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Revision 1

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# **Westinghouse Performance Analysis and Design Model (PAD5)**



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Revision 1

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(PAD5)**

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## Revision 1 Changes

Note: These changes are identified throughout the document with vertical bars on the left margin.

No.	Section/Page(s)	Description/Reference
1.	3.5.3/p 3-15.	Enhanced description of variable $\delta$ in Eq. (3-30) for the thermal accommodation coefficient (LTR-NRC-16-5).
2.	3.5.3/p 3-15.	Specified minimum value in Eq. (3-31) for the thermal accommodation coefficient (LTR-NRC-16-5).
3.	3.5.3/p 3-15.	Corrected plots in Figure 3.5-1 for the thermal accommodation coefficient (LTR-NRC-16-5).
4.	4.2.1/pp 4-7 – 4-8.	Corrected units for time in Table 4.2-1 for the IFBA helium release model per RAI-6f (LTR-NRC-15-87).
5.	4.2.2/p 4-10.	Corrected units for time in Figure 4.2-2 for the IFBA helium release model per RAI-6f (LTR-NRC-15-87).
6.	5.4/p 5-8	Corrected units for strain rate in Eq. (5-20) for the cladding creep model per RAI-9b (LTR-NRC-16-5).
7.	5.5.1/pp 5-13 – 5-15.	Modified Figures 5.5-1 and 5.5-2 and removed Figure 5.5-3 for the cladding diametral growth model per RAI-13 (LTR-NRC-15-87).
8.	5.5.1/pp 5-12 – 5-13.	Updated value of $R_0$ for <b>Optimized ZIRLO</b> <sup>TM1</sup> cladding material and modified text for consistency with modification and removal of figures per RAI-13 (see item 7) (LTR-NRC-15-87).
9.	5.6/p 5-16.	Corrected units for LHGR in Eq. (5-30) for the pellet fragment relocation model per RAI-14b (LTR-NRC-16-5).
10.	5.7.1/pp 5-19 – 5-20.	Modified text to incorporate solid swelling model for gadolinia per RAI-15b (LTR-NRC-15-87).
11.	5.8/p 5-31.	Corrected unit for temperature in Eq. (5-39) for the fission gas swelling model per RAI-16a (LTR-NRC-15-69).
12.	6.1.5/p 6-9	Removed proprietary markings from Eqs. (6-14) through (6-16). No applicable previous reference.
13.	6.5/pp 6-21 – 6-22.	Modified Figures 6.5-2 for the cladding yield stress and 6.5-3 for the cladding ultimate tensile strength per RAI 18 (LTR-NRC-15-69).
14.	7/pp 7-1 – 7-115.	Complete re-write per RAI-32 (LTR-NRC-16-34).
15.	A.1.1/p A-2.	Reworded text about the use of calibration and validation datasets per RAI-30 (LTR-NRC-15-69).
16.	A.2.2.1/p A-31.	Corrected reference to test rod in Table A.2.2-2 per LTR-NRC-16-16.
17.	A.2.3.1/p A-65.	Added missing plant in Table A.2.3-1 per LTR-NRC-16-5.
18.	A.2.3.2/p A-69.	Added missing plant in Table A.2.3-2 per LTR-NRC-16-5.
19.	A.2.4.1/p A-92.	Corrected ramp holding times for three rods in Table A.2.4-4 per RAI-10 a-e (LTR-NRC-16-5).
20.	A.2.5.2/pp A-103 – A-106.	Corrected fuel type for several rods and deleted repeated rods in Table A.2.5-2 per RAI-12a (LTR-NRC-16-5).

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**LIST OF ACRONYMS AND TRADEMARKS**

AC	Alternating Current
ANC	Advanced Nodal Code
ANSI	American National Standards Institute
AO	Axial Offset
AOO	Anticipated Operational Occurrence
AOR	Analysis of Record
ASI	Axial Shape Index
ASME	American Society Mechanical Engineers
BE	Best Estimate
BOL	Beginning of Life
BWR	Boiling Water Reactor
CAOC	Constant Axial Offset Control
CE	Combustion Engineering
CEA	Commissariat à l'énergie atomique
CFFF	Columbia Fuel Fabrication Facility
CVCS	Chemical and Volume Control System
CWSR	Cold-Worked Stress Relieved
C+V	Calibration and Validation
DNB	Departure from Nucleate Boiling
DOE	Department of Energy
ECCS	Emergency Core Cooling System
EOC	End of Cycle
EOL	End of Life
EOS	Equation of State
EPRI	Electric Power Research Institute
FCEP	Fuel Criteria Evaluation Process
FCL	Fuel Centerline
FGR	Fission Gas Release
FSAR	Final Safety Analysis Report
GDC	General Design Criteria
HB	High Burnup
HBC	High Burnup Chemistry
HBEP	High Burnup Effects Program
HBWR	Halden Boiling Heavy-Water Reactor
ID	Inner Diameter
IFA	Instrumented Fuel Assembly
IFBA	Integral Fuel Burnable Absorber
LB	Lower Bound
LEU	Low-Enriched Uranium
LHGR	Linear Heat Generation Rate
LOCA	Loss of Coolant Accident
LTA	Lead Test Assembly
MAP	Maximum Allowable Power
MHI	Mitsubishi Heavy Industries

## LIST OF ACRONYMS AND TRADEMARKS (cont.)

MMIP	Models and Methods Improvement Process
NO	Normal Operation
NRC	Nuclear Regulatory Commission
NSSS	Nuclear Steam Supply System
OD	Outer Diameter
OECD	Organization for Economic Co-Operation and Development
PA	Postulated Accident
PAD	Performance Analysis and Design
PCI	Pellet Clad Interaction
PCMI	Pellet Clad Mechanical Interaction
PIE	Post Irradiation Examination
PNNL	Pacific Northwest National Laboratory
PRTR	Plutonium Recycle Test Reactor
RAOC	Relaxed Axial Offset Control
RCCA	Rod Cluster Control Assembly
RIP	Rod Internal Pressure
SCIP	Studsвик Clad Integrity Project
SG	Steam Generator
SRA	Stress Relieved Anneal
SRP	Standard Review Plan
SRSS	Square Root of Sum of Squares
TCD	Thermal Conductivity Degradation
T.D.	Theoretical Density
TRD	Thermal Reaction accumulated Duty
TRE	Thermal Reaction Energy
UB	Upper Bound
UE	Uniform Elongation
UTS	Ultimate Tensile Stress
VV	Void Volume

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# 1 INTRODUCTION AND SUMMARY

## 1.1 INTRODUCTION

The principal design tool for evaluating fuel rod performance is the Performance Analysis and Design (PAD) code. This computer program iteratively calculates the interrelated effects of fuel and cladding deformations including fuel densification, fuel swelling, fuel relocation, fuel rod temperatures, fission gas release (FGR), and rod internal pressure (RIP) as a function of time and linear power. PAD evaluates the power history of a fuel rod as a series of steady-state power levels with instantaneous jumps from one power level to another. The length of the fuel rod is divided into several axial segments, and each segment is assumed to operate at a constant set of conditions over its length. Fuel densification and swelling, cladding stresses and strains, temperatures, burnup and fission gas releases are calculated separately for each axial segment and the effects are integrated to obtain the overall fission gas release and resulting internal pressure for each time step. The coolant temperature rise along the fuel rod is calculated based on the flow rate and axial power distribution, and the cladding surface temperature is determined with consideration of corrosion effects and the possibility of local boiling.

The fuel pellet is modeled as a solid cylinder with allowances for dishing, edge chamfering and pellet chipping. For purposes of evaluating thermal expansion, fuel densification and swelling, and fission gas release, the fuel pellet is divided into ten equal-volume concentric rings with each ring assumed to be at its average temperature during a given time step. Axial and radial thermal expansion, swelling and densification are determined for each ring and these effects are integrated over the entire fuel rod to determine the length of the fuel column and evaluate the void volumes (VV) required to compute the internal gas pressure.

The PAD fuel performance models have evolved in several stages from the original application (References 1.1 through 1.9). This report provides a compilation of the licensing basis documentation for all of the significant PAD fuel and clad performance models including descriptions for new models. Where appropriate, model validation data are provided to demonstrate the acceptability of the individual fuel and clad performance models relative to available fuel performance data.

Fuel performance data has been obtained over the past five decades from a number of sources. Commercial reactors provide a substantial portion of the in-reactor fuel performance database, beginning with the examinations of the first Zircaloy-4 clad fuel in the Plant Z reactor in Spain, conducted in the late 1960's and through the 1970's. Data for Westinghouse Improved Zircaloy-4 cladding, and the Westinghouse ZIRLO<sup>®</sup> cladding material were obtained from commercial irradiations. Data acquisition continues to the present with the recent examinations of **Optimized ZIRLO<sup>™</sup>** cladding material fuel rods irradiated to very high burnup (HB). Data from these commercial fuel programs has been obtained from both on-site and hot cell examinations to characterize fuel and cladding behavior under normal operation conditions.

Fuel and cladding performance data from commercial irradiation programs are augmented by data obtained under test reactor conditions. Test reactor programs allow for the examination of fuel and cladding performance under off-nominal conditions, such as transient power increases or very high power operation. Test reactor irradiations also allow for additional instrumentation to measure critical parameters such as fuel temperature, internal pressure and cladding elongation. The PAD5 fuel performance database

includes measurement data from jointly sponsored industry programs such as the Halden Project, the Studsvik Clad Integrity Project (SCIP), and the INTERRAMP, SUPERRAMP and TRANSRAMP ramp test programs. The combination of both normal commercial operation and test reactor operation data produces a substantial basis for the calibration and validation of the PAD5 fuel and cladding performance models over the broad range of operating conditions addressed in fuel rod design. Data selected from the database for model Calibration and Validation (C+V) will be identified, as appropriate, for each PAD5 performance model.

## 1.2 SCOPE AND SUMMARY

The PAD5 models are the latest versions in the Westinghouse PAD code. Model updates incorporated into the PAD code address all of the fuel and cladding performance models required for high burnup fuel design. Key fuel performance updates to the PAD5 models include fuel Thermal Conductivity Degradation (TCD) with burnup, enhanced high burnup athermal fission gas release (pellet rim effects) and enhanced high burnup fission gas bubble swelling. Cladding creep and growth models are also updated to reflect high burnup cladding performance.

In addition to high burnup analysis capability, a key driver for the implementation of the PAD5 models in fuel design is to address regulatory concerns associated with fuel thermal conductivity degradation with burnup. It is intended that, following the Nuclear Regulatory Commission (NRC) review and approval, the PAD5 model will be applied on a forward fit basis to all domestic Westinghouse PWR fuel rod design analyses, including both Westinghouse and Combustion Engineering Nuclear Steam Supply System (CE NSSS) fuel.

A comprehensive description of all PAD5 models is provided in this topical. Some models are unaffected by increased fuel burnup and thermal conductivity degradation and remain the same as in the prior PAD 4.0 version, some models were adjusted to address high burnup and fuel thermal conductivity degradation, and some new models have been incorporated in the code. All PAD models are described in this topical, but the report clearly identifies models that are the same as in PAD 4.0, modified models and new models. Section 2 provides an overview of the PAD5 model changes.

The major performance models in PAD5 are discussed in Sections 3 through 6 of this report, as follows:

- Section 3 Thermal Model, including Thermal Conductivity Degradation
- Section 4 Gas Release and Internal Pressure Model
- Section 5 Fuel and Clad Deformation Models
- Section 6 Fuel and Clad Material Properties

Details of the PAD5 model calibration and validation are provided in Appendix A. Appendix B provides PAD5 model results for several sample cases using the PAD code, including both Westinghouse and CE NSSS fuel rod designs, with and without integral burnable absorbers (either  $\text{ZrB}_2$ ,  $\text{Gd}_2\text{O}_3\text{-UO}_2$ , or  $\text{Er}_2\text{O}_3\text{-UO}_2$ ).

To facilitate implementation of the PAD5 models, this topical combines the description of the PAD5 fuel and clad performance models with an assessment of the fuel rod design criteria to be addressed with the PAD code, including a description of the analysis methods for each criterion. Section 7 summarizes these



fuel rod design criteria and application methods which address the fuel rod design related aspects of the fuel system design as discussed in the US NRC Standard Review Plan (SRP), Section 4.2 (Reference 1.10). The design criteria presented in Section 7 will be applied to all Westinghouse PWR fuel rod designs, and they supersede criteria and evaluation methods previously approved for application to Westinghouse and CE NSSS fuel rod designs.

Section 8 describes a process by which the PAD5 fuel and clad performance models or methods may be improved and implemented in design. The process specifies criteria to be applied to model/model improvements such that, when satisfied, the improved model or method will be acceptable for implementation without requiring additional NRC review and approval. Any model or method improvement that does not satisfy the specified criteria will require explicit NRC review and approval prior to implementation.

In summary, this topical report requests NRC review and approval of:

1. The PAD5 fuel performance models for application to lead rod average burnup up to [ ]<sup>a,c</sup> for **Optimized ZIRLO** cladding, and 62 GWd/MTU for all ZIRLO and Zircaloy-4 cladding.
2. The fuel rod design criteria and associated PAD5 model evaluation methods for design application to rod average burnup up to [ ]<sup>a,c</sup>
3. The PAD5 Model and Method Improvement Process (MMIP) that will be applied to future PAD5 model and method changes to determine if these changes are acceptable for implementation without an explicit licensing submittal

### 1.3 SECTION 1 REFERENCES

- 1.1 NS-SL-521, NS-SL-524, and NS-SL-543, (Proprietary), and NS-SL-527, and NS-SL-544, (Non-Proprietary), Supplemental information on fuel design transmitted from R. Salvatori, Westinghouse, to D. Knuth, AEC, January 1973.
- 1.2 WCAP-8218-P-A, (Proprietary) and WCAP-8219-A, (Non-Proprietary), "Fuel Densification Experimental Results and Model for Reactor Application," March 1975.
- 1.3 WCAP-8720 (Proprietary), "Improved Analytical Models Used in Westinghouse Fuel Rod Design Computations," October 1976.
- 1.4 WCAP-8270 Addendum 2 (Proprietary), "Revised PAD Code Thermal Safety Model," October 1982.
- 1.5 WCAP-10851-P-A, (Proprietary) and WCAP-11873-A, (Non-Proprietary), "Improved Fuel Performance Models for Westinghouse Fuel Rod Design and Safety Evaluations," August, 1988.

- 1.6 WCAP-15063-P-A Revision 1, with Errata, (Proprietary) and WCAP-15064-NP-A Revision 1, with Errata, (Non-Proprietary), "Westinghouse Improved Performance Analysis and Design Model (PAD 4.0)," July 2000.
- 1.7 WCAP-12610-P-A (Proprietary), June 1990 and WCAP-14342-A (Non-Proprietary), "VANTAGE+ Fuel Assembly Reference Core Report," April 1995.
- 1.8 CENPD-404-P-A, (Proprietary), and CENPD-404-NP-A, (Non-Proprietary), "Implementation of ZIRLO™ Cladding Material in CE Nuclear Power Fuel Assembly Designs," November 2001.
- 1.9 WCAP-12610-P-A & CENPD-404-P-A, Addendum 1-A (Proprietary) and WCAP-14342-A & CENPD-404-NP-A, Addendum 1-A, (Non-Proprietary), "Optimized ZIRLO™," July 2006.
- 1.10 NUREG-0800, Standard Review Plan for the Review of Safety Analysis Reports for Nuclear Power Plants, Chapter 4.2, "Fuel System Design," March, 2007.

## 2 FUEL PERFORMANCE MODELS INTRODUCTION

### 2.1 BACKGROUND AND PURPOSE

The primary objectives for upgrading the PAD 4.0 Code (Reference 2.1) to PAD5 are to (a) improve and/or validate the fuel and clad performance models for application to high burnup, with a target lead rod average burnup of [ ]<sup>a,c</sup> GWd/MTU for fuel with **Optimized ZIRLO** cladding material and 62 GWd/MTU for Westinghouse PWR fuel with Zircaloy-4 or ZIRLO<sup>®</sup> High Performance Fuel Cladding Material and (b) to explicitly model fuel Thermal Conductivity Degradation (TCD) with burnup.

The PAD5 models describe fuel and clad behavior influenced by time, temperature, or burnup dependent phenomena for individual fuel rods. A fuel pin history is followed through a series of time increments during each of which the power is assumed to be constant. Independent radial thermal equilibrium calculations are performed for each discrete axial segment. Fuel burnup, fuel and clad temperatures, fuel fission gas release, and fuel and cladding deformations are calculated for each axial segment, with convergence based on fuel-cladding gap size for each axial segment. The converged results for each segment are coupled to those of other segments through the assumption of complete and instantaneous mixing of the free gases within the fuel rod. The coupling permits integrated, whole rod predictions of fuel rod fission gas release and internal pressure, with convergence based on the total moles of free gas.

This report is written such that the basis for all of the models in the code are justified and supported in a single document, including models that are not changed relative to prior code versions.

### 2.2 OVERVIEW OF MODEL CHANGES

Table 2.2-1 summarizes the PAD5 models discussed in detail in later sections of this report. Comments in the table highlight whether the models are the same as or changed relative to PAD 4.0.

Table 2.2-1 Fuel Performance Models in PAD5			
System	Section	Model	Comments
Thermal	3.1	Thermal Hydraulic	Same as PAD 4.0
	3.2	Film Temperature Drop	Same as PAD 4.0
	3.3	Clad Corrosion	Same as Reference 2.2
	3.4	Clad Temperature Drop	Same as PAD 4.0
	3.5	Gap Conductance	[ ] <sup>a,c</sup>
	3.6	Fuel Temperature Drop	[ ] <sup>a,c</sup> Radial power distribution within pellet updated to be consistent with PARAGON (Reference 2.3)
	3.7	Plenum Temperature	Same as PAD 4.0

**Table 2.2-1 Fuel performance models in PAD5 (cont.)**

System	Section	Model	Comments
Gas Release and Internal Pressure	4.1	Helium Solubility and Release	Same as PAD 4.0
	4.2	Helium Release (Zirconium Diboride)	Updated model using Halden test data
	4.3	Fission Gas Release	Updated athermal release model; Added high burnup athermal release model; Recalibrated thermal fission gas release model
	4.4	Equation of State (EOS)	Same as PAD 4.0
	4.5	Gas Absorption In Cladding Effect	Same as PAD 4.0
	4.6	Void Volumes and Internal Pressure	Updated crack volume and gap volume due to relocation model
Fuel and Clad Deformation	5.1	Clad Stresses	Added deviatoric stress for creep and plasticity analysis
	5.2	Clad Elastic Deformation	Same as PAD 4.0
	5.3	Clad Plasticity	[ $J^{a,c}$ ]
	5.4	Clad Creep	Revised irradiation creep model. Recalibrated thermal and irradiation creep model with addition of newer data
	5.5	Cladding Diametral Growth	Updated for ZIRLO and <b>Optimized ZIRLO</b> cladding
	5.6	Fuel Relocation	Updated to add explicit relocation model
	5.7	Fuel Swelling and Densification	Validated to data with different manufacture processes
	5.8	Fission Gas Bubble Swelling	Recalibrated PAD 4.0 model
	5.9	Rod Axial Growth	Updated with addition of newer data for ZIRLO and <b>Optimized ZIRLO</b> cladding
Materials Properties	6.1	Fuel: $UO_2/UO_2-Gd_2O_3/UO_2-Er_2O_3$	Updated fuel melting temperature [ $J^{a,c}$ ]  Updated Urania – Gadolinia thermal conductivity model; Added Urania- Erbium Properties; Other properties remain the same as PAD 4.0
	6.2	Zirconium Diboride Integral Fuel Burnable Absorber	Same as PAD 4.0, but updated operating experience

Table 2.2-1 Fuel performance models in PAD5 (cont.)			
System	Section	Model	Comments
Materials Properties (cont.)	6.3	Cladding: Zircaloy-4,/ZIRLO and / <b>Optimized ZIRLO</b>	Added irradiation hardening on yield strength and ultimate tensile strength; Other properties remain the same as PAD 4.0
	6.4	Fill Gas & Fission Gas Material Properties	Same as PAD 4.0

## 2.3 LIMITS OF APPLICABILITY

The ranges of applicability for PAD5 fuel rod design and safety analyses are defined by the aggregate of the calibration and validation data used in PAD5, summarized as follows:

- Pressurized water reactor designs using Low-Enriched Uranium (LEU) fuel loading
- Zircaloy-4 (including OPTIN, conventional Zircaloy-4, and improved Zircaloy-4) cladding, ZIRLO cladding, and **Optimized ZIRLO** cladding
- Uranium  $U^{235}$  enrichments up to [ ]<sup>a,c</sup>
- $ZrB_2$  fuel pellet coating
- Gadolinia concentrations up to [ ]<sup>a,c</sup>
- Erbium concentrations up to [ ]<sup>a,c</sup>
- Fuel grain sizes ranging from [ ]<sup>a,c</sup> to [ ]<sup>a,c</sup>
- Nominal true pellet density ranging from [ ]<sup>a,c</sup> to [ ]<sup>a,c</sup> of the theoretical density of  $UO_2$ .
- Rod average burnups up to 62 GWd/MTU for Zircaloy-4 and ZIRLO cladding and up to [ ]<sup>a,c</sup> for **Optimized ZIRLO** cladding
- Steady-state rod average linear heat generation up to [ ]<sup>a,c</sup> with local powers up to [ ]<sup>a,c</sup>
- Transient power and associated fuel centerline temperatures up to the melt temperature defined in subsection 6.1.5

The PAD5 models and application methods will be implemented on a forward fit basis with the fuel design criteria of Section 7 for all Westinghouse PWR fuel, including CE-NSSS applications that

formerly applied FATES3B fuel performance code models and methods. All PAD5 fuel performance models and the ZIRLO and **Optimized ZIRLO** cladding performance models are applicable to current Westinghouse fuel fabricated for CE-NSSS cores. PAD5 may also be used to perform analyses for legacy fuel fabricated with Zircaloy-4 clad, though it is not anticipated that fuel of this vintage would be further irradiated in current reload cores. The PAD5 Zircaloy-4 cladding models are directly applicable to legacy Westinghouse NSSS Zircaloy-4 clad fuel rods. For analysis of CE-NSSS fuel with legacy CE Zircaloy-4 or OPTIN cladding, PAD5 may be applied [  
] <sup>a,c</sup>

## 2.4 SECTION 2 REFERENCES

- 2.1 WCAP-15063-P-A Revision 1, with Errata, (Proprietary) and WCAP-15064-NP-A Revision 1, with Errata, (Non-Proprietary), "Westinghouse Improved Performance Analysis and Design Model (PAD 4.0)," July 2000.
- 2.2 WCAP-12610-P-A & CENPD-404-P-A, Addendum 2 "Westinghouse Clad Corrosion Model for ZIRLO™ and **Optimized ZIRLO™**," November 2008.
- 2.3 WCAP-16045-P-A (Proprietary) and WCAP-16045-NP-A (Non-Proprietary), "Qualification of the Two-Dimensional Transport Code PARAGON," August 2004.

### 3 THERMAL MODEL

Section 3 describes the PAD5 models for calculating the temperature distribution for the fuel rod, beginning with the coolant conditions surrounding the individual fuel rod and proceeding inward to the fuel centerline. Independent radial temperature calculations are performed at each axial segment in PAD5. The models described are:

- Coolant thermal-hydraulic model,
- Coolant-to-clad film temperature model, including crud,
- Clad corrosion models, including
  - The oxide-to-metal ratio
  - The clad corrosion temperature drop model
- Clad temperature model,
- Fuel-cladding gap conductance, including
  - Gap gas mixture thermal conductivities
  - Gap gas accommodation
  - Open gap conductance
  - Contact gap conductance,
- Fuel temperature model, including
  - Fuel thermal conductivity, and
- Fuel rod plenum temperature model

#### 3.1 THERMAL HYDRAULIC MODEL

*This model is the same as that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in Attachment P of NS-SL-521 (Reference 3.2) with additional information in Reference 3.9.*

The thermal hydraulic model calculates the bulk coolant temperature,  $T_{bulk}(z)$ , at any axial position:

$$\left[ \frac{q''(z)}{C_p G} \right]^{a,c} \quad (3-1)$$

where:

$q''(z)$  is the heat flux at axial elevation  $z$ ,

$C_p$  is the coolant heat capacity,

$G$  is the mass flow rate,

$De$  is the hydraulic diameter,

$T$  is the inlet water temperature,

$\left[ \right]^{a,c}$ .

Appropriate values for  $\left[ \right]$

$\left[ \right]^{a,c}$ . A typical value for  $\left[ \right]$  is  $\left[ \right]^{a,c}$ .

### 3.2 FILM TEMPERATURE DROP

*This model is the same as that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in NS-SL-521 (Reference 3.2), Attachment P.*

The surface temperature of the zirconium oxide layer is derived from the bulk coolant temperature with consideration of the following three modes of heat transfer:

1. Forced convection without boiling using the [  $h_{fc}$  ]<sup>a,c</sup> and accounting for conduction heat transfer across the crud layer, if present.
2. Nucleate boiling from an essentially clean surface using the [  $h_{nb}$  ]<sup>a,c</sup>.
3. Nucleate boiling through a crud layer using the [  $h_{nbl}$  ]<sup>a,c</sup>.

The mode of heat transfer is a function primarily of the heat flux and, if assumed to be present, the crud thickness, and to a lesser extent, of fluid and system conditions. For the ZIRLO cladding and **Optimized ZIRLO** cladding corrosion models (References 3.7 and 3.8) [  $q''$  ]<sup>a,c</sup>

The clad oxide surface temperature under forced convection heat transfer,  $T_C$ , is given by:

$$T_C = T_{bulk} + \frac{q''}{h_{fc}} + \frac{q'' \delta_{crud}}{K_f} \quad (3-2)$$

where:

- $q''$  is the heat flux (BTU/hr-ft<sup>2</sup>),
- $T_{bulk}$  is the bulk coolant temperature (°F),
- $h_{film} =$  the [  $h_{fc}$  ]<sup>a,c</sup> (BTU/hr-ft<sup>2</sup>-°F),
- $K_f$  is the crud conductivity under forced convection (BTU/hr-ft-°F),
- $\delta_{crud}$  is the crud thickness (ft).

If  $T_C$  is less than the saturation temperature,  $T_{sat}$ , the calculation would proceed to the consideration of the temperature rise through the oxide layer. Otherwise, the film temperature calculation considers the impact of subcooled nucleate boiling.

For typical calculations involving ZIRLO and **Optimized ZIRLO** cladding, [  $q''$  ]<sup>a,c</sup>, as given by:



$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c} \quad (3-3)$$

where:

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c}$$

and,

$T_{sat}$  is the coolant saturation temperature (°F),  
 $P$  is the system pressure (psi).

The clad surface temperature using the  $\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c}$  heat transfer becomes  
 (References 3.9 and 3.10):

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c} \quad (3-4)$$

where:

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c}$$

If crud is explicitly modeled, then the  $\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c}$  is also calculated using:

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c} \quad (3-5)$$

where:

$K_b$  is the effective crud conductivity in nucleate boiling (BTU /hr-ft-°F)  
 $\delta_{crud}$  is the crud thickness (ft.)

The temperature at the clad oxide surface under nucleate boiling heat transfer is:

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c} \quad (3-6)$$

### 3.2.1 Crud Model

*This model is the same as that used in PAD 4.0 (Reference 3.1).*

$$\left[ \frac{h_{a,c}}{k_{crud}} \right]^{a,c}$$

[  
] <sup>a,c</sup>

The crud formation model in PAD provides a means to [  
] <sup>a,c</sup>

The crud thickness model allows for [  
] <sup>a,c</sup> as shown below:

$$[ \quad ]^{a,c} \quad (3-7)$$

where:

$\delta_{crud}$  is the total crud thickness (mils)

[  
] <sup>a,c</sup> is an input constant crud thickness (mils) assumed to be formed under [  
] <sup>a,c</sup>

[  
] <sup>a,c</sup> is the crud thickness (mils) assumed to be formed under [  
] <sup>a,c</sup> conditions.

$$[ \quad ]^{a,c} \quad (3-8)$$

where:

$$[ \quad ]^{a,c} \quad (3-9)$$

$q''_{boil}$  is the fractional heat transfer through the coolant film layer due to [  
] <sup>a,c</sup>

$q''$  is the total heat transfer

$A_{crud}$  and  $B_{crud}$  are input constants.

### 3.3 CLAD CORROSION MODELS

#### 3.3.1 ZIRLO Cladding and *Optimized ZIRLO* Cladding

*The ZIRLO and **Optimized ZIRLO** cladding corrosion models have been approved by the NRC (References 3.7 and 3.8) and are incorporated into the PAD code.*

The corrosion model form consists of a Thermal Reaction Accumulated Duty (TRD) term which is an [  
] <sup>a,c</sup>

$$\left[ \frac{a,c}{\dots} \right] \quad (3-10)$$

where:

$$\left[ \dots \right]^{a,c} \quad (3-11)$$

where:

$$\left[ \dots \right]^{a,c} \quad (3-12)$$

where:

$$\left[ \dots \right]^{a,c} \quad (3-13)$$

where:

$T$  is the total time duration in hours.

The ZIRLO cladding best estimate oxide thickness,  $CR_{ZIRLO}$ , is calculated with the expression:

$$\left[ \dots \right]^{a,c} \quad (3-14)$$

The coefficients  $A$  and  $B$  are model constants.

The **Optimized ZIRLO** cladding best estimate oxide thickness,  $CR_{OPTZIRLO}$  is calculated from the  $\left[ \dots \right]^{a,c}$  as follows:

$$\left[ \dots \right]^{a,c} \quad (3-15)$$

Validation of the ZIRLO and **Optimized ZIRLO** cladding corrosion models was presented in detail in Reference 3.7 and approved in Reference 3.8. Figures 3.3-1 and 3.3-2 (Reference 3.9) illustrate the measured oxide thickness data as a function of TRD for ZIRLO and **Optimized ZIRLO** cladding, respectively, with the upper 95% uncertainty curve.



**Figure 3.3-1      ZIRLO Cladding Material Measured Oxide Thickness Versus TRD With Upper 95% Uncertainty Curve (Reference 3.9)**

**Figure 3.3-2      *Optimized ZIRLO Cladding Material Measured Oxide Thickness Versus TRD With Upper 95% Uncertainty Curve (Reference 3.9)***

### 3.3.2 Zircaloy-4 Cladding

*The following model was in PAD 4.0 (Reference 3.1) for application to Zircaloy-4 cladding. This model was presented to the NRC in Reference 3.11. Current fuel does not employ Zircaloy-4 cladding. This model is used as required to analyze Zircaloy-4 clad fuel designs. It is applicable for rod average burnup up to [ ]<sup>a,c</sup>. Zircaloy-4 clad fuel will not be used in design at burnups greater than this value.*

The Zircaloy-4 cladding corrosion model in PAD describes the in-pile corrosion of Zircaloy-4 fuel cladding in terms of weight gain rate,  $dw/dt$ , as follows:

$$\left[ \right]^{a,c} \quad (3-16)$$

where:

$$\left[ \right]^{a,c} \quad (3-17)$$

is the out-of-reactor thermal corrosion rate, in (mg/ dm<sup>2</sup>)/ day,

$T_K$  is the oxide surface temperature in kelvins, and:

$$F_I = \left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c}$$

and the in-reactor corrosion acceleration factors are:

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c}$$

where:

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c}$$

Figure 3.3-3 shows the measured versus predicted oxide comparisons for the Zircaloy-4 corrosion model applied to Westinghouse conventional and improved Zircaloy-4. Figure 3.3-4 provides the measured minus predicted versus predicted oxide comparisons for Westinghouse conventional and improved Zircaloy-4. The 95% uncertainty bounds are seen in Figure 3.3-4.

a,b,c

**Figure 3.3-3      Measured Versus Predicted Oxide Thickness – Westinghouse Zircaloy-4 Model**

**Figure 3.3-4 Measured Minus Predicted Oxide Thickness – Westinghouse Zircaloy-4 Model**

### 3.3.3 Oxide-to-Metal Ratio

*This model is the same as presented in PAD 4.0 (Reference 3.1) and in the recently licensed ZIRLO and Optimized ZIRLO cladding corrosion model Topical (References 3.7 and 3.8).*

Due to the differences in densities of the oxide and the base metal, there is volumetric change from the metal consumed to the oxide formation. This volumetric difference results in a thicker oxide than the equivalent thickness of the base metal that was consumed. The ratio of the volumes is characterized by the oxide-to-metal ratio (O/M). The theoretical O/M ratio,  $O/M_{th}$ , is referred to as the Pilling-Bedworth ratio, and for zirconium-based alloys, the value of 1.56 is commonly used. However, during the in-reactor generation of  $ZrO_2$ , different mechanisms occur that cause the oxide density to be less than theoretical resulting in higher O/M ratios at increasing oxide thickness. As the oxide grows, it transitions from a protective to a non-protective structure. The non-protective oxide contains cracks and pores and this transition occurs when the oxide is about [ ]<sup>a,c</sup>.



The equations governing the O/M ratio as a function of oxide thickness are as follows:

$$\left[ \frac{\delta_{ZrO_2}}{O/M} \right] = \frac{a_c}{a_o} \quad (3-18a)$$

$$\left[ \frac{\delta_{ZrO_2}}{O/M} \right] = \frac{a_c}{a_o} \quad (3-18b)$$

where:

$\delta_{ZrO_2}$  is the oxide thickness (mils).

The cladding base metal wall thickness corrected for oxide,  $t'_{wall}$ , is then:

$$t'_{wall} = t_{wall} - \frac{\delta_{ZrO_2}}{O/M} \quad (3-19)$$

where:

$t_{wall}$  is the initial clad wall thickness.

### 3.3.4 ZrO<sub>2</sub> Temperature Drop Model

*This model is the same as that used in PAD 4.0 (Reference 3.1) and prior PAD versions (References 3.2, 3.12, and 3.14).*

The temperature drop across the zirconium oxide layer,  $\delta_{ZrO_2}$ , is given by:

$$\left[ \frac{\delta_{ZrO_2}}{K_{ZrO_2}} \right] = \frac{a_c}{a_o} \quad (3-20)$$

where:

$K_{ZrO_2}$  is the zirconium oxide thermal conductivity (BTU/hr-ft-°F) (see subsection 6.3.2) and  $\delta_{ZrO_2}$  is the zirconium oxide thickness (ft).

The temperature at the clad metal-to-oxide interface,  $T_{clad}^{m/o}$ , is then:

$$T_{clad}^{m/o}(z) = T_C(z) + \Delta T_{ZrO_2} \quad (3-21)$$

where  $T_C$  is the temperature at the oxide surface.

### 3.4 CLAD TEMPERATURE DROP MODEL

*This model is the same as presented in PAD 4.0 (Reference 3.1) and prior PAD versions (References 3.2, 3.12, and 3.14.*

When the thermal conductivity of the cladding is a [ ]<sup>a,c</sup> (Reference 3.1 and Subsection 6.3.2):

$$[ ]^{a,c} \quad (3-22)$$

then the heat conduction equation can be solved for the cladding inner surface temperature:

$$\left[ \right]^{a,c} \quad (3-23)$$

where:

$R_o$  is the clad outer radius,

$R_i$  is the clad inner radius,

$T_{clad}^{m/o}$  is the metal-oxide interface temperature from Equation (3-21), and

$TCI$  is the cladding inner surface temperature.

For the calculations of cladding creep rates and thermal expansion, the cladding average temperature is used. This is found by numerically calculating the volumetric average of the cladding temperature.

### 3.5 GAP CONDUCTANCE MODEL

*This model is modified from that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in Attachment F of NS-SL-521 (Reference 3.2) and Reference 3.12.*

The temperature drop across the pellet to clad gap is given by:

$$[ ]^{a,c} \quad (3-24)$$

The gap conductance,  $h_{gap}$ , contained in the fuel rod design model is dependent on the [ ]<sup>a,c</sup>.

#### 3.5.1 Gap Conductance Used for Finite Gaps

The PAD5 gap conductance model for an open gap is given by:

$$[ ]^{a,c} \quad (3-25)$$

where:

$h_{gap}$  is the gap conductance (BTU/hr-ft<sup>2</sup>-°F),  
 $K_{mix}$  is the effective gas thermal conductivity (BTU/hr-ft-°F),  
 $GAP$  is the diametral gap (in),  
 $\delta_r$  is the effective surface roughness (ft).

The gap conductance calculated with Equation (3-33) is used to calculate the temperature drop across the annular gap. The fuel-clad gap is determined by first radially deforming both the fuel and cladding. The cladding is radially deformed considering thermal expansion (subsection 6.3.3), elastic deformation in response to the rod internal pressure and coolant pressure (Section 5.2), creep (Section 5.4), clad diametral growth (Section 5.5), and plastic deformation (Section 5.3). The fuel is radially deformed considering thermal expansion (subsection 6.1.3), relocation (Section 5.6), fuel densification and swelling (Section 5.7), and high temperature fission gas bubble swelling (Section 5.8).

### 3.5.2 Gas Mixture Thermal Conductivity Model

*This model is the same as that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in Attachment F of Reference 3.2.*

[  
<sup>a,c</sup> The relationship for calculating the thermal conductivity of a monatomic gas mixture is based on the work of [  
<sup>a,c</sup>]

$$\left[ \sum_{j=1}^n \left( \frac{K_j}{M_j} \right) \right]^{a,c} \quad (3-26)$$

where the case  $i=j$  is excluded from the sum over  $j$  (innermost sum) and:

$$\left[ \sum_{j=1}^n \left( \frac{K_j}{M_j} \right) \right]^{a,c} \quad (3-27)$$

$$\left[ \sum_{j=1}^n \left( \frac{K_j}{M_j} \right) \right]^{a,c} \quad (3-28)$$

and

$n$  is the number of component gases in the mixture,  
 $M$  represents the molecular weight,  
 $x$  is the mole fraction,  
 $K$  stands for the thermal conductivity (BTU/hr-ft-°F).

The correlations used for the thermal conductivities of the individual gases in the pellet-cladding gap are described in Section 6.4.

In calculating the gas mixture thermal conductivity for the gas occupying the pellet-cladding gap, the average temperature between the pellet surface and the cladding inner surface is employed.

### 3.5.3 Accommodation Coefficient

*This model is the same as that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in Reference 3.12.*

When the ratio of the mean free path of the gas molecules to the characteristic dimension (gap) of the body exceeds [ ]<sup>a,c</sup>. Heat transfer from a solid to a gas in this, [ ]<sup>a,c</sup> is then dependent on an accommodation coefficient which is defined as the ratio of the actual energy interchange to the maximum possible energy interchange between a surface and a gas. Generally the accommodation coefficient factor of a heavy gas such as argon or xenon is near one and this effect can be neglected. But the effect cannot be neglected when the gap is filled with a light gas such as helium. For helium, PAD calculates the following factor by which the bulk thermal conductivity of helium should be divided when calculating temperature drops in [ ]<sup>a,c</sup>:

$$f_{\alpha} = 1 + \left( \frac{\alpha_1 + \alpha_2 - \alpha_1 \alpha_2}{\alpha_1 \alpha_2} \right) \left( \frac{4}{C_p + C_v} \right) (K_{He} / \mu) (\lambda / \delta) \quad (3-29)$$

where:

- $\alpha_f$  is the accommodation coefficient of the fuel,
- $\alpha_c$  is the accommodation coefficient of the cladding,
- $C_p$  is the constant pressure heat capacity,
- $C_v$  is the constant volume heat capacity,
- $K_{He}$  is the gas thermal conductivity,
- $\mu$  is the gas viscosity,
- $\lambda$  is the mean free path,
- $\delta$  is the characteristic dimension (e.g., radial gap).

Assuming that [ ]<sup>a,c</sup> and utilizing relationships from the [ ]<sup>a,c</sup> it can be shown that Equation (3-29) reduces to:

$$f = 1 + \frac{\left[ 1.6787 \cdot 10^{-5} \left( \frac{2-A}{A} \right) K_{He} T^{1/2} \right]}{P_f \delta} \quad (3-30)$$

where:

$K$  is in BTU/hr-ft-°F  
 $T$  is the gap gas temperature in degrees Rankine,  
 $P_f$  is the absolute gas pressure in psi,  
 $\delta$  is the radial gap in feet [ ]<sup>a, c</sup>

and where  $A$ , the accommodation coefficient, is a function of pressure (Reference 3.17), as shown in Figure 3.5-1, and is given by:

$$[ ]^{a,c} \quad (3-31)$$

The effective thermal conductivity of helium used in Equation (3-26) accounts for the accommodation factor, so that:

$$K_{eff} = K_{He} / f \quad (3-32)$$

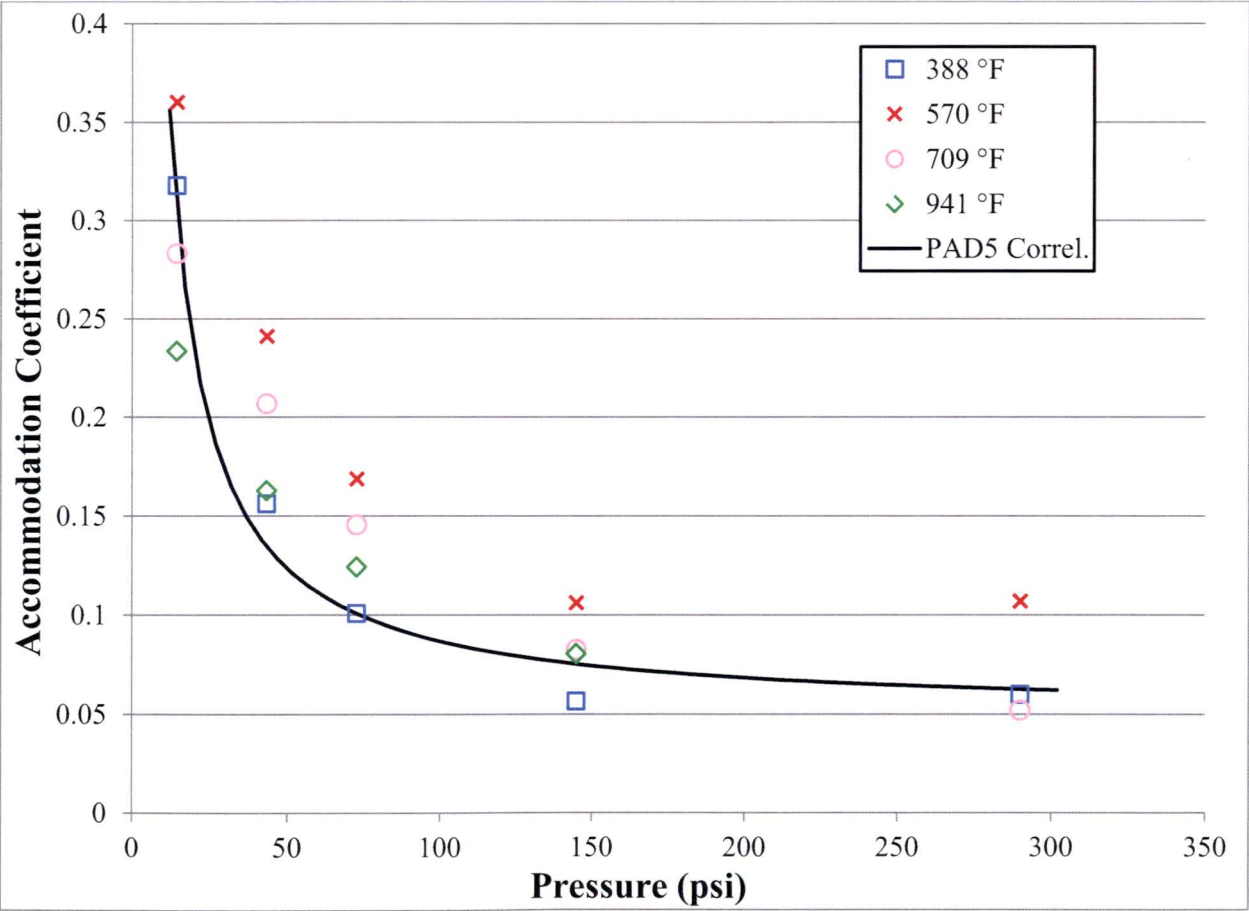


Figure 3.5-1 Variation in Accommodation Coefficient with Pressure (Reference 3.17)

### 3.5.4 Gap Conductance Used for Pellet-Clad Contact

*This model is the same as that used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in Reference 3.12.*

When the pellet is in contact with the cladding, the PAD5 gap (contact) conductance is determined by:

$$\left[ \frac{h_{gap}}{P_c} \right]^{a,c} \quad (3-33)$$

where:

$h_{gap}$  is the contact conductance (BTU/hr-ft<sup>2</sup>-°F),  
 $P_c$  is the contact pressure (psi), [ $\quad$ ]<sup>a,c</sup>  
 $K_{mix}$  is the thermal conductivity of the gas mixture (BTU/hr-ft-°F),  
 $\delta_r$  is the effective surface roughness (ft).

