

1.0 OBJECTIVE

Perform the thermal analyses of the EnergySolutions 8-120B Cask under hypothetical fire accident conditions, using a 3-dimensional finite element model.

2.0 REFERENCES

1. EnergySolutions Drawing No. C-110-E-0007, Rev. 14, 8-120B Shipping Cask.
2. Code of Federal Regulations, Title 10, Part 71, Packaging and Transportation of Radioactive Material, January 2003.
3. U.S. NRC Regulatory Guide 7.8, Revision 1, March 1989, Load Combinations for the Structural Analysis of Shipping Casks for Radioactive Material.
4. Heat Transfer, J.P. Holman, McGraw Hill Book Company, New York, Fifth Edition, 1981.
5. Cask Designers Guide, L.B. Shappert, et. al, Oak Ridge National Laboratory, February 1970, ORNL-NSIC-68.
6. CRC Handbook of Chemistry and Physics, Robert C. Weast and Melvin J. Astel, eds., CRC Press, Inc., Boca Raton, Florida, 62nd ed., 1981.
7. ASME Boiler & Pressure Vessel Code, 2001, Section II, Part D, Materials, The American Society of Mechanical Engineers, New York, NY, 2001.
8. Rohsenow and Hartnett, Handbook of Heat Transfer, McGraw Hill Publication, 1973.
9. ANSYS, Release 13.0, ANSYS Inc., Canonsburg, PA, 2010.
10. IAEA Safety Series No.37, Advisory Material for the IAEA Regulations for the Safe Transport of Radioactive Material - 1985 Edition, International Atomic Energy Agency, Vienna, 1990.
11. Kreith, Frank, *Principles of Heat Transfer*, 3rd Edition, Harper & Row, 1973.
12. Energy Solutions Document No. TH-027, Rev. 1, Steady State Thermal Analyses of the 8-120B Cask Using Finite Element Models.

3.0 INTRODUCTION

EnergySolutions 8-120B Cask (Reference 1) is designed as a Type B radioactive-material shipping package. To be certified by the U.S.N.R.C., the cask needs to meet the requirements of 10 CFR 71 (Reference 2) and follow the guidelines of U.S.N.R.C. Regulatory Guide 7.8 (Ref. 3).

This document presents the thermal load analysis of the 8-120B Cask for the hypothetical accident conditions (HAC) fire test. The normal conditions of transport (NCT) analyses have been performed in a separate document. The analyses in this document are performed using the finite element modeling techniques. Models of the cask that includes all its major components have been employed in the analyses. Temperature dependent material properties of the major components of the cask are used in the analyses.

The effect of HAC fire test on the waste contents of the 8-120B cask has been included in this document. It has also been shown that after the puncture drop test if the fire-shield loses the air gap separation, the HAC fire test will not cause a melting of the lead at this location.

The results of the analyses for various load cases are presented pictorially in temperature contour plots as well as digital data format, suitable for use with a structural finite element model to obtain the thermal stresses.

4.0 MATERIAL PROPERTIES

Temperature-Independent Metal Thermal Properties

Material	Property	Reference: Page	Value
Steel	Density	4: 536	0.2824 lb/in ³
	ε (Outside)	2: 648	0.8
	ε (Inside)	5:133	0.15
Lead	Density	4: 535	0.4109 lb/in ³
	Spec. Heat	4: 535	0.0311 Btu/lb-°F
	Melting Point	6: B-29	621.5 °F

Temperature-Dependent Metal Thermal Properties

Temp. (°F)	Stainless Steel (Ref. 7)		Carbon Steel (Ref.7)		Lead (Ref.8)
	Sp. Heat	Conductivity	Sp. Heat	Conductivity	Conductivity
	Btu/lb-°F	×10 ⁻³ Btu/sec-in-°F	Btu/lb-°F	×10 ⁻³ Btu/sec-in-°F	×10 ⁻³ Btu/sec-in-°F
70	0.117	0.199	0.104	0.813	0.465
100	0.117	0.201	0.106	0.803	0.461
150	0.120	0.208	0.109	0.789	0.455
200	0.122	0.215	0.113	0.778	0.448
250	0.125	0.222	0.115	0.762	0.441
300	0.126	0.227	0.118	0.748	0.435
350	0.128	0.234	0.122	0.731	0.428
400	0.129	0.241	0.124	0.715	0.422
450	0.130	0.245	0.126	0.701	0.415
500	0.131	0.252	0.128	0.677	0.409
550	0.132	0.257	0.131	0.667	0.402
600	0.133	0.262	0.133	0.648	0.395
650	0.134	0.269	0.135	0.632	0.389
700	0.135	0.273	0.139	0.616	0.389
750	0.136	0.278	0.142	0.600	0.389
800	0.136	0.282	0.146	0.583	0.389
900	0.138	0.294	0.154	0.551	0.389
1,000	0.139	0.306	0.163	0.519	0.389
1,100	0.141	0.315	0.172	0.484	0.389
1,200	0.141	0.324	0.184	0.451	0.389

Temperature-Dependent Air Thermal Properties

Temp. (°F)	Air (Ref.4)		
	Density $\times 10^{-5}$ lb/in ³	Sp. Heat Btu/lb-°F	Conductivity $\times 10^{-7}$ Btu/sec-in-°F
70	4.3507	0.2402	3.4491
100	4.1117	0.2404	3.5787
150	3.7517	0.2408	3.9028
200	3.4676	0.2414	4.1759
250	3.2361	0.2421	4.4468
300	3.0307	0.2429	4.7037
350	2.8310	0.2438	4.9560
400	2.6730	0.2450	5.2037
450	2.5220	0.2461	5.4491
500	2.3964	0.2474	5.6875
550	2.2778	0.2490	5.9213
600	2.1684	0.2511	6.1435
650	2.0706	0.2527	6.3634
700	1.9803	0.2538	6.5810
750	1.8981	0.2552	6.7894
800	1.8177	0.2568	6.9954
900	1.6898	0.2596	7.4097
1,000	1.5712	0.2628	7.8032
1,100	1.4722	0.2659	8.1759
1,200	1.3848	0.2689	8.5440
1,300	1.3044	0.2717	8.8981
1,400	1.2350	0.2742	9.2847
1,500	1.1707	0.2766	9.7060

5.0 MODEL DESCRIPTION

The thermal analyses of the 8-120B Cask under hypothetical fire accident test conditions have been performed using finite element modeling techniques. ANSYS finite element analysis code (Ref. 9) has been employed to perform the analyses.

Two finite element models have been employed in performing the HAC thermal analyses. A three-dimensional and a 2-d axisymmetric model were used in the analyses. For the structural analyses of the loading conditions where the bolt loadings are non-uniform a three dimensional finite element model, in which the primary and secondary lid bolts can be appropriately modeled, is needed. To obtain the temperature distribution for these loading conditions a three-dimensional thermal model is used. To obtain the temperature distribution in the cask where the bolt loadings have no effect on the results, a two-dimensional axisymmetric finite element model has been used.

3-Dimensional Finite Element Model

The cask is symmetrical about a vertical plane. Therefore, a one-half model of the cask is employed in the three-dimensional model. Figure 1 shows the finite element model used in various thermal load analyses. Figures 2 and 3 are detailed views of various regions of the finite element model. Figure 4 shows the material property model numbers of various components of the cask.

The finite element model is made of 3-dimensional thermal solid elements (ANSYS SOLID70) that represent the major components of the cask, the cask body, the lid, the bolts, and the fire shield. The interstitial air gaps between the impact limiter plates and the cask end as well as that between the fire-shield and the cask body are modeled by 3-dimensional thermal solid elements.

The poured lead in the body is not bonded to the steel. It is free to slide over the steel surface. Therefore, the interface between the lead and the steel is modeled by pairs of 3-d 8 node thermal contact element (CONTA174) and 3-d target segment (TARGE170) elements. These elements allow the lead to slide over the steel at the same time prevent it from penetrating the steel surface. The interface between the two plates of the lid at the weld location is also modeled by the contact-target pairs. The transition from a coarser mesh to a finer mesh, as well as bondage between various parts of the model, is also achieved by using these elements. Figure 5 shows the contact-target pair elements used in the finite element model.

The heat transfer by radiation between the fire-shield and the ambient air is modeled by 3-d thermal surface (ANSYS SURF152). The radiation between the outer shell and the fire-shield is modeled by superelements (ANSYS MATRIX50). These elements are formed by modeling the radiating surfaces with thermal shells (ANSYS SHELL57) and specifying the appropriate emissivity of the surfaces and the Stefan-Boltzmann constant. The heat transfer by natural convection between the fire-shield and the ambient air is simulated by 3-d thermal surface (ANSYS SURF152). The outer surfaces of the impact limiter plates, which are covered by foam, are considered to be totally insulated.

For the HAC fire the foam of the impact limiters is assumed to provide complete thermal insulation. Therefore, only the area of the model not covered by the impact limiter is subjected to the fire in the analysis.

The real constants used in the ANSYS finite element model, to define the characteristic of various finite elements to simulate the heat transfer by convection and radiation, are based on the derivation in the following section. The heat flux representing the internal heat load and the solar insolation is also presented in the following section.

2-Dimensional Axisymmetric Finite Element Model

The 2-d axisymmetric FEM uses the same modeling techniques as described under the 3-dimensional model, except that it uses the corresponding 2-dimensional elements. Figure 6 shows the 2-dimensional axisymmetric model.

A print-out of the model data is included in Appendix 1.

5.1 Natural Convection Modeling

The convective heat transfer per unit area between the cask and the atmosphere, q , is governed by the equation:

$$q = hA (T_s - T_a)$$

where:

h = Heat transfer coefficient (Btu/hr-ft²-°F)

A = Area (sq ft)

T_s = cask surface temperature (°F)

T_a = ambient temperature (°F)

The heat transfer coefficient for the natural convection is given by the following relationship (see for example Ref. 5, page 135).

$$h = C (T_s - T_a)^{1/3}$$

where, for horizontal cask,

$$\begin{aligned} C &= 0.18 \text{ (Btu/hr-ft}^2\text{-°F}^{4/3}\text{)} \\ &= 3.4722 \times 10^{-7} \text{ (Btu/sec-in}^2\text{-°F}^{4/3}\text{)} \end{aligned}$$

5.2 Forced Convection Modeling

The heat transfer coefficient for the forced convection during the fire is based on the explanatory material for the IAEA regulations in Safety Series No.37 (Ref. 10), the pool fire gas velocity is taken to be 10 m/sec (32.8 ft/sec). The forced convection heat transfer coefficient for large casks, according to Ref. 10, is:

$$h = 10 \frac{W}{m^2 \cdot ^\circ C}$$

For conversion, using,

$$1 \text{ W} = 9.4804 \times 10^{-4} \text{ Btu/sec}$$

$$1 \text{ m} = 39.37 \text{ inch}$$

$$1^\circ C = 1.8^\circ F$$

Therefore,

$$h = \frac{10 \times 9.4804 \times 10^{-4}}{39.37^2 \times 1.8} = 3.398 \times 10^{-6} \frac{\text{Btu}}{\text{sec-in}^2\text{-}^\circ F}$$

The heat transfer coefficients for forced convection are also obtained from additional sources and compared with the IAEA recommended value. It is shown that the value calculated above is larger than the values calculated from other sources. Therefore, the IAEA recommended

value will result in a conservative prediction of the cask temperatures during the HAC fire test.

A comprehensive heat transfer coefficient formula for a horizontal cylinder, which is applicable over a large range of Reynolds number values, is listed in Reference 4. This formula is referred to as Churchill-Bernstein formula, which relates Reynolds and Prandtl number with Nusselt number as follows:

$$Nu_d = 0.3 + \frac{0.62 Re^{1/2} Pr^{1/3}}{\left[1 + \left(\frac{0.4}{Pr}\right)^{1/4}\right]^{3/4}} \left[1 + \left(\frac{Re}{282,000}\right)^{5/8}\right]^{4/5}$$

.....Equation (1)

Where, $Re = \text{Reynolds Number} = \frac{\rho_f u_\infty d}{\mu_f}$

ρ_f = Air density at film temperature (1475°F use 1500°F)

$$= 1.1707 \times 10^{-5} \text{ lb/in}^3 \quad (\text{see Table in Section 4.0})$$

u_∞ = Air velocity = 32.8 ft/sec = 394 in/sec

d = diameter of the cylinder = 73.2 in

μ_f = Air viscosity at film temperature

$$= 91.75 \times 10^{-8} \text{ lb}_f\text{-sec/ft}^2 \quad (\text{Reference Mark's Handbook})$$

$$= 246.2 \times 10^{-8} \text{ lb/(in-sec)}$$

Therefore,

$$Re = \frac{\left(1.1707 \times 10^{-5} \frac{\text{lb}}{\text{in}^3}\right) \times \left(394 \frac{\text{in}}{\text{sec}}\right) \times (73.2 \text{ in})}{246.2 \times 10^{-8} \frac{\text{lb}}{\text{in} \cdot \text{sec}}} = 137,140$$

Also, $Pr = \text{Prandtl Number} = \frac{\mu_f C_{pf}}{k_f}$

C_{pf} = Specific heat of the air at constant pressure at film temperature

$$= 0.2766 \text{ Btu/lb-}^\circ\text{F} \quad (\text{see Table in Section 4.0})$$

k_f = Conductivity of air at film temperature

$$= 9.706 \times 10^{-7} \text{ Btu/sec-in-}^\circ\text{F}$$

Therefore,

$$Pr = \frac{\left(246.2 \times 10^{-8} \frac{lb}{in \cdot sec}\right) \times \left(0.2766 \frac{Btu}{lb \cdot deg F}\right)}{9.706 \times 10^{-7} \frac{Btu}{sec \cdot in \cdot deg F}} = 0.7$$

Substitution of these values in Equation 1 results in:

$$Nu_d = 0.3 + \frac{0.62 \times (137,140)^{1/2} \times (0.7)^{1/3} \left[1 + \left(\frac{137,140}{282,000}\right)^{5/8}\right]^{4/5}}{\left[1 + \left(\frac{0.4}{0.7}\right)^{2/3}\right]^{3/4}} = 0.3 + \left(\frac{203.86}{1.481} \times [1 + 0.6373]^{4/5}\right)$$

$$= 0.3 + [(137.65) \times (1.4836)] = 204.5 \approx 205$$

Where, $Nu_d = \text{Nusselt Number} = \frac{hd}{k_f}$

The forced convection heat transfer coefficient is, therefore;

$$h = \frac{205 \times (9.706 \times 10^{-7})}{73.2} = 2.718 \times 10^{-6} \frac{Btu}{sec \cdot in^2 \cdot deg F} \quad \text{.....Equation 2}$$

For vertical cylinders Reference 11 gives the following equation relating the Reynolds number, Prandtl number and Nusselt number as follows.

$$Nu = (0.664) \times (Re_L)^{0.5} \times (Pr)^{1/3} \quad \text{.....Equation 3}$$

The limits of applicability of this equation is:

$$Re_L < 5 \times 10^5 \quad \text{and} \quad Pr > 0.1$$

Where the Reynolds number Re_L is based on the cylinder length,

$$L = 87.0 \text{ in}$$

$$Re_L = \frac{(1.1707 \times 10^{-5}) \times (394) \times (87.0)}{246.2 \times 10^{-8}} = 162,995 \approx 1.63 \times 10^5$$

Substitution of values in Equation 3 gives,

$$Nu = (0.664) \times (Re_L)^{0.5} \times (Pr)^{1/3} = (0.664) \times (1.63 \times 10^5)^{0.5} \times (0.7)^{1/3} = 238.03$$

The forced convection heat transfer coefficient is, therefore;

$$h = \frac{(238.03) \times (9.706 \times 10^{-7})}{87} = 2.656 \times 10^{-6} \frac{Btu}{sec \cdot in^2 \cdot deg F}$$

The forced convection heat transfer coefficient for both horizontal and vertical cylinders calculated above are both smaller than IAEA recommended value for the 8-120B cask geometry. Therefore, in order to predict the cask temperature conservatively during the HAC fire test, the IAEA recommended value has been used in all the analyses.

5.3 Radiation Modeling

The heat transfer by radiation between two nodes of a finite element model is governed by the following equation (see for example Ref. 4).

$$q = \sigma \varepsilon F A (T_I^4 - T_J^4)$$

where:

q = heat flow rate (Btu/hr)

σ = Stefan-Boltzmann Constant

$$= 1.7136 \times 10^{-9} \text{ (Btu/hr-ft}^2\text{-R}^4\text{)}$$

$$= 3.3056 \times 10^{-15} \text{ (Btu/sec-in}^2\text{-R}^4\text{)}$$

ε = emissivity

F = geometric form factor

A = area (sq ft)

T = temperature (°R)

I = first node number

J = second node number

Two radiation heat transfer systems are modeled: (1) radiation heat transfer between the fire shield outside surface and the environment, and (2) radiation between the fire shield inside surface and the cask shell outside surface and radiation between the cask ends and the impact limiter plates. Emissivity, area, and geometric form factors are defined in both systems.

The overall emissivity for radiation heat transfer between the fire shield and the environment is set equal to the overall emissivity, ε , for heat transfer between two infinite parallel planes as given by the following equation (Ref. 4, page 336).

$$\varepsilon = \frac{\varepsilon_1 \varepsilon_2}{\varepsilon_2 + \varepsilon_1 - \varepsilon_1 \varepsilon_2}$$

Where: ε = overall emissivity

ε_1 = surface 1 emissivity

ε_2 = surface 2 emissivity

The regulations (Article 71.73 of Reference 2) require that an average environment emissivity coefficient of 0.9 must be used for HAC fire test. It also requires that for purpose of calculation, the surface absorptivity coefficient must be either that which the package may be

expected to possess if exposed to fire specified or 0.8, whichever is greater. It is conservatively assumed in the analyses that the package has an absorptivity of 1.0. Therefore, an emissivity coefficient of 0.9 has been conservatively specified for all the elements that radiate heat to the environment. Form factor value of 1.0 is used and the area of the surface is automatically calculated by the computer program.

The radiation between the outer shell and the fire-shield is modeled by superelements (ANSYS MATRIX50). These elements are formed by modeling the radiating surfaces with thermal shells (ANSYS SHELL57). The thermal shield forms an air annulus between itself and the cask body. This annulus is formed by separating the two components with the help of helically wound 5/32" steel wires. The ends of the fire-shield are seal welded to the cask body. Thus, the two surfaces forming the annulus will remain in polished condition throughout the operating life of the cask. The emissivity coefficient of the as-received steel is 0.15 (see Section 4.0). Therefore, the emissivity coefficient of 0.15 is used for the heat transfer between the inside surface of the fire shield and the outside surface of the outer shell.

5.4 Solar Insolation Modeling

The total insolation is required to be 400 gcal/cm² for a 12-hour period for curved surfaces according to the Code of Federal Regulations 10CFR71.71 (Ref 2). The total insolation of 400 gcal/cm² is divided by 12 hours of assumed sunlight to yield an average insolation rate. The average insolation rate is then multiplied by the surface emissivity of 0.7347 (See Reference 12) yielding an insolation rate of 1.742×10^{-4} Btu/sec-in². This insolation heat load is applied to the outside surface of the fire shield.

5.5 Internal Heat Loading

The internal heat load content of the 8-120B Cask has been represented in the finite element models by two different ways. In the first, the heat load is implicitly represented by a uniform heat flux over the cask cavity. In the second, it is explicitly represented by the finite element model of the waste container and the cavity air. The heat load in this case is applied as a constant heat flux over the waste container wall.

5.5.1 Implicit Internal Heat Load Modeling

The 200-Watt decay heat load is modeled as a constant heat flux over the exposed sidewall inner surface of the cask.

$$\text{Internal heat load, } q = 200 \text{ W} = 200 \times 9.4804 \times 10^{-4} = 0.1896 \text{ Btu/sec}$$

The cask inside diameter is 61.8" and the cavity height is 75". Thus, the heat flux on the inside surface of the cask is:

$$\begin{aligned} q_s &= 0.1896 / (\pi \times 61.8 \times 75 + 2 \times \pi / 4 \times 61.8^2) \\ &= 9.2216 \times 10^{-6} \text{ Btu/(sec-in}^2\text{)} \end{aligned}$$

5.5.2 Explicit Internal Heat Load Modeling

In order to obtain the temperature of the waste container and the cavity air during the NCT, they are explicitly represented in the 2-d axisymmetric finite element model with appropriate material properties. The waste container is conservatively assumed to be a cylindrical shell

having the diameter and length to be approximately ½ of the corresponding cavity dimensions.

