

Primary Water Stress Corrosion Crack Growth Analysis - OD Surface Flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesan

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM -"28" Degree Nozzle, Downhill Azimuth,
1.544" above Nozzle Bottom

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " -- between 1.0 and 300.0

Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.

The correction is applied in the determination of the crack extension to obtain the value in inch/hr .

OD Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

Ref_{Point} := 1.544

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)
- 2) The Center of the flaw at the reference point (Enter 2)
- 3) The lower "C- tip" located at the reference point (Enter 3).

Val := 2

Upper Limit to be selected for stress distribution (e.g. Weld bottom). This is the elevation from Nozzle Bottom. Enter this value below

UL_{Strs.Dist} := 1.704

Upper Axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.32$	Initial Flaw Length
$a_0 := 0.661 \cdot 0.12$	Initial Flaw Depth
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
$Years := 4$	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth {MRP}
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left\lfloor \frac{I_{lim}}{50} \right\rfloor \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:
 Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)
 Column "1" = ID Stress data at each Elevation (ksi)
 Column "2" = Quarter Thickness Stress data at each Elevation (ksi)
 Column "3" = Mid Thickness Stress data at each Elevation (ksi)
 Column "4" = Three Quarter Thickness Stress data at each Elevation (ksi)
 Column "5" = OD Stress data at each Elevation (ksi)

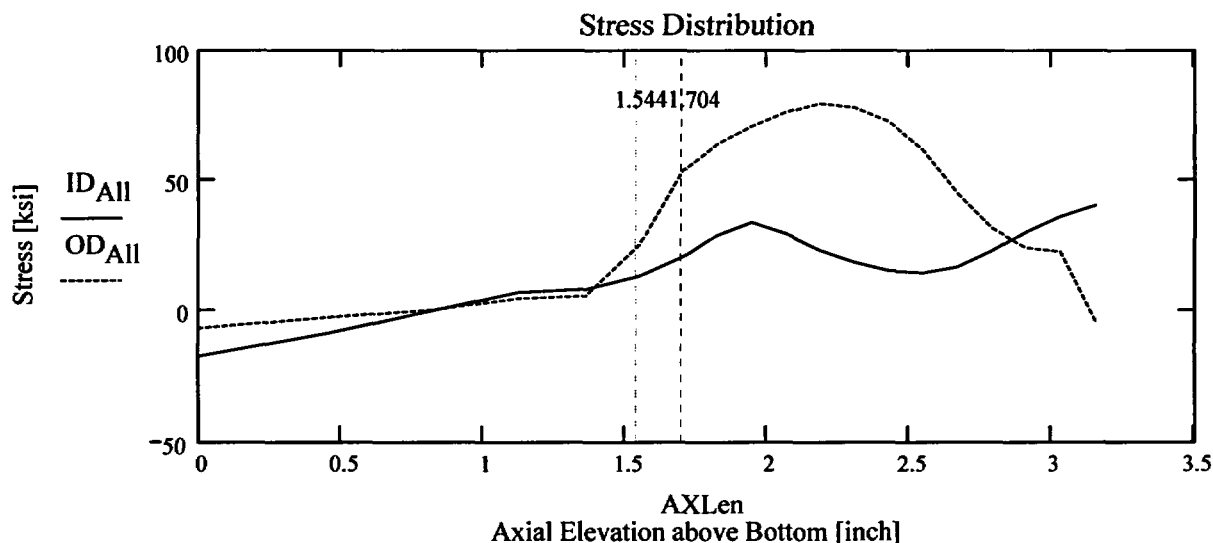
AllData :=

	0	1	2	3	4	5
0	0	-17.41	-13.55	-11.11	-8.88	-6.63
1	0.46	-8.49	-6.31	-4.92	-3.71	-2.54
2	0.83	0.09	0.18	0.11	0.19	0.28
3	1.13	7.03	6.95	6.31	5.21	4.65
4	1.36	8.22	10.95	10.85	9.51	5.65
5	1.55	13.27	16.41	16.06	17.13	25.26
6	1.7	20.63	22.24	25.41	43.58	53.78
7	1.83	29.04	28.83	31.29	53.55	64.08
8	1.95	33.95	30.93	36.41	61.6	71.01
9	2.07	29.59	31.79	40.54	64.61	76.42
10	2.19	23.26	29.74	41.2	64.19	79.63

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

$$Data := \begin{pmatrix} 0 & -17.414 & -13.552 & -11.113 & -8.884 & -6.628 \\ 0.461 & -8.494 & -6.31 & -4.924 & -3.706 & -2.541 \\ 0.83 & 0.089 & 0.179 & 0.11 & 0.186 & 0.284 \\ 1.126 & 7.025 & 6.953 & 6.314 & 5.208 & 4.646 \\ 1.363 & 8.215 & 10.954 & 10.85 & 9.512 & 5.646 \\ 1.552 & 13.266 & 16.41 & 16.061 & 17.131 & 25.256 \\ 1.704 & 20.627 & 22.237 & 25.413 & 43.58 & 53.784 \\ 1.825 & 29.036 & 28.83 & 31.285 & 53.547 & 64.082 \\ 1.946 & 33.945 & 30.929 & 36.407 & 61.6 & 71.01 \\ 2.066 & 29.591 & 31.788 & 40.536 & 64.612 & 76.418 \\ 2.187 & 23.26 & 29.738 & 41.2 & 64.193 & 79.626 \end{pmatrix}$$

$$Axl := Data^{(0)} \quad MD := Data^{(3)} \quad ID := Data^{(1)} \quad TQ := Data^{(4)} \quad QT := Data^{(2)} \quad OD := Data^{(5)}$$

$$R_{ID} := \text{regress}(Axl, ID, 3)$$

$$R_{QT} := \text{regress}(Axl, QT, 3)$$

$$R_{OD} := \text{regress}(Axl, OD, 3)$$

$$R_{MD} := \text{regress}(Axl, MD, 3)$$

$$R_{TQ} := \text{regress}(Axl, TQ, 3)$$


$$FL_{Cntr} := \begin{cases} Ref_{Point} - c_0 & \text{if } Val = 1 \\ Ref_{Point} & \text{if } Val = 2 \\ Ref_{Point} + c_0 & \text{otherwise} \end{cases}$$

Flaw center Location Location above Nozzle Bottom

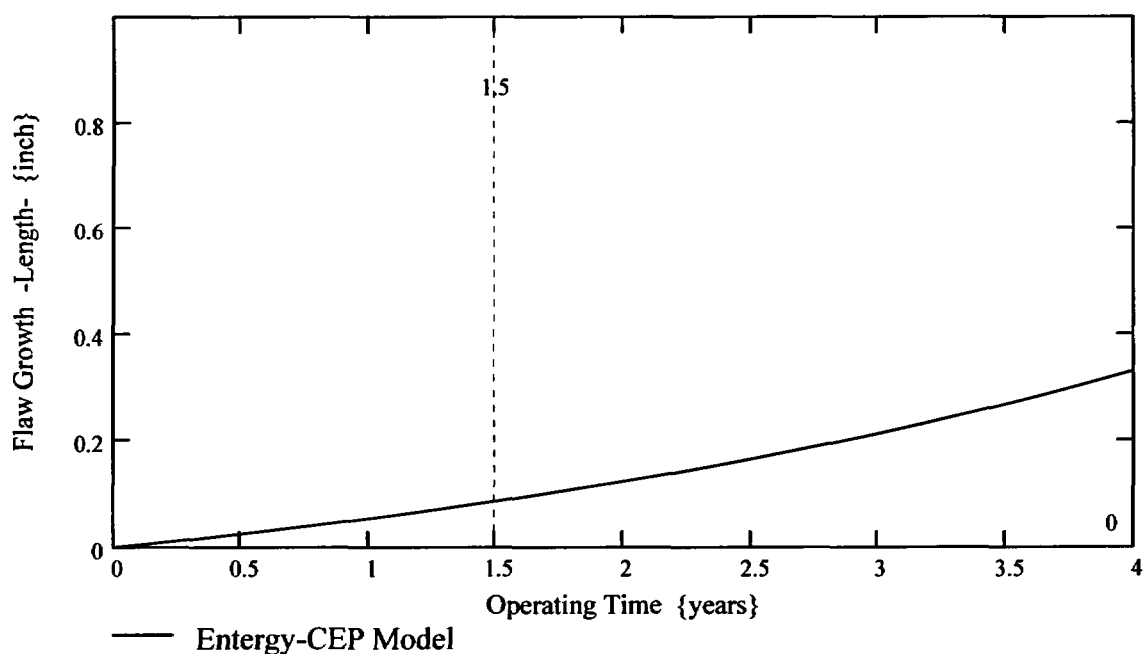
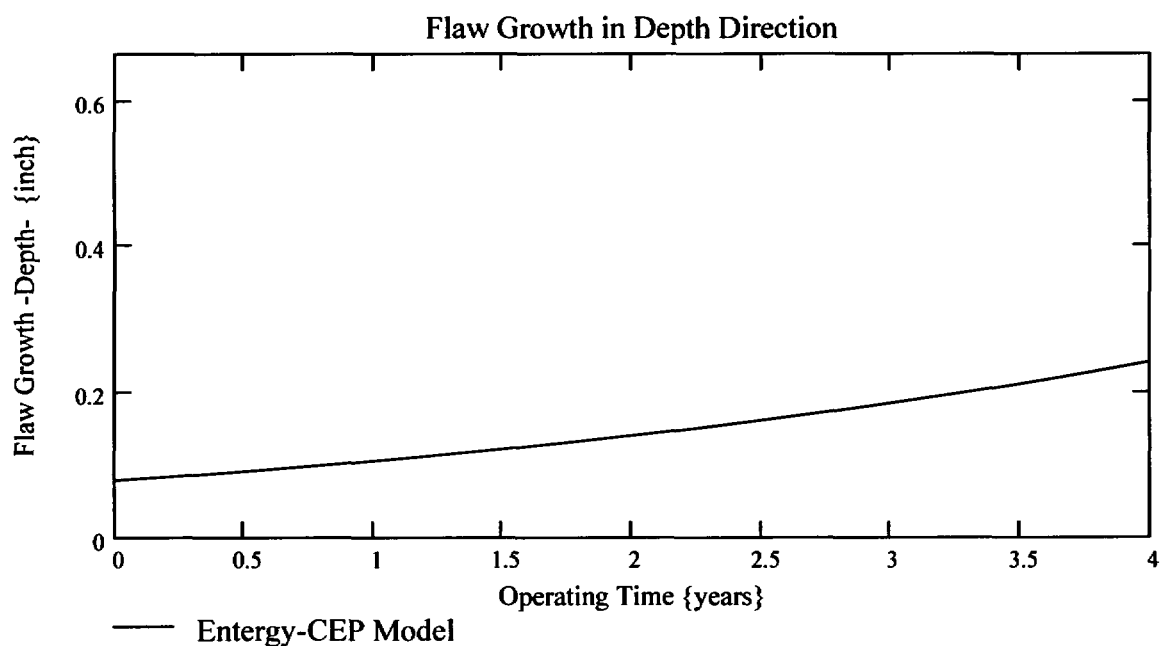
$$U_{Tip} := FL_{Cntr} + c_0$$

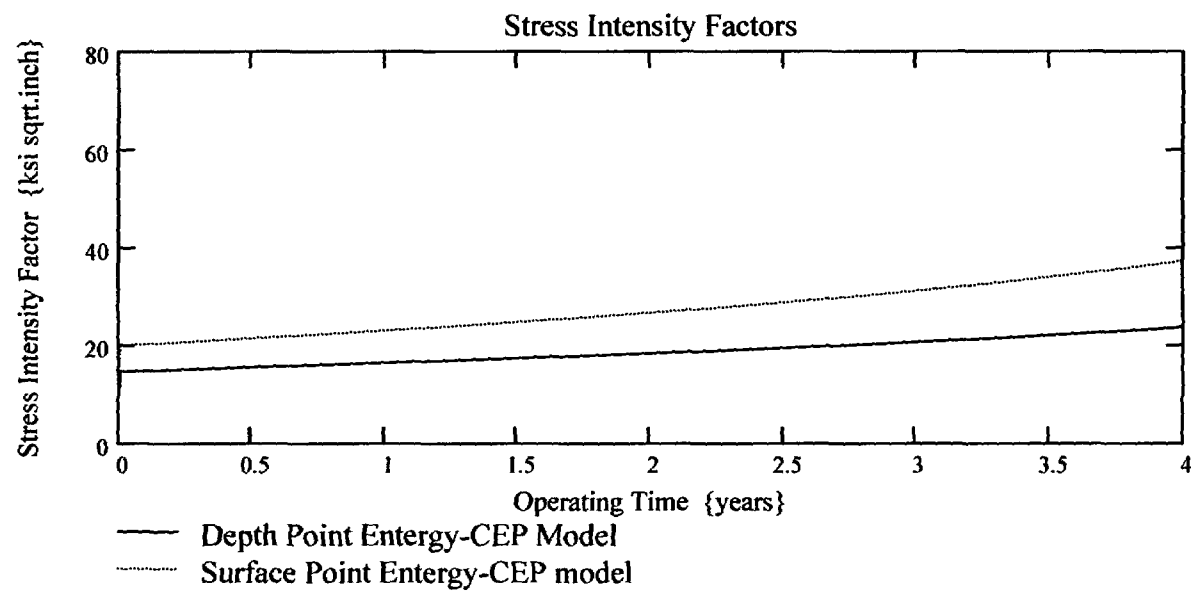
$$IncStrs.avg := \frac{UL_{Strs}.Dist - U_{Tip}}{20}$$

No User Input is required beyond this Point

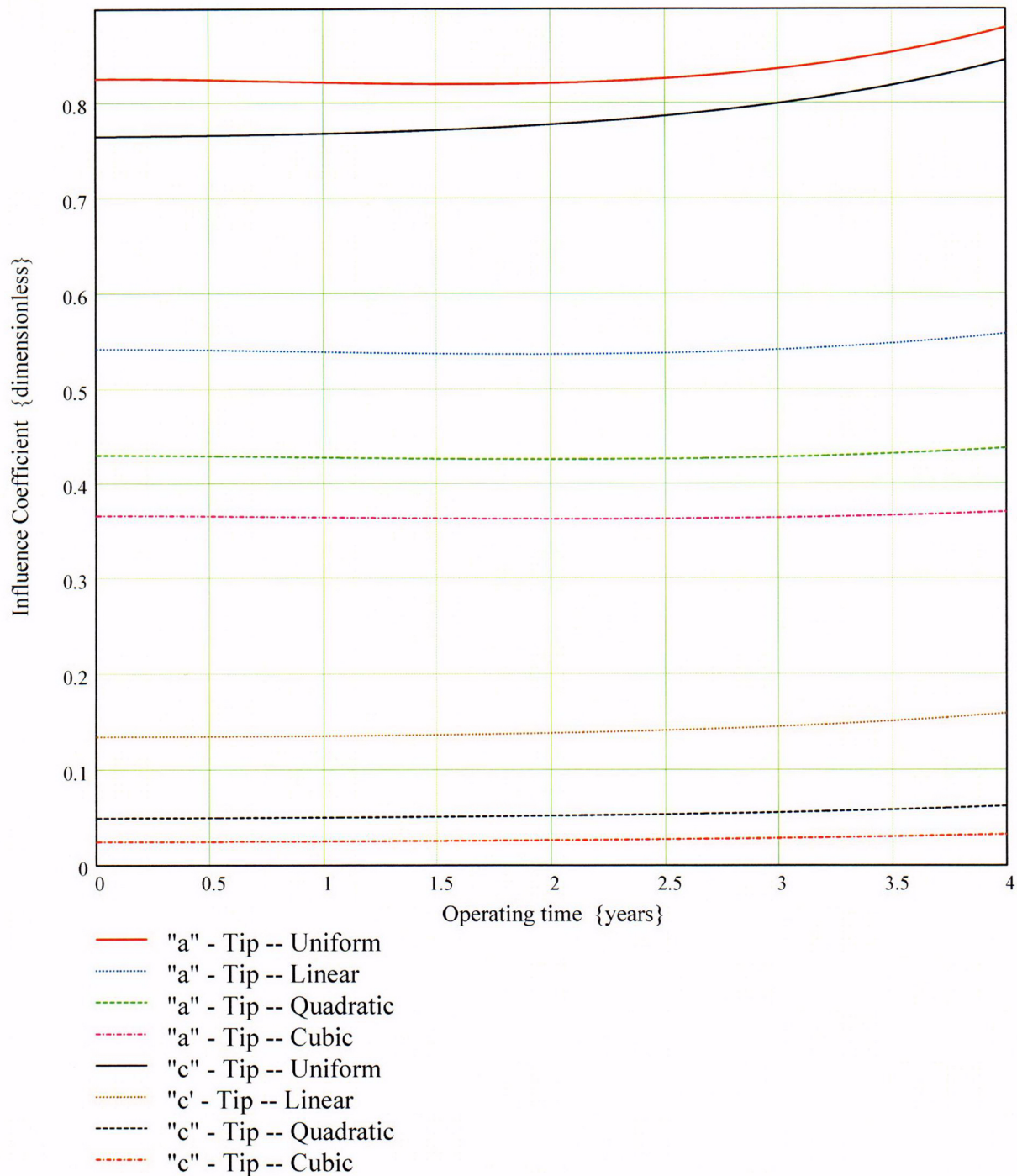
 Sat Aug 09 10:21:18 AM 2003

PropLength = 0





Influence Coefficients - Flaw



$$CGR_{sambi(k, 8)} =$$

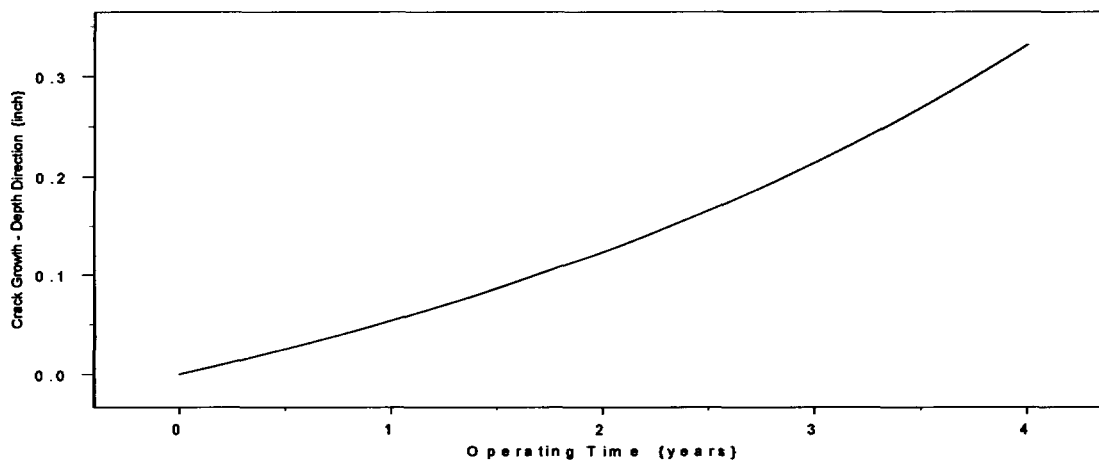
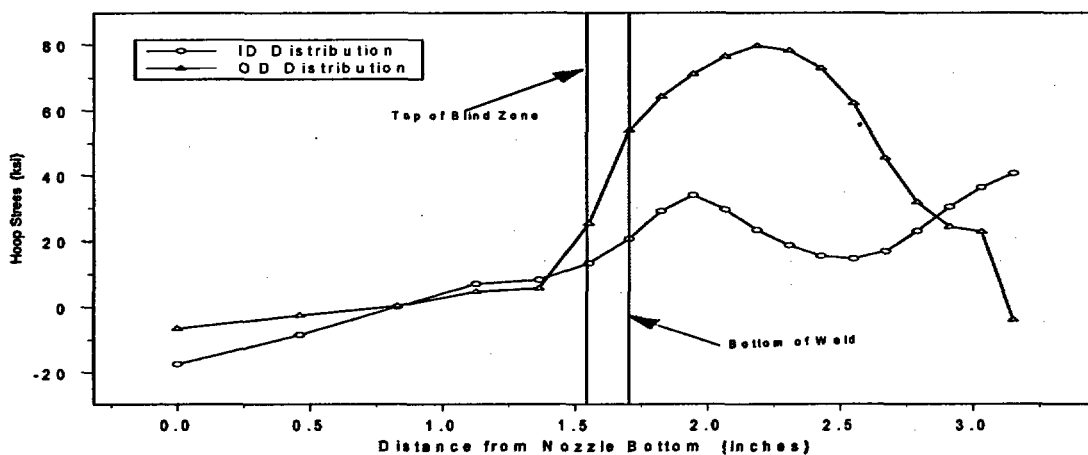
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827

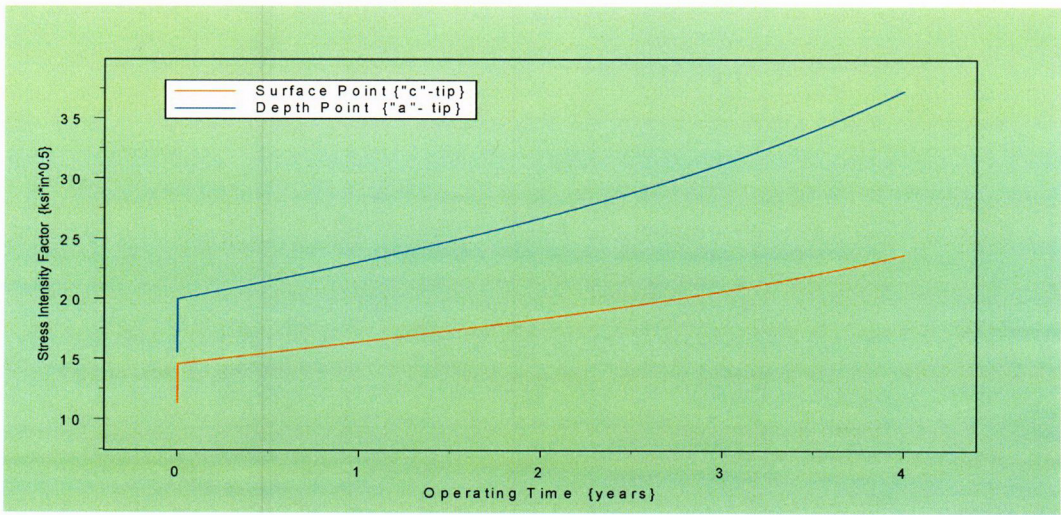
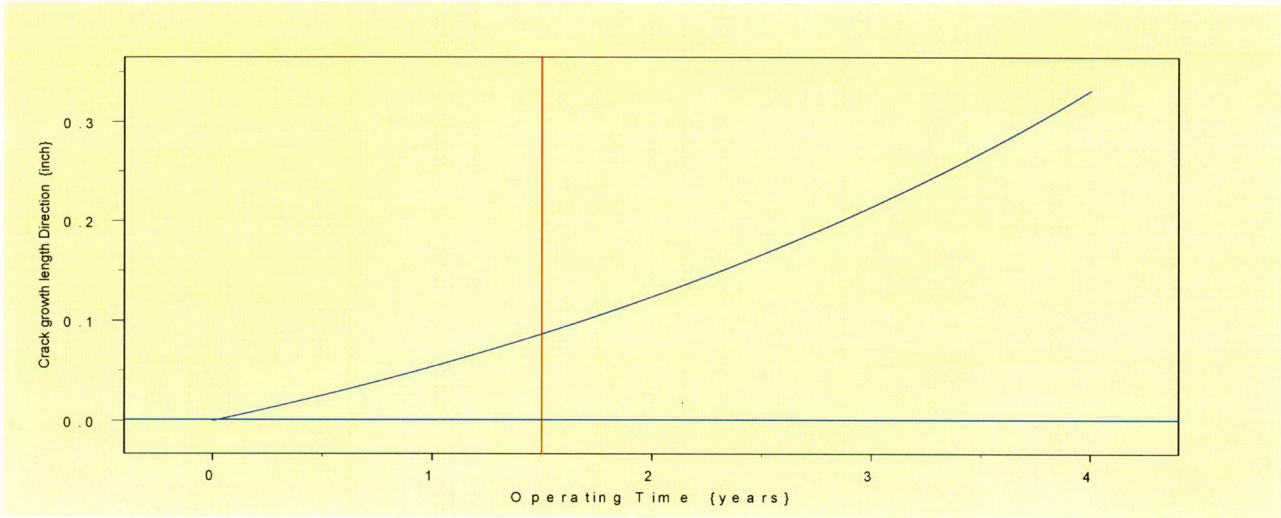
$$CGR_{sambi(k, 6)} =$$

15.529
19.971
19.978
19.986
19.994
20.002
20.009
20.017
20.025
20.033
20.04
20.048
20.056
20.064
20.072
20.079

$$CGR_{sambi(k, 5)} =$$

11.322
14.55
14.555
14.56
14.565
14.57
14.575
14.58
14.585
14.59
14.595
14.6
14.605
14.61
14.615
14.62





Stress Corrosion Crack Growth Analysis Throughwall flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesar

Verified by: B. C. Gray

Note : Only for use when $R_{outside}/t$ is between 2.0 and 5.0 (Thickwall Cylinder)

References :

- 1) ASME PVP paper PVP-350, Page 143; 1997 (Fracture Mechanics Model)
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM - "28" Degree Nozzle, Downhill Azimuth,
1.544 inch above Nozzle Bottom

Calculation Reference: MRP 75 th Percentile and Flaw Pressurized

Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.

The correction is applied in the determination of the crack extension to obtain the value in inch/hr.

Through Wall Axial Flaw

The first Input is to locate the Reference Line (eg. top of the Blind Zone). The throughwall flaw "Upper Tip" is located at the Reference Line.

Enter the elevation of the Reference Line (eg. Blind Zone) above the nozzle bottom in inches.

BZ := 1.544

Location of Blind Zone above nozzle bottom (inch)

The Second Input is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL-Strs.Dist := 1.704

Upper axial Extent for Stress Distribution to be used in the analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.25$ Initial Flaw Length TW axial (Based on 10 Ksi average stress)

$od := 4.05$ Tube OD

$id := 2.728$ Tube ID

$P_{Int} := 2.235$ Design Operating Pressure (internal)

$Years := 4$ Number of Operating Years

$l_{lim} := 1500$ Iteration limit for Crack Growth loop

$T := 604$ Estimate of Operating Temperature

$\nu := 0.307$ Poissons ratio @ 600 F

$\alpha_{0c} := 2.67 \cdot 10^{-12}$ Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F

$Q_g := 31.0$ Thermal activation Energy for Crack Growth {MRP}

$T_{ref} := 617$ Reference Temperature for normalizing Data deg. F

$$C_0 := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3} \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right)} \right]} \cdot \alpha_{0c}$$

$$Tim_{opr} := Years \cdot 365 \cdot 24$$

$$R_o := \frac{od}{2}$$

$$R_i := \frac{id}{2}$$

$$t := R_o - R_i$$

$$R_m := R_i + \frac{t}{2}$$

$$CF_{inhr} := 1.417 \cdot 10^5$$

$$C_{blk} := \frac{Tim_{opr}}{l_{lim}}$$

$$Prnt_{blk} := \left\lfloor \frac{l_{lim}}{50} \right\rfloor$$

$$l := \frac{L}{2}$$

Stress Distribution in the tube. The outside surface is the reference surface for all analysis in accordance with the reference.