If the deformed cladding inner diameter is predicted to be radially inside of the deformed fuel diameter in a particular axial segment, i.e., an interference fuel-cladding gap, the fuel-cladding contact pressure is calculated based on the interference gap size assuming both the fuel and cladding are in the elastic state (see subsection 5.1.1).

Equation (3-33) is based on the work of [ $\quad$ ]<sup>a,c</sup> and has been selected as a mean value between the published curves. The contact conductance described by Equation (3-33) is shown in Figure 3.5-2, and is compared to the published work. The adequateness of this selection is confirmed by the ability of the overall model to correlate the data.

a,c

**Figure 3.5-2      Contact Conductance for  $\text{UO}_2$ -Zr at 500°F in Argon**

### 3.6 FUEL TEMPERATURE DROP MODEL

*This model is the same as that used in PAD 4.0 (References 3.1) and previous PAD versions (References 3.12, and 3.14). A brief description of this model is included below.*

The fuel temperature profile across a fuel pellet, from the fuel surface to the fuel centerline, is calculated by solving the heat conduction equation for a cylindrical pellet (assuming radial heat flow) using [ ]<sup>a,c</sup> method:

$$\left[ \frac{1}{r'} \frac{d}{dr'} \left( K(T) r' \frac{dT}{dr'} \right) + q'''(r') \right] = 0 \quad (3-34)$$

where:

$K(T)$  is the fuel thermal conductivity, assumed to depend on temperature and burnup,  
 $T_S$  is the temperature at the pellet surface,  
 $T(r)$  is the pellet temperature at radial position  $r$ ,  
 $R_S$  is the pellet radius at the outer surface,  
 $q''(r')$  is the heat flux at radial position  $r'$ .

The heat flux is calculated as:

$$q''(r') = \frac{1}{r'} \int_{R_{IN}}^{r'} q'''(r'') r'' dr'' \quad (3-35)$$

where:

$q'''(r'')$  is the volumetric power density at pellet radius  $r''$ .

In PAD, the fuel pellet is divided into ten equal-volume radial rings. The volumetric power density in each ring is given by the pellet radial power distributions discussed in subsection 3.6.2.

The PAD code assumes that the burnup, fuel density, and volumetric power density in each radial ring are uniform over the ring. Using Equations (3-34) and (3-35), the value of [ ]<sup>a,c</sup> is calculated for each ring. The [ ]<sup>a,c</sup> correlation for the fuel thermal conductivity presented in subsection 6.1.2 is then solved for the temperature at each ring boundary.

#### 3.6.1 Fuel Thermal Conductivity

*This model is revised for PAD5. Original documentation of this model is located in Reference 3.14.*

The fuel thermal conductivity model should account for the dependence of thermal conductivity on local fuel temperature and local burnup, as well as on the burnable absorber concentration [ ]<sup>a,c</sup>. The model used in PAD5 has been updated to explicitly account for fuel thermal conductivity degradation with burnup and is described in subsection 6.1.2.



The fuel thermal conductivity model can be qualified directly, based on thermal conductivity data from laser flash thermal diffusivity measurements, and indirectly, based on measured fuel temperatures. The validation against measured fuel temperatures is presented in the thermal calibration results in Appendix A.

Figures 3.6-1 through 3.6-3 show plots of the fuel thermal conductivity of  $\text{UO}_2$ , gadolinia, and erbia fuel, respectively, as a function of temperature for various burnups. The plots have been extended to the melting point. The correlations for fuel thermal conductivity and fuel melting point from subsections 6.1.2 and 6.1.5, respectively, were used.

a,c

**Figure 3.6-1      Thermal Conductivity for  $\text{UO}_2$  Fuel at Various Burnups**

a,c

**Figure 3.6-2 Thermal Conductivity for 8% Gadolinia Fuel at Various Burnups**

**Figure 3.6-3 Thermal Conductivity for 2.5% Erbia Fuel at Various Burnups**

### 3.6.2 Pellet Radial Power Distribution Model

*This model is revised for PAD5.*

The radial power distribution (and the resulting radial power depression factor) in the PAD code is a function of [ ]<sup>a,c</sup>. Local distributions are determined for each axial segment by a parabolic interpolation to data which are tabulated as a function of [ ]<sup>a,c</sup>. It has been shown that these distributions [ ]<sup>a,c</sup>.

Radial power distributions can be accurately calculated by PARAGON (Reference 3.13) for various fuel types (UO<sub>2</sub>, erbia, gadolinia, and ZrB<sub>2</sub>-coated fuel), at various burnups, and for different reactor types.

Figures 3.6-4 through 3.6-7 show sample radial power distributions used in PAD5. Figure 3.6-4 displays radial power distributions for UO<sub>2</sub> fuel with 4.5% enrichment. Figure 3.6-5 presents graphs for radial power distributions for ZrB<sub>2</sub>-coated fuel with 3% uranium enrichment. For this graph the focus is on

lower burnups, as the distributions approach those of pure  $\text{UO}_2$  fuel at burnups starting from about 15 GWd/MTU. Figure 3.6-6 shows radial power distribution for 2.5% erbia fuel of 4% uranium enrichment, and Figure 3.6-7 shows radial power distributions for 8.1% gadolinia fuel of 4.5% uranium enrichment. PAD divides the pellet into ten equal-volume radial rings. In the graphs shown in Figures 3.6-4 through 3.6-7, ring number one corresponds to the innermost ring and ring number ten corresponds to the outermost ring.

a,c

**Figure 3.6-4      Radial Power Distributions for 4.5% Enriched  $\text{UO}_2$  Fuel at Various Burnups**

a,c

**Figure 3.6-5      Radial Power Distributions for  $\text{ZrB}_2$ -Coated 3% Uranium Enriched Fuel for Various Burnups**

a,c

**Figure 3.6-6      Radial Power Distributions for 2.5% Erbia Fuel with 4% Uranium Enrichment at Various Burnups**

**Figure 3.6-7 Radial Power Distributions for 8.1% Gadolinia Fuel with 4.5% Uranium Enrichment at Various Burnups**

### **3.7 PLENUM TEMPERATURE MODEL**

*This model is the same as what was used in PAD 4.0 (Reference 3.1). Original documentation of this model is located in NS-SL-521 (Reference 3.2), Attachment M.*

The plenum gas temperature calculation is based on the data obtained from two in-pile experiments. The first experiment was performed in the Organization for Economic Co-Operation and Development (OECD) Halden Boiling Heavy Water Reactor in Norway during the second half of 1968. The second experiment was performed in the Plutonium Recycle Test Reactor (PRTR) during several months of operation as part of the Plutonium Utilization Program at Pacific Northwest National Laboratory (PNNL). For both experiments, both internal gas pressure and plenum gas temperature were recorded during the irradiation period.

The plenum geometries, the axial heat flux distributions and the fuel rod characteristics were significantly different in the two experiments. However, it was possible to derive from these data a very simple relationship between the gas plenum temperature, the coolant temperature, and the fuel rod operating power. This relationship reveals that the plenum gas temperature is directly related to the surrounding coolant temperature and is a weak function of the fuel rod power. Although the plenum geometries in these two experiments are somewhat different from our reference fuel rod plenum geometry, this simple

model can be used to predict the operating plenum gas temperature in the Westinghouse fuel rods to a sufficient degree of accuracy.

The resulting equation is:

$$T_{plenum} = T_{coolant} + \left[ \frac{\quad}{\quad} \right]^{a,c} \quad (3-36)$$

where:

$T_{plenum}$  is the average temperature of the gas space in the plenum of a PWR fuel rod (°F),

$T_{coolant}$  is the rod coolant channel outlet temperature. (°F),

[  
] <sup>a,c</sup>

### 3.8 SECTION 3 REFERENCES

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- 3.7 WCAP-12610-P-A & CENPD-404-P-A, Addendum 2 "Westinghouse Clad Corrosion Model for ZIRLO™ and Optimized ZIRLO™," November 2008.
- 3.8 Letter from Sher Bahadur (US NRC) to James Gresham (Westinghouse), "Final Safety Evaluation for Westinghouse Electric Company (Westinghouse) Topical Report WCAP-12610-P-A & CENPD-404-P-A, Addendum 2/WCAP-14342-A & CENPD-404-NP-A, Addendum 2, 'Westinghouse Clad Corrosion Model for ZIRLO™ AND Optimized ZIRLO™' (TAC Nos. ME0222 and ME2317)," July 18, 2013.



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- 3.10 LTR-NRC-10-27," Responses to the NRC's Request for Additional Information RE: Westinghouse Electric Company Topical Report WCAP-12610-P-A & CENPD-404-P-A, Addendum 2 / WCAP-14342-A & CENPD-404-NP-A, Addendum 2, 'Westinghouse Clad Corrosion Model For ZIRLO™ and **Optimized ZIRLO™**' (TAC No. ME0222) (Proprietary/Non-Proprietary)," April 30, 2010.
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## 4 GAS RELEASE AND INTERNAL PRESSURE

Section 4 describes the PAD5:

- Fuel helium solubility model,
- $ZrB_2$  helium release model,
- Fission gas release model,
- Free gas equation of state,
- Cladding gas absorption, and
- Open void volume and rod internal pressure models.

### 4.1 HELIUM SOLUBILITY AND RELEASE

*This model is the same as in PAD 4.0 (Reference 4.1). Original documentation of this model is located in PAD 3.3 (Reference 4.2).*

Following initial manufacturing, the fuel pellets in fuel rods containing pressurized helium are assumed to absorb a portion of the helium atmosphere within a short period of time. During operation an additional 0.3 atoms of helium are produced for every 100 fissions and a portion of the produced helium is released.

Review of published literature indicates that dissolution of helium into  $UO_2$  occurs to a small extent. Westinghouse has carried out tests to determine the extent of helium dissolution in  $UO_2$  during irradiation in helium pressurized fuel rods. Knowledge of the extent of dissolution of helium is important in calculating fuel rod internal pressures.

Data from 90% Theoretical Density (T.D.) and 95% T.D. fuel irradiated in the Plant AA reactor to burnups between 1000 and 5000 MWd/MTU (Table 4.1-1) and higher burnup data for fuel between 91.5% T.D. and 96.0% T.D. (Table 4.1-2) have been fit with a standard multiple linear regression equation to obtain the helium solubility equation used in the PAD5 code:

$$S = [ \text{ } ]^{a,c} \quad (4-1)$$

where:

S	is the helium solubility, (cc (STP)/gm $UO_2$ )
$\rho_0$	is the initial fuel density, (% T.D.)
$P_0$	is the initial gas pressure, (psia)
BU	is the fuel burnup, (MWd/MTU)

Table 4.1-3 compares the predicted and measured solubility data using this equation.

The use of the helium content (solubility) equation in design consists essentially of the following steps.

- The extent of helium dissolution is calculated for the initial fuel density assuming dissolution to be complete at the beginning of irradiation.
- The initial backfill pressure is calculated to be consistent with the loss of helium due to solution. The internal gas pressure (P) is then given by:

$$P = P_O - \Delta P_S \quad (4-2)$$

where:

$P_O$  is the initial fill pressure

$\Delta P_S$  is the pressure reduction due to the initial absorption of helium.

- At any point during irradiation the helium content in the fuel is re-evaluated using Equation (4-1) and assuming that helium produced by fissioning is 0.30 atoms/100 fissions. Thus at any time:

$$H_R = S_O + H_p - S(t) \quad (4-3)$$

and the fractional release (relative to the total) is:

$$F_H = \frac{H_R}{S_O + H_p} \quad (4-4)$$

where:

$S(t)$  is the helium content at any time

$S_O$  is the initial helium absorbed into the fuel

$H_R$  is the helium release during irradiation

$H_p$  is the helium produced by fissioning.

Figure 4.1-1 illustrates how the helium content varies as a function of density and burnup.

The helium which has been released from the fuel ( $H_R$ ) increases the internal gas pressure proportionally.

**Table 4.1-1 Plant AA Helium Solubility Data**

Rod Identification	Fuel Density (%T. D.)	Design Pressure (psia)	Burnup (MWd/MTU)	Helium Content (cc/gm)
(1) 506-A1	90.1	720	4600	
(2) 506-A2	90.1	720	4600	
(3) 506-B1	90.2	720	3300	
(4) 506-B2	90.2	720	3300	
(5) PM-1	94.7	500	3310	
(6) PM-2	94.7	500	3500	
(7) PM-3	94.7	500	3350	
(8) PM-4	94.7	500	3500	
(9) PM-5	94.7	500	1070	

a,b,c

**Table 4.1-2 Plant A and Plant Z Helium Solubility Data**

Rod Identification	Fuel Density (% T. D.)	Design Pressure (psia)	Burnup (MWd/MTU)	Helium Content (cc/gm)
(1) Plant A 007-110	96	600	14030	
(2) Plant A 007-4.5	96	600	7788	
(3) Plant A 013-110	96	380	10707	
(4) Plant A 11-110	94	200	13774	
(5) Plant A/3/C18-H1-1A	91.5	275	23500	
(6) Plant A /3/C18-H1-1B	91.5	275	23500	
(7) Plant A /3/C18-J5-1A	91.5	275	22600	
(8) Plant A /3/C18-J5-1B	91.5	275	22600	
(9) Plant Z 285/75-76	94.6	500	31000	
(10) Plant Z 285/75-76	94.6	500	31000	
(11) Plant Z 285/9.9-10.4	94.6	500	30000	
(12) Plant Z 285/4.8-5.2	94.6	500	25000	

a,b,c

**Table 4.1-3 Comparison of Predicted and Measured Helium Solubility Data**

Data Point	Helium Content (cc/gm)	
	Predicted	Measured
Table 4.1-1 (1)		
Table 4.1-1 (2)		
Table 4.1-1 (3)		
Table 4.1-1 (4)		
Table 4.1-1 (5)		
Table 4.1-1 (6)		
Table 4.1-1 (7)		
Table 4.1-1 (8)		
Table 4.1-1 (9)		
Table 4.1-2 (1)		
Table 4.1-2 (2)		
Table 4.1-2 (3)		
Table 4.1-2 (4)		
Table 4.1-2 (5)		
Table 4.1-2 (6)		
Table 4.1-2 (7)		
Table 4.1-2 (8)		
Table 4.1-2 (9)		
Table 4.1-2 (10)		
Table 4.1-2 (11)		
Table 4.1-2 (12)		

a,b,c

a,b,c

**Figure 4.1-1 Comparison of Predicted and Measured Helium Content**

## 4.2 HELIUM RELEASE (ZIRCONIUM DIBORIDE)

The helium generation from  $ZrB_2$  is the same as in PAD 4.0. The helium release model for  $ZrB_2$  is revised for PAD5.

### 4.2.1 Background and Helium Release Data

Helium generation from the  $ZrB_2$  coating is equal to the Boron-10 ( $B^{10}$ ) depletion of the boron coating. The model for the fractional  $B^{10}$  depletion of boron coated pellets is

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \\ \text{ } \end{array} \right]^{a,c} \quad (4-5)$$

where:

$B_{10}$  = fraction remaining of  $B^{10}$   
 the units of Rod Burnup are GWd/MTU  
 the units of Enrichment are weight percent  
 the constants are [ ]<sup>a,c</sup>

The  $ZrB_2$  helium release model in PAD 4.0 conservatively assumes [ ]<sup>a,c</sup> for best estimate results since the licensing of the Vantage 5 fuel assembly in Reference 4.3.

This model is largely based on results from early [ ]<sup>a,c</sup>.

There are helium release data from [ ]

[ ]<sup>a,c</sup>

A recent Halden test has allowed a better understanding of the magnitude and trend of the release during the time of helium production and after the completion of the production. Two rods with  $ZrB_2$  coating [ ]<sup>a,c</sup>. The fraction of helium gas released from the  $ZrB_2$  coating was determined to be around [ ]

[ ]<sup>a,c</sup>

[

] <sup>a,c</sup>**Table 4.2-1 Boron Coated Fuel Helium Release Data from LTA Programs**

<b>Test Number</b>	<b>Plant</b>	<b>Thickness (mil)</b>	<b>B<sup>10</sup> Enrichment (%)</b>	<b>Percent B (%)</b>	<b>Helium Release (Fraction)</b>	<b>Burnup (GWd/MTU)</b>	<b>Time (days)</b>
------------------------	--------------	----------------------------	--	--------------------------	--	-----------------------------	------------------------

a,b,c



**Table 4.2-1 Boron Coated Fuel Helium Release Data from LTA Programs**  
(cont.)

Test Number	Plant	Thickness (mil)	B <sup>10</sup> Enrichment (%)	Percent B (%)	Helium Release (Fraction)	Burnup (GWd/MTU)	Time (days)
----------------	-------	--------------------	--------------------------------------	------------------	---------------------------------	---------------------	----------------

a,b,c

**Figure 4.2-1 Helium Release Fraction as Function of Irradiation Time**

#### 4.2.2 ZrB<sub>2</sub> Helium Release Model

The temperature of the ZrB<sub>2</sub> coating was maintained at [ ]<sup>a,c</sup> which is higher than the temperature of the ZrB<sub>2</sub> coating expected in normal commercial reactor operation. While the absolute helium release fraction may vary in the Halden test rod due to this temperature difference, the helium release trend and the time-dependence indicated by the Halden data should be applicable to commercial fuel rods with ZrB<sub>2</sub> coated pellets.

From Beginning of Life (BOL) to the time the helium release reaches a maximum release fraction, the release fraction as a function of time can be modeled as three linear segments – the Best Estimate (BE) model shown in Figure 4.2-2. [

] <sup>a,c</sup>

[

] <sup>a,b,c</sup>] <sup>a,b,c</sup>

**Figure 4.2-2      ZrB<sub>2</sub> Helium Release Model and Data**

In summary, the proposed helium release fraction model as a function of irradiation time for PAD5 is [

] <sup>a,c</sup>

### 4.3 FISSION GAS RELEASE MODEL

*The model function form for the thermal release was developed as part of PAD 3.4 (Reference 4.4). The model continued to be used in PAD 4.0 (Reference 4.1). The coefficients for the thermal release model are recalibrated with an updated database. [*

*]<sup>a,c</sup>*

A two component gas release model is used to describe the fission gas release, with one component giving the gas release from the high fuel temperature regions of the high power rods and the second component giving the major contribution to the gas release from the lower power rods and the cooler outer regions of the fuel rods. The fission gas release model is based on a fission gas production rate of 30 atoms per 100 fissions.

#### 4.3.1 Thermal Fission Gas Release Model

The high temperature gas release model is based on concepts drawn from mechanistic models of high temperature gas release through interlinking of grain edge fission gas bubbles. Fission gas produced in the interior of the fuel grains migrates to the grain boundaries by diffusion or grain boundary sweeping, depending on the local fuel temperature. Fission gas bubbles form on the grain boundaries and a saturated gas bubble density develops on the grain edges as irradiation continues. Eventually the grain edge bubbles interlink, and the fission gas stored in the bubbles is vented to the fuel rod void volume. An equilibrium release rate is eventually established such that the net gas release rate equals the fission gas production rate.

The functional form of the thermal fission gas release model developed in Reference 4.4 is a phenomenological model based on the irradiation dependence of the gas release from mechanistic gas release models for the diffusion of the fission gas generated in the fuel grains to the grain boundaries, formation of fission gas bubbles on the grain boundaries, and release of the fission gas when these grain boundary bubbles interlink. At low fuel temperatures the grain boundary bubbles do not form, since the gas diffusion rates are very low, and no fission gas is released by this gas release mechanism. At higher fuel temperatures, grain boundary gas bubble formation and grain boundary gas bubble interlinking occurs. There is a delay in the gas release at the higher fuel temperatures to account for the time needed for the grain boundary bubbles to form and interlink. This delay decreases with increasing temperature, as the gas diffusion rates increase, and at very high fuel temperatures the thermal fission gas release is virtually instantaneous. The principal feature of this gas release model is that fission gas is released as fast as it is produced once a burnup threshold is reached. To give a smooth variation in the thermal fission gas release, a transition region is used through which the gas release rate varies continuously between no thermal gas release at low temperatures to a release rate equal to the fission gas production rate at high temperatures.

The temperature dependence of the fission gas release model burnup thresholds is determined by fitting the model with the fission gas release data. The burnup thresholds are shown in Figure 4.3-1. The burnup thresholds  $B_{INC}$  (for incubation) and  $B_{REP}$  (for Release Equal Production) are defined as functions of the local temperature  $T$ , in K, by pairs of burnup and temperature values, with the threshold curves given by:

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c} \quad (4-6)$$

The burnups (in MWd/MTU) and temperatures (in K) defining the inflection points for the  $B_{INC}$  and  $B_{REP}$  curves are:

[

] <sup>a,c</sup>

The fractional gas release in the three regions shown in Figure 4.3-1 are given by:

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c} \quad (4-7a)$$

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c} \quad (4-7b)$$

$$\left[ \begin{array}{c} \\ \\ \end{array} \right]^{a,c} \quad (4-7c)$$

where:

$R_{thermal}$  is fractional thermal fission gas release

$B$  is the local burnup (MWd/MTU)

and  $B_{REP}$  and  $B_{INC}$  are calculated from the current local temperature.

Equation (4-7b) is a simple interpolation formula between no release for burnup less than  $B_{INC}$  to release equal to production for burnup greater than  $B_{REP}$ , with the constants chosen so that the fractional release  $R_{thermal}$  is a continuous function of burnup.

For time dependent fuel temperatures, the thermal fission gas release model is applied by assuming that the thermal fission gas release is path independent when the fuel temperatures increase, so that there is rapid release of stored fission gas when the fuel temperatures increase. When the fuel temperatures decrease, there is no change in the amount of fission gas that has been released until the burnup reaches a value where the model predicts additional fission gas release at the lower temperature. Though the fractional thermal fission gas release will decrease with decreasing temperatures, the total amount of fission gas released will either remain constant or increase.

Because of the path independence of the steady-state thermal fission gas release with increasing temperatures, a transient fission gas release model is used to obtain the fission gas release during short duration overpower transients. The transient fission gas release model is given by:

$$R_{transient}(t) = f(t)R_{thermal} \quad (4-8)$$

where  $f(t)$  is an approximation of the Booth expression (Reference 4.5) for the diffusion of the fission gas in the fuel grain to the grain boundary:

$$\left[ \frac{a, c}{\tau} \right] \quad (4-9a)$$

$$\left[ \frac{a, c}{\tau} \right] \quad (4-9b)$$

where:

$$\tau = D_{Xe}t/XDIFF^2$$

$D_{Xe}$  is the Xenon diffusion coefficient ( $\text{cm}^2/\text{sec}$ )

$t$  is the time (sec)

$XDIFF$  is an effective grain radius (cm),  $\left[ \frac{a, c}{\tau} \right]$

and the Xenon diffusion coefficient, (Reference 4.6), is:

$$D_{Xe} = 7.6 \times 10^{-6} \exp(-35000/T) + 1.41 \times 10^{-18} \phi^{1/2} \exp(-13800/T) + 2 \times 10^{-30} \phi \quad (4-10)$$

where:

$T$  is the local temperature (K)

$\phi$  is the local fission rate (fissions/cm<sup>3</sup>-sec)

The  $f(t)$  transient factor is always used in calculating the thermal fission gas release. It has a relatively minor impact on the thermal fission gas release fractions calculated during steady state operation, where the fuel operates at effectively constant temperatures for a sufficiently long time for  $f(t)$  to reach its asymptotic value,  $f(t) = 1$ .

[

]<sup>a,c</sup>

a,b,c

**Figure 4.3-1       $B_{REP}$  and  $B_{INC}$  Burnup Dependence**

### 4.3.2 Athermal Fission Gas Release Models

Two models are used for the athermal fission gas release at low fuel temperatures. At all fuel burnups, there is a relatively small amount of fission gas release due to recoil and knockout. The fission gas release fraction due to knockout and recoil,  $R_{KOR}$ , is given by:

$$\left[ \frac{Bu_{loc}}{Bu_{enh}} \right]^{a,c} \quad (4-11)$$

where:

$Bu_{loc}$  is the fuel local radial burnup (MWd/MTU) and

$$\left[ \frac{Bu_{loc}}{Bu_{enh}} \right]^{a,c}$$

In addition, a significant increase in the fission gas release of rods that operated at relatively low power at high burnups has been measured. The fuel temperatures achieved for this fuel during high burnup operation are too low for there to be any significant release due to the thermal fission gas release effects included in the PAD5 thermal fission gas release model.

To account for this enhanced athermal fission gas release at high burnup, an enhanced athermal fission gas release fraction,  $R_{ath,enh}$ , is calculated:

$$\left[ \frac{Bu_{loc}}{Bu_{enh}} \right]^{a,c} \quad (4-12a)$$

$$\left[ \frac{Bu_{loc}}{Bu_{enh}} \right]^{a,c} \quad (4-12b)$$

where  $Bu_{loc}$  is the fuel radial local burnup, in MWd/MTU, and  $Bu_{enh}$  is the threshold burnup for the enhanced athermal fission gas release, in MWd/MTU. The value of these model constants for the high burnup enhanced athermal fission gas release are:

$$\left[ \frac{Bu_{loc}}{Bu_{enh}} \right]^{a,c}$$

## 4.4 EQUATION OF STATE (EOS) MODEL

*This model is the same as in PAD 4.0 (Reference 4.1).*

### 4.4.1 Background

The relationship between pressure, temperature, and mass for the fill gas or fission gas in PAD 3.4 (Reference 4.4) and earlier models was based on the Ideal Gas Law. The Ideal Gas Law relationship is valid for many gases near room temperature and pressure, and is good for noble gases such as helium, neon, and argon up to moderate pressures (400–500 psia). At high pressures ( $P > 500$  psia) however, the



Ideal Gas Law becomes increasingly inaccurate. Figure 4 from Chapter VII of Reference 4.7 shows that above about 500 psia, inert gases do not exhibit ideal behavior. At end of life, when the rod internal pressure can exceed 2000 psia, none of these gases will exhibit ideal behavior. Therefore, use of the Ideal Gas Law to estimate rod pressure given the temperature and specific volume will be inaccurate.

A survey was conducted to determine the most appropriate EOS. It was determined that the Peng-Robinson equation, (Reference 4.8), gave the most accurate predictions for the range of interest. However, this EOS consistently under predicts the measurements over the whole range of helium mole fraction. A positive aspect of this EOS is that the prediction error is insensitive to the helium concentration, thus leading to the following (calibrated) gas EOS:

$$P^c = \alpha P \quad (4-13)$$

where  $P^c$  is the corrected or calibrated Peng-Robinson gas EOS,  $P$  is the original Peng-Robinson gas EOS, and  $[\alpha]^{a,c}$  is a correction factor determined from the data depicted in Figure 4.4-1, which shows the prediction error  $DP/P$  as defined below as a function of helium mole fraction for the corrected Peng-Robinson EOS. The prediction errors are very balanced and range between -2% and 2%.

For fission gas mixtures even at end of life, helium generally has the highest mole fraction. Measured data was obtained from References 4.9 through 4.13 for the various gas matrix evaluations.

$DP/P$  is defined as:

$$\frac{DP}{P} = \frac{P_{predicted} - P_{measured}}{P_{measured}} \quad (4-14)$$

This quantity is positive when the pressure is over predicted, and negative when under predicted.

The calibrated Peng-Robinson EOS performs well at high He composition. Figure 4.4-2 shows the calibrated Peng-Robinson Predicted vs. Measured plot results for various pressures.

a,c

**Figure 4.4-1      Calibrated Peng-Robinson Equation of State DP/P versus Helium Mole Fraction**



**Figure 4.4-2 Calibrated Peng-Robinson Equation of State Predicted versus Measured Pressure**

In summary, the Ideal Gas Law was found to potentially underpredict pressure for compositions with high helium mole fraction and the Peng-Robinson EOS was found to better fit measured data.

#### 4.4.2 EOS Model

The pressure-temperature-volume relationship for a pure fluid is often represented by a cubic Equation of State, which has the general form:

$$P = \frac{RT}{v-b} - \frac{a}{v^2 + ubv + wb^2} \quad (4-15)$$

where  $P$  is the pressure,  $T$  is temperature,  $v$  is specific volume, and  $R$  is the Universal Gas constant.

For the Peng-Robinson equation of state,  $u = 2$  and  $w = -1$  with:

$$b = \frac{0.07780RT_c}{P_c} \quad (4-16)$$

and:

$$a = \frac{0.45724R^2T_c^2}{P_c} [1 + f\omega(1 - T_r^{0.5})]^2 \quad (4-17)$$

where:

$$f\omega = 0.37464 + 1.54226\omega - 0.26992\omega^2 \quad (4-18)$$

In Equations (4-16) and (4-17), the subscript “c” denotes properties at the critical point. The reduced temperature is defined as:

$$T_r = \frac{T}{T_c} \quad (4-19)$$

The function for  $f\omega$  given by Equation (4-18) uses the acentric factor  $\omega$ , which is a parameter that represents the complexity of a molecule with respect to geometry and polarity. For mono-atomic gases,  $\omega$  is usually zero or very small.

In the PAD5 code up to eight different gases can be present in the gas mixture. Table 4.4-1 lists these components and the properties assigned in the code as taken from Reference 4.8.

Table 4.4-1 Pure Gas Component Properties List for PAD5			
Component	$T_c$ (K)	$P_c$ (bar)	$\omega$
Helium	5.19	2.27	-0.365
Xenon	289.7	58.4	+0.008
Krypton	209.4	55.0	+0.005
Argon	150.8	48.7	+0.001
Nitrogen	126.2	33.9	+0.039
Water Vapor	647.3	221.2	+0.344
Hydrogen	33.2	13.0	-0.218
Oxygen	154.6	50.4	+0.025

For gas mixtures, the attraction and repulsion between molecules of different components causes non-linear variation of some properties with composition. To account for this in Equation (4-15), a set of mixing rules can be defined to this non-linearity. The values of “a” and “b” in Equation (4-15) are re-defined. Based on the recommendations in Reference 4.8, the following mixing rules are used in the PAD5 code:

$$a_m = \sum_i \sum_j y_i y_j (a_i a_j)^{0.5} (1 - k_{ij}) \quad (4-20)$$

$$b_m = \sum_i y_i b_i \quad (4-21)$$

The  $b_i$  and  $a_i$  for each pure component are given by Equations (4-16) and (4-17) respectively. The term  $k_{ij}$  is used for some binary pairs to adjust for strong interactions and is determined from experimental data. In the PAD5 code  $k_{ij} = 0$ , is assumed for all binary combinations.

#### 4.5 GAS ABSORPTION IN CLADDING EFFECT

*This model is the same as in PAD 4.0 (Reference 4.1).*

The fuel rod internal free gas mixture includes: (1) the fission gases produced during operation, (2) gas from the pellets including the gas from the  $ZrB_2$  coating if present, (3) the gas from the rod pre-pressurization, and (4) the [ ]<sup>a,c</sup>. When the rod is pre-pressurized and sealed during fabrication, the rod [ ]

with [ ]<sup>a,c</sup>. Zirconium alloys are known to react [ ]<sup>a,c</sup>.

For example, assuming a plenum volume of about [ ]<sup>a,c</sup> and a gas mixture of [ ]

[ ]<sup>a,c</sup>. With about [ ]<sup>a,c</sup> the corresponding weight gain for total [ ]<sup>a,c</sup>. Based on reaction rates in Reference 4.15, [ ]<sup>a,c</sup> will occur within [ ]<sup>a,c</sup>. Thus, all of the [ ]<sup>a,c</sup>.

Zirconium preferentially reacts with [ ]<sup>a,c</sup> are present. When the [ ]

[ ]<sup>a,c</sup> The absorption rate of [ ]<sup>a,c</sup>. Based upon the weight of [ ]<sup>a,c</sup>. Thus, it will take about [ ]<sup>a,c</sup>. This may be a lower than actual rate since the rate is temperature dependent. [ ]<sup>a,c</sup>

Irradiated rods were punctured in the hot cell and the gas present in the rod was captured and analyzed. In the 22 Plant B rods measured and in 7 other rods from Plant C and Plant D, there was [ ]<sup>a,b,c</sup>. The

measurement sensitivity is reported as less than 0.01% by volume. If the [ ]<sup>a,c</sup> pressurization and a resultant internal rod gas mixture with [ ]<sup>a,c</sup> by volume. These levels, if present in the rod at end of life, are above the detection limit by a factor of over 100.

These evaluations indicate that the [ ]<sup>a,c</sup> the Zircaloy-4, ZIRLO, or Optimized ZIRLO cladding microstructure, and [ ]

[ ]<sup>a,c</sup>. Therefore, PAD5

does not account for [  $\sum_j^{a,c}$  ]

#### 4.6 VOID VOLUMES AND INTERNAL PRESSURE

*This model has minor updates to what was used in PAD 4.0 (Reference 4.1). Original documentation for the model is located in NS-SL-521 (Reference 4.16 Attachment C). PAD5 updates the crack volume and the gap volume due to the pellet relocation model. The central hole volume for the annular pellets was also modeled in PAD 4.0, but was not discussed in the PAD 4.0 topical report.*

The fuel rod void volume at power is the sum of the following void components:

- Plenum volume
- Dish volume
- Fuel-clad gap volume
- Crack volume
- Open porosity volume
- Fuel pellet surface imperfection volumes
- Central hole volume for annular pellets

Plenum volume is calculated from the initial clad, fuel, and spring dimensions and subsequently accounts for temperature and irradiation effects (such as clad growth and fuel swelling).

The dish volume is calculated at any given power and time by applying the appropriate thermal expansion and fuel swelling to ten radial rings, recalculating the shape of the pellet, and then summing up the volume at the pellet end face.

The fuel-clad gap volume is calculated by subtracting a pellet cross-sectional area based on radial fuel pellet expansion due to fuel densification, swelling, thermal expansion, and pellet relocation, from a cladding ID cross-sectional area which includes the effects of elastic and plastic cladding deformation.

The crack volume, as a result of pellet cracks from the thermal stress, is calculated at any given power and time by applying the linear thermal expansion to ten radial rings (appropriate to its average temperature), calculating the pellet volume based on the pellet radius, and then subtracting the sum of the volumetric expansions of each ring. [  $\sum_j^{a,c}$  ]

$$\left[ \sum_j^{a,c} \left( \frac{\Delta l_j}{DP_{hot}} \left( \frac{DP_{hot}^2}{4} - \frac{(\delta D_{reloc,j})^2}{4} \right) \right) \right]^{a,c} \quad (4-22)$$

where  $j$  is the axial section index,  $\Delta l_j$  is the height of the axial section,  $DP_{hot}$  is the hot pellet diameter, and  $\delta D_{reloc}$  is the diametral relocation.  $\Delta V_{reloc,j}$  is added to the pellet crack volume in each axial section.

The open porosity volume is calculated as a function of initial fuel density and fuel temperature at power. The open porosity at room temperature is:

$$\left[ \frac{\rho_0}{\rho} \right]^{a,c} \quad (4-23)$$

where  $\rho_0$  is geometric density in % T.D. The above relationship was obtained from open porosity measurements on a large number of unirradiated production fuel pellets.

The open porosity dependence on temperature,  $T$ , is calculated as follows:

$$\left[ \frac{T - T_0}{T_0} \right]^{a,c}$$

Fuel pellet surface imperfection volume is the difference between the geometric and true density. It has been determined to be  $\left[ \frac{\rho_0}{\rho} \right]^{a,c}$  of the fuel volume, based on a large number of measurements on unirradiated production pellets.

The central hole volume for the annular pellets is calculated from the initial central hole volume and subsequently accounts for temperature and irradiation effects (such as fuel thermal expansion and fuel swelling and densification).

All the volumes thus calculated are summed to give the total fuel rod internal void volume.

Rod internal pressure is calculated using the Peng-Robinson EOS discussed in Section 4.4. The various volumes and the corresponding temperatures used in the equation are:

Plenum volume	$\left[ \frac{T - T_0}{T_0} \right]^{a,c}$	
Dish volume		
Fuel-clad gap volume		
Crack volume		
Open porosity volume		
Fuel pellet surface imperfection volumes		
Central hole volume for annular pellet		

## 4.7 SECTION 4 REFERENCES

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## 5 CLAD AND FUEL DEFORMATION MODELS

Section 5 begins by describing the PAD5 cladding models for:

- Clad stresses, including fuel-cladding contact pressure, differential pressure stresses, stress for clad creep and plasticity analysis, and thermal stresses,
- Clad elastic deformation,
- Clad plasticity,
- Clad creep, and
- Clad diametral growth.

Following a description of these cladding models, Section 5 then describes fuel models for:

- Fission gas bubble swelling,
- Fuel swelling and densification, and
- Fuel relocation

Lastly, the fuel rod cladding axial growth model is described. Clad and fuel thermal expansion models are described in subsections 6.3.3 and 6.1.3, respectively.

### 5.1 CLAD STRESSES

#### 5.1.1 Fuel – Clad Interference Gap and Contact Pressure

*This model is the same as in PAD 4.0 (Reference 5.1) except that the fuel relocation deformation is considered in the interference gap in PAD5.*

The fuel-clad gap is determined by first radially deforming both the fuel and cladding. The cladding is radially deformed considering thermal expansion (subsection 6.3.3), elastic deformation in response to the rod internal pressure and coolant pressure (Section 5.2), creep (Section 5.4), clad diametral growth (Section 5.5), and plastic deformation (Section 5.3). The fuel is radially deformed considering thermal expansion (subsection 6.1.3), relocation (Section 5.6), fuel densification and swelling (Section 5.7), and high temperature fission gas bubble swelling (Section 5.8).

If the deformed cladding inner diameter is predicted to be radially inside of the deformed fuel diameter in a particular axial segment, i.e., an interference fuel-cladding gap, the fuel-cladding contact pressure is calculated based on the interference gap size. The interference gap, GAP, is represented as a positive quantity for this calculation. The fuel-cladding contact pressure is calculated assuming both the fuel and cladding are in the elastic state. The contact pressure is given by:

$$\left[ \frac{P_{ct}}{E_c} \right] \frac{a_c}{a_c} \quad (5-1)$$

where:

$P_{ct}$  is the contact pressure (psi)

GAP is the diametral interference (in)

$R_i$  is the clad inside radius (in)

E is Young's Modulus (psi)

$\mu$  is Poisson's Ratio

$K_c$  is the cladding radius ratio,  $R_o/R_i$

and the subscripts c and p refer to cladding and fuel pellet, respectively.

Some deformations (for example, fuel relocation, clad plasticity, and cladding creep) may depend on the calculated stress or contact pressure. These deformations and the contact pressure for a give time step have to be solved iteratively. The solution procedure is summarized in Section 2.1.

### 5.1.2 Cladding Stresses from Differential Pressure

*This model is the same as in PAD 4.0 (Reference 5.1).*

The cladding stresses are calculated assuming no interaction between the axial segments in the fuel rod geometry. For elastic analysis, the cladding stresses for radial, hoop and axial directions are:

$$\left[ \begin{array}{c} \sigma_r \\ \sigma_\theta \\ \sigma_z \end{array} \right] = \left[ \begin{array}{c} P_{gas} \\ P_{sys} \\ P_{con} \end{array} \right] \quad (5-2a)$$

$$\left[ \begin{array}{c} \sigma_r \\ \sigma_\theta \\ \sigma_z \end{array} \right] = \left[ \begin{array}{c} P_{gas} \\ P_{sys} \\ P_{con} \end{array} \right] \quad (5-2b)$$

$$\left[ \begin{array}{c} \sigma_r \\ \sigma_\theta \\ \sigma_z \end{array} \right] = \left[ \begin{array}{c} P_{gas} \\ P_{sys} \\ P_{con} \end{array} \right] \quad (5-2c)$$

where:

$\sigma_r$  = radial stress (psi)

$\sigma_\theta$  = hoop stress (psi)

$\sigma_z$  = axial stress (psi)

$P_{gas}$  = rod internal pressure (psi)

$P_{sys}$  = coolant system pressure (psi)

$P_{con}$  = pellet-cladding contact pressure (psi)

$R_o$  = cladding outer radius prior to accounting for elastic strains (in)

$R_i$  = cladding inner radius prior to accounting for elastic strains (in)

$K_c$  =  $R_o/R_i$ .

The radial and hoop stresses depend on the radial position 'R' in the cladding wall, and the axial stress is constant across the cladding wall.

### 5.1.3 Stress for Clad Creep and Plasticity Analysis

*This model is new for PAD5.*

PAD5 uses the Prandtl-Reuss flow rule to calculate the cladding hoop creep and plastic strain increments, with the stresses calculated at the cladding mid-wall:

$$\delta\epsilon_{\theta} = \frac{\delta\epsilon_{eff}}{\sigma_{eff}} \sigma_{\theta,d} \quad (5-3)$$

where:

- $\delta\epsilon_{\theta}$  = hoop strain increment (in/in)
- $\delta\epsilon_{eff}$  = effective strain increment, calculated from the creep rate model (in/in)
- $\sigma_{eff}$  = effective stress (psi or MPa)
- $\sigma_{\theta,d}$  = hoop deviatoric stress (psi or MPa)

with:

$$\sigma_{\theta,d} = \sigma_{\theta} - 1/2(\sigma_r + \sigma_z)$$

and:

$$\sigma_{eff} = \frac{1}{\sqrt{2}} \left[ (\sigma_{\theta} - \sigma_r)^2 + (\sigma_r - \sigma_z)^2 + (\sigma_z - \sigma_{\theta})^2 \right]^{1/2} \quad (5-4)$$

This flow rule can be generalized to calculate the hoop, radial, and axial creep and plasticity strain increments, using the definitions:

$$\sigma_{\theta,d} = \sigma_{\theta} - 1/2(\sigma_r + \sigma_z) \quad (5-5a)$$

$$\sigma_{r,d} = \sigma_r - 1/2(\sigma_{\theta} + \sigma_z) \quad (5-5b)$$

$$\sigma_{z,d} = \sigma_z - 1/2(\sigma_r + \sigma_{\theta}) \quad (5-5c)$$

With these definitions,

$$\sigma_{eff} = \sqrt{\frac{2}{3}} \left[ \sigma_{\theta,d}^2 + \sigma_{r,d}^2 + \sigma_{z,d}^2 \right]^{1/2} \quad (5-6)$$

### 5.1.4 Thermal Stresses

*This model is the same as in PAD 4.0 (Reference 5.1).*

Thermally induced hoop stresses caused by the temperature gradient through the cladding are considered in the fatigue calculation. These stresses are given by:

$$\left[ \frac{a, c}{\lambda} \right] \quad (5-7)$$

and

$$\left[ \frac{a, c}{\eta} \right] \quad (5-8)$$

where  $\lambda$  and  $\eta$  are cladding anisotropic property coefficients,  $k_{\text{clad}}$  is the clad thermal conductivity (BTU/hr-ft-°F),  $q''(Z)$  is the heat flux (BTU/hr-ft<sup>2</sup>) at axial elevation  $z$ , and  $\sigma_{\theta,0}$  and  $\sigma_{\theta,i}$  are at the cladding outside and inside, respectively.

