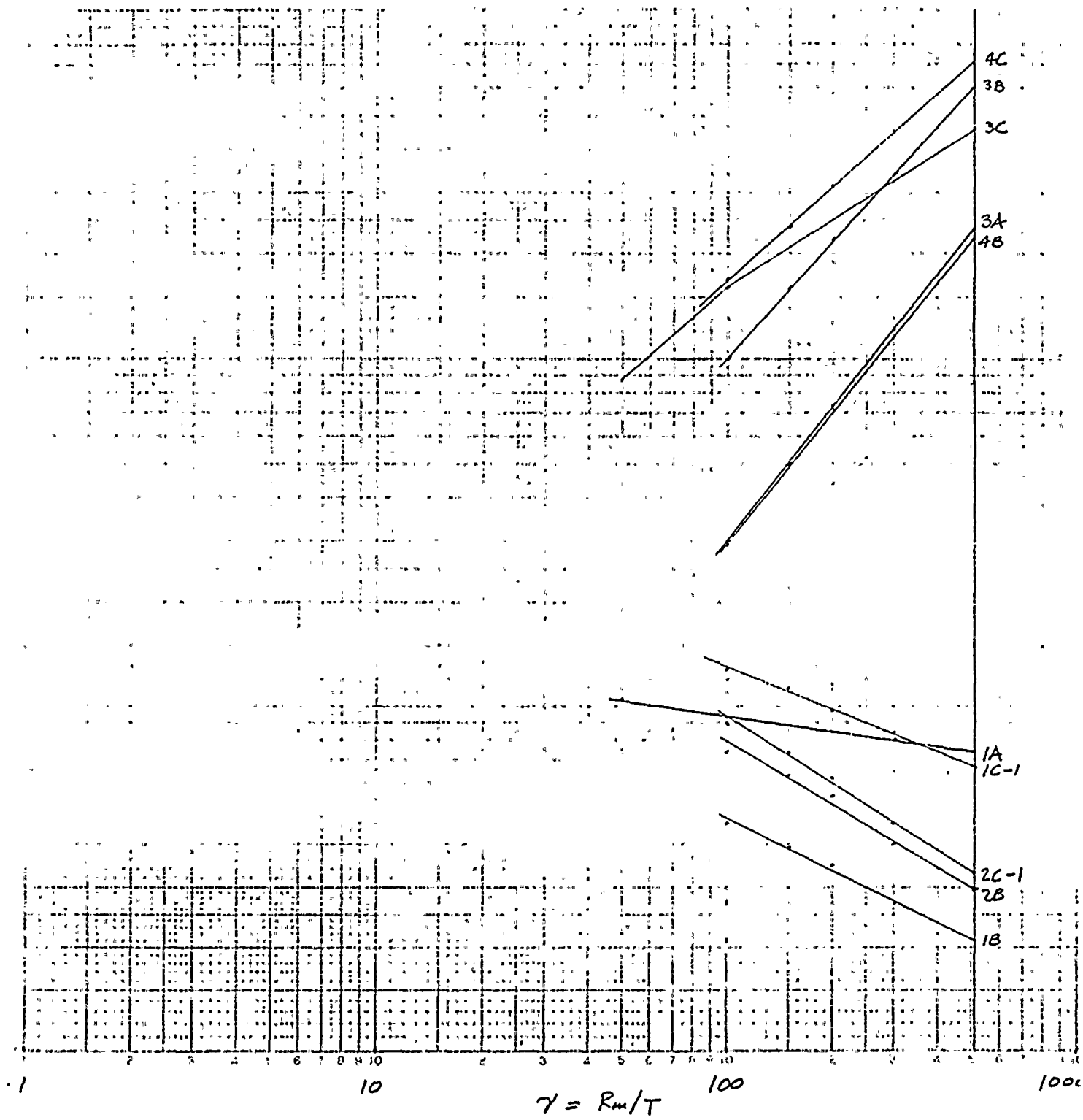


CYLINDRICAL SHELL

From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES - if load is opposite that shown, reverse signs shown							
			Au	AL	Bu	BL	Cu	CL	Du	DL
2C or 4C	$\frac{M_b}{P/R_n} = 71$	$K_n \left(\frac{M_b}{P/R_n} \right) \cdot \frac{P}{R_n T} = 2200$	-2200	-2200	-2200	-2200	-	-	-	-
1C or 2C-1	$\frac{M_c}{P} =$	$K_b \left(\frac{M_c}{P} \right) \cdot \frac{6P}{T^2} =$	-	+	-	+	-	+	-	+
3A	$\frac{M_b}{M_c/R_n \beta} =$	$K_n \left(\frac{M_b}{M_c/R_n \beta} \right) \cdot \frac{M_c}{R_n \beta T} =$					-	-	+	+
1A	$\frac{M_b}{M_c/R_n \beta} =$	$K_b \left(\frac{M_b}{M_c/R_n \beta} \right) \cdot \frac{6M_c}{R_n \beta T^2} =$					-	+	+	-
3B	$\frac{M_b}{M_L/R_n \beta} = 60$	$K_n \left(\frac{M_b}{M_L/R_n \beta} \right) \cdot \frac{M_L}{R_n \beta T} = 1270$	-1270	-1270	-1270	+1270	+1270			
1B or 1B-1	$\frac{M_b}{M_L/R_n \beta} =$	$K_b \left(\frac{M_b}{M_L/R_n \beta} \right) \cdot \frac{6M_L}{R_n \beta T^2} =$								
Add algebraically for combination of stresses, σ_{SRSS}			2540	2540	2540	2540				
3C or 4C	$\frac{M_b}{P/R_n} = 46$	$K_n \left(\frac{M_b}{P/R_n} \right) \cdot \frac{P}{R_n T} = 1430$	-1430	-1430	-1430	-1430	-	-	-	-
1C-1 or 2C	$\frac{M_c}{P} =$	$K_b \left(\frac{M_c}{P} \right) \cdot \frac{6P}{T^2} =$	-	+	-	+	-	+	-	+
4A	$\frac{M_b}{M_c/R_n \beta} =$	$K_n \left(\frac{M_b}{M_c/R_n \beta} \right) \cdot \frac{M_c}{R_n \beta T} =$					-	-	+	+
2A	$\frac{M_b}{M_c/R_n \beta} =$	$K_b \left(\frac{M_b}{M_c/R_n \beta} \right) \cdot \frac{6M_c}{R_n \beta T^2} =$					-	+	+	-