Stress Input Data

Import the Required data from applicable Excel spread Sheet. The column designations are as follows:

Cloumn "0" = Axial distance from Minimum to Maximum recorded on the data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

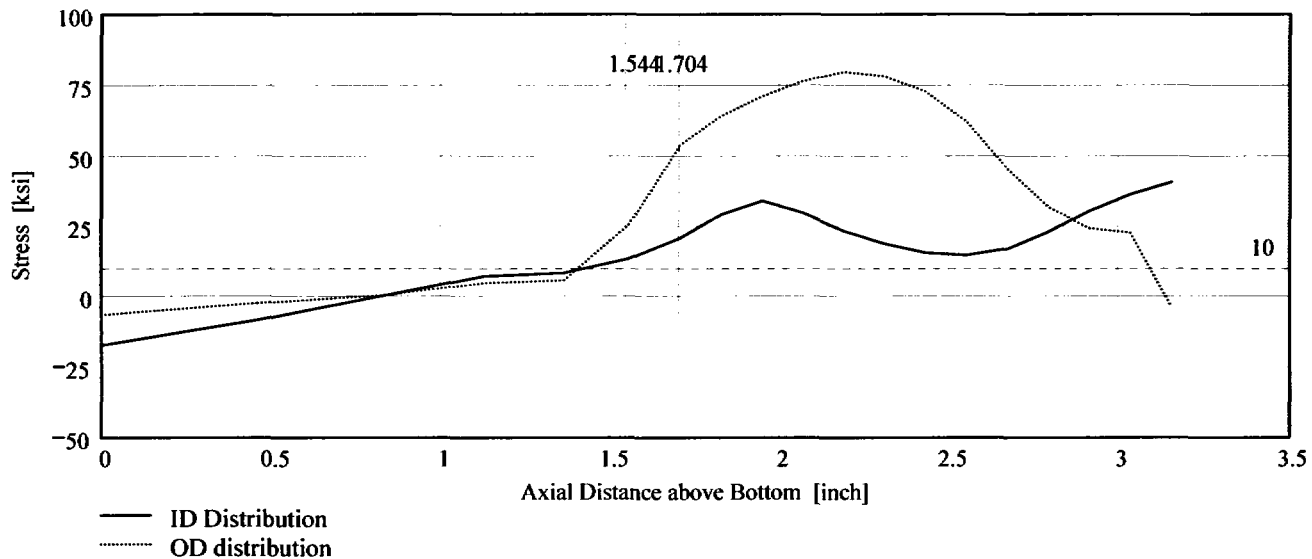
DataAll :=

	0	1	2	3	4	5
0	0	-17.41	-13.55	-11.11	-8.88	-6.63
1	0.46	-8.49	-6.31	-4.92	-3.71	-2.54
2	0.83	0.09	0.18	0.11	0.19	0.28
3	1.13	7.03	6.95	6.31	5.21	4.65
4	1.36	8.22	10.95	10.85	9.51	5.65
5	1.55	13.27	16.41	16.06	17.13	25.26
6	1.7	20.63	22.24	25.41	43.58	53.78
7	1.83	29.04	28.83	31.29	53.55	64.08
8	1.95	33.95	30.93	36.41	61.6	71.01
9	2.07	29.59	31.79	40.54	64.61	76.42
10	2.19	23.26	29.74	41.2	64.19	79.63
11	2.31	18.69	27.73	41.29	61.78	78.12

AllAxI := DataAll⁽⁰⁾

AllID := DataAll⁽¹⁾

AllOD := DataAll⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

Data :=

0	-17.414	-13.552	-11.113	-8.884	-6.628
0.461	-8.494	-6.31	-4.924	-3.706	-2.541
0.83	0.089	0.179	0.11	0.186	0.284
1.126	7.025	6.953	6.314	5.208	4.646
1.363	8.215	10.954	10.85	9.512	5.646
1.552	13.266	16.41	16.061	17.131	25.256
1.704	20.627	22.237	25.413	43.58	53.784
1.825	29.036	28.83	31.285	53.547	64.082
1.946	33.945	30.929	36.407	61.6	71.01
2.066	29.591	31.788	40.536	64.612	76.418
2.187	23.26	29.738	41.2	64.193	79.626
2.308	18.689	27.734	41.29	61.777	78.117

$Ax1 := Data_{(0)}$

$ID := Data_{(1)}$

$OD := Data_{(5)}$

$R_{ID} := \text{regress}(Ax1, ID, 3)$


$R_{OD} := \text{regress}(Ax1, OD, 3)$

$FL_{Cntr} := BZ - 1$

Flaw Center above Nozzle Bottom

$$IncStrs.avg := \frac{ULStrs.Dist - BZ}{20}$$

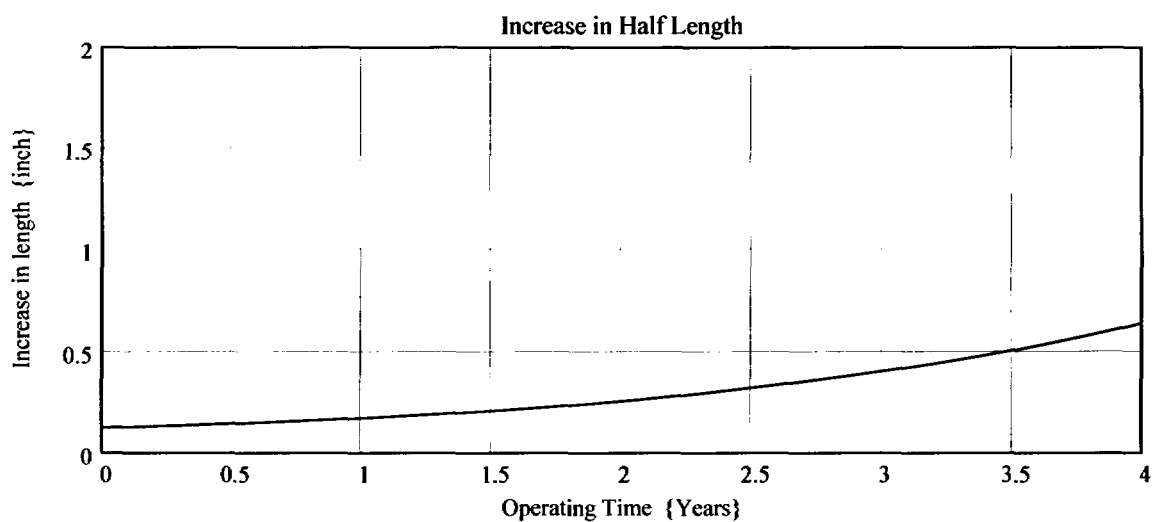
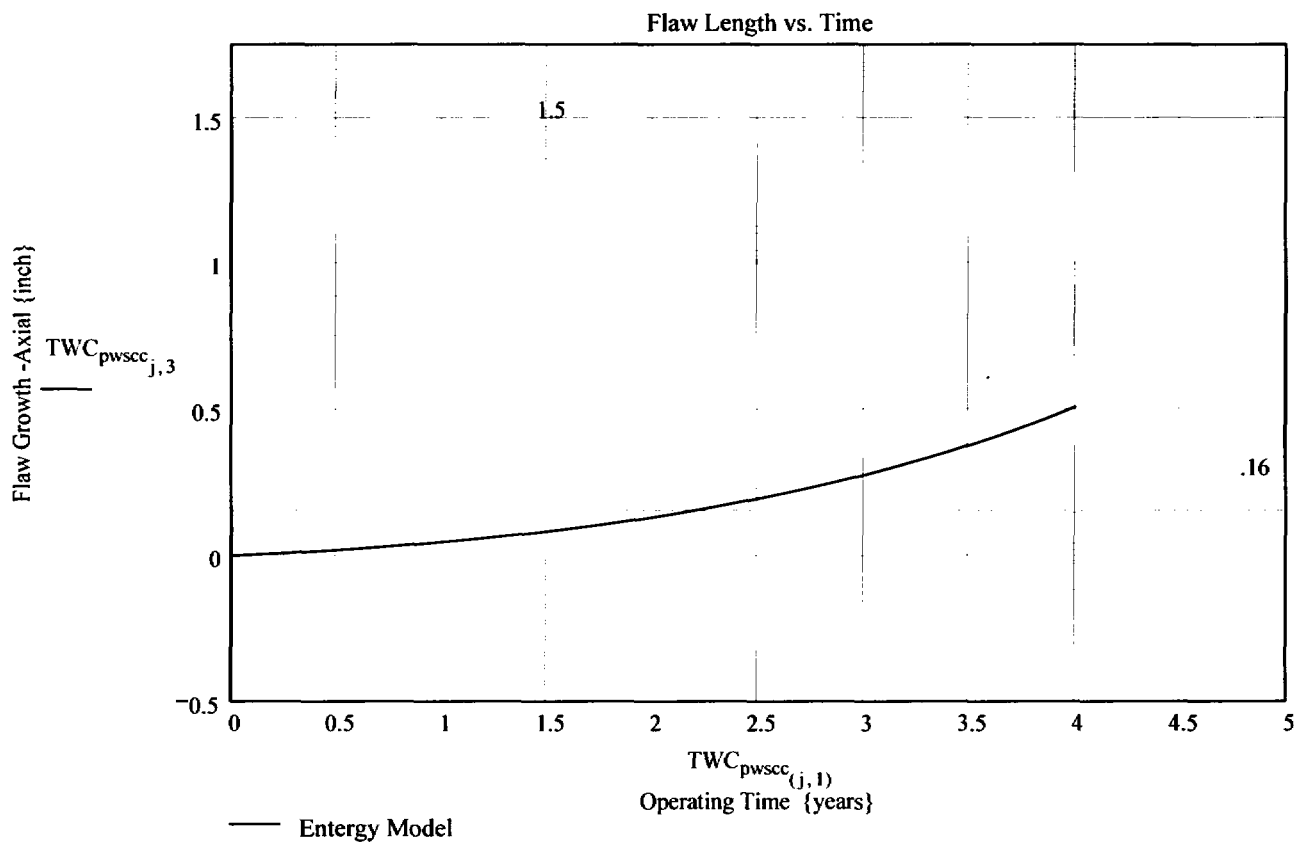
No User Input required beyond this Point

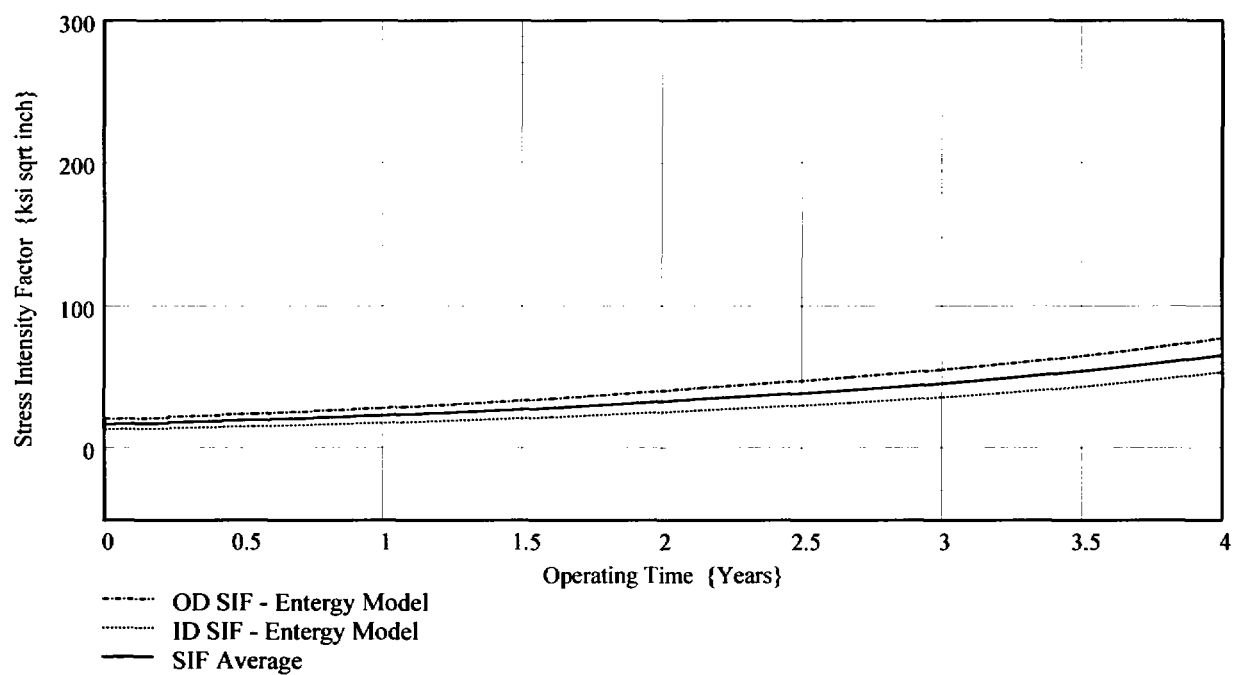
 Sat Aug 09 11:44:49 AM 2003

Developed by:

Verified by:

PropLength = 0.16





$TWC_{pwscc_{(j,6)}} =$

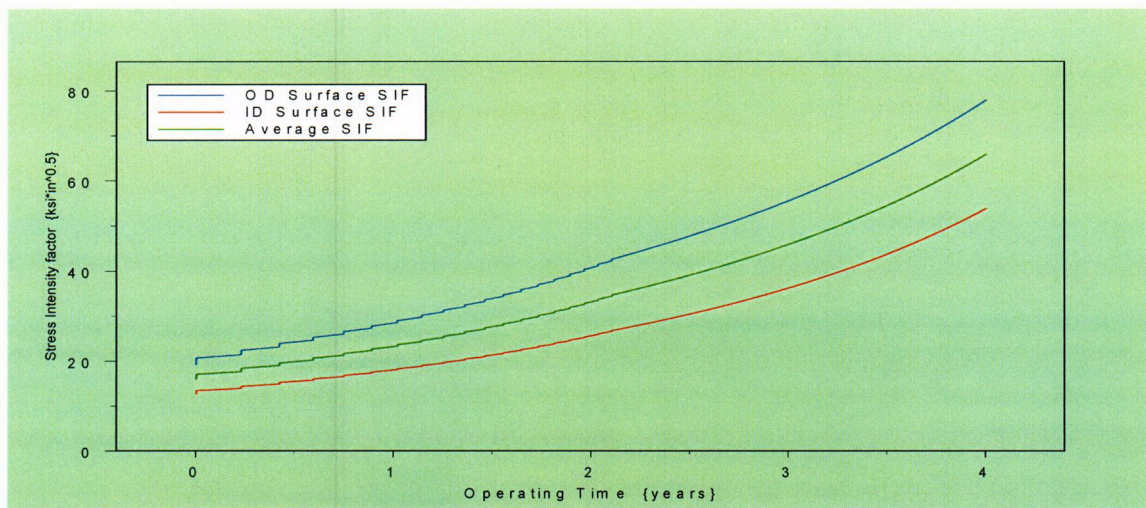
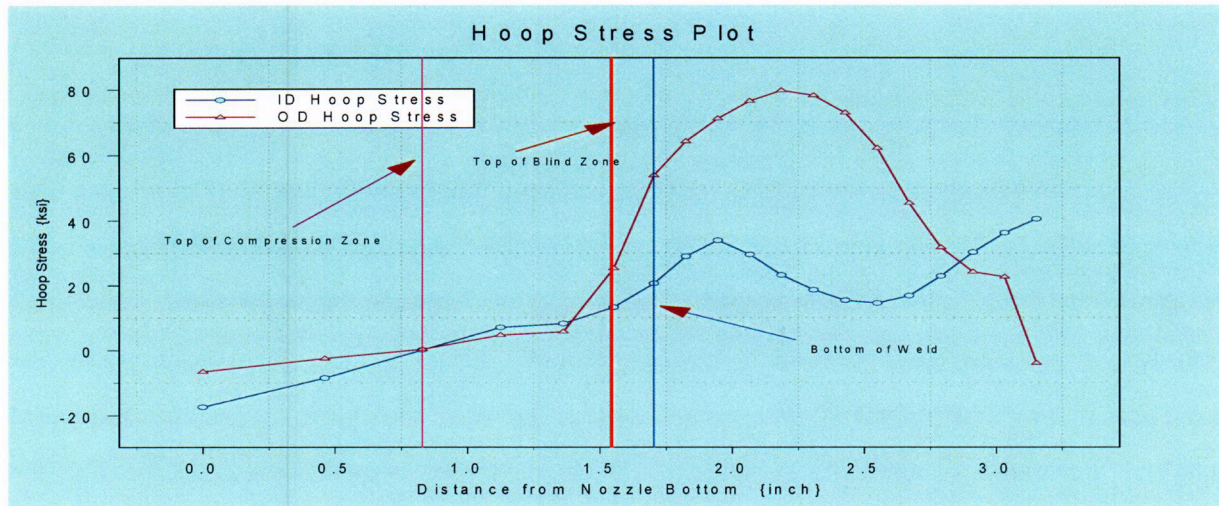
19.249
20.707
20.715
20.724
20.732
20.741
20.749
20.758
20.766
20.774
20.783
20.791
20.8
20.808
20.817
20.826

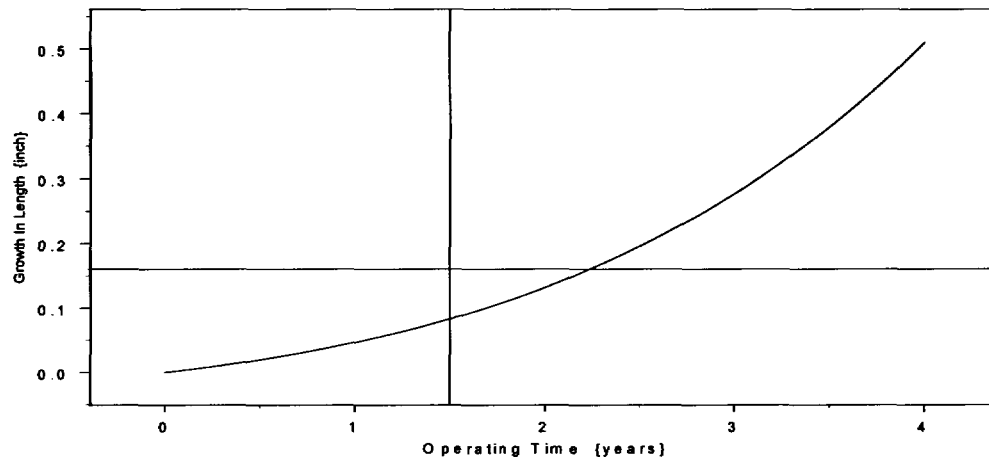
$TWC_{pwscc_{(j,7)}} =$

12.744
13.424
13.43
13.436
13.442
13.447
13.453
13.459
13.465
13.471
13.476
13.482
13.488
13.494
13.5
13.506

$TWC_{pwscc_{(j,8)}} =$

16.124
17.198
17.205
17.212
17.22
17.227
17.234
17.242
17.249
17.257
17.264
17.271
17.279
17.286
17.293
17.301





**Primary Water Stress Corrosion Crack Growth Analysis ID flaw;
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Developed by: J. S. Brihmadesam

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

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Mean Radius -to- Thickness Ratio:- " R_m/t " – between 1.0 and 300.0

**Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1 .
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .**

ID Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

Ref_{Point} := 1.544

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)*
- 2) The Center of the flaw at the reference point (Enter 2)*
- 3) The lower "C- tip" located at the reference point (Enter 3).*

Val := 2

The Input Below is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL_{Strs.Dist} := 4.268 Upper axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom).

Input Data :-

$L := 0.32$	Initial Flaw Length (Twice detectable length)
$a_0 := 0.661 \cdot 0.07$	Initial Flaw Depth (Minimum Detectable Depth was 5% TW)
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
Years := 4	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := \text{Years} \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left| \frac{I_{lim}}{50} \right| \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:

- Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)
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- Column "4" = Three quarter Thickness Stress data at each Elevation (ksi)
- Column "5" = OD Stress data at each Elevation (ksi)

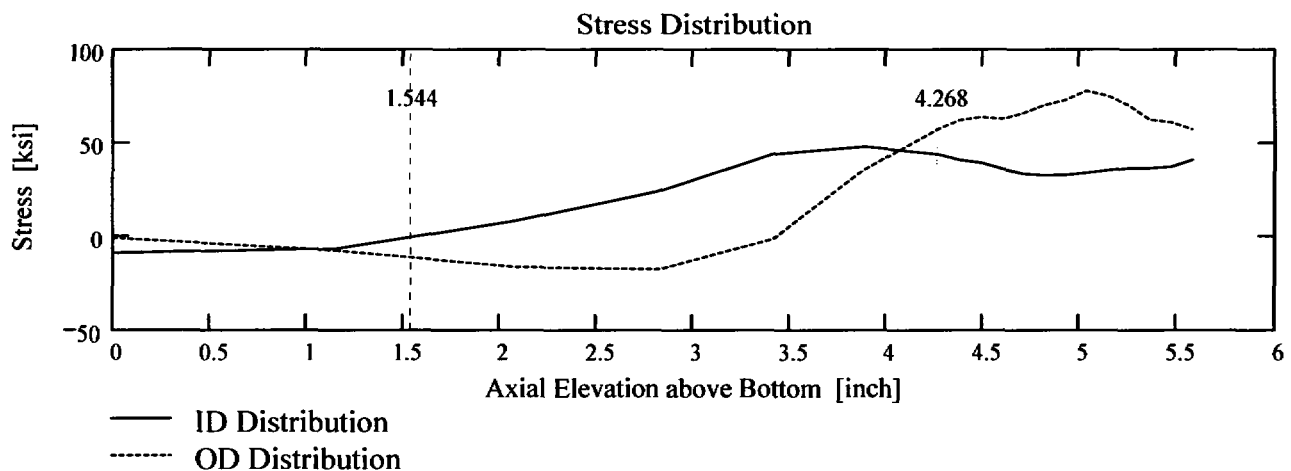
AllData :=

	0	1	2	3	4	5
0	0	-9.03	-5.86	-4.25	-2.69	-1.03
1	1.15	-6.76	-6.74	-7.24	-7.66	-7.8
2	2.08	7.97	1.74	-6.23	-11.85	-16.39
3	2.82	23.85	21.76	8.56	-6.39	-17.65
4	3.41	43.99	38.07	29.83	13.47	-1.63
5	3.89	47.95	41.75	35.45	33.32	35.85
6	4.27	43.76	39.21	38.4	53.02	57.54
7	4.38	40.77	36.24	41.27	61.45	62.19
8	4.49	39.28	35.33	44.86	64.2	63.9
9	4.6	36.02	35.39	46.84	64.32	62.93

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Highlight the region in the above table representing the region to be selected (click on the first cell for selection and drag the mouse whilst holding the left mouse button down. Once this is done click the right mouse button and select "Copy Selection"; this will copy the selected area on to the clipboard. Then click on the "Matrix" below (to the right of the $data$ statement) to highlight the entire matrix and delete it from the edit menu. When the Mathcad input symbol appears, use the paste function in the tool bar to paste the selection.

$$Data := \begin{pmatrix} 0 & -9.034 & -5.855 & -4.246 & -2.689 & -1.031 \\ 1.154 & -6.761 & -6.739 & -7.237 & -7.662 & -7.803 \\ 2.078 & 7.965 & 1.742 & -6.23 & -11.848 & -16.387 \\ 2.819 & 23.851 & 21.763 & 8.555 & -6.39 & -17.647 \\ 3.412 & 43.99 & 38.072 & 29.826 & 13.47 & -1.632 \\ 3.888 & 47.954 & 41.753 & 35.453 & 33.324 & 35.846 \\ 4.268 & 43.756 & 39.214 & 38.4 & 53.023 & 57.543 \\ 4.377 & 40.773 & 36.237 & 41.27 & 61.453 & 62.189 \\ 4.486 & 39.277 & 35.327 & 44.863 & 64.204 & 63.895 \end{pmatrix}$$

$$Axl := Data^{(0)} \quad MD := Data^{(3)} \quad ID := Data^{(1)} \quad TQ := Data^{(4)} \quad QT := Data^{(2)} \quad OD := Data^{(5)}$$

$$R_{ID} := \text{regress}(Axl, ID, 3)$$

$$R_{QT} := \text{regress}(Axl, QT, 3)$$

$$R_{OD} := \text{regress}(Axl, OD, 3)$$

$$R_{MD} := \text{regress}(Axl, MD, 3)$$


$$R_{TQ} := \text{regress}(Axl, TQ, 3)$$

$$FL_{Cntr} := \begin{cases} Ref_{Point} - c_0 & \text{if } Val = 1 \\ Ref_{Point} & \text{if } Val = 2 \\ Ref_{Point} + c_0 & \text{otherwise} \end{cases} \quad \text{Flaw center Location above Nozzle Bottom}$$

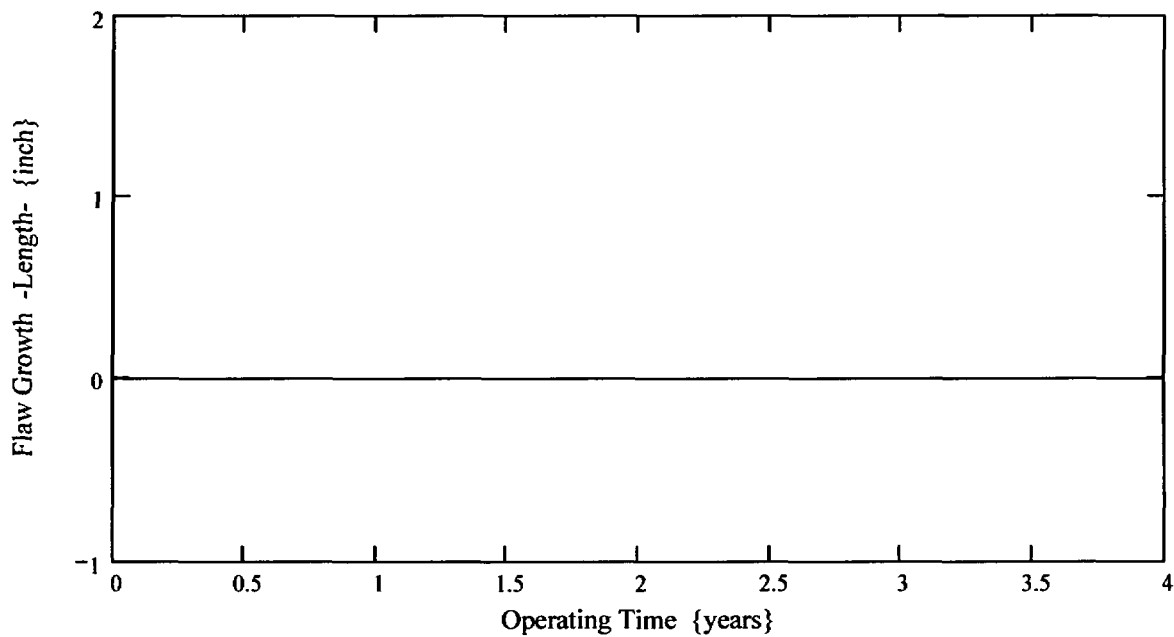
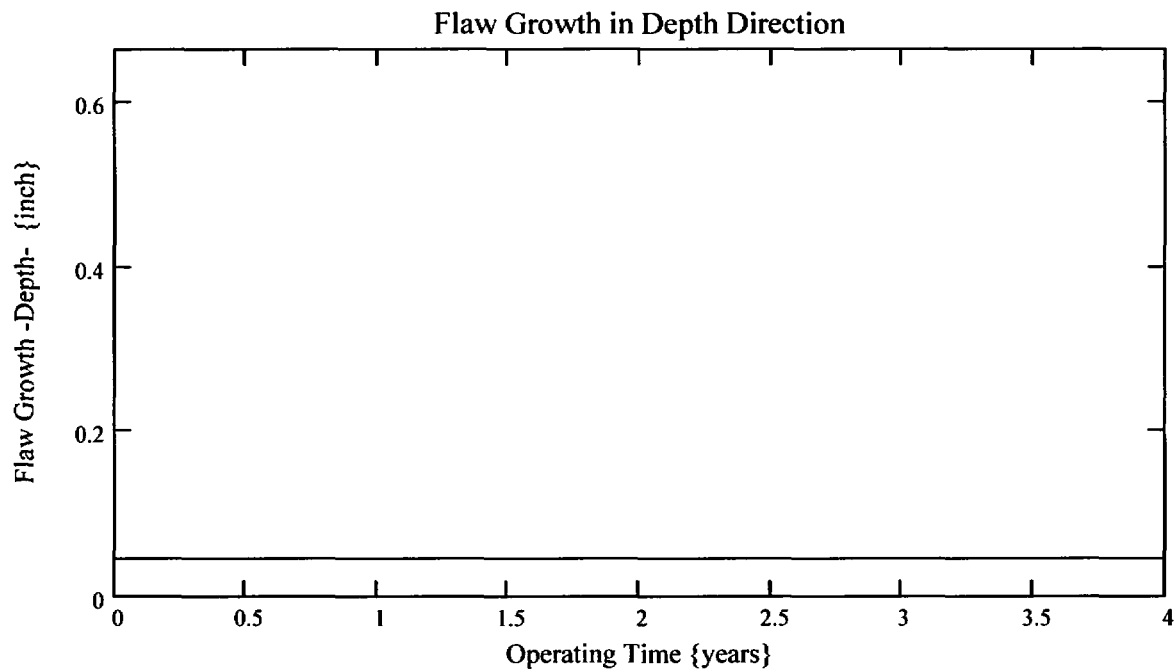
$$U_{Tip} := FL_{Cntr} + c_0$$

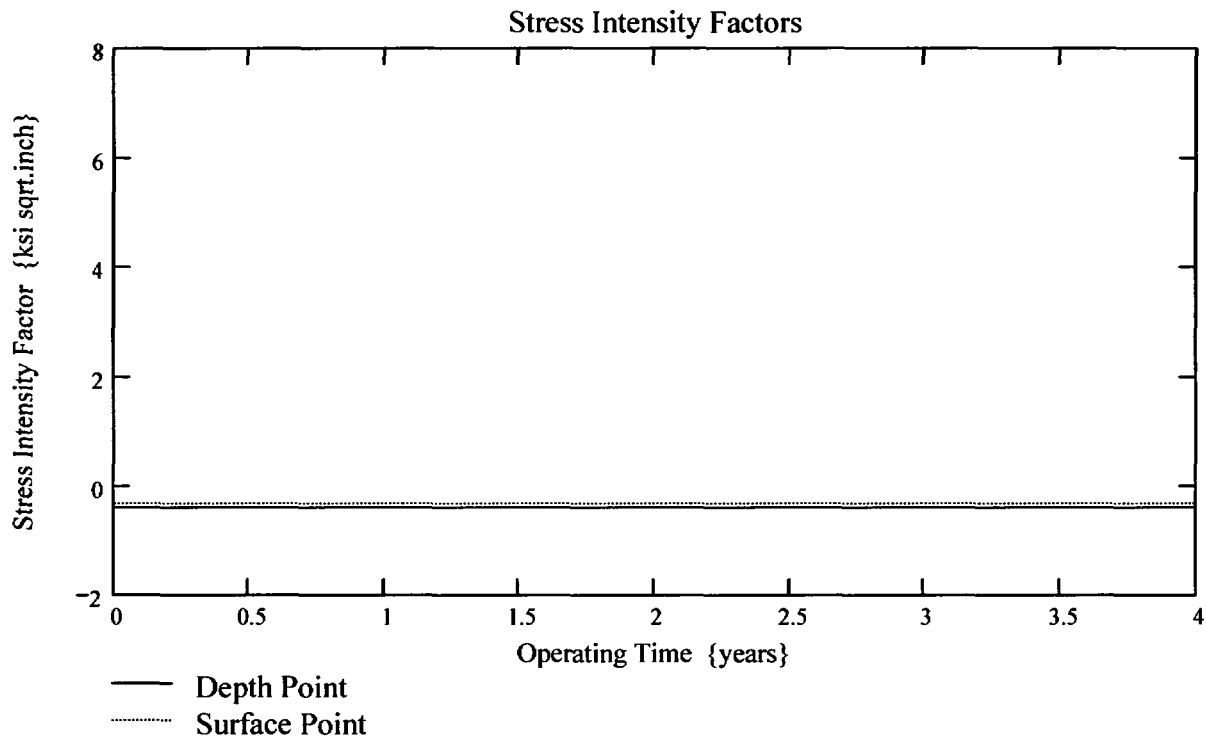
$$IncStrs.avg := \frac{ULStrs.Dist - U_{Tip}}{20}$$

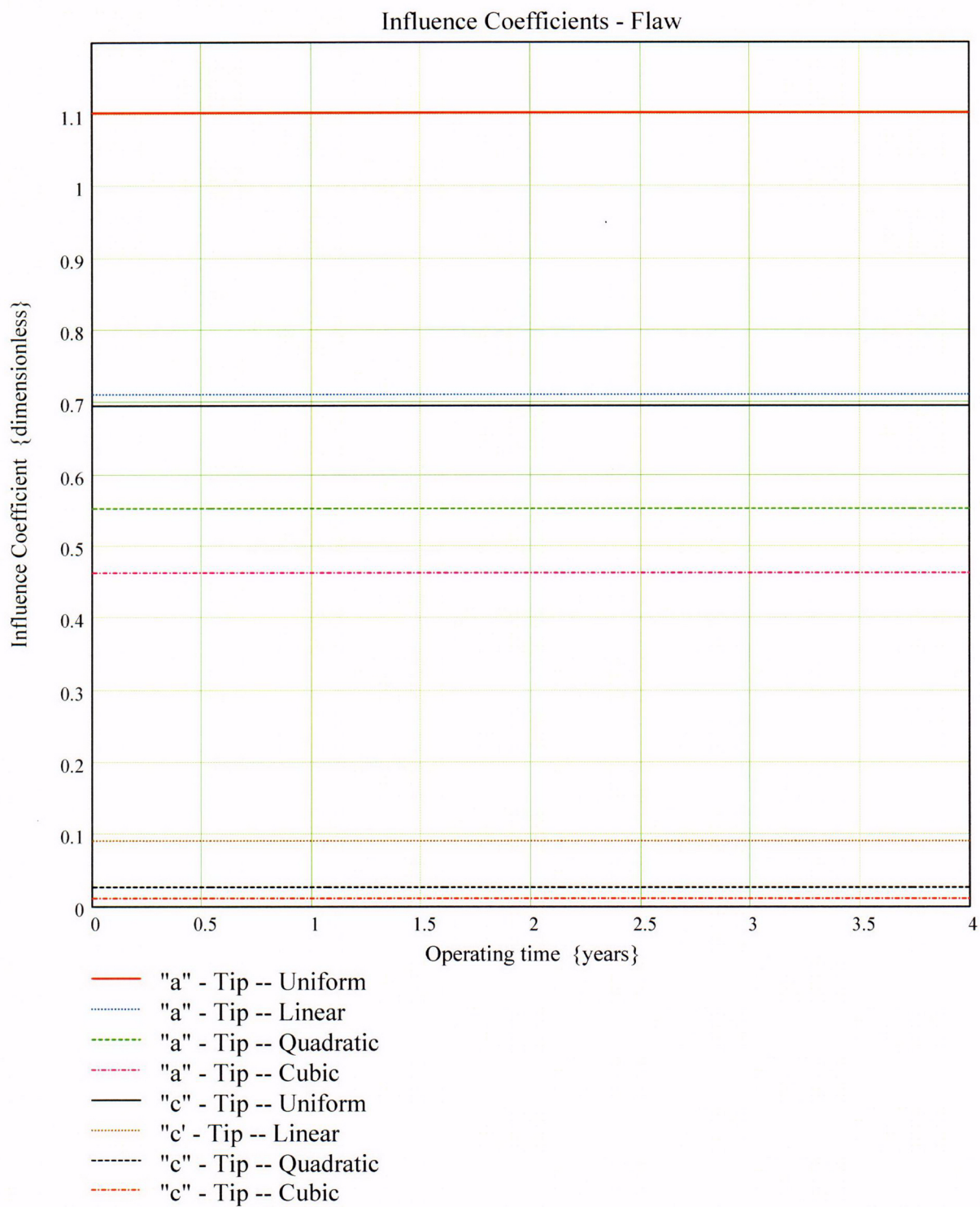
No User Input is required beyond this Point