Waste Container Outside Diameter = 31", Length = 37.5", Wall Thickness = 1.0"

Container Inside Surface Area = $\pi \times (31-2) \times (37.5-2) + 2 \times [\pi \times (31-2)^2 / 4] = 4,555 \text{ in}^2$

The heat flux is applied on the inside surface of the waste container. Its magnitude is:

$$q_w = 0.1896 / 4,555 = 4.163 \times 10^{-5} \text{ Btu/sec-in}^2$$

The cavity air is conservatively assumed to be stagnant - no convective heat transfer is assumed to take place between the waste container and the cask cavity. Thus, the heat transfer between these components takes place by means of conduction through the air and radiation between the two bodies. The air conduction is accounted for by the appropriate material properties of the finite elements representing it. The heat transfer by radiation is achieved by the ANSYS radiation sub-matrix method. Since the objective of this analysis is to obtain the waste temperature conservatively, the coefficients of emissivity for both the surfaces are assigned lower than actual values. The inside surface of the cask is either carbon steel (emissivity 0.6 to 0.7) or, if it has a liner, stainless steel (emissivity 0.54 to 0.63). For both kinds of surfaces the emissivity is assumed to be 0.2. The waste containers are made of painted carbon steel (emissivity 0.8 to 0.9), stainless steel (emissivity 0.54 to 0.63), or plastic (emissivity 0.8 to 0.9). The assumed value of the emissivity of the waste container surface is 0.4. Please note that the emissivity values quoted here have been obtained from Reference 4. The form-factor is automatically calculated by the computer program from the geometry of the two bodies. The finite element model of the cask with explicit internal heat load modeling is shown in Figure 6.

5.6 Puncture Drop Damage Modeling

During the regulatory puncture drop test specified in Article 71.73(c)(3) of Reference 2, the structural analyses have shown that no damage to the cask body will occur. However, the fire-shield which is separated from the cask body with the help of helically-wound wires, may jam into the cask body at the puncture bar contact location. An analysis of the 8-120B cask has been performed assuming that such a contact occurred at the exposed location closest to the top of the fire-shield. The 2-d axisymmetric model is used for this analysis, which conservatively makes the contact area in shape of a circular ring having a width equal to the contact diameter instead of a circle. Full thermal contact is assumed between the fire-shield and the outer shell of the cask at the assumed location of the puncture drop. In other words, the outer shell is assumed to be directly exposed to the fire at this location.

6.0 ANALYSES

The finite element model described in Section 5.0 is analyzed in the following manner:

1. The initial temperature condition is obtained by running the finite element model with the following boundary conditions:
 - ❑ Internal heat load – 200 W
 - ❑ Solar insolation - yes

- ☐ Heat Transfer to the ambient by radiation – yes
 - ☐ Heat transfer to the ambient by natural convection – yes
 - ☐ Ambient air temperature - 100°F
2. The fire transient is run with the body temperature resulting from the above initial conditions. The fire transient is run for 30 minutes (1,800 sec) with the following boundary conditions:
- ☐ Internal heat load – 200 W
 - ☐ Solar insolation - no
 - ☐ Heat Transfer to the ambient by radiation – yes
 - ☐ Heat transfer to the ambient by forced convection – yes
 - ☐ Ambient air temperature - 1475°F
3. The end of fire analysis of the model is performed with the body temperature resulting from the above fire transient to 1801 sec with the following boundary conditions:
- ☐ Internal heat load – 200 W
 - ☐ Solar insolation - no
 - ☐ Heat Transfer to the ambient by radiation – yes
 - ☐ Heat transfer to the ambient by natural convection – yes
 - ☐ Ambient air temperature - 100°F
4. The cool-down analysis of the model is performed with the body temperature resulting from the above fire transient to 14,000 sec with the following boundary conditions:
- ☐ Internal heat load – 200 W
 - ☐ Solar insolation - yes
 - ☐ Heat Transfer to the ambient by radiation – yes
 - ☐ Heat transfer to the ambient by natural convection – yes
 - ☐ Ambient air temperature - 100°F

Figure 7 shows the boundary conditions used during the fire transient analysis.

The finite element models are analyzed to obtain the limiting thermal loading on the 8-120B cask. The internal heat load has been modeled implicitly for the cask body temperature evaluation. It has been demonstrated in the NCT thermal analysis that this model predicts the cask temperature conservatively.

7.0 RESULTS

7.1 Cask Temperature Results

As described earlier the model representing the internal heat load implicitly is used for obtaining the cask body temperature. From the transient analyses of the finite element model, using the loading described in Section 6.0, a time-history data of the temperature in various

components of the cask is obtained. The fire shield, outer shell, inner shell, lead, and seal were considered as the critical components of the cask. The temperatures at representative nodes in these locations are monitored during the entire fire and cool down transient analysis. The nodes that are monitored at these critical components are shown in Figure 8.

Figure 9 gives the plot of the time-history data at the representative nodes of the cask components. Figure 10 gives the same data in cask components that are not directly exposed to the fire. Figure 11 shows the time-history plot of the waste container temperature at a node where the temperature attains a maximum value (Node 2,013 of the 2-d model).

The ANSYS printout of the modeling is provided in Appendix 1 and the summary of the time-history data are provided in Appendix 2.

Table 1 summarizes the maximum temperature of various components of the cask during the entire analysis period.

To capture the maximum stresses in the cask components during the fire accident test conditions, the structural finite element model, with identical node numbers, needs to be run at some critical time instances. From the time-history plot several critical time instants were identified. The criterion used for determining these time instants was that one of the components achieved the highest temperature during the fire transient. Figure 10 identifies these time instants. These time instants are identified as 0.1, 1,800.3, 1,810, 5,000, 7,500, 12,500, and 22,500 seconds after the start of the fire test. The temperature profiles at these time instants are shown in Figures 12 through 18.

7.2 Cavity Air Temperature Results

The temperature profile at the nodes located on the inside (cavity) of the cask is shown in Figure 19. The maximum value of the temperature in the cavity is 320.5°F. The entire cask cavity volume is conservatively considered to be at 320.5°F.

The temperature profile of the cask with the implicit waste modeling (2-d model) at the time when the waste attains a maximum value during the HAC fire (t = 40,289 sec) is shown in Figure 20. The waste attains a maximum temperature of 239.7°F.

7.3 Puncture Drop Damage Results

The temperature profile of the cask obtained from the finite element model with damaged fire shield at the location of puncture drop is shown in Figure 21. From this figure it is observed that due to the assumed damage to the fire shield the temperature increase in the cask body under HAC fire is highly localized. The temperature plot in the lead column near the assumed damage location is shown in Figure 22. A maximum value of 617.7° is attained by the lead column.

Table 1
Summary of Maximum Hypothetical Fire Temperatures

Component	Maximum Calculated Temp.			Maximum Allowable Temperature (°F)
	Location (Node Nos.)	Time (Sec.)	Value (°F)	
Fire Shield	42,910	1,800	1,392	N.A
Outer Shell	12,531	1,800.3	464.4	800
Inner Shell	8,015	4,461.7	295.5	800
Lead	14,338	4,461.7	295.8	622 ⁽¹⁾
Baseplate	2,430	936.48	206.3	800
Primary Lid	37,675	612.66	202.9	800
Secondary Lid	27,023	1,566.13	192.6	800
Primary Lid Seals	25,430	18,225	212.4	235 ⁽²⁾
Secondary Lid Seals	37,678	24,000	202.9	235 ⁽²⁾
Vent Seal	34,802	24,000	206.9	235 ⁽²⁾
Impact Limiter	27,594	24,000	205.1	500 ⁽⁴⁾
Cask Cavity	⁽³⁾	1,800	320.5	⁽⁵⁾
Waste Contents	2,013	40,289	239.7	⁽⁶⁾

NOTES:

- (1) Lead melting point temperature.
- (2) Established based on 110% of the maximum seal temperatures.
- (3) Obtained from the temperature contour plot. See Figure 19.
- (4) Temperature at which the foam material shows 0% thermal decomposition. Obtained from the General Plastics' sales brochure.
- (5) Temperature used for calculating the cavity pressure.
- (6) Waste contents temperature is obtained for reference purpose.

8.0 ANSYS PRINTOUT AND DATA FILES

The print-out of the model data is included in Appendix 1. Appendix 2 included the print-out of the results from the analyses presented in this document.

The complete electronic data of the input and output of all the analyses are included on a CDROM in Appendix 2.

9.0 APPENDICES

Appendix 1 Print-out of the ANSYS model data

Appendix 2 Print-out of the results

Appendix 3 Electronic data on CDROM

Figures
(22 Pages)

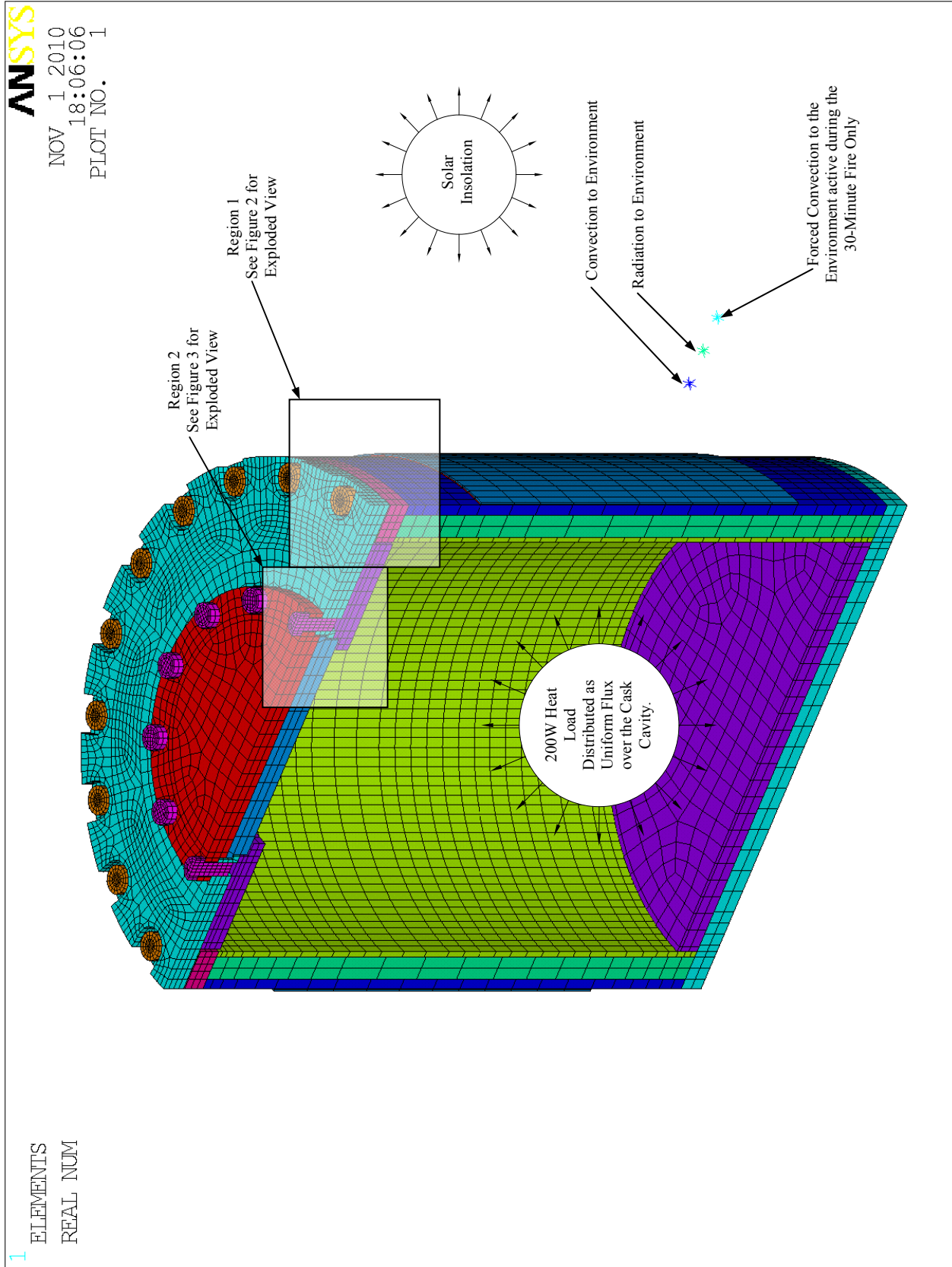


Figure 1 – 3-D Finite Element Model of the 8-120B Cask Used for the Thermal Analyses