4B	$\frac{M_b}{M_L/R_n \beta} = 22$	$K_n \left(\frac{M_b}{M_L/R_n \beta} \right) \cdot \frac{M_L}{R_n \beta T} = 460$	-460	-460	-460	+460	+460			
2B or 2B-1	$\frac{M_b}{M_L/R_n \beta} =$	$K_b \left(\frac{M_b}{M_L/R_n \beta} \right) \cdot \frac{6M_L}{R_n \beta T^2} =$								
Add algebraically for combination of stresses, σ_{SRSS}			1500	1500	1500	1500				
Shear stress due to Torsion, M_T										
$\tau_{M_T} = T \cdot \phi = \frac{M_T}{2\pi r_o^2 T} = 62$			+62	+62	+62	+62	+62	+	+	+
Shear stress due to load, V_c										
$\tau_{V_c} = \frac{V_c}{\pi r_o^2 T} = 200$			+200	+200	+200	-200	-200			
Shear stress due to load, V_L										
$\tau_{V_L} = \frac{V_L}{\pi r_o^2 T} =$										
Add algebraically for combination of shear stresses, τ_{SRSS}			210	210	210	210				

SHEET 12/27
 BY TMT 8/31/84
 KBM 9/19/84

$$\beta = 0.063$$





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SHEET 13 OF 27

RWST NOZZLE LOADS

TMT 8/31/84

KSM 9/19/84

REVISIONS

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads

Radial load, $P = 310$ lb.
Circ. Moment, $M_c = 2600$ in. lb.
Long. Moments, $M_L = 6240$ in. lb.
Torsion Moment, $M_T = 320$ in. lb.
Shear Load, $V_c = 90$ lb.
Shear Load, $V_L = 269$ lb.

2. Geometry

Vessel thickness, $T = 0.3125$ in.
Attachment radius, $r_o = 14.13$ in.
Vessel radius, $R_m = 159$ in.

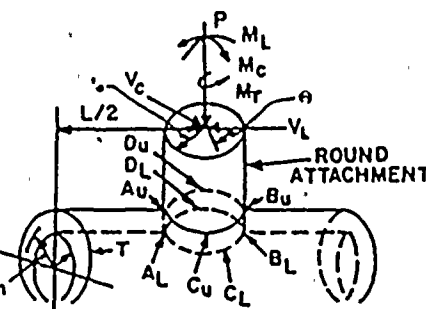
3. Geometric Parameters

$$\gamma = \frac{R_m}{T} = 509$$

$$\beta = (0.875) \frac{r_o}{R_m} = 0.078$$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = R_m$