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PropLength = 2.564







$$CGR_{sambi(k,8)} =$$

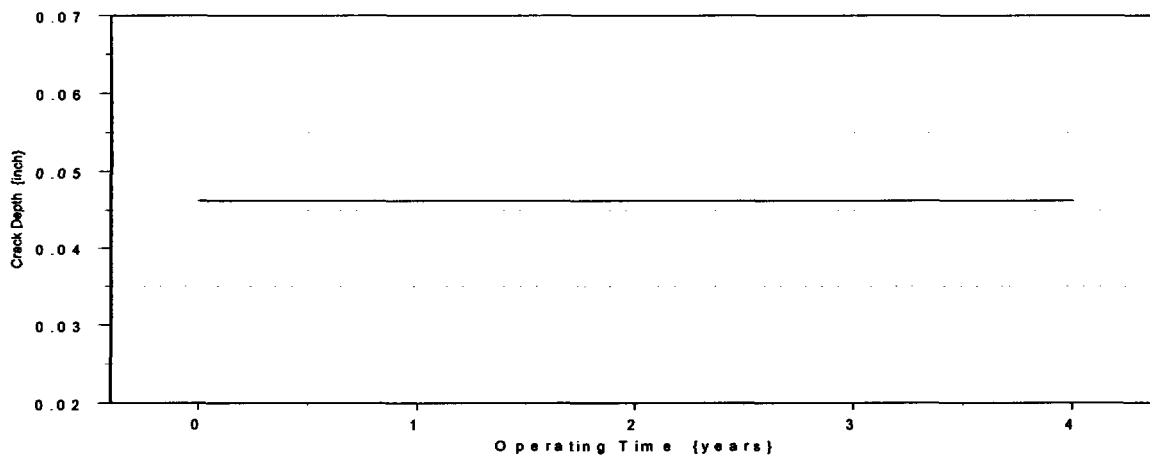
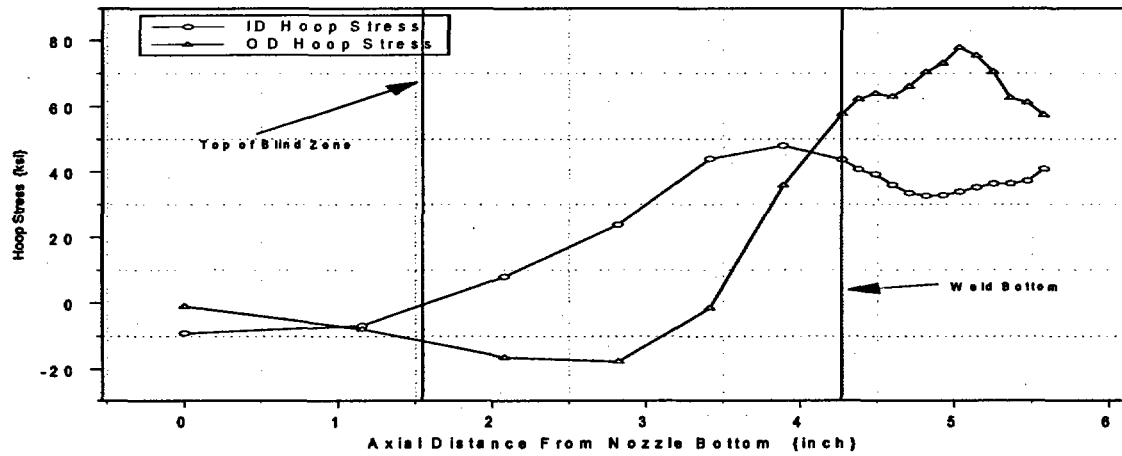
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103
1.103

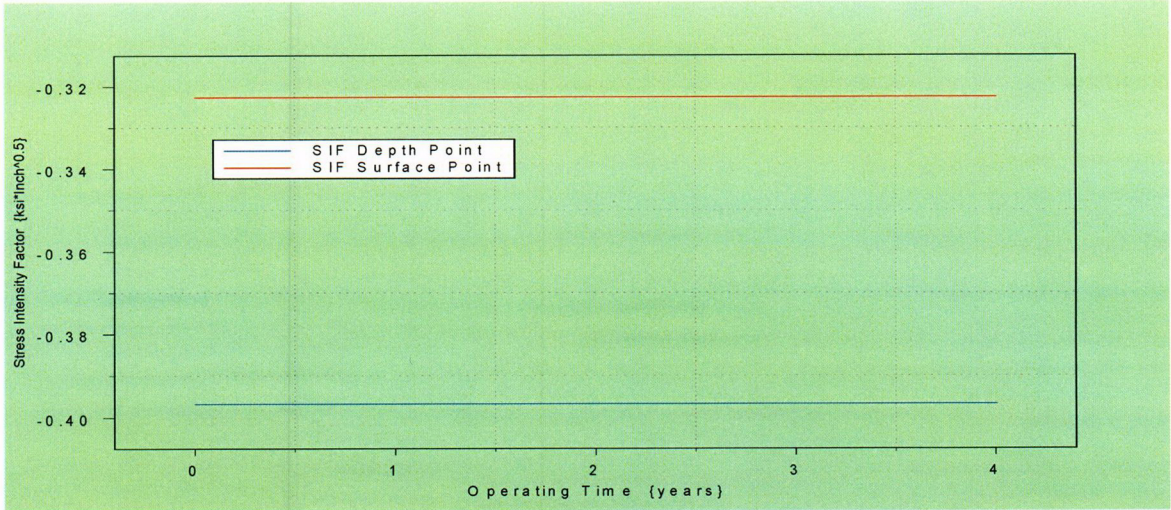
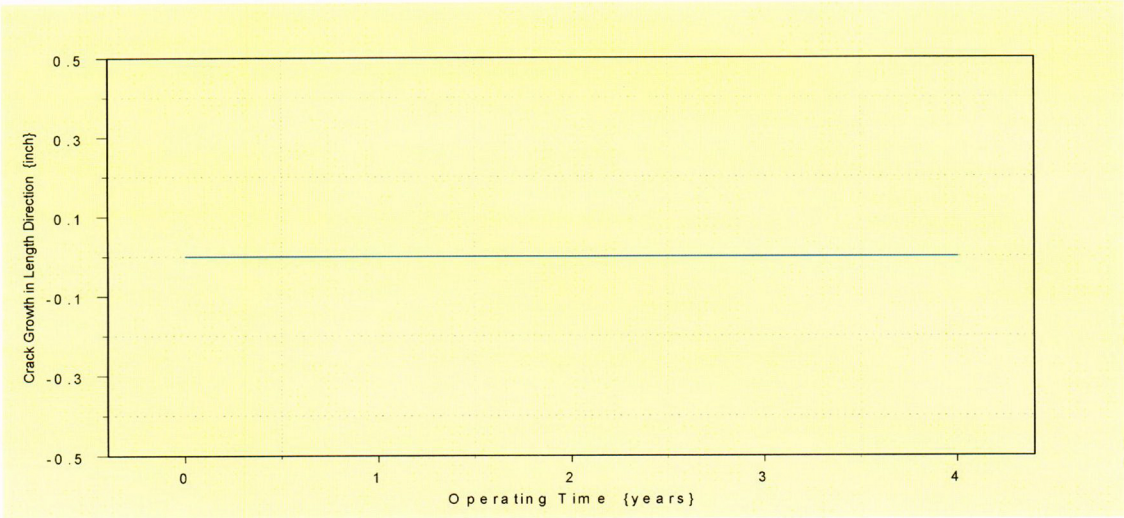
$$CGR_{sambi(k,6)} =$$

-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323
-0.323

$$CGR_{sambi(k,5)} =$$

-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397
-0.397





Primary Water Stress Corrosion Crack Growth Analysis - OD Surface Flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesar

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM -"28" Degree Nozzle, Uphill Azimuth,
1.544" above Nozzle Bottom

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " -- between 1.0 and 300.0

Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.

The correction is applied in the determination of the crack extension to obtain the value in inch/hr .

OD Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

$Ref_{point} := 1.544$

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)
- 2) The Center of the flaw at the reference point (Enter 2)
- 3) The lower "C- tip" located at the reference point (Enter 3).

$Val := 2$

Upper Limit to be selected for stress distribution (e.g. Weld bottom). This is the elevation from Nozzle Bottom. Enter this value below

$UL_{Strs.Dist} := 4.268$

Upper Axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.32$	Initial Flaw Length
$a_0 := 0.661 \cdot 0.12$	Initial Flaw Depth
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
$Years := 4$	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left| \frac{I_{lim}}{50} \right| \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:
Column "0" = Axial distance from minimum to maximum recorded on data sheet(inches)
Column "1" = ID Stress data at each Elevation (ksi)
Column "2" = Quarter Thickness Stress data at each Elevation (ksi)
Column "3" = Mid Thickness Stress data at each Elevation (ksi)
Column "4" = Three Quarter Thickness Stress data at each Elevation (ksi)
Column "5" = OD Stress data at each Elevation (ksi)

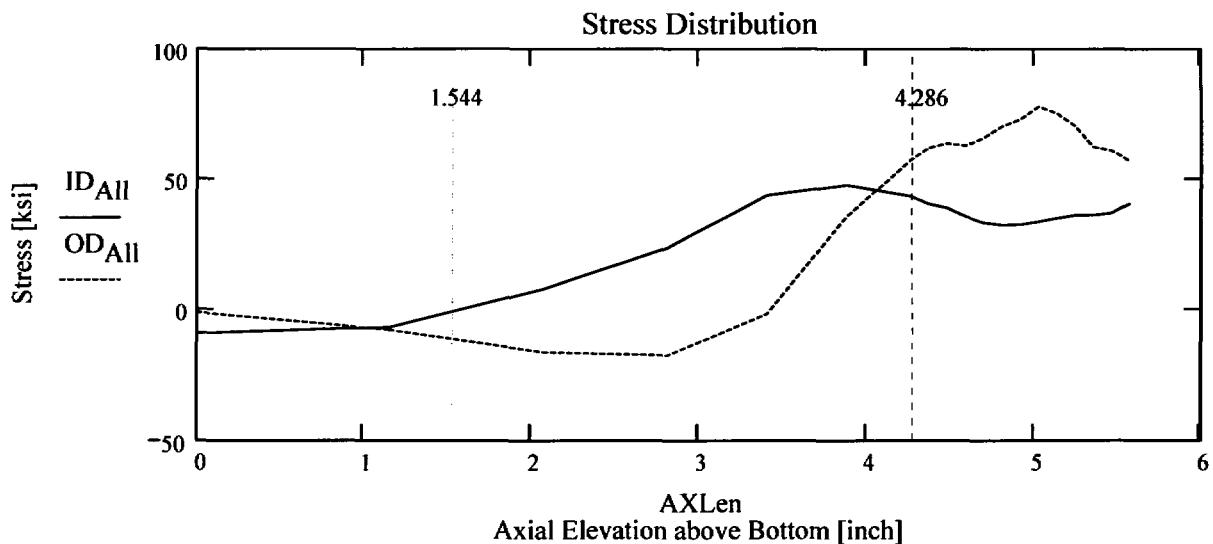
AllData :=

	0	1	2	3	4	5
0	0	-9.03	-5.86	-4.25	-2.69	-1.03
1	1.15	-6.76	-6.74	-7.24	-7.66	-7.8
2	2.08	7.97	1.74	-6.23	-11.85	-16.39
3	2.82	23.85	21.76	8.56	-6.39	-17.65
4	3.41	43.99	38.07	29.83	13.47	-1.63
5	3.89	47.95	41.75	35.45	33.32	35.85
6	4.27	43.76	39.21	38.4	53.02	57.54
7	4.38	40.77	36.24	41.27	61.45	62.19
8	4.49	39.28	35.33	44.86	64.2	63.9
9	4.6	36.02	35.39	46.84	64.32	62.93

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled Data_{All} that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

$$\text{Data} := \begin{pmatrix} 0 & -9.034 & -5.855 & -4.246 & -2.689 & -1.031 \\ 1.154 & -6.761 & -6.739 & -7.237 & -7.662 & -7.803 \\ 2.078 & 7.965 & 1.742 & -6.23 & -11.848 & -16.387 \\ 2.819 & 23.851 & 21.763 & 8.555 & -6.39 & -17.647 \\ 3.412 & 43.99 & 38.072 & 29.826 & 13.47 & -1.632 \\ 3.888 & 47.954 & 41.753 & 35.453 & 33.324 & 35.846 \\ 4.268 & 43.756 & 39.214 & 38.4 & 53.023 & 57.543 \\ 4.377 & 40.773 & 36.237 & 41.27 & 61.453 & 62.189 \\ 4.486 & 39.277 & 35.327 & 44.863 & 64.204 & 63.895 \end{pmatrix}$$

$$\text{Axl} := \text{Data}^{(0)} \quad \text{MD} := \text{Data}^{(3)} \quad \text{ID} := \text{Data}^{(1)} \quad \text{TQ} := \text{Data}^{(4)} \quad \text{QT} := \text{Data}^{(2)} \quad \text{OD} := \text{Data}^{(5)}$$

$$R_{ID} := \text{regress}(\text{Axl}, \text{ID}, 3)$$

$$R_{QT} := \text{regress}(\text{Axl}, \text{QT}, 3)$$

$$R_{OD} := \text{regress}(\text{Axl}, \text{OD}, 3)$$

$$R_{MD} := \text{regress}(\text{Axl}, \text{MD}, 3)$$

$$R_{TQ} := \text{regress}(\text{Axl}, \text{TQ}, 3)$$


$$\text{FL}_{\text{Cntr}} := \begin{cases} \text{Ref}_{\text{Point}} - c_0 & \text{if Val} = 1 \\ \text{Ref}_{\text{Point}} & \text{if Val} = 2 \\ \text{Ref}_{\text{Point}} + c_0 & \text{otherwise} \end{cases}$$

Flaw center Location Location above Nozzle Bottom

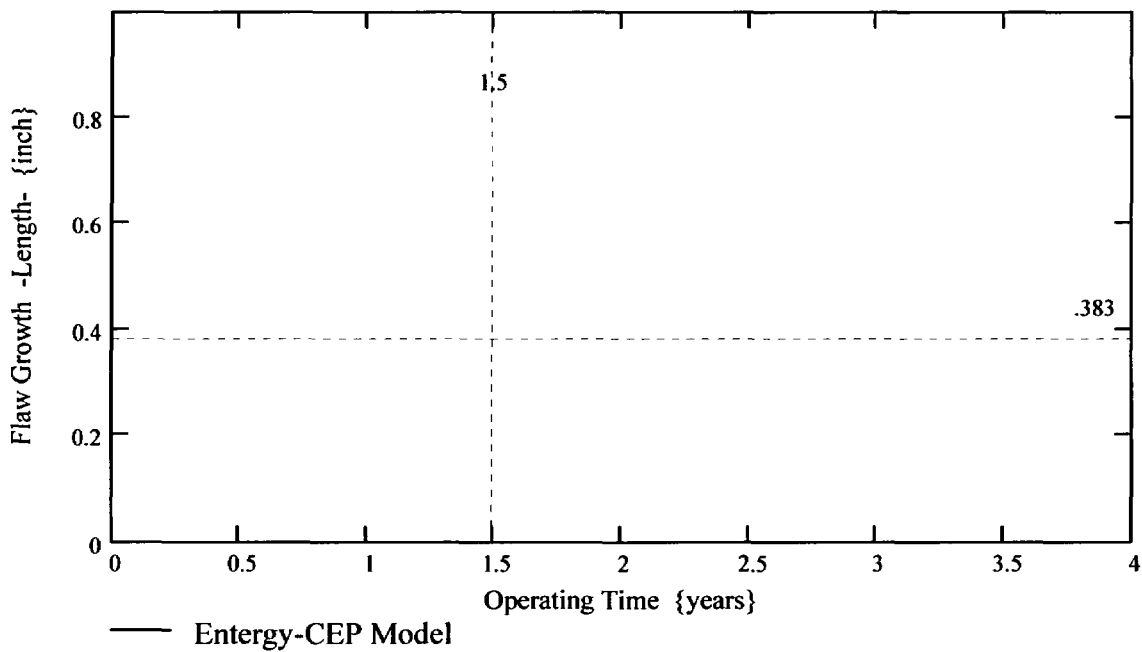
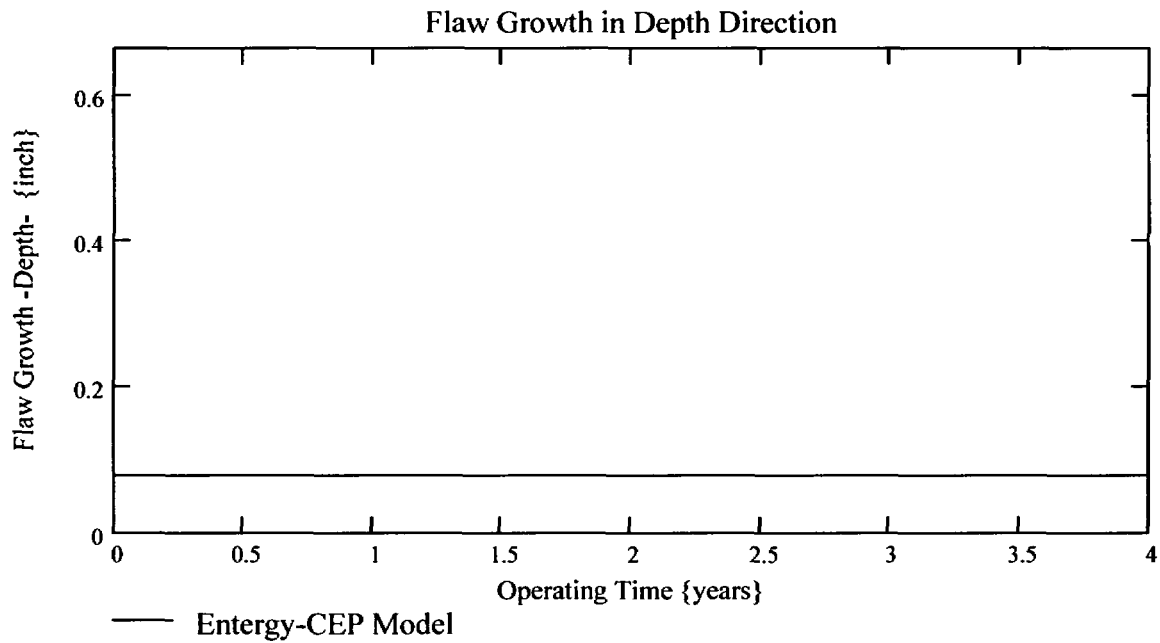
$$U_{\text{Tip}} := \text{FL}_{\text{Cntr}} + c_0$$

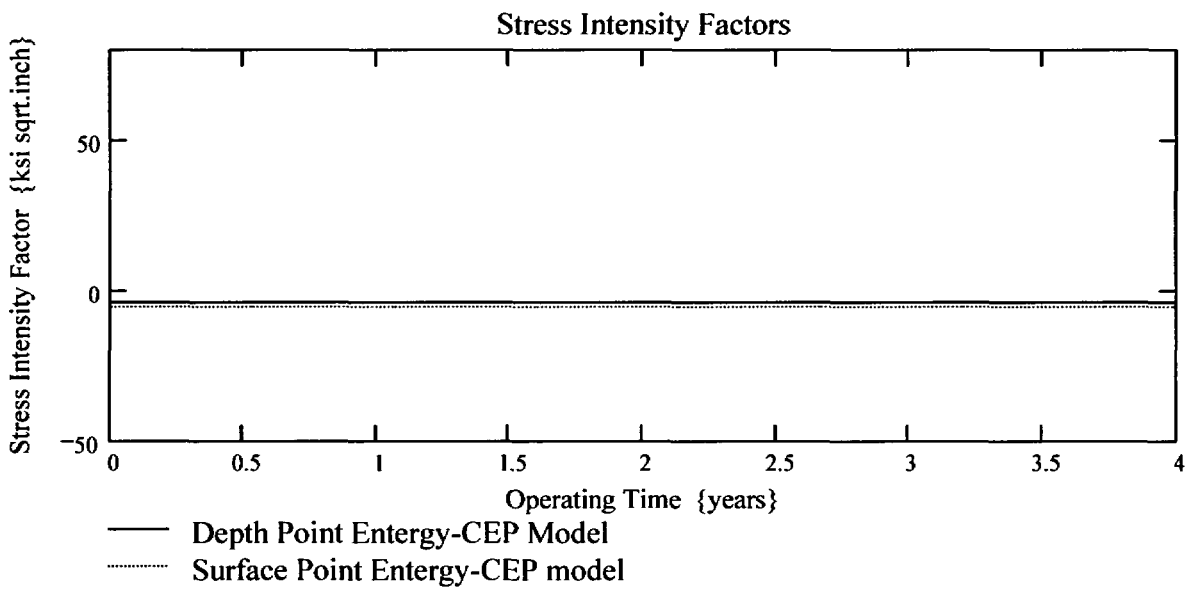
$$\text{IncStrs.avg} := \frac{U_{\text{LStrs.Dist}} - U_{\text{Tip}}}{n}$$

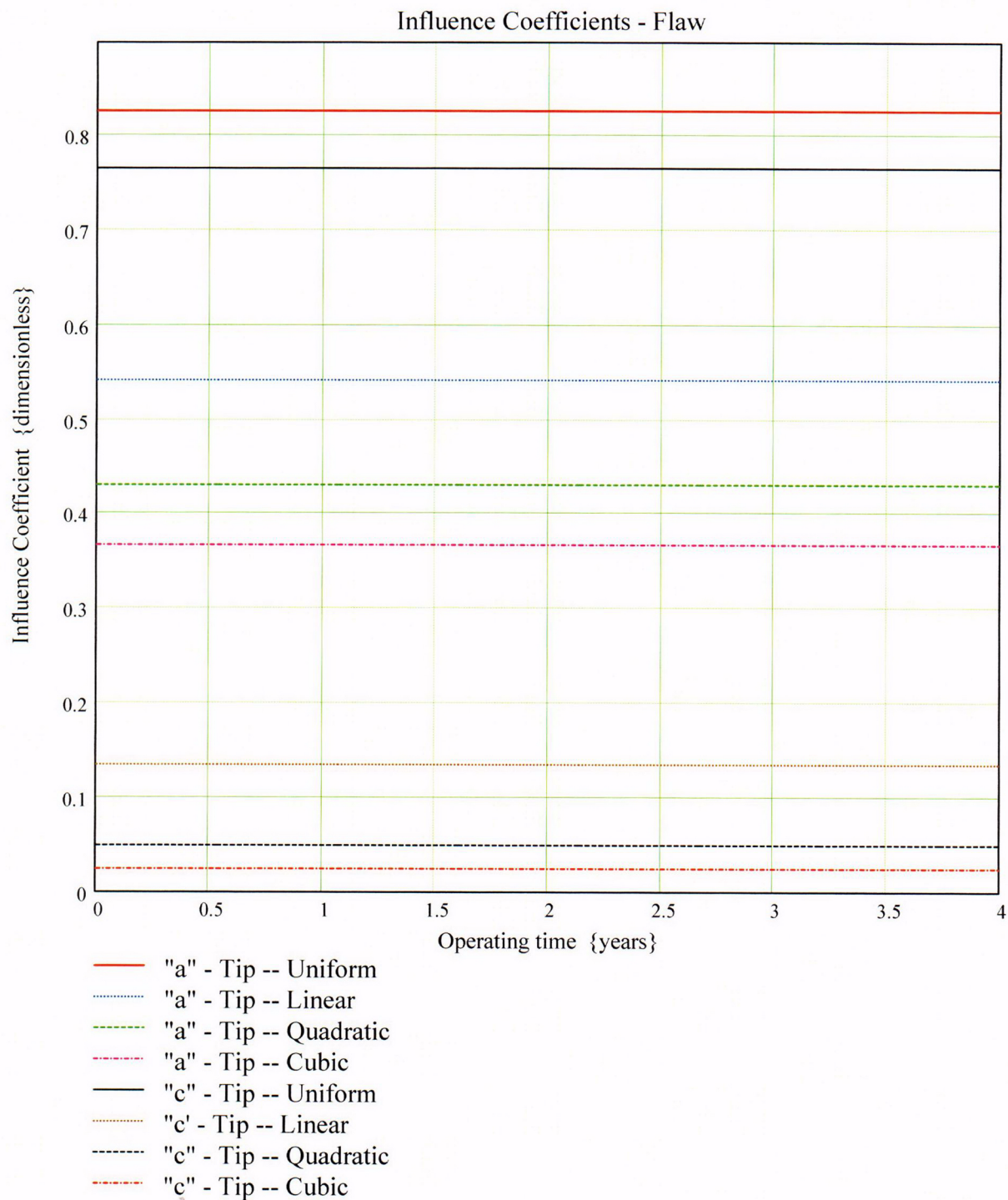
No User Input is required beyond this Point

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$$\text{Prop}_{\text{Length}} \approx 2.564$$







$CGR_{sambi(k,8)} =$

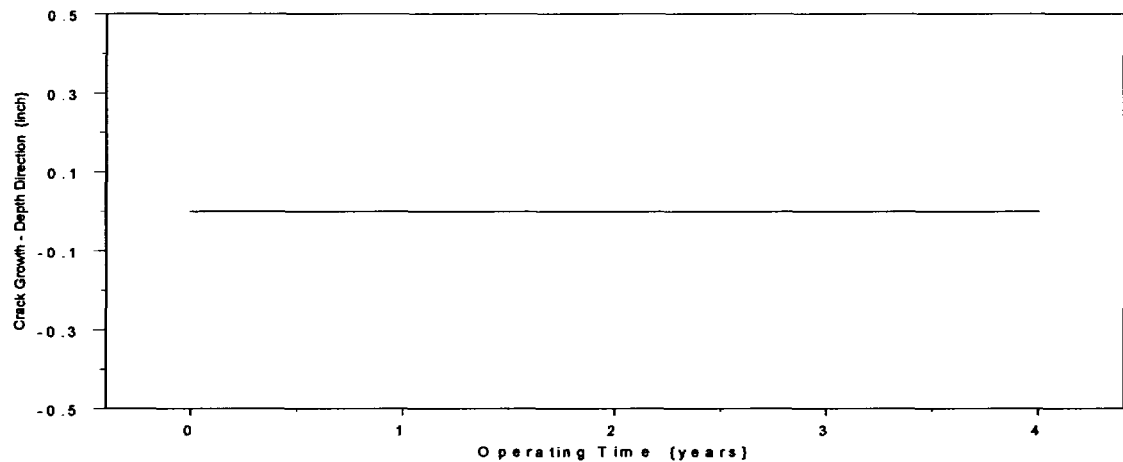
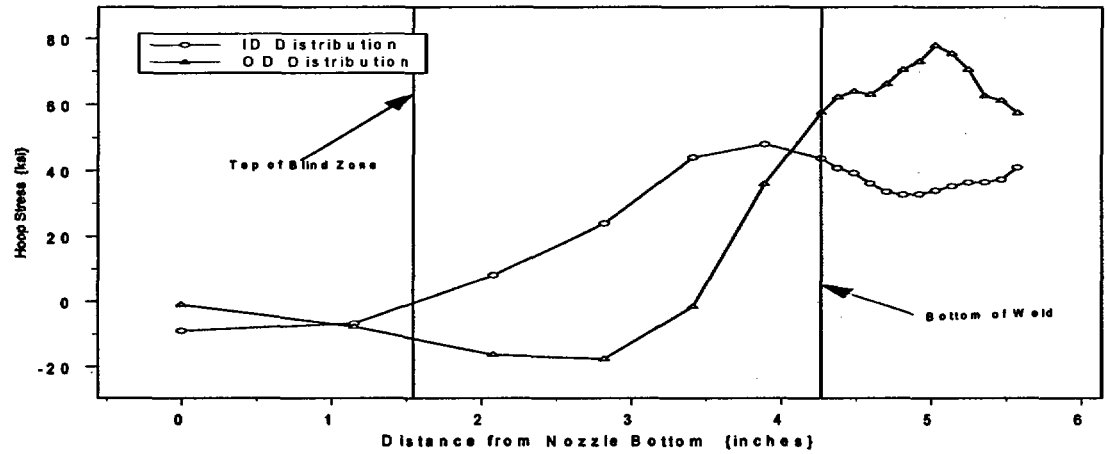
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
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0.827
0.827

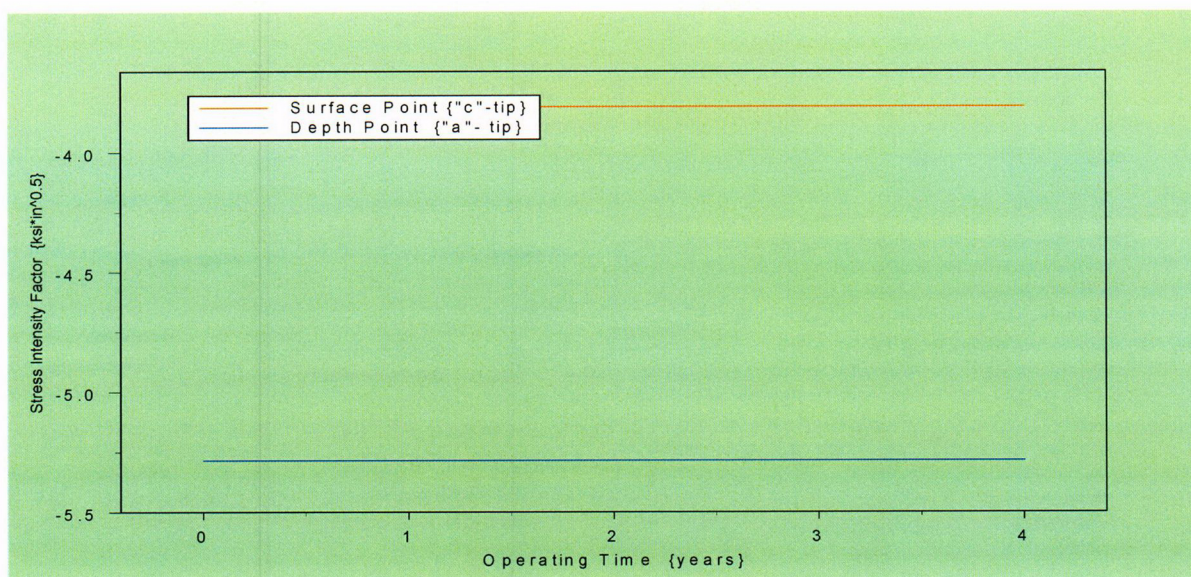
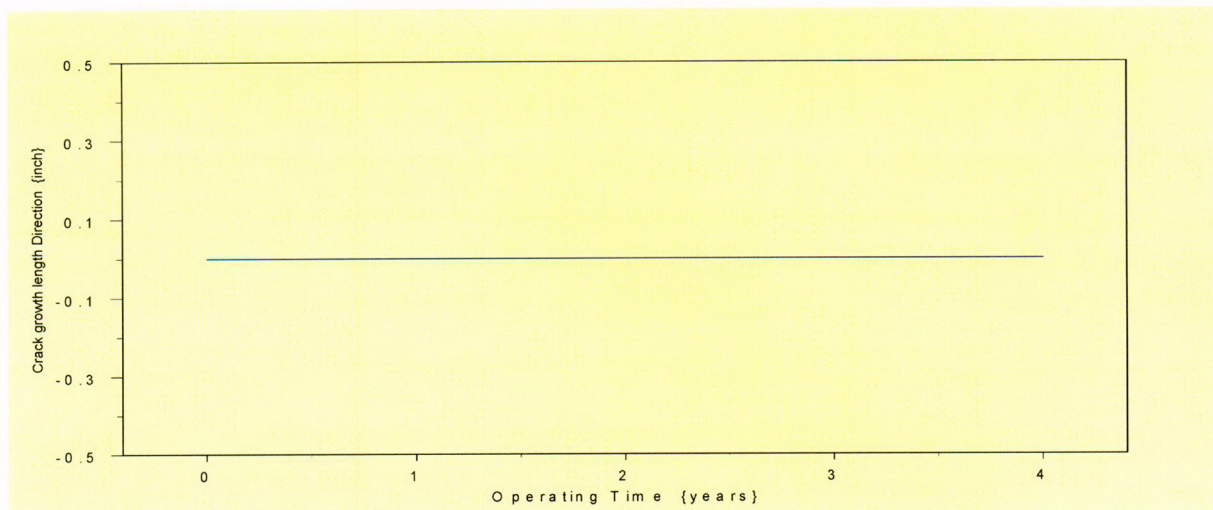
$CGR_{sambi(k,6)} =$

-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284
-5.284

$CGR_{sambi(k,5)} =$

-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812
-3.812





Stress Corrosion Crack Growth Analysis Throughwall flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadeseam

Verified by: B. C. Gray

Note : Only for use when $R_{outside}/t$ is between 2.0 and 5.0 (Thickwall Cylinder)

References :

- 1) ASME PVP paper PVP-350, Page 143; 1997 (Fracture Mechanics Model)
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM - "28" Degree Nozzle, Uphill Azimuth,
1.544 inch above Nozzle Bottom

Calculation Reference: MRP 75 th Percentile and Flaw Pressurized

Note : *Used the Metric form of the equation from EPRI MRP 55-Rev. 1.*
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .