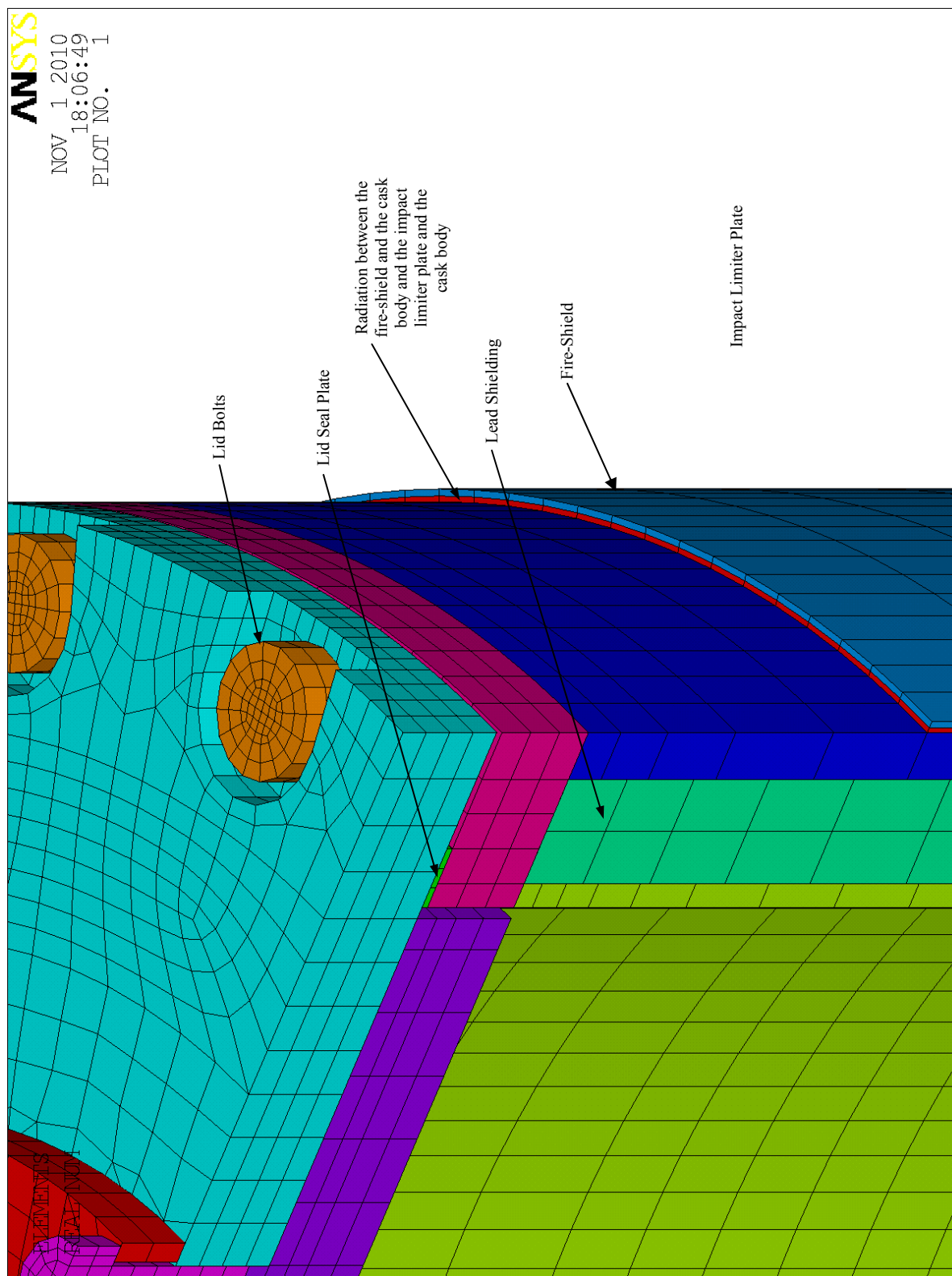


Figure 2 – Exploded View of Region 1 of Figure 1

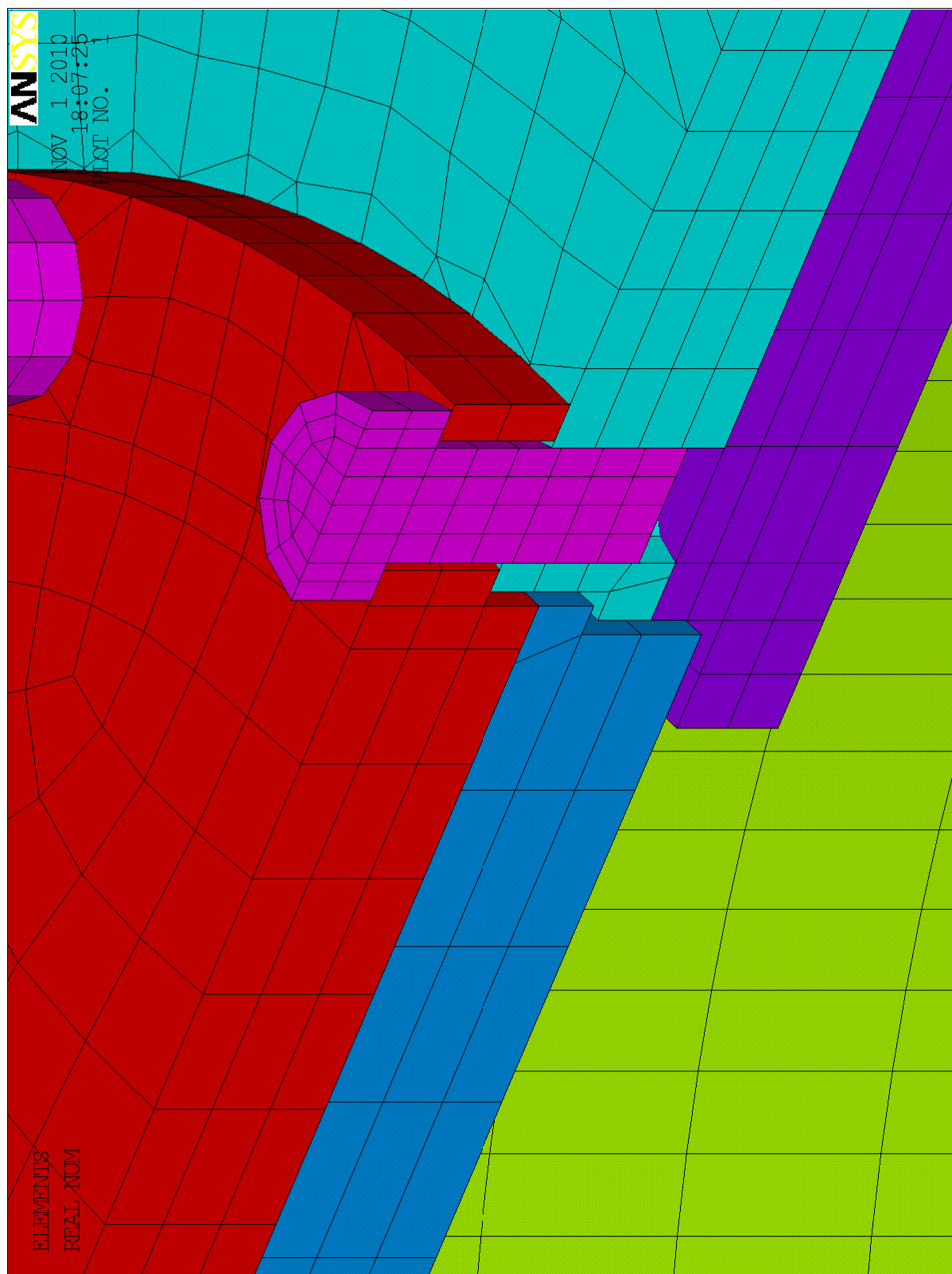


Figure 3 – Exploded View of Region 2 of Figure 1

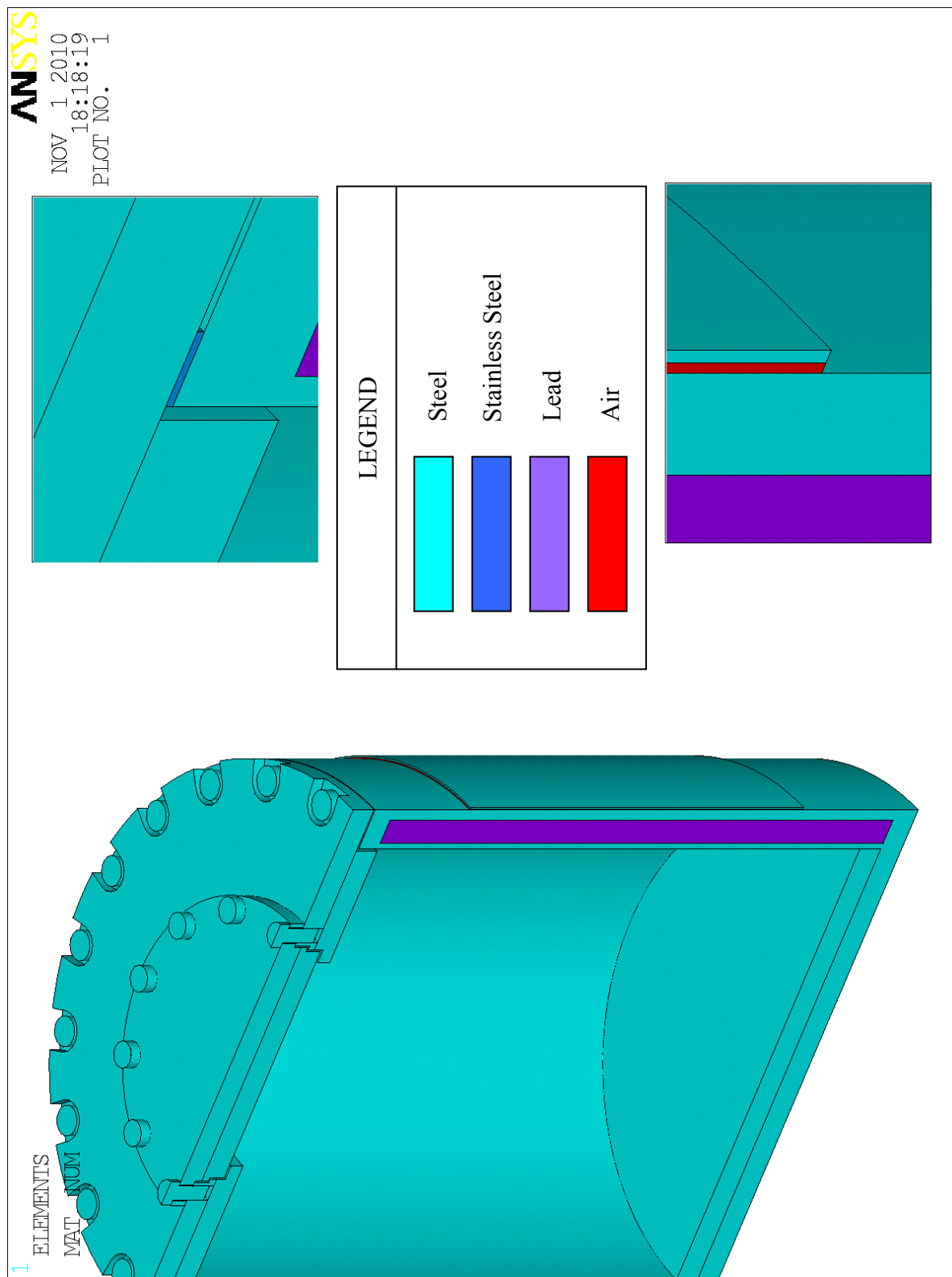


Figure 4 – Materials Used in the Finite Element Model

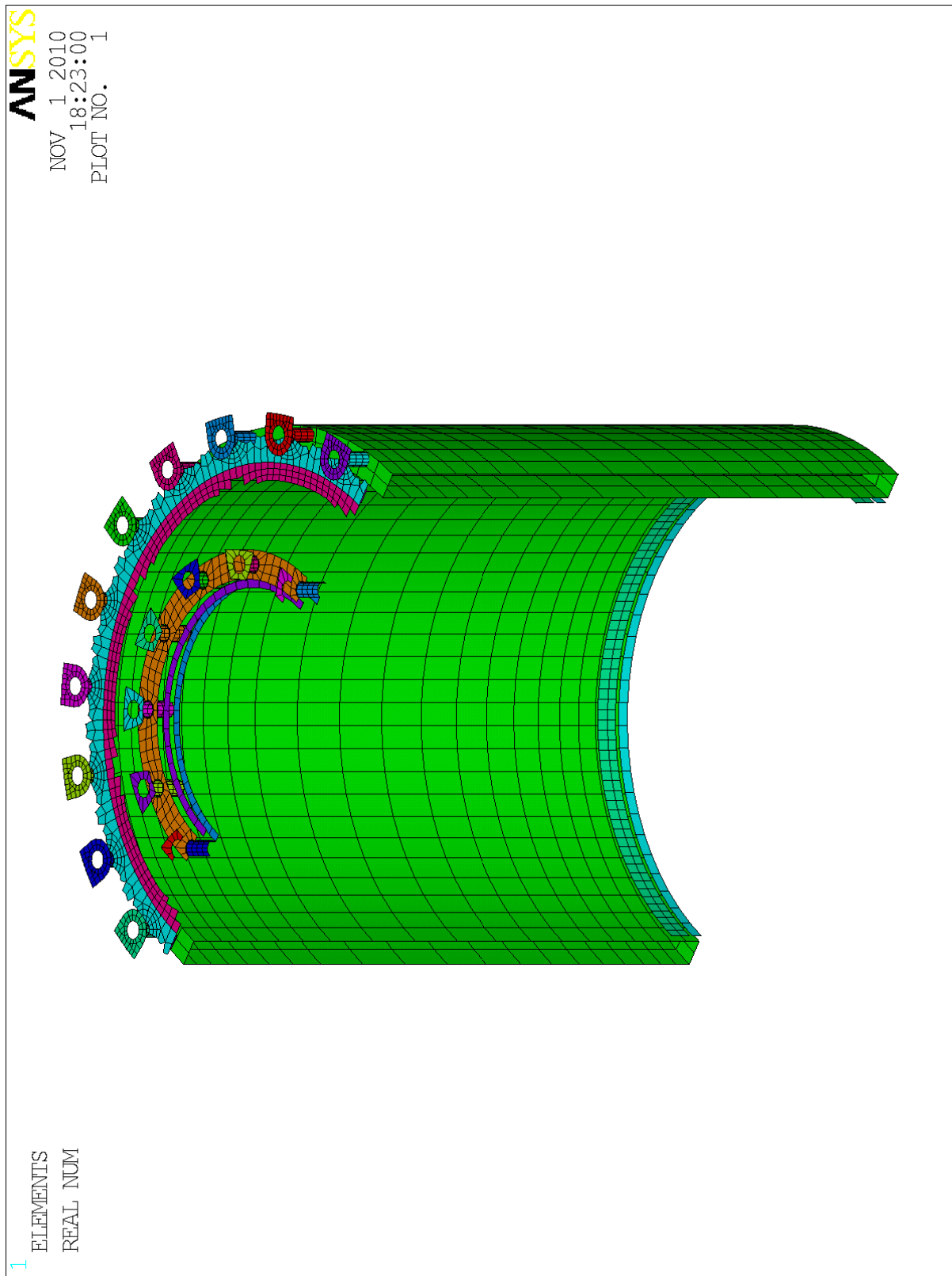
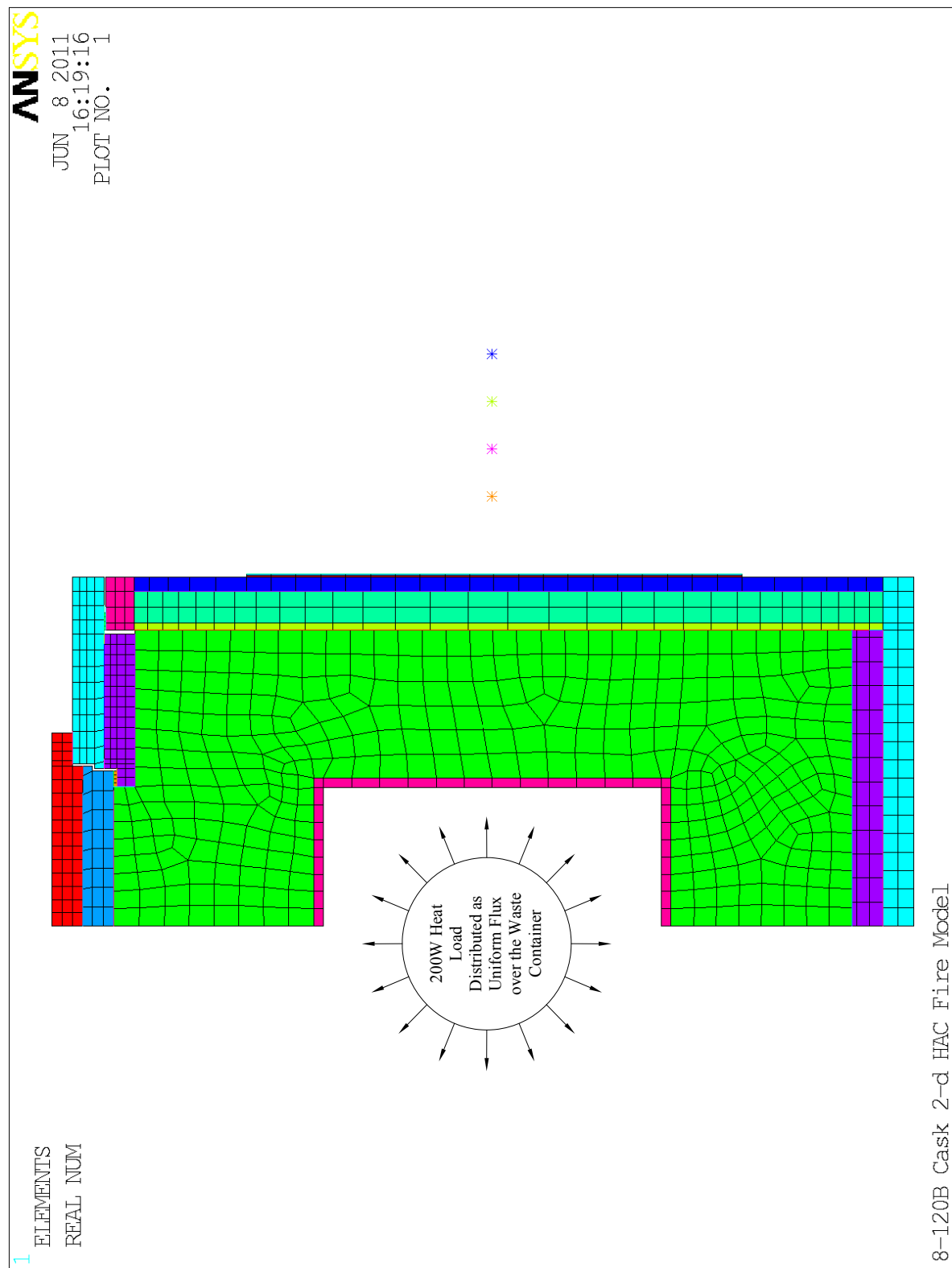


Figure 5 – Contact-Target Pair Elements (Only Target Elements Shown for Clarity)