Assuming isotropic properties and rearranging, simplified forms of these expressions are obtained:

$$\left[ \frac{a, c}{\alpha} \right] \quad (5-9)$$

and

$$\left[ \frac{a, c}{\mu} \right] \quad (5-10)$$

where,  $\alpha$  is the thermal expansion coefficient,  $\mu$  is Poisson's ratio,  $\Delta T$  is the temperature (°F) difference between clad inner and outer surface.

## 5.2 CLAD ELASTIC DEFORMATION

*This model is the same as in PAD 4.0 (Reference 5.1).*

Considering the fuel rod as a closed-end transversely isotropic thick-walled cylinder in a three dimensional system subjected to the external coolant pressure,  $P_s$ , and the internal fission gas pressure,  $P_f$ , the elastic diametric deflection at the inner surface due to the pressure differential is given by:

$$\left[ \frac{a, c}{E_c} \right] \quad (5-11)$$

where:

- $K_c$  = cladding radius ratio, ( $R_o/R_i$ );
- $\mu_c$  = Poisson's ratio for cladding;
- $E_c$  = Young's Modulus for cladding (psi).

### 5.3 CLAD PLASTICITY

*This model is revised for PAD5.*

Cladding plastic strains follow a power law expression:

$$\left[ \frac{\epsilon_{pl}}{\sigma_{eff}} \right]^{a,c} = \frac{\sigma_y}{E} \quad (5-12)$$

where:

$\epsilon_{pl}$  is the plastic effective strain (in/in)

$\sigma_{eff}$  is the cladding effective stress (psi or MPa)

$\sigma_y$  is the irradiated cladding 0.2% offset yield stress (psi or MPa)

The correlation for the plasticity model stress exponent  $N$  can be developed using the materials properties correlations for Young's modulus ( $E$ ), the yield stress, and the ultimate tensile strength.

Though Equation (5-12) will be used as an equation for engineering strain as a function of engineering stress in PAD5, it is an equation for true strain as a function of true stress. This distinction is not significant for plastic strains  $< 2\%$ , which is probably the maximum plastic strain that could occur in fuel rod overpower ramp tests and is certainly more than the plastic strain that would be calculated in design analyses, which will be no more than the 1% allowable transient total strain increment. However, the distinction is significant for the evaluation of unirradiated cladding tubing plasticity data when it is used to obtain data for the stress exponent  $N$  in Equation (5-12).

During a uniaxial test,

$$\epsilon_{true} = \ln(1 + \epsilon_{eng}) \quad (5-13)$$

and:

$$\sigma_{true} = \sigma_{eng} e^{\epsilon_{true}} \quad (5-14)$$

Including the elastic strains, the relationship between the total true strain and the true stress is:

$$\left[ \frac{\sigma_{true}}{\sigma_y} \right]^{a,c} = \epsilon_{true} \quad (5-15)$$

The engineering strain at the maximum is the Uniform Elongation (UE) and the engineering stress at the maximum is the Ultimate Tensile Stress (UTS). The condition defining the UE is:

$$\left[ \frac{\sigma_y}{E} \right]^{a,c} \quad (5-16)$$

From Equations (5-15) and (5-16), it can be shown that the stress exponent  $N$  and the cladding Uniform Elongation (UE) are related by two coupled non-linear equations:

$$\left[ \frac{\sigma_y}{E} \right]^{a,c} \quad (5-17)$$

$$\left[ \frac{\sigma_y}{E} \right]^{a,c} \quad (5-18)$$

where  $E$  is the cladding Young's modulus and UTS is the cladding Ultimate Tensile Strength.

Using the Westinghouse materials properties correlations for  $E$ , the irradiated yield stress  $\sigma_y$ , and the irradiated Ultimate Tensile Stress in Section 6, the stress exponent  $N$  for irradiated cladding is temperature dependent and is given by:

$$\left[ \frac{\sigma_y}{E} \right]^{a,c} \quad (5-19a)$$

$$\left[ \frac{\sigma_y}{E} \right]^{a,c} \quad (5-19b)$$

where:

$T_c$  is the cladding temperature ( $^{\circ}\text{C}$ ).

This correlation for  $N$  is applicable to irradiated Zircaloy-4, ZIRLO, and **Optimized ZIRLO** cladding.

Figure 5.3-1 shows the behavior of the PAD5 cladding plasticity model. This plasticity model is implemented using the von Mises criterion and the Prandtl-Reuss flow law to calculate the hoop, radial, and axial components of the plastic strains as discussed in subsection 5.1.3, in the same way that the hoop, radial, and axial components of the cladding creep strains are calculated.



**Figure 5.3-1 Plasticity Model Schematic Behavior**

## **5.4 CLAD CREEP**

*The thermal creep model is the same as in PAD 4.0 (Reference 5.1). The irradiation creep model is revised in PAD5. The coefficients are updated for PAD5 based on calibration to in-reactor and test reactor data.*

In-reactor diametral creep is predominantly a combination of the following mechanisms: thermal creep, irradiation enhanced creep, and irradiation induced diametral growth.

The in-reactor cladding creep determines the total effective strain rate by adding together the terms for thermal, irradiation, and diametral growth:

$$\left[ \begin{array}{c} \text{a,c} \\ \hline \end{array} \right] \quad (5-20)$$

where:

- $\dot{\epsilon}_{th}$  = thermal creep rate (in/in/hour),  
 $\dot{\epsilon}_{irr}$  = irradiation creep rate (in/in/hour), and  
 $\dot{\epsilon}_{gr}$  = diametral stress-free irradiation growth rate (in/in/hour).

$\dot{\epsilon}_{th}$ ,  $\dot{\epsilon}_{irr}$ , and  $\dot{\epsilon}_{gr}$  are treated as mid-wall quantities, i.e., mid-wall clad strain and mid-wall clad strain rate. The thermal and irradiation creep components are distributed to the hoop, radial, and axial directions based on the flow rule relationship of the deviatoric stress, effective stress, and effective strain that was established in subsection 5.1.3. Only the diametral stress free irradiation growth is considered here. The axial stress free irradiation growth is considered in Section 5.9 rod axial growth.

For each cladding type [

]<sup>a,c</sup> determined with the final functional form such that the average Predicted-Measured creep value for each clad material is approximately zero; and thus the total effective strain rate is given by:

$$\left[ \begin{array}{c} \text{a,c} \\ \hline \end{array} \right] \quad (5-21)$$

Thermal effective creep rate  $\dot{\epsilon}_{th}$  is a time derivative of:

$$\left[ \begin{array}{c} \text{a,c} \\ \hline \end{array} \right] \quad (5-22)$$

where:

- $\epsilon_{th}$  = thermal creep effective strain (-),  
 $t$  = time (hour),  
 $\phi$  = cumulative fast fluence (n/cm<sup>2</sup>),  
 HTCAL and DTCAL are calibration coefficients.

and the secondary creep rate is:

$$\left[ \begin{array}{c} \text{a,c} \\ \hline \end{array} \right] \quad (5-23)$$

where:

- $\dot{\epsilon}_s$  = secondary creep rate (1/hour)



$T$  = temperature (K),

$\sigma_{eff}$  = effective stress (MPa),

$E$  = Young's modulus (MPa),

STCAL and QTCAL are calibration coefficients.

and the saturated primary creep strain is:

$$\left[ \frac{a,c}{\sigma_{eff}} \right] \quad (5-24)$$

and the time constant is:

$$\left[ \frac{a,c}{\sigma_{eff}} \right] \quad (5-25)$$

where, PTCAL and CTCAL are calibration coefficients.

From a review of the available irradiation creep data on Cold-Worked Stress Relieved (CWSR) Zircaloy-4, ZIRLO, and **Optimized ZIRLO** cladding, [

] <sup>a,c</sup>

a,b,c

**Figure 5.4-1 B&W/EPRI CWSR Zircaloy-4 Irradiation Creep Data (103 MPa) for Various Temperatures**

The revised effective irradiation creep rate  $\dot{\epsilon}_{irr}$  is:

$$\left[ \right]^{a,c} \quad (5-26)$$

where:

$\Phi$  = cumulative fast fluence (n/cm<sup>2</sup>),  
 $\phi$  = instantaneous fast flux (n/cm<sup>2</sup>-sec),

$$\left[ \right]^{a,c} \quad (5-27)$$

$T$  = temperature (K),  
 AICAL, DICAL, FICAL, and TSATIRR are calibration coefficients,

$$\left[ \right]^{a,c}$$

In PAD 4.0, [

$\epsilon^{a,c}$ . This change of temperature dependence is based on the following observations:

1. [

$\epsilon^{a,c}$

2. [

$\epsilon^{a,c}$

3. [

$\epsilon^{a,c}$

However, [

$\epsilon^{a,c}$ .

The diametral growth  $\epsilon_{gr}$  is described in Section 5.5.

The creep model calibration and model uncertainties are discussed in the creep model calibration section of Appendix A. The calibrated model coefficients are:

Table 5.4-1 PAD5 Creep Model Coefficients			
Creep Model	Calibration Parameters	Values	Comments
Thermal Creep	STCAL		
	PTCAL		
	CTCAL		
	HTCAL		
	QTCAL		
	DTCAL		
Irradiation Creep	AICAL		
	DICAL		
	FICAL		
	TSATIRR		

The coefficients CCOEF1 and CCOEF2 are cladding alloy specific material constants, with values given in the following table:

Table 5.4-2 PAD5 Creep Model Alloy Specific Material Constants			a,c
Cladding Material	CCOEF1	CCOEF2	

## 5.5 CLAD DIAMETRICAL GROWTH

The rod growth model for Zircaloy-4 cladding is the same as in PAD 4.0 (Reference 5.1). Rod growth models for ZIRLO cladding and **Optimized ZIRLO** cladding are revised for PAD5.

### 5.5.1 ZIRLO Cladding and **Optimized ZIRLO** Cladding

The fluence dependence of the experimental data for both ZIRLO and **Optimized ZIRLO** cladding is different from the PAD 4.0 model. The PAD 4.0 model has an exponential decay fluence dependence. The data exhibit an approximately linear fluence dependence. Figures 5.5-1 and 5.5-2 show that:

- The fluence dependence is linear and
- The measured strains (note that the data have negative hoop strain values) are significantly lower than the PAD 4.0 model.

The models for the stress-free diametral growth strains of ZIRLO cladding and **Optimized ZIRLO** cladding are:

$$\left[ \begin{array}{c} \text{a,c} \\ \end{array} \right] \quad (5-28)$$

where  $R_0$  are the low fluence irradiation growth rates and  $\Phi$  is the fast fluence. The values for these material parameters, for  $\epsilon_0$  in units of % and fast fluence in units of  $10^{21}$  n/cm<sup>2</sup>, are:

Parameter (Units)	ZIRLO Cladding	<b>Optimized ZIRLO</b> Cladding	a,c
$R_0$ (%/ $10^{21}$ n/cm <sup>2</sup> , E>1 MeV)			

The nominal tin content for **Optimized ZIRLO** cladding is [ ]<sup>a,c</sup>.

The hoop strain data are shown in Figures 5.5-1 for ZIRLO cladding and 5.5-2 for **Optimized ZIRLO** cladding as a function of fluence.

In the case of ZIRLO cladding, [ ]<sup>a,c</sup>. Figure 5.5-1 shows that the data are consistent.

In the case of Optimized ZIRLO cladding, [  
]a,c .

a,b,c

**Figure 5.5-1      ZIRLO Cladding Irradiation Growth Diameter Strain**



**Figure 5.5-2**     *Optimized ZIRLO* Cladding Diameter Irradiation Growth

### 5.5.2 Zircaloy-4 Cladding

*This model is the same as in PAD 4.0 (Reference 5.1).*

$$\left[ \begin{array}{c} \text{a,c} \\ \text{ } \end{array} \right] \quad (5-29)$$

where  $\dot{\epsilon}_z$  = axial growth rate (time derivative of Equation [5-48] in subsection 5.9.3).

## 5.6 FUEL RELOCATION

*This model is new for PAD5.*

Fuel pellets crack as a result of thermal stress during operation. When there is a fuel-cladding gap, the fuel pellet fragments move towards the cladding and reduce the gap size. Previous versions of PAD have accounted for this effect in the thermal calculations, but not in the pellet-cladding mechanical interaction calculations.

When the gap is open, the relocation depends on the pellet-cladding gap size. The relocation may also change due to mutual interaction between the pellet fragments caused by thermal expansion during a power change or long term interaction effects due to fuel swelling. When the pellet-cladding gap closes, the relocation may reduce due to the contact forces exerted on the pellet fragments. This relocation reduction is not expected to be instantaneous. The magnitude of the relocation reduction depends on the contact pressure and the nature of the interface, which may change during irradiation as a result of failure or localized creep of the asperities on the fuel and/or cladding surface.

The pellet relocation phenomenon has been studied in detail by Oguma (Reference 5.2). Oguma's model is based on the analyses of pellet cracking (crack pattern and diameter increase) using out-of-pile experiments of cracked pattern behavior under soft and hard pellet-clad interaction. Oguma's work also included analyses of gap closure during power ramps of instrumented fuel rods irradiated in the Halden Boiling Water Reactor for determining the burnup and power dependence of pellet relocation. Similar models were used in developing the Westinghouse fuel performance code FATES (Reference 5.3) and the EPRI fuel performance code ESCORE (Reference 5.4).

The pellet fragment relocation diameter change is added to the hot pellet Outer Diameter (OD), calculated without considering relocation, and reduces the pellet-cladding mechanical gap. For consistency with the PAD mechanical analysis, the fragment relocation is defined as an increment in the pellet outer diameter,  $\delta D_{\text{reloc}}$  (in).

The initial pellet fragment diametral relocation, before the pellet-cladding gap initially closes, is given by:

$$\left[ \frac{G_o}{D_p} \right]^{a,c} \quad (5-30)$$

Where:

$$\left[ \frac{G_o}{D_p} \right]^{a,c} = \left[ \frac{G_o}{D_p} \right]^{a,c} \left[ \frac{Bu}{Q} \right]^{a,c} \left[ \frac{Q}{Q_c} \right]^{a,c}$$

The meaning of the variables found in these equations is as follows:

$$\begin{aligned} G_o &= \text{as-fabricated pellet-cladding diametral gap (in)} \\ DP &= \text{as-fabricated pellet diameter (in)} \\ Bu &= \text{local burnup (MWd/MTU)} \\ Q &= \text{linear heat generation rate (kW/m)} \\ Q_c &= \text{critical power for pellet cracking (onset of relocation), = } \left[ \right]^{a,c} \end{aligned}$$



and CRELOC0G, CRELOC0P, CRELOC1, CRELOC2, CRELOC3, and CRELOC4 are calibration constants that have been adjusted in the PAD5 calibration.

For Westinghouse fuel designs, the current calibration of the relocation model results in maximum initial diametral relocations (at high powers and low burnups) of [ ]<sup>a,c</sup>, increasing to saturation at about [ ]<sup>a,c</sup> at burnups of about [ ]<sup>a,c</sup>, provided the gap remains open. Figure 5.6-1 shows fuel diametral relocations for a pellet diameter of 0.3444 in, and an as-fabricated diametral gap of 7 mil as a function of burnup at various power levels.

a,c

**Figure 5.6-1 Behavior of the Initial Relocation as a Function of Burnup at Constant Powers**

It is assumed that the relocation never decreases when the gap is open. In addition, it is assumed that when the power increases, the relocation immediately increases to the value it would have had if the fuel had been operated at the current power from Beginning of Life (BOL). Therefore, for cases where the Linear Heat Generation Rate (LHGR) changes from time step to time step before the pellet-cladding gap initially closes, the relocation is given by the maximum of Equation (5-30), calculated using the burnup and LHGR in the current time step, and the relocation at the end of the previous time step.

If the gap is closed, relocation accommodation, i.e., inward motion of the pellet fragments due to the forces applied by the cladding on the fragments, is modeled using [ ]<sup>a,c</sup>:

$$\left[ \right]^{a,c} \quad (5-31)$$

where  $P_c$  is the contact pressure (psi) and  $\eta$  (psi-hr) is a calibration constant. The solution of this rate equation over a time interval  $\delta t$  is:

$$\left[ \frac{\delta D_{\text{reloc},o}}{\delta t} \right]^{a,c} \quad (5-32)$$

where  $\delta D_{\text{reloc},o}$  is the pellet relocation diameter increment at the start of the time interval. Assuming  $P_c$  varies relatively slowly over the time interval,  $\delta t$ :

$$\left[ \frac{\delta D_{\text{reloc},o}}{\delta t} \right]^{a,c} \quad (5-33)$$

[ ]<sup>a,c</sup> gives an exponential decay time of [ ]<sup>a,c</sup>, roughly consistent with the RECOR test data (Reference 5.5), for a contact pressure of [ ]<sup>a,c</sup>. In the implementation,  $\eta$  was used as the calibration factor, CRELOC5.

When the gap re-opens, it is assumed that there will be recovery of the accommodated relocation, such that the relocation eventually returns to the maximum value given by Equation (5-30) if the gap remains open for a sufficiently long time. This recovery ( $\delta D_{\text{reloc},\text{reopen}}$ ) of the accommodated relocation is given by:

$$[ ]^{a,c} \quad (5-34a)$$

$$[ ]^{a,c} \quad (5-34b)$$

$$[ ]^{a,c} \quad (5-34c)$$

Where:

$\delta Bu$  is the burnup increment (MWd/MTU) in the time interval  $\delta t$ ,

$\delta D_{\text{reloc},o}$  is the pellet relocation diameter increment at the start of the time interval, and

$\delta D_{\text{reloc},\text{max}}$  is the maximum value of the relocation throughout the previous operation of the fuel.

[

] <sup>a,c</sup>

The pellet fragment relocation reduces the pellet-cladding gap void volume and increases the pellet internal crack void volume. The relocation's impact on void volume is discussed in Section 4.6.

Validation of the relocation model is based on fuel temperature data for open gap conditions and ramp test cladding diameter change data for both open and closed gap conditions. These calibration results are shown in Appendix A. The calibrated coefficients are:

[

] <sup>a,c</sup>

## 5.7 FUEL SWELLING AND DENSIFICATION MODELS

*This model is the same as in PAD 4.0 (Reference 5.1). Validation of the model is extended to fuel with manufacture processes that are different from those used at the Westinghouse Columbia Fuel Fabrication Facility (CFFF).*

Densification and swelling are dependent on the manufacturing process. The densification and swelling model in PAD 4.0 is limited to fuel with microstructure close to Westinghouse fuel manufactured in the Columbia Fuel Fabrication Facility (CFFF). For fuel manufactured with different microstructure, the NRC densification model (Reference 5.6) is used in PAD 4.0. In PAD5, the densification model was extended to model fuel manufactured outside of the Columbia Fuel Fabrication Facility based on measured volume change data for fuel manufactured in the Westinghouse Sweden Vasteras Nuclear Fuel Factory and for the Combustion Engineering (CE) fuel manufactured in the Hematite plant.

### 5.7.1 Model Description

The PAD densification and swelling model is from Reference 5.7. Justification of the correlation and its coefficients are provided in Section 2-2 of Reference 5.7.

The PAD model for the fractional volume change due to solid fission product swelling and densification for UO<sub>2</sub> fuel is:

$$\left[ \dots \right]^{a,c} \quad (5-35)$$

where the first term gives the solid fission product swelling and the second term gives the densification, with F the burnup in fissions/cm<sup>3</sup>, P<sub>0</sub> is the initial porosity, and N is a function of the fuel sintering temperature:

$$\left[ \dots \right]^{a,c} \quad (5-36)$$

where  $T_{\text{sint}}$  is the pellet fabrication sintering temperature in K. Based on Halden tests, for gadolinia fuel the solid swelling in the first term in the right-hand side of Eq. (5-35) is adjusted from [

]<sup>a,c</sup>.

The PAD equation relating the burnup  $F$  in fissions/cm<sup>3</sup> to the burnup  $Bu$  in MWd/MTU is:

$$[ \quad ]^{a,c} \quad (5-37)$$

where  $\rho$  is the as-fabricated fuel fractional density (in terms of as-fabricated initial porosity  $P_o$ ,  $\rho = 1 - P_o$ ). In terms of  $Bu$ , the PAD model for solid fission product swelling and densification for  $UO_2$  fuel is:

$$[ \quad ]^{a,c} \quad (5-38)$$

Note that the units of  $Bu$  in Equation (5-38) are MWd/MTU. Based on Halden tests, for gadolinia fuel the solid swelling in the first term in the right-hand side of Eq. (5-38) is adjusted from [

]<sup>a,c</sup>.

PAD5 will use this model as the basis for fuel densification and swelling with [

]<sup>a,c</sup>

## 5.7.2 Data and Qualification

The combined densification and swelling model is validated using volume change data. Data provided in Tables 5.7-1, 5.7-2, and 5.7-3 are for fuel manufactured in Westinghouse Columbia, Vasteras, and the former CE Hematite fuel manufacture facilities. All Westinghouse (Columbia, Vasteras, and Hematite) data are plotted in Figure 5.7-1 as function of burnup.

For fuel manufactured in the Westinghouse Columbia Fuel Fabrication Facility, the fuel densification and swelling model has been validated in Reference 5.7. [

]<sup>a,c</sup> The predicted vs. measured results are shown in Figure 5.7-2, and the predicted minus measured versus burnup results are shown in Figure 5.7-3. [

]<sup>a,c</sup>

[<sup>a,c</sup>  
the model predicts the PWR fuel pellets with less uncertainty, as shown in Figure 5.7-4.

**Table 5.7-1 Volume Change ( $\Delta V/V$ ) Data for Fuel Manufactured in Columbia Fuel Fabrication Facility**

<b>Rod</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
------------	-----------------------------	--	--------------------------------

<sup>a,b,c</sup>

\_\_\_\_\_

a,b,c

11010.

\_\_\_\_\_

 $\top_{a,b,c}$

**Table 5.7-2 Volume Change ( $\Delta V/V$ ) Data for Fuel Manufactured in Vasteras Nuclear Fuel Factory**

<b>Reactor</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
----------------	-------------------------	------------------------------------	----------------------------

a,b,c

**Table 5.7-2 Volume Change ( $\Delta V/V$ ) Data for Fuel Manufactured in Vasteras Nuclear Fuel Factory (cont.)**

<b>Reactor</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
----------------	-------------------------	------------------------------------	----------------------------

a,b,c

**Table 5.7-3 Volume Change ( $\Delta V/V$ ) Data for CE Fuel (References 5.8 through 5.11)**

<b>Reactor</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
----------------	-------------------------	------------------------------------	----------------------------

a,b,c



**Table 5.7-3 Volume Change ( $\Delta V/V$ ) Data for CE Fuel (References 5.8 thru 5.11 )**  
**(cont.)**

<b>Reactor</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
----------------	-----------------------------	--	--------------------------------

a,b,c

**Table 5.7-3 Volume Change ( $\Delta V/V$ ) Data for CE Fuel (References 5.8 thru 5.11 )**  
(cont.)

<b>Reactor</b>	<b>Burnup (GWd/MTU)</b>	<b><math>\Delta V/V</math> (%)</b>	<b>Initial Density (%)</b>
----------------	-----------------------------	--	--------------------------------

a,b,c

a,b,c

**Figure 5.7-1      Columbia, Vasteras, and CE Fuel Volume Change Data Comparison**



**Figure 5.7-2 Predicted vs. Measured Volume Change ( $\Delta V/V$ ) Using PAD Densification and Swelling Model**

a,b,c

**Figure 5.7-3 Predicted – Measured Volume Change ( $\Delta V/V$ ) Versus Burnup for All Westinghouse Fuel Using PAD Densification and Swelling Model**

**Figure 5.7-4 Predicted – Measured Volume Change ( $\Delta V/V$ ) Versus Burnup for Columbia and CE Fuel Using PAD Densification and Swelling Model**

## **5.8 FISSION GAS BUBBLE SWELLING**

*This model is revised for PAD5. The fission gas swelling model used in PAD 4.0 (Reference 5.1) and previous PAD versions (Reference 5.7) and developed in PAD3.3 (Reference 5.12) has been improved and recalibrated.*

For irradiated fuel, fission gas atoms will diffuse through the fuel grains and accumulate at the grain boundaries. When the gas concentration reaches a certain amount, release tunnels will form and fission gas will be released to the rod internal free volume. During this process, fission gas bubbles form inside the grains as very fine bubbles and on the grain boundaries as larger bubbles. The grain boundary bubbles contribute to the majority of the gas swelling.

A schematic representation of the PAD5 fission gas swelling model is shown in Figure 5.8-1. Two significant mechanisms are modeled, depending on the temperature of the fuel. [

] <sup>a,c</sup> This model is consistent with fission gas bubble/porosity data from unrestrained fuel (Section 2.9.3 of Reference 5.13) and restrained fuel (Reference 5.14).

a,c

**Figure 5.8-1 Schematic of PAD5 Fission Gas Swelling Porosity Distribution**

The PAD5 gas swelling is given by:

a,c

(5-39)

[

] a,c

Some irradiation time is needed for fission gas build up and bubble growth. [

]<sup>a,c</sup>

For conditions allowing sufficient time for gas bubbles to develop (i.e., steady state operation or a ramp with a long hold period at high power), the burnup effect on the gas bubble swelling is modeled through BPORCM as:

[

]<sup>a,c</sup>

(5-40)

[

]<sup>a,c</sup>

For short transients, there is not enough time for the fission gas bubble to fully develop, and there will be no additional gas swelling at lower burnup and reduced gas swelling at high burnup compared to long ramps. The burnup effect on the gas bubble swelling during a short transient is modeled through BPORCM as:

[

]<sup>a,c</sup>

(5-41)

where:

[

]<sup>a,c</sup>



The fission gas swelling model is calibrated in Appendix A to the diameter change measurements from various ramp test programs at higher burnup when relocation is expected to be fully accommodated (i.e., some RISO III and Studsvik Clad Integrity Program (SCIP) ramp test rods). The calibrated BPORC, FRBUP, XBUP, FGSWFACT, BPORCTR, XBUPTR1, and XBUPTR2 are:

[

]<sup>a,c</sup>

5.9 ROD AXIAL GROWTH MODEL

*Rod growth model for Zircaloy-4 is the same as in PAD 4.0 (Reference 5.1). Rod growth models for ZIRLO and **Optimized ZIRLO** clad fuel are revised for PAD5.*

Separate rod growth models were developed for ZIRLO cladding, **Optimized ZIRLO** cladding, and Zircaloy-4 cladding materials.

5.9.1 ZIRLO Cladding Material

[

]<sup>a,b,c</sup>

Table 5.9-1 ZIRLO Cladding Fuel Rod Growth Data Base	
Plant	Number of Measurements

a,b,c

Table 5.9-1 ZIRLO Cladding Fuel Rod Growth Data Base (cont.)	
Plant	Number of Measurements

The PAD5 ZIRLO cladding rod growth model [

]<sup>a,c</sup>

Equations (5-42) through (5-44) describe the ZIRLO cladding rod growth model developed.

[

]

a,c

(5-42)

[

]

a,c

(5-43)

[

]

a,c

(5-44)

where:

- $(\Delta L/L)_{BE}$

=

Best Estimate (BE) fuel rod average growth (%)
- $(\Delta L/L)_{UB}$

=

Upper Bound (UB) (95% bound) fuel rod average growth (%)
- $(\Delta L/L)_{LB}$

=

Lower Bound (LB) (95% bound) fuel rod average growth (%)
- $\Phi$

=

fuel rod average fast fluence ( $E > 1$  MeV) ( $10^{21} \times \text{n/cm}^2$ )

The PAD5 ZIRLO cladding rod growth model is plotted with the ZIRLO cladding rod growth data in Figure 5.9-1. The predicted minus measured ZIRLO cladding growth as a function of fast fluence in Figure 5.9-2 confirms good agreement over the whole fast fluence range.

a,b,c

**Figure 5.9-1      ZIRLO Cladding Rod Growth Data and PAD5 ZIRLO Cladding Rod Growth Model**



**Figure 5.9-2      PAD5 ZIRLO Cladding Rod Growth Model Predicted Minus Measured Residual Plot**

### **5.9.2    *Optimized ZIRLO* Cladding Material**

[

] <sup>a,b,c</sup>

Table 5.9-2 *Optimized ZIRLO Cladding Fuel Rod Growth Data Base*

Plant	Number of Measurements	a,b,c

The PAD5 **Optimized ZIRLO** cladding growth model has the same form as the PAD 4.0 rod growth model, but the coefficients are adjusted to get the PAD5 **Optimized ZIRLO** cladding growth model.

Equations (5-45) through (5-47) describe the **Optimized ZIRLO** cladding growth model developed.

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \\ \text{ } \end{array} \right] \begin{array}{c} a,c \\ \text{ } \\ \text{ } \end{array} \quad (5-45)$$

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \\ \text{ } \end{array} \right] \begin{array}{c} a,c \\ \text{ } \\ \text{ } \end{array} \quad (5-46)$$

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \\ \text{ } \end{array} \right] \begin{array}{c} a,c \\ \text{ } \\ \text{ } \end{array} \quad (5-47)$$

where:

$(\Delta L/L)_{BE}$  = Best Estimate (BE) fuel rod average growth (%)

$(\Delta L/L)_{UB}$  = Upper Bound (UB) (95% bound) fuel rod average growth (%)

$(\Delta L/L)_{LB}$  = Lower Bound (LB) (95% bound) fuel rod average growth (%)

$\Phi$  = fuel rod average fluence ( $E > 1\text{MeV}$ ) ( $10^{21} \times \text{n/cm}^2$ )

The PAD5 **Optimized ZIRLO** cladding growth model is plotted with the measured **Optimized ZIRLO** cladding growth as a function of fast fluence in Figure 5.9-3. Predicted minus measured **Optimized ZIRLO** cladding growth is plotted as a function of fast fluence in Figure 5.9-4.



**Figure 5.9-3**      *Optimized ZIRLO* Cladding Rod Growth Data and PAD5 ZIRLO Cladding Rod Growth Model

**Figure 5.9-4      PAD5 *Optimized ZIRLO* Cladding Rod Growth Model Predicted Minus Measured Residual Plot**

### 5.9.3 Zircaloy-4 Cladding Material

The Zircaloy-4 cladding growth model was developed in PAD 3.4 (Reference 5.7) by regression analysis of peripheral rod growth data from assemblies irradiated in the Plant AB, Plant E, and Plant AC reactors. The rod growth data base is summarized in Table 5.9-3. This data base extends to fast fluence levels of  $11.8 \times 10^{21} \text{ n/cm}^2$ ,  $E > 1 \text{ MeV}$ . The rod growth model for Zircaloy-4 cladding is:

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \\ \text{ } \end{array} \right] \quad \text{a,c} \quad (5-48)$$

where:

$$\begin{aligned} \frac{\Delta L}{L} &= \text{fractional rod length change (\%)} \\ \Phi &= \text{neutron fast fluence (n/cm}^2\text{, } E > 1 \text{ MeV)} \end{aligned}$$

Table 5.9-3 Zircaloy-4 Cladding Fuel Rod Growth Data Base						
Reactor	Assembly	Creepdown Data				
		EOC1	EOC2	EOC3	EOC4	EOC5

a,b,c

## 5.10 SECTION 5 REFERENCES

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  - 5.6 Regulatory Guide 1.126, Revision 2, "An Acceptable Model and Related Statistical Methods for the Analysis of Fuel Densification," March 2010.
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  - 5.8 DOE/ET/34010-10, CEND-414, "The Evaluation and Demonstration of Methods for Improved Fuel Utilization," October 1983.
  - 5.9 DOE/ET/34030-11, CEND-427, "Hot Cell Examination of Extended Burnup Fuel From Fort Calhoun," September 1986.
  - 5.10 DOE/ET/34013-15 UC-523, CEND-432, "The Evaluation and Demonstration of Methods for Improved Nuclear Fuel Utilization, Final Report," August 1994.
  - 5.11 EPRI TR-103302-V2, CE NPSD-780-P, "Hot Cell Examination of Extended Burnup Fuel from Calvert Cliffs-1, Final Report Volume 2," May 1994.
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  - 5.13 NUREG/CR-6150, Volume 4, Rev. 2, INEL-96/0422, "SCDAP/RELAP5/MOD3.3 Code Manual: MATPRO - A Library of Materials Properties for Light-Water-Reactor Accident Analysis," January 2001.
  - 5.14 M. Mogensen, et al, "Local Fission Gas Release and Swelling in Water Reactor Fuel During Slow Power Transients," Journal of Nuclear Materials, 131 (1985) 162-171.

## 6 FUEL AND CLAD MATERIAL PROPERTIES

Section 6.1 describes the PAD5  $\text{UO}_2$ ,  $\text{UO}_2\text{-Gd}_2\text{O}_3$ , and  $\text{UO}_2\text{-Er}_2\text{O}_3$  fuel material properties for:

- Density,
- Thermal conductivity,
- Thermal expansion,
- Specific heat,
- Melting point,
- Young's modulus, and
- Poisson's ratio.

Section 6.2 describes PAD5 treatment of  $\text{ZrB}_2$  coating on  $\text{UO}_2$  fuel.

Section 6.3 provides descriptions of the Zircaloy, ZIRLO and **Optimized ZIRLO** cladding models for:

- Density,
- Thermal conductivity,
- Thermal expansion,
- Specific heat and enthalpy,
- Young's modulus,
- Poisson's ratio,
- Yield strength,
- Ultimate tensile strength,
- Irradiation hardening

The fill gas and fission gas thermal conductivity expressions are provided in Section 6.4.

Lastly, the applicability of the material property models to rod average burnup up to [ ]<sup>a,c</sup> is supported in Section 6.5.

### 6.1 FUEL: $\text{UO}_2/\text{UO}_2\text{-GD}_2\text{O}_3/\text{UO}_2\text{-ER}_2\text{O}_3$

#### 6.1.1 Density

*The density for  $\text{UO}_2$  and  $\text{UO}_2\text{-Gd}_2\text{O}_3$  is the same as what is in PAD 4.0. The density for  $\text{UO}_2\text{-Er}_2\text{O}_3$  is new for PAD5, but the same as what was used in FATES3B.*

$\rho_o$  theoretical density (T. D.,  $\text{g/cm}^3$ )

**$\text{UO}_2$  (Reference 6.1)**

$$\rho_o = 10.96 \text{ g/cm}^3 \quad (6-1)$$

**UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub>**

$$[ \quad ]^{a,c} \quad (6-2)$$

$$[ \quad ]^{a,c}$$

**UO<sub>2</sub>-Er<sub>2</sub>O<sub>3</sub> (Reference 6.3)**

$$[ \quad ]^{a,c} \quad (6-3)$$

$$[ \quad ]^{a,c}$$

**6.1.2 Thermal Conductivity**

*The model for UO<sub>2</sub> and UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub> are updated for PAD5. The model for UO<sub>2</sub>-Er<sub>2</sub>O<sub>3</sub> is new for PAD5.*

**UO<sub>2</sub>**

The fuel thermal conductivity model used in PAD 4.0 (Reference 6.1) was updated to incorporate thermal conductivity degradation with burnup. The burnup dependent term, f(BU), is modeled after the [  $\quad ]^{a,c}$ , to account for thermal conductivity degradation as a function of burnup.

The model coefficients were updated based on calibration to measured fuel centerline temperatures. The updated model is:

$$\left[ \quad \right]^{a,c} \quad (6-4)$$

where:

$$[ \quad ]^{a,c} \quad (6-5)$$

K<sub>95</sub> = thermal conductivity for fuel with 95% theoretical density (W/cm-°C)

Bu = local burnup (GWd/MTU)

TC = fuel temperature (°C)

For thermal conductivity for fuel with any other density,

$$K_{density} = \frac{1-P}{1+\beta P} K_{100} = \left( \frac{1-P}{1+\beta P} \right) \times 1.08 \times K_{95} \quad (6-6)$$

where:

P = Fractional porosity of the fuel ring (1-density)

β = a coefficient which is equal to 0.5 for P ≤ 0.05 and which is equal to 1.0 for P > 0.05

K<sub>100</sub> = thermal conductivity for fuel with 100% theoretical density

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Additional details on the calibration of PAD5 to measured fuel centerline temperature data are presented in Appendix A.

### **UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub> and UO<sub>2</sub>-Er<sub>2</sub>O<sub>3</sub>**

The thermal conductivity model for fresh urania-gadolinia/erbium used in PAD5 is developed from the same data and using the same methodology as was the CE urania- gadolinia/erbium thermal conductivity model licensed for use in the FATES3B code (Reference 6.3).

The PAD5 model is based on measured thermal conductivity data of urania/erbium and verified for gadolinium using the same measured gadolinium thermal conductivity data used to bench mark the FATES3B urania-erbium/gadolinium thermal conductivity model.

Figure 6.1-1 compares the predicted values versus the measured data for the urania-erbium/gadolinium model. Excellent agreement was obtained between the measured and predicted values. Figure 6.1-2 and 6.1-3 are plots of the residuals (measured minus predicted value) versus erbium/gadolinium content and temperature.



**Figure 6.1-1 Predicted vs. Measured Data for the PAD5 Urania-erbia/Gadolinia Thermal Conductivity Model**

a,b,c

**Figure 6.1-2 Residuals (Measured Minus Predicted Values) Versus Erbium/Gadolinia Content**



**Figure 6.1-3      Residuals (Measured Minus Predicted Values) Versus Temperature**

The burnup dependent term for the urania-erbia/gadolinia model is the same as for the urania model in Equation (6-5). The final derived urania-erbia/gadolinia thermal conductivity model for PAD5 is:

$$\left[ \frac{\Delta L}{L_o} = 6.797 \times 10^{-6} T + 2.896 \times 10^{-9} T^2 - 1.723 \times 10^{-4} \right]^{a,c} \quad (6-7)$$

### 6.1.3 Thermal Expansion

*This model is the same as what is in PAD 4.0*

#### UO<sub>2</sub> (Reference 6.1)

$$\Delta L/L_o = 6.797 \times 10^{-6} T + 2.896 \times 10^{-9} T^2 - 1.723 \times 10^{-4} \text{ (in/in)} \quad (6-8)$$

where T is temperature, in °C, for 25°C < T < melting temperature.

#### UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub>

Thermal expansion of UO<sub>2</sub> with low Gd<sub>2</sub>O<sub>3</sub> concentrations is approximated by that of pure UO<sub>2</sub> (Reference 6.2), which is:

$$\Delta L/L_o = 6.797 \times 10^{-6} T + 2.896 \times 10^{-9} T^2 - 1.723 \times 10^{-4} \text{ (in/in)} \quad (6-9)$$

for 25°C ≤ T ≤ melting temperature, and [ ]<sup>a,c</sup>

#### UO<sub>2</sub>-Er<sub>2</sub>O<sub>3</sub> (Reference 6.3)

Considering the similarity between erbia and gadolinia, the thermal expansion of a urania-erbia mixture is concluded to be the same as that of urania gadolinia mixtures and, therefore, identical to the thermal expansion of pure urania (Reference 6.1):

$$\Delta L/L_o = 6.797 \times 10^{-6} T + 2.896 \times 10^{-9} T^2 - 1.723 \times 10^{-4} \text{ (in/in)} \quad (6-10)$$

for 25°C < T < melting temperature, and [ ]<sup>a,c</sup>



### 6.1.4 Specific Heat

*This model is the same as what is used with PAD 4.0.*

#### UO<sub>2</sub> (Reference 6.1)

The relationship derived by Kerrisk and Clifton (Reference 6.6) is used for calculating the specific heat for UO<sub>2</sub>.

$$C_p(UO_2) = \frac{5.486 \times 10^6 \exp\left(\frac{535.3}{T}\right)}{T^2 \times (\exp\left(\frac{535.3}{T}\right) - 1)^2} + 1.569 \times 10^{-3} \times T + \frac{1.0707 \times 10^{11}}{T^2} \exp\left(\frac{-1.897 \times 10^4}{T}\right) \text{ (cal/mole} \cdot \text{K)} \quad (6-11)$$

T = temperature, K, for 298 K < T < melting temperature

#### UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub>

Specific heat determinations for urania with small additions of gadolinia indicate a small increase over the specific heat of urania. These [ ]<sup>a,c</sup>. The relationship for the specific heat of UO<sub>2</sub>-Gd<sub>2</sub>O<sub>3</sub> solid solutions is given below (Reference 6.2):

$$[ ]^{a,c} \quad (6-12)$$

where:

$$\begin{aligned} C_p(UO_2-Gd_2O_3) &= \text{Specific heat of } UO_2-Gd_2O_3 \text{ mixture, cal/mole} \cdot \text{K} \\ W_g &= \text{Weight percent of } Gd_2O_3, [ ]^{a,c} \\ C_p(UO_2) &= \text{Specific heat of } UO_2, \text{ cal/mole} \cdot \text{K, and is given in Equation (6-11):} \end{aligned}$$

#### UO<sub>2</sub>-Er<sub>2</sub>O<sub>3</sub>

The specific heat of dilute mixtures of urania-erbia is expected to behave similarly to the urania-gadolinia data (Reference 6.3). Hence, a [ ]<sup>a,c</sup> is applicable for calculating the specific heat of urania-erbia mixtures.

$$[ ]^{a,c} \quad (6-13)$$

where:

$$\begin{aligned} C_p(UO_2-Er_2O_3) &= \text{Specific heat of } UO_2-Er_2O_3 \text{ mixture, cal/mole} \cdot \text{K} \\ W_E &= \text{Weight percent of } Er_2O_3, [ ]^{a,c} \\ C_p(UO_2) &= \text{Specific heat of } UO_2, \text{ cal/mole} \cdot \text{K, and is given in Equation (6-11)} \end{aligned}$$

### 6.1.5 Melting Point

The melting point for fresh  $\text{UO}_2$  is the same as what is used with PAD 4.0. [

]

#### $\text{UO}_2$ (Reference 6.1)

The improved melting temperature model with an updated burnup dependence term (References 6.7 and 6.8) is used in PAD5:

$$T_m = 2805 - 0.5 \text{ Bu} \quad (6-14)$$

where:

$T_m$  = Melting temperature in  $^{\circ}\text{C}$   
 $\text{Bu}$  = Burnup in GWd/MTU.

#### $\text{UO}_2\text{-Gd}_2\text{O}_3$

The published experimental data indicate small  $\text{Gd}_2\text{O}_3$  additions to  $\text{UO}_2$  only slightly depress the melting temperatures. The following conservative correlation is used for the melting point of a  $\text{UO}_2\text{-Gd}_2\text{O}_3$  solid solution (Reference 6.2):

$$T_m = 2805 - 5.8W_g - 0.5\text{Bu} \quad (6-15)$$

where:

$W_g$  = wt. %  $\text{Gd}_2\text{O}_3$ , for [ ]<sup>a,c</sup>  
 $T_m$  = temperature in  $^{\circ}\text{C}$   
 $\text{Bu}$  = fuel burnup in GWd/MTU.

#### $\text{UO}_2\text{-Er}_2\text{O}_3$

The expression for melt temperature as a function of erbia content (Reference 6.2 and 6.3) and burnup (References 6.7 and 6.8) is:

$$T_{\text{melt, erbia}} = 2805 - 5.8W_E - 0.5\text{Bu} \quad (6-16)$$

where:

$T_{\text{melt, erbia}}$  is the melting point of urania-erbia,  $^{\circ}\text{C}$   
 $W_E$  = wt% erbia, [ ]<sup>a,c</sup>  
 $\text{Bu}$  = burnup in GWd/MTU.