*NOTE: Enter all force values in accordance with sign convention



CYLINDRICAL SHELL

From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES — if load is opposite that shown, reverse signs shown							
			Au	AL	Bu	BL	Cu	CL	Du	DL
3C or 3C-1	$\frac{M\phi}{P/R_m} = 36$	$K_n \left(\frac{M\phi}{P/R_m} \right) \cdot \frac{P}{R_m T} =$	220	—	—	—	—220	—220	—220	—220
1C or 2C-1	$\frac{M\phi}{P} =$	$K_b \left(\frac{M\phi}{P} \right) \cdot \frac{6P}{T^2} =$	—	—	—	—	—	—	—	—
3A	$\frac{M\phi}{M_c/R_m \beta} = 25$	$K_n \left(\frac{M\phi}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} =$	310	—	—	—	—310	—310	+310	+310
1A	$\frac{M\phi}{M_c/R_m \beta} =$	$K_b \left(\frac{M\phi}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$	—	—	—	—	—	—	—	—
3B	$\frac{M\phi}{M_L/R_m \beta} =$	$K_n \left(\frac{M\phi}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	—	—	—	—	—	—	—	—
1B or 1B-1	$\frac{M\phi}{M_L/R_m \beta} =$	$K_b \left(\frac{M\phi}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$	—	—	—	—	—	—	—	—
Add algebraically for summation of ϕ stresses, σ_ϕ			—	—	—	—	—530	—530	90	90
3C or 4C	$\frac{M_n}{P/R_m} = 65$	$K_n \left(\frac{M_n}{P/R_m} \right) \cdot \frac{P}{R_m T} =$	410	—	—	—	—410	—410	—410	—410
1C-1 or 2C	$\frac{M_n}{P} =$	$K_b \left(\frac{M_n}{P} \right) \cdot \frac{6P}{T^2} =$	—	—	—	—	—	—	—	—
4A	$\frac{M_n}{M_c/R_m \beta} = 48$	$K_n \left(\frac{M_n}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} =$	590	—	—	—	—590	—590	+590	+590
2A	$\frac{M_n}{M_c/R_m \beta} =$	$K_b \left(\frac{M_n}{M_c/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$	—	—	—	—	—	—	—	—
4B	$\frac{M_n}{M_L/R_m \beta} =$	$K_n \left(\frac{M_n}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	—	—	—	—	—	—	—	—
2B or 2B-1	$\frac{M_n}{M_L/R_m \beta} =$	$K_b \left(\frac{M_n}{M_L/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$	—	—	—	—	—	—	—	—
Add algebraically for summation of N stresses, σ_n			—	—	—	—	—1000	—1000	180	180
Shear stress due to Torsion, M_T		$\tau_{\phi} = \tau_{\phi} = \frac{M_T}{2\pi r_o T} = 8$	+	+	+	+	+8	+8	+8	+8
Shear stress due to load, V_c		$\tau_{\phi} = \tau_{\phi} = \frac{V_c}{\pi r_o T} =$	+	+	—	—	—	—	—	—
Shear stress due to load, V_L		$\tau_{\phi} = \tau_{\phi} = \frac{V_L}{\pi r_o T} = 190$	—	—	—	—	—190	—190	+190	+190
Add Algebraically for summation of shear stresses, τ_{ϕ}			—	—	—	—	—180	—180	200	200



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RWST NOZZLE LOADS

REVISIONS	0	TMT 8/31/84
		KPM 9/19/84

Table 5—Computation Sheet for Local Stresses in Cylindrical Shells

1. Applied Loads

Radial load, $P = 1540$ lb.
Circ. Moment, $M_c = 112000$ in. lb.
Long. Moments, $M_L = 10500$ in. lb.
Torsion Moment, $M_T = 16200$ in. lb.
Shear Load, $V_c = 2240$ lb.
Shear Load, $V_L = 440$ lb.

2. Geometry

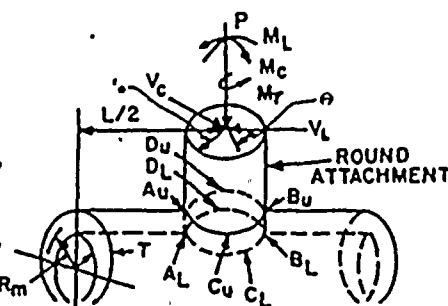
Vessel thickness, $T = 0.3125$ in.
Attachment radius, $r_a = 14.13$ in.
Vessel radius, $R_m = 159$ in.

3. Geometric Parameters

$$\gamma = \frac{R_m}{T} = 509$$

$$\beta = (0.873) \frac{r_a}{R_m} = 0.078$$

Stress Concentration due to:
a) membrane load, $K_n = 1.0$
b) bending load, $K_b = R_m$
NOTE: Enter all force values in accordance with sign convention