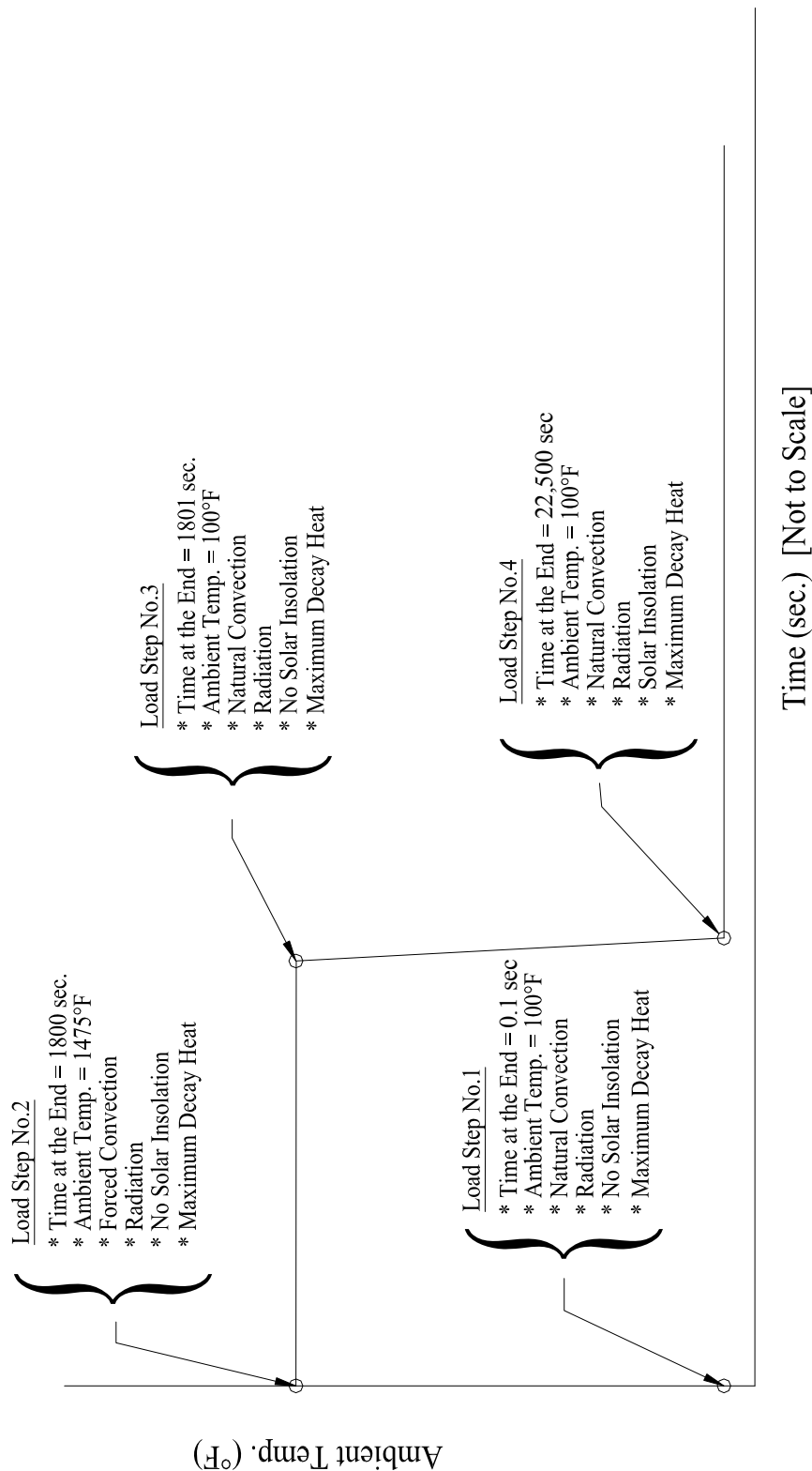


Figure 7 – HAC Fire Analysis Load Steps and Boundary Conditions

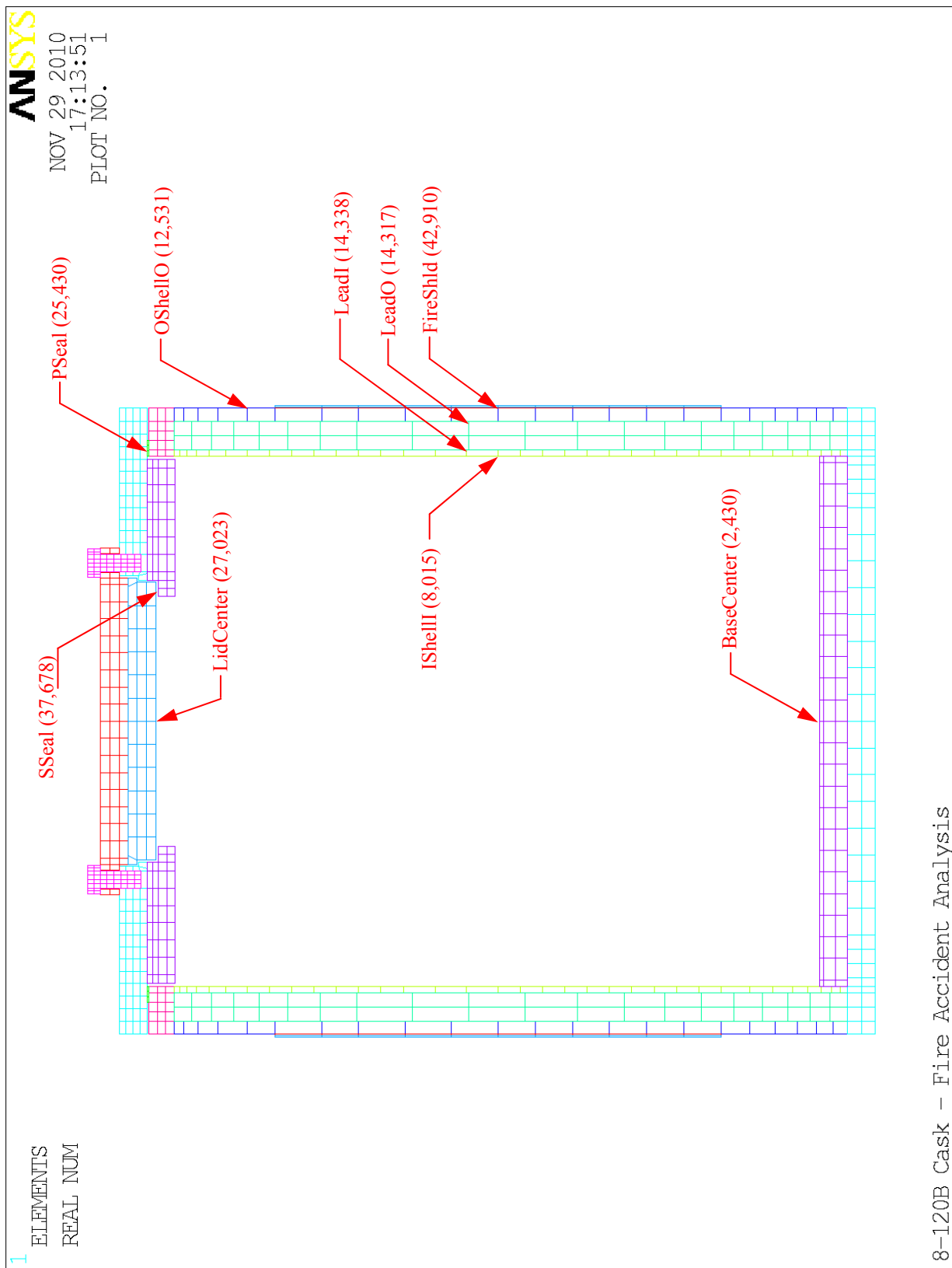
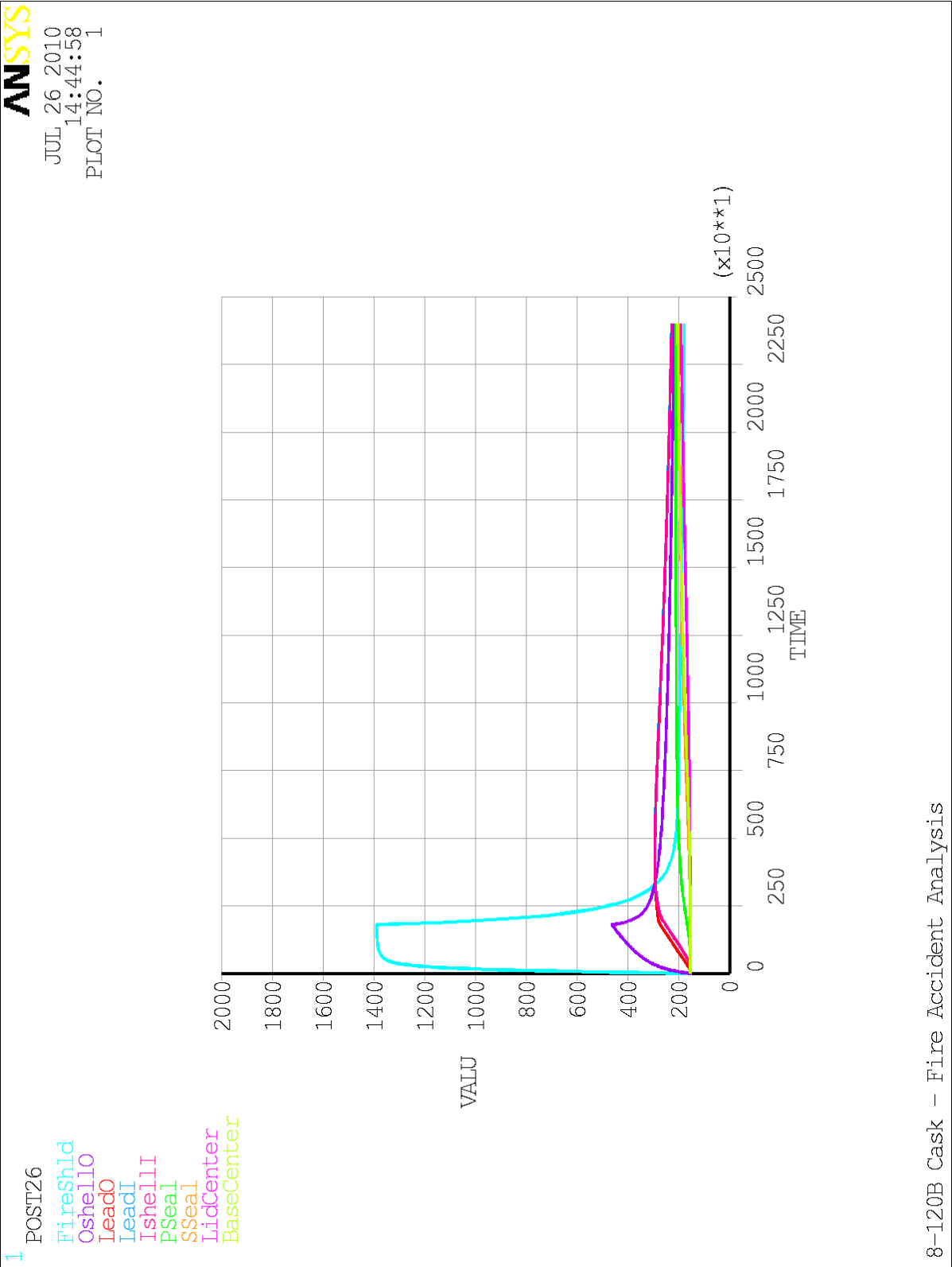
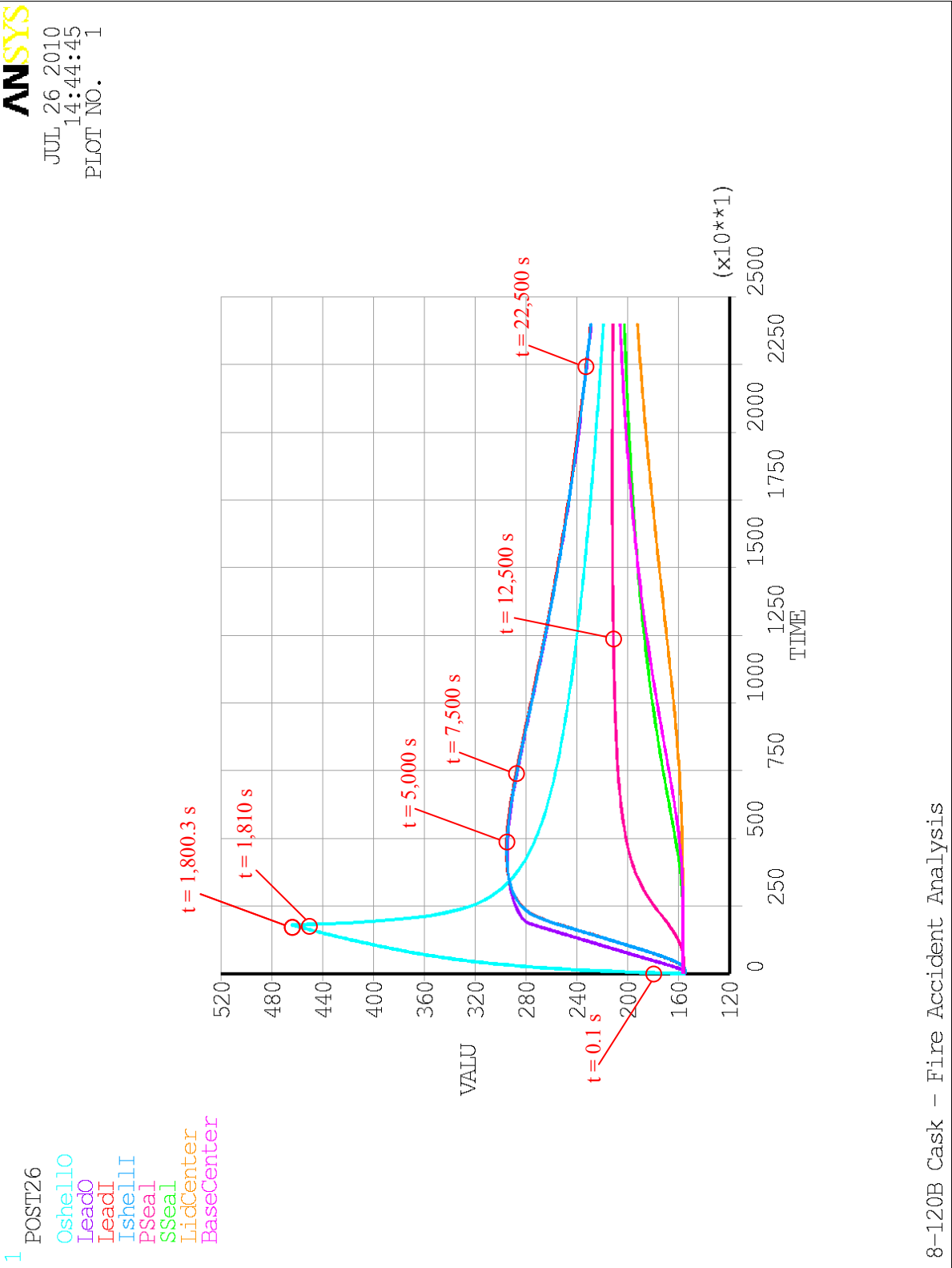


Figure 8 – Identification of the Nodes where Time-History is Monitored





8-120B Cask - Fire Accident Analysis

Figure 10 –Temperature Plot of Nodes not Directly Exposed

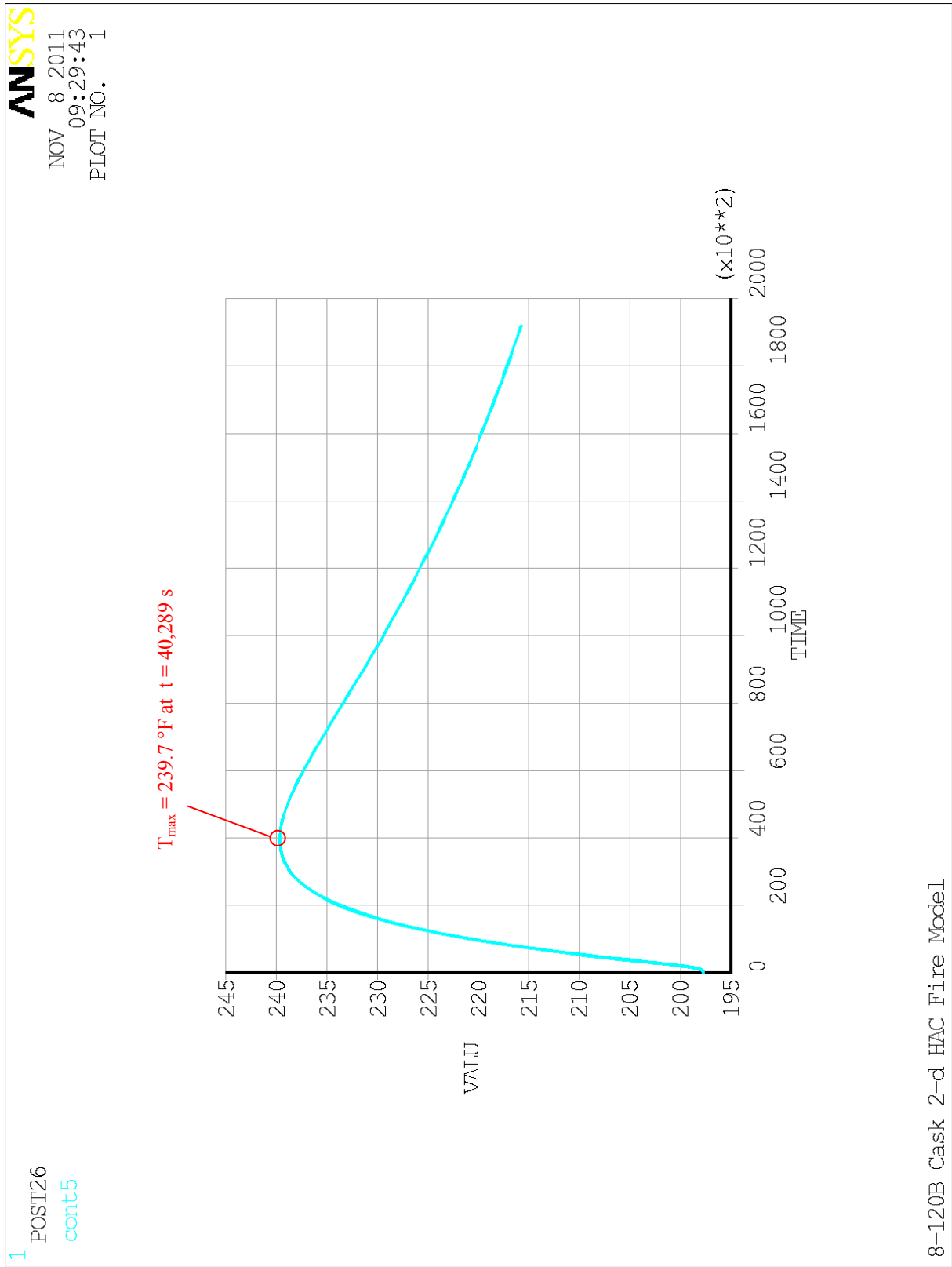
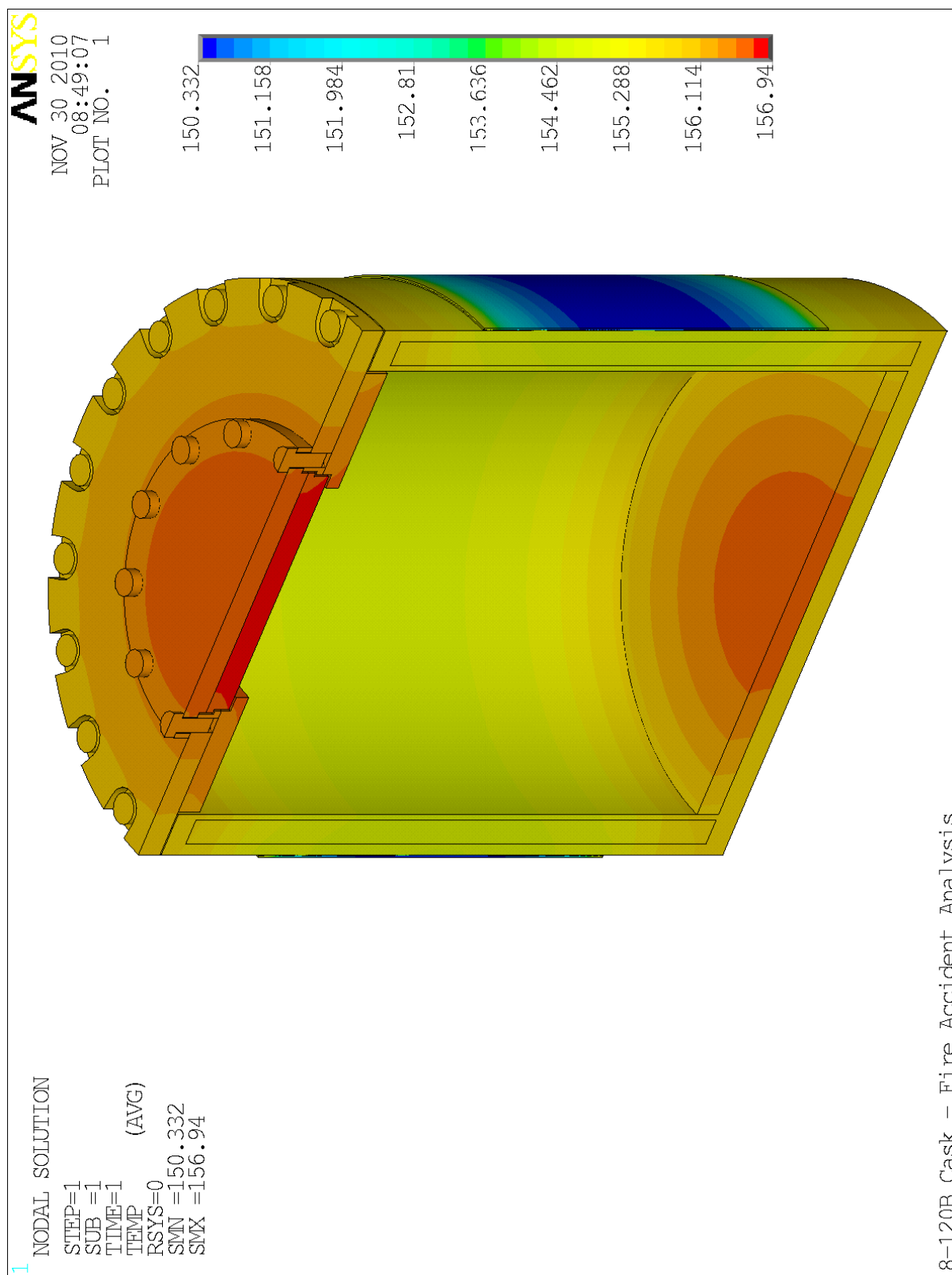


Figure 11 –Time-History Plot of the Waste Container Temperature during the HAC Fire
(from the 2-D Model)