### 6.1.6 Young's Modulus

*This model is the same as what is in PAD 4.0.*

The model for the Young's modulus of  $\text{UO}_2$  (Reference 6.1) is given as follows:

$$E = (32.548 \times 10^6 - 1.859 \times 10^3 TF_o)(1 - 2.277P_o) \text{ (psi)} \quad (6-17)$$

where:

$P_o$  is the volume fraction of porosity;

$TF_o$  is the temperature in  $^{\circ}\text{F}$

The Young's modulus for  $\text{UO}_2\text{-Gd}_2\text{O}_3$  and  $\text{UO}_2\text{-Er}_2\text{O}_3$  fuel is the same as for  $\text{UO}_2$  fuel. The Young's modulus depends on the inter-atomic forces between neighboring atoms of the major ceramic component.

### 6.1.7 Poisson's Ratio

*This model is the same as what is in PAD 4.0.*

The recommended model for the Poisson's Ratio of  $\text{UO}_2$  (Reference 6.1) is given as follows:

$$\mu = (0.361 - 1.439P_o) / (1 - 1.660P_o) \quad (6-18)$$

where:

$P_o$  is the porosity fraction

Poisson's ratio for  $\text{UO}_2\text{-Gd}_2\text{O}_3$  and  $\text{UO}_2\text{-Er}_2\text{O}_3$  fuel is the same as for  $\text{UO}_2$  fuel. The Poisson's ratio depends on the inter-atomic forces between neighboring atoms of the major ceramic component.

## 6.2 ZIRCONIUM DIBORIDE INTEGRAL FUEL BURNABLE ABSORBER (IFBA)

Zirconium diboride ( $\text{ZrB}_2$ ) Integral Fuel Burnable Absorber (IFBA) can be used in all of the Westinghouse fuel rod designs. The  $\text{ZrB}_2$  is applied as a thin coating layer onto the outer surface of the uranium dioxide ( $\text{UO}_2$ ) fuel pellet stack prior to loading into the fuel rod cladding tubes rather than being mixed with the  $\text{UO}_2$  as is done with other absorber materials (e.g., erbia or gadolinia). As the  $\text{B}^{10}$  isotope in the  $\text{ZrB}_2$  IFBA absorber burns out, the fuel rod is left with no residual absorber, unlike other absorber materials like erbium or gadolinium. The burnout of the  $\text{B}^{10}$  absorber results in production of helium gas which is released into the fuel rod plenum, increasing internal gas pressure. [ <sup>a,c</sup>]. The helium production effect on internal gas pressure and gas conductivity is taken into account in the design and safety evaluations for Westinghouse fueled PWRs. Neutronics codes already contain the capability to predict neutronics related behavior of the  $\text{ZrB}_2$  IFBA absorber. In the PAD code only a  $\text{ZrB}_2$  IFBA helium generation and release model is required. Although PAD predicted fuel rod internal conditions (pressures, temperatures, etc.) are  $\text{ZrB}_2$  IFBA

specific for input to other analyses, no coding modifications are required for other design and safety analysis codes.

The ZrB<sub>2</sub> coating currently used in all Westinghouse ZrB<sub>2</sub> IFBA designs contains boron that has been enriched in the B<sup>10</sup> isotope. [

] <sup>a,c</sup> The ZrB<sub>2</sub> IFBA coating is applied over the center of the UO<sub>2</sub> pellet stack length and does not extend to either end of the fuel rod. The ends without ZrB<sub>2</sub> IFBA are referred to as cutback regions. The fuel pellets in the cutback regions may be solid, annular, or a combination of solid and annular geometry (i.e., solid pellet at the bottom of the pellet stack with annular pellets at the top of the pellet stack) and may be at reduced U<sup>235</sup> enrichment (blankets). However, the ZrB<sub>2</sub> IFBA coating is applied only to the central solid fuel pellet stack. ZrB<sub>2</sub> IFBA fuel rods are loaded into an assembly in specific fuel rod locations as a matrix of ZrB<sub>2</sub> IFBA and UO<sub>2</sub> fuel rods.

The thickness of the ZrB<sub>2</sub> IBFA coating [

] <sup>a,c</sup>

The stability of the B<sup>10</sup> depleted ZrB<sub>2</sub> layer has been confirmed through hotcell examinations of commercial rods ranging in burnup from [ <sup>a,c</sup>. No increase in pellet roughness is observed in irradiated ZrB<sub>2</sub> coated fuel compared to UO<sub>2</sub> fuel of similar burnups.

## 6.3 CLADDING: ZIRCALOY/ZIRLO/OPTIMIZED ZIRLO

### 6.3.1 Density

*This model is the same as what is in PAD 4.0.*

The density (References 6.11 and 6.12) for Zircaloy, ZIRLO or **Optimized ZIRLO** alloy at 25°C is:

$$\rho_o = 6.56 \pm 0.02 \text{ g/cm}^3 \quad (6-19)$$

### 6.3.2 Thermal Conductivity

*This model is the same as what is in PAD 4.0.*

#### Zircaloy-4 (Reference 6.13)

$$[ \quad ]^{a,c} \quad (6-20)$$

where:

$$\begin{aligned} k &= \text{thermal conductivity in (W/cm-}^{\circ}\text{C)} \\ T &= \text{temperature in } ^{\circ}\text{C, [ } \quad \quad \quad ]^{a,c} \end{aligned}$$

### **ZIRLO Cladding/Optimized ZIRLO cladding (References 6.11, 6.12 and 6.13)**

$$[ \quad \quad \quad ]^{a,c} \quad (6-21)$$

where:

$$\begin{aligned} k &= \text{thermal conductivity in (W/cm-}^{\circ}\text{C)} \\ T &= \text{temperature in } ^{\circ}\text{C, [ } \quad \quad \quad ]^{a,c} \end{aligned}$$

### **CRUD (Reference 6.14)**

The following are typical values used in PAD5.

$$K_f=0.5 \text{ (BTU/hr-ft-}^{\circ}\text{F)} \text{ for forced convection without boiling (Reference 6.15)} \quad (6-22)$$

$$K_b=4.0 \text{ (BTU/hr-ft-}^{\circ}\text{F)} \text{ for nucleate boiling through a crudded surface (Reference 6.16)} \quad (6-23)$$

$$K_b=0 \text{ (BTU/hr-ft-}^{\circ}\text{F)} \text{ for nucleate boiling from an essentially clean surface (Reference 6.17)} \quad (6-24)$$

### **ZrO<sub>2</sub>**

For thermal calculations (Reference 6.13 and 6.18):

$$[ \quad \quad \quad ]^{a,c} \quad (6-25)$$

### **6.3.3 Thermal Expansion**

*This model is the same as what is in PAD 4.0.*

The thermal expansion models for Zircaloy-4, ZIRLO cladding and **Optimized ZIRLO** cladding are (References 6.1, 6.9, 6.11 and 6.12):

$$\frac{\Delta L}{L}(\text{axial}) = (2.92 \times 10^{-6}) \times (T - 70) \quad (6-26)$$

$$\frac{\Delta R}{R_0}(\text{radial}) = (3.22 \times 10^{-6})(T - 70) \quad (6-27)$$

where:

T is temperature in °F,  $70^{\circ}\text{F} < T < 800^{\circ}\text{F}$

### 6.3.4 Specific Heat and Enthalpy

*This model is the same as what is used with PAD 4.0.*

For approximate calculations, the following values are used up to the temperature where the alpha-to-beta phase change starts to occur. The specific heat of ZIRLO cladding is different from that of Zircaloy-4 primarily because of the difference in the temperature range over which the alpha-to-beta phase change occurs for the two alloys.

For Zircaloy-4 (References 6.1 and 6.18):

#### Enthalpy

$$H_0^t = -0.379 + 0.0738T \text{ (cal/g), } T \text{ in } ^{\circ}\text{C} \quad (0^{\circ}\text{--}492^{\circ}\text{C}), (32^{\circ}\text{--}918^{\circ}\text{F}) \quad (6-28)$$

$$H_0^t = -6.607 + 0.08648T \text{ (cal/g), } T \text{ in } ^{\circ}\text{C} \quad (492^{\circ}\text{--}823^{\circ}\text{C}), (918^{\circ}\text{--}1513^{\circ}\text{F}) \quad (6-29)$$

#### Specific Heat

$$C_p = 0.0738 \text{ (cal/(g}\cdot\text{K))} \quad (0^{\circ}\text{--}492^{\circ}\text{C}), (32^{\circ}\text{--}918^{\circ}\text{F}) \quad (6-30)$$

$$C_p = 0.08648 \text{ (cal/(g}\cdot\text{K))} \quad (492^{\circ}\text{--}823^{\circ}\text{C}), (918^{\circ}\text{--}1513^{\circ}\text{F}) \quad (6-31)$$

For ZIRLO cladding and **Optimized ZIRLO** cladding (References 6.11 and 6.12):

#### Enthalpy

$$H_0^t = -0.379 + 0.073T \text{ (cal/g), } T \text{ in } ^{\circ}\text{C} \quad (0^{\circ}\text{--}516^{\circ}\text{C}), (32^{\circ}\text{--}961^{\circ}\text{F}) \quad (6-32)$$

$$H_0^t = -6.607 + 0.0878T \text{ (cal/g), } T \text{ in } ^{\circ}\text{C} \quad (516^{\circ}\text{--}751^{\circ}\text{C}), (961^{\circ}\text{--}1384^{\circ}\text{F}) \quad (6-33)$$

#### Specific Heat

$$C_p = 0.073 \text{ (cal/(g}\cdot\text{K))} \quad (0^{\circ}\text{--}516^{\circ}\text{C}), (32^{\circ}\text{--}961^{\circ}\text{F}) \quad (6-34)$$

$$C_p = 0.0878 \text{ (cal/(g}\cdot\text{K))} \quad (516^{\circ}\text{--}751^{\circ}\text{C}), (961^{\circ}\text{--}1384^{\circ}\text{F}) \quad (6-35)$$

### 6.3.5 Young's Modulus

*This model is the same as what is in PAD 4.0.*

For temperatures from 0° to 1200°F, the Young's modulus for Zircaloy-4, ZIRLO cladding and **Optimized ZIRLO** cladding is (References 6.1, 6.9, 6.12 and 6.20):

Circumferential elastic modulus:

$$[ \quad ]^{a,c} \quad (6-36)$$

Longitudinal elastic modulus:

$$[ \quad ]^{a,c} \quad (6-37)$$

### 6.3.6 Poisson's Ratio

*This model is the same as what is in PAD 4.0.*

For temperatures from 0° to 1200°F, the Poisson's ratio for Zircaloy-4, ZIRLO cladding and **Optimized ZIRLO** cladding is (References 6.1, 6.9, 6.12 and 6.20):

for the Poisson's ratio (circumferential strain /imposed longitudinal strain):

$$[ \quad ]^{a,c} \quad (6-38)$$

for the Poisson's ratio ( radial strain/imposed longitudinal strain):

$$[ \quad ]^{a,c} \quad (6-39)$$

### 6.3.7 Yield Strength

*This model is the same as what is in PAD 4.0.*

#### a. Cold Worked Stress Relieved Anneal (SRA) Zircaloy-4

1. Best estimate unirradiated (Reference 6.1)

$$\sigma_y = 92300 - 52.571 \text{ TF (psi)} \quad \text{for TF} \leq 700^\circ\text{F} \quad (6-40)$$

$$\sigma_y = 137750 - 117.500 \text{ TF (psi)} \quad \text{for } 700 < \text{TF} \leq 900^\circ\text{F} \quad (6-41)$$

2. Best estimate irradiated (Reference 6.1)

$$[ \quad \quad \quad ]^{a,c} \quad (6-42)$$

$$[ \quad \quad \quad ]^{a,c} \quad (6-43)$$

$$[ \quad \quad \quad ]^{a,c}$$

**b. Cold Worked Stress Relieved Anneal (SRA) ZIRLO Cladding**

1. Best estimate unirradiated (Reference 6.9)

$$[ \quad \quad \quad ]^{a,c} \quad (6-44)$$

$$[ \quad \quad \quad ]^{a,c} \quad (6-45)$$

2. Best estimate irradiated

$$[ \quad \quad \quad ]^{a,c} \quad (6-46)$$

$$[ \quad \quad \quad ]^{a,c} \quad (6-47)$$

$$[ \quad \quad \quad ]^{a,c}$$

**c. Partially Recrystallized *Optimized* ZIRLO Cladding**

1. Best estimate unirradiated

$$[ \quad \quad \quad ]^{a,c} \quad (6-48)$$

2. Best estimate irradiated

$$[ \quad \quad \quad ]^{a,c} \quad (6-49)$$

$$[ \quad \quad \quad ]^{a,c} \quad (6-50)$$

$$[ \quad \quad \quad ]^{a,c}$$



### 6.3.8 Ultimate Tensile Strength

*This model is the same as what is used with PAD 4.0.*

#### a. Cold Worked Stress Relieved Anneal (SRA) Zircaloy-4

1. Best estimate unirradiated

$$[ \quad ]^{a,c} \quad (6-51)$$

$$[ \quad ]^{a,c} \quad (6-52)$$

2. Best estimate irradiated

$$[ \quad ]^{a,c} \quad (6-53)$$

$$[ \quad ]^{a,c} \quad (6-54)$$

$$[ \quad ]^{a,c}$$

#### b. Cold Worked Stress Relieved Anneal (SRA) ZIRLO Cladding

1. Best estimate unirradiated

$$[ \quad ]^{a,c} \quad (6-55)$$

$$[ \quad ]^{a,c} \quad (6-56)$$

2. Best estimate irradiated

$$[ \quad ]^{a,c} \quad (6-57)$$

$$[ \quad ]^{a,c} \quad (6-58)$$

$$[ \quad ]^{a,c}$$

c. **Partially Recrystallized *Optimized ZIRLO* Cladding**

1. Best estimate unirradiated

$$[\text{ } ]^{a,c} \quad (6-59)$$

2. Best estimate irradiated

$$[\text{ } ]^{a,c} \quad (6-60)$$

$$[\text{ } ]^{a,c} \quad (6-61)$$

$$[\text{ } ]^{a,c}$$

### 6.3.9 Irradiation Hardening

*This model is new for PAD5.*

An irradiation hardening model in the form of,  $[\text{ } ]^{a,c}$ , has been developed based on published data (References 6.22, 6.23, and 6.24) to take into account the phenomenon of irradiation hardening that occurs at the beginning of life. The database contains literature data at different test temperatures and starting material conditions. To simplify data analysis, the data were normalized by  $[\text{ } ]^{a,c}$ . An exponential form,  $[\text{ } ]^{a,c}$ , was selected to approximate the irradiation hardening phenomena. This exponential form is fitted to the normalized data and a coefficient of  $[\text{ } ]^{a,c}$  is obtained. The normalized data along with the curve fit are plotted in Figure 6.3-1. The model can be applied as a multiplier to the fully irradiation hardened mechanical strength model in the following manner:

$$[\text{ } ]^{a,c} \quad (6-62)$$

where:

- $\sigma_o$  is the unirradiated cladding yield strength or ultimate tensile strength (psi or MPa)
- $\sigma_f$  is the fully hardened irradiated cladding yield strength or ultimate tensile strength (psi or MPa)
- $\Phi$  is the fast neutron fluence ( $\text{n/cm}^2$ ,  $E > 1 \text{ MeV}$ )

The model above is applicable to zirconium based alloys with a maximum of 3 percent alloying elements. The same functional form will be used for ZIRLO and **Optimized ZIRLO** cladding.

**Figure 6.3-1 Zircaloy Irradiation Hardening Data Plotted with Proposed Irradiation Hardening Model**

#### **6.4 FILL GAS & FISSION GAS MATERIAL PROPERTIES**

*This model is the same as what is in PAD 4.0.*

The thermal conductivity equations of fission gases xenon and krypton (Reference 6.14) were fit to the correlated data of Gandhi and Saxena (Reference 6.21):

$$K_{Xe} = 1.395 \times 10^{-5} T^{0.872} \text{ BTU /hr-ft-}^{\circ}\text{F} \quad (6-63)$$

$$K_{Kr} = 1.588 \times 10^{-5} T^{0.92331} \text{ BTU /hr-ft-}^{\circ}\text{F} \quad (6-64)$$

where T is in degrees Rankine. It is assumed that [ ]<sup>a,c</sup>.

The thermal conductivity equations for air, nitrogen, and hydrogen used in the PAD5 code are:

$$K_{\text{air}} = 7.35 \times 10^{-5} T^{0.846} \quad \text{BTU /hr-ft-}^{\circ}\text{F} \quad (6-65)$$

$$K_{\text{N}_2} = 7.35 \times 10^{-5} T^{0.846} \quad \text{BTU /hr-ft-}^{\circ}\text{F} \quad (6-66)$$

$$K_{\text{H}_2} = 5.834 \times 10^{-4} T^{0.8213} \quad \text{BTU /hr-ft-}^{\circ}\text{F} \quad (6-67)$$

where:

T is in degrees Rankine.

For fill gas argon, the thermal conductivity is given by:

$$[ \quad ]^{a,c} \quad (6-68)$$

$$[ \quad ]^{a,c} \quad (6-69)$$

The thermal conductivity of fill and fission gas helium is based on the results presented by Tsederberg (Reference 6.4). The equation, which is a function of pressure and temperature, is:

$$[ \quad ]^{a,c} \quad (6-70)$$

where:

T = gas temperature, in degrees Rankine, and

P = gas pressure, in psia.

For the thermal conductivity of water vapor, the steam tables are used.

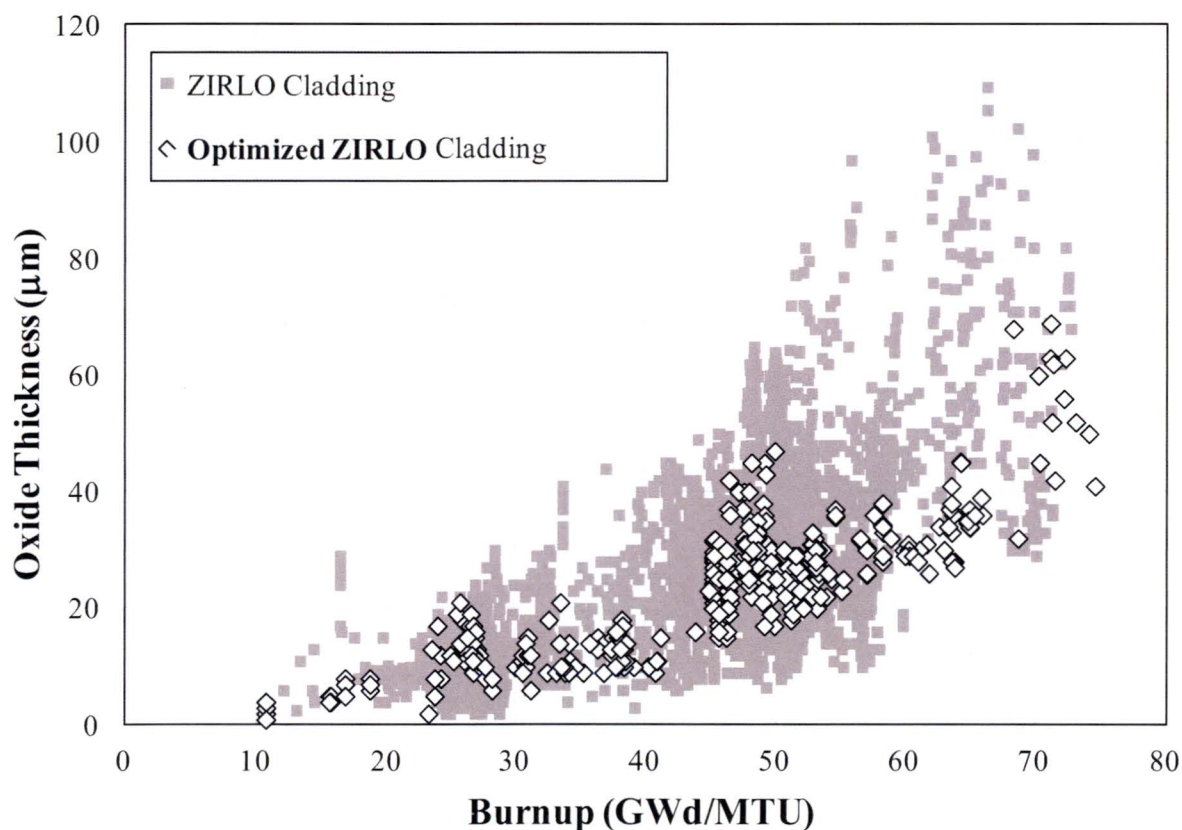
## 6.5 APPLICABILITY OF MATERIAL MODELS TO HIGH BURNUP

The models in PAD5 are applicable to a burnup of [ ]<sup>a,c</sup> and the material performance supports burnups of [ ]<sup>a,c</sup> for **Optimized ZIRLO** clad fuel rods. The materials properties for **Optimized ZIRLO** cladding are applicable to burnups of [ ]<sup>a,c</sup> based on evaluation of post irradiation exam data. The evaluation of poolside and hotcell data collected to a maximum fuel rod burnup of [ ]<sup>a,c</sup> indicates that there is a consistent behavior for the fuel rods in general, but especially with respect to cladding corrosion, cladding yield, ultimate tensile strength (UTS) and cladding ductility.

In Figure 6.5-1, corrosion data for ZIRLO and **Optimized ZIRLO** cladding are plotted as a function of rod average burnup (Reference 6.10). Figure 6.5-2 plots irradiated clad yield strength for ZIRLO and **Optimized ZIRLO** cladding as a function of fast neutron fluence. Figure 6.5-3 plots irradiated clad UTS for ZIRLO and **Optimized ZIRLO** cladding as a function of fast neutron fluence. In the latter two cases there is no distinct trend with increasing burnup (fluence).

Since the maximum waterside oxide thickness associated with the cladding corrosion increases with burnup, the maximum cladding hydrogen content will also increase with burnup. There is some decrease in ductility with increasing hydrogen, but the dominant impact on the cladding irradiated ductility is due to irradiation damage. This can be observed in Figure 6.5-4 where cladding uniform plastic strain is plotted as a function of cladding metal hydrogen content. Only a slight decrease in uniform strain is observed as hydrogen increases.

The evaluation of the data indicates that the **Optimized ZIRLO** cladding will maintain sufficient mechanical capability to insure that no unanticipated failure modes will occur at burnups of [ ]<sup>a,c</sup>.



**Figure 6.5-1**      **ZIRLO and *Optimized* ZIRLO Cladding Corrosion Waterside Oxide Thickness as a Function of Burnup**

a,b,c

**Figure 6.5-2     *ZIRLO* and *Optimized ZIRLO* Cladding High Temperature Yield Stress as a Function of Rod Average Fast Fluence**

a,b,c

**Figure 6.5-3      ZIRLO and *Optimized ZIRLO* Cladding High Temperature Ultimate Tensile Strength as a Function of Rod Average Fast Fluence**

a,b,c

**Figure 6.5-4      ZIRLO and *Optimized ZIRLO* Cladding Uniform Plastic Strain as a Function of Hydrogen Content of the Cladding Metal**



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## **7 FUEL ROD DESIGN CRITERIA AND EVALUATION METHODOLOGY**

### **7.1 FUEL ROD DESIGN SCOPE**

This chapter describes the Westinghouse analysis methods that will be used with PAD5 in fuel rod design and safety analysis input calculations. With the implementation of PAD5, the fuel rod design processes for both Westinghouse Nuclear Steam Supply System (NSSS) and Combustion Engineering (CE) NSSS applications are integrated, so that the fuel rod design bases and criteria presented in this chapter will apply to all Westinghouse United States (US) Pressurized Water Reactor (PWR) fuel rod analyses. These design criteria and associated application methods supersede prior approved design criteria applied to Westinghouse and CE NSSS plant analyses when PAD5 is applied unless otherwise noted.

PAD5 is the primary Westinghouse fuel rod design tool. It incorporates all relevant fuel performance phenomena, including fuel thermal conductivity degradation with burnup and enhanced fission gas bubble swelling at high burnup, as an integrated set of interrelated performance models. Using appropriate input describing fuel rod design and operating conditions, PAD5 calculates key fuel performance parameters such as cladding stress, strain, oxidation and hydriding, fuel temperature and volume changes, and rod internal pressure.

The primary objectives of the fuel design and safety analyses are to provide assurance that:

1. The fuel system is not damaged as a result of normal operation and anticipated operational occurrences;
2. Fuel system damage is never so severe as to preclude control rod insertion when required;
3. The number of fuel rod failures is not underestimated for postulated accidents; and
4. Core coolability is always maintained.

Standard classifications of reactor operating conditions have been defined in both the SRP Sections 4.2 (Reference 7.1) and 15 (Reference 7.2) and in published American National Standards Institute (ANSI) standards (References 7.3 and 7.4), as summarized in Table 7.1-1. This table also indicates the relationship between the two classification sets as established in the SRP Section 15. As noted in SRP Section 15, either classification set may be used. Westinghouse fuel system designs have typically employed the ANS 51.1 classification which uses four classifications: Condition I for normal operation and Conditions II, III and IV for transient events of varying probability or frequency of occurrence. The breakdown of events by frequency of occurrence as defined in the ANS 51.1 standard is consistent with the event frequency classifications in Chapter 15 of US NRC Regulatory Guide 1.70 (Reference 7.5).

Examples of plant operating conditions or events considered as Condition I, Condition II, Condition III or Condition IV operation are provided in Tables 7.1-2, 7.1-3, 7.1-4 and 7.1-5, respectively. Under the SRP classifications, Condition I is equivalent to Normal Operation (NO), Condition II events are Anticipated

Operational Occurrences (AOOs), Condition III events are either AOOs or Postulated Accidents, and Condition IV events are Postulated Accidents (PA).

General Design Criteria (GDC) 10, in Appendix A to 10 CFR Part 50, states that "The reactor core and associated coolant, control, and protection systems shall be designed with appropriate margin to assure that specified acceptable fuel design limits are not exceeded during any condition of normal operation, including the effects of anticipated operational occurrences." The Westinghouse fuel rod design process assures good fuel performance during Condition I (NO) and Condition II (AOO) operation. Condition III and Condition IV (PA) events, which allow for some fuel rod failure provided that radiological dose limits are satisfied, are addressed as part of the safety analysis using established licensed analysis methods. PAD5 analyses are performed, however, to provide steady state initial conditions such as fuel temperatures, rod internal pressures, and core stored energy, as input to these safety analyses.

The key interfaces which provide input to the fuel rod design or which receive and process results from the fuel rod design are reviewed in Section 7.2. Section 7.3 presents the fundamental principles that are the bases for the fuel rod design analysis methods. Westinghouse PWR fuel rod design limits are defined in Section 7.4 along with a description of the evaluation methods used to verify that these design limits are satisfied. Section 7.5 provides a description of the PAD5 analyses performed to generate input to safety analysis calculations, including the associated analysis methods.

**Table 7.1-1 SRP Section 15 and ANSI18.2 /ANS 51.1 Operating Condition Classifications**

<b>SRP 15 Operating Conditions</b>	<b>ANSI/ANS 51.1 Operating Conditions</b>
Normal Operation (NO) - Frequent and/or expected modes of operation that occur in the normal course of plant operation.	Condition I Operation - Normal operation and operational transients.
Anticipated Operational Occurrences (AOO) - Conditions expected to occur one or more times during the life of a nuclear power plant.	Condition II Operation - Events of moderate frequency, i.e., an event could occur once per cycle of operation.
	Condition III Operation - Infrequent incidents, i.e., an event that may occur on a frequency of once during the life of the plant. Some limited fuel failures may occur but are accounted for in radiological dose calculations.
Postulated Accidents (PA) - Postulated events not expected to occur during the life of a nuclear power plant.	Condition IV Operation - Limiting faults, i.e., design basis events that are not expected to occur during the life of the plant.

**Table 7.1-2 Plant Conditions for Condition I Operation**

Condition I occurrences are operations that are expected frequently or regularly in the course of power operation, refueling, maintenance, or maneuvering of the plant. Examples include:

- Steady-state and shutdown operations
  - Power operation (~15 to 100% of full power)
  - Startup (or standby) (critical, 0 to 15% of full power)
  - Hot shutdown (subcritical, Residual Heat Removal System isolated)
  - Cold shutdown (subcritical, Residual Heat Removal System in operation)
  - Refueling
- Operational transients including load follow
  - Plant heatup and cooldown (up to 100°F/hour for the Reactor Coolant System)
  - Step load changes (up to 10%)
  - Ramp load changes (up to 5%/minute)
  - Load rejection up to and including design load rejection transient

**Table 7.1-3 Condition II Operation – Incidents of Moderate Frequency**

Condition II events shall be accommodated with, at most, a shutdown of the reactor with the plant capable of returning to operation after corrective action. Examples of Condition II events (References 7.5 and 7.6) include:

- Feedwater Malfunctions Causing a Decrease in Feedwater Temperature
- Feedwater Malfunction Causing an Increase in Feedwater Flow
- Excessive Increase in Secondary Steam Flow
- Inadvertent Opening of a Steam Generator (SG) Safety or Relief Valve
- Loss of Electrical Load and/or Turbine Trip
- Loss of Non-Emergency Alternating Current (AC) Power
- Loss of Normal Feedwater
- Partial Loss of Forced Reactor Coolant Flow
- Uncontrolled Rod Cluster Control Assembly (RCCA) Bank Withdrawal from a Subcritical Condition
- Uncontrolled RCCA Bank Withdrawal at Power
- RCCA Mis-Operation (RCCA Misalignment, Rod Drop)
- Startup of an Inactive Reactor Coolant Loop
- Uncontrolled Boron Dilution
- Inadvertent Emergency Core Cooling System (ECCS) Actuation at Power
- Chemical and Volume Control System (CVCS) Malfunction Causing an Increase in Reactor Coolant Inventory
- Inadvertent Opening of a Pressurizer Safety or Relief Valve

**Table 7.1-4 Condition III Faults – Infrequent Incidents**

Condition III occurrences are faults which may occur very infrequently during the life of a plant. Examples of Condition III events (References 7.5 and 7.6) include:

- Minor Reactor Coolant System Pipe Ruptures
- Minor Secondary System Pipe Breaks
- Complete Loss of Forced Reactor Coolant Flow
- Single Rod Cluster Control Assembly Withdrawal at Power
- Inadvertent Loading of a Fuel Assembly in an Improper Location

**Table 7.1-5 Condition IV Faults – Limiting Faults**

Condition IV occurrences are faults which are not expected to take place, but are postulated because their consequences would include the potential for the release of significant amounts of radioactive material. Examples of Condition IV faults (References 7.5 and 7.6) include:

- Major Reactor Coolant System Pipe Ruptures
- Major Rupture of a Main Steam Line
- Major Rupture of a Main Feedwater Pipe
- Single Reactor Coolant Pump Locked Rotor
- Rupture of a Control Rod Drive Mechanism Housing (RCCA Ejection)

## 7.2 FUEL ROD DESIGN INTERFACES REVIEW

Figure 7.2-1 provides a schematic illustration of typical fuel rod design interfaces. Interfaces that provide input to the fuel rod design are described in Section 7.2.1. Interfaces that receive output from the fuel rod design process are described in Section 7.2.2.

a,c

**Figure 7.2-1 Schematic Illustration of Typical Fuel Rod Design Interfaces**

## 7.2.1 Fuel Rod Design Input Interfaces

Design input is required from several sources in order to perform fuel rod design analyses. Standard procedures require that design inputs used in the fuel rod design and safety analyses be appropriately documented and verified for each design analysis. In this section, the primary sources of design related input are identified along with a description of the input data that each discipline provides.

### 7.2.1.1 Product Design & Manufacturing

The fuel rod and fuel assembly mechanical design function establishes the physical parameters required to define the fuel rod. Several components comprise the fuel rod system: the fuel tube, top and bottom end plugs, the fuel pellet stack, spacer components, if required, and pellet stack hold down springs. Figure 7.2-2 illustrates the mechanical components of a fuel rod. While not an integral part of a fuel rod, the fuel assembly structural grid design is also important for establishing, among other things, the flow channel spacing between fuel rods. The assembly design also establishes the spacing between the ends of the fuel rods and the top and bottom nozzles.

The critical dimensions for the components that are needed for PAD5 input are defined in design drawings for each assembly and the components that make up that assembly, including the fuel rods, grids and assembly skeleton. These drawings provide both the nominal dimensions used in fuel rod design and the fabrication geometric tolerances that are considered in the uncertainties analyses. [

] <sup>a,c</sup> The types of fuel pellets contained within each rod are also specified. These would include whether there are solid or annular pellets in the fuel stack and whether the non-blanket fuel pellets are  $\text{UO}_2$ ,  $\text{ZrB}_2$  IFBA coated  $\text{UO}_2$ ,  $\text{Gd}_2\text{O}_3$ - $\text{UO}_2$  or  $\text{Er}_2\text{O}_3$ - $\text{UO}_2$ .

Information on fuel rod fill gas pressure, pellet  $\text{U}^{235}$  enrichment, integral fuel burnable absorber pellet compositions and nominal fuel density and sintering temperature is also obtained from region specific product design documents. Product and process specifications provide data such as fuel density, fuel pellet sintering temperature, and fill gas composition, including allowances for variability.

When analyses are performed for fuel that has already been fabricated, quality assurance inspection data obtained during the fuel fabrication process provide actual component dimensions and associated uncertainties. Measured data are typically available for fuel pellet density, sintering temperature, pellet and cladding diameters, and fuel rod plenum length. Pellet thermal stability testing results may also be available. [

] <sup>a,c</sup>

Westinghouse fuel and cladding fabrication processes are stable and controlled. [

] <sup>a,c</sup>



I

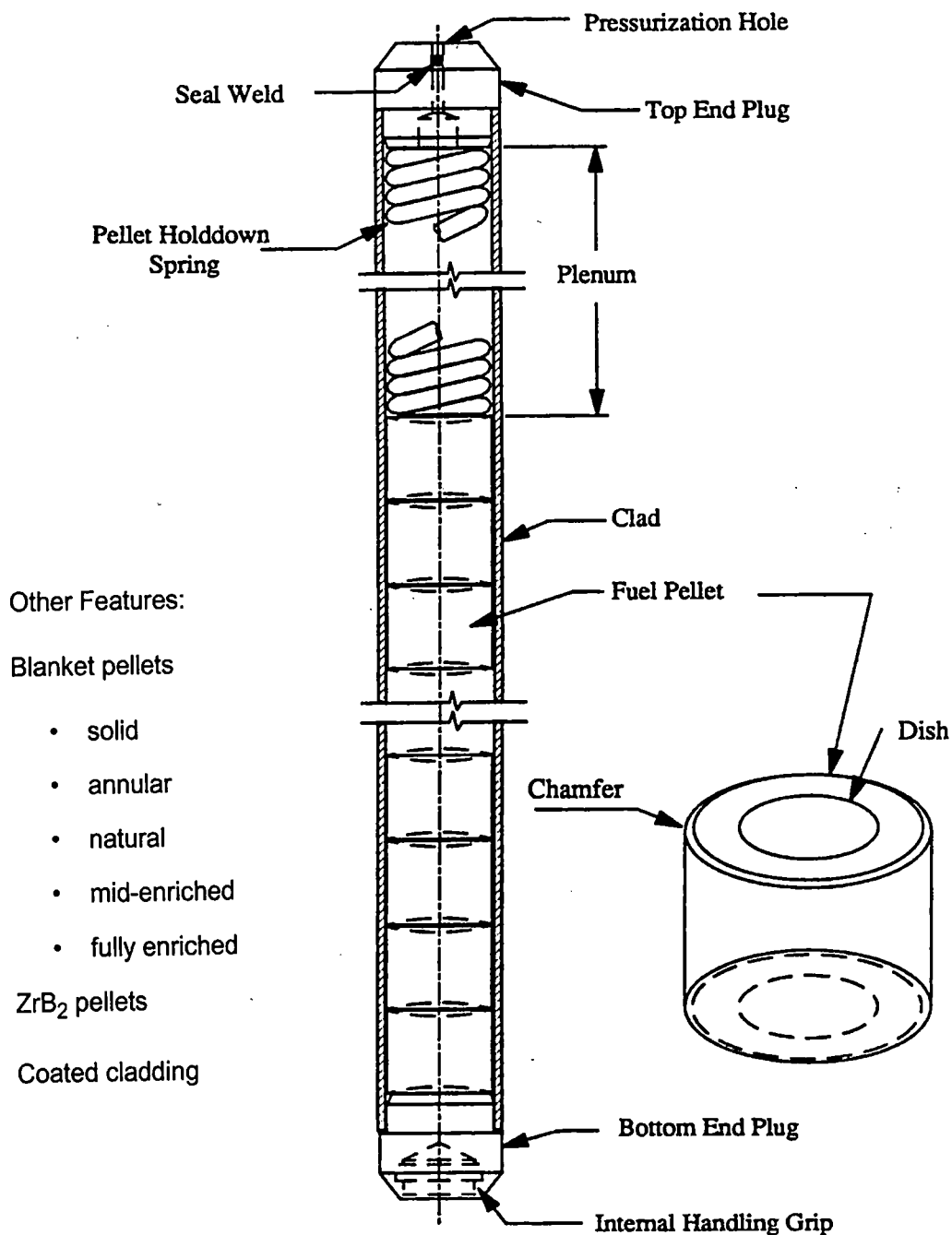
] <sup>a,c</sup>

Figure 7.2-2 Illustration of Typical Westinghouse Fuel Rod Components

### 7.2.1.2 Thermal/Hydraulic Design

The core thermal-hydraulic design provides the design basis plant operating conditions for system pressure, core inlet temperature and coolant mass flow rate. As-operated values for these parameters are also available for past operating cycles from plant operation data. [ ]<sup>a,c</sup> may be plant specific.

### 7.2.1.3 Plant Operation

The fuel rod design accounts for the planned operation of fuel in the design cycle with consideration of the past operating history of the fuel in the design cycle that operated in prior cycles.

Data obtained from prior plant cycles of operation includes measured core power, core coolant flow, coolant inlet temperature and core and vessel coolant average temperatures. [ ]<sup>a,c</sup> may be plant specific.

In the event that a prior operating cycle experienced a Condition II overpower transient, the details of this event are provided so that the impact of the event can be accounted for in the fuel rod design.

### 7.2.1.4 Nuclear Design

The nuclear design provides a substantial portion of the required input for fuel rod design, including the following types of information:

- Fuel management data
- Steady state power history data;
- Transient power duty;
- Fast ( $E > 1.0$  MeV) neutron flux and fluence data; and
- Fuel pellet radial power distribution data.

Each of these information types are described in more detail in the following sub-sections.

#### 7.2.1.4.1 Fuel Management Data

The nuclear design establishes the core loading pattern and the fuel inventory for the design cycle based on cycle specific energy requirements. The nuclear design considers several parameters which define the fuel operation in the design cycle, including the following:

- Current operating cycle planned shutdown and associated burnup windows;
- Energy requirements for the design cycle which establish the planned cycle length;
- Core power level;
- Technical Specification limits on power peaking factors (e.g.,  $F_{\Delta H}$  and  $F_Q$ ), if any;
- Available burned fuel inventory;
- Planned load follow, if any; and
- Plant operation parameters such as core control strategy and axial shape control limits (e.g., Axial Offset (AO), Power Imbalance ( $\Delta I$ ) or Axial Shape Index (ASI) bands).

The fuel loading pattern for the design cycle is defined with consideration of all of these parameters. It specifies the fuel assemblies included in the core inventory, including the number of assemblies from each fuel batch, initial  $U^{235}$  enrichment(s), and burnable absorber ( $ZrB_2$ ,  $Gd_2O_3-UO_2$  or  $Er_2O_3-UO_2$ ) design details. The operating history of each burned fuel assembly (prior cycles in which the fuel assemblies operated) is also derived from the loading pattern for the design cycle in combination with prior cycle loading patterns.

Table 7.2.1-1 provides an example of the fuel management summary data provided by the nuclear design. In this example, burned fuel in the design cycle (Cycle 19) also operated in cycles 17 and 18. The cycle lengths in burnup (MWd/MTU) and effective full power hours (EFPH) are given for each cycle along with the total uranium loading and core average power in each cycle. Other basic design limits, such as Technical Specification peaking factor limits, core axial power distribution limits, such as AO,  $\Delta I$  or ASI bands, and core control strategy, such as Relaxed Axial Offset Control (RAOC) or Constant Axial Offset Control (CAOC), are specified. These parameters are provided for all cycles for which fuel assemblies specified in the design core loading plan have operated.

The design cycle loading pattern specifies the fuel assemblies included in the core inventory, including the number of assemblies in each fuel batch, as illustrated in the example in Table 7.2.1-2. The region ID indicates the specific batches of fuel resident in the design cycle. Using these region identifiers, the specific fuel rod design data for each fuel batch is obtained from the manufacturing history and design drawings.

**Table 7.2.1-1 Example of Nuclear Design Fuel Management Parameters Data**

Cycle	Cycle Burnup	Uranium Loading	Core Power		Time	Tech Spec	Core Control Strategy	
							AO Band	Type
	MWD/MTU	MTU	MWt	kW/ft	EFPH	$F_{AH}$	percent	
17	21137	80.421	3587	5.71	11375	1.7	10. -15.	RAOC
18	24625	80.265	3645	5.81	13014	1.7	10. -15.	RAOC
19	24389	80.399	3645	5.81	12911	1.7	15. -20.	RAOC

<b>Table 7.2.1-2 Example Irradiation History Data Region History</b>	
<b>Region ID</b>	<b>Cycles in Core</b>
19B	17, 18, 19
19C	17, 18, 19
19C	17, 19
20A	18, 19
20B	18, 19
20C	18, 19
21A	19
21B	19
21C	19

#### 7.2.1.4.2 Steady State Power History Data

The fuel rod power history is a key factor in determining the expected performance of each fuel rod. In absolute terms, the power history is the power in each axial segment of the fuel rod defined as a series of time or burnup steps through the fuel rod operation. Typically, the power history is divided into two components as a function of time or burnup: the rod average power and the normalized axial power shape.

Both components of these power history components are calculated in core depletion calculations performed using NRC approved core physics codes (e.g., Westinghouse Advanced Nodal Code (ANC)). The rod average power and axial power shape [

] <sup>a,c</sup> rod power and cycle time for each cycle of operation.

When performing fuel temperature analyses for input to safety analyses, it is important that the power history used to generate maximum fuel temperatures [

] <sup>a,c</sup> are most limiting for the fuel temperature analysis.

Fuel rod inputs generated for safety analyses are typically generated early in the design process to support the design and licensing schedule. [

] <sup>a,c</sup> Figure 7.2.1-1 provides an example of a [

] <sup>a,c</sup> Figure 7.2.1-2 illustrates a comparison of [

] <sup>a,c</sup>

[

]<sup>a,c</sup> for this cycle.