Through Wall Axial Flaw

The first Input is to locate the Reference Line (eg. top of the Blind Zone). The throughwall flaw "Upper Tip" is located at the Reference Line.

Enter the elevation of the Reference Line (eg. Blind Zone) above the nozzle bottom in inches.

BZ := 1.544

Location of Blind Zone above nozzle bottom (inch)

The Second Input is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

ULStrs.Dist := 4.268

Upper axial Extent for Stress Distribution to be used in the analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.25$ Initial Flaw Length TW axial (Based on 10 Ksi average stress)

$od := 4.05$ Tube OD

$id := 2.728$ Tube ID

$P_{Int} := 2.235$ Design Operating Pressure (internal)

$Years := 4$ Number of Operating Years

$I_{lim} := 1500$ Iteration limit for Crack Growth loop

$T := 604$ Estimate of Operating Temperature

$\nu := 0.307$ Poissons ratio @ 600 F

$\alpha_{0c} := 2.67 \cdot 10^{-12}$ Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F

$Q_g := 31.0$ Thermal activation Energy for Crack Growth (MRP)

$T_{ref} := 617$ Reference Temperature for normalizing Data deg. F

$$C_0 := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3} \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right)} \right]} \cdot \alpha_{0c} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$R_o := \frac{od}{2} \quad R_i := \frac{id}{2} \quad t := R_o - R_i \quad R_m := R_i + \frac{t}{2} \quad CF_{inhr} := 1.417 \cdot 10^5$$

$$C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left\lfloor \frac{I_{lim}}{50} \right\rfloor \quad l := \frac{L}{2}$$

Stress Distribution in the tube. The outside surface is the reference surface for all analysis in accordance with the reference.

Stress Input Data

Import the Required data from applicable Excel spread Sheet. The column designations are as follows:

Cloumn "0" = Axial distance from Minimum to Maximum recorded on the data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

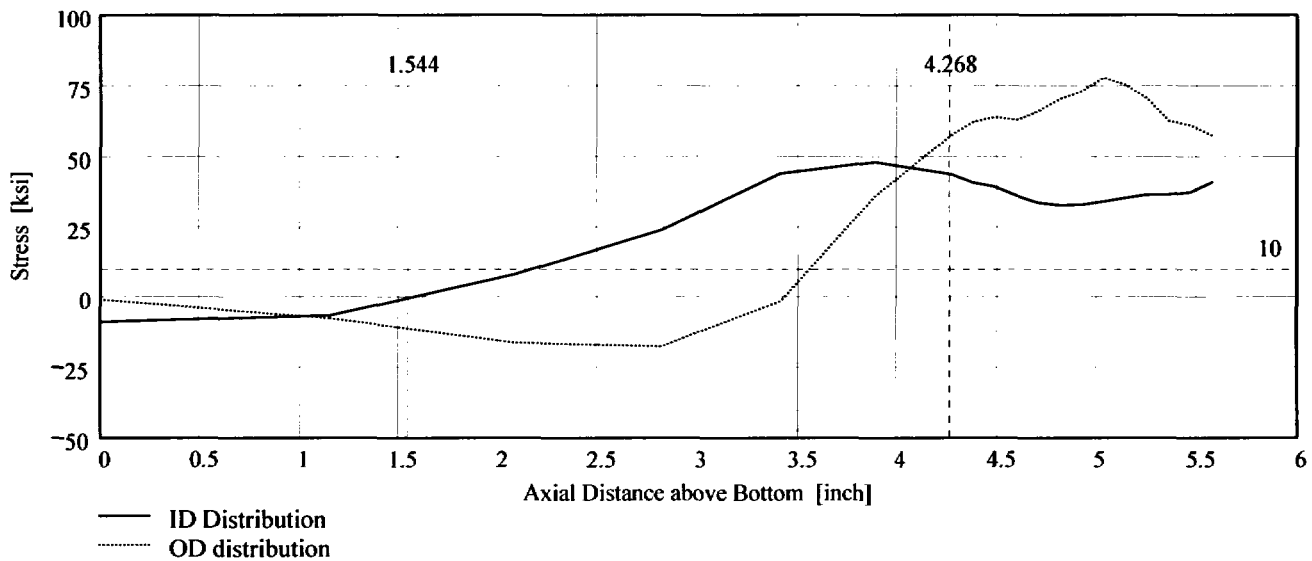
DataAll :=

	0	1	2	3	4	5
0	0	-9.03	-5.86	-4.25	-2.69	-1.03
1	1.15	-6.76	-6.74	-7.24	-7.66	-7.8
2	2.08	7.97	1.74	-6.23	-11.85	-16.39
3	2.82	23.85	21.76	8.56	-6.39	-17.65
4	3.41	43.99	38.07	29.83	13.47	-1.63
5	3.89	47.95	41.75	35.45	33.32	35.85
6	4.27	43.76	39.21	38.4	53.02	57.54
7	4.38	40.77	36.24	41.27	61.45	62.19
8	4.49	39.28	35.33	44.86	64.2	63.9
9	4.6	36.02	35.39	46.84	64.32	62.93
10	4.7	33.54	36.17	48.06	64.48	66.03
11	4.81	32.63	36.62	47.78	67.61	70.36

AllAx1 := DataAll⁽⁰⁾

AllID := DataAll⁽¹⁾

AllOD := DataAll⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

	0	-9.034	-5.855	-4.246	-2.689	-1.031
	1.154	-6.761	-6.739	-7.237	-7.662	-7.803
	2.078	7.965	1.742	-6.23	-11.848	-16.387
	2.819	23.851	21.763	8.555	-6.39	-17.647
	3.412	43.99	38.072	29.826	13.47	-1.632
	3.888	47.954	41.753	35.453	33.324	35.846
Data :=	4.268	43.756	39.214	38.4	53.023	57.543
	4.377	40.773	36.237	41.27	61.453	62.189
	4.486	39.277	35.327	44.863	64.204	63.895
	4.595	36.022	35.389	46.842	64.323	62.934
	4.704	33.54	36.173	48.06	64.483	66.03
	4.813	32.631	36.616	47.779	67.612	70.356

$Ax1 := Data^{(0)}$

$ID := Data^{(1)}$

$OD := Data^{(5)}$

$R_{ID} := \text{regress}(Ax1, ID, 3)$

$R_{OD} := \text{regress}(Ax1, OD, 3)$

$$FL_{Cntr} := BZ - I$$

Flaw Center above Nozzle Bottom

$$IncStrs.avg := \frac{ULStrs.Dist - BZ}{20}$$

No User Input required beyond this Point

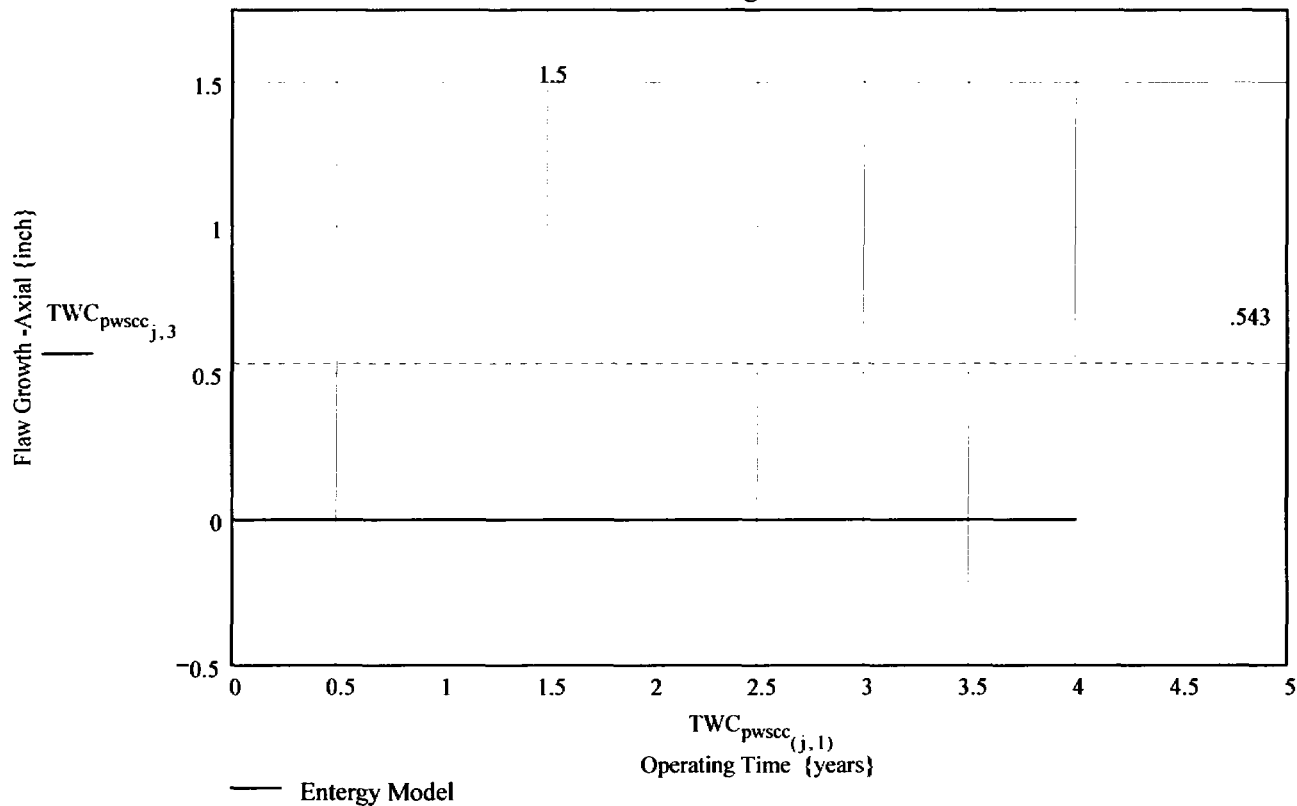
 Sat Aug 09 11:44:49 AM 2003

Developed by:

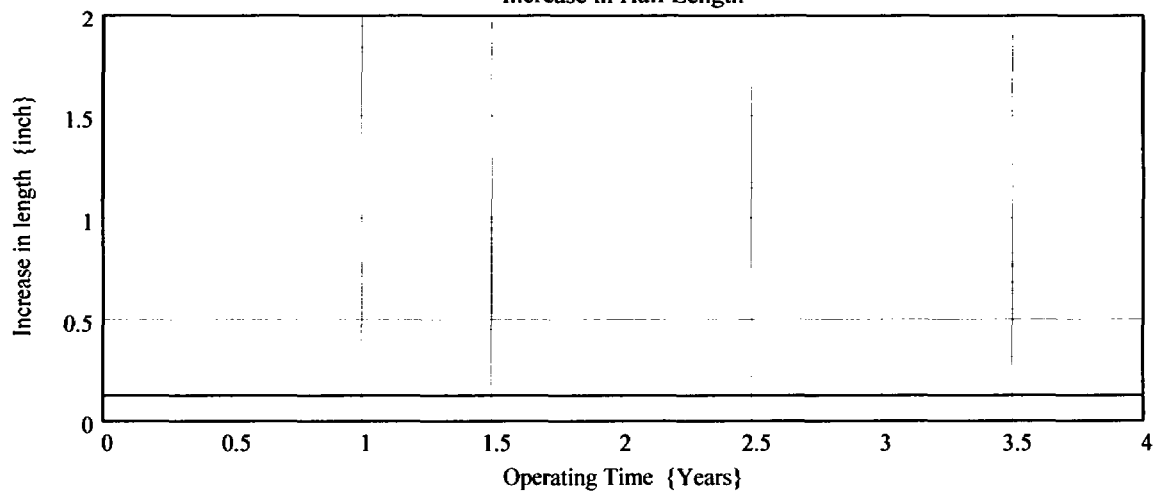
Verified by:

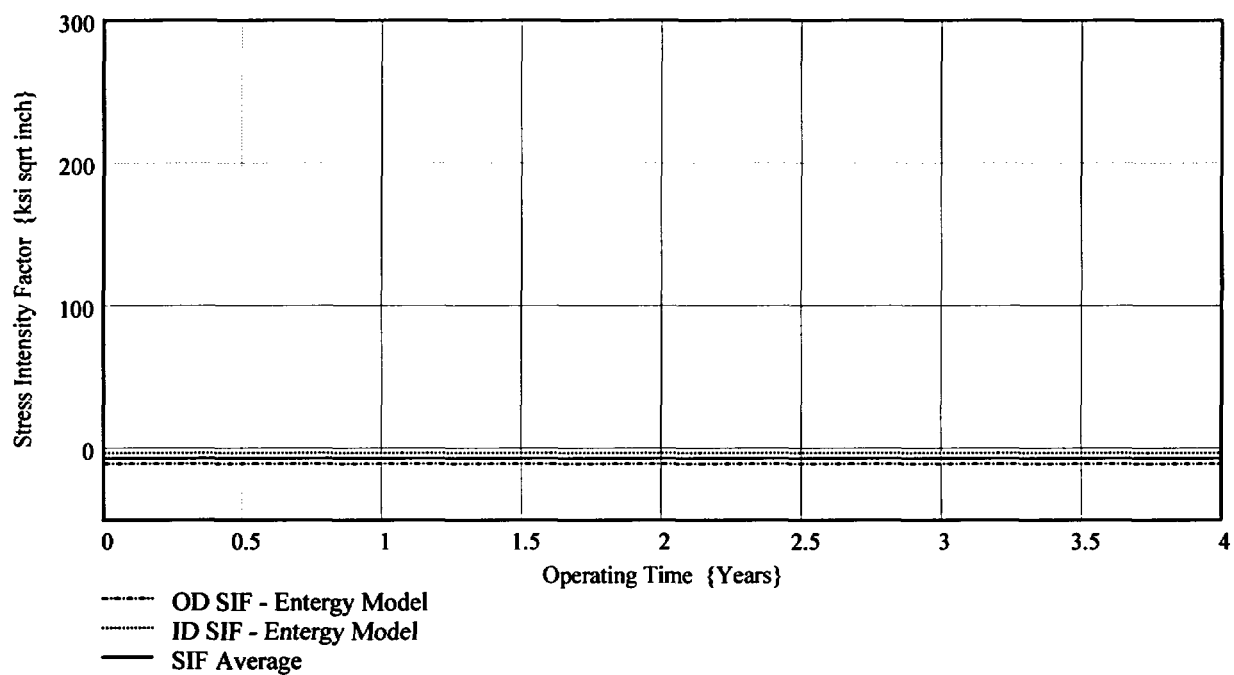
PropLength = 2.724

Flaw Length vs. Time



Increase in Half Length





$TWC_{pwscc(j,6)} =$

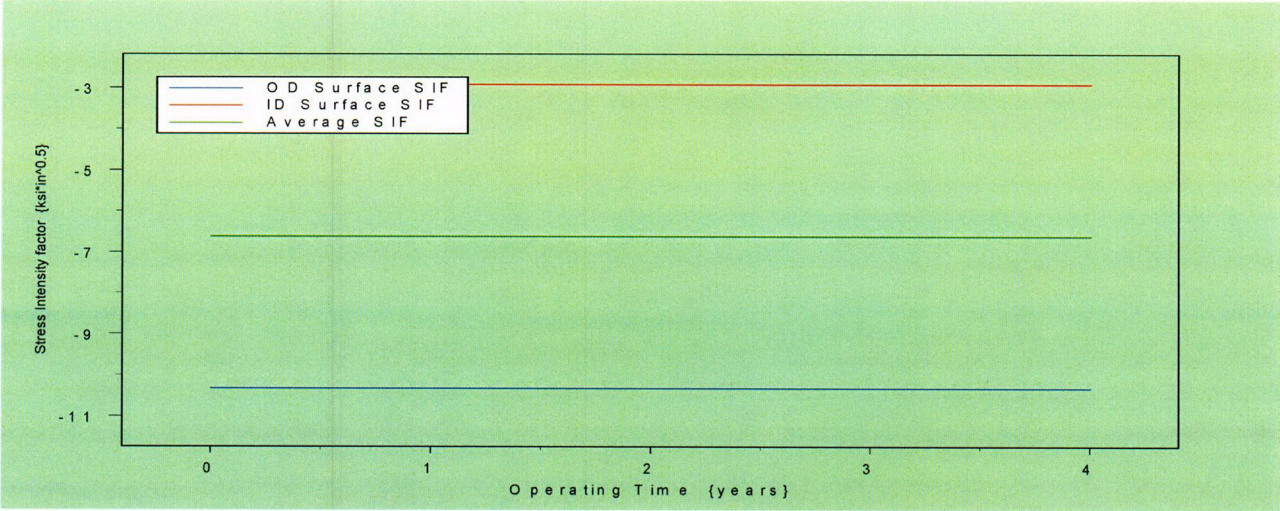
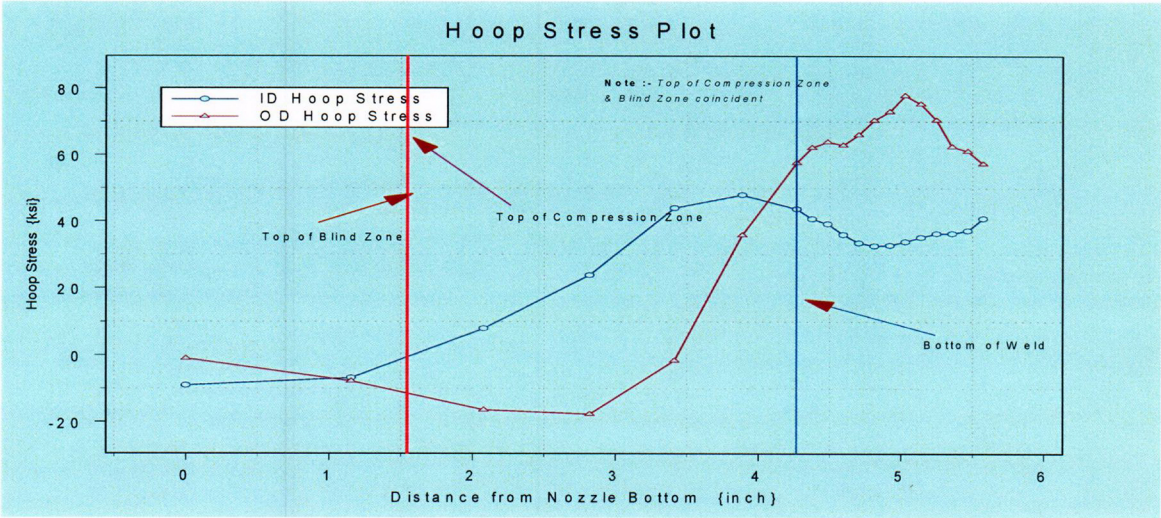
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322
-10.322

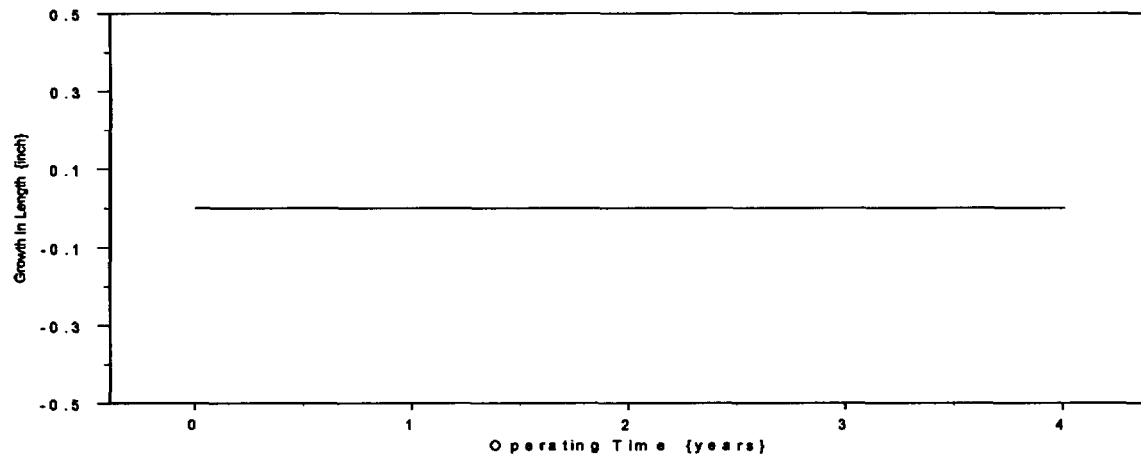
$TWC_{pwscc(j,7)} =$

-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923
-2.923

$TWC_{pwscc(j,8)} =$

-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63
-6.63





Developed by:

Verified by:

**Primary Water Stress Corrosion Crack Growth Analysis ID flaw;
Developed by Central Engineering Programs, Entergy Operations Inc.**

Developed by: J. S. Brihmadesam

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

**Component : Reactor Vessel CEDM -"28" Degree Nozzle, Mid-Plane Azimuth,
1.544" above Nozzle Bottom**

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " – between 1.0 and 300.0

**Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1 .
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .**

ID Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

Ref_{Point} := 1.544

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)*
- 2) The Center of the flaw at the reference point (Enter 2)*
- 3) The lower "C- tip" located at the reference point (Enter 3).*

Val := 2

The Input Below is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL_{Strs.Dist} := 2.999 Upper axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom).

Input Data :-

$L := 0.32$	Initial Flaw Length (Twice detectable length)
$a_0 := 0.661 \cdot 0.07$	Initial Flaw Depth (Minimum Detectable Depth was 5% TW)
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
$Years := 4$	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left| \frac{I_{lim}}{50} \right| \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:

Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "2" = Quarter Thickness Stress data at each Elevation (ksi)

Column "3" = Mid Thickness Stress data at each Elevation (ksi)

Column "4" = Three quarter Thickness Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

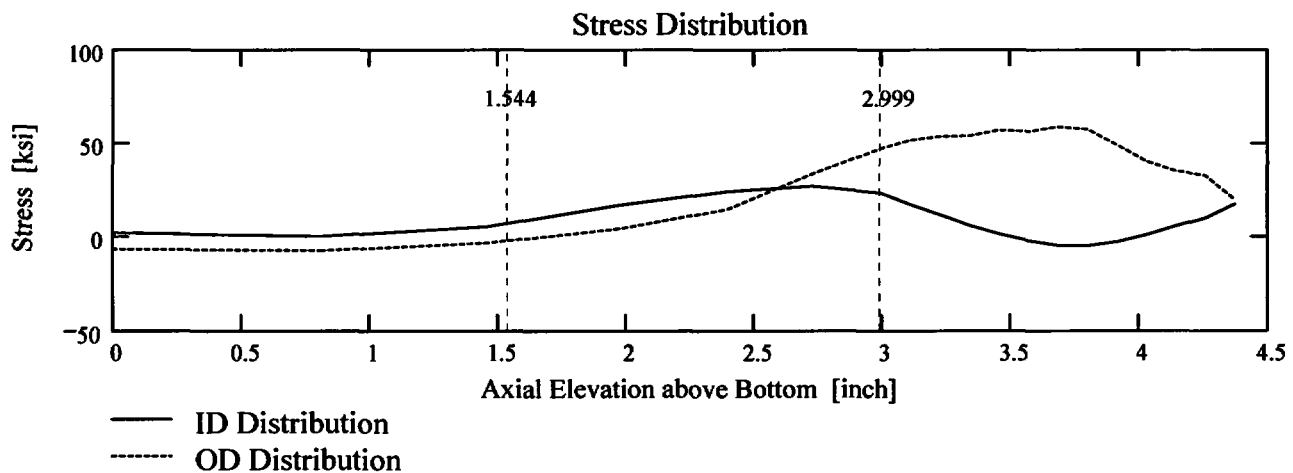
AllData :=

	0	1	2	3	4	5
0	0	2.08	-0.87	-2.96	-4.82	-6.75
1	0.81	0.09	-2.37	-4.27	-6	-7.55
2	1.46	5.28	1.69	-0.79	-2.49	-3.47
3	1.98	16.88	12.42	9.56	6.91	4.32
4	2.4	24.14	20.89	18.11	16.59	14.51
5	2.73	26.96	22.67	20.69	24.84	33.52
6	3	23.28	20.9	21.71	37.11	47.4
7	3.11	17.16	17.1	20.74	41.09	51.76
8	3.23	11.72	14.42	21.34	43.54	53.69
9	3.34	6	11.11	20.91	43.83	54.15
10	3.46	1.44	8.09	20.38	43.02	57.02
11	3.57	-2.17	5.89	19.93	42.41	56.41
12	3.69	-4.72	4.86	19.99	40.42	58.85

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Highlight the region in the above table representing the region to be selected (click on the first cell for selection and drag the mouse whilst holding the left mouse button down. Once this is done click the right mouse button and select "Copy Selection"; this will copy the selected area on to the clipboard. Then click on the "Matrix" below (to the right of the $data$ statement) to highlight the entire matrix and delete it from the edit menu. When the Mathcad input symbol appears, use the paste function in the tool bar to paste the selection.

$$Data := \begin{pmatrix} 0 & 2.079 & -0.875 & -2.96 & -4.82 & -6.75 \\ 0.811 & 0.091 & -2.37 & -4.267 & -6.004 & -7.552 \\ 1.46 & 5.283 & 1.686 & -0.786 & -2.49 & -3.469 \\ 1.98 & 16.881 & 12.419 & 9.564 & 6.907 & 4.319 \\ 2.397 & 24.144 & 20.894 & 18.115 & 16.59 & 14.513 \\ 2.731 & 26.962 & 22.672 & 20.686 & 24.842 & 33.523 \\ 2.999 & 23.279 & 20.902 & 21.706 & 37.111 & 47.395 \\ 3.113 & 17.161 & 17.101 & 20.743 & 41.091 & 51.762 \\ 3.228 & 11.722 & 14.424 & 21.34 & 43.543 & 53.688 \\ 3.343 & 6.004 & 11.108 & 20.912 & 43.833 & 54.154 \\ 3.457 & 1.439 & 8.085 & 20.38 & 43.021 & 57.025 \\ 3.572 & -2.175 & 5.891 & 19.929 & 42.405 & 56.415 \\ 3.687 & -4.725 & 4.858 & 19.994 & 40.425 & 58.85 \end{pmatrix}$$

$$Axl := Data^{(0)} \quad MD := Data^{(3)} \quad ID := Data^{(1)} \quad TQ := Data^{(4)} \quad QT := Data^{(2)} \quad OD := Data^{(5)}$$

$$R_{ID} := \text{regress}(Axl, ID, 3)$$

$$R_{QT} := \text{regress}(Axl, QT, 3)$$

$$R_{OD} := \text{regress}(Axl, OD, 3)$$

$$R_{MD} := \text{regress}(Axl, MD, 3)$$

$$R_{TQ} := \text{regress}(Axl, TQ, 3)$$


$$FL_{Cntr} := \begin{cases} Ref_{Point} - c_0 & \text{if } Val = 1 \\ Ref_{Point} & \text{if } Val = 2 \\ Ref_{Point} + c_0 & \text{otherwise} \end{cases}$$