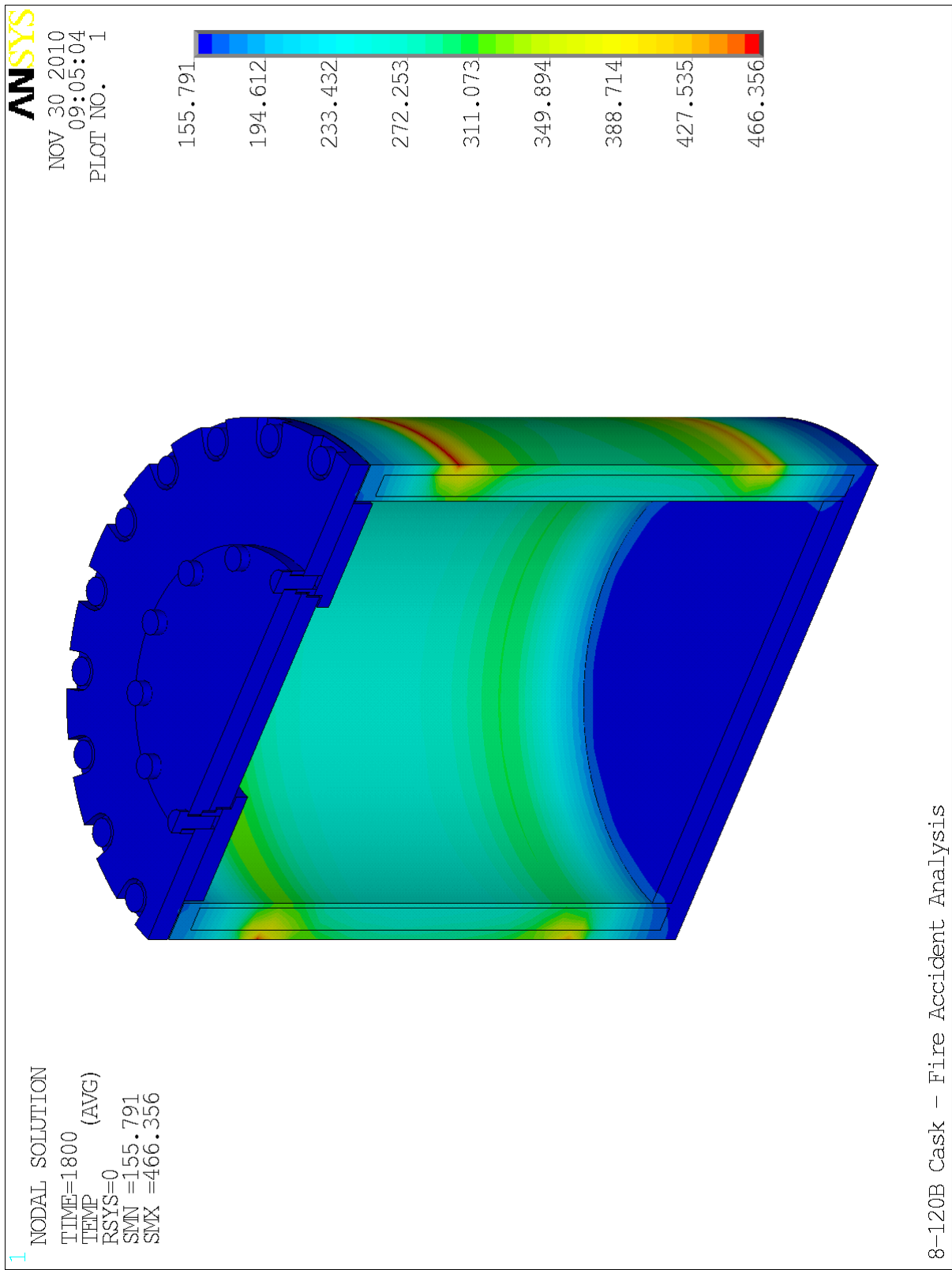


Figure 13 - Temperature Distribution – 1,800.3 Sec. After the Start of the Fire

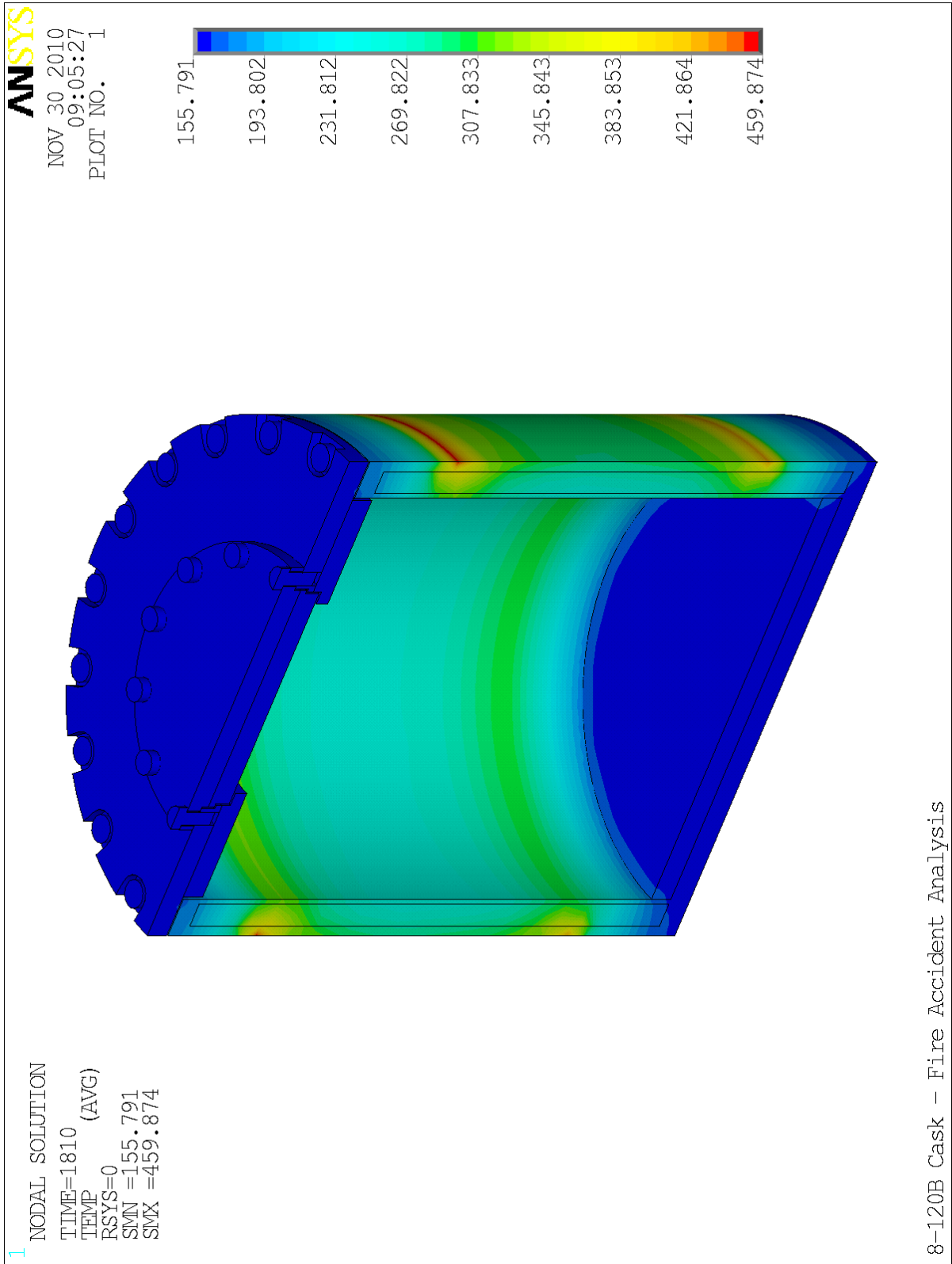


Figure 14 - Temperature Distribution – 1,810 Sec. After the Start of the Fire

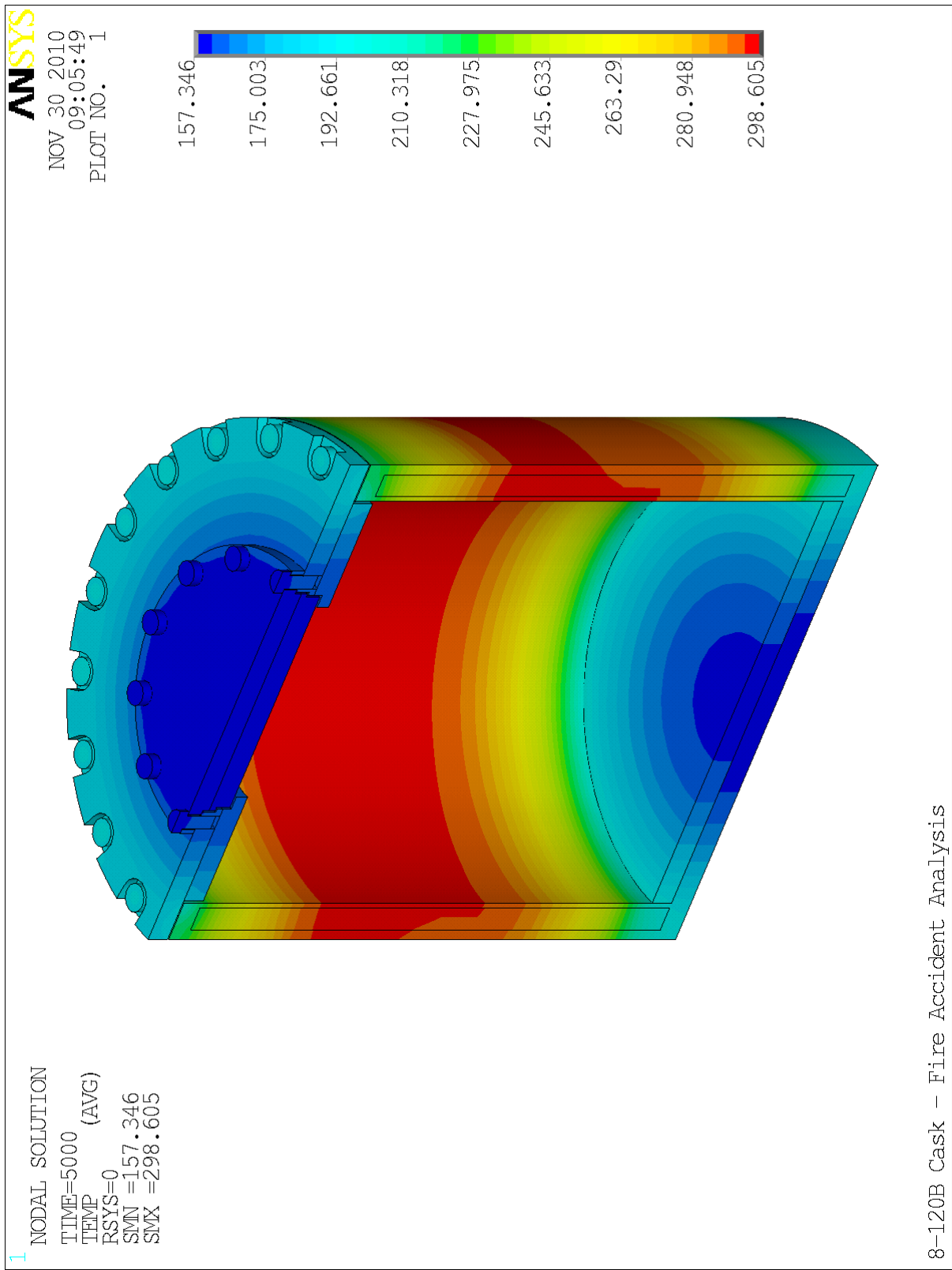


Figure 15 - Temperature Distribution – 5,000 Sec. After the Start of the Fire

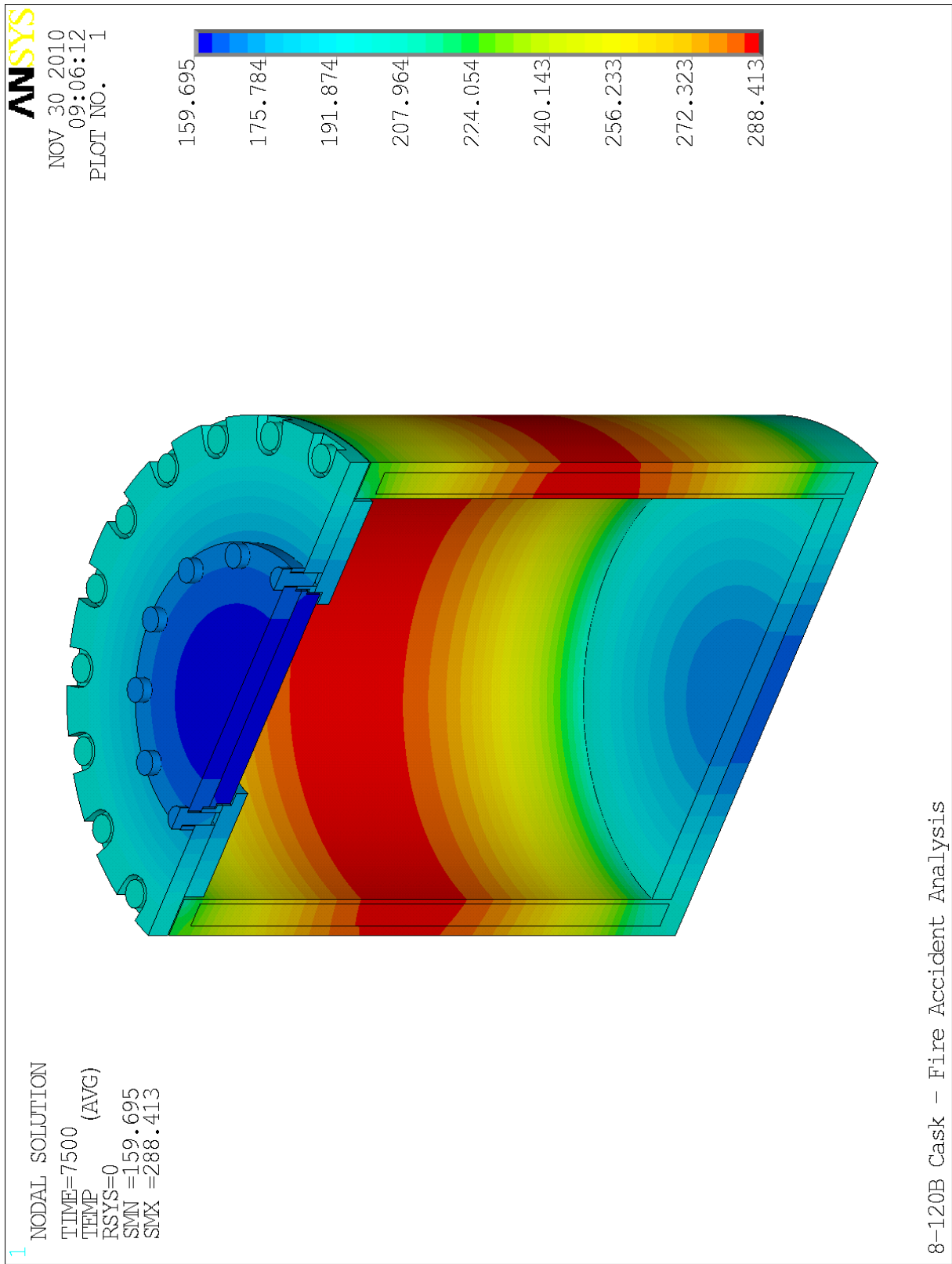


Figure 16 - Temperature Distribution – 7,500 Sec. After the Start of the Fire

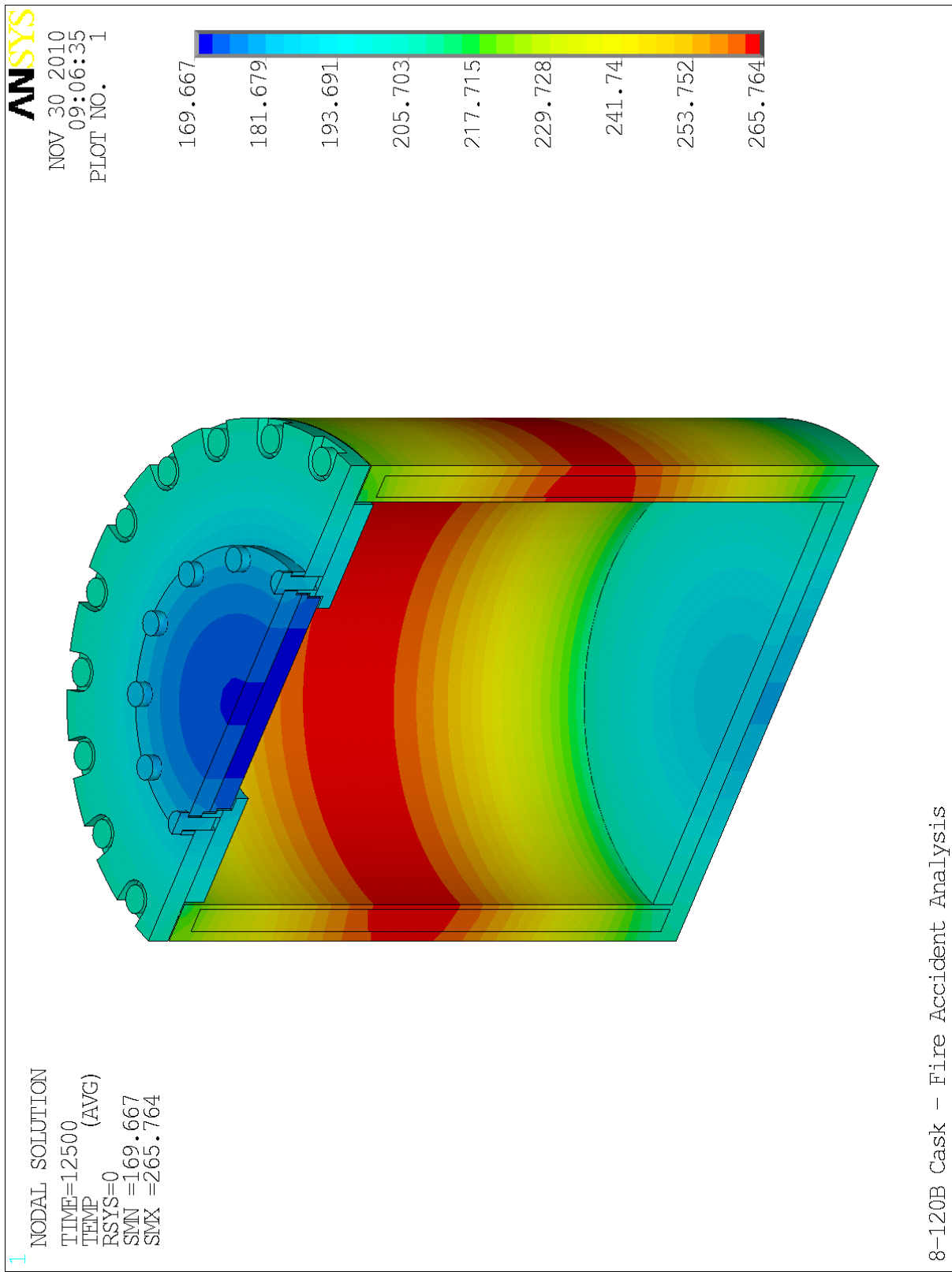


Figure 17 - Temperature Distribution – 12,500 Sec. After the Start of the Fire

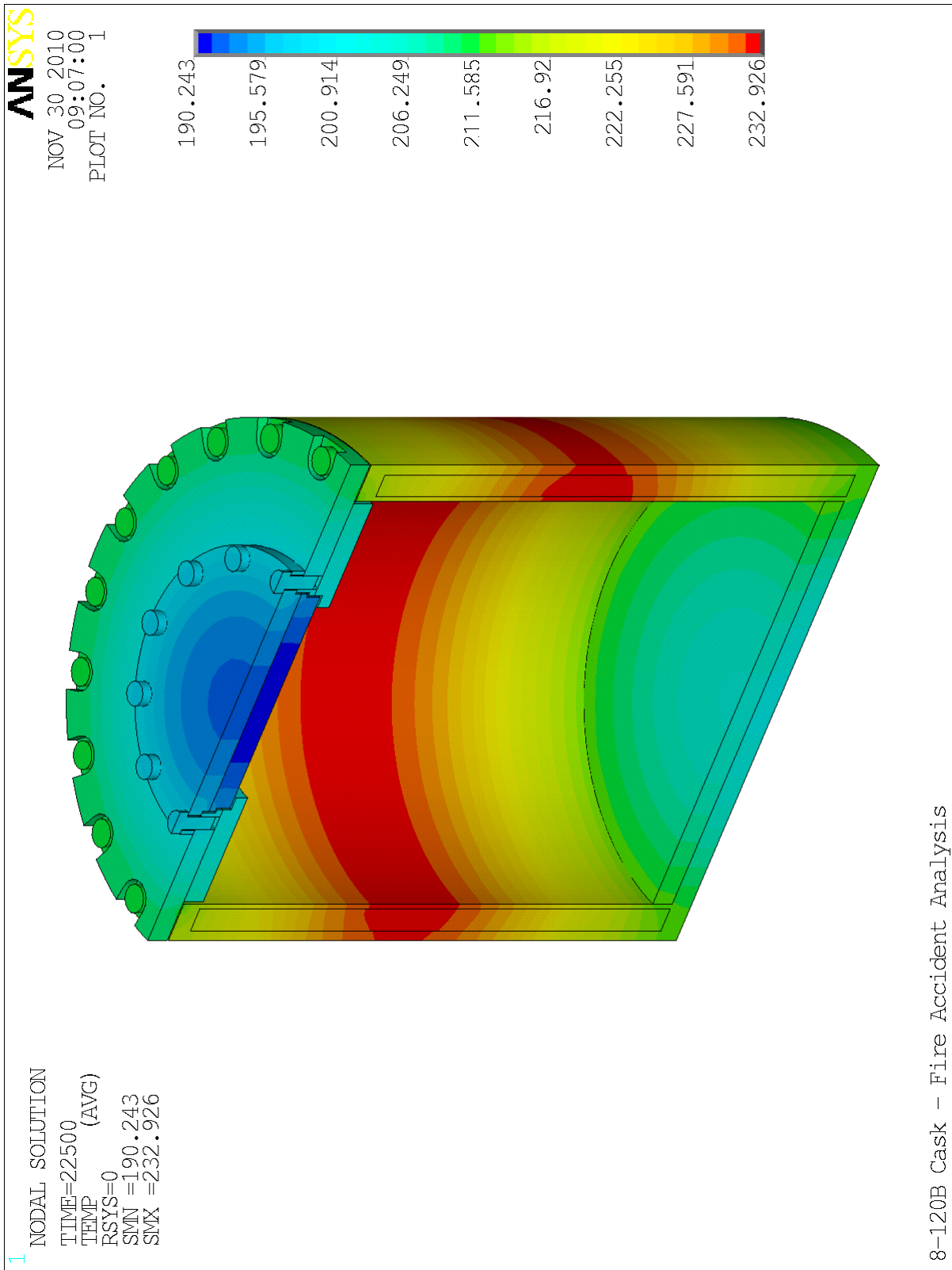


Figure 18 - Temperature Distribution – 22,500 Sec. After the Start of the Fire

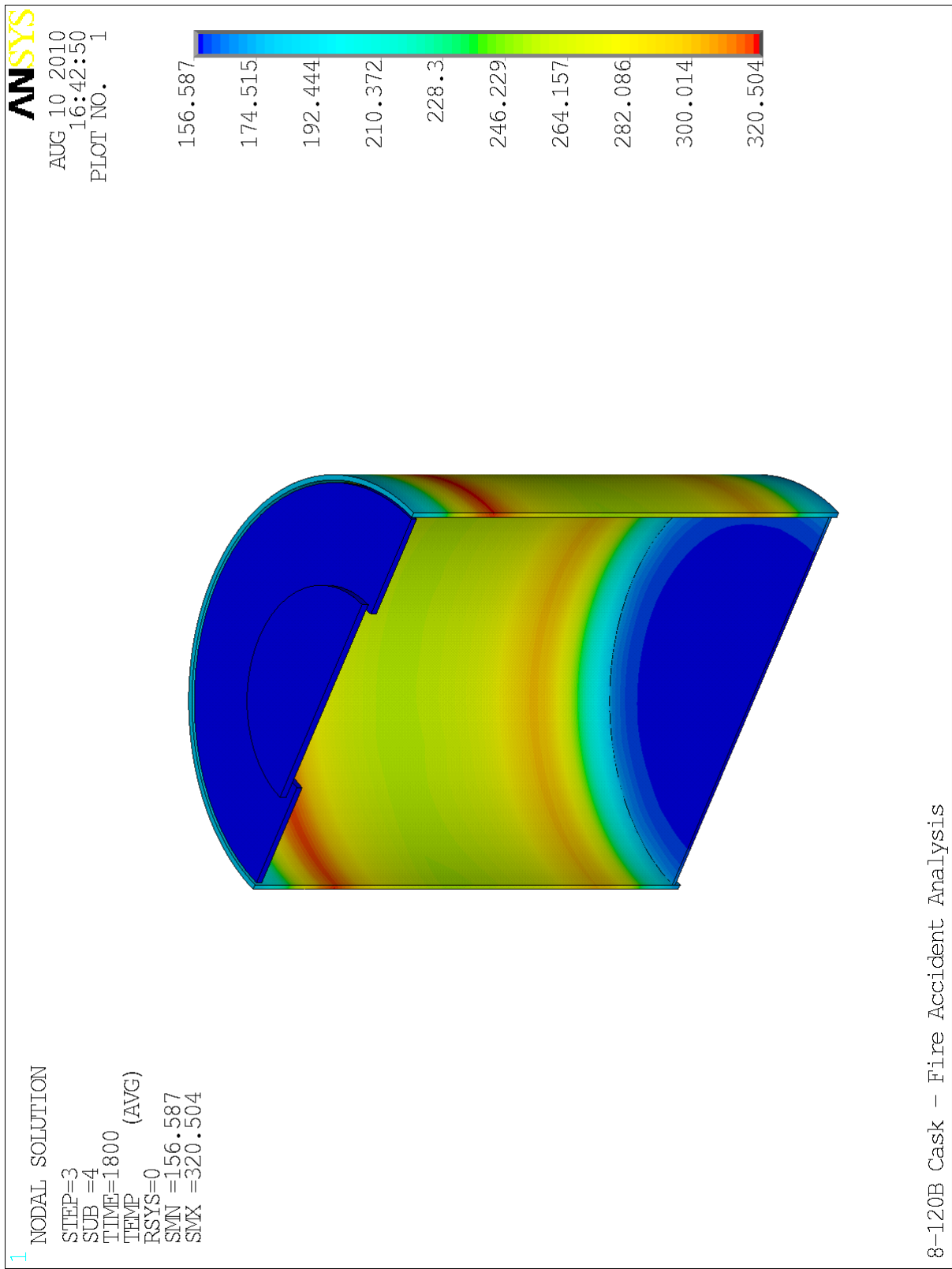


Figure 19 - Temperature Profile of the Cavity Surface

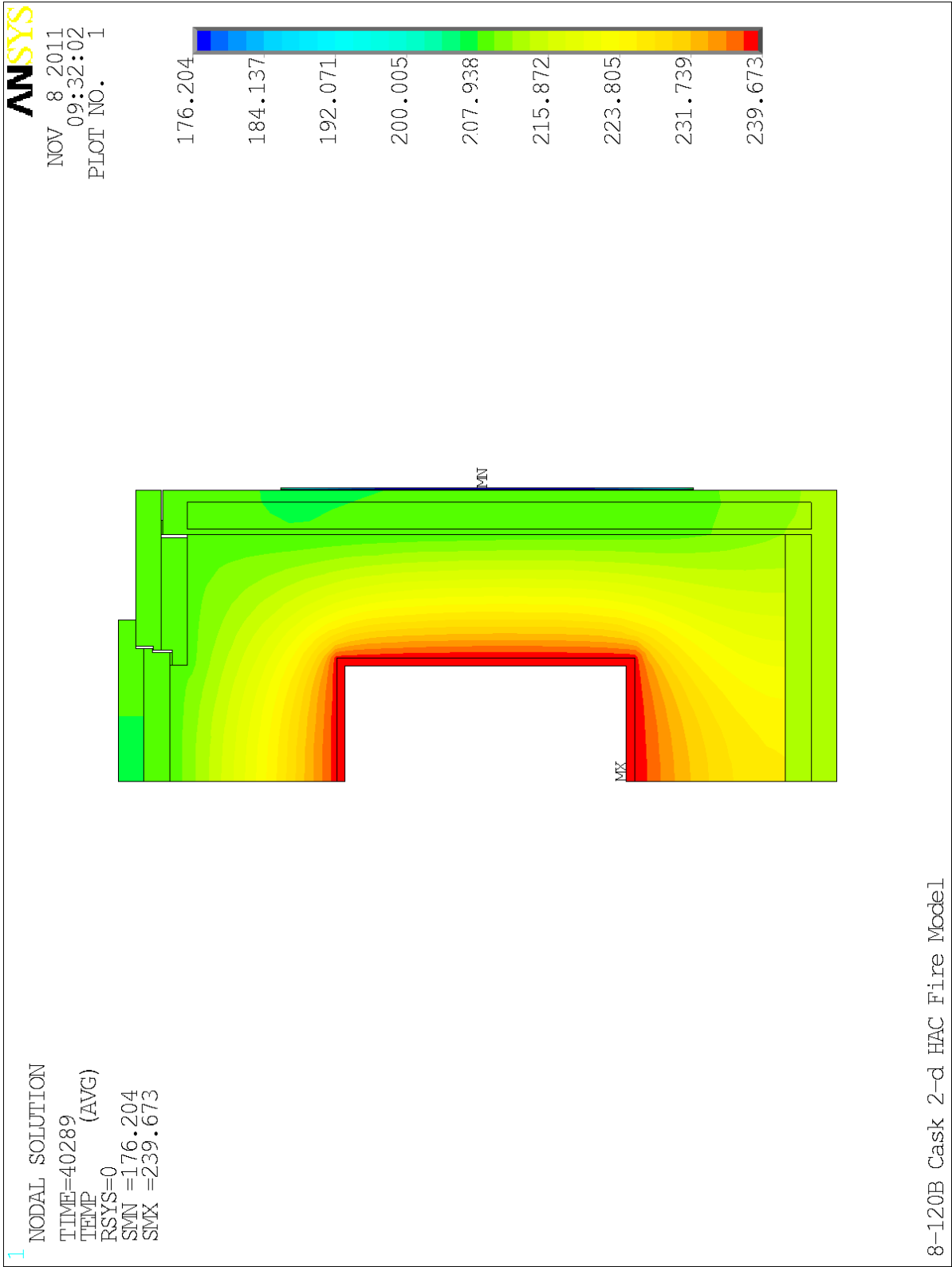


Figure 20 - Temperature Profile of the Cask at time when the Waste Container Temperature Peaks
(from the 2-D Model)

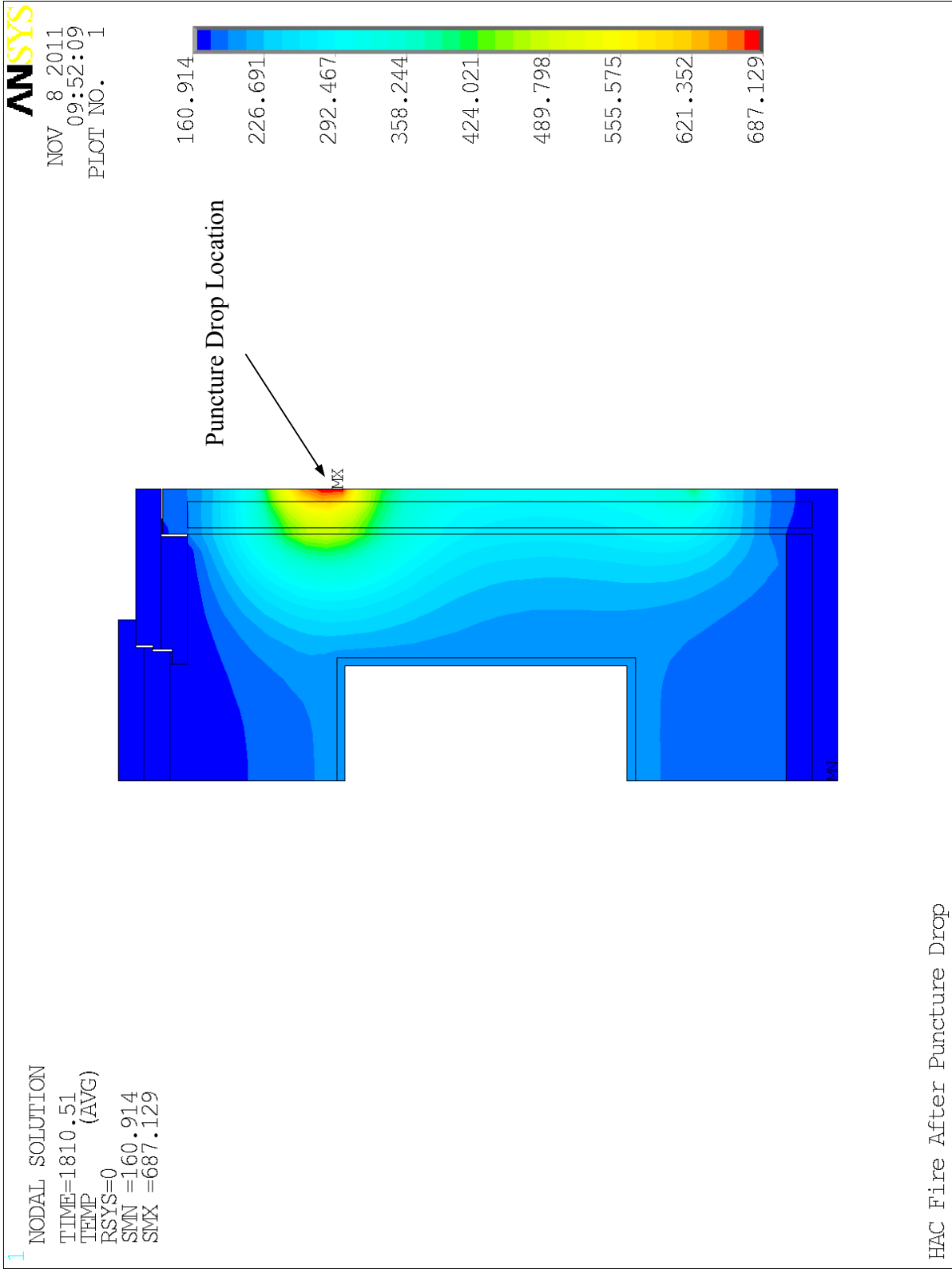
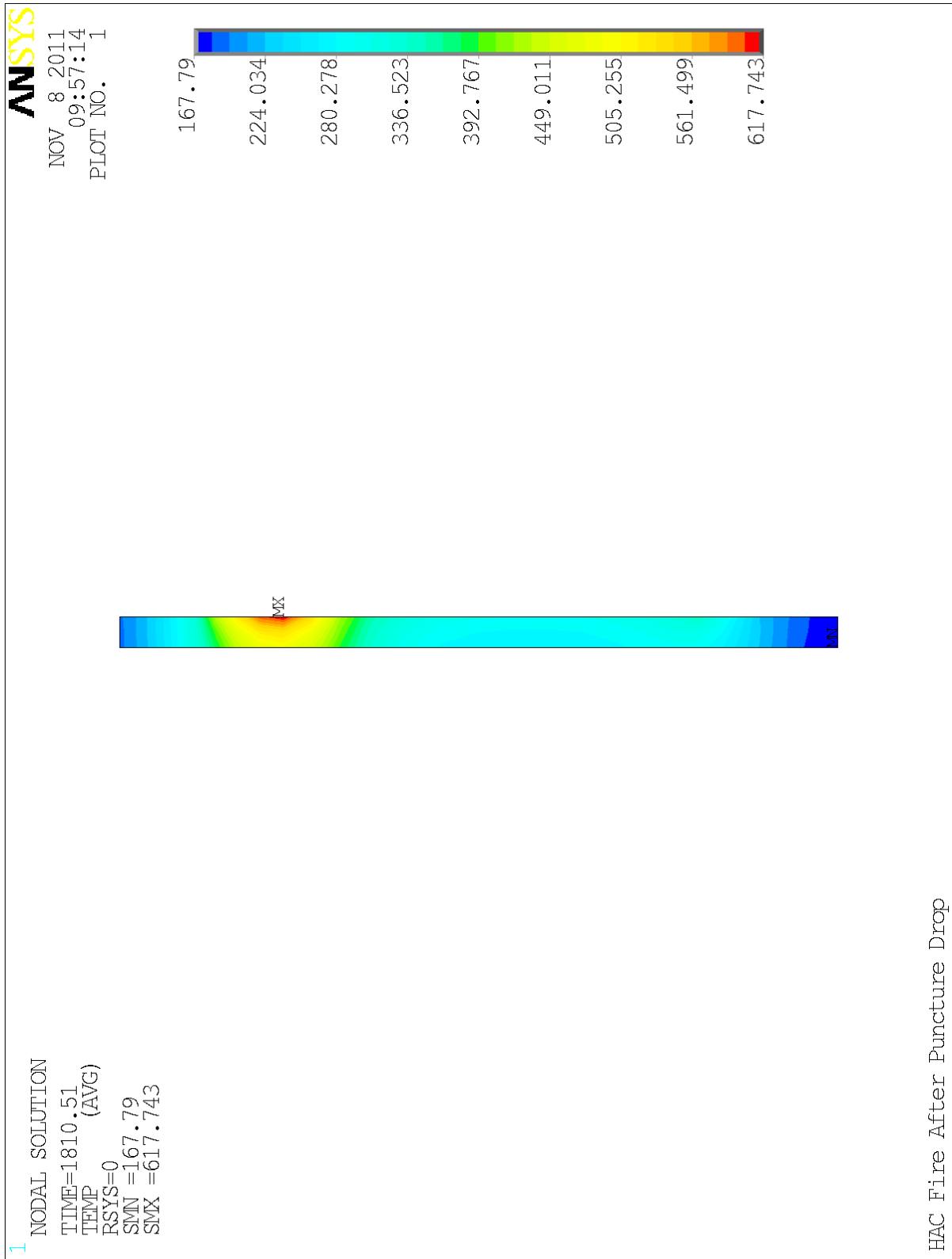


Figure 21 –Temperature Distribution in the Cask with Puncture Drop Damage
(from the 2-D Model)



Appendix 1

Print-out of the ANSYS model data
(33 Pages)

|

ANSYS Finite Element Model Partial Printout3-d Finite Element Model Data

(Note: The complete data printout is included on the file "3d Model\model.out", which is included on the electronic media included in the package)

```

***** TITLES *****

*** YOU ARE IN   ANSYS - ENGINEERING ANALYSIS SYSTEM ***
ANSYS Mechanical
RELEASE  12.1      UPDATE 20091102    CUSTOMER  00222442

INITIAL JOBNAME = file
CURRENT JOBNAME = file

Current Working Directory: D:\ANSYS Analyses\8-120B\Thermal\HAC

TITLE= 8-120B Cask - Fire Accident Analysis

MENULIST File: C:\Program Files\ANSYS Inc\v121\ANSYS\gui\en-
us\UIDL\menulist121.ans

                G L O B A L   S T A T U S

ANSYS - Engineering Analysis System          Nov 01, 2010      11:42
Release 12.1                                00222442          WINDOWS x64   Version

Current working directory: D:\ANSYS Analyses\8-120B\Thermal\HAC

MENULIST File: C:\Program Files\ANSYS Inc\v121\ANSYS\gui\en-
us\UIDL\menulist121.ans

Product(s) enabled: ANSYS Mechanical

Total connect time. . . . . 0 hours  1 minutes
Total CP usage. . . . . 0 hours  0 minutes  2.5 seconds

J O B   I N F O R M A T I O N  -----