The second component of the power history is the definition of the axial power distribution as a function of time or burnup. Typically, time or burnup dependent axial power distributions are defined [

]<sup>a,c</sup>

a,c

**Figure 7.2.1-1 Example [**

] <sup>a,c</sup>

a,c

**Figure 7.2.1-2 Example [**

] <sup>a,c</sup>

a,c



**Figure 7.2.1-3 Example 4-Loop 17x17 18-month Cycle Normalized Axial Power Distributions for Fresh Fuel in the First Third of the Cycle**

a,c



**Figure 7.2.1-4 Example 4-Loop 17x17 18-month Cycle Normalized Axial Power Distributions for Fresh Fuel in the Middle Third of the Cycle**



a,c



**Figure 7.2.1-5 Example 4-Loop 17x17 18-month Cycle Normalized Axial Power Distributions for Fresh Fuel in the Final Third of the Cycle**

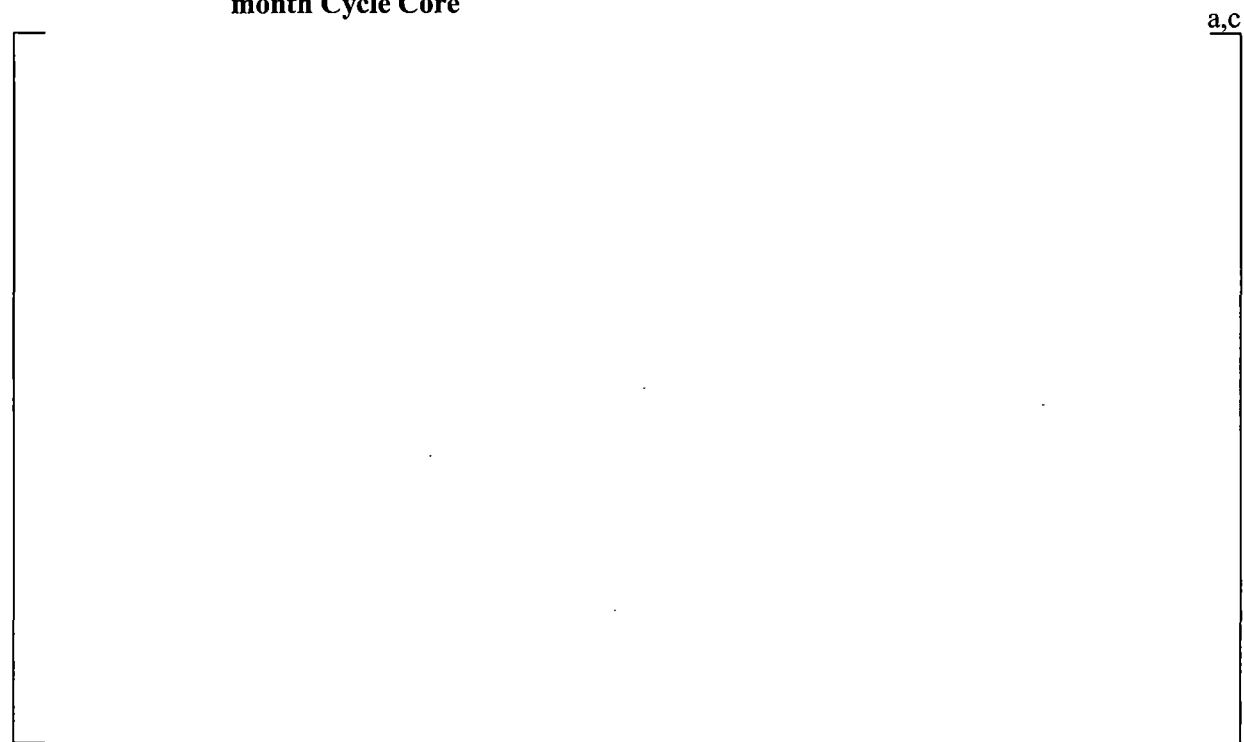
a,c



**Figure 7.2.1-6 Example Fresh Fuel Time Average Normalized Axial Power Shapes over Each Third of an 18-month Cycle of Operation**



**Figure 7.2.1-7** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Fresh Fuel in the First Third of the Cycle for a 4-Loop 17x17 18-month Cycle Core



**Figure 7.2.1-8** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Fresh Fuel in the Middle Third of the Cycle for a 4-Loop 17x17 18-month Cycle Core



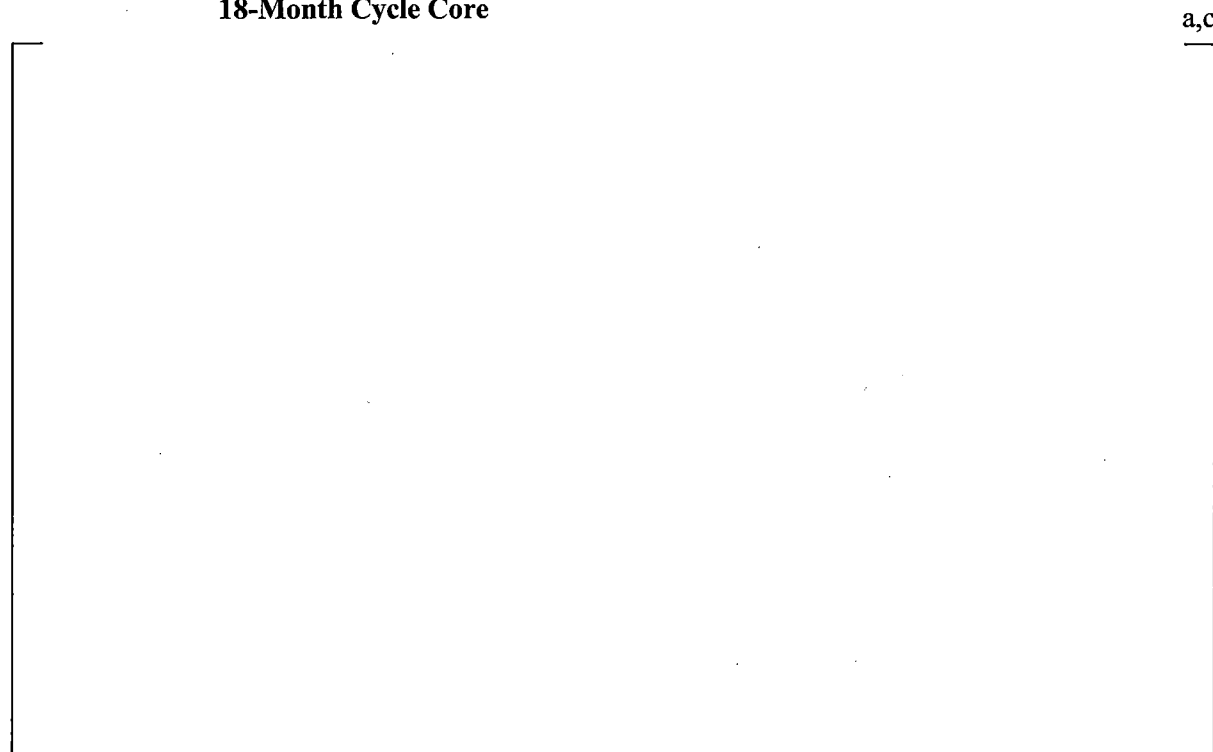
**Figure 7.2.1-9** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Fresh Fuel in the Last Third of the Cycle for a 4-Loop 17x17 18-month Cycle Core



**Figure 7.2.1-10** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Burned Fuel in the First Third of the Cycle for a 4-Loop 17x17 18-month Cycle Core



**Figure 7.2.1-11** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Burned Fuel in the Middle Third of the Cycle for a 4-Loop 17x17 18-Month Cycle Core



**Figure 7.2.1-12** Comparison of [ ]<sup>a,c</sup> Normalized Axial Power Distributions for Burned Fuel in the Last Third of the Cycle for a 4-Loop 17x17 18-month Cycle Core

#### 7.2.1.4.3 Transient Power Duty

The fuel rod design addresses fuel rod operation under both Condition I and Condition II operation. The Condition I and Condition II operational transients considered in the fuel rod design are defined by the nuclear design. The process for generating these transient fuel duty inputs for fuel rod design with PAD5 is based on that described for PAD 3.4 in Reference 7.8, as discussed below for Condition I and Condition II events.

##### A. CONDITION I EVENTS

[

] <sup>a,c</sup>

[

]<sup>a,c</sup>

## B. CONDITION II TRANSIENT DUTY

Condition II events, as summarized in Table 7.1-3, are events which are expected to occur at moderate frequency during the operating life of a nuclear plant. The events of importance to the fuel rod design are those [

]<sup>a,c</sup>

[

] <sup>a,c</sup>

[

] <sup>a,c</sup> Further discussion of how these transient duty limit options are implemented in the fuel rod design process is given in Section 7.4.2 for transient clad strain and in Section 7.4.3 for rod internal pressure analyses.



a,c



**Figure 7.2.1-13 Example Condition I Transient Normalized Axial Power Shapes [**  
**]**<sup>a,c</sup>

a,c



**Figure 7.2.1-14 Condition I Transient Normalized Axial Power Shapes [**  
**]**<sup>a,c</sup>

a,c

**Figure 7.2.1-15 Example Condition I Transient Normalized Axial Power Shapes [**  
**]<sup>a,c</sup>**

a,c

**Figure 7.2.1-16 Illustration of Condition II Transient Power Shape Modeling in PAD5 for**  
**Transient Peaking at the Middle of the Rod**

a,c

**Figure 7.2.1-17 Illustration of Condition II Transient Power Shape Modeling in PAD5 for Transient Peaking at the Bottom of the Rod**

a,c

**Figure 7.2.1-18 Illustration of Condition II Transient Power Shape Modeling in PAD5 for Transient Peaking at the Top of the Rod**

a,c

**Figure 7.2.1-19 Example Transient Rod Average Power Limit as a Function of Rod Average Burnup**

a,c

**Figure 7.2.1-20 Example Transient Rod Average Power Limit as a Function of Steady State Rod Average Power**

a,c

**Figure 7.2.1-21 Example of Binned Transient Local Power Duty**

a,c

**Figure 7.2.1-22 Example Fitted Transient Local Power Duty versus Initial Local Power and Initial Local Burnup**

#### 7.2.1.4.4 Fast Neutron Flux and Fluence Data

Fast ( $E > 1.0$  MeV) neutron flux, in neutrons/cm<sup>2</sup>-sec, and fast fluence, in neutrons/cm<sup>2</sup>, are calculated using an NRC approved core physics code (e.g., Westinghouse Advanced Nodal Code (ANC) Reference 7.7). Fast neutron fluence is the integral of the fast neutron flux over time. [

]<sup>a,c</sup> As burnup increases, the flux spectrum hardens due to fission product burnup and U<sup>235</sup> depletion, and the fluence / burnup ratio increases.

[

]<sup>a,c</sup>

(7-3)

The second PAD5 fast fluence input option is based on [

]<sup>a,c</sup>

The fast fluence inputs for the Westinghouse 17X17 fuel assembly lattice with 0.360 inch fuel rod OD provide a typical example of this fast fluence input. The fast neutron fluences calculated [

]<sup>a,c</sup>



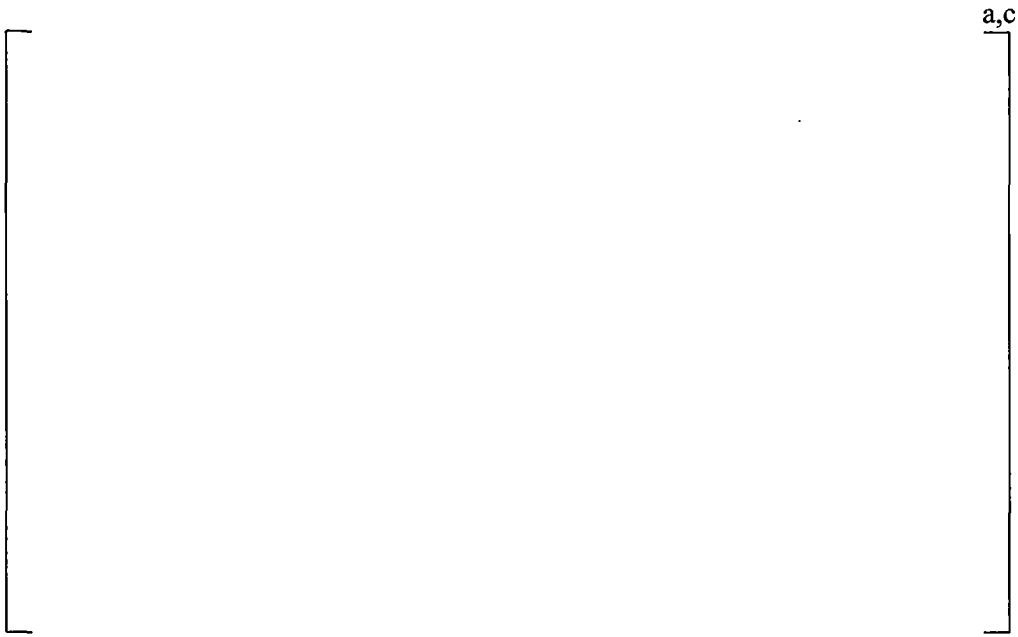


Figure 7.2.1-23 Example Axial Fast Fluence [  
] <sup>a,c</sup>

a,c

**Figure 7.2.1-24 [ ]<sup>a,c</sup> Fast Fluence for 17X17 Fuel Assembly with  
0.360 inch OD Fuel Rods**

#### 7.2.1.4.5 Fuel Pellet Radial Power Distribution Data

Pellet radial power distributions for fuel rod design and licensing analyses are calculated [

]<sup>a,c</sup>



**Figure 7.2.1-25 Comparison of Pellet Radial Power Distributions by Fuel Type for 5 wt%  $U^{235}$  Enrichment Pellets with 2X IFBA  $B^{10}$  Loading at 0 GWd/MTU**



**Figure 7.2.1-26 Comparison of Pellet Radial Power Distributions by Fuel Type for 5 wt%  $U^{235}$  Enrichment Pellets with 2X IFBA  $B^{10}$  Loading at 5 GWd/MTU**

a,c

**Figure 7.2.1-27 Comparison of Pellet Radial Power Distributions by Fuel Type for 5 wt%  $U^{235}$  Enrichment Pellets with 2X IFBA  $B^{10}$  Loading at 10 GWd/MTU**

## **7.2.2 Fuel Rod Design Output Interfaces**

This section provides a summary overview of the flow of fuel rod design analysis results to interfacing disciplines.

### **7.2.2.1 Nuclear Design**

The fuel rod design provides confirmation to nuclear design that the cycle under design satisfies all of the fuel rod design criteria described in Section 7.4. This confirms the acceptability of the core loading plan and planned cycle operation with respect to fuel rod design limits.

The fuel rod design also provides safety related information to nuclear design. For example, the fuel rod design establishes power to fuel melt limits as a function of burnup that are used to confirm that no fuel melting will occur under normal operation and Condition II transient duty conditions. Fuel rod design also provides information to nuclear design to confirm that the transient strain criterion and the rod internal pressure no gap reopening criterion are satisfied with consideration of the limiting transient fuel duty generated by the nuclear design.

### **7.2.2.2 Thermal/Hydraulic Design**

Fuel temperature data generated with PAD5 for safety analyses inputs are provided to both the safety analyses functions and the thermal /hydraulic design function. These temperature data are used as input for initialization of VIPRE (Reference 7.10) transient calculations such as those performed as part of the three-dimension control rod ejection analysis methods in Reference 7.11.

Limiting rod internal pressure data is also provided to the thermal hydraulic function for DNB propagation analysis.

### **7.2.2.3 SAFETY ANALYSIS**

The fuel rod design provides fuel temperature and rod internal pressure inputs to plant safety analyses. These inputs, described below, are typically provided on a conservative basis for use in safety analyses of record (AOR) and are then confirmed to remain applicable on a cycle specific basis to ensure that the safety analyses of record remain valid. Should cycle specific assessments indicate that the fuel rod design inputs used in the AOR are no longer bounding, updated inputs are provided for evaluation and / or reanalysis, as required, in reload specific safety analyses.

#### **7.2.2.3.1 Non-LOCA**

To support non-LOCA safety analyses, the fuel rod design provides the following types of data for [

] <sup>a,c</sup>

[

] <sup>a,c</sup>

#### **7.2.2.3.2 LOCA**

To support LOCA safety analyses, fuel rod design provides the following types of data for [

] <sup>a,c</sup>

#### **7.2.2.3.3 Containment & Radiological Analyses**

The fuel rod design provides the following types of data for [

] <sup>a,c</sup>

[

] <sup>a,c</sup>

#### 7.2.2.4 Product Design & Manufacturing

Fuel rod design analyses for new fuel rod designs provide confirmation to product design that fuel rod design limits can be satisfied for the proposed design. For reload fuel designs, fuel rod design confirms the design fill gas pressure for the feed fuel region to manufacturing.

The PAD5 code also provides [

] <sup>a,c</sup>

#### 7.2.2.5 Fuel Licensing

The fuel rod design provides confirmation to fuel licensing that the fuel rod design satisfies all licensed fuel rod design limits. Typically this is documented in the reload safety evaluation document or its equivalent.

### 7.3 FUEL ROD DESIGN METHODS OVERVIEW

The primary function of a fuel rod is to generate and transfer heat to the reactor coolant. In the process of generating this heat via fissioning, both radioactive and stable fission products are produced in the fuel. A second critical function of the fuel rod is to contain these fission products within the rod so that the reactor coolant does not become contaminated. To meet this goal, the structural integrity of the fuel rod must be maintained (i.e., fuel damage or penetration of fuel rod clad is to be precluded).

The integrity of the fuel rods during Condition I and Condition II modes of operation is ensured by designing the fuel rods so that specific design criteria are satisfied. The design process considers the effects of variations and fluctuations in core and local power and in reactor coolant temperature, pressure and flow which occur during these Condition I and II modes of operation. Examples of Condition I and II operation are given in Tables 7.1-2 and 7.1-3, respectively.

Extremely low probability occurrences which have the potential to cause significant fuel damage are classified as Condition III and Condition IV events. Examples of Condition III are given in Table 7.1-4, and examples of Condition IV events are given in Table 7.1-5. Fuel rod integrity cannot be guaranteed during these hypothetical occurrences. However, the safety analyses and offsite dose calculations for these events confirm that expected fuel failures are conservatively accounted for in radioactive release calculations to confirm that dose limits are satisfied.

The Westinghouse fuel rod design [

] <sup>a,c</sup>



[

] <sup>a,c</sup> All design limits shall be satisfied with appropriate consideration of uncertainties.

Section 7.3.1 provides additional discussion of limiting rods and how they are defined. Section 7.3.2 describes the uncertainties that are considered in the fuel rod design.

### 7.3.1 Definition of Limiting Rods

The fuel rod design for each plant and cycle under design must consider all of the fuel rods included in the operation of that cycle. Each fuel rod is defined by the fuel mechanical design, which establishes the initial fuel rod geometry, fuel enrichment, and, if applicable, burnable absorber type; and by the operating history of the rod, which is established by the reactor coolant conditions and by the nuclear design for each cycle in which a fuel rod has operated. The fuel rod power history, which is defined by both the rod average power and the axial power shape as a function of time, is a key element of the operating history that differentiates the performance of the individual fuel rods.

Limiting rods are defined as those rods whose predicted performance results in the minimum margin to an established design criterion. While all fuel rods are considered in the fuel rod design, not every rod is expected to be a limiting rod. Furthermore, a rod that is limiting for one design criterion may not be limiting for other criteria. For each unique fuel rod design, defined by fuel enrichment and integral burnable absorber type, and pellet and cladding dimensions, the individual rod power histories of the fuel rods of that design distinguish the performance of each of the rods with respect to each design criterion. The one fuel rod which exhibits the minimum margin to a design criterion is the limiting rod so that the limiting rod can be identified. If there is acceptable design margin for this limiting rod, all of the other rods in the core of that design will have even greater margin to the design criterion.

For some design criteria, the operating characteristics that produce the most limiting rod behavior with respect to that criterion are well known. For example, in the analysis for the limiting rod internal pressure, it is known that high pressure rods are typically characterized by high power and burnup operation, particularly those rods with maximum power at near end of life, since this type operation results in maximum fission gas release. This information is used to identify potential candidates for the limiting rod for the rod internal pressure related criteria. For the transient strain design criteria, the limiting rod is characterized as the rod with the largest transient local power increase following pellet clad contact. Design experience of this type is utilized to focus on those rods which are most likely limiting rods.

### 7.3.2 Treatment of Uncertainties

For each limiting rod, the conservative case analysis for each design criterion is defined as that analysis which considers the impacts of all significant uncertainties in determining the results to be compared to the design limit. Uncertainty exists in fuel rod fabrication characteristics, such as fuel pellet and cladding dimensions; in operating parameters, such as system pressure, power history, or inlet temperature; and in PAD5 fuel and cladding performance models. All of these uncertainties are considered as potentially significant uncertainty candidates.

[

] <sup>a,c</sup>

Table 7.3.2-1 provides a listing of the uncertainty parameters that are considered in the fuel rod design process. [

] <sup>a,c</sup>

[

] <sup>a,c</sup>

[ ]<sup>a,c</sup>  
As noted, the limiting rod for each criterion is that rod power history which has the minimum margin to the design limit.

The PAD5 model uncertainties considered in PAD5 uncertainty assessments for fuel rod design and safety analysis input calculations are summarized in Table 7.3.2-1. Upper and lower bound model parameters are shown for each model. For fuel and cladding performance models that are unchanged from PAD 4.0, documented in References 7.8 and 7.32, model uncertainties remain the same.

Fabrication parameter uncertainties considered in PAD5 uncertainty assessments are summarized in Table 7.3.2-2. [

] <sup>a,c</sup>

Plant operating parameter inputs considered in PAD5 uncertainty assessments are summarized in Table 7.3.2-3. [

] <sup>a,c</sup>

[

] <sup>a,c</sup>

**Table 7.3.2-1 List of PAD5 Model Uncertainties Considered in Significant Uncertainty Assessments**

a,c

#	Parameter Description	Value Basis <sup>(1)</sup>	Upper Bound Uncertainty	Lower Bound Uncertainty
---	-----------------------	----------------------------	-------------------------	-------------------------

**Table 7.3.2-1 List of PAD5 Model Uncertainties Considered in Significant Uncertainty Assessments.**  
**Continued**

#	Parameter Description	Value Basis <sup>(1)</sup>	Upper Bound Uncertainty	Lower Bound Uncertainty	a,c
---	-----------------------	----------------------------	-------------------------	-------------------------	-----

**Table 7.3.2-2 List of Fabrication Parameter Uncertainties Considered in Significant Uncertainty Assessments**

#	Parameter Description	Value Basis	a,c

<sup>1</sup> Rods with annular blanket pellets only.

<sup>2</sup> ZrB<sub>2</sub> IFBA rods only.

**Table 7.3.2-3 List of Plant Operation Uncertainty Parameters Considered in Significant Uncertainty Assessments**

a,c

**Table 7.3.2-4 Summary of Power Related Uncertainties (ANC)**

#	Parameter	Uncertainty Value (95/95% upper limit)	Comment
---	-----------	---	---------

a,c

<sup>3</sup> Plant specific input uncertainties are provided as input and are defined on a 95/95% basis.

<sup>4</sup> Rods modeled with generic fast fluence fitting coefficients.



### 7.3.3 Steady State Power Histories

Steady state power history inputs used in fuel rod design are described in Section 7.2.1.4.1. The details of how power histories are evaluated are included in Section 7.4 as part of the evaluation methodology discussion for each design criterion and in Section 7.5 with respect to safety analysis input generation.

### 7.3.4 Transient Power Duty

The fuel rod design considers Condition I and II operation, including Condition I and II transients. A description of the Condition I and Condition II transient power duty input generated by the nuclear design for input to the fuel rod design is given in Section 7.2.1.4.3. As noted, [

]<sup>a,c</sup>

Additional details on the application of the Condition I and II transient power duty in fuel rod design are presented on a criterion specific basis in Section 7.4 and with respect to safety analysis input generation in Section 7.5.

## 7.4 FUEL ROD DESIGN CRITERIA & EVALUATION METHODS

For each fuel rod design criterion, the Design Category, Design Parameter, Design Basis, Acceptance Limit, and Design Evaluation Methodology are summarized. The definitions of these items are:

Design Category - The NRC groups acceptance criteria for design limits into three categories in the SRP (Reference 7.1): Fuel System Damage, Fuel Rod Failure, and Fuel Coolability.

Design Parameter - The specific parameter that will be evaluated.

Design Basis - The reason that the design parameter needs to be considered in the safety evaluation

Acceptance Limit - The acceptance limit is the value that must be demonstrated to be satisfied to satisfy the requirements of SRP Section 4.2 to provide an acceptable margin to fuel failure.

Design Evaluation - A description of the evaluation methods used to evaluate the design or acceptance limits.

### 7.4.1 Clad Stress

Design Category - Fuel System Damage

Design Parameter - Clad Stress

Design Basis - The fuel system will not be damaged due to excessive fuel clad stress.

Acceptance Limit – Consistent with NRC SRP Section 4.2 (Reference 7.1), maximum cladding stress intensities excluding PCI induced stress shall be evaluated based on ASME Boiler and Pressure Vessel Code (BPVC) guidelines (Reference 7.13). Stresses in the cladding are combined to calculate a maximum stress intensity which is then compared to the criteria in Table 7.4.1-1.

These clad stress criteria under Condition I and Condition II loading have been updated relative to those approved by NRC in Reference 7.12 to support the use of improved cladding materials such as **Optimized ZIRLO** cladding. Modified stress intensity limits [

]<sup>a,c</sup> The ASME BPVC recommended limits for Condition III and IV events are retained without modification.

For cladding with the debris mitigating oxide coating on the bottom 6 to 7 inches of the fuel rod, the same clad stress criteria are applied as shown in Table 7.4.1-1 with [

]<sup>a,c</sup>

Design Evaluation – Based on the ASME BPVC, the Primary Membrane Stress (Pm) is defined as the component of normal stress, uniformly distributed and equal to the average value of stress across the thickness of the section under consideration, developed by the imposed loading which is necessary to satisfy the simple laws of equilibrium of external and internal forces and moments. Primary Bending Stress (Pb) is defined as the component of normal stress varying from the average value of stress across the thickness of the section under consideration, developed by the imposed loading which is necessary to satisfy the simple laws of equilibrium of external and internal forces and moments. Secondary Stress (Q) is defined as a normal or a shear stress developed by the constraint of adjacent material or by self-constraint of the structure. The basic characteristic of a secondary stress is that it is self-limiting. Local yielding and minor distortions can satisfy the conditions which cause the stress to occur, and failure from one application of the stress is not to be expected.

Design loadings considered in the clad stress analysis and the associated stress category are shown in Table 7.4.1-2. [

] <sup>a,c</sup>

The average differential pressure induced stress in the cladding is accounted for as a primary membrane stress. The variation in the clad stresses due to differential pressure loads relative to the average differential pressure stress is classified as secondary stress. The PAD5 fuel performance code is used to calculate rod internal pressure (and clad average temperature) input to the clad stress analyses. PAD5 rod internal pressure analyses are performed at transient conditions for the limiting pressure fuel rod in the core, with consideration of model, fabrication and operational uncertainties, to define the limiting internal pressures. Analyses are performed at BOL conditions to obtain the rod internal pressures used in calculating the limiting compressive differential pressure stresses. The maximum calculated upper bound rod internal pressure over the life of the fuel rods is used to obtain the rod internal pressures used in calculating the limiting tensile differential pressure stresses. [

] <sup>a,c</sup> The rod internal pressure uncertainty analysis methods used in these calculations are the same as are described in Section 7.4.3.

Cladding ovality is considered in the clad stress analysis [

] <sup>a,c</sup>

[

] <sup>a,c</sup>

Dynamic axial and lateral loads associated with seismic or combined seismic/LOCA events are classified as primary membrane and primary bending stresses, respectively, in the cladding stress analysis. These stresses are calculated using NRC approved methods, such as those described in References 7.26 and 7.27 for Westinghouse and Combustion Engineering NSSS cores, respectively, and are applied in the calculation of the fuel rod cladding stresses and stress intensities.

Table 7.4.1-1 Westinghouse Clad Stress Criteria

Parameter	Clad Stress Criteria	
	Compressive Loading	Tensile Loading

a,c

Where:

- $\sigma_y$  is the lower bound 0.2% offset cladding yield strength with consideration of temperature and irradiation effects;
- $\sigma_u$  is the lower bound cladding ultimate tensile strength with consideration of temperature and irradiation effects;
- $P_m$  is the primary membrane stress;
- $P_b$  is the primary bending stress; and
- $Q$  is the secondary stress.

<sup>5</sup> For the oxide coated cladding region, [

]a,c

**Table 7.4.1-2 Summary of Design Loadings and Stress Classifications**

Design Loading	Stress Classification

a,c

## 7.4.2 Clad Strain

Design Category - Fuel Rod Failure / Fuel System Damage

Design Parameter - Clad Strain

Design Basis - The fuel rod will not fail due to excessive fuel clad strain.

Acceptance Limit - The design limit for the fuel rod clad strain is that the total tensile strain, elastic plus plastic, due to uniform cylindrical fuel pellet deformation during any single Condition I or II transient shall be less than 1% from the pre-transient value.

Design Evaluation – Design analyses are performed with the Westinghouse PAD5 code to confirm that the 1% transient clad strain criterion is satisfied for the limiting rod in the core. [

]a,c



[

strain is calculated based on the change in the cladding diameter at transient power conditions relative to the cladding diameter at the pre-transient steady state condition. ]<sup>a,c</sup> The

All significant uncertainties are accounted for in the transient clad strain analysis [

respect to the established design limit of 1%. ]<sup>a,c</sup> The design margin is evaluated with

[

] <sup>a,c</sup>

Table 7.4.2-1 Example of [

]<sup>a,c</sup> Two Cycle Rod Power Histories for Transient Clad Strain

Table 7.4.2-2 Significant Uncertainty Parameters Included in the Transient Clad Strain Total Uncertainty Calculation			
#	Uncertainty Parameter	Limiting Direction	Value

a,c

<sup>1</sup> Fuel rods with ZrB<sub>2</sub> IFBA fuel pellets only

<sup>2</sup> Fuel rods with Gd<sub>2</sub>O<sub>3</sub>-UO<sub>2</sub> pellets only

a,c

**Figure 7.4.2-1 Comparison of Transient Clad Strain Uncertainties for a 17X17 RFA ZrB<sub>2</sub> IFBA Fuel Rod**

a,c

**Figure 7.4.2-2 Comparison of Transient Clad Strain Uncertainties for a 17X17 RFA non-IFBA Fuel Rod**

a,c

**Figure 7.4.2-3 Example Transient Clad Strain [**

**]<sup>a,c</sup>**

### 7.4.3 Rod Internal Pressure

Design Category - Fuel System Damage

Design Parameter - Rod Internal Pressure

Design Basis - The fuel system will not be damaged due to excessive fuel rod internal pressure.

Acceptance Limit - The internal pressure of the lead fuel rod in the reactor will:

- be limited to a value below that which could cause the diametral gap to increase (cladding liftoff) due to outward cladding creep during normal (Condition I) operation (Reference 7.14);
- be limited to a value below that which could result in cladding hydride reorientation in the radial direction; and
- be limited to preclude extensive DNB propagation (References 7.15, 7.16 and 7.17).

Design Evaluation – The PAD5 code is used to calculate the rod internal pressure (RIP) in the fuel rod, with consideration of appropriate uncertainties. The rod internal pressure design limits and the associated evaluation methods are discussed in the following sections.

#### 7.4.3.1 Rod Internal Pressure No Clad Liftoff

The no clad liftoff criterion assures that the pellet-clad gap does not reopen due to the cladding creep rate exceeding the fuel swelling rate. The concern is that gap reopening will initiate a positive feedback loop, where the increased local temperature associated with a reopened gap will result in additional thermal fission gas release, thus increasing the rod internal pressure and further increasing the creep rate such that the gap increases in size and the process accelerates.

Design Evaluation – PAD5 analyses are performed to confirm that the rod internal pressure for the most limiting rod in the core remains below that pressure which would result in cladding liftoff. PAD5 is used both to calculate the internal pressure and also to calculate the clad liftoff limit. Since the liftoff limit is also a function of the power history and internal pressure, the clad liftoff limit is calculated in the same PAD5 analysis with the rod internal pressure

The rod internal pressure increases during fuel rod operation due to changes in the internal void volume, due to cladding creep and fuel swelling, and due to increased gas inventory driven primarily by fission gas release. For fuel rods with  $\text{ZrB}_2$  IFBA coated pellets, there is also increased helium in the rods as the  $\text{B}^{10}$  is depleted to produce helium which is then released. Both void volume and gas inventory changes are modeled in PAD5.

The clad liftoff limit, calculated by PAD5 on a case specific basis, is also a function of power history, since the cladding creep rate is a function of fast flux, clad temperature and differential pressure stresses and the fuel swelling rate is a function of the burnup rate.

The rod internal pressure analysis has a well-defined sensitivity to fuel rod power history. [

] <sup>a,c</sup>

All rod power histories in the core can be analyzed for rod internal pressure to identify the most limiting rod. However, [

] <sup>a,c</sup> Separate PAD5 rod internal pressure analyses are performed for UO<sub>2</sub> fuel rods and for rods with integral burnable absorbers (e.g., ZrB<sub>2</sub> IFBA, Gd<sub>2</sub>O<sub>3</sub>-UO<sub>2</sub>, and Er<sub>2</sub>O<sub>3</sub>-UO<sub>2</sub>).

As noted, the PAD5 code calculates both the rod internal pressure and the pressure level at which the cladding outward creep rate is predicted to exceed the pellet solid swelling rate. At each time step where the internal pressure exceeds the system pressure, [

] <sup>a,c</sup>



All significant uncertainties that affect the calculated internal pressure and the liftoff limit are accounted for in these rod internal pressure analyses. [

] <sup>a,c</sup>

Power history uncertainties are explicitly accounted for in the rod internal pressure analysis. Figure 7.4.3-2 illustrates an example upper bound power history uncertainty as applied to a power history used in rod internal pressure analyses.

The impact of Condition I and Condition II operation events are accounted for in the rod internal pressure analysis. For the design cycle, [

] <sup>a,c</sup>

Similar to the transient clad strain analysis, [

] <sup>a,c</sup> including the specified transient power duty.

[

] <sup>a,c</sup>

[

] <sup>a,c</sup>

**Table 7.4.3-1 Significant Uncertainty Parameters Included in the Rod Internal Pressure No Clad Liftoff Total Uncertainty Calculation**

#	Uncertainty Parameter	Limiting Direction	Value
---	-----------------------	--------------------	-------

a,c

<sup>1</sup> [

] <sup>a,c</sup>

<sup>2</sup> Rods with annular axial blankets only

<sup>3</sup> ZrB2 IFBA rods only

a,c

**Figure 7.4.3-1 Comparison of Rod Internal Pressure No Clad Liftoff Uncertainties for a 17X17 OFA Fuel Rod with Optimized ZIRLO Cladding**

a,c

**Figure 7.4.3-2 Illustration of Upper Bound Power History Application for Rod Internal Pressure Analyses**

a,c

Figure 7.4.3-3 Example Rod Internal Pressure [ $J^{a,c}$ ]



applicable to **Optimized ZIRLO**. However, the application of the mechanistic DNB propagation methods may be used with any cladding material given that the appropriate licensed high temperature creep model is used. Details of the high temperature creep model implementation are provided in Reference 7.15.

Limiting Condition III and Condition IV DNB events, such as control rod ejection, single rod withdrawal at power, and locked rotor, are modeled with a sub-channel thermal-hydraulic code such as VIPRE-01 (Reference 7.10) at design limiting conditions to establish the required input to model the DNB propagation. The inputs included: [

]<sup>a,c</sup>

The DNBR limit is used to identify the time in the transient for the onset of DNB. The sub-set of the [

]<sup>a,c</sup> to

preclude DNB propagation to neighboring fuel rods. In addition, fuel rods must be demonstrated to not burst prior to reaching that critical strain limit during the DNB transient. If fuel rods are predicted to exceed the strain limit of [

]<sup>a,c</sup>

In general, the most limiting transients would be ones which occur for [

]<sup>a,c</sup>

As noted in Reference 7.15, the analysis of DNB propagation is extremely conservative. The primary conservatisms are listed below:

1. The DNB rod ballooning is conservatively assumed to occur symmetrically. In reality, rod-to-rod gap closure from a ballooning fuel rod experiencing DNB clearly degrades the surface heat transfer of an adjacent rod only at a local area on the circumference. Thus, occurrence of DNB on an adjacent rod will be highly circumferentially oriented and high temperature deformation would likely occur only on the surface of the adjacent fuel rod facing the original fuel rod experiencing DNB. Consequently, DNB propagation and fuel rod failure are construed to involve at most only one additional row of adjacent fuel rods .
2. [

]<sup>a,c</sup>

3. [ ]<sup>a,c</sup>
4. DNB transients are modeled conservatively assuming the hot rod in DNB when the plant DNB Ratio (DNBR) limit is violated. The DNBR limit is determined based on test data so that there is at least a 95 percent probability at a 95 percent confidence level that DNB will not occur on the hot rod.
5. During the DNB transient, it is conservatively assumed that film boiling heat transfer occurs if the DNBR is below the DNBR limit.
6. The strain criterion of [ ]<sup>a,c</sup> is less than allowed by the [ ]<sup>a,c</sup> gap reduction strain (References 7.15 and 7.16).

Therefore, while DNB propagation is conservatively assumed in the analysis, the physical mechanisms involved do not actually support the occurrence of DNB propagation.

#### 7.4.3.3 Rod Internal Pressure Clad Hydride Reorientation

Hydride reorientation occurs when hydride precipitates formed during reactor operation reorient from the circumferential to the radial direction. It is a precipitation driven process that occurs when fuel cladding is cooled under a tensile stress from a temperature where hydrides are dissolved. The formation of radial hydrides can reduce the cladding ductility and increase the potential for brittle failure due to subsequent fuel rod handling.

Investigations into this phenomenon have been conducted by three organizations: CEA, Westinghouse and ANL. The CEA tests were part of an industry effort that tested Zircaloy-4, Zircaloy-2 and ZIRLO in both un-irradiated and irradiated conditions. These investigations found that the critical stress for initiation of hydride reorientation of ZIRLO is about [ ]<sup>a,c</sup> in both the unirradiated and irradiated condition. In these hydride reorientation (HRO) tests, the cladding segment is heated, then placed under a constant tensile stress, held for a time and then cooled. A typical test is presented in Figure 7.4.3-4 where both the sample temperature and internal pressure are plotted as a function of time. In the CEA tests following the HRO treatment, some of the cladding samples were fabricated into test specimens and subjected to a ring tensile test (RTT) at various temperatures. Other samples were inspected to determine the fraction of radial hydrides and hydride distribution. The tensile tests results were converted to Displacement Energy Density (DED)

$$DED = \int_0^{\Delta X_R} R(\Delta X) * d(\Delta X)$$

Where:

R: Engineering Stress

d: Displacement – Diameter Increase.

A set of DED results at 420°C are plotted in Figure 7.4.3-5. Here it can be observed that [ ]<sup>a,c</sup> is a conservative estimate of the impact of increasing radial hydride fraction.

The use of [ ]<sup>a,c</sup> tensile stress in the cladding is a very conservative limit for hydride reorientation and associated ductility loss, since ductility loss does not actually occur at the onset of hydride reorientation. In experiments simulating dry cask storage conditions (Reference 7.287.28), ZIRLO cladding was placed in tension at various stress levels while cooling from a high temperature of 400°C to room temperature. The cladding was then tested using a ring compression test (RCT), and tensile stresses of greater than 100 MPa (14.5 ksi) were needed to observe a brittle RCT result. The typical test sequenced used at ANL is plotted in Figure 7.4.3-6 as pressure and temperature vs time. The results of the tests performed on Zircaloy-4 and ZIRLO cladding are presented in Figure 7.4.3-7 as hoop stress vs cladding hydrogen. The results of the test are shown as open symbols (ductile) or filled (brittle).

Westinghouse testing has compared the hydride reorientation behavior of ZIRLO and **Optimized ZIRLO** cladding at [ ]<sup>a,c</sup> which is slightly above the threshold where HRO is first observed [ ]<sup>a,c</sup> for ZIRLO cladding. It was observed that the percentage of hydrides that reoriented in **Optimized ZIRLO** was [ ]<sup>a,c</sup>

The results are listed in Table 7.4.3-2. The same conservative hydride reorientation limits are applied to both ZIRLO and **Optimized ZIRLO** cladding.

These hydride reorientation threshold data are converted to maximum internal pressure in the clad using the thick wall stress equation for the cladding mid-wall hoop stress and solving for the internal pressure.

$$\sigma_{\theta} = \left[ \frac{a_0^2 P_i - b_o^2 P_o}{(b_o^2 - a_0^2)} \right] - \left[ \frac{a_0^2 b_o^2 (P_o - P_i)}{r_m^2 (b_o^2 - a_0^2)} \right] \quad (7-5)$$

Where

- $\sigma_{\theta}$  = mid-wall hoop stress, psi
- $b_o$  = outer radius, inches
- $a_o$  = inner radius, inches
- $r_m$  = mid-wall radius, inches
- $P_o$  = external pressure, psi, and
- $P_i$  = internal pressure, psi.

Solving for the internal pressure,  $P_i$ , this equation becomes



$$P_i = \left[ \frac{(b_o^2 - a_o^2)\sigma_\theta + P_o(b_o^2 + a_o^2 b_o^2 / r_m^2)}{(a_o^2 + a_o^2 b_o^2 / r_m^2)} \right] \quad (7-6)$$

Assuming a hoop stress threshold of [ ]<sup>a,c</sup> for onset of ductility loss and a typical system pressure of 15.5 MPa (2250 psi), the rod internal pressure limits to preclude ductility loss due to hydride reorientation are calculated as shown in Table 7.4.3-3 for Westinghouse fuel rod designs. The rod internal pressure values for the onset of hydride reorientation [ ]<sup>a,c</sup> are also shown in this table. Rod internal pressures must exceed [ ]<sup>a,c</sup> before hydride reorientation would cause a reduction in the cladding ductility. The initiation of hydride reorientation is precluded at rod internal pressures up to and exceeding [ ]<sup>a,c</sup>. These thresholds exceed the maximum rod internal pressure allowed by the clad no liftoff criterion. Rod internal pressure analyses performed for the no liftoff criterion confirm that the threshold pressures for hydride reorientation are not exceeded.