CYLINDRICAL SHELL

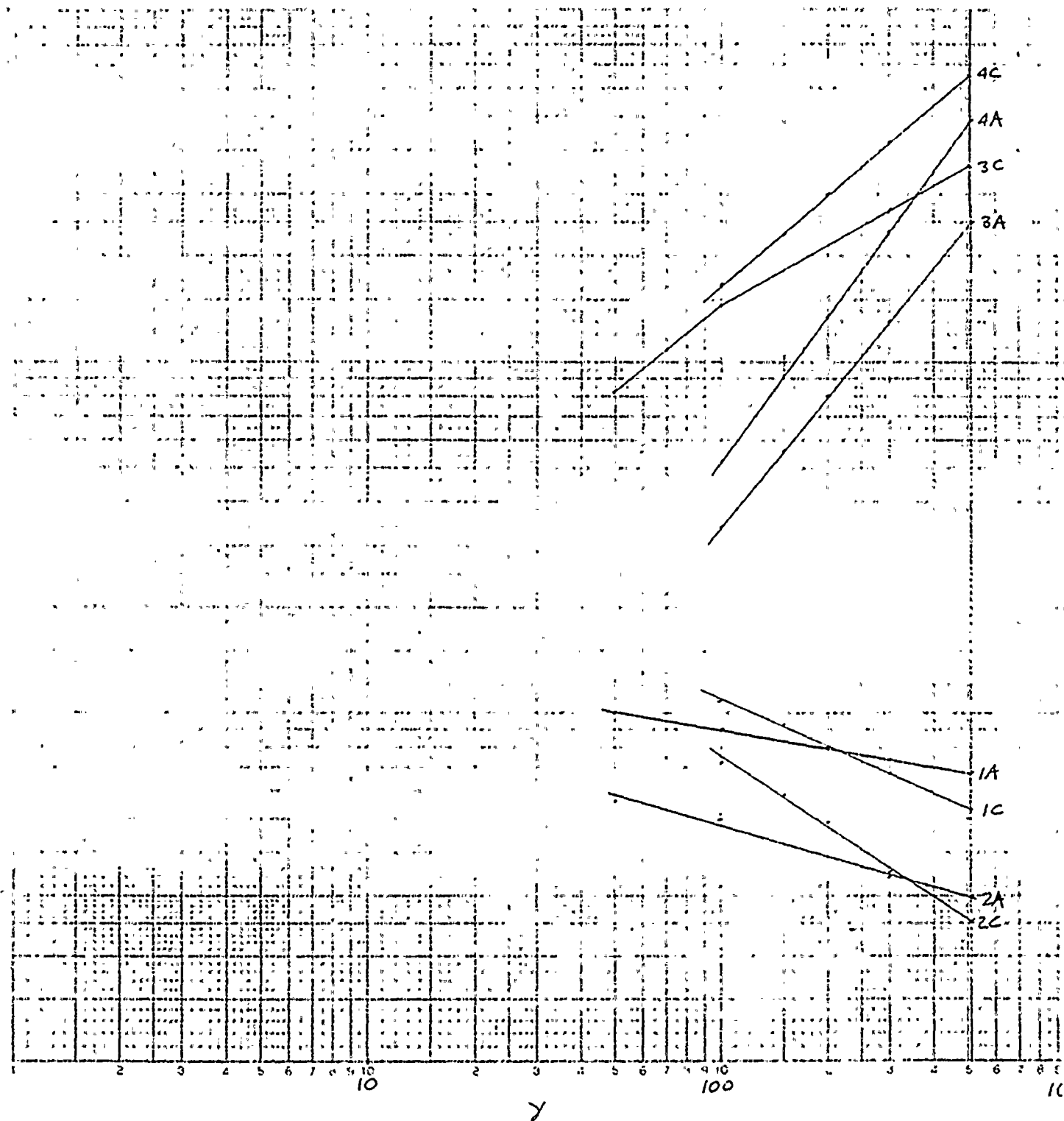
From Fig.	Read curves for	Compute absolute values of stress and enter result	STRESSES — if load is opposite that shown, reverse signs shown							
			A _u	A _L	B _u	B _L	C _u	C _L	D _u	D _L
3C or 4C	$\frac{M_c}{P/R_m} = 36$	$K_n \left(\frac{M_c}{P/R_m} \right) \cdot \frac{P}{R_m T} = 1120$	—	—	—	—	-1120	-1120	-1120	-1120
1C or 2C-1	$\frac{M_L}{P} =$	$K_b \left(\frac{M_L}{P} \right) \cdot \frac{6P}{T^2} =$	—	—	—	—	—	—	—	—
3A	$\frac{M_c}{M_c/R_m \beta} = 25$	$K_n \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} = 4580$					-4580	-4580	+4580	+4580
1A	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$					—	—	—	—
3B	$\frac{M_L}{M_L/R_m \beta} =$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	—	—	+	+				
1B or 1B-1	$\frac{M_c}{M_c/R_m \beta} =$	$K_b \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$	—	+	+	—				
Add algebraically for summation of stresses, σ_{ϕ}							4700	4700	4700	4700
3C or 4C	$\frac{M_c}{P/R_m} = 65$	$K_n \left(\frac{M_c}{P/R_m} \right) \cdot \frac{P}{R_m T} = 2010$	—	—	—	—	-2010	-2010	-2010	-2010
1C-1 or 2C	$\frac{M_L}{P} =$	$K_b \left(\frac{M_L}{P} \right) \cdot \frac{6P}{T^2} =$	—	—	—	—	—	—	—	—
4A	$\frac{M_c}{M_c/R_m \beta} = 48$	$K_n \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{M_c}{R_m \beta T} = 8800$					-8800	-8800	+8800	+8800
2A	$\frac{M_L}{M_L/R_m \beta} =$	$K_b \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{6M_c}{R_m \beta T^2} =$					—	—	—	—
4B	$\frac{M_L}{M_L/R_m \beta} =$	$K_n \left(\frac{M_L}{M_L/R_m \beta} \right) \cdot \frac{M_L}{R_m \beta T} =$	—	—	+	+				
2B or 2B-1	$\frac{M_c}{M_c/R_m \beta} =$	$K_b \left(\frac{M_c}{M_c/R_m \beta} \right) \cdot \frac{6M_L}{R_m \beta T^2} =$	—	+	+	—				
Add algebraically for summation of stresses, σ_{ϕ}							9000	9000	9000	9000
Shear stress due to Torsion, M_T										
$\tau_{\phi} = T \cdot \phi = \frac{M_T}{2\pi r_o^2 T} = 40$			+	+	+	+	+40	+40	+40	+40
Shear stress due to load, V_c										
$\tau_{\phi} = \frac{V_c}{\pi r_o T}$			+	+	—	—				
Shear stress due to load, V_L										
$\tau_{\phi} = \frac{V_L}{\pi r_o T} = 30$							-30	-30	+30	+30
Add algebraically for summation of shear stresses, τ_{ϕ}							50	50	50	50

SHEET 15/27

By TMT 8/31/84

KDM 9/19/84

$$\beta = 0.078$$





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SUBJECT _____ JOB No. 93C2209 SHEET 16 OF 27