8-120B Cask - Fire Accident Analysis

Current jobname . . . . . .file
Initial jobname . . . . . .file

Units . . . . . .unknown

                Available          Used
Scratch Memory Space. . . . 512.000 mb      6.839 mb ( 1.3%)
Database space . . . . . 65535.750 mb      107.485 mb ( 0.2%)

User menu file in use . . . %ANSYS121_DIR%\gui\en-us\UIDL\UIMENU.GRN
User menu file in use . . . %ANSYS121_DIR%\gui\en-us\UIDL\UIFUNC1.GRN

```

User menu file in use . . . %ANSYS121_DIR%\gui\en-us\UIDL\UIFUNC2.GRN
 User menu file in use . . . %ANSYS121_DIR%\gui\en-us\UIDL\MECHTOOL.AUI
 Beta features are not shown in the user interface

MODEL INFORMATION -----

Solid model summary:

	Largest Number	Number Defined	Number Selected
Keypoints	0	0	0
Lines	0	0	0
Areas	0	0	0
Volumes	0	0	0

Finite element model summary:

	Largest Number	Number Defined	Number Selected
Nodes	50005	39845	39845
Elements	52666	40315	40315
Element types	150	101	n.a.
Real constant sets	92	51	n.a.
Material property sets	6	6	n.a.
Coupling	162	162	n.a.
Constraint equations	0	0	n.a.
Master DOFs	0	0	n.a.
Dynamic gap conditions	0	0	n.a.

BOUNDARY CONDITION INFORMATION -----

	Number Defined
Constraints on nodes	2
Constraints on keypoints	0
Constraints on lines	0
Constraints on areas	0
Forces on nodes	0
Forces on keypoints	0
Surface loads on elements	5838
Number of element flagged surfaces	0
Surface loads on lines	0
Surface loads on areas	0
Body loads on elements	0
Body loads on areas	0
Body loads on lines	0
Body loads on nodes	0
Body loads on keypoints	0
Temperatures	
Uniform temperature	0.000
Offset from absolute scale	460.000

	X	Y	Z
Linear acceleration	0.0000	0.0000	0.0000
Angular velocity (about global CS)	0.0000	0.0000	0.0000
Angular acceleration (about global CS) . .	0.0000	0.0000	0.0000
Location of reference CS.	0.0000	0.0000	0.0000
Angular velocity (about reference CS) . .	0.0000	0.0000	0.0000
Angular acceleration (about reference CS)	0.0000	0.0000	0.0000

ROUTINE INFORMATION -----

Current routine.Preprocessing (PREP7)

Active coordinate system 1 (Cylindrical)

Display coordinate system. 0 (Cartesian)

Current element attributes:

Type number	5	(MATRIX50)
Real number	94	
Material number	5	
Element coordinate system number. .	0	

Current mesher type.based on default element shape

Current element meshing shape 2D . . .use default element shape.

Current element meshing shape 3D . . .use default element shape.