Hydride Reorientation H/T Run#1  
Samples: 2102A, 0589A, 0339C, 9095A & 9097B



**Figure 7.4.3-4 Typical Westinghouse HRO Test as Function of Time**



**Figure 7.4.3-5 RCT DED Following a HRO Test as a Function of Stress**

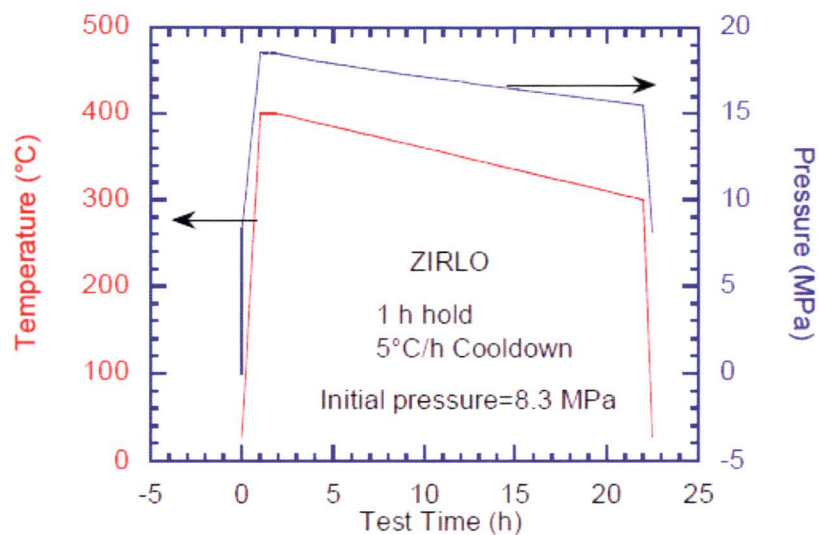


Figure 7.4.3-6 From Reference 7.28, Figure B-1, HRO Test, Temperature and Pressure vs Time

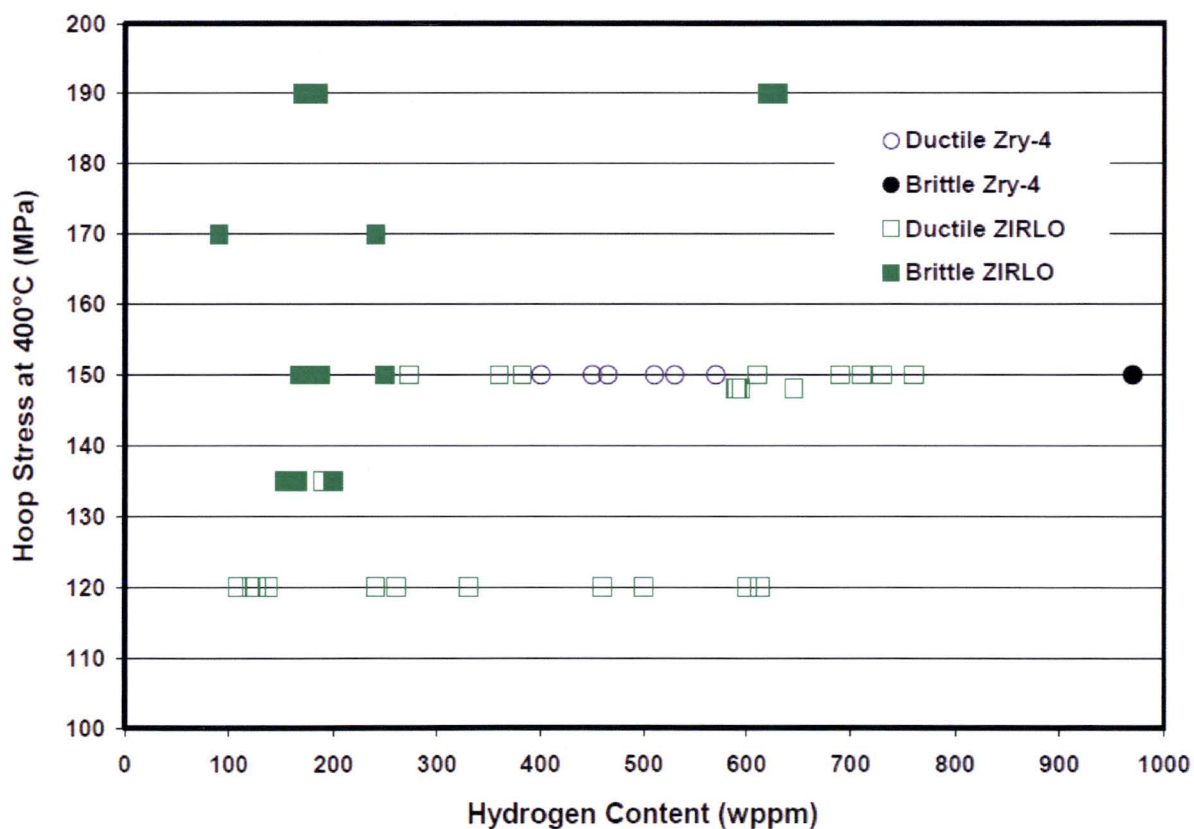


Figure 7.4.3-7 Reference 7.28, Figure 5, RCT Ductility as a Function of Reorientation Stress

**Table 7.4.3-2 Westinghouse HRO Testing Results on ZIRLO and Optimized ZIRLO Cladding**

Cladding	Wt% tin	Heat Treatment	Condition	Minimum % Radial Hydrides	Maximum % Radial Hydrides	Average % Radial Hydrides
----------	------------	-------------------	-----------	---------------------------------	---------------------------------	---------------------------------

a,c

**Table 7.4.3-3 Sample Rod Internal Pressure Limits for No Clad Hydride Reorientation**

Rod Design	15X15, 14X14	16X16, CE16NGF, 17X17 RFA	17X17 OFA	CE 16X16
------------	--------------	---------------------------------	-----------	----------

a,c

#### 7.4.4 Clad Fatigue

Design Category - Fuel System Damage

Design Parameter - Clad Fatigue

Design Basis - The fuel system will not be damaged due to fatigue.

Acceptance Limit - The fatigue life usage factor is limited to less than 1.0 to prevent reaching the material fatigue limit, considering a safety factor of 2 on stress amplitude or a safety factor of 20 on the number of cycles, whichever is more limiting (Reference 7.18).

Design Evaluation - The Langer O'Donnell fatigue model (Reference 7.19) is used in the PAD5 fatigue analysis, with the empirical factors of their fatigue model correlation modified to conservatively bound the results of Westinghouse testing programs for Zircaloy-4, ZIRLO and **Optimized ZIRLO** cladding material. Fatigue model constants developed for Zircaloy-4 cladding (Reference 7.20) were later validated for ZIRLO (Reference 7.21) and **Optimized ZIRLO** cladding (Reference 7.22). The design equations follow the concept for the fatigue design criterion described in the ASME Boiler and Pressure Vessel Code, Section III. The factor of 2 on stress amplitude and the factor of 20 on the number of cycles are conservative adjustment factors to account for known effects that are not included in standard fatigue tests (Reference 7.29) and are not considered as uncertainties but as conservative adjustments of the fatigue limit. The fatigue design curve used in Westinghouse design analyses is given in Figure 7.4.4-1.

Fatigue analyses are primarily concerned with the accumulated effects of short-term, cyclic, cladding stress and strain which result from daily load follow operation and other anticipated core maneuvers. While most Westinghouse PWR plants have historically been operated in base load mode, these plants and the fuel have been designed to accommodate daily load follow operation. Load follow, when implemented, can take the form of multiple different load cycles. They are characterized by 24 hour periods with an assumed time at full power and an assumed time at reduced power. The degree of power reduction during the load cycle would typically be determined by the utility based on their load requirements. The fuel rod design conservatively assumes a bounding load swing of [

]<sup>a,c</sup>

The PAD5 analyses for clad fatigue also include consideration of local cladding defects typically up to [

]<sup>a,c</sup> This local defect is accounted for as a local stress concentration.

The limiting power history for clad fatigue is typically characterized by [

]<sup>a,c</sup>

These observed trends are helpful in assessing which rod power histories will likely result in limiting clad fatigue usage, but they are not conclusive. Therefore, [

]<sup>a,c</sup>

[

<sup>a,c</sup> Based on the maximum change in the principal stresses, the number of cycles to failure,  $N_f$ , is calculated using the Langer-O'Donnell fatigue model given in Figure 7.4.4-1, with consideration of the factors of safety on stress and number of cycles and accounting for a local clad imperfection, or wear scar, as shown in Equation (7-7) below:

$$N_f = \left[ \frac{E}{(S_a \times FSCF - S_e)} \frac{\ln[100/(100 - RA)]}{4} \right]^2 \quad (7-7)$$

where:  $S_a$  = Stress Amplitude calculated by PAD5 for the load cycle (psi);

$E$  = Young's Modulus (psi) =  $f(T)$ ;

$T$  = Clad Average Temperature in °F;

$RA$  = Fatigue Material Constant [ <sup>a,c</sup> related to percent reduction of area in tensile testing;

$S_e$  = Endurance Limit =  $SEEE \times E$ ;

$SEEE$  = Fatigue material constant [ <sup>a,c</sup>; and

$FSCF$  = Fatigue stress concentration factor.

The temperature dependent Young's Modulus for ZIRLO and **Optimized ZIRLO** cladding is given in Section 6.3.5.

The fractional fatigue usage associated with each interval is the ratio of the number of cycles in the interval divided by  $N_f$ , the number of cycles to failure. The cumulative fatigue usage is the sum of these fractional usages over the life of the fuel rod, or

$$\text{Cumulative Usage} = \sum_1^k \frac{n_k}{N_{fk}} < 1.0 \quad (7-8)$$

Where:  $n_k$  = number of cycles of mode  $k$ ; and  
 $N_{fk}$  = number of allowable cycles.

In addition to daily load follow, the fatigue analysis [

] <sup>a,c</sup> in the cumulative fatigue usage calculation.

Model, fabrication, and operating uncertainties are accounted for in PAD5 fatigue analyses. Significant uncertainties are selected based on sensitivity studies performed as described in Section 7.3.2. Table 7.4.4-1 lists the uncertainties determined to be significant in the calculation of the clad fatigue [

] <sup>a,c</sup>

**Table 7.4.4-1 Significant Uncertainty Parameters Included in the Clad Fatigue Total Uncertainty Calculation**

#	Uncertainty Parameter	Limiting Direction	Value	a,c

<sup>1</sup> ZrB<sub>2</sub> IFBA fuel rods only

<sup>2</sup> Limiting direction for power related uncertainties could be upper or lower bound.



$$\epsilon_t = 2 * S_a * 100 / E (\%)$$

$$S_a = \frac{E}{4 \sqrt{N_f}} \ln [ 100 / (100 - RA) ] + S_e$$

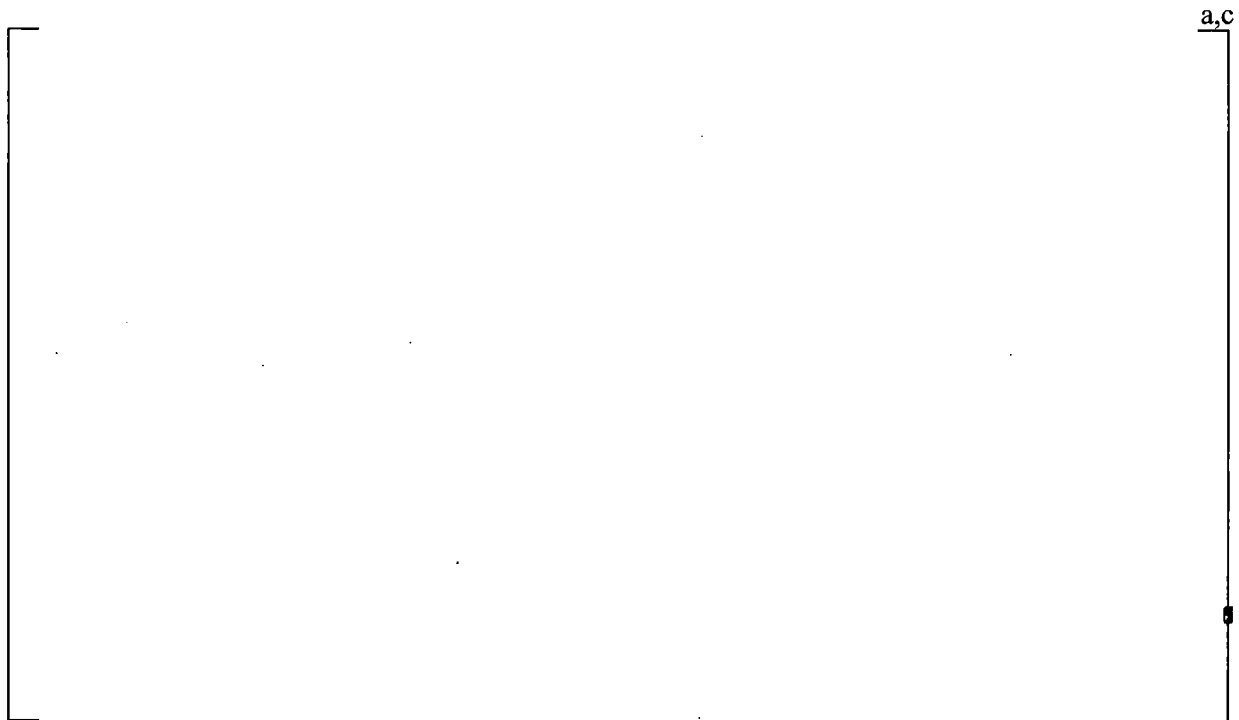
$\epsilon_t$  = Strain Range (%)

$S_a$  = Pseudo Stress Amplitude (psi)

$E$  = Young's Modulus (psi)

$RA$  = Reduction in Area (%)

$S_e$  = Endurance Limit (psi)



**Figure 7.4.4-1 Fatigue Design Curve Based on the Langer-O'Donnell Model**

**Figure 7.4.4-2 Comparison of Upper Bound End of Life Fatigue Usage for All Uncertainties versus Selected Significant Uncertainties**

### 7.4.5 Clad Oxidation

Design Category - Fuel System Damage

Design Parameter - Clad Oxidation

Design Basis - The fuel system will not be damaged due to excessive fuel clad oxidation.

Acceptance Limit - The predicted [ ]<sup>a,c</sup> shall be no greater than 100 microns, (Reference 7.23). The maximum Thermal Reaction Accumulated Duty (TRD) for ZIRLO and **Optimized ZIRLO** cladding is restricted to numbers corresponding to a cladding corrosion thickness of 100 microns for licensing applications.

Design Evaluation – Clad corrosion is calculated in the PAD5 code as a function of axial elevation, using the channel coolant models and corrosion models as described in Sections 3.1, 3.2 and 3.3. This calculation involves only a small subset of the fuel and cladding performance models incorporated in the PAD5 code, since the corrosion calculation is focused on the outer surface of the cladding and clad oxidation is not dependent on any of the other fuel pellet and cladding performance models.

For clad corrosion and hydrogen pickup, the limiting rod power history is [

] <sup>a,c</sup>

In fuel rod design, [

] <sup>a,c</sup> The limiting rod power history for corrosion is that which results in the maximum predicted clad oxide at the most limiting axial node.

Analyses performed to address the corrosion design limit are done [

] <sup>a,c</sup> to support LOCA safety analyses in evaluating the impact of pre-transient oxide. For these analyses, the upper bound corrosion models given in Section 3 by cladding material are used.

[

] <sup>a,c</sup> A similar figure can be generated for upper bound clad oxide as a function of rod burnup, as shown in the example in Figure 7.4.5-2.

a,c

**Figure 7.4.5-1 Example Best Estimate Clad Corrosion as a Function of Rod Average Burnup**  
Maximum Values [ ]<sup>a,c</sup>

a,c

**Figure 7.4.5-2 Example Upper Bound Clad Oxide as a Function of Rod Average Burnup**  
Maximum Values [ ]<sup>a,c</sup>

## 7.4.6 Clad Hydrogen Pickup

Design Category - Fuel System Damage / Fuel Rod Failure

Design Parameter - Clad Hydrogen Pickup

Design Basis - The fuel system will be operated to prevent significant degradation of mechanical properties of the clad at low temperatures, as a result of hydrogen embrittlement caused by the formation of zirconium hydride platelets.

Acceptance Limit - The best estimate, [ ]<sup>a,c</sup> hydrogen pickup level in the most limiting clad axial node will be less than or equal to [ ]<sup>a,c</sup> (Reference 7.23) at the end of fuel operation.

Design Evaluation – The best estimate cladding hydrogen pickup is calculated directly from the best estimate corrosion calculation results discussed in Section 7.4.5, using the hydrogen pickup fraction of [ ]<sup>a,c</sup> licensed in Reference 7.23. The best estimate through-wall average hydrogen content in the cladding is calculated by

$$\left[ \frac{\text{Best Estimate Hydrogen Pickup Fraction} \times \text{Through-Wall Average Hydrogen Content}}{\text{Through-Wall Average Hydrogen Content}} \right]^{a,c} \quad (7-9)$$

Using the maximum calculated best estimate clad oxide thickness result from the corrosion analysis, the best estimate [ ]<sup>a,c</sup> hydrogen content is calculated using Equation (7-9) and the result compared to the [ ]<sup>a,c</sup> design limit.

If the best estimate clad oxidation is known as a function of rod burnup, as illustrated in Figure 7.4.5-1, the best estimate [ ]<sup>a,c</sup> hydrogen can also be derived as illustrated in Figure 7.4.6-1.

a,c

**Figure 7.4.6-1 Example Best Estimate Clad Hydrogen as a Function of Rod Average Burnup  
Maximum Values [ ]<sup>a,c</sup>**

### 7.4.7 Fuel Rod Axial Growth

Design Category - Fuel System Damage

Design Parameter - Fuel Rod Axial Growth

Design Basis - The fuel system will not be damaged due to excessive axial interference between the fuel rods and the fuel assembly structure.

Acceptance Limit - The fuel rods shall be designed with adequate clearance between the fuel rod and the top and bottom nozzles to accommodate the differences in the growth of fuel rods and the growth of the assembly without interference.

Design Evaluation – The PAD5 upper bound fuel rod axial growth models, given in Equation 5-43 for ZIRLO cladding and Equation 5-46 for **Optimized ZIRLO** cladding, are used in the calculation of the fuel rod shoulder gap as a function of fast neutron fluence. The fuel rod growth analysis uses the following equation:

a,c

[

]a,c The

resulting minimum shoulder gap must be greater than zero.

The combination of [

]a,c provides high confidence that no fuel rod interference with the top and bottom nozzles can occur during planned operation up to the design rod average burnup limit.

### 7.4.8 Clad Flattening

Design Category - Fuel Rod Failure

Design Parameter - Clad Flattening

Design Basis - Fuel rod failures will not occur due to clad flattening.

Acceptance Limit - The fuel rod design shall preclude clad flattening during projected exposure.

Design Evaluation - Westinghouse fabricated fuel is sufficiently stable with respect to fuel densification such that the axial column gaps that can form as a result of fuel densification and axial shrinkage are too small to allow clad flattening to occur and that axial column gaps that could occur are sufficiently small that a densification power spike factor of 1.0 is appropriate (Reference 7.25). These conclusions were based on extensive in-reactor operating experience, with associated on-site (visual examination and flux detector traces) and hot cell (gamma scanning) measured data. It was shown that Westinghouse fabrication processes are well controlled with respect to the parameters that impact fuel densification, such as fuel density, grain size and pore size distribution, such that adverse fuel performance issues associated with fuel densification (clad flattening), as observed in very early PWR fuel operation with densifying UO<sub>2</sub> fuel, do not occur.

These conclusions are applicable to current Westinghouse manufactured fuel. The approved (Reference 7.25) process for extending these conclusions to new fuel designs will be based on PAD5 analyses for time of pellet-clad gap closure (hard contact).



### 7.4.9 Clad Free Standing

Design Category - Fuel System Damage

Design Parameter - Clad Free Standing

Design Basis - The fuel system will not be damaged due to excessive fuel clad stress.

Acceptance Limit - The cladding shall be short-term free standing at beginning of life, at power, and during hot hydrostatic testing.

Design Evaluation – This criterion ensures that the fuel rod cladding has sufficient strength to preclude instantaneous collapse, or buckling, of the fuel rod cladding onto the fuel pellet due to compressive differential pressure loads across the clad wall. The limiting time for clad free standing is beginning of life due to the reduced strength of the unirradiated cladding material compared to saturation damaged irradiated cladding. Beginning of life hot conditions are limiting since the yield and ultimate strength of the cladding decrease with increasing temperature. The limiting loading is typically the maximum system pressure for hot hydrostatic testing. The clad free standing criterion is addressed either by analysis or by testing.

Clad free standing analyses are based on standard equations for the buckling of tubes under compressive differential pressure loading, as given in Reference 7.30, where the critical collapse differential pressure for a tube with no ovality,  $q_{cr}$ , is given by

$$q_{cr} = \frac{E}{4(1 - \nu^2)} \left( \frac{h}{r_0} \right)^3 \quad (7.12)$$

Where: E is the elastic modulus

$\nu$  is Poisson's Ratio

$h$  is clad thickness

$r_0$  is the clad mean radius

Actual fuel tubes have some initial ovality which reduces the differential pressure required for buckling of the fuel tube. From Reference 7.30, the critical differential pressure for buckling,  $q_{yp}$ , of a tube with initial radial ovality,  $w_o$ , is obtained by solving

$$q_{yp}^2 - \left[ \frac{\sigma_{yp}}{m} + (1 + 6mn)q_{cr} \right] q_{yp} + \frac{\sigma_{yp}}{m} q_{cr} = 0 \quad (7.13)$$

Where:  $\sigma_{yp}$  is the tube yield strength, psi

$q_{cr}$  is the critical collapse pressure for a circular tube from Eq. 7.8, psi

$m = R/h$

$n = w_o/R$

$R$  = mean tube radius, inches

$h$  = tube thickness, inches

As noted in Reference 7.30, this calculation of the critical pressure for collapse is conservative, since it represents the beginning of yielding in the extreme fibers of the clad wall. Buckling strength of actual cladding further exceeds this limit due to the as-fabricated condition of the cladding initial ovality where the orientation of the initial ovality is not constant but spirals around the tube due to the pilgering processes used to reduce the tube dimensions. As a result, there is internal resistance to collapse until a more uniform ovality orientation is developed due to cladding creep.

[

]<sup>a,c</sup>

#### 7.4.10 Fuel Pellet Overheating (Power to Melt only)

Design Category - Fuel Rod Failure

Design Parameter - Fuel Pellet Overheating

Design Basis - The fuel rods will not fail due to fuel centerline melting for Condition I and Condition II events.

Acceptance Limit - The fuel rod centerline temperature shall not exceed the fuel melt temperature during Condition I and II operation, accounting for degradation of the melt temperature due to burnup and the addition of integral burnable absorbers.

Design Evaluation - The Westinghouse PAD5 fuel performance code is used to evaluate fuel rod centerline temperatures. Analyses are performed to predict best estimate and upper bound fuel centerline temperatures as a function of rod local burnup and power based on a limiting steady state power history. These fuel centerline temperature analyses are the same as those performed to provide maximum fuel temperature input to LOCA and Non-LOCA safety analysis, Section 7.5.1. Upper bound fuel temperature results are summarized in tabular form as shown in the Table 7.4.10-1 example for centerline temperatures. Using these upper bound fuel centerline temperatures in conjunction with the burnup dependent fuel melting temperature limits given in Equation (6-14) in Section 6.1.5 for  $\text{UO}_2$  fuel and Equation (6-15) for  $\text{UO}_2\text{-Gd}_2\text{O}_3$  fuel, the local power level at which fuel centerline melt will occur is determined for each burnup step. Figure 7.4.10-1 provides an example power to melt limit as a function of burnup derived from PAD5 analyses. It is noted that fuel temperatures in excess of the melting temperature are shown in this example. These are calculated by PAD5 for use in the determination of the power to melt limit. However, PAD5 does not model the pellet changes that occur upon melting, so these results are not used except for the interpolation of the melt limit.

Additional details of the maximum fuel temperature calculations, including the treatment of uncertainties, are presented in Section 7.5.1 in the context of the fuel temperature analyses performed to generate safety analysis inputs.

---

**Table 7.4.10-1 Example PAD5 Upper Bound Fuel Centerline Temperature Table<sup>1</sup>**

--

---

<sup>1</sup> Shaded cells indicate centerline temperatures in excess of the fuel melt limit.

**Figure 7.4.10-1 Example Power to Melt Limit Results**

#### 7.4.11 Pellet/Clad interaction (PCI)

Design Category - Fuel Rod Failure

Design Parameter – Clad strain and fuel pellet overheating

Design Basis - The fuel rod will not fail due to pellet clad interaction.

Acceptance Limit – The NRC SRP does not require a specific design criterion for PCI. Two related criteria, the one percent clad strain criterion (Section 7.4.2) and the fuel overheating (no centerline fuel melt) criterion (Section 7.4.10), must be met.

Design Evaluation – As noted, the NRC SRP states that no criterion for fuel failure due to PCI or pellet/clad mechanical interaction (PCMI) has been established. Rather, two existing design limits, one percent transient clad strain and no fuel centerline melt, must be satisfied to provide protection against PCI or PCMI fuel failure. PCI addresses stress corrosion cracking mechanisms due to fission product embrittlement of the cladding, while PCMI is a stress driven failure mechanism. The one percent uniform clad strain criterion limits the clad strain during a transient to a range where the cladding has sufficient ductility to preclude strain related fuel failures. The fuel pellet overheating criterion precludes fuel melting and the associated large volume increase in the fuel due to the phase change which could result in excessive cladding stresses and strains. Classical stress corrosion cracking related PCI failures associated with transient power increases have not been observed as an active fuel failure mechanism in commercial PWR fuel operation. The few leaking fuel rods associated with local power changes that have been observed were determined to be caused by manufacturing defects such as missing pellet surfaces (MPS). These have been addressed as part of the overall industry fuel integrity improvement efforts. Therefore, no further analysis of PCI is required in the fuel rod design except that the transient clad strain and no fuel centerline melt design limits must be satisfied.

## 7.5 INTERFACE TO SAFETY ANALYSES

The PAD5 code is used to generate fuel temperature, rod internal pressure, core stored energy and additional fuel and cladding parameter inputs for LOCA and Non-LOCA safety analyses. These results are used to define the steady state initial conditions used to initialize the transient safety analysis codes. The following data are provided:

[

]<sup>a,c</sup>

PAD5 analyses performed to generate input to safety analyses are typically performed in advance of reload fuel rod design analyses. Detailed fuel rod power histories are not well defined for the future operating cycles to be bounded in the safety analyses of record. The basic fuel assembly and fuel rod mechanical design parameters are established as part of the design bases, but as-fabricated fuel and cladding data are not available. The above inputs for safety analysis are therefore generated using a

[

]<sup>a,c</sup> and they will be discussed

in each of the following subsections.

### 7.5.1 Maximum Fuel Temperatures

Fuel centerline, fuel average, and fuel surface temperatures, with and without uncertainties, are generated as a function of rod average burnup and local power covering the expected range of fuel operation. These temperatures are used as initial conditions in LOCA, containment, and non-LOCA safety analyses calculations. They are also used in the calculation of the core stored energy, as discussed in Section 7.5.5, and in the power to melt calculation discussed in Section 7.4.10.

Maximum fuel temperatures are generated to model steady state initial conditions for the limiting rods in the safety analysis. These are [

]<sup>a,c</sup>

[

] <sup>a,c</sup>

Table 7.5.1-1 provides an example list of uncertainty parameters [

] <sup>a,c</sup>



[

] <sup>a,c</sup>

Maximum fuel temperature calculations are performed for all fuel types to be analyzed in the plant safety analyses (i.e.,  $\text{UO}_2$  and/or  $\text{ZrB}_2$  IFBA,  $\text{Gd}_2\text{O}_3$ - $\text{UO}_2$ , or  $\text{Er}_2\text{O}_3$ - $\text{UO}_2$ ). For each unique fuel rod design, best estimate maximum fuel average temperatures are [

]<sup>a,c</sup> The final results are given as tables of best estimate and upper bound fuel average, fuel surface and fuel centerline temperature. Tables 7.5.1-2, 7.5.1-3 and 7.5.1-4 provide an example tables of upper bound maximum fuel average temperatures, upper bound fuel centerline temperatures, and upper bound fuel surface temperatures, respectively.

[

] <sup>a,c</sup>

Fuel temperatures generated for safety analyses inputs are typically generated in advance of the actual fuel operation and they are intended to bound the fuel temperatures expected for the actual fuel to be fabricated and operated in the core. They are confirmed to be applicable on a reload specific basis. [

] <sup>a,c</sup>

[

]<sup>a,c</sup>

**Table 7.5.1-1 Significant Uncertainty Parameters Included in the Maximum Fuel Temperature Uncertainty Calculation<sup>1</sup>**

Uncertainty <sup>2</sup>	Fuel Centerline	Fuel Average	Fuel Surface	a,c
--------------------------	-----------------	--------------	--------------	-----

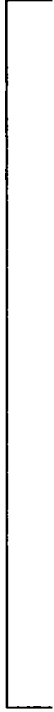
<sup>1</sup> Uncertainty values are based on Tables 7.3.2-1, 7.2.3-2, 7.3.2-3 and 7.3.2.4 for model, fabrication, plant operation and nuclear uncertainties, respectively.

<sup>2</sup> Note that for some uncertainty parameters, [  
] <sup>a,c</sup>

<sup>3</sup> Fuel modeled with Westinghouse densification model.

<sup>4</sup> [  
] <sup>a,c</sup> in fuel  
centerline or fuel average temperature calculations.

**Table 7.5.1-2 Example Upper Bound Maximum Fuel Average Temperatures (°F)**



---

**Table 7.5.1-3 Example Upper Bound Maximum Fuel Centerline Temperatures (°F) <sup>1</sup>**

--

---

<sup>1</sup> Shaded cells indicate centerline temperatures in excess of the fuel melt limit.



a,c

**Figure 7.5.1-1 Schematic Illustration of Burnup Step Selection for Maximum Fuel Temperature Calculations**

a,c

**Figure 7.5.1-2 Illustration of Power Ramping Steps in Fuel Temperature Calculations**

## 7.5.2 Rod Internal Pressure

Maximum and minimum rod internal pressure inputs corresponding to the lead power fuel rod are generated with PAD5 for input to safety analysis. The LOCA analysis uses these rod internal pressures as initial conditions. The methods used to generate these rod internal pressure results are consistent with those used for maximum fuel temperatures as described in Section 7.5.1.

[  
] <sup>a,c</sup> Table 7.5.3-1 provides an example set of significant uncertainty parameters for both upper and lower bound rod internal pressure analyses. To provide assurance that the rod internal pressure results will bound future as-fabricated fuel regions, the best estimate cases for the upper and lower bound rod internal pressure analyses are [

] <sup>a,c</sup>

The same steady state power history and axial power shapes used in the calculation of the maximum fuel average temperatures are used for the rod internal pressure analysis so that rod internal pressure results are on a consistent basis with the maximum fuel average temperature results. Analyses are performed for the same burnup steps as are used for the maximum fuel temperature analysis. Power ramps are modeled at each burnup step to obtain rod internal pressure results as a function of rod average burnup and power. In the rod internal pressure analysis power ramps, the rod power is increased [

] <sup>a,c</sup>

Rod internal pressure calculations are performed for all fuel types to be analyzed in the plant safety analyses (i.e., UO<sub>2</sub> and/or ZrB<sub>2</sub> IFBA, Gd<sub>2</sub>O<sub>3</sub>-UO<sub>2</sub>, or Er<sub>2</sub>O<sub>3</sub>-UO<sub>2</sub>). For each unique fuel rod design, [

] <sup>a,c</sup> The final results are given as tables of best estimate and upper bound and best estimate and lower bound rod internal pressure as a function of rod average burnup and power. Table 7.5.3-2 provides an example upper bound rod internal pressure table for fuel rod burnup from 0.0 to 62000 MWd/MTU.



**Table 7.5.3-1 Example Selection of Significant Uncertainties for Upper Bound and Lower Bound Rod Internal Pressure**

#	Uncertainty Parameter	Limiting Direction		Value	a,c
		Maximum RIP	Minimum RIP		

<sup>1</sup> For plants with calorimetric power uncertainty greater than or equal to 0.6% of rated core power.

<sup>2</sup> For designs using the Westinghouse fuel densification model.

**Table 7.5.3-2 Example Upper Bound Rod Internal Pressures (psi)**

--

### 7.5.3 Minimum Fuel Temperatures

Minimum fuel temperatures are also provided as input to safety analyses for initialization of non-LOCA transients. These temperatures are calculated based [

] <sup>a,c</sup> are generated as a function of rod average burnup and local power covering the expected range of fuel operation.

For minimum fuel temperature calculations, [

] <sup>a,c</sup>

All significant fabrication tolerances, model uncertainties, and operating uncertainties are accounted for in the calculation of lower bound minimum fuel temperatures, [

] <sup>a,c</sup> Table 7.5.2-1 provides an example list of uncertainty parameters which have been identified as significant using the sensitivity analysis process outlined in Section 7.3.2. As with maximum fuel temperatures, [

] <sup>a,c</sup> that can contribute significantly to the total uncertainty.

Minimum fuel temperature analyses are performed as a function of rod average burnup and power similar to the manner used to perform maximum fuel temperature calculations, as illustrated in Figures 7.5.1-1 and 7.5.1-2. Minimum fuel temperature tables, with uncertainties, are provided to safety analysis in tabular format as shown in the example lower bound minimum fuel average temperature table in Table 7.5.2-2. Similar to the maximum fuel temperatures, to conservatively envelope the PAD5-calculated fuel rod temperature inputs, the [

] <sup>a,c</sup>

Fuel temperatures generated for safety analyses inputs are typically generated in advance of the actual fuel operation and they are intended to bound the fuel temperatures expected for the actual fuel to be fabricated and operated in the core. They are confirmed to be applicable on a reload specific basis. To provide added confidence that fuel temperatures used in safety analyses address the full range of expected fabrication and plant operating conditions and therefore minimize the risk of reload fuel temperatures exceeding values used in analyses of record, [

] <sup>a,c</sup>

[ ]<sup>a,c</sup> values provides added assurance that the fuel temperature results provided for safety analyses will remain bounding for actual fuel to be operated in the plant.

**Table 7.5.2-1 Significant Uncertainty Parameters Included in the Minimum Fuel Temperature Uncertainty Calculation<sup>1</sup>**

Uncertainty	Fuel Centerline	Fuel Average	Fuel Surface
-------------	-----------------	--------------	--------------

a,c

<sup>1</sup> Uncertainty values are based on Tables 7.3.2-1, 7.2.3-2, 7.3.2-3 and 7.3.2.4 for model, fabrication, plant operation and nuclear uncertainties, respectively.

**Table 7.5.2-2 Example Lower Bound Minimum Fuel Average Temperatures (°F)****a,c****a,c**

#### 7.5.4 Fuel Rod Parameters

Additional fuel rod parameter data is provided to safety analysis to characterize the fuel analyzed in the maximum fuel temperature calculations. These data are used in pre-transient initialization for LOCA transient analyses. The following types of data are provided for each burnup step in the fuel temperature calculations:

[

] <sup>a,c</sup>

Parameter data are provided from the [

] <sup>a,c</sup> fuel temperatures are calculated.

Table 7.5.4-1 provides an example of the [ ] <sup>a,c</sup> data provided, Table 7.5.4-2 provides an example of the [ ] <sup>a,c</sup> data provided, and Table 7.5.4-3 shows an example of the [ ] <sup>a,c</sup> data provided.

---

**Table 7.5.4-1 Example Fuel Parameters Data – Dimensional Data, Gap Conductance, and Pellet Porosity**

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**Table 7.5.4-2 Example Fuel Parameters Data – Void Volumes**

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**Table 7.5.4-3 Example Fuel Parameters Data – Gas Inventory**

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### 7.5.5 Core Stored Energy

Core stored energy is defined as the amount of energy in the fuel rods in the core above the local coolant temperature. The following equation is used to calculate core stored energy:

$$\text{CSE} = \frac{(\rho \cdot C \cdot V)_{\text{pellet}} (T_{\text{pellet}} - T_{\text{coolant}}) + (\rho \cdot C \cdot V)_{\text{clad}} (T_{\text{clad}} - T_{\text{coolant}})}{q'} \quad (7-11)$$

Where:

- CSE = core stored energy, fps
- $\rho$  = pellet or cladding densities, lb/in<sup>3</sup>
- C = pellet or cladding specific heats, Btu/lb-°F
- V = pellet or cladding volumes per foot of length, in<sup>3</sup>/ft
- $T_{\text{pellet}}$  = pellet average temperature, °F
- $T_{\text{clad}}$  = cladding average temperature, °F
- $T_{\text{coolant}}$  = local coolant temperature, °F
- $q'$  = local linear power, Btu/sec-ft

When divided by  $q'$ , the units of core stored energy are full power seconds (fps).

Maximum fuel average temperatures with uncertainties are used to calculate core stored energy as a function of linear power and burnup. For a whole-core parameter such as core stored energy, the application of [

]<sup>a,c</sup>

As implied in the name, core stored energy is not specific to a single fuel rod, but it is representative of the energy in the entire core. To capture the effects of burnup and thermal conductivity degradation, the core stored energy is calculated with consideration of [

]<sup>a,c</sup> the core stored energy calculations are confirmed on a reload specific basis.

## 7.6 SECTION 7 REFERENCES

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## 8 PAD5 MODELS AND METHODS IMPROVEMENT PROCESS

The PAD5 fuel performance models and associated evaluation methods described in the prior sections of this report represent the current state of the art for Westinghouse fuel rod design. It is desirable to maintain these models and methods to a high degree of accuracy, since they are the basis for fuel rod design and safety analysis input calculations. Fuel performance models have a theoretical basis, but are also based on a developing technical understanding of material behaviors in-reactor with fitting coefficients to best predict key performance parameters. While much of the empirical database is well established, new performance data are being accumulated due to healthy fuel examinations, ongoing vendor development programs, joint industry test programs, and detailed failed fuel examinations. These new data may reflect an extension of prior test conditions (e.g., power levels, coolant chemistry) or significant improvements in measurement accuracy due to advancing developments in measurement techniques and equipment. As new data are acquired and as in-reactor material behaviors become better understood, improvements in performance models and associated methods may be developed as required to reflect these industry developments. An improvement is defined as a change in a model or method that improves predictions for key performance parameters relative to qualified measured performance data. Improvements may be assessed in terms of bias, ability to match data trends, or scatter (e.g., standard deviation).

The standard model development and licensing process can result in significant delay between the time that new data or new technology are developed and when it can be reflected in design applications. An ongoing model improvement process could lead to multiple submittals of incremental changes to fuel performance models and associated methods. The full review and approval process requires extensive time and resources, making incremental improvements impractical. The consequence can be that once models are approved by the NRC they remain fixed until there is sufficient safety and/or commercial incentive to engage in the model improvement and licensing process.

Fuel performance model and method improvements are developed under a controlled process with established documentation and verification and internal review following the Westinghouse Quality Management System. This section defines how this process will be applied to provide for a streamlined Models and Methods Improvement Process (MMIP) for models and methods improvements that satisfy established applicability criteria. Improvements that satisfy these applicability criteria may be implemented in design applications following NRC notification of the model(s) and/or method(s) change. Improvements that don't satisfy the applicability criteria for the MMIP would follow the standard licensing process, requiring NRC review and approval prior to implementation.

Section 8.1 summarizes the Westinghouse fuel rod performance model and design application method improvement process. The MMIP applicability criteria for determining the appropriate approval path for model and method improvements are defined in Section 8.2. The model and method improvement licensing process flow is defined in more detail in Section 8.3.

This PAD5 MMIP provides an alternative to the Westinghouse Fuel Criteria Evaluation Process (FCEP) methodology for fuel and cladding model updates outlined in Chapter 7 of Reference 8.1. The FCEP will continue to be applied for implementing fuel design changes as explained in Reference 8.1.

## **8.1 FUEL PERFORMANCE MODEL AND METHOD IMPROVEMENT PROCESS**

### **8.1.1 Process Fundamentals**

The fundamental objective of Westinghouse fuel rod design is to assure that:

1. The fuel system is not damaged as a result of normal operation and anticipated operational occurrences;
2. Fuel system damage is never so severe as to preclude control rod insertion when required;
3. The number of fuel rod failures is not underestimated for postulated accidents; and
4. Core coolability is always maintained.

Section 7.4 identifies the acceptance limits that must be met by the fuel rod design during Condition I and Condition II operation. Condition III and IV events are addressed by Westinghouse safety analysis methods.

To perform design analyses, fuel and cladding material property and in-reactor performance models are developed and incorporated into integrated software (e.g., PAD5 as described in Sections 3 through 6) to predict key fuel and cladding performance parameters. These models are generally developed on a best estimate basis by adjusting the performance model coefficients to best fit qualified measured fuel and cladding performance data with minimal bias. (The data qualification process is discussed in more detail in subsection 8.1.2.) Model performance trends are evaluated to confirm that there are no systematic biases with respect to key model inputs (e.g., fuel burnup, temperature, fluence). Engineering judgment may be used to determine if it is appropriate to allow a conservative model bias, based on evaluation of the sufficiency and/or quality of the measured data. Following completion of model development, model uncertainties are defined to bound at least 95% of the qualified database (i.e., the upper bound model will predict a value greater than or equal to the measured value for at least 95% of the relevant measured data points). In some cases, the model uncertainty may include multiple contributors, when it is not possible to separate out the contributions of uncertainty components. For example, measurement uncertainty in fission gas release measurements may not be adequately defined to justify removing it from the total fission gas release uncertainty.

Design methods are established to specify how design and safety analyses are to be performed to ensure safe and reliable fuel operation. For each type of analysis, appropriate input requirements are defined, including consideration of operating duty. Interfaces with other design disciplines are established to assure that inputs used in the analysis are consistent with their intended use. Basic assumptions are stated to define the limits of applicability of the methods. The treatment of model, fabrication and operational uncertainties is defined to ensure that significant uncertainties are accounted for. For each design parameter (e.g., rod internal pressure or clad strain), sensitivity studies are performed to identify those uncertainties that significantly impact that parameter. These studies consider the typical range of operating duty over which design analyses are performed. All uncertainty parameters that produce a significant impact on the design parameter are included in the design application methods for analysis of that parameter.

The model and method development process is performed in a controlled manner, with appropriate internal documentation and verification as per the Westinghouse Quality Management System.

### 8.1.2 Development Process

Fuel rod performance models are verified and validated against qualified experimental data. Methods used to perform fuel rod design analysis and to generate inputs to safety analysis have been defined based on established performance code response to models, fuel and cladding fabrication parameters, and plant operation inputs, with appropriate consideration of established uncertainties.

It is the objective of the model and methods development process to maintain or improve the accuracy of the fuel rod performance models and associated methods, with consideration of new fuel and cladding performance data and associated technical development in the understanding of fuel and cladding performance mechanisms and modeling techniques.

New fuel performance data continue to be obtained by Westinghouse and the nuclear industry. Much of this data for commercial reactor fuel is the result of healthy fuel examinations that are intended to validate continued good fuel performance while providing a means to identify potential performance concerns before they impact fuel performance in reactor. More detailed fuel and cladding examinations may be performed when fuel rod failures occur to identify cause of failure. Vendor specific and joint industry test programs are also conducted to further establish current fuel and cladding performance trends under more challenging operating conditions, to provide confirmatory data to validate established performance trends, and to investigate new fuel and cladding material behaviors. All of these data sources are subject to measurement and operational uncertainties, such that the data must be qualified.

The initial step for data review and qualification is to assess the new data relative to known trends for each measured parameter of interest. Data that trends as expected provides added confirmation that current performance models are performing accurately. When data exhibit a deviation from expected trends, a more detailed review of the test series or measurement campaign may be required. The purpose of this review is to validate the experimental procedures, (i.e., calibration and data reduction) to confirm consistency with those used to establish the existing database. The operating history of the fuel rods under examination relative to the operating history for rods in the existing database is also reviewed to identify potential differences that could explain the deviation in the trends. Manufacturing information for the test rods may also be reviewed to verify consistency with the database. The objective of these initial reviews is to verify that the measured data are valid, to establish the relevance of the data with respect to design applications, and to further confirm whether the observations are consistent with the known uncertainties in the current performance models.

Data that exhibit unexpected trends may require further validation before being included in the model development database. Additional measurements or measurement campaigns may be conducted to obtain additional data under similar operating conditions to confirm that the initial observations are valid. The extent of these additional investigations is dependent on the potential significance of the observed performance trend. Once sufficient data have been accumulated that validates the initial observations, and appropriate reviews of the measurements have been completed and documented, the data are qualified for use in assessing fuel rod design and performance impacts. Westinghouse internal procedures, such as the Westinghouse corrective action process, are used as required to document and track the observations and



the resolution with respect to design and safety impacts. Any observations that could adversely impact nuclear safety would be addressed consistent with the requirements of 10 CFR 21.

In addition to new data acquisition, fuel and cladding performance modeling technology is also under development both within Westinghouse and throughout the industry. Westinghouse participates with utilities individually and collectively in the development of increased understanding of in-reactor material behaviors and improvement in fuel performance modeling. Joint industry projects that sponsor ongoing in-reactor test programs also provide a forum for the discussion and development of fuel performance modeling technology. New modeling approaches may be investigated to determine if the accuracy of the predictions may be improved by reducing the variability in predicted versus measured results for the existing database and/or for improving predictions for new qualified measurement data.