RWST NOZZLE LOADS

REVISIONS

0 TMT 8/31/84
KBM 9/19/84

SUMMARY OF STRESSES

		Vessel		Piping		Combined	Maximum Principal stress	Factor of Safety
POINT	STRESS	DW	SSE	DW+IP	SSE			
$DW + DW_{\text{piping}} + IP_{\text{piping}} + \sqrt{SSE_{\text{vessel}}^2 + SSE_{\text{piping}}^2}$								
A	σ_ϕ	8.9	2.4	2.9	0.9	14.4		
	σ_x	0.1	5.9	0.9	0.7	6.9	14.7*	2.9
	T_{xy}	0	1.6	0	0.3	1.6		
B	σ_ϕ	8.9	2.4	2.6	0.9	14.1		
	σ_x	0.1	5.9	0.6	0.7	6.6	14.4	3.0
	T_{xy}	0	1.6	0	0.3	1.6		
C	σ_ϕ	8.9	2.4	0.2	1.6	12.0		
	σ_x	0.1	5.9	0.3	1.9	6.6	12.6	3.4
	T_{xy}	0	1.6	0.3	0.2	1.9		
D	σ_ϕ	8.9	2.4	0	1.6	11.8		
	σ_x	0.1	5.9	0	1.9	6.3	12.3	3.5
	T_{xy}	0	1.6	0.2	0.2	1.8		
A'	σ_ϕ	17.7	4.7	8.0	2.5	31.0		
	σ_x	0.2	11.7	3.1	1.5	15.1	31.6	1.35
	T_{xy}	0	3.2	0	0.2	3.2		
B'	σ_ϕ	17.7	4.7	7.1	2.5	30.1		
	σ_x	0.2	11.7	2.5	1.5	14.5	30.7	1.39
	T_{xy}	0	3.2	0	0.2	3.2		
C'	σ_ϕ	17.7	4.7	0.5	4.7	24.8		
	σ_x	0.2	11.7	1.0	9.0	16.0	26.0	1.64
	T_{xy}	0	3.2	0.2	0.1	3.4		
D'	σ_ϕ	17.7	4.7	0.1	4.7	24.4		
	σ_x	0.2	11.7	0.2	9.0	15.2	25.5	1.67
	T_{xy}	0	3.2	0.2	0.1	3.4		

* Maximum principle stress calculated by $(\sigma_x + \sigma_\phi) / 2 + \sqrt{(\sigma_x - \sigma_\phi)^2 / 4 + T_{xy}^2}$



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SUBJECT RWST

JOB No. 83C2209 SHEET 17 OF 27

Anchor Bolt

REVISIONS	0	TMT 8-14-84
	(W)	9/18/84

Anchor bolt shear cone pull out capacity

RWST has 30 - 2 1/2" anchor bolts at the base uniformly distributed around a circle of radius 18'-7". The distance between two adjacent anchor bolts is $S = 2'-10.8"$ (see p. 18). The embedment length of the anchors is 30" (see p. 19).

The diameter of the anchor head at the bottom of the embedment is 9" according to Gilbert Associates Drawing S-423-005 (p. 20 value P). Thus, the radius of the shear cone is

$$R = 30 + \frac{9}{2} = 34.5"$$

Since $R > S/2$, reduction of capacity due to overlapping needs to be made. The reduction should equal twice the shaded area (A_s) below

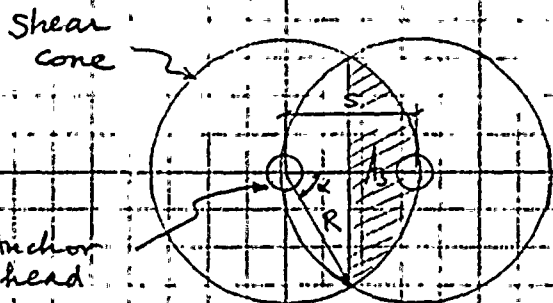
$$\alpha = \cos^{-1}\left(\frac{S}{2R}\right) = 60.4^\circ$$

$$A_s = \pi R^2 \times \frac{2\alpha}{360} = \sqrt{R^2 - \left(\frac{S}{2}\right)^2} \times \frac{S}{2}$$