Flaw center Location above Nozzle Bottom

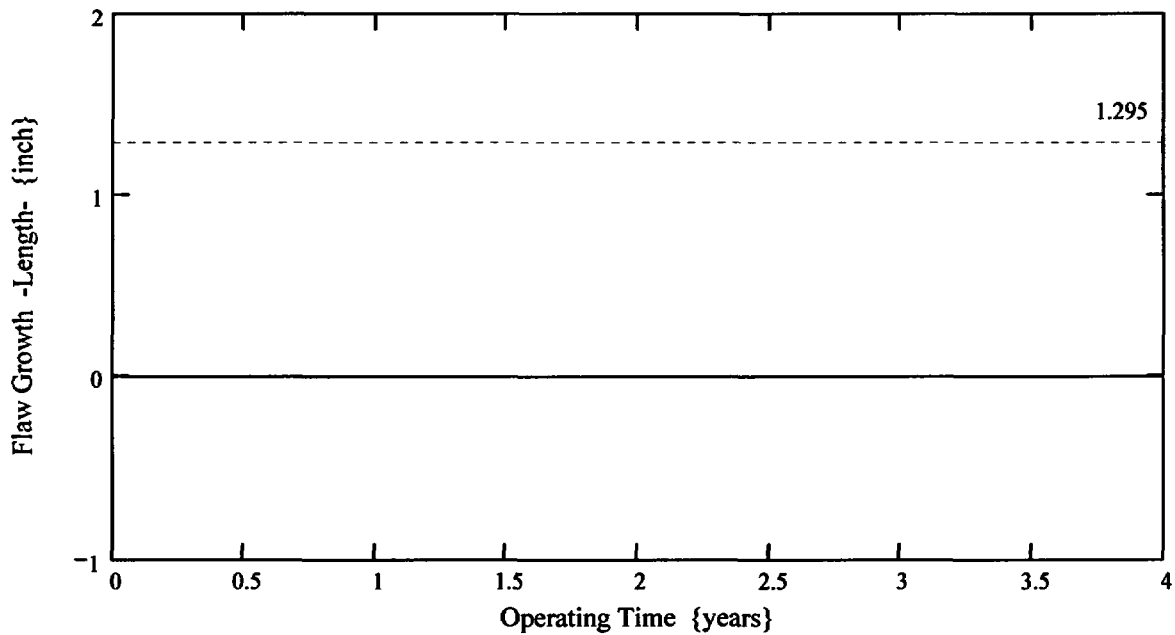
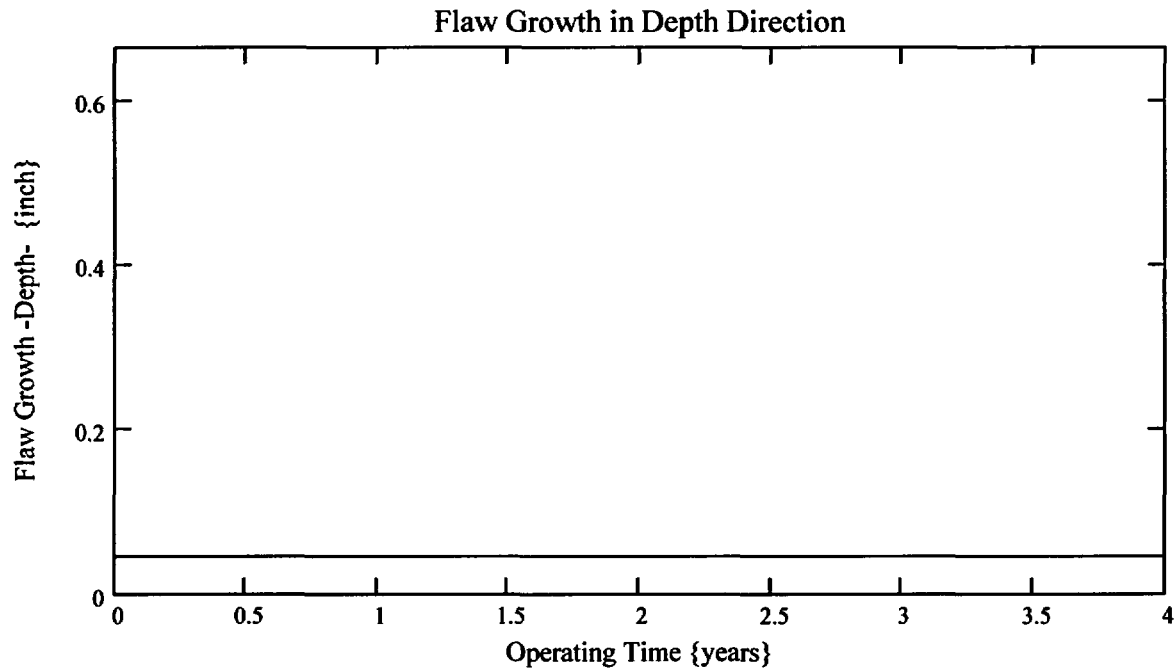
$$U_{Tip} := FL_{Cntr} + c_0$$

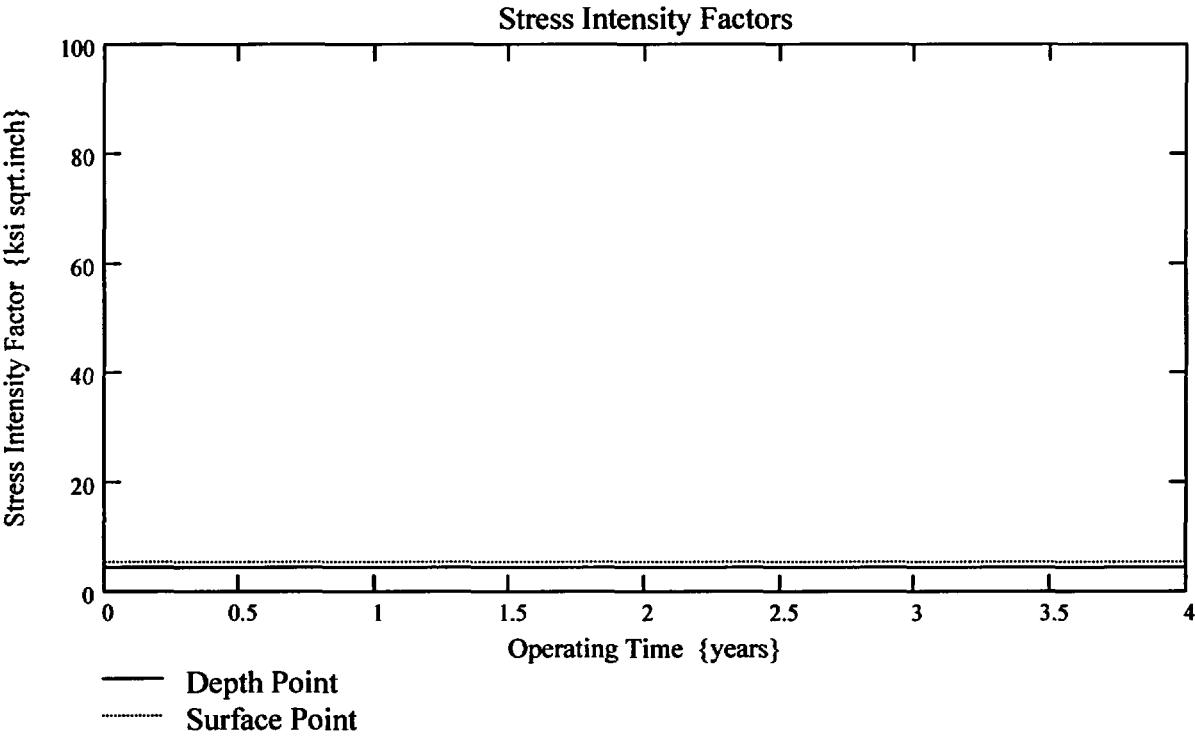
$$IncStrs.avg := \frac{UL_{Strs.Dist} - U_{Tip}}{20}$$

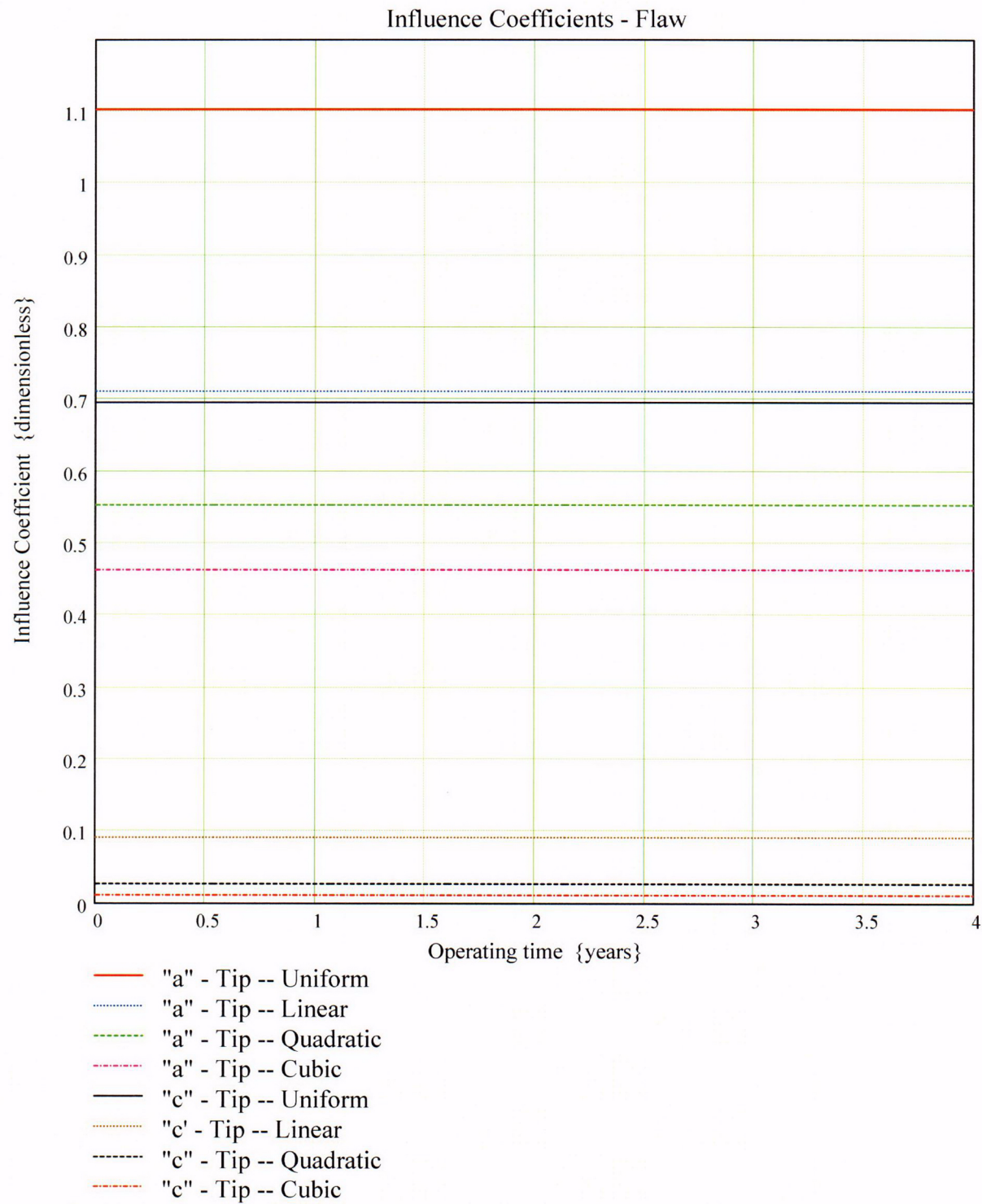
No User Input is required beyond this Point

 Sat Aug 09 10:59:39 AM 2003

PropLength = 1.295







$$\text{CGR}_{\text{sambi}(k,8)} =$$

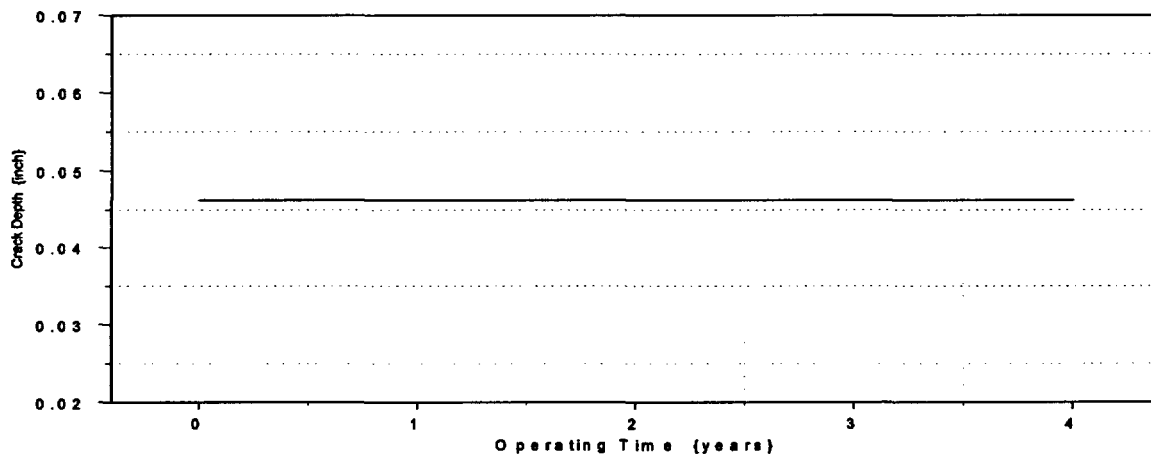
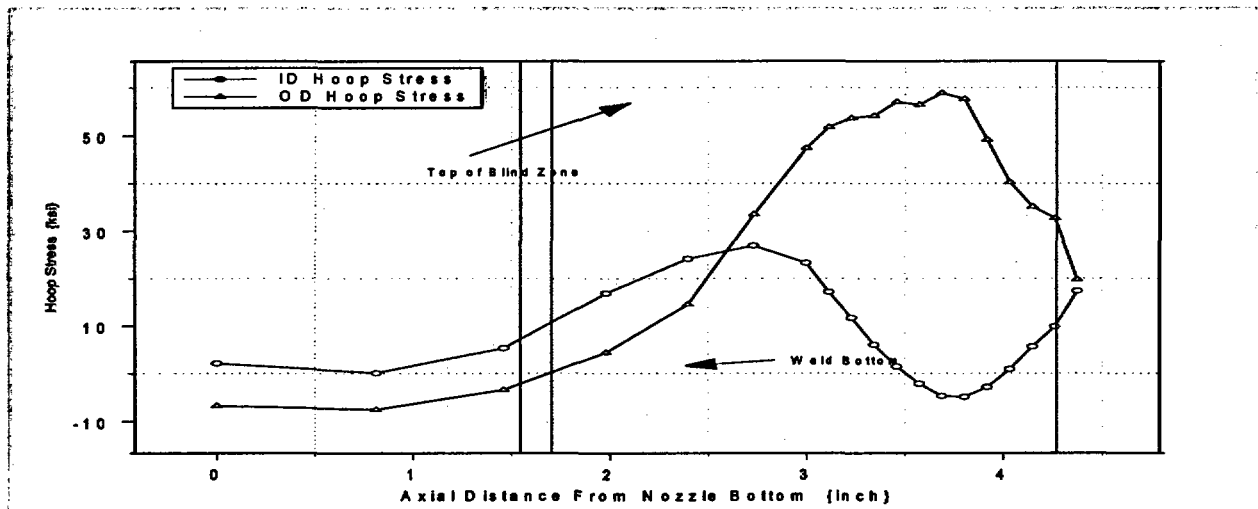
1.103
1.103
1.103
1.103
1.103
1.103
1.103
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1.103
1.103
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1.103
1.103
1.103
1.103
1.103
1.103

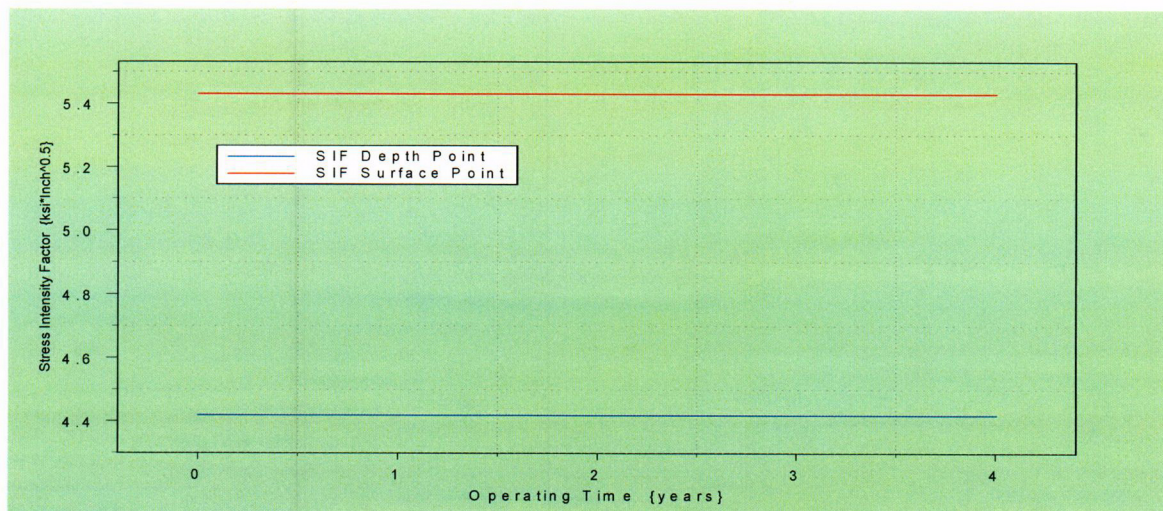
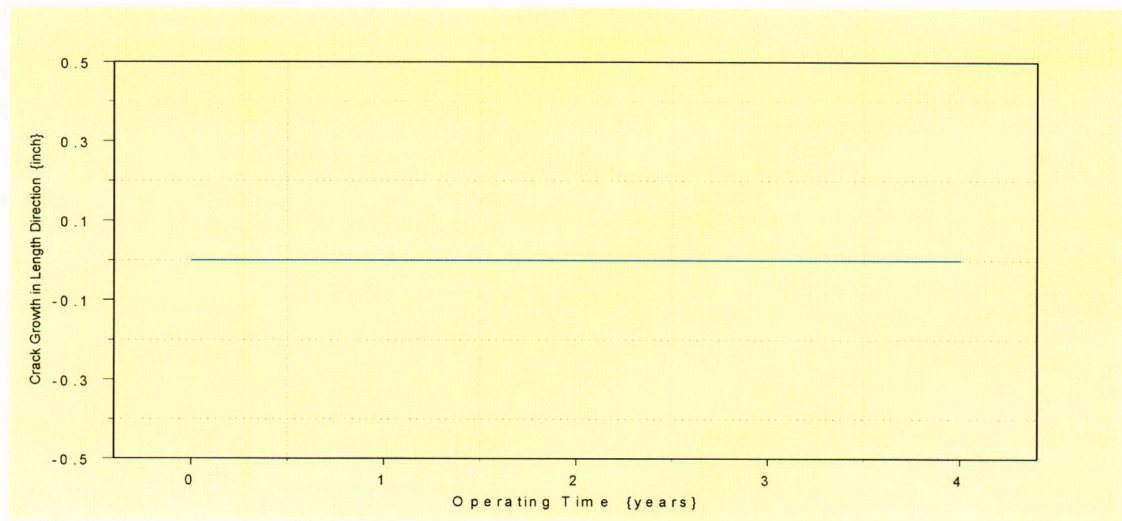
$$\text{CGR}_{\text{sambi}(k,6)} =$$

5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43
5.43

$$\text{CGR}_{\text{sambi}(k,5)} =$$

4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419
4.419





Primary Water Stress Corrosion Crack Growth Analysis - OD Surface Flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesam

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM -"28" Degree Nozzle, Mid-Plane Azimuth,
1.544" above Nozzle Bottom

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " -- between 1.0 and 300.0

Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.

The correction is applied in the determination of the crack extension to obtain the value in inch/hr .

OD Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

$Ref_{Point} := 1.544$

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)
- 2) The Center of the flaw at the reference point (Enter 2)
- 3) The lower "C- tip" located at the reference point (Enter 3).

$Val := 2$

Upper Limit to be selected for stress distribution (e.g. Weld bottom). This is the elevation from Nozzle Bottom. Enter this value below

$UL_{Strs.Dist} := 2.999$

Upper Axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.32$	Initial Flaw Length
$a_0 := 0.661 \cdot 0.12$	Initial Flaw Depth
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
$Years := 4$	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left\lceil \frac{I_{lim}}{50} \right\rceil \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:
 Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)
 Column "1" = ID Stress data at each Elevation (ksi)
 Column "2" = Quarter Thickness Stress data at each Elevation (ksi)
 Column "3" = Mid Thickness Stress data at each Elevation (ksi)
 Column "4" = Three Quarter Thickness Stress data at each Elevation (ksi)
 Column "5" = OD Stress data at each Elevation (ksi)

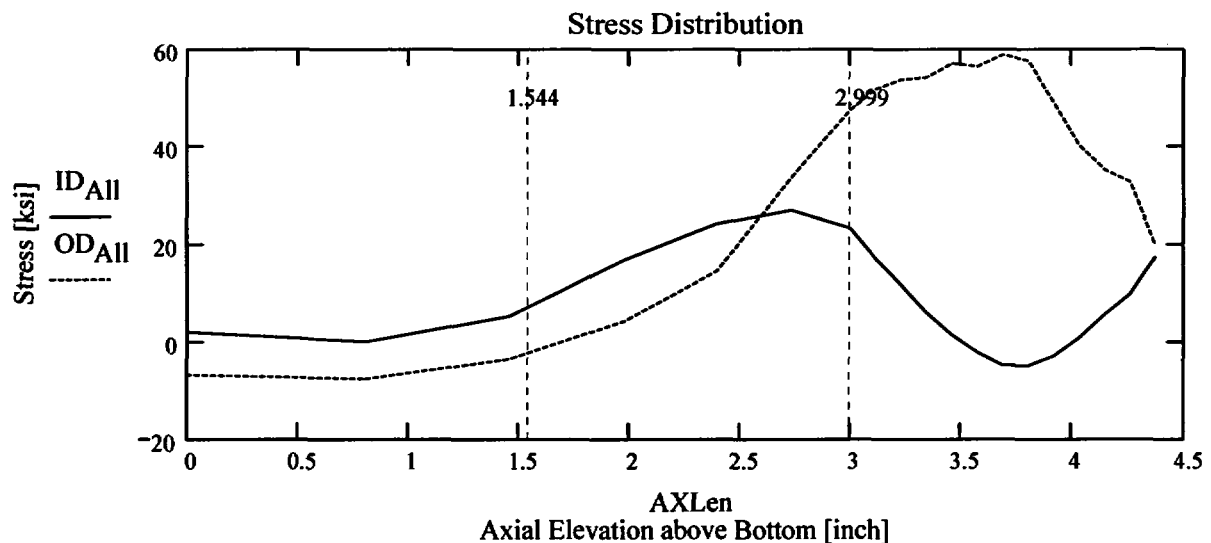
AllData :=

	0	1	2	3	4	5
0	0	2.08	-0.87	-2.96	-4.82	-6.75
1	0.81	0.09	-2.37	-4.27	-6	-7.55
2	1.46	5.28	1.69	-0.79	-2.49	-3.47
3	1.98	16.88	12.42	9.56	6.91	4.32
4	2.4	24.14	20.89	18.11	16.59	14.51
5	2.73	26.96	22.67	20.69	24.84	33.52
6	3	23.28	20.9	21.71	37.11	47.4
7	3.11	17.16	17.1	20.74	41.09	51.76
8	3.23	11.72	14.42	21.34	43.54	53.69
9	3.34	6	11.11	20.91	43.83	54.15
10	3.46	1.44	8.09	20.38	43.02	57.02
11	3.57	-2.17	5.89	19.93	42.41	56.41
12	3.69	-4.72	4.86	19.99	40.42	58.85
13	3.8	-4.92	4.88	20.34	38.45	57.62

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled Data_{All} that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

$$\text{Data} := \begin{pmatrix} 0 & 2.079 & -0.875 & -2.96 & -4.82 & -6.75 \\ 0.811 & 0.091 & -2.37 & -4.267 & -6.004 & -7.552 \\ 1.46 & 5.283 & 1.686 & -0.786 & -2.49 & -3.469 \\ 1.98 & 16.881 & 12.419 & 9.564 & 6.907 & 4.319 \\ 2.397 & 24.144 & 20.894 & 18.115 & 16.59 & 14.513 \\ 2.731 & 26.962 & 22.672 & 20.686 & 24.842 & 33.523 \\ 2.999 & 23.279 & 20.902 & 21.706 & 37.111 & 47.395 \\ 3.113 & 17.161 & 17.101 & 20.743 & 41.091 & 51.762 \\ 3.228 & 11.722 & 14.424 & 21.34 & 43.543 & 53.688 \\ 3.343 & 6.004 & 11.108 & 20.912 & 43.833 & 54.154 \\ 3.457 & 1.439 & 8.085 & 20.38 & 43.021 & 57.025 \\ 3.572 & -2.175 & 5.891 & 19.929 & 42.405 & 56.415 \\ 3.687 & -4.725 & 4.858 & 19.994 & 40.425 & 58.85 \end{pmatrix}$$

$$\text{Axl} := \text{Data}^{(0)} \quad \text{MD} := \text{Data}^{(3)} \quad \text{ID} := \text{Data}^{(1)} \quad \text{TQ} := \text{Data}^{(4)} \quad \text{QT} := \text{Data}^{(2)} \quad \text{OD} := \text{Data}^{(5)}$$

$$R_{ID} := \text{regress}(\text{Axl}, \text{ID}, 3)$$

$$R_{QT} := \text{regress}(\text{Axl}, \text{QT}, 3)$$

$$R_{OD} := \text{regress}(\text{Axl}, \text{OD}, 3)$$

$$R_{MD} := \text{regress}(\text{Axl}, \text{MD}, 3)$$

$$R_{TQ} := \text{regress}(\text{Axl}, \text{TQ}, 3)$$


$$\text{FL}_{\text{Cntr}} := \begin{cases} \text{Ref}_{\text{Point}} - c_0 & \text{if Val} = 1 \\ \text{Ref}_{\text{Point}} & \text{if Val} = 2 \\ \text{Ref}_{\text{Point}} + c_0 & \text{otherwise} \end{cases}$$

Flaw center Location Location above Nozzle Bottom

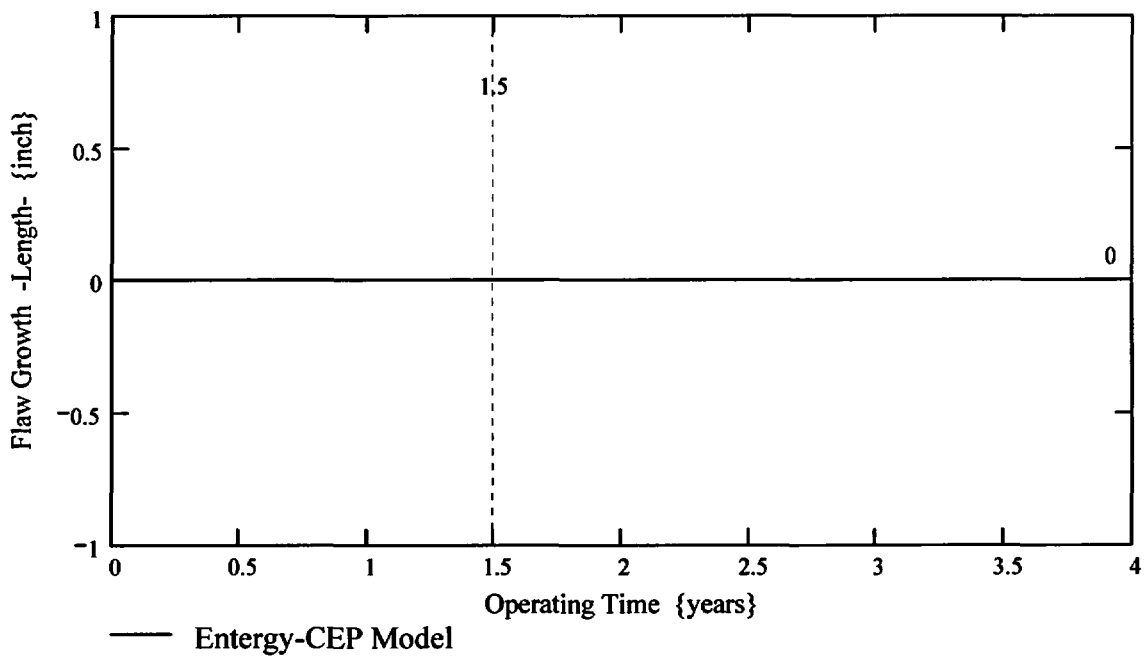
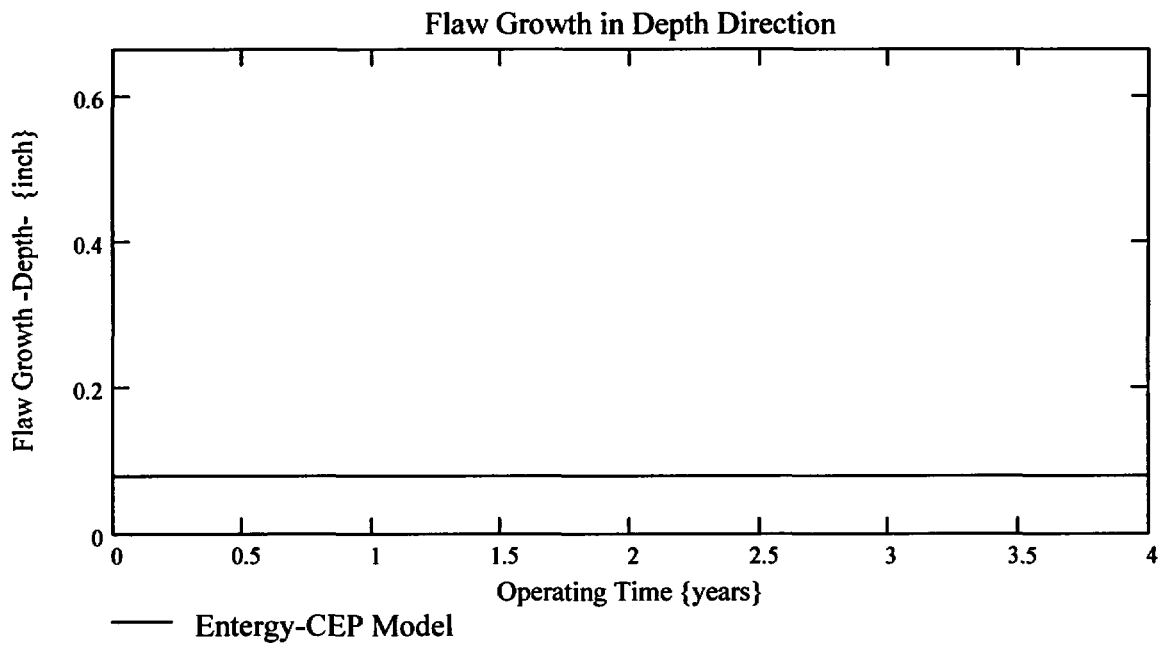
$$U_{\text{Tip}} := \text{FL}_{\text{Cntr}} + c_0$$

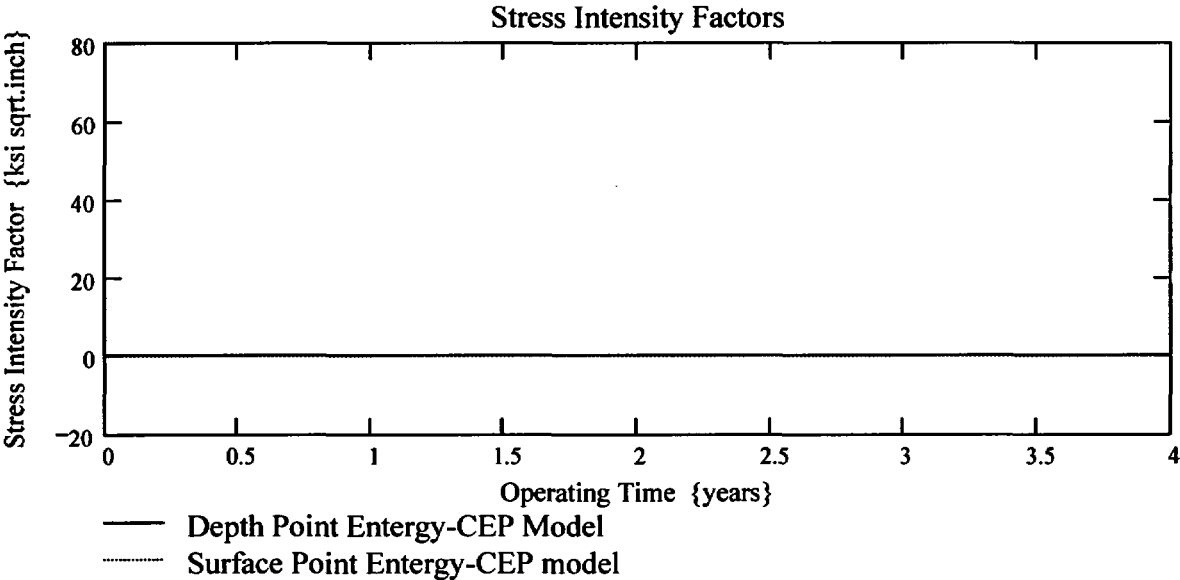
$$\text{IncStrs.avg} := \frac{U_{\text{Strs.Dist}} - U_{\text{Tip}}}{n}$$