Modifications to existing analysis methods may also be developed as part of improvement in the database and fuel performance technology. These methods improvements may be initiated in response to model improvements or modifications to plant operation strategies, to improve the accuracy of design and safety analyses, or to reduce the cycle time for performing design analyses. Methods modifications may also be required to provide additional input to safety analysis in response to emerging issues or changes in regulatory requirements. The modified methods must still satisfy the fundamental objective of the design process which is to assure the all fuel rod design and safety limits are satisfied with appropriate consideration of all significant uncertainties.

As models and/or methods improvements are identified and developed, the potential impact of methods improvements on plant safety and operation is assessed. Improvements that could adversely impact design and/or safety margins are tracked using the Westinghouse corrective action process.

As improvements to fuel performance models and methods are developed, the licensing requirements for implementation of these improvements must be addressed. The first step in this evaluation is to apply the MMIP applicability criteria presented in Section 8.2 to determine whether the improvements can be implemented under the MMIP streamlined process. Under the streamlined MMIP, after the model and methods development is completed within Westinghouse with appropriate documentation, independent verification and internal review requirements satisfied, the model and/or method improvements are summarized in a notification letter that is transmitted to the US NRC. In addition to the description of the improvement to be implemented, this letter will identify when the improvements are expected to be implemented in design application. Explicit NRC review and approval is not required under the streamlined process. If the MMIP applicability requirements in Section 8.2 are not satisfied, appropriate NRC review and approval of the changes must be obtained prior to use in Westinghouse design applications.

## 8.2 STREAMLINED MMIP APPLICABILITY CRITERIA

The streamlined process for fuel performance model and method improvements may be followed provided that the following applicability criteria are satisfied:

- [ ]<sup>a,c</sup>
- [ ]<sup>a,c</sup>
- [ ]<sup>a,c</sup>
- [ ]<sup>a,c</sup>

## 8.3 MODEL & METHOD IMPROVEMENT PROCESS SUMMARY

Figure 8.3-1 provides a schematic illustration of the Westinghouse fuel performance model and method licensing process.

Potential model and method improvements are defined based on ongoing data acquisition and qualification efforts and fuel performance technology developments. The decision to pursue specific model and method improvements is the responsibility of Westinghouse management based on the safety significance of the proposed improvements and other considerations. Once an improvement is accepted for development, the proposed changes are evaluated relative to the MMIP applicability criteria to determine the required licensing path.

Model and / or method development is then completed, with all appropriate testing, documentation, independent verification, and internal reviews. This stage of the development process is the same regardless of the licensing path that is determined from the MMIP applicability criteria review.

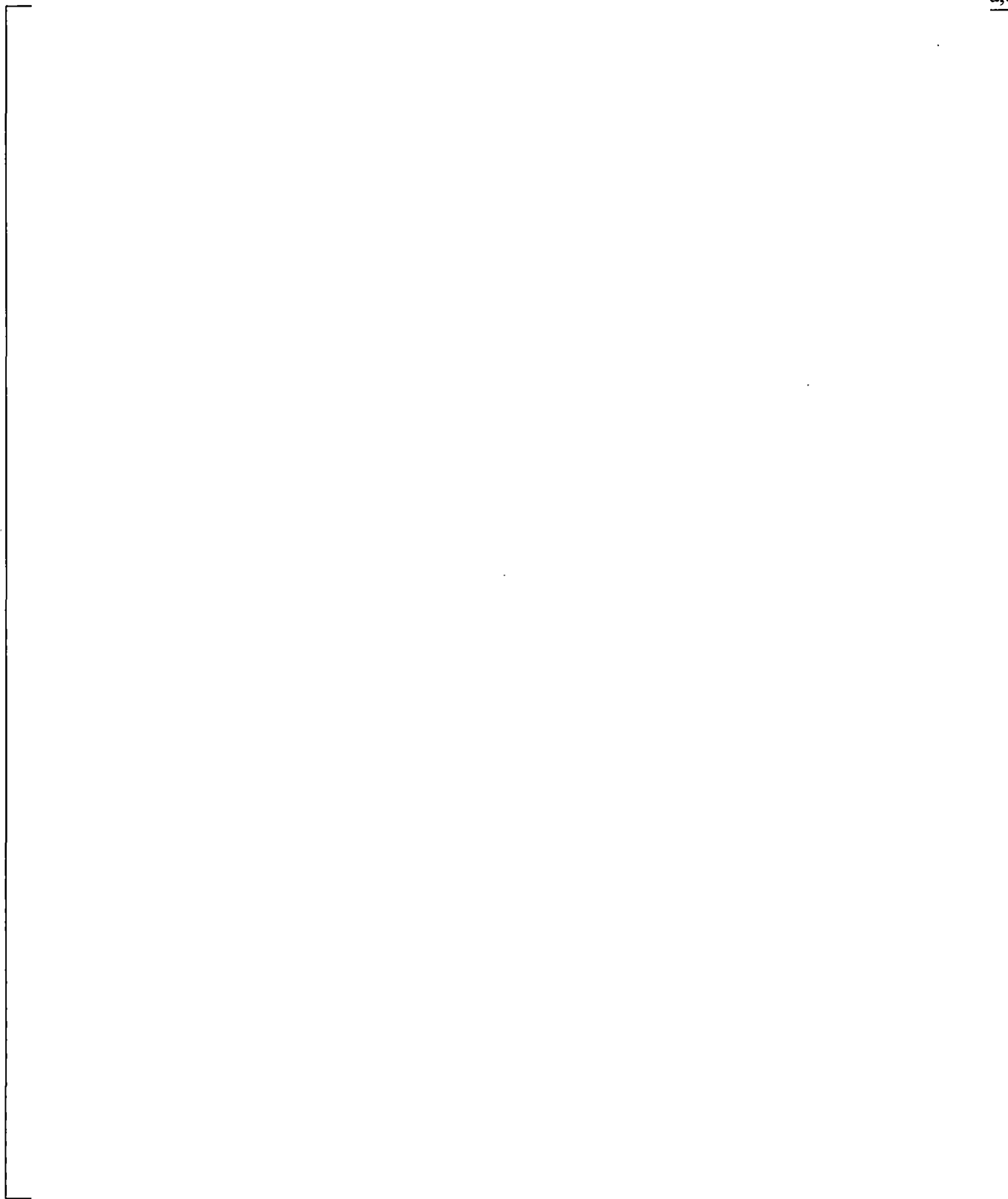
The licensing phase follows completion of the development effort. If the standard licensing process is required based on the MMIP applicability criteria or if it is deemed appropriate to request explicit NRC review and approval, appropriate licensing documentation will be developed to present the improvements to the NRC. Implementation of the change would then follow the completion of the NRC review and approval. If the model and/or method improvements satisfy the streamlined process criteria, a letter summarizing the improvements and the implementation plan will be sent to NRC for information only. No explicit review and approval is required by the NRC under the streamlined process, though the NRC may elect to audit the improvement development process.

Following completion of licensing requirements, the model and method improvements will be implemented in Westinghouse PWR fuel rod design and safety analyses, typically on a forward fit basis. After this PAD5 topical report becomes part of a plant's licensing basis, an improvement implemented under the streamlined process described in this section is not considered to be a departure from a method of evaluation described in the Final Safety Analysis Report (FSAR) (as updated) used in establishing the design bases or in the safety analyses. Therefore, changes to PAD5 models or methods using the streamlined MMIP do not result in a departure from a method of evaluation as defined in 10 CFR 50.59(a)(2).

#### **8.4 PROCESS APPLICATION**

[

] <sup>a,c</sup>



**Figure 8.3-1      PAD5 Fuel Performance Models and Methods Improvement Licensing Process Map**

## **8.5 SECTION 8 REFERENCES**

- 8.1 WCAP-12488-A (Proprietary) and WCAP-14204-A (Non-Proprietary), "Westinghouse Fuel Criteria Evaluation Process," October 1994.

## APPENDIX A

### PAD5 CALIBRATION/VALIDATION/UNCERTAINTY RESULTS

#### A.1 CALIBRATION PROCESS

##### A.1.1 Description of the Model Calibration Process

Many fuel rod performance models and material properties can be independently validated against measured data because they do not depend on or are only weakly dependent on the interrelated effects of the other fuel rod models. No integral code calibration is required to validate these models and material properties. These models and material properties include the cladding Young's modulus, cladding Poisson's ratio, fuel and cladding thermal expansion, cladding yield strength and ultimate strength, cladding and oxide thermal conductivity, cladding oxide thickness, and rod growth. However, certain models are more strongly dependent on how the overall code responds, and thus will be validated by an integral code calibration. Such models each have a direct impact on the following important fuel rod performance parameters:

- Fuel centerline temperature
- Steady state and transient fission gas release
- Cladding creep and deformation
- Fuel rod internal void volume.

An integral code calibration was performed as described in this appendix by evaluating these fuel rod performance parameters. The models that most strongly influence the calibration for the specific fuel rod performance parameter are discussed below.

Table A.1-1 summarizes the detailed PAD5 integral calibration steps and describes which model is primarily used to tune and optimize the best estimate predictions. The dependent fuel rod performance parameters of interest listed above (e.g. fuel centerline temperature) have been considered over specific ranges of fuel rod condition (e.g. open fuel-cladding gap) and operation (e.g. burnup or during power ramps). Each primary model is tuned to predict the calibration database for the fuel performance parameter of interest on [

]<sup>a,c</sup> The process is repeated as necessary to converge on the final set of calibrated models.

**Table A.1-1 PAD5 Integral Calibration Strategy**

Fuel Rod Performance Parameter	Primary Model Used for Code Calibration
--------------------------------	---

a,c

**Table A.1-1 PAD5 Integral Calibration Strategy  
(cont.)**

a,c

Fuel Rod Performance Parameter	Primary Model Used for Code Calibration
--------------------------------	---

The detailed calibration process for:

- fuel temperature,
- fission gas release,
- cladding creep,
- cladding diameter in response to ramp tests, and
- integral calibration confirmation
  - cladding outer diameter due to steady state operation to high burnup,
  - rod internal cold void volume at high burnup

is described in the following sections of this appendix. In each of these sections is an explanation of the calibration that has been performed, a description of the database used to perform the calibration, a presentation of the results of the calibration, and a description of the uncertainties, if appropriate, associated with the calibration.

The databases for each calibration category will be divided into calibration and validation sets, indicated by C and V in the tables listing the calibration and validation datasets in each fuel rod parameter calibration section that follows. [

] a,c

### A.1.2 Process for Generation of Model Uncertainties

The process of generating the PAD5 model uncertainties from the calibration data is based on the following process. For each fuel performance parameter or model for which an uncertainty is desired for use in design analysis, [

] <sup>a,c</sup>

This process is conservative because the resulting model uncertainties will [

] <sup>a,c</sup>

The following sections of this appendix include a description of the uncertainties, if appropriate, associated with the calibration of the fuel performance parameter in that section.

## A.2 CALIBRATION RESULTS

### A.2.1 Fuel Temperatures

#### A.2.1.1 Thermal Database

The thermal database used to calibrate PAD5 fuel centerline temperatures has been significantly expanded from that used for PAD 4.0 (Reference A.34). New high burnup data has been included from [

] <sup>a,c</sup> Table A.2.1-1 lists the available tests for the thermal calibration and validation of PAD5 and provides references to the reports where the experiments are described. While many rods have been added to the thermal database, others were excluded for a variety of reasons. These rods have been listed in Table A.2.1-2 along with justifications for their exclusion.



**Table A.2.1-1 PAD5 Thermal Calibration Database Test Program Reports**

Test	Number of Rods	Reference(s)
------	----------------	--------------

a,b,c

**Table A.2.1-2 Rods Excluded from Thermal Database**

Tests	Rod ID(s)	TC/ET	Reason for Exclusion
-------	-----------	-------	----------------------

a,b,c

**Table A.2.1-2 Rods Excluded from Thermal Database (cont.)**

Tests	Rod ID(s)	TC/ET	Reason for Exclusion	a,b,c
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]<sup>a,c</sup> The resulting database is shown in Table A.2.1-3, along with rod design parameters relevant to temperature predictions. The PAD5 thermal database includes both thermocouple measurements and expansion thermometer measurements. The column labeled ‘TC/ET’ refers to the temperature measuring device: TC for thermocouple (local centerline temperatures) or ET for expansion thermometer (axially averaged centerline temperatures).

**Table A.2.1-3 Summary of Rods in PAD5 Thermal Database**

Plant/Program	Rod ID	TC/ET	Gap <sup>(1)</sup> (mil)	Fill Gas	Backfill Pressure (psi)	EOL RABU <sup>(2)</sup> (MWd/MTU)	C/V	Gad (%)	a,b,c
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**Table A.2.1-3 Summary of Rods in PAD5 Thermal Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>TC/ ET</b>	<b>Gap<sup>(1)</sup> (mil)</b>	<b>Fill Gas</b>	<b>Backfill Pressure (psi)</b>	<b>EOL RABU<sup>(2)</sup> (MWd/MTU)</b>	<b>C/V</b>	<b>Gad (%)</b>	a,b,c
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**Table A.2.1-3 Summary of Rods in PAD5 Thermal Database (cont.)**

Plant/Program	Rod ID	TC/ ET	Gap <sup>(1)</sup> (mil)	Fill Gas	Backfill Pressure (psi)	EOL RABU <sup>(2)</sup> (MWd/MTU)	C/V	Gad (%)

**Notes:**

1. As-fabricated pellet-cladding diametral gap
2. End of Life Rod Average Burnup
3. Including extended operation to high burnup
4. [ ]<sup>a,b,c</sup>

a,b,c

The database described in Table A.2.1-3 contains [ ]<sup>a,c</sup>. However, points with measured temperatures [ ]<sup>a,c</sup> These points do not sufficiently exercise the code calculation of temperature rise and contribute little to the thermal model development. [ ]<sup>a,c</sup> Table A.2.1-4 provides further information of the distribution of data in the thermal database.

**Table A.2.1-4 Distribution of the Number of Rods and Data Points in the PAD5 Thermal Database**

Role	All Fuel		UO <sub>2</sub>		Gadolinia	
	Rods	Data Points	Rods	Data Points	Rods	Data Points

a,b,c

In addition to the thermal model database described in Table A.2.1-3, the results of the [ ]

[ ]<sup>a,b,c</sup> The predictions obtained with PAD5 are compared to the results of the experiment in Section A.2.1.2.

### A.2.1.2 Thermal Model Results

#### Thermal Model Calibration and Validation

As discussed in Section A.1.1, the first step in the thermal model calibration process was calibrating the thermal model to match BOL data. [

included data [ ]<sup>a,c</sup> The data used for BOL calibrations ]<sup>a,c</sup>

Figures A.2.1-1 through A.2.1-6 show BOL thermal calibration results. The figures present plots of predicted versus measured temperature results as well as residual graphs (predicted minus measured temperatures) versus LHGR. Separate plots are presented for the calibration and validation datasets, as well as for the entire database combined. The plots show good overall agreement between measurements and predictions. No anomalous trends with increasing LHGR are seen.

Note that thermocouple data points are a function of local values, while expansion thermometer data points are a function of rod average values. The figures themselves are an aggregate of local and rod average values for the quantities involved.

a,b,c

**Figure A.2.1-1 Predicted vs. Measured BOL Fuel Centerline (FCL) Temperatures for the Calibration Database**

a,b,c

**Figure A.2.1-2 Predicted Minus Measured BOL Fuel Centerline Temperatures vs. LHGR for the Calibration Database**

a,b,c

**Figure A.2.1-3 Predicted vs. Measured BOL Fuel Centerline Temperatures for the Validation Database**



a,b,c

**Figure A.2.1-4 Predicted Minus Measured BOL Fuel Centerline Temperatures vs. LHGR for the Validation Database**

a,b,c

**Figure A.2.1-5 Predicted vs. Measured BOL Fuel Centerline Temperatures for the Calibration and Validation Database.**

**Figure A.2.1-6 Predicted Minus Measured BOL Fuel Centerline Temperatures vs. LHGR for the Calibration and Validation Database**

Figures A.2.1-7 through A.2.1-15 show calibration and validation results for fuel centerline temperature from all burnups throughout life, including burnups below 500 MWd/MTU. The figures are organized in sets of three for each database subset. Each set includes the following:

- Predicted versus measured fuel centerline temperatures,
- Predicted minus measured fuel centerline temperature residuals versus LHGR, and
- Predicted minus measured fuel centerline temperature residuals versus burnup.

Figures A.2.1-7 through A.2.1-9 show results for UO<sub>2</sub> fuel (calibration and validation datasets).

Figures A.2.1-10 through A.2.1-12 show results for gadolinia fuel (calibration and validation datasets).

Figures A.2.1-13 through A.2.1-15 show results for UO<sub>2</sub> and gadolinia fuel combined.

Table A.2.1-5 provides statistics for PAD5 thermal calibration and validation.

Table A.2.1-5 Statistics for the Predicted Minus Measured (P-M) Residuals in the PAD5 Thermal Calibration and Validation									
Role	All Fuel			UO <sub>2</sub>			Gadolinia		
	Data Points	P-M Average (°F)	P-M Standard Dev. (°F)	Data Points	P-M Average (°F)	P-M Standard Dev. (°F)	Data Points	P-M Average (°F)	P-M Standard Dev. (°F)
									a,b,c
									a,b,c

**Figure A.2.1-7 Predicted vs. Measured In-Life Fuel Centerline Temperatures for the UO<sub>2</sub> Calibration and Validation Database**

a,b,c

**Figure A.2.1-8 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. LHGR for the UO<sub>2</sub> Calibration and Validation Database**

a,b,c

**Figure A.2.1-9 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. Burnup for the UO<sub>2</sub> Calibration and Validation Database**

a,b,c

**Figure A.2.1-10 Predicted vs. Measured In-Life Fuel Centerline Temperatures for the Gadolinia Calibration and Validation Database**

a,b,c

**Figure A.2.1-11 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. LHGR for the Gadolinia Calibration and Validation Database**



a,b,c

**Figure A.2.1-12 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. Burnup for the Gadolinia Calibration and Validation Database**

a,b,c

**Figure A.2.1-13 Predicted vs. Measured In-Life Fuel Centerline Temperatures**

a,b,c

**Figure A.2.1-14 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. LHGR for the UO<sub>2</sub> and Gadolinia Database**

**Figure A.2.1-15 Predicted Minus Measured In-Life Fuel Centerline Temperatures vs. Burnup for the UO<sub>2</sub> and Gadolinia Database**

It is concluded from Figures A.2.1-1 through A.2.1-15 that:

1. There are no anomalous trends with burnup (as shown in Figures A.2.1-9, A.2.1-12, and A.2.1-15) or LHGR (as shown in Figures A.2.1-2, A.2.1-4, A.2.1-6, A.2.1-8, A.2.1-11, and A.2.1-14) for any of the datasets.
2. There are no significant differences in the quality or behavior of the predictions for gadolinia and UO<sub>2</sub> rods. This is also seen in the statistics shown in Table A.2.1-5.
3. The validation dataset shows a somewhat larger scatter, but otherwise the results are similar to those seen in the calibration dataset (as shown in Figures A.2.1-7 through A.2.1-10)
4. For burnups [ ]<sup>a,c</sup> (as shown in Figure A.2.1-15), and the average value of the predicted minus measured residuals is [ ]<sup>a,c</sup>. This confirms that the degradation of the fuel thermal conductivity with burnup has been adequately modeled.
5. A significant number of points are contained [ ]<sup>a,c</sup> (as shown in Figures A.2.1-14 and A.2.1-15).

Therefore, based on the results and discussion presented in this section and in Figures A.2.1-1 through A.2.1-15, it is concluded that the PAD5 code fuel temperatures are generally in [

] <sup>a,c</sup>

### Comparison with the Results from the HBC Power-to-Melt Test

As described earlier, the results of [

] <sup>a,b,c</sup>

For the UO<sub>2</sub> rod, [

] <sup>a,b,c</sup>

Figures A.2.1-16 and A.2.1-17 show the PAD5 calculated radial fuel temperature profile for the UO<sub>2</sub> rod at nodes 8, and 11, respectively. Node 7 had essentially equivalent conditions as did node 8 and the results obtained for both nodes are also essentially equivalent. Fuel melting temperatures as calculated using the correlation in Equation (6-14), Section 6.1.5 are also shown in each figure for comparison.

The analysis shows that PAD5 predicts [

] <sup>a,b,c</sup> The results

obtained with PAD5 for the UO<sub>2</sub> rod are consistent with the observations reported for this rod.

For the gadolinia rod, [

] <sup>a,b,c</sup>

[

] <sup>a,b,c</sup>

Figure A.2.1-18 shows the calculated radial fuel temperature profile for the gadolinia rod at node 8 [ u ]<sup>a,c</sup> and Figure A.2.1-19 show the result for [

] <sup>a,c</sup> With the fuel melt temperature calculated using Equation (6-15) in Section 6.1.5, [

] <sup>a,c</sup>

These direct melting point comparisons show that PAD5 is acceptable for predicting high fuel temperatures approaching fuel melting.

a,b,c

**Figure A.2.1-16 Pellet Temperature Results for Nodal Elevation 8 of the UO<sub>2</sub> Rod of the HBC Test**

a,b,c

**Figure A.2.1-17 Pellet Temperature Results for Nodal Elevation 11 of the UO<sub>2</sub> Rod of the HBC Test**

a,b,c

**Figure A.2.1-18 Pellet Temperature Results for Node 8 of the Gadolinia Rod of the HBC Test with More Surface Peaked Radial Power Distribution**

**Figure A.2.1-19 Pellet Temperature Results for Node 8 of the Gadolinia Rod of the HBC Test with Less Surface Peaked Radial Power Distribution**

### A.2.1.3 Thermal Calibration Uncertainties

Upper and lower bound uncertainties were developed from the [ ]<sup>a,c</sup> It is observed that the prediction residuals for the cases [ ]

] <sup>a,c</sup>

[

] <sup>a,c</sup>

For the predicted [ ] <sup>a,c</sup> data, the upper-bound temperatures are calculated as:

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \end{array} \right] \begin{array}{c} \text{a,c} \\ \text{ } \end{array} \quad (\text{A.2-1})$$

For the predicted [ ] <sup>a,c</sup> data, the upper-bound temperatures are calculated as:

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \end{array} \right] \begin{array}{c} \text{a,c} \\ \text{ } \end{array} \quad (\text{A.2-2})$$



For the predicted [ ]<sup>a,c</sup> data, the lower-bound temperatures are calculated as:

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \end{array} \right]^{a,c} \quad (\text{A.2-3})$$

For the predicted [ ]<sup>a,c</sup> data, the lower-bound temperatures are calculated as:

$$\left[ \begin{array}{c} \text{ } \\ \text{ } \end{array} \right]^{a,c} \quad (\text{A.2-4})$$

where:

$T_{BE}$  is the best-estimate temperature predicted by PAD5 (°F),

$T_{UB}$  is the temperature to be used as upper-bound (°F),

$T_{LB}$  is the temperature to be used as lower-bound (°F), and

[ ]<sup>a,c</sup>

Figures A.2.1-20 through A.2.1-23 show the predicted minus measured temperature residuals calculated using the biased temperatures for upper and lower bound settings, as a function of LHGR and as a function of burnup.

a,b,c

**Figure A.2.1-20 Upper-Bound Fuel Centerline Temperatures Predicted Minus Measured vs. LHGR**

a,b,c

**Figure A.2.1-21 Upper-Bound Fuel Centerline Temperatures Predicted Minus Measured vs. Burnup**

a,b,c

**Figure A.2.1-22 Lower-Bound Fuel Centerline Temperatures Predicted Minus Measured vs. LHGR**

**Figure A.2.1-23 Lower-Bound Fuel Centerline Temperatures Predicted Minus Measured vs. Burnup**

## **A.2.2 Fission Gas Release**

### **A.2.2.1 Fission gas release database**

The PAD5  $\text{UO}_2$  fueled steady state Fission Gas Release (FGR) database, the  $\text{UO}_2$  fueled transient FGR database, and the  $\text{UO}_2\text{-Gd}_2\text{O}_3$  fueled steady state and transient FGR database are described in this section. Table A.2.2-1 summarizes the industry available FGR measurements by data set, which has been combined with Westinghouse's own data. In reviewing and modeling these data, PAD5 models did not accurately model certain datasets due to atypical fuel feature or behavior. The data summarized in Table A.2.2-2 will be excluded from calibration and validation, with justification for the dataset exclusion.

**Table A.2.2-1 FGR Test Program by Data Set**

<b>Validation Data Set</b>	<b>Number of Rods</b>	<b>References</b>
----------------------------	-----------------------	-------------------

a,b,c

**Table A.2.2-2 Data Sets Excluded from Calibration and Validation**

<b>Test Program/Rods</b>	<b>Atypical Fuel Feature and Behavior</b>	<b>Reason for Exclusion</b>
--------------------------	---	-----------------------------

a,b,c

The PAD5 steady-state  $\text{UO}_2$  fission gas release data has been [

] <sup>a,c</sup>. It should be noted that [

] <sup>a,c</sup>.

There are [

] <sup>a,c</sup>. The rods in each of these databases are listed in Tables A.2.2-3 (Thermal FGR database), A.2.2-4 (High Burnup Athermal FGR database), and A.2.2-5 (Low Burnup Athermal FGR database).

**Table A.2.2-3 PAD5 FGR Model Calibration Results: Thermal FGR Database**

Plant/Program	Rod ID	Rod Burnup (MWd/MTU)	Measured FGR (%)	Predicted FGR (%)	C/V
---------------	--------	-------------------------	---------------------	----------------------	-----

a,b,c

**Table A.2.2-3 PAD5 FGR Model Calibration Results: Thermal FGR Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	------------------------------	------------	-------

**Table A.2.2-3 PAD5 FGR Model Calibration Results: Thermal FGR Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
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**Table A.2.2-4 PAD5 FGR Model Calibration Results: High Burnup Athermal FGR Database**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	------------------------------	------------	-------



**Table A.2.2-4 PAD5 FGR Model Calibration Results: High Burnup Athermal FGR Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	------------------------------	------------	-------

**Table A.2.2-4 PAD5 FGR Model Calibration Results: High Burnup Athermal FGR Database (cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	------------------------------	------------	-------

**Table A.2.2-4 PAD5 FGR Model Calibration Results: High Burnup Athermal FGR Database (cont.)**

Plant/Program	Rod ID	Rod Burnup (MWd/MTU)	Measured FGR (%)	Predicted FGR (%)	C/V
---------------	--------	-------------------------	---------------------	----------------------	-----

a,b,c

**Table A.2.2-5 PAD5 FGR Model Calibration Results: Low Burnup Athermal FGR Database**

Plant/Program	Rod ID	Rod Burnup (MWd/MTU)	Measured FGR (%)	Predicted FGR (%)	C/V
---------------	--------	-------------------------	---------------------	----------------------	-----

a,b,c

**Table A.2.2-5 PAD5 FGR Model Calibration Results: Low Burnup Athermal FGR Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	<b>C/V</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	------------------------------	------------	-------

**Table A.2.2-5 PAD5 FGR Model Calibration Results: Low Burnup Athermal FGR Database (cont.)**

Plant/Program	Rod ID	Rod Burnup (MWd/MTU)	Measured FGR (%)	Predicted FGR (%)	C/V	a,b,c
---------------	--------	----------------------	------------------	-------------------	-----	-------

There are [ ]<sup>a,c</sup> in the UO<sub>2</sub> fueled transient fission gas release database. Note that rods with less than 3% measured FGR were not used as part of calibration or validation, however, they were used as part of the uncertainty evaluation. The rods in each of these databases are listed in Table A.2.2-6 (Transient FGR database).

**Table A.2.2-6 PAD5 FGR Model Calibration Results: Transient FGR Database**

Plant/Program	Rod ID	Rod Burnup (MWd/MTU)	Measured FGR (%)	Predicted FGR (%) <sup>(1)</sup>	C/ V	Refabricated	a,b,c
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**Table A.2.2-6 PAD5 FGR Model Calibration Results: Transient FGR Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)<sup>(1)</sup></b>	<b>C/ V</b>	<b>Refabricated</b>	a,b,c
----------------------	---------------	---------------------------------	-----------------------------	--	-------------	---------------------	-------

a,b,c

1. Predicted FGR values are the FGR due to [ ]<sup>a,c</sup>

<sup>a,c</sup>). The rods inMay 2016  
Revision 1

**Table A.2.2-7 PAD5 FGR Model Calibration Results: Gadolinia Fuel Steady-State FGR Database**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Gadolinia Concentration (w/o)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	a,b,c
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**Table A.2.2-8 PAD5 FGR Model Calibration Results: Gadolinia Fuel Transient FGR Database**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Burnup (MWd/MTU)</b>	<b>Gadolinia Concentration (w/o)</b>	<b>Measured FGR (%)</b>	<b>Predicted FGR (%)</b>	a,b,c
----------------------	---------------	---------------------------------	--	-----------------------------	------------------------------	-------



### A.2.2.2 Fission Gas Release Model Results

The low burnup athermal fission gas release model is described as Equation (4-11) of Section 4.3.2. The constant  $F_{ath1}$  has been determined by fitting the equation for the athermal knockout/recoil release to the data in the low burnup athermal fission gas release database for which negligible thermal fission gas release is predicted. There is no high burnup athermal release due to the low burnup of the selected rods. The fit of the athermal fission gas release model to the 24 rods selected by this criterion is shown in Figure A.2.2-1. The remaining rods in the low burnup athermal database were used to validate the model. Figures A.2.2-2 and A.2.2-3 show the predicted versus measured fission gas release and predicted minus measured fission gas release as a function of rod average burnup, respectively, for the low burnup athermal fission gas release database (calibration and validation data sets combined). It is expected that some portion of the release is coming from thermal release for the validation data sets. Figures A.2.2-1 through A.2.2-3 show that PAD5 predicts the FGR for these low burnup rods with athermal dominated release.

a,b,c

**Figure A.2.2-1 PAD5 Athermal Fission Gas Release Model Predictions vs. Measured Gas Release with Essentially No Thermal Fission Gas Release**

a,b,c

**Figure A.2.2-2 PAD5 Fission Gas Release Predictions vs. Measured Fission Gas Release for the Low burnup Athermal Fission Gas Release Database**

**Figure A.2.2-3 PAD5 Fission Gas Release Predicted Minus Measured Results vs. Rod Average Burnup for the Low Burnup Athermal Fission Gas Release Database**

The high burnup athermal FGR model is described as Equations (4-12a) and (4-12b) in Section 4.3.2. The model has been calibrated based on a subset of the available data base and validated by comparisons with the remaining database. About half of the database was selected for the calibration. The selection of calibration data was random with the constraint that the fuel rods from any assembly should not be heavily over represented in either the calibration or validation subsets. Although the athermal (both low and high burnup release) are the dominate mechanism for these rods, there may still be a fair amount of thermal FGR in these rods. The coefficients ( $Bu_{enh}$ ,  $1/a$ , and  $b$ ) in the high burnup athermal FGR model along with the thermal FGR model, which are discussed next, have been determined to give a best estimate fit to the data in the high burnup athermal FGR database. The model constants are adjusted until the average predicted minus measured FGR is approximately 0. Figures A.2.2-4 and A.2.2-5 show the predicted versus measured fission gas release and predicted minus measured fission gas release as a function of rod average burnup for the high burnup athermal fission gas release database.

a,b,c

**Figure A.2.2-4 PAD5 Fission Gas Release Predictions vs. Measured Fission Gas Release for the High Burnup Athermal Fission Gas Release Database**

**Figure A.2.2-5 PAD5 Fission Gas Release Predicted Minus Measured Results vs. Rod Average Burnup for the High Burnup Athermal Fission Gas Release Database**

The thermal FGR model is described with Equations (4-7a), (4-7b), and (4-7c) in Section 4.3.1. Similar to the high burnup athermal model, the thermal model has been calibrated based on a subset of the available thermal database and validated by comparisons with the remaining database. About half of the database was selected for the calibration. The selection of calibration data was random with the constraint that the fuel rods from any assembly should not be heavily over represented in either the calibration or validation subsets. The thermal release is the dominate release mechanism for these data. The model parameters ( $B_0$ ,  $T_0$ ,  $BL$ ,  $TL$ ,  $BM$ ,  $TM$ ,  $BH$ ,  $TH1$ ,  $BSL_0$ ,  $TSL_0$ ,  $BSLL$ ,  $TSLL$ ,  $BSLM$ ,  $TSLM$ ,  $BSLH$ ,  $TSLH$ ,  $BSLH1$ ,  $TSLH1$ ), which determine the release threshold  $B_{INC}(T)$  and  $B_{REP}(T)$  (Equation (4-6) in Section 4.3.1) are adjusted to give a best estimate fit to the data in the FGR databases. The model constants are adjusted until the average predicted minus measured FGR is approximately 0. Figures A.2.2-6 and A.2.2-7 show the predicted versus measured fission gas release and predicted minus measured fission gas release as a function of rod average burnup for the thermal fission gas release database.

a,b,c

**Figure A.2.2-6 PAD5 Fission Gas Release Predictions vs. Measured Fission Gas Release for the Thermal Fission Gas Release Database**

**Figure A.2.2-7 PAD5 Fission Gas Release Predicted Minus Measured Results vs. Rod Average Burnup for the Thermal Fission Gas Release Database**

The transient fission gas release model is described as Equations (4-8), (4-9a), and (4-9b) in Section 4.3.1. With the calibrated steady-state thermal FGR model, the transient FGR model is calibrated by adjusting the effective grain size parameter XDIF that results in a best estimate fit, again determined by obtaining an average predicted minus measured fission gas release approaching zero. Figures A.2.2-8 and A.2.2-9 show the predicted versus measured fission gas release and the predicted minus measured fission gas release as a function of rod average burnup for the transient fission gas release database.

a,b,c

**Figure A.2.2-8 PAD5 Fission Gas Release Predictions vs. Measured Fission Gas Release for the Transient Fission Gas Release Database**



**Figure A.2.2-9 PAD5 Fission Gas Release Predicted Minus Measured Results vs. Rod Average Burnup for the Transient Fission Gas Release Database**

[ ]<sup>a,c</sup>  
The gadolinia fuel thermal fission gas release multiplication factor is determined to give an average predicted minus measured fission gas release of approximately zero for the gadolinia fuel with measured fission gas releases greater than 3%, since these are the gadolinia fuel data which are expected to have a significant sensitivity to the thermal fission gas release model. Figures A.2.2-10 through A.2.2-12 show the predicted versus measured fission gas release, predicted minus measured fission gas release as a function of rod average burnup for the gadolinia fission gas release database, and predicted minus measured fission gas release as a function of gadolinia concentration, respectively. There is no trend observed as a function of burnup or gadolinia concentration.

a,b,c

**Figure A.2.2-10 PAD5 Fission Gas Release Predictions vs. Measured Fission Gas Release for the Gadolinia Fuel Fission Gas Release Database**

a,b,c

**Figure A.2.2-11 PAD5 Fission Gas Predicted Minus Measured Results vs. Rod Average Burnup for the Gadolinia Fuel Fission Gas Release Database**

a,b,c

**Figure A.2.2-12 PAD5 Fission Gas Predicted Minus Measured Results vs. Gadolinia Concentration for the Gadolinia Fuel Fission Gas Release Database**

Tables A.2.2-3 through A.2.2-8 in Section A.2.2.1 give a complete listing of the fission gas release data and the PAD5 fission gas release predictions for all the rods in the fission gas release database. Statistics for the PAD5 FGR calibration results are given in Table A.2.2-9.

<b>Table A.2.2-9 PAD5 FGR Model Results Statistics</b>			
<b>Thermal Fission Gas Release Database</b>			
	<b>Calibration Database</b>	<b>Validation Database</b>	<b>Entire Database</b>
Number of rods			
P – M average (%)			
P – M $\sigma$ (%)			
<b>High Burnup Athermal (Greater than 45 GWd/MTU) Fission Gas Release Database</b>			
	<b>Calibration Database</b>	<b>Validation Database</b>	<b>Entire Database</b>
Number of rods			
P – M average (%)			
P – M $\sigma$ (%)			

Table A.2.2-9 PAD5 FGR Model Results Statistics (cont.)				
Low Burnup Athermal (Less than 45 GWd/MTU) Fission Gas Release Database				
	Calibration Database	Validation Database	Entire Database	
Number of rods				
P – M average (%)				
P – M $\sigma$ (%)				
Complete Athermal Fission Gas Release Database				
	Calibration Database	Validation Database	Entire Database	
Number of rods				
P – M average (%)				
P – M $\sigma$ (%)				
Transient Fission Gas Release Database				
	Calibration Database	Validation Database	Entire Database Measured FGR $\geq 3\%$	Entire Database
Number of rods				
P – M average (%)				
P – M $\sigma$ (%)				
Gadolinia Fission Gas Release Database, Measured FGR $> 3\%$				
	Steady State FGR Data	Transient FGR Data	Entire Database	
Number of rods				
P – M average (%)				
P – M $\sigma$ (%)				

### A.2.2.3 Fission Gas Release Model Uncertainties

In PAD5, [

] <sup>a,c</sup>. The athermal FGR model

uncertainty is given by [

] <sup>a,c</sup>. The

thermal FGR model uncertainty is derived by [

] <sup>a,c</sup>. For transient FGR, [

] <sup>a,c</sup>. The UB/LB

FGR results are obtained by using [

] <sup>a,c</sup>.

For example for the UB FGR results, the EOL FGR for each rod is calculated with PAD5 for [

]<sup>a,c</sup>

The UB SRSS value is calculated with the expression:

$$\left[ \begin{array}{c} \text{a,c} \\ \text{ } \end{array} \right] \quad (\text{A.2-5})$$

The final FGR model uncertainties and BE model parameters are summarized in Table A.2.2-10.

Figures A.2.2-13 through A.2.2-19 compare the bounding models with the measured fission gas release data. The UB FGR model [ ]<sup>a,c</sup> of the UO<sub>2</sub> fuel thermal FGR data, [ ]<sup>a,c</sup> of the UO<sub>2</sub> fuel athermal FGR data, [ ]<sup>a,c</sup> of the transient UO<sub>2</sub> fuel FGR data, and [ ]<sup>a,c</sup> of the total steady-state and transient UO<sub>2</sub> fuel thermal FGR data. The LB FGR model bounds [ ]<sup>a,c</sup> of the UO<sub>2</sub> fuel thermal FGR data, [ ]<sup>a,c</sup> of the UO<sub>2</sub> fuel athermal FGR data, and [ ]<sup>a,c</sup> of the total steady-state and transient UO<sub>2</sub> fuel thermal FGR data. The model uncertainties are appropriate for gadolinia fuel without further modification.

Model Coefficient	BE	UB	LB
-------------------	----	----	----

a,c

a,c

\_\_\_\_\_

1. Note that the [

Ja,c

a,b,c

**Figure A.2.2-13 Upper Bound  $\text{UO}_2$  Steady-State Fission Gas Release Model, Thermal Release Dominated**

a,b,c

**Figure A.2.2-14 Upper Bound  $\text{UO}_2$  Steady-State Fission Gas Release Model, Athermal Release Dominated**



a,b,c

**Figure A.2.2-15 Lower Bound  $\text{UO}_2$  Steady-State Fission Gas Release Model, Thermal Release Dominated**

a,b,c

**Figure A.2.2-16 Lower Bound  $\text{UO}_2$  Steady-State Fission Gas Release Model, Athermal Release Dominated**

a,b,c

**Figure A.2.2-17 Upper Bound  $\text{UO}_2$  Transient Fission Gas Release Model**

a,b,c

**Figure A.2.2-18 Upper Bound Gadolinia Fission Gas Release Model**

**Figure A.2.2-19 Lower Bound Gadolinia Fission Gas Release Model**

### **A.2.3 Cladding Creep**

#### **A.2.3.1 Cladding Creep Database**

Table A.2.3-1 provides a summary of the creep database used to calibrate and validate the PAD5 cladding creep model and to quantify the model uncertainties. Table A.2.3-1 includes a brief description of the test characteristics for each set of data. Except for the [

] <sup>a,c</sup>, all of the ZIRLO and **Optimized ZIRLO** in-reactor data is [

] <sup>a,c</sup>, while the Zircaloy-4 in-reactor data is [

] <sup>a,c</sup>. These rods have not

experienced significant pellet-clad contact, hence the measured diametral profilometry reflects the evolution of the permanent deformation of the clad geometry. The Plant R samples also do not experience any pellet-clad contact, by the design of the samples.

**Table A.2.3-1 Summary PAD5 Cladding Creep Database**

Campaign	Material	Comments
----------	----------	----------

a,b,c

For comparison purposes, the measured in-reactor profilometry data has been [ ]<sup>a,c</sup>, and [ ]<sup>a,c</sup> of the rod have been considered. [ ]<sup>a,c</sup>

### **A.2.3.2 Cladding Creep Model Calibration**

The ZIRLO and **Optimized ZIRLO** cladding creep database has been separated into two roughly equal subsets, namely the Calibration (C) and Validation (V) datasets. The separation process was achieved by first sorting the list of unique rod names and then selecting every other rod (this ensures a roughly 50%/50% selection from each campaign over all campaigns). The Zircaloy-4 database was similarly separated. Table A.2.3-2 summarizes the separation of the database into C and V data sets.

**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)<sup>(1)</sup></b>	<b>Cladding</b>	<b>Calibration (C)/Validation (V)</b>	<b>a,b,c</b>
----------------------	---------------	---	-----------------	---	--------------

**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)<sup>(1)</sup></b>	<b>Cladding</b>	<b>Calibration (C)/Validation (V)</b>	<b>a,b,c</b>
----------------------	---------------	---	-----------------	---	--------------



**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)<sup>(1)</sup></b>	<b>Cladding</b>	<b>Calibration (C)/Validation (V)</b>	a,b,c
----------------------	---------------	---	-----------------	---	-------

**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)<sup>(1)</sup></b>	<b>Cladding</b>	<b>Calibration (C)/Validation (V)</b>	<b>a,b,c</b>
----------------------	---------------	---	-----------------	---	--------------

**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)<sup>(1)</sup></b>	<b>Cladding</b>	<b>Calibration (C)/Validation (V)</b>	a,b,c
----------------------	---------------	---	-----------------	---	-------

**Table A.2.3-2 PAD5 Clad Creep Calibration and Validation Database (cont.)**

Plant/Program	Rod ID	Rod Average Burnup (MWd/MTU) <sup>(1)</sup>	Cladding	Calibration (C)/Validation (V)	a,b,c
---------------	--------	---	----------	--------------------------------	-------

PAD5 clad creep model is discussed in Section 5.4. For ZIRLO and **Optimized ZIRLO** cladding, the calibration of the irradiation creep model (Equation (5-26) in Section 5.4) is [ ]<sup>a,c</sup> and the calibration of the thermal creep model (Equation (5-22) in Section 5.4) is [ ]<sup>a,c</sup>

The final calibration was [ ]

] <sup>a,c</sup>

The validation dataset was used to confirm that the predictions remain for a fully independent set of data. Table A.2.3-3 (for ZIRLO and **Optimized ZIRLO** cladding) and Table A.2.3-4 (for Zircaloy-4) show that the combined calibration and validation datasets (All) have very similar best estimate creep statistics.