$$= \pi(34.5)^2 \times \frac{120.8}{360}$$

$$= 17.06 \sqrt{34.5^2 - 17.06^2}$$

$$= 743.1 \text{ in}^2$$



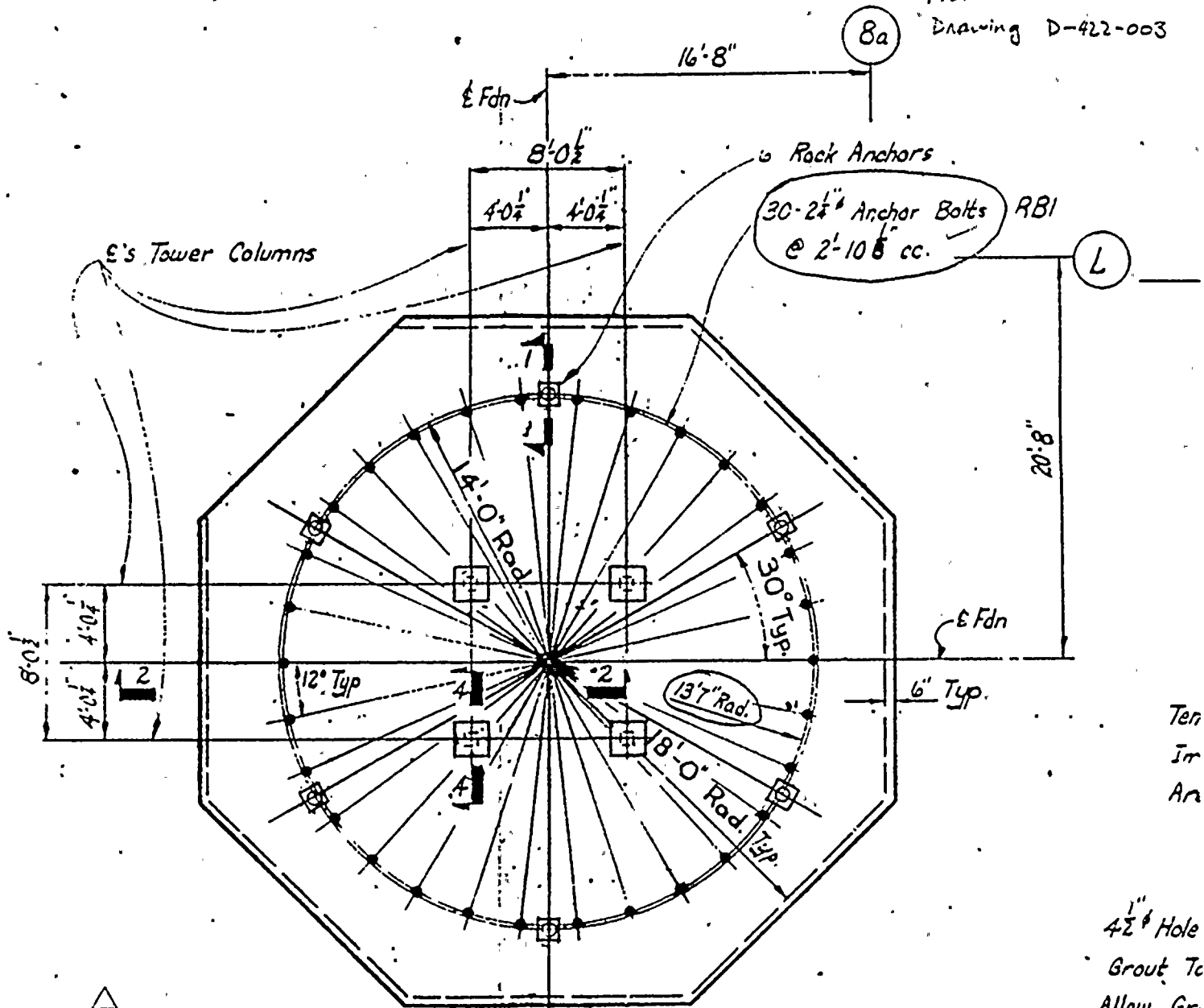
$$\text{Reduction of anchor head} = \pi(4.5)^2 = 63.6 \text{ in}^2$$

$$\begin{aligned} \text{Effective stress area for one bolt } A_e &= \pi(34.5)^2 - 2 \times 743.1 - 63.6 \\ &= 7189 \text{ in}^2 \end{aligned}$$

Bottom Mat Reinforcement

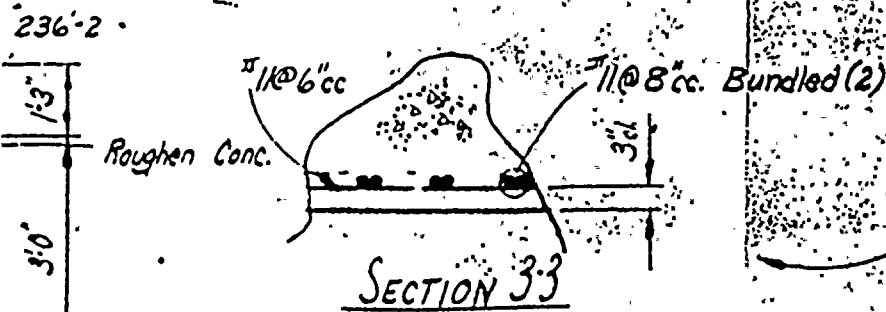
P.18/27

From Gilbert Associates
Drawing D-422-003



TANK FOUNDATION ANCHOR DETAIL

1" Hole
4 1/2" Grout
Allow Gr.
Stress Ter



Basement Floor
Datum Elev. 235'-8"

2'
2'
3'

1

2

3

4

5

6

7

8

9

10

11

12

13

14

15

16

17

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29

30

4' O.D. BBRV
32 Wire Tendon

Pipe & R
By Ceners

P. 19/27
From Gilbert Associates
Drawing D-422-003

6" Steel Pipe
Pipe & It Must Be In
Place When Concrete Is Poured

DETAIL 'A'

W.W.F. 6:6:90
(Bend Down over Dowels)
#5 Circular Bar

14'-0" (N.T.S.)

13'-7" (N.T.S.)

- Eley. 236-2 -

TANK

SECT

SECTION 2-2

8a

6.5 BENTON 18.40

*627866

45 e 18 c.c.

GILBERT ASSOCIATES, INC.