SmrtSize Level OFF

Global element size. 0 divisions per line

Active coordinate system 1 (Cylindrical)

Display coordinate system. 0 (Cartesian)

Analysis type.Transient

Active options for this analysis type:

Solution method	Full
Equation solver to use.	Sparse

Results filefile.rth

Load step number 2

Number of substeps:

Starting number of substeps	5000
Maximum number of substeps. . . .	5000
Minimum number of substeps. . . .	100
Step change boundary conditions .	Yes

SOLUTION OPTIONS

PROBLEM DIMENSIONALITY.3-D

DEGREES OF FREEDOM. TEMP

```

ANALYSIS TYPE . . . . .TRANSIENT
  SOLUTION METHOD. . . . .FULL
OFFSET TEMPERATURE FROM ABSOLUTE ZERO . . . . . 460.00
EQUATION SOLVER OPTION. . . . .SPARSE
NEWTON-RAPHSON OPTION . . . . .PROGRAM CHOSEN
GLOBALLY ASSEMBLED MATRIX . . . . .SYMMETRIC

```

LOAD STEP OPTIONS

LOAD STEP NUMBER.	2	
TIME AT END OF THE LOAD STEP.	24001.	
AUTOMATIC TIME STEPPING	ON	
INITIAL NUMBER OF SUBSTEPS	5000	
MAXIMUM NUMBER OF SUBSTEPS	5000	
MINIMUM NUMBER OF SUBSTEPS	100	
MAXIMUM NUMBER OF EQUILIBRIUM ITERATIONS.	15	
STEP CHANGE BOUNDARY CONDITIONS	YES	
TRANSIENT (INERTIA) EFFECTS		
THERMAL DOFS	ON	
TRANSIENT INTEGRATION PARAMETERS		
THETA.	1.0000	
OSCILLATION LIMIT CRITERION.	0.50000	
TOLERANCE.	0.0000	
PRESSURE LOAD STIFFNESS	NEVER USED	
TERMINATE ANALYSIS IF NOT CONVERGED	YES (EXIT)	
CONVERGENCE CONTROLS.	USE DEFAULTS	
PRINT OUTPUT CONTROLS	NO PRINTOUT	
DATABASE OUTPUT CONTROLS		
ITEM	FREQUENCY	COMPONENT
BASI	ALL	

LIST ELEMENT TYPES FROM 1 TO 150 BY 1

ELEMENT TYPE	2	IS	SOLID70	3-D	THERMAL	SOLID	
KEYOPT(1- 6)=	0		0	0		0	0
KEYOPT(7-12)=	0		0	0		0	0
KEYOPT(13-18)=	0		0	0		0	0

ELEMENT TYPE	4	IS	SHELL57	THERMAL	SHELL	
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0

ELEMENT TYPE	5	IS	MATRIX50	SUPERELEMENT	(SUBSTRUCTURE)
KEYOPT(1- 6)=	1	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0

ELEMENT TYPE	7	IS	TARGET170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0

ELEMENT TYPE	8 IS	CONTA174	3D 8-NODE	THERMAL	CONTACT
KEYOPT (1- 6) =	2	2	0	2	0
KEYOPT (7-12) =	0	0	0	2	5

KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	11	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	12	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	17	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	18	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	19	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	20	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	21	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	22	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	23	IS	TARGE170	3-D	TARGET	SEGMENT
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KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	24	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	25	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0

ELEMENT TYPE	26	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	27	IS	TARGE170	3-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	28	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	29	IS	TARGE170	3-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	30	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	31	IS	TARGE170	3-D TARGET SEGMENT
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KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	32	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	33	IS	TARGE170	3-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	34	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	35	IS	TARGE170	3-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	36	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	37	IS	TARGE170	3-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0

KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	38	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	39	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	40	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	41	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	42	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	43	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	44	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	45	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	46	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	47	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	48	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0

ELEMENT TYPE	49	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	50	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	51	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	52	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	53	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	54	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	55	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	56	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	57	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	58	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	59	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	60	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT

KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 61 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 62 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 63 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 64 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 65 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 66 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 67 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 68 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 69 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 70 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 71 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0

KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	72	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	73	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0 0
KEYOPT(7-12)=	0		0	0	0	0 0
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	74	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	75	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0 0
KEYOPT(7-12)=	0		0	0	0	0 0
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	76	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	77	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0 0
KEYOPT(7-12)=	0		0	0	0	0 0
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	78	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	79	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0 0
KEYOPT(7-12)=	0		0	0	0	0 0
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	80	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	81	IS	TARGE170	3-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0 0
KEYOPT(7-12)=	0		0	0	0	0 0
KEYOPT(13-18)=	0		0	0	0	0 0
ELEMENT TYPE	82	IS	CONTA174	3D	8-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	2	0 5
KEYOPT(13-18)=	0		0	0	0	0 0

ELEMENT TYPE	83	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	84	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	85	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	86	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	89	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	90	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	91	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	92	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	93	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	94	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0
KEYOPT(7-12)=			0 0	0	2	0	5
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	95	IS	TARGE170	3-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0
ELEMENT TYPE	96	IS	CONTA174	3D	8-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 2	0	2	0	0

KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 97 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 98 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 99 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 100 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 101 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 102 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 103 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 104 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 105 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 106 IS CONTA174 3D 8-NODE THERMAL CONTACT						
KEYOPT(1- 6)=	2	2	0	2	0	0
KEYOPT(7-12)=	0	0	0	2	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE 109 IS TARGE170 3-D TARGET SEGMENT						
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0

ELEMENT TYPE	110	IS	CONTA174	3D 8-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0

ELEMENT TYPE	147	IS	SURF152	3-D THERMAL SURFACE
KEYOPT(1- 6)=	0		0	0 1 1 0
KEYOPT(7-12)=	0		0	1 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0

ELEMENT TYPE	148	IS	SURF152	3-D THERMAL SURFACE
KEYOPT(1- 6)=	0		0	0 1 1 0
KEYOPT(7-12)=	1		4	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0

ELEMENT TYPE	149	IS	SURF152	3-D THERMAL SURFACE
KEYOPT(1- 6)=	0		0	0 1 1 0
KEYOPT(7-12)=	0		4	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0

ELEMENT TYPE	150	IS	SURF152	3-D THERMAL SURFACE
KEYOPT(1- 6)=	0		0	0 1 0 0
KEYOPT(7-12)=	0		1	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0

CURRENT NODAL DOF SET IS TEMP
THREE-DIMENSIONAL MODEL

LIST MATERIALS 1 TO 6 BY 1
PROPERTY= ALL

MATERIAL NUMBER 1

TEMP	DENS
	0.2830000

TEMP	KXX
70.000	0.81300E-03
100.00	0.80300E-03
200.00	0.77800E-03
300.00	0.74800E-03
400.00	0.71500E-03
500.00	0.67700E-03
600.00	0.64800E-03
700.00	0.61600E-03
800.00	0.58300E-03
900.00	0.55100E-03
1000.0	0.51900E-03
1100.0	0.48400E-03
1200.0	0.45100E-03
1300.0	0.41700E-03
1400.0	0.38000E-03
1500.0	0.36300E-03

TEMP	C
70.000	0.10330

100.00	0.10530
200.00	0.11210
300.00	0.11770
400.00	0.12340
500.00	0.12780
600.00	0.13220
700.00	0.13810
800.00	0.14520
900.00	0.15350
1000.0	0.16240
1100.0	0.17100
1200.0	0.18290
1300.0	0.20450
1400.0	0.40100
1500.0	0.19820

MATERIAL NUMBER 2

TEMP	DENS
70.000	0.41090
100.00	0.41090
150.00	0.41090
200.00	0.41090
250.00	0.41090
300.00	0.41090
350.00	0.41090
400.00	0.41090
450.00	0.41090
500.00	0.41090
550.00	0.41090
600.00	0.41090
650.00	0.41090
700.00	0.41090
750.00	0.41090
800.00	0.41090
900.00	0.41090
1000.0	0.41090
1100.0	0.41090
1200.0	0.41090
1300.0	0.41090
1400.0	0.41090
1500.0	0.41090

TEMP	KXX
70.000	0.46500E-03
100.00	0.46100E-03
150.00	0.45500E-03
200.00	0.44800E-03
250.00	0.44100E-03
300.00	0.43500E-03
350.00	0.42800E-03
400.00	0.42200E-03
450.00	0.41500E-03
500.00	0.40900E-03
550.00	0.40200E-03
600.00	0.39500E-03
650.00	0.38900E-03

700.00	0.38900E-03
750.00	0.38900E-03
800.00	0.38900E-03
900.00	0.38900E-03
1000.0	0.38900E-03
1100.0	0.38900E-03
1200.0	0.38900E-03
1300.0	0.38900E-03
1400.0	0.38900E-03
1500.0	0.38900E-03

TEMP	C
70.000	0.31050E-01
100.00	0.31050E-01
150.00	0.31050E-01
200.00	0.31050E-01
250.00	0.31050E-01
300.00	0.31050E-01
350.00	0.31050E-01
400.00	0.31050E-01
450.00	0.31050E-01
500.00	0.31050E-01
550.00	0.31050E-01
600.00	0.31050E-01
650.00	0.31050E-01
700.00	0.31050E-01
750.00	0.31050E-01
800.00	0.31050E-01
900.00	0.31050E-01
1000.0	0.31050E-01
1100.0	0.31050E-01
1200.0	0.31050E-01
1300.0	0.31050E-01
1400.0	0.31050E-01
1500.0	0.31050E-01

MATERIAL NUMBER	3
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TEMP	DENS
70.000	0.43510E-04
100.00	0.41120E-04
150.00	0.37520E-04
200.00	0.34680E-04
250.00	0.32360E-04
300.00	0.30310E-04
350.00	0.28310E-04
400.00	0.26730E-04
450.00	0.25220E-04
500.00	0.23960E-04
550.00	0.22780E-04
600.00	0.21680E-04
650.00	0.20710E-04
700.00	0.19800E-04
750.00	0.18980E-04
800.00	0.18180E-04
900.00	0.16900E-04
1000.0	0.15710E-04

1100.0	0.14720E-04
1200.0	0.13850E-04
1300.0	0.13040E-04
1400.0	0.12350E-04
1500.0	0.11710E-04

TEMP	KXX
70.000	0.34490E-06
100.00	0.36210E-06
150.00	0.39030E-06
200.00	0.41770E-06
250.00	0.44460E-06
300.00	0.47040E-06
350.00	0.49570E-06
400.00	0.52040E-06
450.00	0.54480E-06
500.00	0.56880E-06
550.00	0.59210E-06
600.00	0.61430E-06
650.00	0.63630E-06
700.00	0.65810E-06
750.00	0.67900E-06
800.00	0.69960E-06
900.00	0.74090E-06
1000.0	0.78040E-06
1100.0	0.81750E-06
1200.0	0.85450E-06
1300.0	0.88970E-06
1400.0	0.92850E-06
1500.0	0.97070E-06

TEMP	C
70.000	0.24020
100.00	0.24040
150.00	0.24080
200.00	0.24140
250.00	0.24210
300.00	0.24290
350.00	0.24380
400.00	0.24500
450.00	0.24610
500.00	0.24740
550.00	0.24900
600.00	0.25110
650.00	0.25270
700.00	0.25380
750.00	0.25520
800.00	0.25680
900.00	0.25960
1000.0	0.26280
1100.0	0.26590
1200.0	0.26890
1300.0	0.27170
1400.0	0.27420
1500.0	0.27660

TEMP	DENS
70.000	0.28240
100.00	0.28240
150.00	0.28240
200.00	0.28240
250.00	0.28240
300.00	0.28240
350.00	0.28240
400.00	0.28240
450.00	0.28240
500.00	0.28240
550.00	0.28240
600.00	0.28240
650.00	0.28240
700.00	0.28240
750.00	0.28240
800.00	0.28240
900.00	0.28240
1000.0	0.28240
1100.0	0.28240
1200.0	0.28240
1300.0	0.28240
1400.0	0.28240
1500.0	0.28240

TEMP	KXX
70.000	0.19900E-03
100.00	0.20100E-03
150.00	0.20800E-03
200.00	0.21500E-03
250.00	0.22200E-03
300.00	0.22700E-03
350.00	0.23400E-03
400.00	0.24100E-03
450.00	0.24500E-03
500.00	0.25200E-03
550.00	0.25700E-03
600.00	0.26200E-03
650.00	0.26900E-03
700.00	0.27300E-03
750.00	0.27800E-03
800.00	0.28200E-03
900.00	0.29400E-03
1000.0	0.30600E-03
1100.0	0.31500E-03
1200.0	0.32400E-03
1300.0	0.33600E-03
1400.0	0.34500E-03
1500.0	0.35400E-03

TEMP	C
70.000	0.11700
100.00	0.11700
150.00	0.12000
200.00	0.12200
250.00	0.12500

300.00	0.12600
350.00	0.12800
400.00	0.12900
450.00	0.13000
500.00	0.13100
550.00	0.13200
600.00	0.13300
650.00	0.13400
700.00	0.13500
750.00	0.13600
800.00	0.13600
900.00	0.13800
1000.0	0.13900
1100.0	0.14100
1200.0	0.14100
1300.0	0.14300
1400.0	0.14400
1500.0	0.14500

MATERIAL NUMBER 5

TEMP	EMIS
	0.1500000

MATERIAL NUMBER 6

TEMP	EMIS
	0.9000000

2-d Finite Element Model Data

(Note: The complete data printout is included on the file "3d Model\model.out", which is included on the electronic media included in the package)

G L O B A L S T A T U S

ANSYS - Engineering Analysis System	Nov 08, 2011	10:02
Release 13.0	00222442	WINDOWS x64 Version

Current working directory: D:\ANSYS Analyses\8-120B\Thermal\Rev2\2-d
Analyses\HAC

MENULIST File: C:\Program Files\ANSYS Inc\v130\ANSYS\gui\en-
us\UIDL\menulist130.ans

Product(s) enabled: ANSYS Mechanical

Total connect time.	0 hours	0 minutes	
Total CP usage.	0 hours	0 minutes	2.0 seconds

J O B I N F O R M A T I O N -----

8-120B Cask 2-d HAC Fire Model

Current jobnamefile
Initial jobnamefile

Unitsunknown

	Available	Used
Scratch Memory Space. . . .	9600.000 mb	517.840 mb (5.4%)
Database space	65535.750 mb	2.748 mb (0.0%)

User menu file in use . . .%ANSYS130_DIR%\gui\en-us\UIDL\UIMENU.GRN
User menu file in use . . .%ANSYS130_DIR%\gui\en-us\UIDL\UIFUNC1.GRN
User menu file in use . . .%ANSYS130_DIR%\gui\en-us\UIDL\UIFUNC2.GRN
User menu file in use . . .%ANSYS130_DIR%\gui\en-us\UIDL\MECHTOOL.AUI
Beta featuresare not shown in the user interface

M O D E L I N F O R M A T I O N -----

Solid model summary:

	Largest Number	Number Defined	Number Selected
Keypoints	101	101	18
Lines	136	136	23
Areas	43	43	6
Volumes	0	0	0

Finite element model summary:

	Largest Number	Number Defined	Number Selected
Nodes	2443	1074	1068
Elements.	1350	1210	823
Element types	65	46	n.a.
Real constant sets.	60	24	n.a.
Material property sets. . . .	14	10	n.a.
Coupling.	2	2	n.a.
Constraint equations.	0	0	n.a.
Master DOFs	0	0	n.a.
Dynamic gap conditions. . . .	0	0	n.a.

B O U N D A R Y C O N D I T I O N I N F O R M A T I O N -----

	Number Defined
Constraints on nodes.	4
Constraints on keypoints.	0
Constraints on lines.	0
Constraints on areas.	0
Forces on nodes	0
Forces on keypoints	0
Surface loads on elements	88
Number of element flagged surfaces	0

Surface loads on lines.	0
Surface loads on areas.	0
Body loads on elements.	0
Body loads on areas	0
Body loads on lines	0
Body loads on nodes	0
Body loads on keypoints	0

Temperatures

Uniform temperature.	70.000
Offset from absolute scale	460.000

	X	Y	Z
Linear acceleration	0.0000	0.0000	0.0000
Angular velocity (about global CS).	0.0000	0.0000	0.0000
Angular acceleration (about global CS).	0.0000	0.0000	0.0000
Location of reference CS.	0.0000	0.0000	0.0000
Angular velocity (about reference CS)	0.0000	0.0000	0.0000
Angular acceleration (about reference CS)	0.0000	0.0000	0.0000

R O U T I N E I N F O R M A T I O N -----

Current routine.Preprocessing (PREP7)

Active coordinate system 0 (Cartesian)

Display coordinate system. 0 (Cartesian)

Current element attributes:

Type number	50	(MATRIX50)
Real number	50	
Material number	8	
Element coordinate system number.	0	

Current mesher type.free mesher

Current element meshing shape 2D . . .use default element shape.

Current element meshing shape 3D . . .use default element shape.

SmrtSize Level OFF

Global element size. 0 divisions per line

Active coordinate system 0 (Cartesian)

Display coordinate system. 0 (Cartesian)

Analysis type.Transient

Active options for this analysis type:

Solution method	Full
Equation solver to use.	Sparse

Results filefile.rth

Load step number 1

Number of substeps:

Starting number of substeps 40000
 Maximum number of substeps. . . . 40000
 Minimum number of substeps. . . . 100
 Step change boundary conditions . .Yes

Analysis Options

New, Restart, or Expansion Pass: NEW ANALYSIS
 Discipline (based on active DOF): THERMAL
 Analysis type: TRANSIENT
 Analysis method: FULL

Newton-Raphson option PROGRAM CHOOSES
 Newton-Raphson adaptive descent DO NOT USE ADAPT DESCENT

Equation solver to be used SPARSE DIRECT SOLVER

Difference (in degrees) between absolute zero and
 the temperature system being used 460.00

LIST ELEMENT TYPES FROM 1 TO 65 BY 1

ELEMENT TYPE	1 IS PLANE55	AXI. THERMAL SOLID
KEYOPT(1- 6)=	0 0	1 0 0 0
KEYOPT(7-12)=	0 0	0 0 0 0
KEYOPT(13-18)=	0 0	0 0 0 0
ELEMENT TYPE	2 IS TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0 0	0 0 0 0
KEYOPT(7-12)=	0 0	0 0 0 0
KEYOPT(13-18)=	0 0	0 0 0 0
ELEMENT TYPE	3 IS CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2 2	0 2 0 0
KEYOPT(7-12)=	0 0	0 2 0 5
KEYOPT(13-18)=	0 0	0 0 0 0
ELEMENT TYPE	4 IS TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0 0	0 0 0 0
KEYOPT(7-12)=	0 0	0 0 0 0
KEYOPT(13-18)=	0 0	0 0 0 0
ELEMENT TYPE	5 IS CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2 0	0 2 0 0
KEYOPT(7-12)=	0 0	0 0 0 0
KEYOPT(13-18)=	0 0	0 0 0 0
ELEMENT TYPE	6 IS TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0 0	0 0 0 0
KEYOPT(7-12)=	0 0	0 0 0 0
KEYOPT(13-18)=	0 0	0 0 0 0

ELEMENT TYPE	7	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 2 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	8	IS	TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	9	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 0 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	10	IS	TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	11	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 0 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	12	IS	TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	13	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 0 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	14	IS	TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	15	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0 2 0 0
KEYOPT(7-12)=	0		0	0 0 0 5
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	16	IS	TARGE169	2-D TARGET SEGMENT
KEYOPT(1- 6)=	0		0	0 0 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	17	IS	CONTA171	2D 2-NODE THERMAL CONTACT
KEYOPT(1- 6)=	2		0	0 2 0 0
KEYOPT(7-12)=	0		0	0 0 0 0
KEYOPT(13-18)=	0		0	0 0 0 0
ELEMENT TYPE	18	IS	TARGE169	2-D TARGET SEGMENT

KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	19	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	5
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	20	IS	TARGE169	2-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	21	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	5
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	22	IS	TARGE169	2-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	23	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	5
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	24	IS	TARGE169	2-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	25	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	5
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	26	IS	TARGE169	2-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	27	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		2	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	5
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	28	IS	TARGE169	2-D	TARGET	SEGMENT
KEYOPT(1- 6)=	0		0	0	0	0
KEYOPT(7-12)=	0		0	0	0	0
KEYOPT(13-18)=	0		0	0	0	0
ELEMENT TYPE	29	IS	CONTA171	2D	2-NODE	THERMAL CONTACT
KEYOPT(1- 6)=	2		0	0	2	0 0
KEYOPT(7-12)=	0		0	0	0	0

KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	49	IS	LINK33	3-D CONDUCTION BAR		
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	50	IS	MATRIX50	SUPERELEMENT (SUBSTRUCTURE)		
KEYOPT(1- 6)=	1	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	51	IS	SURF151	2-D THERMAL SURFACE		
KEYOPT(1- 6)=	0	0	1	1	1	0
KEYOPT(7-12)=	1	4	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	52	IS	SURF151	2-D THERMAL SURFACE		
KEYOPT(1- 6)=	0	0	1	1	1	0
KEYOPT(7-12)=	0	0	1	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	53	IS	SURF151	2-D THERMAL SURFACE		
KEYOPT(1- 6)=	0	0	1	1	1	0
KEYOPT(7-12)=	0	4	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	54	IS	SURF151	2-D THERMAL SURFACE		
KEYOPT(1- 6)=	0	0	1	1	1	0
KEYOPT(7-12)=	0	0	1	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	55	IS	SURF151	2-D THERMAL SURFACE		
KEYOPT(1- 6)=	0	0	1	1	0	0
KEYOPT(7-12)=	0	1	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	56	IS	TARGE169	2-D TARGET SEGMENT		
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	57	IS	CONTA171	2D 2-NODE THERMAL CONTACT		
KEYOPT(1- 6)=	2	0	0	2	0	0
KEYOPT(7-12)=	0	0	0	0	0	5
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	58	IS	TARGE169	2-D TARGET SEGMENT		
KEYOPT(1- 6)=	0	0	0	0	0	0
KEYOPT(7-12)=	0	0	0	0	0	0
KEYOPT(13-18)=	0	0	0	0	0	0
ELEMENT TYPE	59	IS	CONTA171	2D 2-NODE THERMAL CONTACT		
KEYOPT(1- 6)=	2	0	0	2	0	0
KEYOPT(7-12)=	0	0	0	0	0	5
KEYOPT(13-18)=	0	0	0	0	0	0