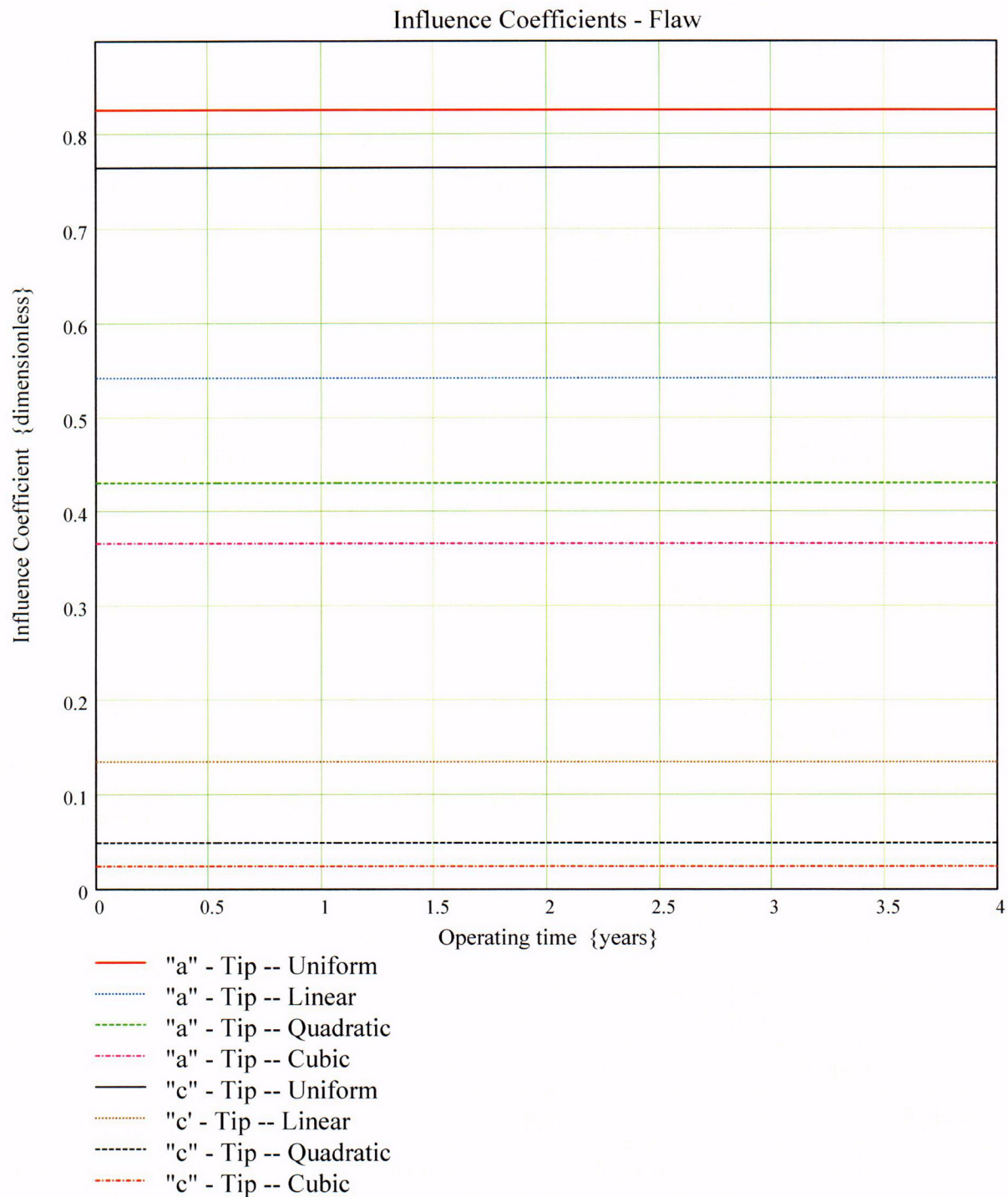
No User Input is required beyond this Point

 Sat Aug 09 10:21:18 AM 2003

PropLength = 1.295







$$CGR_{sambi(k,8)} =$$

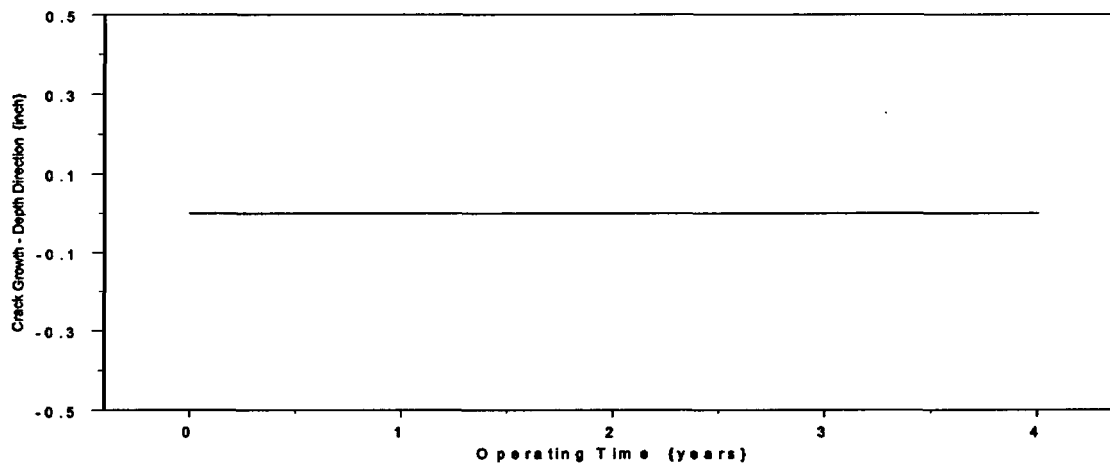
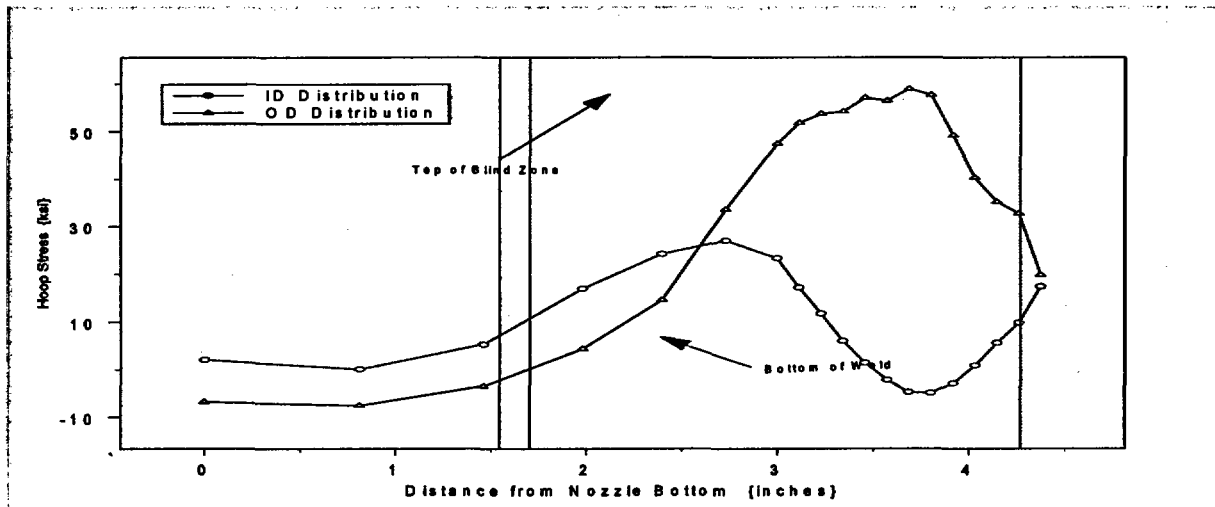
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
0.827
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0.827
0.827

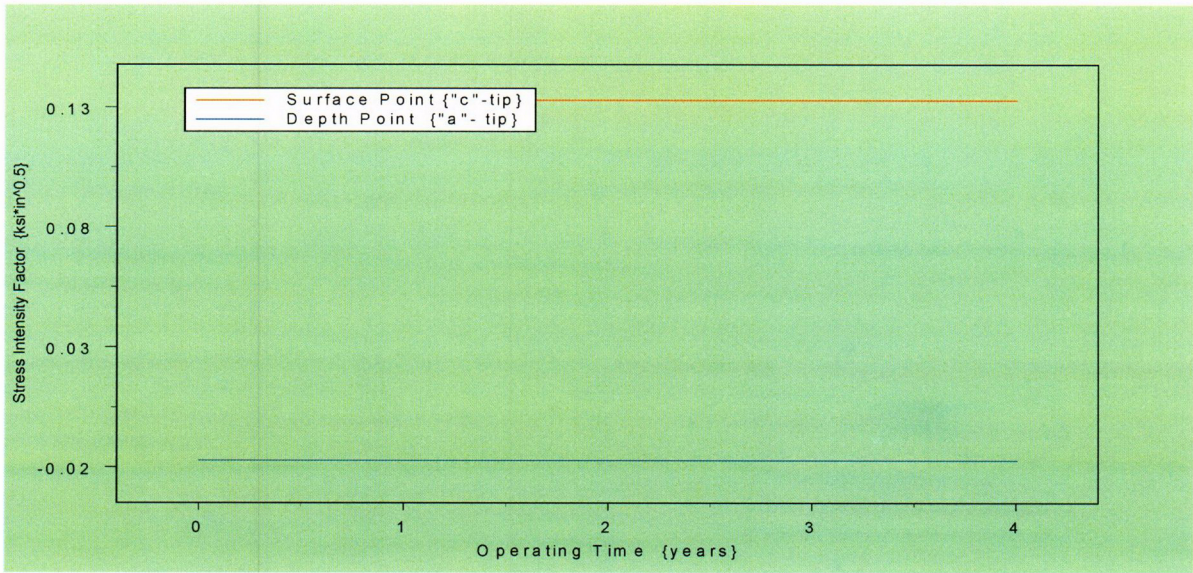
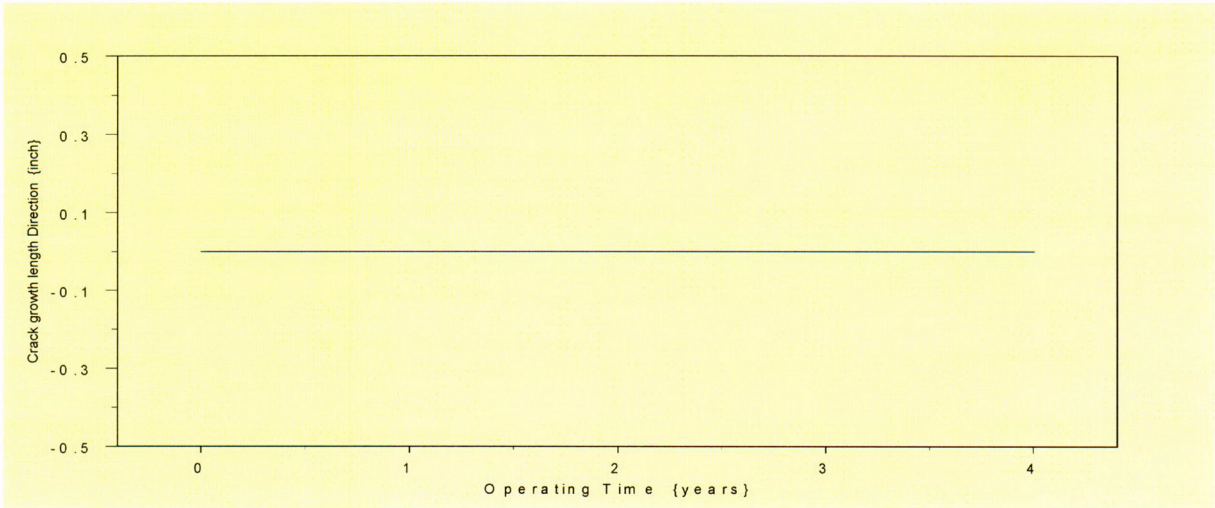
$$CGR_{sambi(k,6)} =$$

-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017
-0.017

$$CGR_{sambi(k,5)} =$$

0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133
0.133





Stress Corrosion Crack Growth Analysis Throughwall flaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesan

Verified by: B. C. Gray

Note : Only for use when $R_{outside}/t$ is between 2.0 and 5.0 (Thickwall Cylinder)

References :

- 1) ASME PVP paper PVP-350, Page 143; 1997 {Fracture Mechanics Model}
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM - "28" Degree Nozzle, Mid-Plane Azimuth,
1.544 inch above Nozzle Bottom

Calculation Reference: MRP 75 th Percentile and Flaw Pressurized

Note : *Used the Metric form of the equation from EPRI MRP 55-Rev. 1.
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .*

Through Wall Axial Flaw

The first Input is to locate the Reference Line (eg. top of the Blind Zone). The throughwall flaw "Upper Tip" is located at the Reference Line.

Enter the elevation of the Reference Line (eg. Blind Zone) above the nozzle bottom in inches.

BZ := 1.544

Location of Blind Zone above nozzle bottom (inch)

The Second Input is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL-Strs.Dist := 2.999

Upper axial Extent for Stress Distribution to be used in the analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.25$ Initial Flaw Length TW axial (Based on 10 Ksi average stress)

$od := 4.05$ Tube OD

$id := 2.728$ Tube ID

$P_{Int} := 2.235$ Design Operating Pressure (internal)

$Years := 4$ Number of Operating Years

$I_{lim} := 1500$ Iteration limit for Crack Growth loop

$T := 604$ Estimate of Operating Temperature

$\nu := 0.307$ Poissons ratio @ 600 F

$\alpha_{0c} := 2.67 \cdot 10^{-12}$ Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F

$Q_g := 31.0$ Thermal activation Energy for Crack Growth {MRP}

$T_{ref} := 617$ Reference Temperature for normalizing Data deg. F

$$C_0 := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3} \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right)} \right]} \cdot \alpha_{0c}$$

$$Tim_{opr} := Years \cdot 365 \cdot 24$$

$$R_o := \frac{od}{2}$$

$$R_i := \frac{id}{2}$$

$$t := R_o - R_i$$

$$R_m := R_i + \frac{t}{2}$$

$$CF_{inhr} := 1.417 \cdot 10^5$$

$$C_{blk} := \frac{Tim_{opr}}{I_{lim}}$$

$$Prnt_{blk} := \left\lfloor \frac{I_{lim}}{50} \right\rfloor$$

$$l := \frac{L}{2}$$

Stress Distribution in the tube. The outside surface is the reference surface for all analysis in accordance with the reference.

Stress Input Data

Import the Required data from applicable Excel spread Sheet. The column designations are as follows:

Cloumn "0" = Axial distance from Minimum to Maximum recorded on the data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

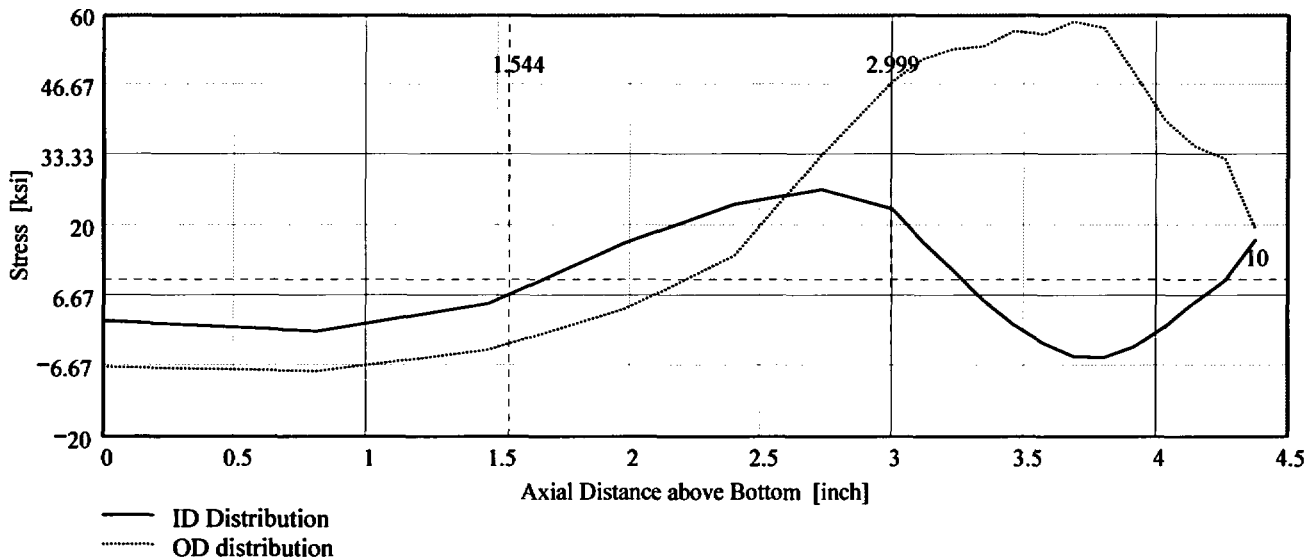
Data_{All} :=

	0	1	2	3	4	5
0	0	2.08	-0.87	-2.96	-4.82	-6.75
1	0.81	0.09	-2.37	-4.27	-6	-7.55
2	1.46	5.28	1.69	-0.79	-2.49	-3.47
3	1.98	16.88	12.42	9.56	6.91	4.32
4	2.4	24.14	20.89	18.11	16.59	14.51
5	2.73	26.96	22.67	20.69	24.84	33.52
6	3	23.28	20.9	21.71	37.11	47.4
7	3.11	17.16	17.1	20.74	41.09	51.76
8	3.23	11.72	14.42	21.34	43.54	53.69
9	3.34	6	11.11	20.91	43.83	54.15
10	3.46	1.44	8.09	20.38	43.02	57.02
11	3.57	-2.17	5.89	19.93	42.41	56.41

AllAx1 := Data_{All}⁽⁰⁾

AllID := Data_{All}⁽¹⁾

AllOD := Data_{All}⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

	0	2.079	-0.875	-2.96	-4.82	-6.75
	0.811	0.091	-2.37	-4.267	-6.004	-7.552
	1.46	5.283	1.686	-0.786	-2.49	-3.469
	1.98	16.881	12.419	9.564	6.907	4.319
	2.397	24.144	20.894	18.115	16.59	14.513
Data :=	2.731	26.962	22.672	20.686	24.842	33.523
	2.999	23.279	20.902	21.706	37.111	47.395
	3.113	17.161	17.101	20.743	41.091	51.762
	3.228	11.722	14.424	21.34	43.543	53.688
	3.343	6.004	11.108	20.912	43.833	54.154
	3.457	1.439	8.085	20.38	43.021	57.025

$Axl := Data^{(0)}$

$ID := Data^{(1)}$

$OD := Data^{(5)}$

$R_{ID} := \text{regress}(Axl, ID, 3)$

$R_{OD} := \text{regress}(Axl, OD, 3)$

$FL_{Cntr} := BZ - I$

Flaw Center above Nozzle Bottom

$$IncStrs.avg := \frac{ULStrs.Dist - BZ}{20}$$

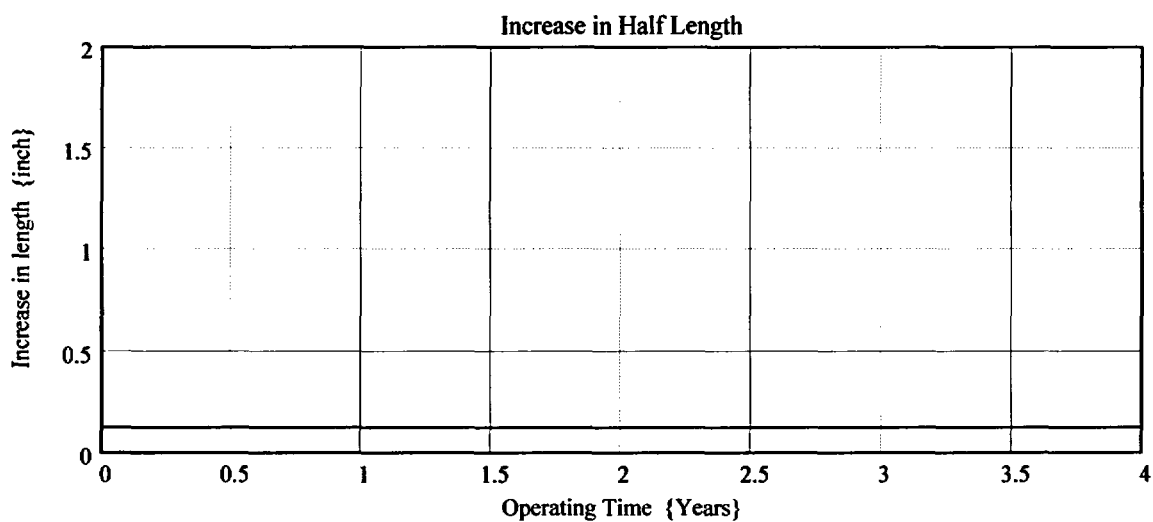
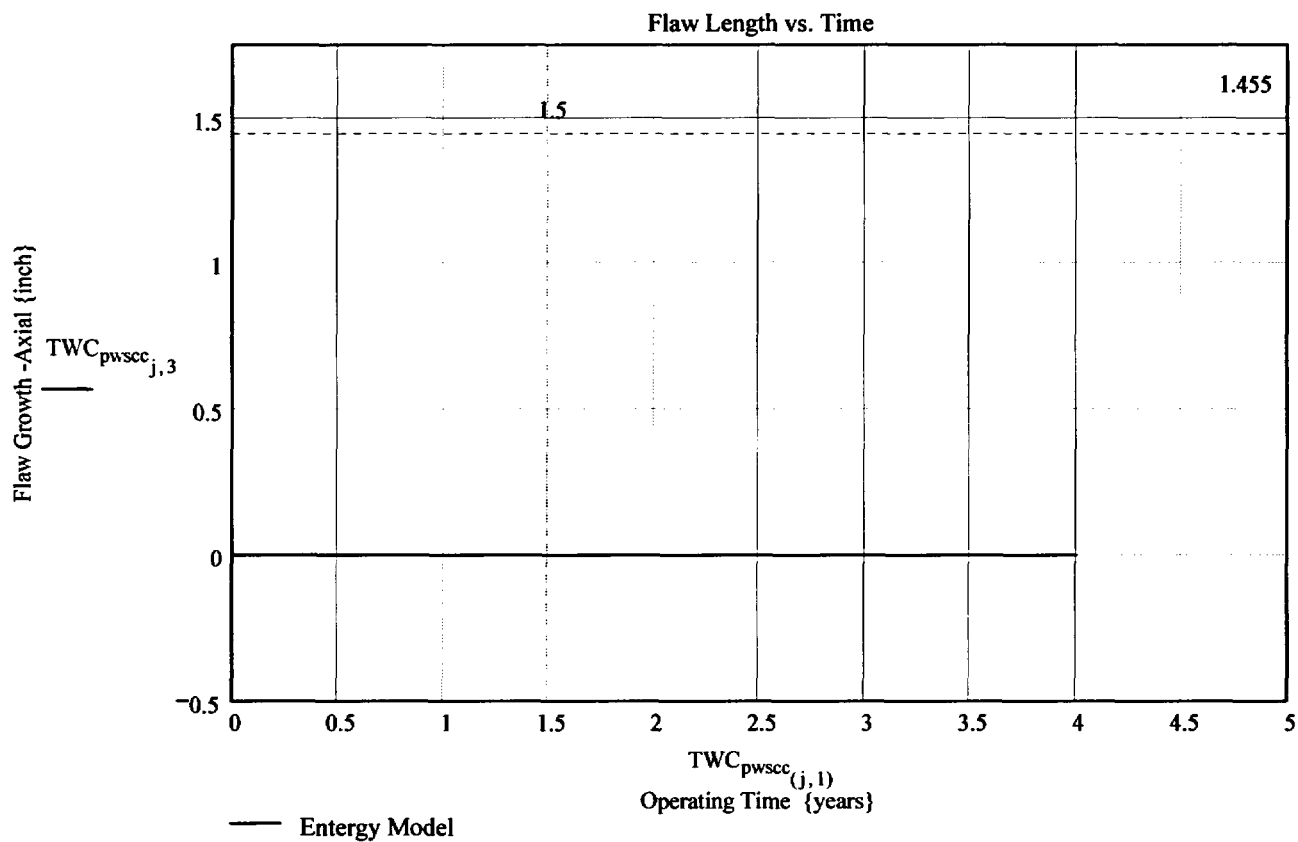
No User Input required beyond this Point

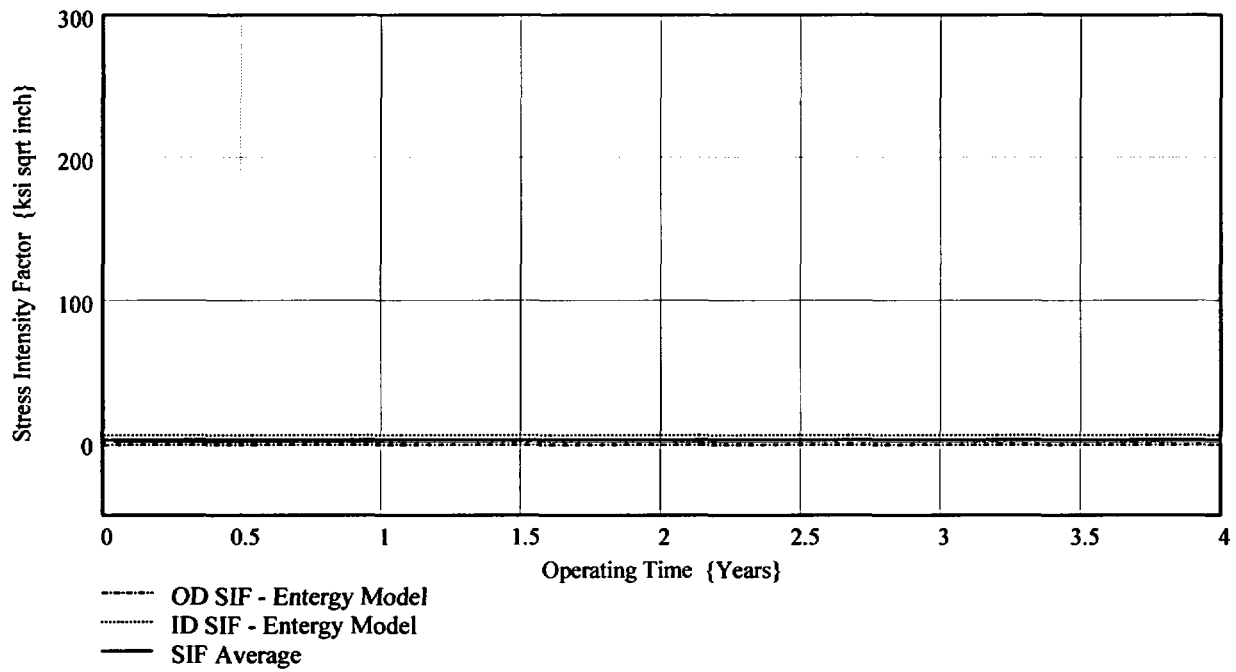
 Sat Aug 09 11:44:49 AM 2003

Developed by:

Verified by:

PropLength = 1.455





$TWC_{pwscc(j,6)} =$

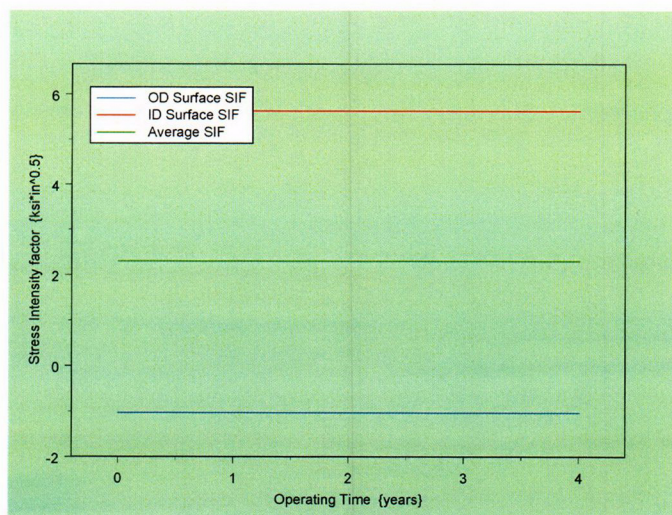
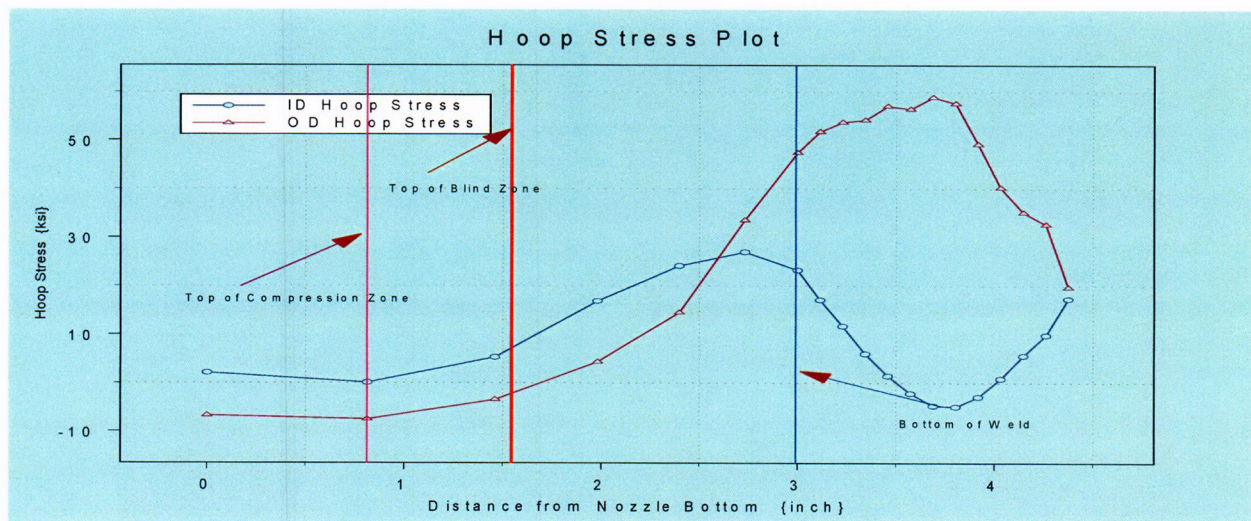
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027
-1.027

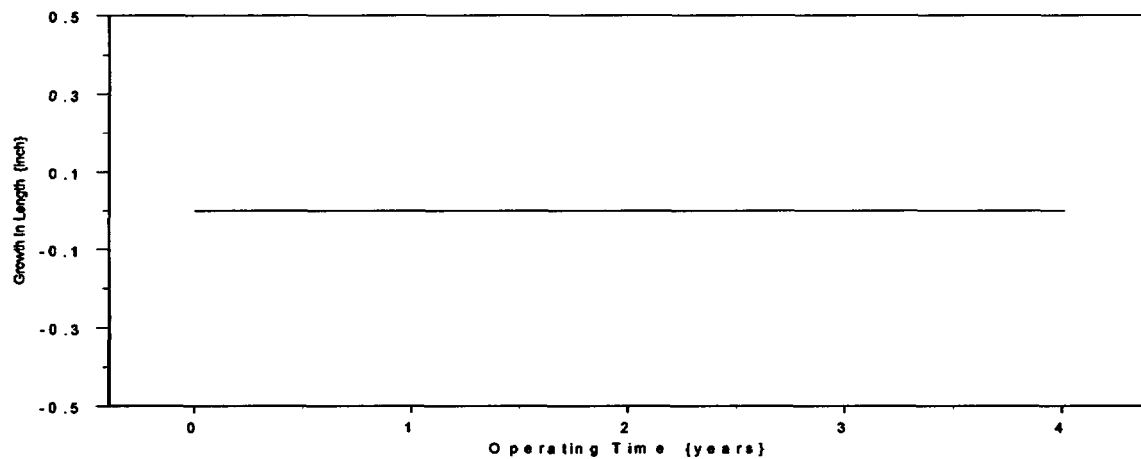
$TWC_{pwscc(j,7)} =$

5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625
5.625

$TWC_{pwscc(j,8)} =$

2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389
2.389





Developed by:

Verified by:

**Primary Water Stress Corrosion Crack Growth Analysis ID flaw;
Developed by Central Engineering Programs, Entergy Operations Inc.**

Developed by: J. S. Brihmadesam

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

**Component : Reactor Vessel CEDM -"49" Degree Nozzle, Downhill Azimuth,
1.544" above Nozzle Bottom**

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " – between 1.0 and 300.0

**Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .**

ID Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

Ref_{point} := 1.544

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)*
- 2) The Center of the flaw at the reference point (Enter 2)*
- 3) The lower "C- tip" located at the reference point (Enter 3).*

Val := 2

The Input Below is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL_{Strs.Dist} := 1.889 Upper axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom).