ATOMIC POWER DIVISION

CHICO VOLKKE

ENGINEERS AND CONSULTANTS

ROBERT CHAMMETT GINNA NUCLEAR POWER STATION UNIT NO. 1

50 of P.N. L. 1954

READING, PENNA. AND NEW YORK, N. Y.

R. G. & E. CORP., ROCHESTER, N. Y.

CP'DEN RW/EL/NMK

455-00.

5:429-005

ENCLOSURE

WORK ORDER

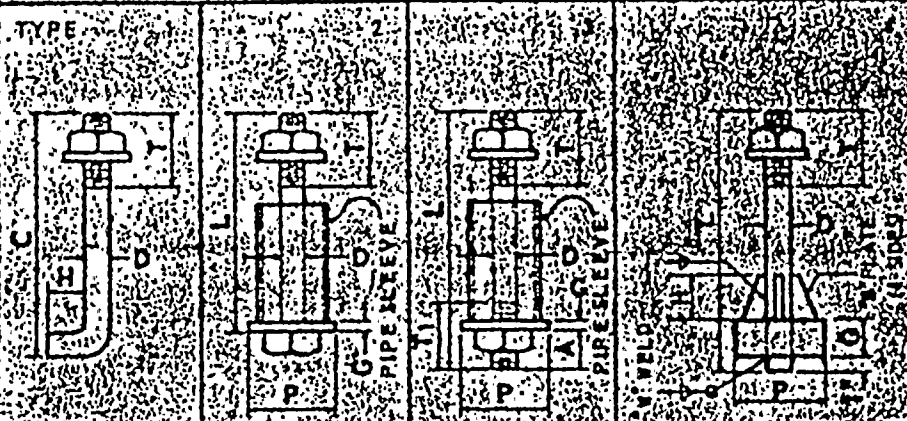
SIZE	6	DRYING	1	RECY.
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ANCHOR BOLT LIST: *Fuel Storage Tank* D: 422,003

REV. COL. ADP 510A-2

STANDARD TYPES

SPECIAL TYPES



RECEIVED
OCT 20 1966
R.G. & E. - ENG. DEPT.

[illegible]

DIMENSIONS ARE
IN INCHES UNLESS
OTHERWISE NOTED

NOTES:-



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SUBJECT RWST

JOB No. _____

SHEET 21 OF 27

Anchor Bolt

REVISIONS

0 TMT 8-14-84
WJ 9/18/84

The pull out strength, according to ACI 349-77, is

$$P_d = 4 \phi \sqrt{f'_c} A_e$$

Conservatively use $\phi = 0.65$ and $f'_c = 3000$ psi

$$\begin{aligned} P_d &= 4 \times 0.65 \times \sqrt{3000} \times 2189 \\ &= 312,000 / 16 = 19,500 \text{ kips} \end{aligned}$$

The ultimate capacity for yielding of the anchor bolt is

$$P_u = (3.25) (40.6) = 132 \text{ kips} < P_d$$

Refer to RWST calc, B3C2209 p. 45 and 57.

Anchor bolt yielding governs the design

$$P_d / P_u = 2.36 = \text{Factor of Safety}$$

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SUBJECT _____ JOB No. B3C2209 SHEET 22 OF 27

GINNA WASTE HOLDUP
TANK

REVISIONS	0	TMT 8-31-84
	WD	9/18/84

ALTERNATIVE ANALYSIS FOR THE WASTE HOLDUP TANK.

In the previous qualification report for the waste holdup tank (Ref. 4), the fundamental frequency in the transverse direction was calculated to be rigid. The review by SMA (Ref. 2) questioned the validity of the calculation and suggested that the fundamental frequency would occur at around half a hertz. Due to the complex geometry of the structure, the basic assumptions in the approximate analysis cannot be justified save for a full scale finite element analysis.

An alternative analysis procedure, similar to the equivalent static analysis, will be used in this section, where 1.5 times the peak of the spectra at 7% damping will be applied as the lateral load. Since the transverse direction is by far the less critical direction in the final stress evaluation, the waste holdup tank still satisfies the acceptance criteria under the conservative spectral assumption assumed.

Before pursuing the response calculation, the effect of sloshing is evaluated. The evaluation concludes that full tank is the most critical condition under the seismic input.



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SUBJECT

JOB No.

8302209

SHEET

23

OF

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GINNA WASTE HOLDUP
TANK

REVISIONS

0 TMT 8-31-84
WJ 9/21/84

SLOSHING EFFECTS

The sloshing frequencies in the transverse direction of the tank are calculated for three water levels, $1/4$, $1/2$, and $3/4$ water depth.

$$f_n = \frac{\lambda_n \sqrt{g}}{2\pi} \quad (\text{Hz})$$

λ_n (Frequency parameter from Ref. 3, p. 48)

0.75	1.4	0.54
0.50	1.15	0.44
0.25	1.05	0.40

where a = radius of tank = 66 inches

The corresponding spectral accelerations for the site specific response spectra are extrapolated as in the RWST calc. p. 34.