ELEMENT TYPE	60	IS	TARGE169	2-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0

ELEMENT TYPE	61	IS	CONTA171	2D	2-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 0	0	2	0	0
KEYOPT(7-12)=			0 0	0	0	0	5
KEYOPT(13-18)=			0 0	0	0	0	0

ELEMENT TYPE	62	IS	TARGE169	2-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0

ELEMENT TYPE	63	IS	CONTA171	2D	2-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 0	0	2	0	0
KEYOPT(7-12)=			0 0	0	0	0	5
KEYOPT(13-18)=			0 0	0	0	0	0

ELEMENT TYPE	64	IS	TARGE169	2-D	TARGET	SEGMENT	
KEYOPT(1- 6)=			0 0	0	0	0	0
KEYOPT(7-12)=			0 0	0	0	0	0
KEYOPT(13-18)=			0 0	0	0	0	0

ELEMENT TYPE	65	IS	CONTA171	2D	2-NODE	THERMAL	CONTACT
KEYOPT(1- 6)=			2 0	0	2	0	0
KEYOPT(7-12)=			0 0	0	0	0	5
KEYOPT(13-18)=			0 0	0	0	0	0

CURRENT NODAL DOF SET IS TEMP
THREE-DIMENSIONAL MODEL

LIST MATERIALS 1 TO 14 BY 1
PROPERTY= ALL

MATERIAL NUMBER 1

TEMP	DENS
70.000	0.28180
100.00	0.28180
150.00	0.28180
200.00	0.28180
250.00	0.28180
300.00	0.28180
350.00	0.28180
400.00	0.28180
450.00	0.28180
500.00	0.28180
550.00	0.28180
600.00	0.28180
650.00	0.28180
700.00	0.28180
750.00	0.28180
800.00	0.28180
900.00	0.28180
1000.0	0.28180

1100.0	0.28180
1200.0	0.28180
1300.0	0.28180
1400.0	0.28180
1500.0	0.28180

TEMP	MU
	0.000000

TEMP	KXX
70.000	0.81300E-03
100.00	0.80300E-03
150.00	0.78900E-03
200.00	0.77800E-03
250.00	0.76200E-03
300.00	0.74800E-03
350.00	0.73100E-03
400.00	0.71500E-03
450.00	0.70100E-03
500.00	0.68300E-03
550.00	0.66700E-03
600.00	0.64800E-03
650.00	0.63200E-03
700.00	0.61600E-03
750.00	0.60000E-03
800.00	0.58300E-03
900.00	0.55100E-03
1000.0	0.51900E-03
1100.0	0.48400E-03
1200.0	0.45100E-03
1300.0	0.41700E-03
1400.0	0.38000E-03
1500.0	0.36300E-03

TEMP	C
70.000	0.10400
100.00	0.10600
150.00	0.10900
200.00	0.11300
250.00	0.11500
300.00	0.11800
350.00	0.12200
400.00	0.12400
450.00	0.12600
500.00	0.12800
550.00	0.13100
600.00	0.13300
650.00	0.13500
700.00	0.13900
750.00	0.14200
800.00	0.14600
900.00	0.15400
1000.0	0.16300
1100.0	0.17200
1200.0	0.18400
1300.0	0.20500
1400.0	0.41100

1500.0	0.19900
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TEMP	EMIS
	0.000000

MATERIAL NUMBER	2
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TEMP	DENS
70.000	0.28240
100.00	0.28240
150.00	0.28240
200.00	0.28240
250.00	0.28240
300.00	0.28240
350.00	0.28240
400.00	0.28240
450.00	0.28240
500.00	0.28240
550.00	0.28240
600.00	0.28240
650.00	0.28240
700.00	0.28240
750.00	0.28240
800.00	0.28240
900.00	0.28240
1000.0	0.28240
1100.0	0.28240
1200.0	0.28240
1300.0	0.28240
1400.0	0.28240
1500.0	0.28240

TEMP	MU
	0.000000

TEMP	KXX
70.000	0.19900E-03
100.00	0.20100E-03
150.00	0.20800E-03
200.00	0.21500E-03
250.00	0.22200E-03
300.00	0.22700E-03
350.00	0.23400E-03
400.00	0.24100E-03
450.00	0.24500E-03
500.00	0.25200E-03
550.00	0.25700E-03
600.00	0.26200E-03
650.00	0.26900E-03
700.00	0.27300E-03
750.00	0.27800E-03
800.00	0.28200E-03
900.00	0.29400E-03
1000.0	0.30600E-03
1100.0	0.31500E-03
1200.0	0.32400E-03
1300.0	0.33600E-03

1400.0	0.34500E-03
1500.0	0.35400E-03

TEMP	C
70.000	0.11700
100.00	0.11700
150.00	0.12000
200.00	0.12200
250.00	0.12500
300.00	0.12600
350.00	0.12800
400.00	0.12900
450.00	0.13000
500.00	0.13100
550.00	0.13200
600.00	0.13300
650.00	0.13400
700.00	0.13500
750.00	0.13600
800.00	0.13600
900.00	0.13800
1000.0	0.13900
1100.0	0.14100
1200.0	0.14100
1300.0	0.14300
1400.0	0.14400
1500.0	0.14500

MATERIAL NUMBER	3
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TEMP	DENS
70.000	0.41090
100.00	0.41090
150.00	0.41090
200.00	0.41090
250.00	0.41090
300.00	0.41090
350.00	0.41090
400.00	0.41090
450.00	0.41090
500.00	0.41090
550.00	0.41090
600.00	0.41090
650.00	0.41090
700.00	0.41090
750.00	0.41090
800.00	0.41090
900.00	0.41090
1000.0	0.41090
1100.0	0.41090
1200.0	0.41090
1300.0	0.41090
1400.0	0.41090
1500.0	0.41090

TEMP	KXX
70.000	0.46500E-03

100.00	0.46100E-03
150.00	0.45500E-03
200.00	0.44800E-03
250.00	0.44100E-03
300.00	0.43500E-03
350.00	0.42800E-03
400.00	0.42200E-03
450.00	0.41500E-03
500.00	0.40900E-03
550.00	0.40200E-03
600.00	0.39500E-03
650.00	0.38900E-03
700.00	0.38900E-03
750.00	0.38900E-03
800.00	0.38900E-03
900.00	0.38900E-03
1000.0	0.38900E-03
1100.0	0.38900E-03
1200.0	0.38900E-03
1300.0	0.38900E-03
1400.0	0.38900E-03
1500.0	0.38900E-03

TEMP	C
70.000	0.31050E-01
100.00	0.31050E-01
150.00	0.31050E-01
200.00	0.31050E-01
250.00	0.31050E-01
300.00	0.31050E-01
350.00	0.31050E-01
400.00	0.31050E-01
450.00	0.31050E-01
500.00	0.31050E-01
550.00	0.31050E-01
600.00	0.31050E-01
650.00	0.31050E-01
700.00	0.31050E-01
750.00	0.31050E-01
800.00	0.31050E-01
900.00	0.31050E-01
1000.0	0.31050E-01
1100.0	0.31050E-01
1200.0	0.31050E-01
1300.0	0.31050E-01
1400.0	0.31050E-01
1500.0	0.31050E-01

MATERIAL NUMBER	4
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TEMP	DENS
70.000	0.43510E-04
100.00	0.41120E-04
150.00	0.37520E-04
200.00	0.34680E-04
250.00	0.32360E-04
300.00	0.30310E-04

350.00	0.28310E-04
400.00	0.26730E-04
450.00	0.25220E-04
500.00	0.23960E-04
550.00	0.22780E-04
600.00	0.21680E-04
650.00	0.20710E-04
700.00	0.19800E-04
750.00	0.18980E-04
800.00	0.18180E-04
900.00	0.16900E-04
1000.0	0.15710E-04
1100.0	0.14720E-04
1200.0	0.13850E-04
1300.0	0.13040E-04
1400.0	0.12350E-04
1500.0	0.11710E-04

TEMP	KXX
70.000	0.34490E-06
100.00	0.36210E-06
150.00	0.39030E-06
200.00	0.41770E-06
250.00	0.44460E-06
300.00	0.47040E-06
350.00	0.49570E-06
400.00	0.52040E-06
450.00	0.54480E-06
500.00	0.56880E-06
550.00	0.59210E-06
600.00	0.61430E-06
650.00	0.63630E-06
700.00	0.65810E-06
750.00	0.67900E-06
800.00	0.69960E-06
900.00	0.74090E-06
1000.0	0.78040E-06
1100.0	0.81750E-06
1200.0	0.85450E-06
1300.0	0.88970E-06
1400.0	0.92850E-06
1500.0	0.97070E-06

TEMP	C
70.000	0.24020
100.00	0.24040
150.00	0.24080
200.00	0.24140
250.00	0.24210
300.00	0.24290
350.00	0.24380
400.00	0.24500
450.00	0.24610
500.00	0.24740
550.00	0.24900
600.00	0.25110
650.00	0.25270

Stiffness matrix reuse options
 Step boundary condition
 Reference temperature

PROGRAM DECIDES
 STEP BOUNDARY KEY = 1 (STEP)
 TREF= 70.000

*** NOTE ***

CP = 1.981 TIME= 10:02:49

No nodal body forces to list.

Inertia load options

Acceleration vector GLOBAL CARTESIAN COMPONENTS ARE:
 0.0000 0.0000 0.0000
 Angular velocity vector GLOBAL CARTESIAN COMPONENTS ARE:
 0.0000 0.0000 0.0000
 SPIN SOFTENING NOT ACTIVATED
 Angular acceleration vector GLOBAL CARTESIAN COMPONENTS ARE:
 0.0000 0.0000 0.0000
 Reference coord. system origin ORIGIN = 0.0000 0.0000 0.0000
 Angular velocity vector REFERENCE COORDINATE COMPONENTS ARE:
 0.0000 0.0000 0.0000
 Angular acceleration vector REFERENCE COORDINATE COMPONENTS ARE:
 0.0000 0.0000 0.0000
 Inertia relief NO INERTIA RELIEF
 Translational acceleration vector on components NONE
 Angular velocity vector on components NONE
 Angular acceleration vector on components NONE

LIST CONSTRAINTS FOR SELECTED NODES 1 TO 2443 BY 1
 CURRENTLY SELECTED DOF SET= TEMP

*** NOTE ***

CP = 1.981 TIME= 10:02:49

No nodal forces to list.

LIST ELEM PRESS FOR SELECTED ELEMENTS IN RANGE 1 TO 1350 BY 1

LIST NODAL SURFACE LOAD HFLU FOR ALL SELECTED NODES

ELEMENT	LKEY	FACE NODES	HEAT FLUX
740	3	2013	4.163000000E-05
		2014	4.163000000E-05
741	3	2014	4.163000000E-05
		2015	4.163000000E-05
742	3	2015	4.163000000E-05
		2016	4.163000000E-05
743	3	2016	4.163000000E-05
		2017	4.163000000E-05
744	3	2017	4.163000000E-05
		2018	4.163000000E-05
745	3	2018	4.163000000E-05
		2019	4.163000000E-05
746	3	2019	4.163000000E-05
		2020	4.163000000E-05
747	3	2020	4.163000000E-05
		2012	4.163000000E-05
749	4	2012	4.163000000E-05
		2046	4.163000000E-05
750	4	2046	4.163000000E-05

2045 4.163000000E-05

ELEMENT	LKEY	FACE NODES	HEAT FLUX
751	4	2045	4.163000000E-05
		2044	4.163000000E-05
752	4	2044	4.163000000E-05
		2043	4.163000000E-05
753	4	2043	4.163000000E-05
		2042	4.163000000E-05
754	4	2042	4.163000000E-05
		2041	4.163000000E-05
755	4	2041	4.163000000E-05
		2040	4.163000000E-05
756	4	2040	4.163000000E-05
		2039	4.163000000E-05
757	4	2039	4.163000000E-05
		2038	4.163000000E-05
758	4	2038	4.163000000E-05
		2037	4.163000000E-05
759	4	2037	4.163000000E-05
		2036	4.163000000E-05
760	4	2036	4.163000000E-05
		2035	4.163000000E-05

ELEMENT	LKEY	FACE NODES	HEAT FLUX
762	1	2050	4.163000000E-05
		2049	4.163000000E-05
763	1	2051	4.163000000E-05
		2050	4.163000000E-05
764	1	2052	4.163000000E-05
		2051	4.163000000E-05
765	1	2053	4.163000000E-05
		2052	4.163000000E-05
766	1	2054	4.163000000E-05
		2053	4.163000000E-05
767	1	2055	4.163000000E-05
		2054	4.163000000E-05
768	1	2056	4.163000000E-05
		2055	4.163000000E-05
769	1	2035	4.163000000E-05
		2056	4.163000000E-05

Appendix 2

Print-out of the Results

(2 Pages)

ANSYS Finite Element Model Partial Printout

3-d Finite Element Model Results

(The complete print-out of the time-history of the following nodal temperatures is available on the file "3d model\prvar1.lis" included on the DVD in Appendix 3)

POST26 SUMMARY OF VARIABLE EXTREME VALUES

NAME	ELEMENT	NODE	MINIMUM	AT TIME	MAXIMUM	AT TIME
FireShld		42910	150.332	1	1392.01	1800
Oshell0		12531	154.965	1	464.422	1800.3
Lead0		14317	154.688	1	295.127	4461.67
LeadI		14338	154.763	1	295.845	4461.67
IshellI		8015	154.767	8.196	295.455	4461.67
PSeal		25430	155.938	1	212.385	18225.1

(The complete print-out of the time-history of the following nodal temperatures is available on the file "3d model\prvar2.lis" included on the DVD in Appendix 3)

POST26 SUMMARY OF VARIABLE EXTREME VALUES

NAME	ELEMENT	NODE	MINIMUM	AT TIME	MAXIMUM	AT TIME
SSeal		37678	156.538	594.67	202.867	24000
LidCenter		27023	156.94	1566.13	192.57	24000
BaseCente		2430	156.715	936.48	206.255	24000

(The complete print-out of the time-history of the following nodal temperatures is available on the file "3d model\prvar3.lis" included on the DVD in Appendix 3)

POST26 SUMMARY OF VARIABLE EXTREME VALUES

NAME	ELEMENT	NODE	MINIMUM	AT TIME	MAXIMUM	AT TIME
PLid		37675	156.552	612.66	202.876	24000
VSeal		34802	156.211	1	206.878	24000
IL		27594	156.28	1	205.084	24000

2-d Finite Element Model HAC fire Results

(The complete print-out of the time-history of the following nodal temperatures is available on the file "2d model\prvar1.lis" included on the DVD in Appendix 3)

POST26 SUMMARY OF VARIABLE EXTREME VALUES

NAME	ELEMENT	NODE	MINIMUM	AT TIME	MAXIMUM	AT TIME
cav1		2057	197.591	54.97	239.047	40289
cav2		2023	197.282	54.97	238.819	38387
cav3		2029	197.865	72.96	239.629	38387
cav4		2021	197.301	54.97	239.212	38387
cav5		2003	197.694	54.97	239.622	40289
cav6		2098	160.971	4.598	302.278	3643.22

(The complete print-out of the time-history of the following nodal temperatures is available on the file “2d model\prvar2.lis” included on the DVD in Appendix 3)

POST26 SUMMARY OF VARIABLE EXTREME VALUES

NAME	ELEMENT	NODE	MINIMUM	AT TIME	MAXIMUM	AT TIME
cont1		2049	197.642	72.96	239.099	40289
cont2		2035	197.329	72.96	238.864	38387
cont3		2041	197.915	72.96	239.679	38387
cont4		2012	197.405	72.96	239.314	38387
cont5		2013	197.745	72.96	239.673	40289

Appendix 3

(Electronic Data - 1 DVD)

Volume in drive E is TH-028

Volume Serial Number is 3C05-E9B0

Directory of E:\

11/08/2011	01:49 PM	<DIR>	2-D Model
11/08/2011	01:52 PM	<DIR>	3-D Model
	0 File(s)		0 bytes

Directory of E:\2-D Model

11/08/2011	01:49 PM	<DIR>	.
11/08/2011	02:28 PM	<DIR>	..
11/08/2011	02:03 PM	<DIR>	Fire
11/08/2011	02:04 PM	<DIR>	Puncture
	0 File(s)		0 bytes

Directory of E:\2-D Model\Fire

11/08/2011	02:03 PM	<DIR>	.
11/08/2011	01:49 PM	<DIR>	..
10/31/2011	12:16 PM		65,536 annulus-sub.sub
10/31/2011	12:16 PM		59,964 ansub.out
10/31/2011	03:06 PM		131,072 cavity-sub.sub
10/31/2011	03:06 PM		337,229 cavsub.out
11/08/2011	09:27 AM		596 EXTREME-1.lis
11/08/2011	09:27 AM		518 EXTREME-2.lis
11/08/2011	11:06 AM		5,111,808 file.db
11/08/2011	11:06 AM		175,308,800 file.rth
11/08/2011	09:29 AM		20,637 file000.png
11/08/2011	09:32 AM		27,119 file001.png
11/08/2011	10:59 AM		179,461 model.out
11/08/2011	09:27 AM		27,224 PRVAR-1.lis
11/08/2011	09:28 AM		23,542 PRVAR-2.lis
	13 File(s)		181,293,506 bytes

Directory of E:\2-D Model\Puncture

11/08/2011	02:04 PM	<DIR>	.
11/08/2011	01:49 PM	<DIR>	..
10/31/2011	12:16 PM		65,536 annulus-sub.sub
10/31/2011	12:16 PM		59,964 ansub.out
10/31/2011	03:06 PM		131,072 cavity-sub.sub
10/31/2011	03:06 PM		337,229 cavsub.out
11/08/2011	09:54 AM		206 EXTREME.lis
11/08/2011	11:08 AM		5,111,808 file.db
11/08/2011	11:08 AM		173,867,008 file.rth
11/08/2011	09:52 AM		28,140 file001.png
11/08/2011	09:57 AM		21,603 file002.png
11/08/2011	10:59 AM		208,098 model.out
11/08/2011	09:54 AM		8,752 PRVAR.lis

11 File(s) 179,839,416 bytes

Directory of E:\3-D Model

11/08/2011	01:52 PM	<DIR>	.
11/08/2011	02:28 PM	<DIR>	..
07/23/2010	04:55 PM		30,474,240 annulus15.sub
11/29/2010	04:20 PM		596 EXTREME1.lis
11/29/2010	04:21 PM		364 EXTREME2.lis
06/07/2011	01:37 PM		362 EXTREME3.lis
11/30/2010	09:15 AM		65,339,392 file.db
08/10/2010	03:42 PM		143,251 file001.png
07/26/2010	01:44 PM		28,688 file002.png
07/26/2010	01:44 PM		24,448 file003.png
11/29/2010	05:13 PM		17,017 file004.png
11/30/2010	08:49 AM		152,049 file005.png
11/30/2010	09:05 AM		151,666 file006.png
11/30/2010	09:05 AM		152,857 file007.png
11/30/2010	09:05 AM		167,156 file008.png
11/30/2010	09:06 AM		167,539 file009.png
11/30/2010	09:06 AM		164,579 file010.png
11/30/2010	09:07 AM		157,256 file011.png
11/01/2010	10:42 AM		2,893,815 model.out
11/29/2010	04:18 PM		29,960 PRVAR1.lis
11/29/2010	04:19 PM		18,606 PRVAR2.lis
06/07/2011	01:37 PM		17,822 PRVAR3.lis
		20 File(s)	100,101,663 bytes

Total Files Listed:

44 File(s)	461,234,585 bytes
12 Dir(s)	0 bytes free