**Table A.2.3-3 ZIRLO/Optimized ZIRLO Best Estimate Cladding Creep Statistics**

Dataset	Sample Size	Average of Predicted Minus Measured Diameter (mils)	Standard Deviation of Predicted Minus Measured Diameter (mils)	a,b,c
All				
Calibration				
Validation				

Table A.2.3-4 Zircaloy-4 Best Estimate Cladding Creep Statistics				a,b,c
Dataset	Sample Size	Average of Predicted Minus Measured Diameter (mils)	Standard Deviation of Predicted Minus Measured Diameter (mils)	
All				
Calibration				
Validation				

The PAD5 prediction versus measured results for the ZIRLO and **Optimized ZIRLO** in-reactor creep and for the Plant R creep data are shown in Figures A.2.3-1 and A.2.3-2, respectively. "In-reactor" in the following figures refers to commercial fuel rods. To assess trends in the ZIRLO and **Optimized ZIRLO**, in-reactor creep predictions the residual (prediction minus measured cladding diameter) versus some key model input parameters (burnup and clad temperature (elevation)) are shown in Figures A.2.3-3 and A.2.3-4. Over the [ ]<sup>a,c</sup> of the rod the PAD5 model residual does not have any significant trends with burnup or elevation. The PAD5 BE model has acceptable stress, temperature (as shown in the elevation plots), and fast fluence (and flux) dependence over the whole range of conditions that are typical of one cycle fuel rod exposure.

a,b,c

**Figure A.2.3-1    ZIRLO and *Optimized ZIRLO* Cladding In-reactor Creepdown**

a,b,c

**Figure A.2.3-2 ZIRLO Cladding Creep From Plant R Creep and Growth Program**

a,b,c

**Figure A.2.3-3    ZIRLO and *Optimized ZIRLO* Cladding In-reactor Residual (M – P) vs. Nodal Burnup**





**Figure A.2.3-4 ZIRLO Cladding In-reactor Residual (P – M) vs. Normalized Elevation**

For all practical purposes, [

] <sup>a,c</sup> Figures A.2.3-5 and A.2.3-6 show the creep results by ZIRLO and **Optimized ZIRLO** data sets.

a,b,c

**Figure A.2.3-5 ZIRLO and *Optimized ZIRLO* Cladding In-reactor Creepdown**

**Figure A.2.3-6 ZIRLO and *Optimized ZIRLO* Cladding In-reactor Residual (P – M) vs. Nodal Burnup**

As discussed in Section 5.4, there is [

]<sup>a,c</sup> For Zircaloy-4, [ ]<sup>a,c</sup> were used in the best estimate calibration. Figures A.2.3-7 through A.2.3-9 show the Zircaloy-4 best estimate results. To assess trends in the Zircaloy-4 creep predictions, the residual (predicted minus measured clad diameter) versus some key model input parameters (burnup and clad temperature (elevation)) are shown in Figures A.2.3-8 and A.2.3-9.

a,b,c

**Figure A.2.3-7 Zircaloy-4 In-reactor Predicted vs. Measured Creepdown**

a,b,c

**Figure A.2.3-8 Zircaloy-4 In-reactor Creepdown Residual ( $P - M$ ) vs. Nodal Burnup**

**Figure A.2.3-9 Zircaloy-4 In-reactor Creepdown Residual ( P – M ) vs. Normalized Elevation**

The online measurement data from the Halden tests (Reference A.18 and A.19) are also used to confirm the capability of the model to predict the history of cladding creep under irradiation. Figures A.2.3-10 through A.2.3-12 show the PAD5 predictions of creep strain as a function of time for the Halden ZIRLO tests. The conditions of the Halden ZIRLO creep test are listed in Table A.2.3-5. These Halden tests were performed at [

] <sup>a,c</sup> hence it is valuable to quantify the steady state creep rate under these conditions, representing the top end of the rod. These tests, due to the higher temperature, have [

] <sup>a,c</sup>. It should be noted that the Halden ZIRLO creep test [

] <sup>a,c</sup>.

**Table A.2.3-5 Halden ZIRLO Creep Test Conditions**

Test	Hoop Stress [MPa]	Fast Flux [ $10^{13}$ n/cm <sup>2</sup> -s E>1 MeV]	Temperature [°F]
IFA 663-1			
IFA 663-2			
IFA 663-3			
IFA 617			

a,b,c

a,b,c

**Figure A.2.3-10 Halden ZIRLO Cladding Creep Tests IFA663-1 and IFA663-2**

a,b,c

**Figure A.2.3-11 Halden ZIRLO Cladding Creep Test IFA663-3**





**Figure A.2.3-12 Halden ZIRLO Cladding Creep Test IFA617**

The online measurement data from the [ ]<sup>a,c</sup> are used to confirm the capability of the model to predict the history of creep strain. Figure A.2.3-13 shows the PAD5 predictions of creep strain as a function of time for the [ ]<sup>a,c</sup>. This test was performed at high temperature (662 °F) and high fast flux ( $16.9 \times 10^{13}$  n/cm<sup>2</sup>/s), hence it is valuable to quantify the steady state creep rate under these prototypic limiting conditions.

**Figure A.2.3-13 The CEA Zircaloy-4 Creep Test**

### A.2.3.3 Cladding Creep Model Uncertainty

To quantify the uncertainty in the PAD5 creep model,  $J^{a,c}$ , is applied to the creep model, which is equal to unity in the best estimate (BE) case. The Upper Bound (UB) and Lower Bound (LB)  $J^{a,c}$  are determined from  $J^{a,c}$  as used in the BE calculations in Section A.2.3.2. The LB and UB ACREEP values for PAD5 are listed in Table A.2.3-6. All alloys (Zircaloy-4, ZIRLO, and **Optimized ZIRLO**) have the same functional form. Further, for convenience, the PAD5 creep model also considers that  $J^{a,c}$ .

The Zircaloy-4, ZIRLO and **Optimized ZIRLO** UB and LB values are determined by  $J^{a,c}$ . Figures A.2.3-14 and A.2.3-15 show the ZIRLO and **Optimized ZIRLO** UB and LB predictions, respectively. Figures A.2.3-16 and A.2.3-17 show the Zircaloy-4 UB and LB predictions respectively.

Table A.2.3-6 Bounding PAD5 Creep Model Alloy Specific Material ACREEP		
Cladding Material	LB ACREEP	UB ACREEP
Zircaloy-4		
ZIRLO and <b>Optimized ZIRLO</b>		

a,b,c


**Figure A.2.3-14 ZIRLO and *Optimized ZIRLO* Cladding Upper Bound Predictions**

a,b,c

**Figure A.2.3-15 ZIRLO and *Optimized ZIRLO* Cladding Lower Bound Predictions**

a,b,c

**Figure A.2.3-16 Zircaloy-4 Upper Bound Predictions**



**Figure A.2.3-17 Zircaloy-4 Lower Bound Predictions**

## **A.2.4 Cladding Diameter Change from Ramp Tests**

### **A.2.4.1 Ramp Diameter Change Database**

Table A.2.4-1 summarizes the available ramp cladding diameter change measurements by data set. In reviewing and modeling these data, PAD5 models did not accurately model certain datasets due to either atypical fuel design or defective measurements. The data summarized in Table A.2.4-2 will be excluded from calibration and validation.

**Table A.2.4-1 Cladding Diameter Change Measurements by Data Set**

Validation Data Set	Number of Rods	References
---------------------	----------------	------------

a,b,c

**Table A.2.4-2 Cladding Diameter Change Data Sets Excluded from Calibration and Validation**

Test Program/Rods	Atypical Fuel Features	Reason for Exclusion
-------------------	------------------------	----------------------

a,b,c

The data used for calibrating fission gas swelling during long ramps are listed in Table A.2.4-3.

[

] <sup>a,c</sup> This data set covers a wide range of burnup levels. For calibration and validation purposes these data are further divided into a Calibration (C) set and Validation (V) set as shown in the C/V column in Table A.2.4-3. The selection of rods for the calibration or the validation data set is random in nature, except that the [

] <sup>a,c</sup> are intentionally kept in the calibration set.

**Table A.2.4-3 Long Term Ramp Tests Data Used for Ramp Diameter Change Calibration and Validation**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Burnup (GWd/MTU)</b>	<b>Ramp Terminal LHGR (RTL) (kW/ft)</b>	<b>Holding Time (h)</b>	<b>C/V</b>
----------------------	---------------	-----------------------------	---	-----------------------------	------------

a,b,c

The data used for calibrating fission gas swelling during short ramps are listed in Table A.2.4-4. [

] <sup>a,c</sup>



**Table A.2.4-4 Ramp Tests Data Used for Justification of Reduced Fission Gas Swelling for Short Ramps**

Program	Rods	Burnup (GWd/MTU)	RTL (KW/ft)	Holding Time
---------	------	------------------	-------------	--------------

a,b,c

#### A.2.4.2 Ramp Diameter Change Calibration Results

Many PAD5 models contribute to the ramp strain, including fuel thermal expansion, fuel relocation, fission gas swelling, cladding thermal expansion, cladding plasticity and cladding creep. Therefore, ramp strain is an integral calibration confirmation of the mechanical models of PAD5. [

] <sup>a,c</sup>

The typical ramp tests have holding times from a few hours to days, which is sufficient time for the fission gas bubbles to develop and cause gaseous swelling. However, [

] <sup>a,c</sup>

It should be noted there are [

] <sup>a,c</sup> The PAD5 fission gas swelling model is discussed in Section 5.8.

The coefficients in Equation (5-40) of Section 5.8 are adjusted to best predict the clad diameter change during long ramps. Figures A.2.4-1 and A.2.4-2 show the calibration and validation results for the ramp test rods with long holding time. These results show good agreement between predicted and measured values with no apparent bias in predicted minus measured data versus burnup. There are expected differences in the behavior of gaseous swelling between low burnup and high burnup. At low burnup there is limited grain boundary gas accumulation. At high burnup there may be fission gas already accumulated at the grain boundary during steady state operation from the fission gas release processes and from microstructure changes. PAD5 is able to treat those effects properly.

a,b,c

**Figure A.2.4-1 PAD5 Predicted vs. Measured Diameter Change – Calibration and Validation Sets for Ramps with Long Hold Times**

**Figure A.2.4-2 PAD5 Predicted Minus Measured Diameter Change vs. Nodal Burnup – Calibration and Validation Sets for Ramps with Long Hold Times**

The coefficients in Equation (5-41) of Section 5.8 are adjusted to best predict the clad diameter change during short ramps. Figures A.2.4-3 and A.2.4-4 show the results for the rods with short holding times as listed in Table A.2.4-4. [

] <sup>a,c</sup>

a,b,c

**Figure A.2.4-3 Predicted vs. Measured Ramp Cladding Diameter Change for Long and Short Ramps**

**Figure A.2.4-4 PAD5 Predicted Minus Measured Ramp Cladding Diameter Change for Long and Short Ramps**

#### **A.2.4.3 Ramp Diameter Change Model Uncertainties**

As discussed in Section A.2.4.2, the PAD5 diameter change predictions are affected by many performance models. The uncertainties in these models contribute to the uncertainty in the permanent cladding deformation results. In addition, uncertainties in key fuel design parameters, particularly gap size, will also affect the permanent cladding deformation results. The Upper Bound (UB) uncertainty is an aggregate of the many model uncertainties that have contributed to the diameter changes during the ramp, i.e., fuel and clad thermal expansion, fuel solid swelling and densification, and cladding creep. [

] <sup>a,c</sup>

#### **A.2.5 Integral Calibration Confirmation**

Following the calibration of the fuel performance parameters and models described in the preceding sections, the integrated code performance can be evaluated further based on the ability of the code to predict certain other fuel rod performance parameters which are a reflection of the whole rod

performance. This integrated code performance assessment shows how well the code predicts end of life or end of cycle fuel rod diameters and end of life cold internal void volumes.

### A.2.5.1 Steady-State Cladding Outer Diameter

The cladding diameter change after fuel-cladding gap closure is impacted by the pellet deformation, i.e., thermal expansion, pellet relocation, fuel solid swelling, and fission gas swelling. The data considered for this evaluation has nodal burnups greater than [ ]<sup>a,c</sup>. Figures A.2.5-1 and A.2.5-2 show the diameter change data and predictions from some multi-cycle high burnup profilometry. For comparison purposes, the measured in-reactor profilometry data has been [

] <sup>a,c</sup>. Table A.2.5-1 shows the data used for the steady-state cladding outer diameter evaluation. Figure A.2.5-1 shows the predicted versus measured cladding outer diameter change. Figure A.2.5-2 shows the predicted minus measured cladding outer diameter change versus rod average burnup. There is [

] <sup>a,c</sup>

**Table A.2.5-1 Steady-state Cladding Outer Diameter Database (Nodal Burnup > [ ]<sup>a,c</sup>)**

Plants	Rod ID	Rod Average Burnup (MWd/MTU)	Cladding Type
--------	--------	---------------------------------	---------------

a,b,c

**Table A.2.5-1 Steady-state Cladding Outer Diameter Database (Nodal Burnup > [ ]<sup>a,c</sup>)**  
(cont.)

<b>Plants</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Cladding Type</b>
---------------	---------------	---	----------------------

a,b,c

**Table A.2.5-1 Steady-state Cladding Outer Diameter Database (Nodal Burnup > [ ]<sup>a,c</sup>)**  
(cont.)

<b>Plants</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Cladding Type</b>
---------------	---------------	---	----------------------

a,b,c



**Table A.2.5-1 Steady-state Cladding Outer Diameter Database (Nodal Burnup > [ ]<sup>a,c</sup>)**  
(cont.)

<b>Plants</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Cladding Type</b>
---------------	---------------	---	----------------------

a,b,c

**Table A.2.5-1 Steady-state Cladding Outer Diameter Database (Nodal Burnup > [ ]<sup>a,c</sup>)**  
**(cont.)**

<b>Plants</b>	<b>Rod ID</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Cladding Type</b>
---------------	---------------	---	----------------------

a,b,c

a,b,c

**Figure A.2.5-1 Predicted Versus Measured Steady-State Clad Diameter Change (Burnup > [ ]<sup>a,c</sup>)**

**Figure A.2.5-2 Predicted Minus Measured Steady-State Clad Diameter Change vs. Nodal Burnup (Burnup > [ ]<sup>a,c</sup>)**

#### **A.2.5.2 Void Volume**

Void volume data are used primarily to assess the integral performance of cladding creep, fuel relocation, rod growth, fuel densification, and the fuel swelling models. Table A.2.5-2 lists the void volume data used in this integral calibration. Void volumes from the refabricated ramp test rods are excluded. PAD5 results at room temperature (cold) are compared with the measured data in Figures A.2.5-3 and A.2.5-4. Figure A.2.5-3 shows the predicted versus measured total free volume at room temperature. Figure A.2.5-4 shows the predicted minus measured total free volume versus rod average burnup. These figures illustrate good agreement between the predicted and measured total free cold void volume.

**Table A.2.5-2 PAD5 Cold Void Volume Database**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	------------------	-------------------------------------	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c



**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

**Table A.2.5-2 PAD5 Cold Void Volume Database  
(cont.)**

<b>Plant/Program</b>	<b>Rod ID</b>	<b>Fuel Type</b>	<b>Rod Average Burnup (MWd/MTU)</b>	<b>Measured Cold Void Volume (in<sup>3</sup>)</b>	<b>Predicted Cold Void Volume (in<sup>3</sup>)</b>
----------------------	---------------	----------------------	---	---	--

a,b,c

a,b,c

**Figure A.2.5-3 Predicted vs. Measured Cold Void Volume**

**Figure A.2.5-4 Predicted Minus Measured Cold Void Volume vs. Rod Average Burnup**

### A.3 CONCLUSIONS

PAD5 calibration and validation is based on a significantly expanded database, particularly data for fuel rods with burnup higher than [ ]<sup>a,c</sup>. As shown by the calibrations presented in this appendix, PAD5 predicts the fuel temperature, fission gas release, ramp cladding diameter change, steady state cladding diameter, and cold internal void volume data well at high rod average burnups up to and above [ ]<sup>a,c</sup>, and exhibits no significant error trend over the range of operation (i.e. burnup, power, and temperatures). The PAD5 results presented in this Appendix, along with the results for individual models presented in Sections 3 through 6, in particular clad corrosion and rod growth, demonstrate that PAD5 models are qualified to model ZIRLO (and legacy Zircaloy-4) fuel rods with lead rod average burnup up to 62 GWd/MTU and **Optimized ZIRLO** clad fuel rods with lead rod average burnup up to [ ]<sup>a,c</sup>.

### A.4 APPENDIX A REFERENCES

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## APPENDIX B

### SAMPLE CALCULATION

#### B.1 INTRODUCTION

Appendix B provides sample PAD5 results for six fuel rods selected from three fuel assembly designs, as summarized in Table B.1-1. A standard  $\text{UO}_2$  fuel rod and an Integral Fuel Burnable Absorber (IFBA) fuel rod were modeled from each assembly type. The 15X15 fuel rods represent the largest cladding outer diameter (0.422" OD) Westinghouse fuel design, and the 17X17 fuel rods represent the smallest cladding outer diameter (0.360" OD) Westinghouse fuel design. The 15X15 and 17X17 IFBA fuel designs use  $\text{ZrB}_2$  coated  $\text{UO}_2$  IFBA. The CE 16X16 IFBA fuel rod (0.382" OD) uses  $\text{Gd}_2\text{O}_3$ - $\text{UO}_2$  IFBA. All of these rods use **Optimized ZIRLO** cladding material.

Current Westinghouse fuel operation is limited to lead rod average burnup up to 62 GWd/MTU (CE NSSS limited to 60 GWd/MTU), and these reference case results are intended to illustrate expected fuel behavior consistent with three-cycles of operation to an extended lead rod average burnup up to [ ]<sup>a,c</sup>. The power histories for these reference rods were developed based on current fuel rod designs but with adjustments to rod power and cycle lengths to produce the desired increase in rod burnup. Axial power shapes, defined by axial node, represent realistic plant operation, but the shape assignments to specific steps were simplified to reduce the total number of axial shapes to less than 20 (consistent with FRAPCON 3.4 input limitations, Reference B.1). The resulting shape assignments produced more limiting end of cycle local power changes in the 17X17 case results. Additional details on the reference case designs are provided in Section B.2.

PAD5 models the rod power history as a series of discrete time steps, with the rod average power and axial power shape held constant during the time step. For example, if the first and second time steps are (in hours) 0.01 and 150.0, respectively, the first power level and axial shape are applied for the time interval from 0.0 to 0.01 hours, and the second power level and axial shape are applied for the time interval from 0.01 to 150.0 hours. This is indicated in the figures that illustrate the reference rod power histories in Section B.2 as a step function.

Typical PAD5 output summarizes the predicted results at the end of each time step. The reference case predicted performance parameters are plotted as a function of burnup as line segments connecting these end of step results.



Table B.1-1 PAD5 Reference Designs					a,c
Reference Case	Fuel Assembly Type	Rod Type	Rod Average Burnup (MWd/MTU)	Cladding Outer Diameter (inches)	
1					
2					
3					
4					
5					
6					

## B.2 KEY FUEL ROD PERFORMANCE PARAMETERS

This section provides additional fuel rod design data and selected PAD5 results for each of the six reference fuel rods described in Table B.1-1. Subsections B.2.1, B.2.2, and B.2.3 summarize the 15X15, 17X17, and CE 16X16 fuel rod designs and best estimate PAD5 predictions.

The following additional fuel rod design information is provided for each reference design:

- Detailed fuel rod design parameters, including fabrication tolerances
- Time-dependent rod average power history
- Time-dependent axial power shapes
- Core operating conditions
- Rod average fast fluence

For each fuel rod, the following PAD5 results are provided on a best-estimate basis:

- Hot void volume versus rod average burnup
- Rod internal pressure versus rod average burnup
- Fission gas release versus rod average burnup
- Pellet-clad gap versus local burnup at axial node 15
- Gap conductance versus local burnup at axial node 15
- Fuel surface temperature versus local burnup at axial node 15
- Fuel average temperature versus local burnup at axial node 15
- Fuel centerline temperature versus local burnup at axial node 15
- Pellet diameter versus local burnup at axial node 15
- Clad OD versus local burnup at axial node 15

## B.2.1 15X15 (0.422" OD) Reference Cases

The following tables and figures summarize the fuel rod design input parameters for 15X15 (0.422" OD) reference fuel rods.

- Table B.2.1-1: 15X15 Reference Case Fuel Rod Parameters
- Table B.2.1-2: 15X15 Reference Case Axial Power Shapes
- Table B.2.1-3: 15X15 Reference Case Rod Power History
- Table B.2.1-4: 15X15 Reference Case Core Operating Conditions
- Figure B.2.1-1: 15X15 Reference Case Rod Average Fast Fluence
- Figure B.2.1-2: 15X15 Reference Case Rod Power History

The following figures illustrate the PAD5 predictions for selected fuel rod parameters. The IFBA fuel rod and the UO<sub>2</sub> fuel rod predictions are provided in the same figures for comparison.

- Figure B.2.1-3: 15X15 Hot Void Volume Versus Rod Average Burnup
- Figure B.2.1-4: 15X15 Rod Internal Pressure Versus Rod Average Burnup
- Figure B.2.1-5: 15X15 Fission Gas Release Versus Rod Average Burnup
- Figure B.2.1-6: 15X15 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)
- Figure B.2.1-7: 15X15 Gap Conductance Versus Local Burnup (Axial Node 15)
- Figure B.2.1-8: 15X15 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.1-9: 15X15 Fuel Average Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.1-10: 15X15 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.1-11: 15X15 Pellet Diameter Versus Local Burnup (Axial Node 15)
- Figure B.2.1-12: 15X15 Clad OD Versus Local Burnup (Axial Node 15)

Table B.2.1-1 15X15 Reference Case Fuel Rod Parameters					a,c
Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	
Clad Material					
Fuel Stack Length (ft)					
Plenum Length (in)					
Pellet Diameter (in)					
Clad Inner Diameter (in)					
Clad Outer Diameter (in)					
Pellet Length (in)					
Chamfer Depth (in)					
Chamfer Width (in)					
Pellet Dish Spherical Radius (in)					
Pellet Dish Depth (in)					
Pellet Sintering Temperature (°F)					

**Table B.2.1-1 15X15 Reference Case Fuel Rod Parameters  
(cont.)**

Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	a,c
Axial Blanket Design					
Axial Blanket Length (in)					
Annulus Inner Diameter (in)					
Annulus Volume (in <sup>3</sup> ) <sup>(1)</sup>					
Plenum with Annulus Volume (in) <sup>(2)</sup>					
Initial Cold Void Volume (in <sup>3</sup> )					
Plenum Spring Diameter (in)					
Plenum Spring Wire Diameter (in)					
Plenum Spring Turns					
Pellet True Density (% Theoretical Density)					
Fuel U-235 Enrichment (w/o)					
Helium Backfill Pressure (psig)					
Rod Pitch (in)					
ZrB <sub>2</sub> IFBA Coating Length (in)					
ZrB <sub>2</sub> IFBA B10 Loading (mg/in)					
ZrB <sub>2</sub> IFBA Coating Thickness (mils)					
Gadolinia Enrichment (w/o)					
Notes:					J <sup>a,c</sup>
1. [					
2. [					J <sup>a,c</sup>

**Table B.2.1-2 15X15 Reference Case Axial Power Shapes**

Axial Segment	Power Shape					
	1	2	3	4	5	6
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						
Axial Segment	Power Shape					
	7	8	9	10	11	
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						

**Table B.2.1-2 15X15 Reference Case Axial Power Shapes  
(cont.)**

Axial Segment	Power Shape					a,c
	7	8	9	10	11	
11						
12						
13						
14						
15						
16						
17						
18						

**Table B.2.1-3 15X15 Reference Case Rod Power History**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					
17					

**Table B.2.1-3 15X15 Reference Case Rod Power History (cont.)**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
18					
19					
20					
21					
22					
23					
24					
25					
26					
27					
28					
29					
30					
31					

**Table B.2.1-4 PAD5 FGR Model Results Statistics 15X15 Reference Case Core Operating Conditions**

Core Operating Condition	Value	a,c
Core Average Linear Power (kW/ft)		
Primary Coolant System Pressure (psi)		
Core Mass Flow Rate (1E+6 lbm/hr-ft <sup>2</sup> )		
Core Inlet Temperature (°F)		

a,c

**Figure B.2.1-1 15X15 Reference Case Rod Average Fast Fluence**

a,c

**Figure B.2.1-2 15X15 Reference Case Rod Power History**



a,c

**Figure B.2.1-3 15X15 Hot Void Volume Versus Rod Average Burnup**

a,c

**Figure B.2.1-4 15X15 Rod Internal Pressure Versus Rod Average Burnup**

a,c

**Figure B.2.1-5 15X15 Fission Gas Release Versus Rod Average Burnup**

a,c

**Figure B.2.1-6 15X15 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)<sup>1</sup>**

1. [

]a,c

a,c

**Figure B.2.1-7 15X15 Gap Conductance Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.1-8 15X15 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.1-9 15X15 Fuel Average Temperature Versus Local Burnup (Axial Node 15)**

a,c

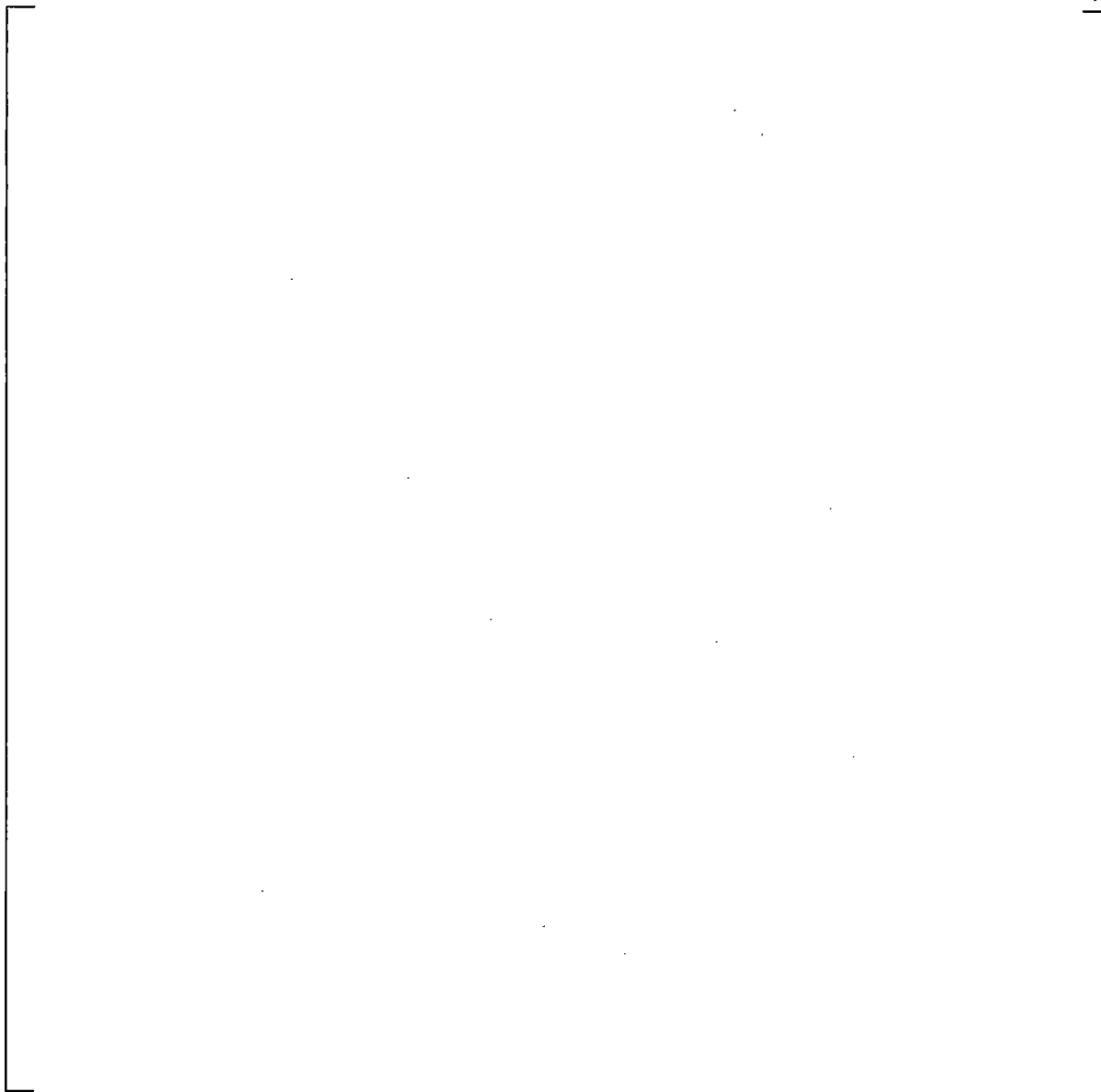
**Figure B.2.1-10 15X15 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)**





**Figure B.2.1-11 15X15 Pellet Diameter Versus Local Burnup (Axial Node 15)**

a,c



**Figure B.2.1-12 15X15 Clad OD versus Local Burnup (Axial Node 15)**

### **B.2.2 17X17 (0.360" OD) Reference Cases**

The following tables and figures summarize the fuel rod design input parameters for 17X17 (0.360" OD) reference fuel rods.

- Table B.2.2-1: 17X17 Reference Case Fuel Rod Parameters
- Table B.2.2-2: 17X17 Reference Case Axial Power Shapes
- Table B.2.2-3: 17X17 Reference Case Rod Power History
- Table B.2.2-4: 17X17 Reference Case Core Operating Conditions
- Figure B.2.2-1: 17X17 Reference Case Rod Average Fast Fluence
- Figure B.2.2-2: 17X17 Reference Case Rod Power History

The following figures illustrate the PAD5 predictions for selected fuel rod parameters. The IFBA fuel rod and the UO<sub>2</sub> fuel rod predictions are provided in the same figures for comparison.

- Figure B.2.2-3: 17X17 Hot Void Volume Versus Rod Average Burnup
- Figure B.2.2-4: 17X17 Rod Internal Pressure Versus Rod Average Burnup
- Figure B.2.2-5: 17X17 Fission Gas Release Versus Rod Average Burnup
- Figure B.2.2-6: 17X17 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)
- Figure B.2.2-7: 17X17 Gap Conductance Versus Local Burnup (Axial Node 15)
- Figure B.2.2-8: 17X17 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.2-9: 17X17 Fuel Average Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.2-10: 17X17 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.2-11: 17X17 Pellet Diameter Versus Local Burnup (Axial Node 15)
- Figure B.2.2-12: 17X17 Clad OD Versus Local Burnup (Axial Node 15)

**Table B.2.2-1 17X17 Reference Case Fuel Rod Parameters**

Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	a,c
Clad Material					
Fuel Stack Length (ft)					
Plenum Length (in)					
Pellet Diameter (in)					
Clad Inner Diameter (in)					
Clad Outer Diameter (in)					
Pellet Length (in)					
Chamfer Depth (in)					
Chamfer Width (in)					
Pellet Dish Spherical Radius (in)					
Pellet Dish Depth (in)					
Pellet Sintering Temperature (°F)					
Axial Blanket Design					
Axial Blanket Length (in)					
Annulus Inner Diameter (in)					
Annulus Volume (in <sup>3</sup> ) <sup>(1)</sup>					
Plenum with Annulus Volume (in <sup>3</sup> ) <sup>(2)</sup>					
Initial Cold Void Volume (in <sup>3</sup> )					

**Table B.2.2-1 17X17 Reference Case Fuel Rod Parameters (cont.)**

Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	a,c
Plenum Spring Diameter (in)					
Plenum Spring Wire Diameter (in)				-	
Plenum Spring Turns					
Pellet True Density (% Theoretical Density)					
Fuel U-235 Enrichment (w/o)					
Helium Backfill Pressure (psig)					
Rod Pitch (in)					
ZrB <sub>2</sub> IFBA Coating Length (in)					
ZrB <sub>2</sub> IFBA B10 Loading (mg/in)					
ZrB <sub>2</sub> IFBA Coating Thickness (mils)					
Gadolinia Enrichment (w/o)					
1. [ <span style="float: right;">] <sup>a,c</sup></span> 2. [ <span style="float: right;">] <sup>a,c</sup></span>					

**Table B.2.2-2 17X17 Reference Case Axial Power Shapes**

Axial Segment	Power Shape					a,c
	1	2	3	4	5	
1						
2						
3						
4						
5						
6						
7						
8						

**Table B.2.2-2 17X17 Reference Case Axial Power Shapes  
(cont.)**

Axial Segment	Power Shape					a,c
	1	2	3	4	5	
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						
Axial Segment	Power Shape					a,c
	6	7	8	9	10	
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						

**Table B.2.2-2 17X17 Reference Case Axial Power Shapes  
(cont.)**

Axial Segment	Power Shape				a,c
	11	12	13	14	
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					
17					
18					
Axial Segment	Power Shape				a,c
	15	16	17	18	
1					
2					
3					
4					
5					
6					
7					
8					
9					

**Table B.2.2-2 17X17 Reference Case Axial Power Shapes  
(cont.)**

Axial Segment	Power Shape				a,c
	15	16	17	18	
10					
11					
12					
13					
14					
15					
16					
17					
18					

**Table B.2.2-3 17X17 Reference Case Rod Power History**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					

**Table B.2.2-3 17X17 Reference Case Rod Power History  
(cont.)**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
17					
18					
19					
20					
21					
22					
23					
24					
25					
26					
27					
28					
29					
30					
31					
32					
33					
34					
35					
36					
37					
38					
39					
40					
41					
42					
43					
44					
45					
46					



**Table B.2.2-3 17X17 Reference Case Rod Power History (cont.)**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
47					
48					
49					
50					
51					
52					
53					
54					
55					

**Table B.2.2-4 17X17 Reference Case Core Operating Conditions**

Core Operating Condition	Value	a,c
Core Average Linear Power (kW/ft)		
Primary Coolant System Pressure (psi)		
Core Mass Flow Rate (1E+6 lbm/hr-ft <sup>2</sup> )		
Core Inlet Temperature (°F)		

a,c

**Figure B.2.2-1 17X17 Reference Case Rod Average Fast Fluence**

a,c

**Figure B.2.2-2 17X17 Reference Case Rod Power History**

a,c

**Figure B.2.2-3 17X17 Hot Void Volume Versus Rod Average Burnup**

a,c

**Figure B.2.2-4 17X17 Rod Internal Pressure Versus Rod Average Burnup**

a,c

**Figure B.2.2-5 17X17 Fission Gas Release Versus Rod Average Burnup**

a,c

**Figure B.2.2-6 17X17 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)<sup>2</sup>**

<sup>2</sup> [

] <sup>a,c</sup>

a,c

**Figure B.2.2-7 17X17 Gap Conductance Versus Local Burnup (Axial Node 15)**



a,c

**Figure B.2.2-8 17X17 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.2-9 17X17 Fuel Average Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.2-10 17X17 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.2-11 17X17 Pellet Diameter Versus Local Burnup (Axial Node 15)**

**Figure B.2.2-12 17X17 Clad OD Versus Local Burnup (Axial Node 15)**

### **B.2.3 CE 16X16 (0.382" OD) Reference Cases**

The following tables and figures summarize the fuel rod design input parameters for CE 16X16 (0.382" OD) reference fuel rods.

- Table B.2.3-1: CE 16X16 Reference Case Fuel Rod Parameters
- Table B.2.3-2: CE 16X16 Reference Case Axial Power Shapes (Gadolinia Fuel)
- Table B.2.3-3: CE 16X16 Reference Case Axial Power Shapes (UO<sub>2</sub> Fuel)
- Table B.2.3-4: CE 16X16 Reference Case Rod Power History (Gadolinia Fuel)
- Table B.2.3-5: CE 16X16 Reference Case Rod Power History (UO<sub>2</sub> Fuel)
- Table B.2.3-6: CE 16X16 Reference Case Core Operating Conditions
- Figure B.2.3-1: CE 16X16 Reference Case Rod Average Fast Fluence
- Figure B.2.3-2: CE 16X16 Reference Case Rod Power History

The following figures illustrate the PAD5 predictions for selected fuel rod parameters. The IFBA fuel rod and the UO<sub>2</sub> fuel rod predictions are provided in the same figures for comparison.

- Figure B.2.3-3: CE 16X16 Hot Void Volume Versus Rod Average Burnup
- Figure B.2.3-4: CE 16X16 Rod Internal Pressure Versus Rod Average Burnup
- Figure B.2.3-5: CE 16X16 Fission Gas Release Versus Rod Average Burnup
- Figure B.2.3-6: CE 16X16 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)
- Figure B.2.3-7: CE 16X16 Gap Conductance Versus Local Burnup (Axial Node 15)
- Figure B.2.3-8: CE 16X16 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.3-9: CE 16X16 Fuel Average Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.3-10: CE 16X16 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)
- Figure B.2.3-11: CE 16X16 Pellet Diameter Versus Local Burnup (Axial Node 15)
- Figure B.2.3-12: CE 16X16 Clad OD Versus Local Burnup (Axial Node 15)

Table B.2.3-1 CE 16X16 Reference Case Fuel Rod Parameters					a,c
Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	
Clad Material					
Fuel Stack Length (ft)					
Plenum Length (in)					
Pellet Diameter (in)					
Clad Inner Diameter (in)					
Clad Outer Diameter (in)					
Pellet Length (in)					
Chamfer Depth (in)					
Chamfer Width (in)					
Pellet Dish Spherical Radius (in)					
Pellet Dish Depth (in)					
Pellet Sintering Temperature (°F)					
Axial Blanket Design					
Axial Blanket Length (in)					
Annulus Inner Diameter (in)					
Annulus Volume (in <sup>3</sup> )					
Plenum with Annulus Volume (in)					

**Table B.2.3-1 CE 16X16 Reference Case Fuel Rod Parameters (cont.)**

Description	Input	Upper Bound Uncertainty	Lower Bound Uncertainty	Uncertainty Distribution	a,c
Initial Cold Void Volume (in <sup>3</sup> )					
Plenum Spring Diameter (in)					
Plenum Spring Wire Diameter (in)					
Plenum Spring Turns					
Pellet True Density (% Theoretical Density)					
Fuel U-235 Enrichment (w/o)					
Helium Backfill Pressure (psig)					
Rod Pitch (in)					
ZrB <sub>2</sub> IFBA Coating Length (in)					
ZrB <sub>2</sub> IFBA B10 Loading (mg/in)					
ZrB <sub>2</sub> IFBA Coating Thickness (mils)					
Gadolinia Enrichment (w/o)					

**Table B.2.3-2 CE 16X16 Reference Case Axial Power Shapes (Gadolinia Fuel)**

Axial Segment	Power Shape					a,c
	1	2	3	4	5	
1						
2						
3						
4						
5						
6						
7						
8						
9						

**Table B.2.3-2 CE 16X16 Reference Case Axial Power Shapes (Gadolinia Fuel)**  
(cont.)

Axial Segment	Power Shape					a,c
	1	2	3	4	5	
10						
11						
12						
13						
14						
15						
16						
17						
18						
Axial Segment	Power Shape					a,c
	6	7	8	9	10	
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						



**Table B.2.3-2 CE 16X16 Reference Case Axial Power Shapes (Gadolinia Fuel)**  
(cont.)

Axial Segment	Power Shape				a,c
	11	12	13	14	
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					
17					
18					
Axial Segment	Power Shape				a,c
	15	16	17	18	
1					
2					
3					
4					
5					
6					
7					
8					
9					

**Table B.2.3-2 CE 16X16 Reference Case Axial Power Shapes (Gadolinia Fuel)**  
(cont.)

Axial Segment	Power Shape				a,c
	15	16	17	18	
10					
11					
12					
13					
14					
15					
16					
17					
18					

**Table B.2.3-3 CE 16X16 Reference Case Axial Power Shapes (UO<sub>2</sub> Fuel)**

Axial Segment	Power Shape					a,c
	1	2	3	4	5	
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						

**Table B.2.3-3 CE 16X16 Reference Case Axial Power Shapes (UO<sub>2</sub> Fuel)**  
(cont.)

Axial Segment	Power Shape					
	1	2	3	4	5	
16						a,c
17						
18						
Axial Segment	Power Shape					
	6	7	8	9	10	
1						a,c
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						

**Table B.2.3-3 CE 16X16 Reference Case Axial Power Shapes (UO<sub>2</sub> Fuel)**  
(cont.)

	Power Shape					a,c
Axial Segment	11	12	13	14	15	
1						
2						
3						
4						
5						
6						
7						
8						
9						
10						
11						
12						
13						
14						
15						
16						
17						
18						
	Power Shape					a,c
Axial Segment	16	17	18	19		
1						
2						
3						
4						
5						
6						
7						
8						
9						

**Table B.2.3-3 CE 16X16 Reference Case Axial Power Shapes (UO<sub>2</sub> Fuel)**  
(cont.)

Axial Segment	Power Shape				a,c
	16	17	18	19	
10					
11					
12					
13					
14					
15					
16					
17					
18					

**Table B.2.3-4 CE 16X16 Reference Case Rod Power History (Gadolinia Fuel)**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					

**Table B.2.3-4 CE 16X16 Reference Case Rod Power History (Gadolinia Fuel)**  
(cont.)

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
17					
18					
19					
20					
21					
22					
23					
24					
25					
26					
27					
28					
29					
30					
31					
32					
33					
34					
35					
36					
37					
38					
39					
40					
41					
42					
43					
44					
45					
46					

**Table B.2.3-4 CE 16X16 Reference Case Rod Power History (Gadolinia Fuel)**  
(cont.)

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
47					
48					
49					
50					
51					
52					

**Table B.2.3-5 CE 16X16 Reference Case Rod Power History (UO<sub>2</sub> Fuel)**

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
14					
15					
16					
17					
18					
19					
20					

**Table B.2.3-5 CE 16X16 Reference Case Rod Power History (UO<sub>2</sub> Fuel)**  
(cont.)

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
21					
22					
23					
24					
25					
26					
27					
28					
29					
30					
31					
32					
33					
34					
35					
36					
37					
38					
39					
40					
41					
42					
43					
44					
45					
46					
47					
48					
49					
50					



**Table B.2.3-5 CE 16X16 Reference Case Rod Power History (UO<sub>2</sub> Fuel)**  
(cont.)

Time Step	Time (Hours)	Rod Average Burnup (MWd/MTU)	Rod Average Power (kW/ft)	Axial Power Shape	a,c
51					
52					
53					
54					
55					
56					
57					

**Table B.2.3-6 CE 16X16 Reference Case Core Operating Conditions**

Core Operating Condition	Value	a,c
Core Average Linear Power (kW/ft)		
Primary Coolant System Pressure (psi)		
Core Mass Flow Rate (1E+6 lbm/hr-ft <sup>2</sup> )		
Core Inlet Temperature (°F)		

a,c

**Figure B.2.3-1 CE 16X16 Reference Case Rod Average Fast Fluence**

a,c

**Figure B.2.3-2 CE 16X16 Reference Case Rod Power History**

a,c

**Figure B.2.3-3 CE 16X16 Hot Void Volume Versus Rod Average Burnup**

a,c

**Figure B.2.3-4 CE 16X16 Rod Internal Pressure Versus Rod Average Burnup**

a,c

**Figure B.2.3-5 CE 16X16 Fission Gas Release Versus Rod Average Burnup**

a,c

**Figure B.2.3-6 CE 16X16 Pellet-Clad Gap Versus Local Burnup (Axial Node 15)<sup>3</sup>**

3. [

] <sup>a,c</sup>

a,c

**Figure B.2.3-7 CE 16X16 Gap Conductance Versus Local Burnup (Axial Node 15)**



a,c

**Figure B.2.3-8 CE 16X16 Fuel Surface Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.3-9 CE 16X16 Fuel Average Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.3-10 CE 16X16 Fuel Centerline Temperature Versus Local Burnup (Axial Node 15)**

a,c

**Figure B.2.3-11 CE 16X16 Pellet Diameter Versus Local Burnup (Axial Node 15)**



**Figure B.2.3-12 CE 16X16 Clad OD Versus Local Burnup (Axial Node 15)**

### **B.3 APPENDIX B REFERENCES**

- B.1 K. J. Geelhood, W. G. Luscher and C. E. Beyer, "FRAPCON-3.4: A Computer Code for the Calculation of Steady-State Thermal-Mechanical Behavior of Oxide Fuel Rods for High Burnup," NUREG/CR-7022, Vol. 1 (PNNL-19418, Vol. 1), March 2011.