Input Data :-

$L := 0.32$	Initial Flaw Length (Twice detectable length)
$a_0 := 0.661 \cdot 0.07$	Initial Flaw Depth (Minimum Detectable Depth was 5% TW)
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
Years := 4	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := \text{Years} \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left| \frac{I_{lim}}{50} \right| \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:

Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "2" = Quarter Thickness Stress data at each Elevation (ksi)

Column "3" = Mid Thickness Stress data at each Elevation (ksi)

Column "4" = Three quarter Thickness Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

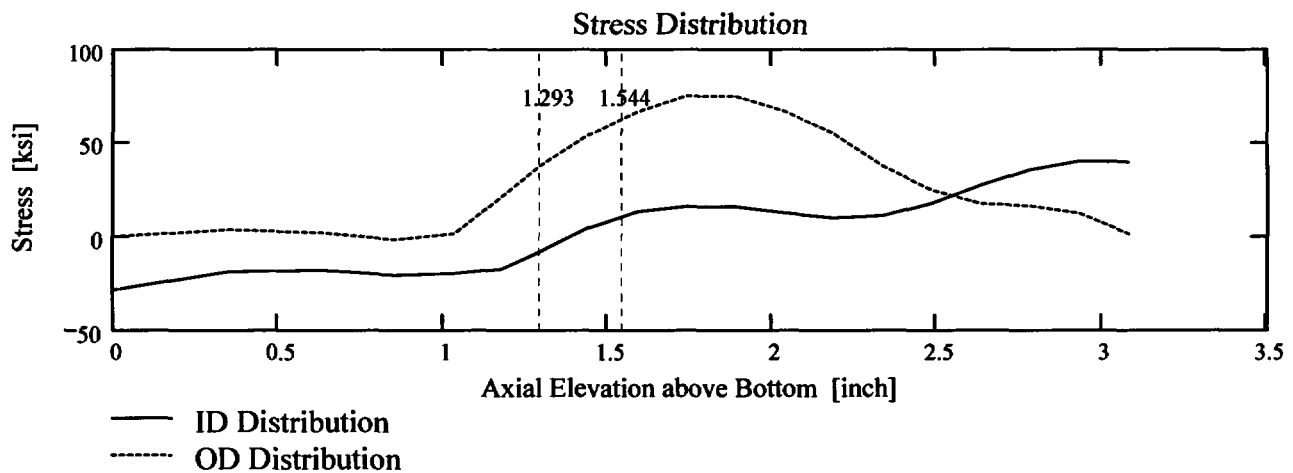
AllData :=

	0	1	2	3	4	5
0	0	-28.32	-18.3	-12.16	-6.2	-0.02
1	0.35	-18.79	-12.49	-6.61	-1.37	3.65
2	0.63	-17.84	-10.52	-4.41	-0.48	2.08
3	0.85	-20.52	-12.97	-5.9	-0.87	-1.54
4	1.03	-19.66	-11.83	-5.29	0.23	1.46
5	1.18	-17.2	-10.59	-0.52	16.33	21.02
6	1.29	-8.02	-2.2	10.46	32.66	37.29
7	1.44	4.78	9.56	24.9	38.18	54.09
8	1.59	13.25	18.57	35.28	52.81	66.52
9	1.74	16	22.02	39.19	62.95	75
10	1.89	15.86	23.14	40.23	64.33	74.87

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Highlight the region in the above table representing the region to be selected (click on the first cell for selection and drag the mouse whilst holding the left mouse button down. Once this is done click the right mouse button and select "Copy Selection"; this will copy the selected area on to the clipboard. Then click on the "Matrix" below (to the right of the $data$ statement) to highlight the entire matrix and delete it from the edit menu. When the Mathcad input symbol appears, use the paste function in the tool bar to paste the selection.

$$Data := \begin{pmatrix} 0 & -28.324 & -18.299 & -12.16 & -6.201 & -0.021 \\ 0.35 & -18.794 & -12.495 & -6.607 & -1.366 & 3.655 \\ 0.63 & -17.838 & -10.518 & -4.407 & -0.477 & 2.08 \\ 0.854 & -20.517 & -12.968 & -5.902 & -0.874 & -1.536 \\ 1.034 & -19.663 & -11.831 & -5.288 & 0.227 & 1.46 \\ 1.178 & -17.203 & -10.587 & -0.515 & 16.326 & 21.019 \\ 1.293 & -8.023 & -2.205 & 10.461 & 32.658 & 37.289 \\ 1.442 & 4.778 & 9.557 & 24.903 & 38.177 & 54.089 \\ 1.591 & 13.252 & 18.569 & 35.278 & 52.808 & 66.517 \\ 1.74 & 16.001 & 22.017 & 39.194 & 62.945 & 75.001 \\ 1.889 & 15.857 & 23.14 & 40.235 & 64.335 & 74.874 \end{pmatrix}$$

$$Axl := Data^{(0)} \quad MD := Data^{(3)} \quad ID := Data^{(1)} \quad TQ := Data^{(4)} \quad QT := Data^{(2)} \quad OD := Data^{(5)}$$

$$R_{ID} := \text{regress}(Axl, ID, 3)$$

$$R_{QT} := \text{regress}(Axl, QT, 3)$$

$$R_{OD} := \text{regress}(Axl, OD, 3)$$

$$R_{MD} := \text{regress}(Axl, MD, 3)$$


$$R_{TQ} := \text{regress}(Axl, TQ, 3)$$

$$FL_{Cntr} := \begin{cases} Ref_{Point} - c_0 & \text{if } Val = 1 \\ Ref_{Point} & \text{if } Val = 2 \\ Ref_{Point} + c_0 & \text{otherwise} \end{cases} \quad \text{Flaw center Location above Nozzle Bottom}$$

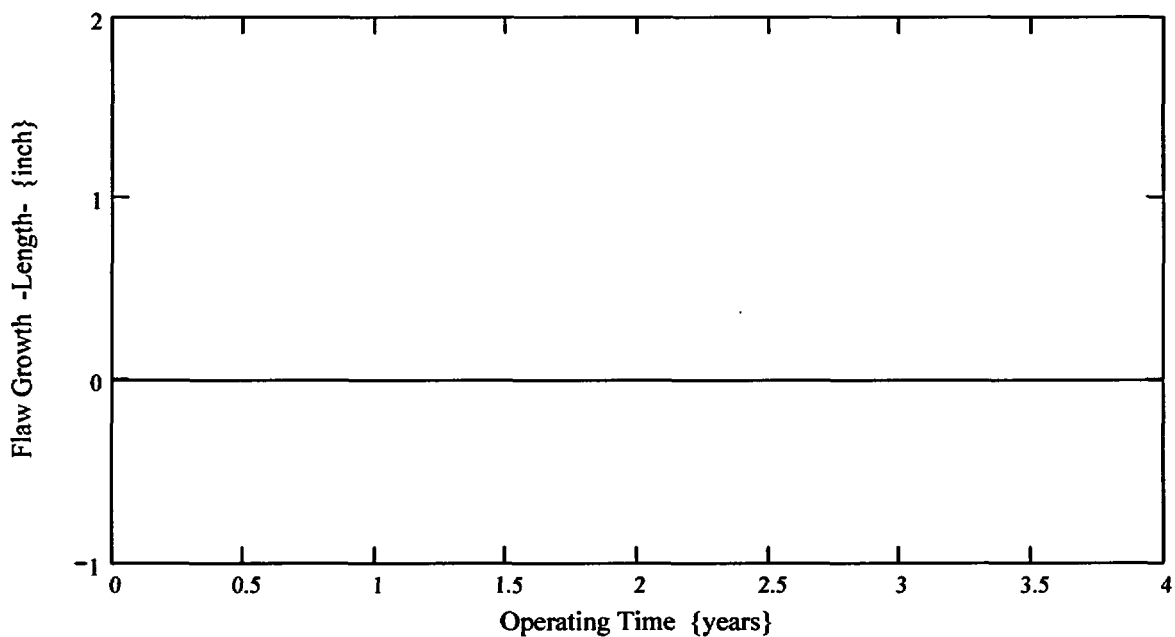
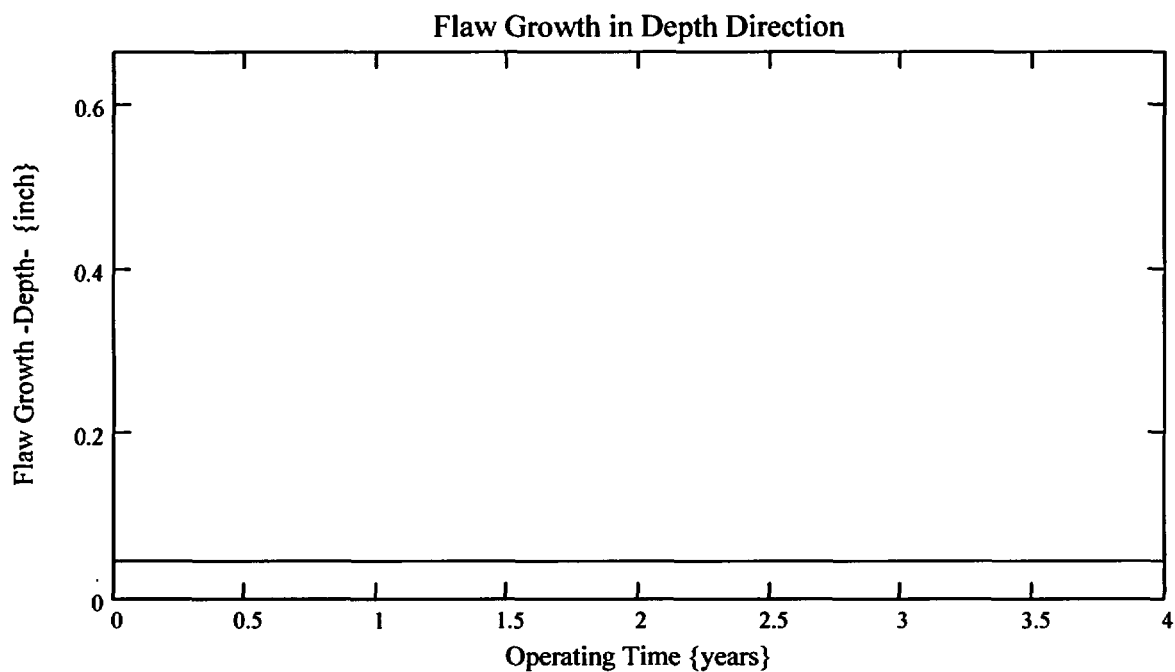
$$U_{Tip} := FL_{Cntr} + c_0$$

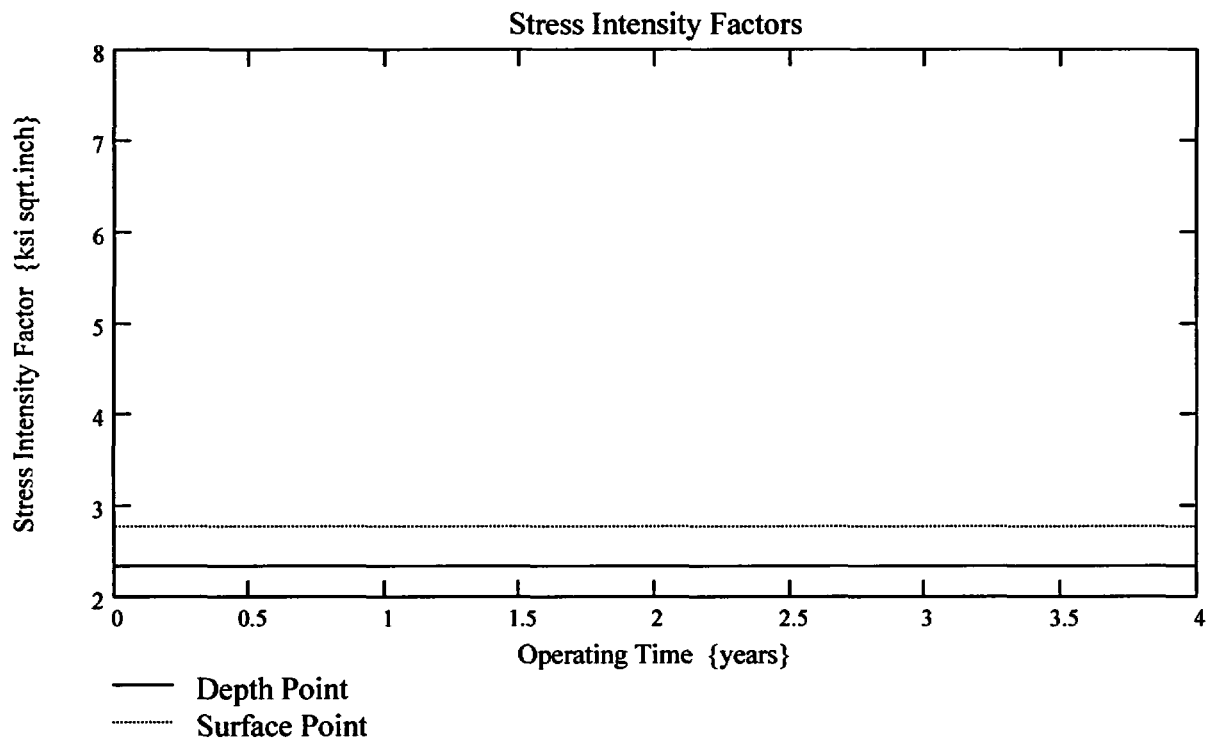
$$Inc_{Strs.avg} := \frac{UL_{Strs.Dist} - U_{Tip}}{20}$$

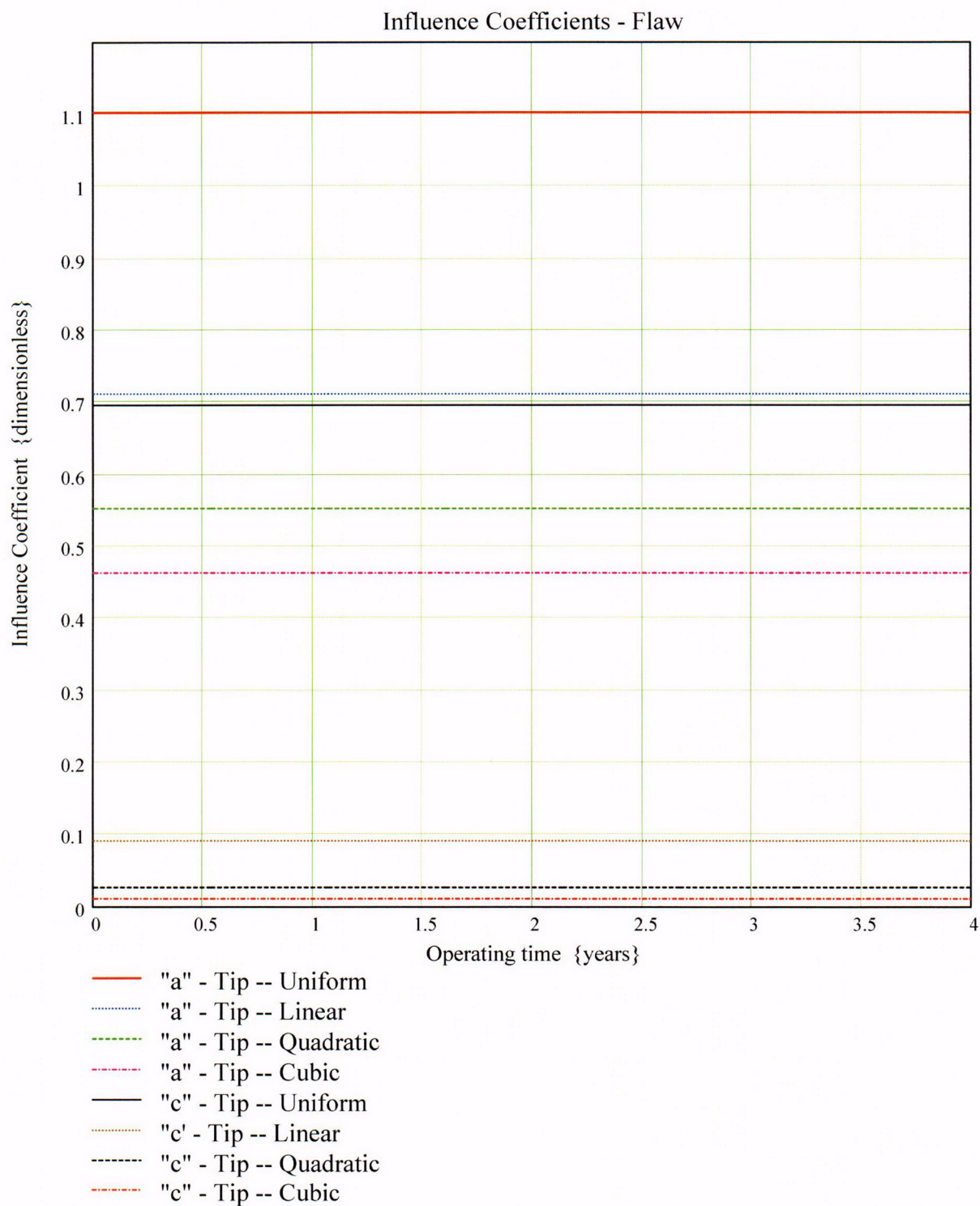
No User Input is required beyond this Point

 Sat Aug 09 10:59:39 AM 2003

$$\text{PropLength} = 0.185$$







$CGR_{sambi(k,8)} =$

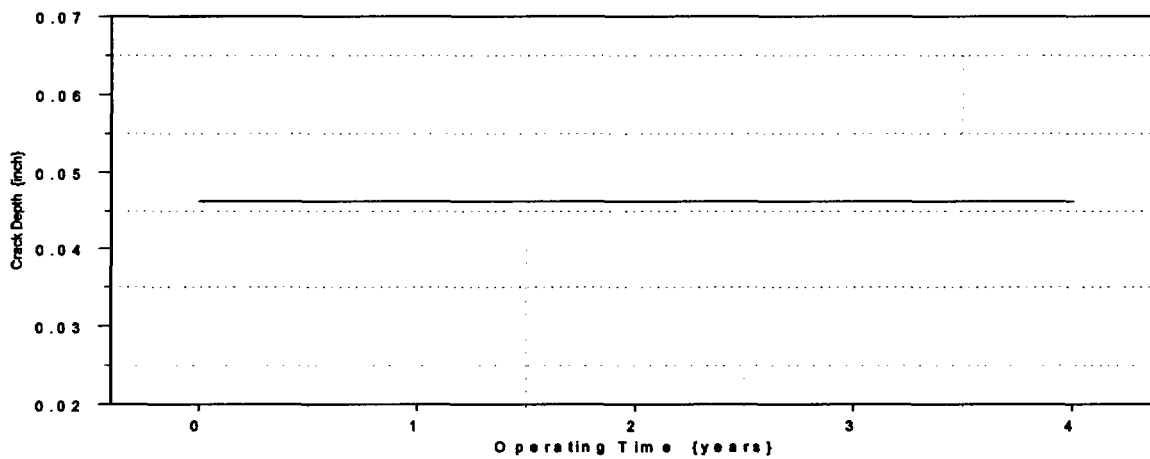
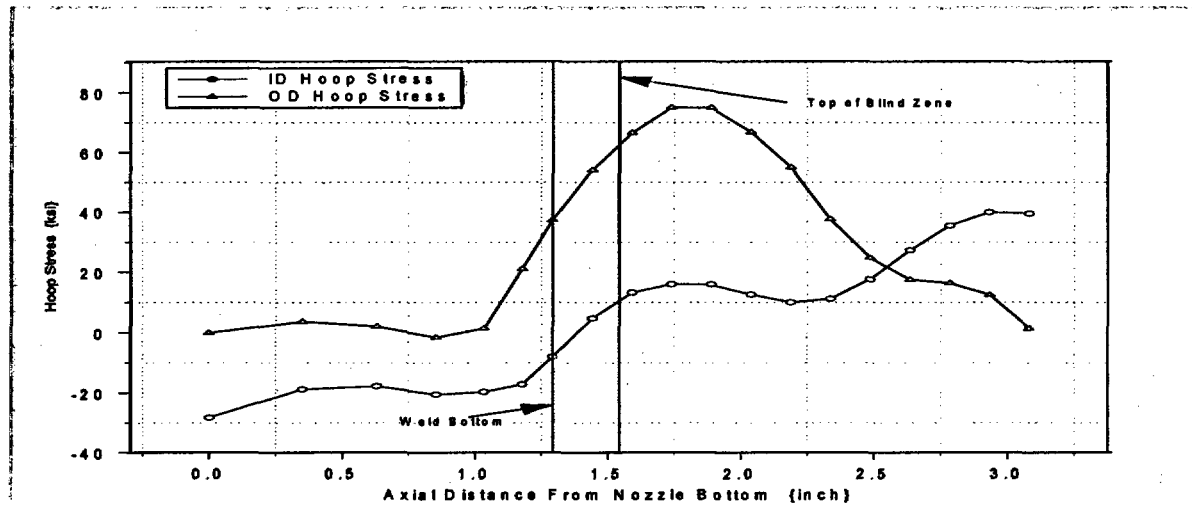
1.103
1.103
1.103
1.103
1.103
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1.103
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1.103
1.103

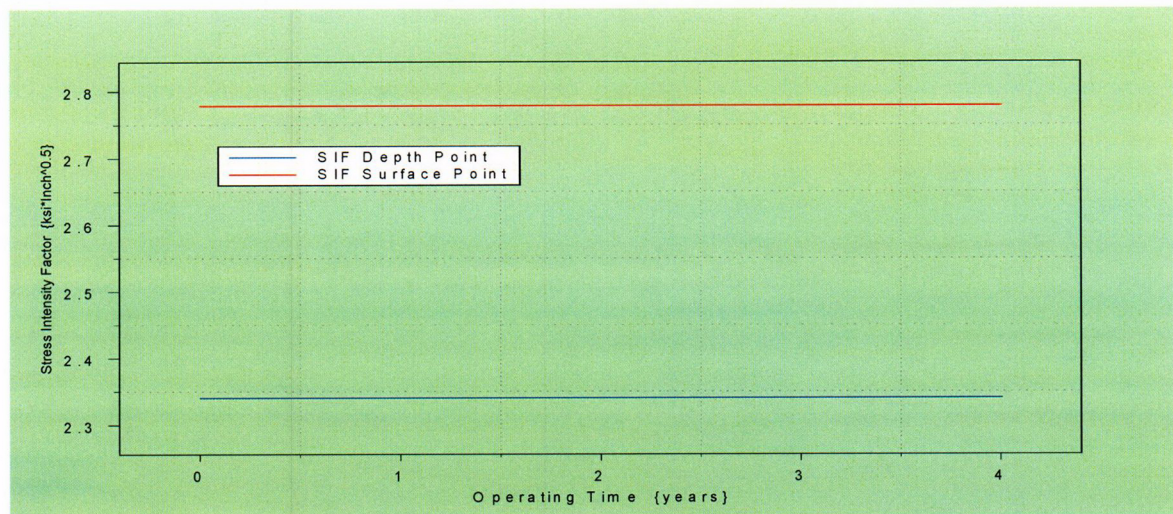
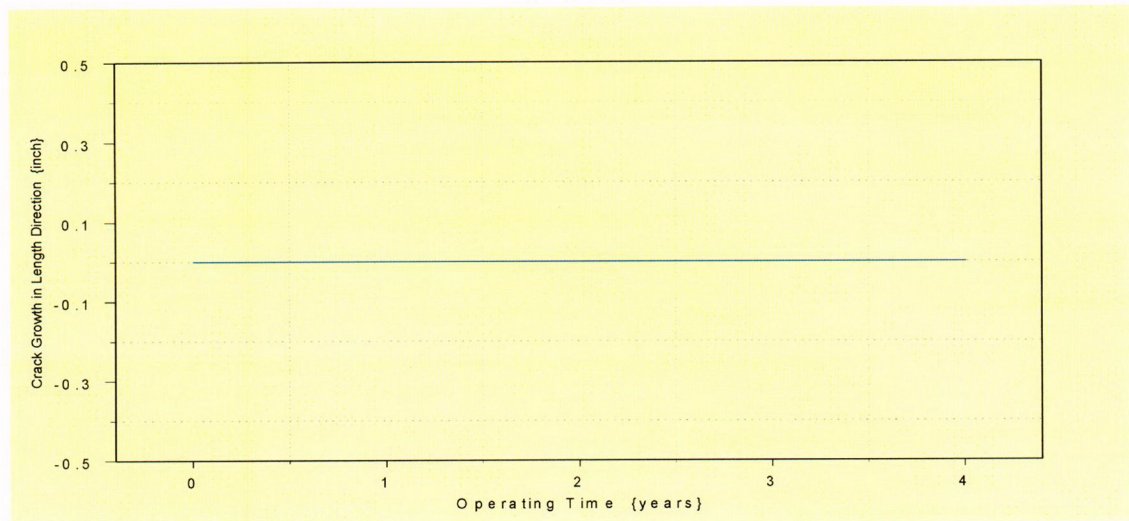
$CGR_{sambi(k,6)} =$

2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778
2.778

$CGR_{sambi(k,5)} =$

2.34
2.34
2.34
2.34
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2.34
2.34





**Primary Water Stress Corrosion Crack Growth Analysis ID flaw;
Developed by Central Engineering Programs, Entergy Operations Inc.**

Developed by: J. S. Brihmadesam

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

**Component : Reactor Vessel CEDM -"49" Degree Nozzle, Uphill Azimuth,
1.544" above Nozzle Bottom**

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " – between 1.0 and 300.0

**Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1 .
The correction is applied in the determination of the crack extension to
obtain the value in inch/hr .**

ID Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

Ref_{Point} := 1.544

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)*
- 2) The Center of the flaw at the reference point (Enter 2)*
- 3) The lower "C- tip" located at the reference point (Enter 3).*

Val := 2

The Input Below is the Upper Limit for the evaluation, which is the bottom of the fillet weld leg. This is shown on the Excel spread sheet as weld bottom. Enter this dimension (measured from nozzle bottom) below.

UL_{Strs.Dist} := 6.628 Upper axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom).