$h/2a$	S_a (5% damping)
0.75	$\exp(\ln 0.336 + (\ln 0.54 - \ln 2.5)(\ln 0.168 - \ln 0.336)/(-\ln 2.5)) = 0.105 g$
0.50	" 0.44 " 0.090
0.25	" 0.40 " 0.084

The factor C_T is also extrapolated

$h/2a$	C_T
0.75	$-\exp(\ln 1.81 + (\ln 0.54 - \ln 1.33)(\ln 1.96 - \ln 1.81)/(-\ln 1.33)) = -2.33$
0.50	" 0.44 " -2.46
0.25	" 0.40 " -2.53

The spectral acceleration for 0.5% damping are

$h/2a$	S_a (0.5%)
0.75	$0.105 \times 10^{-2.33(0.005-0.05)} = 0.134 g$
0.50	$0.090 \times 10^{-2.46(0.005-0.05)} = 0.116 g$
0.25	$0.084 \times 10^{-2.53(0.005-0.05)} = 0.109 g$



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JOB No. 83 C2209 SHEET 24 OF 27

GINNA WASTE HOLDUP
TANK

REVISIONS

0 TRIT B-21-PV
LD 9/18/82

It can be seen that the spectral accelerations at 0.5% damping corresponding to the sloshing modes are much less than either the peak of the spectra or ZPA at 7% damping. This concludes* that full tank is the critical condition in the seismic analysis for the horizontal waste hold-up tank under consideration.

$$\begin{aligned} \text{Peak of the spectra at 7\% damping} &= 0.397 \times 10^{-1.7(0.02)} \\ &= 0.37 g \end{aligned}$$

The same can be concluded for the longitudinal direction sloshing modes, the frequency of which are

h/L	γ_n (from Ref. 3) p. 48	$f_n = \gamma_n \sqrt{\frac{g}{L} \tanh \frac{\pi h}{L}} / 2\pi$
3/4	1.75	0.23
1/2	1.6	0.18
1/4	1.5	0.12

* As for overturning moment, the sloshing mass acting at the free surface tends to have larger moment arm. However, this effect does not invalidate the conclusion.



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SUBJECT _____ JOB No. 83C2209 SHEET 25 OF 27

GINNA WASTE HOLDUP TANK

REVISIONS	0	TMT 8-31-84
		WD 9/21/84

STRESSES DUE TO TRANSVERSE SEISMIC LOAD

- Maximum bending stress at horn of saddle (Ref. 4, calc. P. 23)

$$\sigma_b = 0.007 \times \frac{1.5 \times 0.37}{0.17} = 0.023 \text{ ksi}$$

- Maximum membrane stress (P. 23)

$$\sigma_m = 0.23 \times \frac{1.5 \times 0.37}{0.17} = 0.75 \text{ ksi}$$

- Anchor bolt stress (P. 26)

$$\sigma_f = 6.97 \times \frac{1.5 \times 0.37}{0.17} = 22.8 \text{ ksi}$$

combine with longitudinal response $\sigma_f = (35.6^2 + 22.8^2)^{1/2}$
 $= 42.3 \text{ ksi}$

- End stiffener axial compression (P. 30)

$$\sigma_c = 1.1 \times \frac{1.5 \times 0.37}{0.17} = 3.6 \text{ ksi}$$

Combined
 $\sigma_c = (3.6^2 + 0.8^2)^{1/2}$
 $= 3.7 \text{ ksi}$

- Welding between saddle plate and vessel (P. 31)

$$\tau_f = 1.4 \times \frac{1.5 \times 0.37}{0.17} = 4.6 \text{ ksi}$$

Combined stress $= (1.6^2 + 7.3^2 + 4.6^2)^{1/2} = 10.5 \text{ ksi}$



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SUBJECT

JOB No.

83C2209

SHEET

26 OF 27

GINNA WASTE HOLDUP
TANK

REVISIONS

0 TMT 8-31-84
WJB 9/17/84

SUMMARY TABLE OF SAFETY MARGINS (Ref. 4, Calc. p. 36)

LOCATION	Dead Load stress (ksi)	Vertical Seismic (ksi)	Horizontal Seismic (ksi)	Total	Allowable stress (ksi)
VESSEL SHELL					
• Mid-span					
σ_{xm}	0.04	0	0.02	0.06	35.6
• Plane of saddle					
σ_{xm}	1.9	0.2	0.75	2.7	
$\sigma_{x\phi}$	1.7	0.2	0	1.9	35.6
$\sigma_{\phi m}$	2.8	0.3	0.75	3.6	
$\sigma_{\phi b} + \sigma_{\phi m}$	5.6 + 2.8	6.2 + 0.3	0 + 0.75	65	18.0
Ring compression					
$\sigma_{\phi m}$	10.7	1.2	0	11.9	35.6
• Head					
Tension σ_{xm}	1.1	0.1	0	1.2	35.6
• Anchor					
Tension σ_t	18.9	2.1	42.3	23.5	40.6
• Saddle Support					
Tension in Flange	0.87	0.1	0	0.97	24.0
Flange bending	32	3.5	0	35.5	39.6
Web compression	1.4	0.2	0	1.6	45
Center stiffener	1.1	0.1	2.9	4.0	36
End stiffener	1.5	0.2	3.7	5.2	36
• Welding between Saddle and Vessel	0	0	10.5	10.5	36



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SUBJECT _____ JOB No. 83C2209 SHEET 27 OF 27

RWST NOZZLE LOADS

REVISIONS

0 TMT 8/31/84
WD 9/18/84

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