Input Data :-

$L := 0.32$	Initial Flaw Length (Twice detectable length)
$a_0 := 0.661 \cdot 0.07$	Initial Flaw Depth (Minimum Detectable Depth was 5% TW)
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
Years := 4	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := \text{Years} \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left\lfloor \frac{I_{lim}}{50} \right\rfloor \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:

Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)

Column "1" = ID Stress data at each Elevation (ksi)

Column "2" = Quarter Thickness Stress data at each Elevation (ksi)

Column "3" = Mid Thickness Stress data at each Elevation (ksi)

Column "4" = Three quarter Thickness Stress data at each Elevation (ksi)

Column "5" = OD Stress data at each Elevation (ksi)

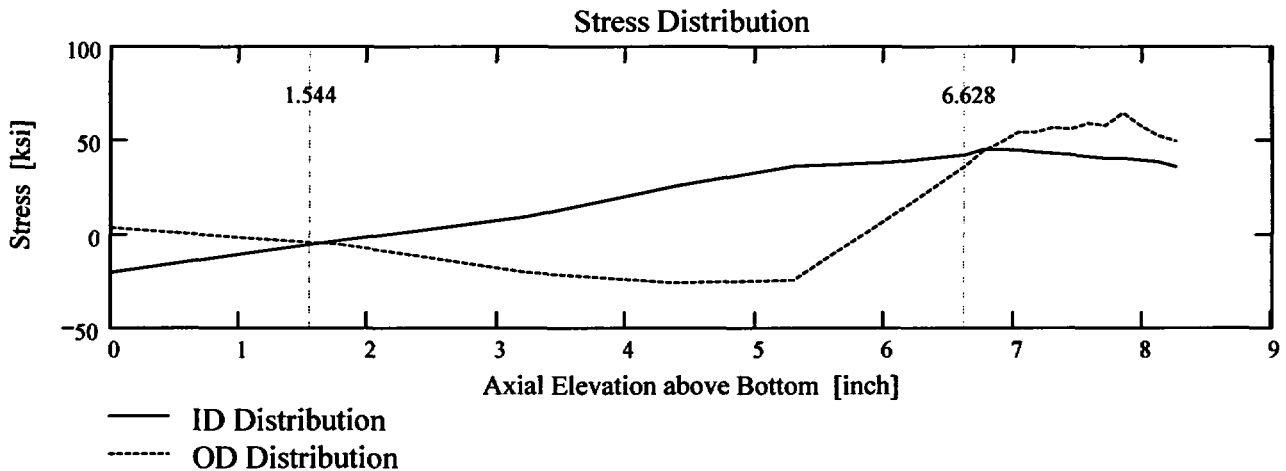
AllData :=

	0	1	2	3	4	5
0	0	-20.18	-11.45	-5.94	-1.16	3.7
1	1.79	-3.02	-4.38	-5.44	-5.51	-5.34
2	3.23	9.4	12.13	-0.26	-12.62	-20.23
3	4.38	25.65	24.71	14.58	-15.3	-25.69
4	5.3	36.18	33.79	26.29	-5.92	-24.31
5	6.04	38.11	35.03	31.43	21.21	8.83
6	6.63	42.19	38.1	36.25	40.68	36.41
7	6.76	45.07	42.22	42.74	47.55	44.23
8	6.9	44.97	43.61	46.01	49.99	48.8
9	7.03	44.7	44.12	47.02	51.04	54.11

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled $Data_{All}$ that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Highlight the region in the above table representing the region to be selected (click on the first cell for selection and drag the mouse whilst holding the left mouse button down. Once this is done click the right mouse button and select "Copy Selection"; this will copy the selected area on to the clipboard. Then click on the "Matrix" below (to the right of the $data$ statement) to highlight the entire matrix and delete it from the edit menu. When the Mathcad input symbol appears, use the paste function in the tool bar to paste the selection.

$$Data := \begin{pmatrix} 0 & -20.175 & -11.45 & -5.94 & -1.163 & 3.704 \\ 1.792 & -3.024 & -4.378 & -5.443 & -5.511 & -5.341 \\ 3.228 & 9.398 & 12.134 & -0.258 & -12.622 & -20.232 \\ 4.378 & 25.65 & 24.71 & 14.577 & -15.299 & -25.689 \\ 5.299 & 36.179 & 33.787 & 26.287 & -5.925 & -24.306 \\ 6.037 & 38.106 & 35.028 & 31.43 & 21.215 & 8.834 \\ 6.628 & 42.186 & 38.102 & 36.248 & 40.684 & 36.405 \\ 6.764 & 45.067 & 42.217 & 42.736 & 47.553 & 44.235 \\ 6.899 & 44.968 & 43.606 & 46.007 & 49.995 & 48.803 \end{pmatrix}$$

$$A_{x1} := Data^{(0)} \quad MD := Data^{(3)} \quad ID := Data^{(1)} \quad TQ := Data^{(4)} \quad QT := Data^{(2)} \quad OD := Data^{(5)}$$

$$R_{ID} := \text{regress}(A_{x1}, ID, 3)$$

$$R_{QT} := \text{regress}(A_{x1}, QT, 3)$$

$$R_{OD} := \text{regress}(A_{x1}, OD, 3)$$

$$R_{MD} := \text{regress}(A_{x1}, MD, 3)$$

$$R_{TQ} := \text{regress}(A_{x1}, TQ, 3)$$

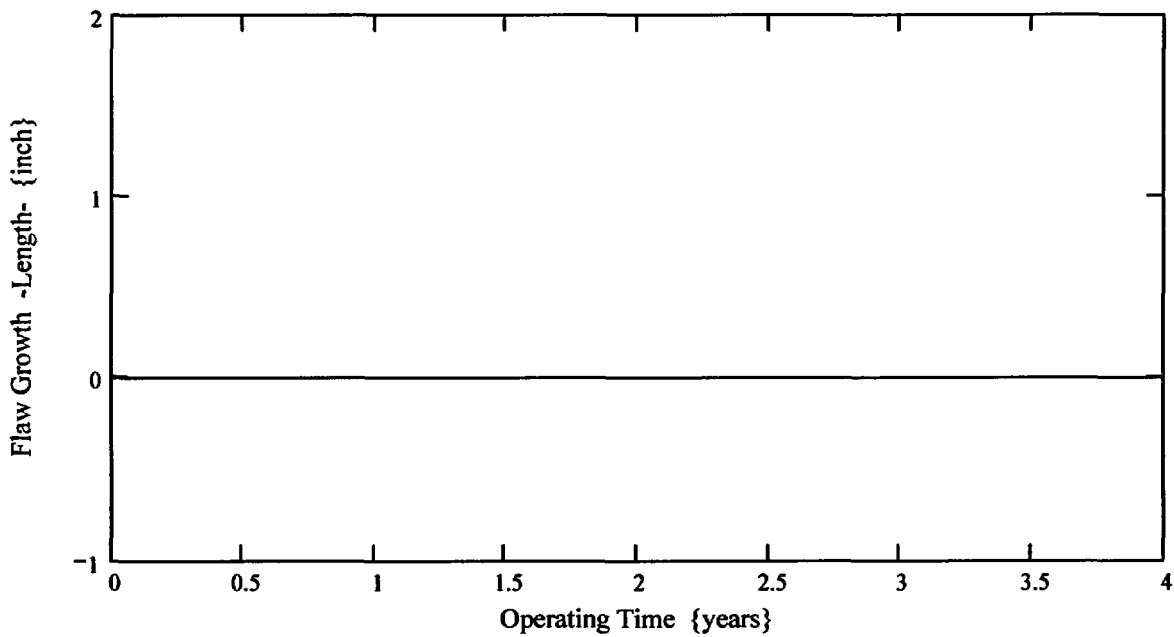
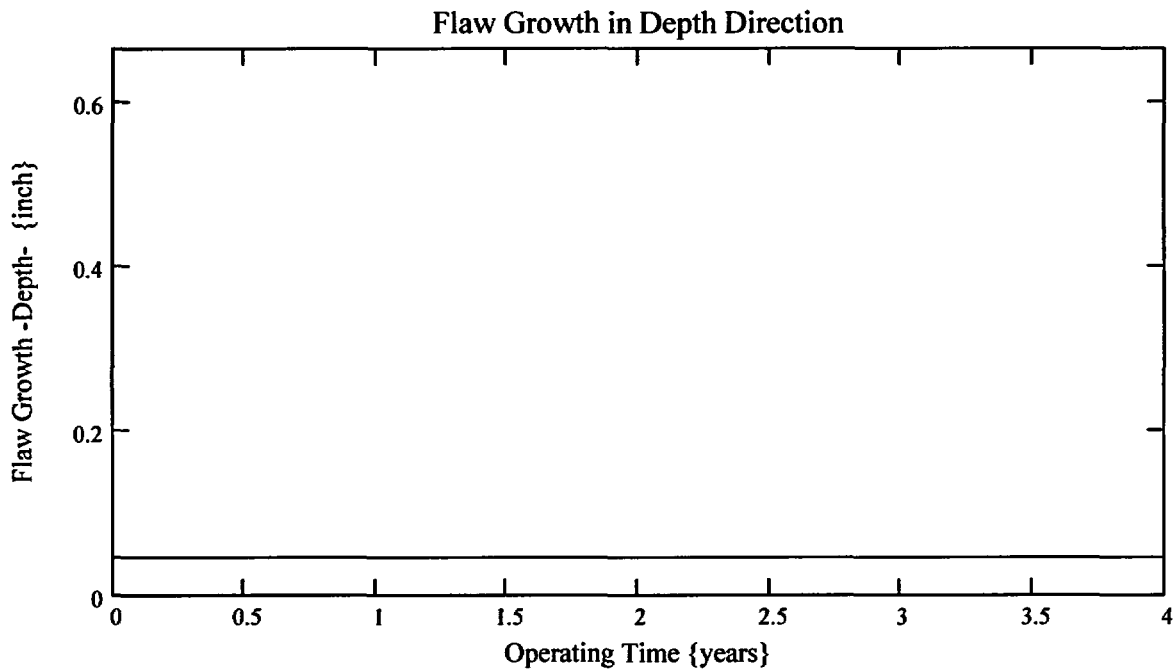
$$FL_{Cntr} := \begin{cases} Ref_{Point} - c_0 & \text{if } Val = 1 \\ Ref_{Point} & \text{if } Val = 2 \\ Ref_{Point} + c_0 & \text{otherwise} \end{cases} \quad \text{Flaw center Location above Nozzle Bottom}$$

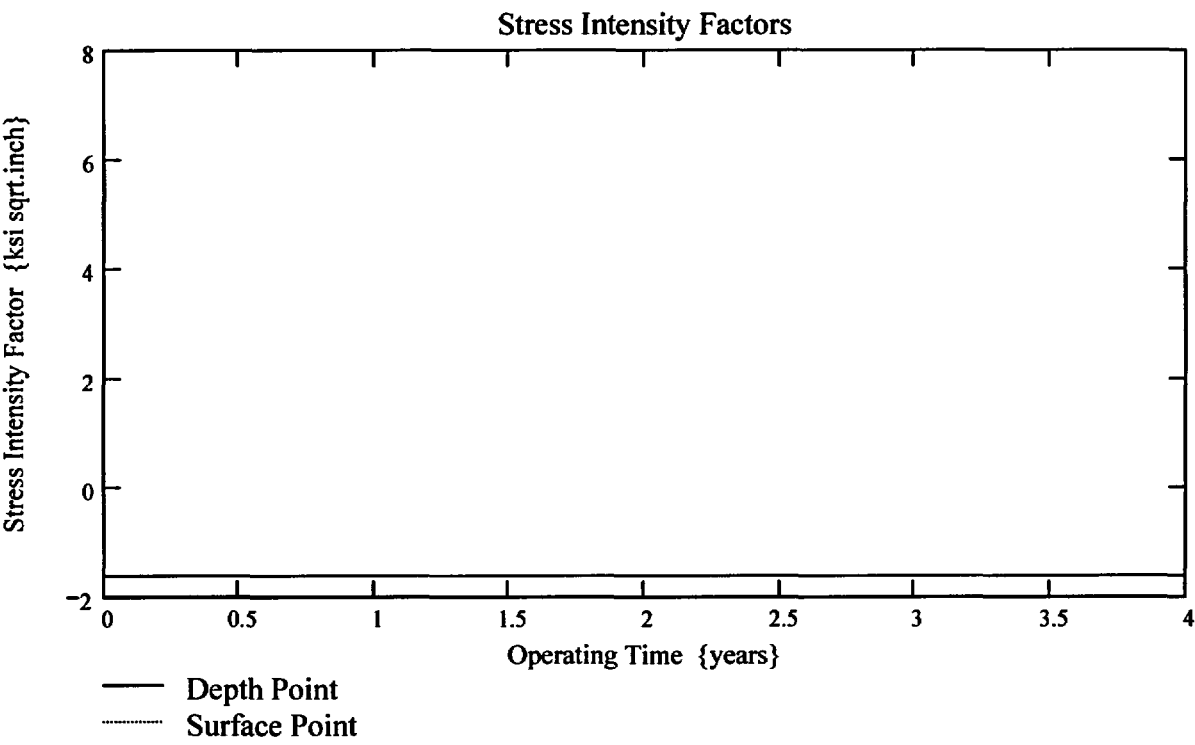
$$U_{Tip} := FL_{Cntr} + c_0 \quad IncStrs.avg := \frac{UL_{Strs.Dist} - U_{Tip}}{20}$$

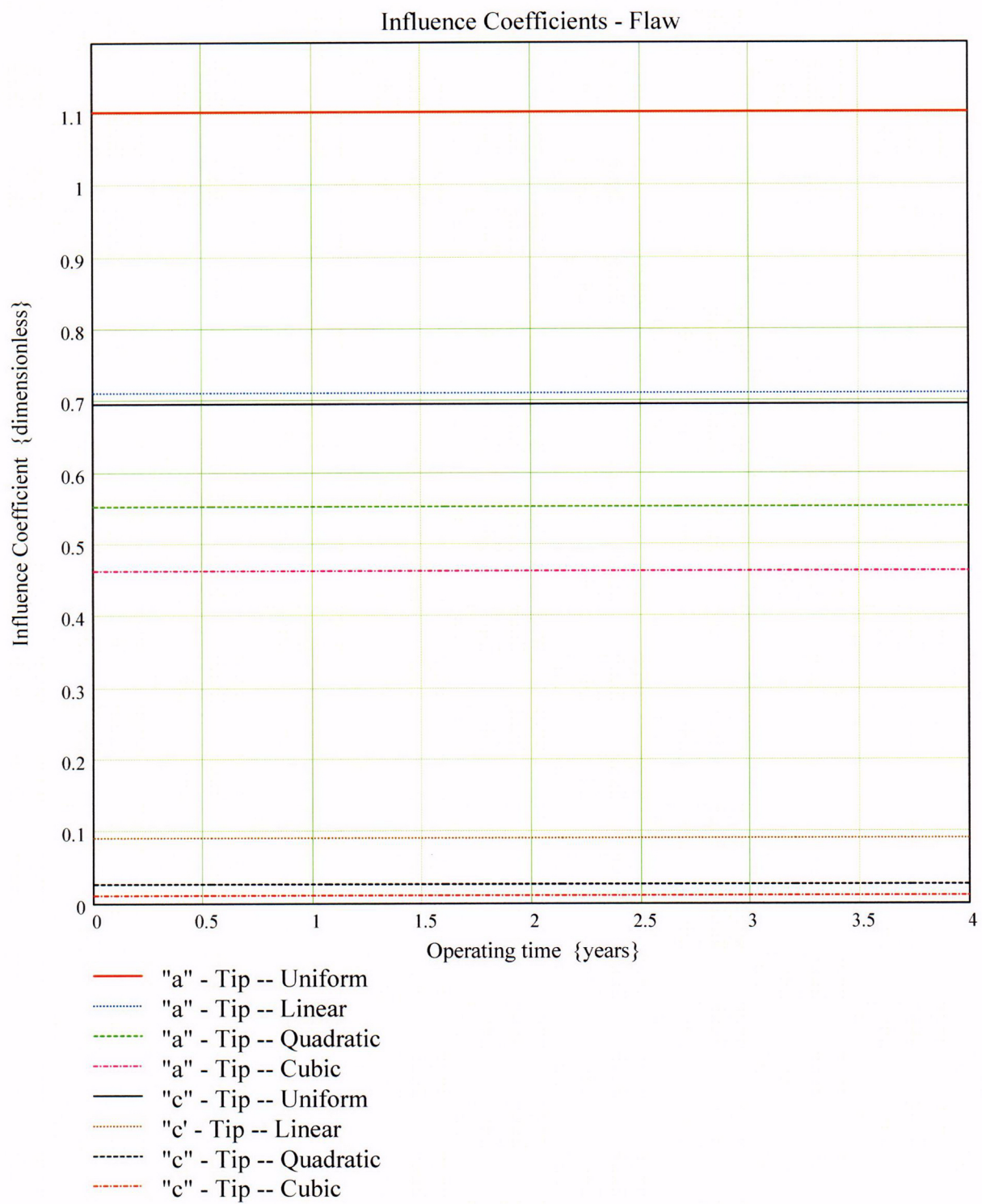
No User Input is required beyond this Point

Sat Aug 09 10:59:39 AM 2003

$\text{PropLength} = 4.924$







C16

$$\text{CGR}_{\text{sambi}(k,8)} =$$

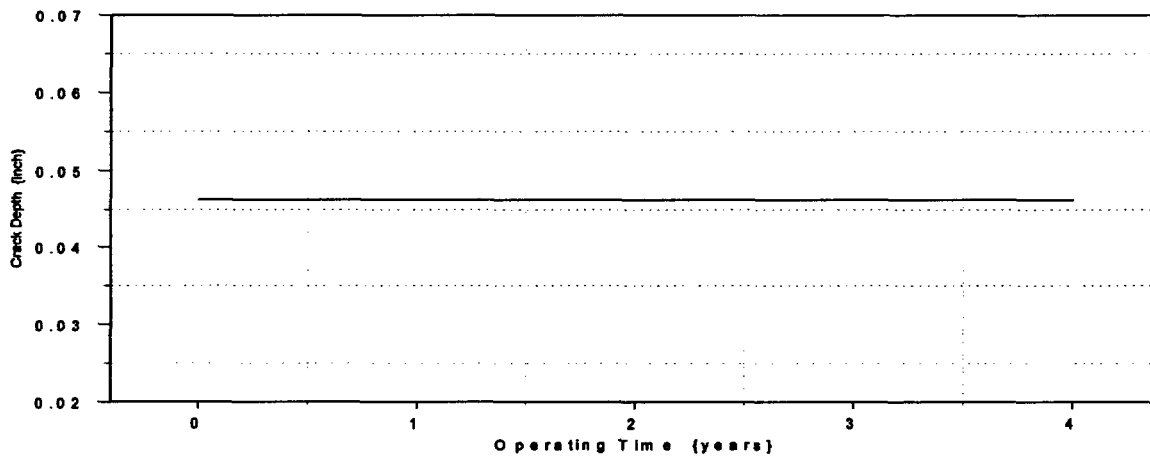
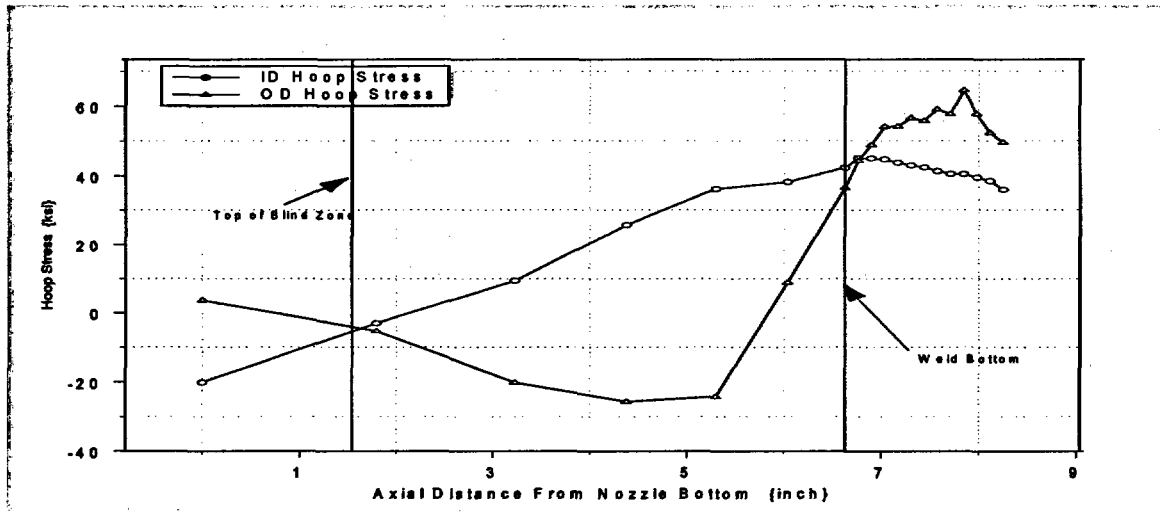
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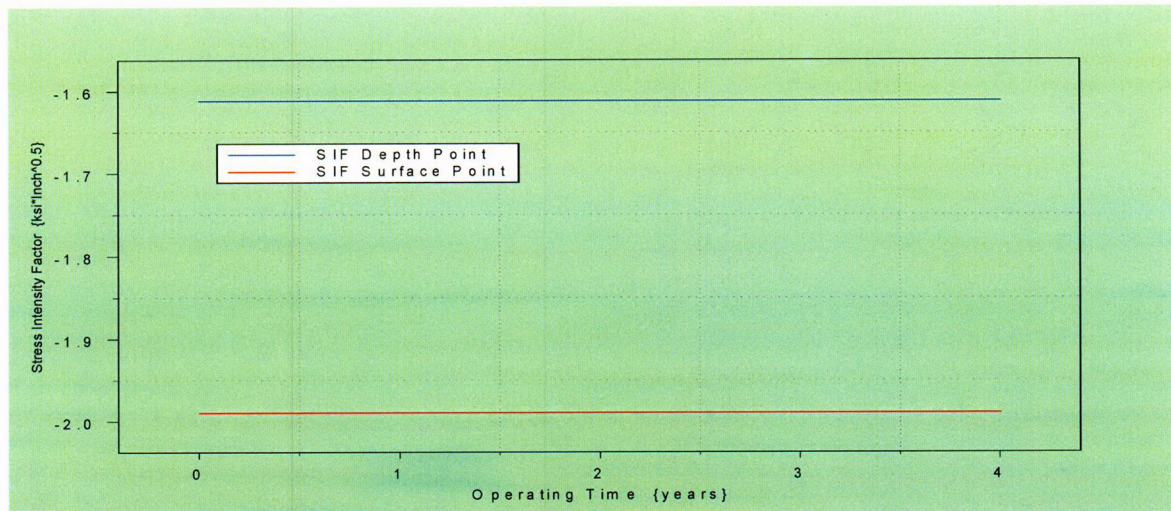
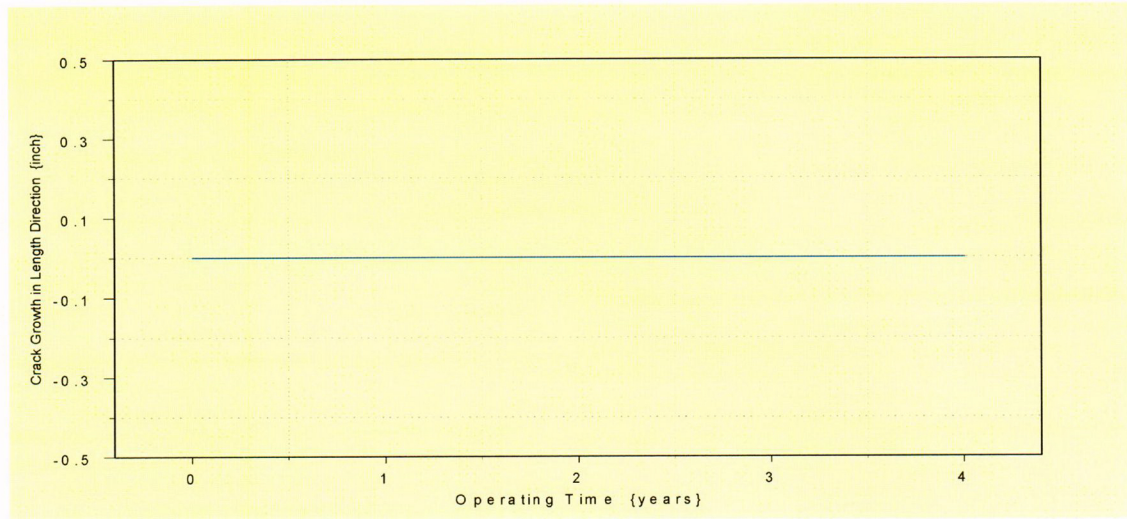
$$\text{CGR}_{\text{sambi}(k,6)} =$$

-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99
-1.99

$$\text{CGR}_{\text{sambi}(k,5)} =$$

-1.612
-1.612
-1.612
-1.612
-1.612
-1.612
-1.612
-1.612
-1.612
-1.612
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-1.612
-1.612
-1.612
-1.612





Primary Water Stress Corrosion Crack Growth Analysis - OD SurfaceFlaw

Developed by Central Engineering Programs, Entergy Operations Inc

Developed by: J. S. Brihmadesan

Verified by: B. C. Gray

References :

- 1) "Stress Intensity factors for Part-through Surface cracks"; NASA TM-11707; July 1992.
- 2) Crack Growth of Alloy 600 Base Metal in PWR Environments; EPRI MRP Report MRP 55 Rev. 1, 2002

Arkansas Nuclear One Unit 2

Component : Reactor Vessel CEDM -"49" Degree Nozzle, Uphill Azimuth,
1.544" above Nozzle Bottom

Calculation Basis: MRP 75 th Percentile and Flaw Face Pressurized

Mean Radius -to- Thickness Ratio:- " R_m/t " -- between 1.0 and 300.0

Note : Used the Metric form of the equation from EPRI MRP 55-Rev. 1.

The correction is applied in the determination of the crack extension to obtain the value in inch/hr .

OD Surface Flaw

The first Required input is a location for a point on the tube elevation to define the point of interest (e.g. The top of the Blind Zone, or bottom of fillet weld etc.). This reference point is necessary to evaluate the stress distribution on the flaw both for the initial flaw and for a growing flaw. This is defined as the reference point. Enter a number (inch) that represents the reference point elevation measured upward from the nozzle end.

$Ref_{point} := 1.544$

To place the flaw with respect to the reference point, the flaw tips and center can be located as follows:

- 1) The Upper "C- tip" located at the reference point (Enter 1)
- 2) The Center of the flaw at the reference point (Enter 2)
- 3) The lower "C- tip" located at the reference point (Enter 3).

$Val := 2$

Upper Limit to be selected for stress distribution (e.g. Weld bottom). This is the elevation from Nozzle Bottom. Enter this value below

$UL_{Strs.Dist} := 6.628$

Upper Axial Extent for Stress Distribution to be used in the Analysis (Axial distance above nozzle bottom)

Input Data :-

$L := 0.32$	Initial Flaw Length
$a_0 := 0.661 \cdot 0.12$	Initial Flaw Depth
$od := 4.05$	Tube OD
$id := 2.728$	Tube ID
$P_{Int} := 2.235$	Design Operating Pressure (internal)
$Years := 4$	Number of Operating Years
$I_{lim} := 1500$	Iteration limit for Crack Growth loop
$T := 604$	Estimate of Operating Temperature
$\alpha_{0c} := 2.67 \cdot 10^{-12}$	Constant in MRP PWSCC Model for I-600 Wrought @ 617 deg. F
$Q_g := 31.0$	Thermal activation Energy for Crack Growth (MRP)
$T_{ref} := 617$	Reference Temperature for normalizing Data deg. F

$$R_o := \frac{od}{2} \quad R_{id} := \frac{id}{2} \quad t := R_o - R_{id} \quad R_m := R_{id} + \frac{t}{2} \quad Tim_{opr} := Years \cdot 365 \cdot 24$$

$$CF_{inhr} := 1.417 \cdot 10^5 \quad C_{blk} := \frac{Tim_{opr}}{I_{lim}} \quad Prnt_{blk} := \left\lceil \frac{I_{lim}}{50} \right\rceil \quad c_0 := \frac{L}{2} \quad R_t := \frac{R_m}{t}$$

$$C_{01} := e^{\left[\frac{-Q_g}{1.103 \cdot 10^{-3}} \cdot \left(\frac{1}{T+459.67} - \frac{1}{T_{ref}+459.67} \right) \right]} \cdot \alpha_{0c} \quad \text{Temperature Correction for Coefficient Alpha}$$

$$C_0 := C_{01}$$

75th percentile MRP-55 Revision 1

Stress Input Data

Input all available Nodal stress data in the table below. The column designations are as follows:
 Column "0" = Axial distance from minimum to maximum recorded on data sheet (inches)
 Column "1" = ID Stress data at each Elevation (ksi)
 Column "2" = Quarter Thickness Stress data at each Elevation (ksi)
 Column "3" = Mid Thickness Stress data at each Elevation (ksi)
 Column "4" = Three Quarter Thickness Stress data at each Elevation (ksi)
 Column "5" = OD Stress data at each Elevation (ksi)

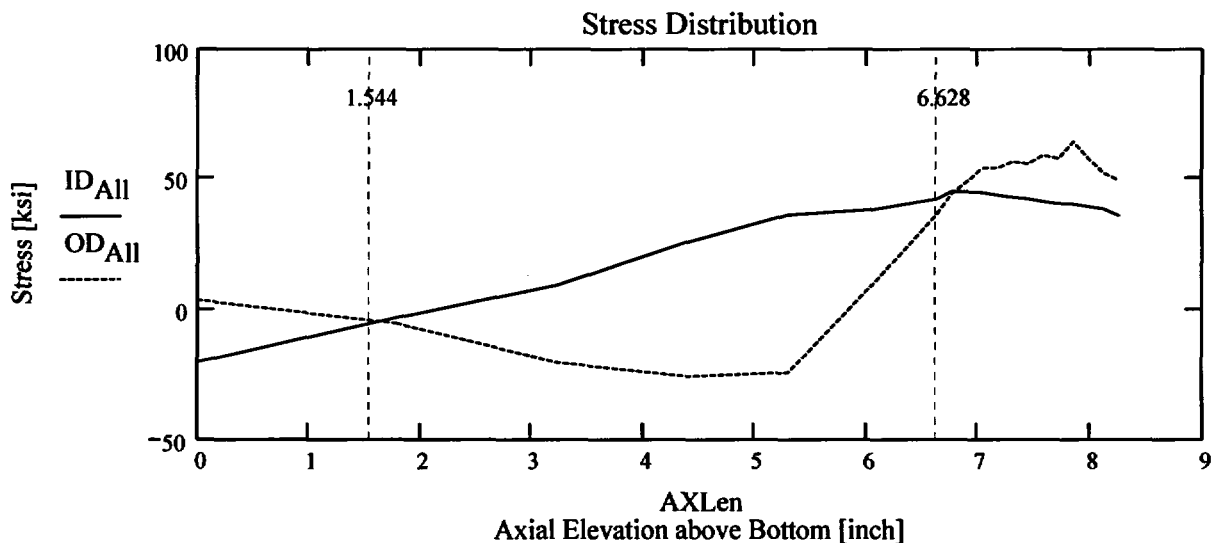
AllData :=

	0	1	2	3	4	5
0	0	-20.18	-11.45	-5.94	-1.16	3.7
1	1.79	-3.02	-4.38	-5.44	-5.51	-5.34
2	3.23	9.4	12.13	-0.26	-12.62	-20.23
3	4.38	25.65	24.71	14.58	-15.3	-25.69
4	5.3	36.18	33.79	26.29	-5.92	-24.31
5	6.04	38.11	35.03	31.43	21.21	8.83
6	6.63	42.19	38.1	36.25	40.68	36.41
7	6.76	45.07	42.22	42.74	47.55	44.23
8	6.9	44.97	43.61	46.01	49.99	48.8
9	7.03	44.7	44.12	47.02	51.04	54.11
10	7.17	43.72	43.97	47.64	50.17	54.17

AXLen := AllData⁽⁰⁾

ID_{All} := AllData⁽¹⁾

OD_{All} := AllData⁽⁵⁾



Observing the stress distribution select the region in the table above labeled Data_{All} that represents the region of interest. This needs to be done especially for distributions that have a large compressive stress at the nozzle bottom and high tensile stresses at the J-weld location. Copy the selection in the above table, click on the "Data" statement below and delete it from the edit menu. Type "Data and the Mathcad "equal" sign (Shift-Colon) then insert the same to the right of the Mathcad Equals sign below (paste symbol).

$$\text{Data} := \begin{pmatrix} 0 & -20.175 & -11.45 & -5.94 & -1.163 & 3.704 \\ 1.792 & -3.024 & -4.378 & -5.443 & -5.511 & -5.341 \\ 3.228 & 9.398 & 12.134 & -0.258 & -12.622 & -20.232 \\ 4.378 & 25.65 & 24.71 & 14.577 & -15.299 & -25.689 \\ 5.299 & 36.179 & 33.787 & 26.287 & -5.925 & -24.306 \\ 6.037 & 38.106 & 35.028 & 31.43 & 21.215 & 8.834 \\ 6.628 & 42.186 & 38.102 & 36.248 & 40.684 & 36.405 \\ 6.764 & 45.067 & 42.217 & 42.736 & 47.553 & 44.235 \\ 6.899 & 44.968 & 43.606 & 46.007 & 49.995 & 48.803 \\ 7.035 & 44.695 & 44.12 & 47.021 & 51.043 & 54.113 \\ 7.17 & 43.723 & 43.973 & 47.639 & 50.172 & 54.17 \end{pmatrix}$$

$$\text{Axl} := \text{Data}^{(0)} \quad \text{MD} := \text{Data}^{(3)} \quad \text{ID} := \text{Data}^{(1)} \quad \text{TQ} := \text{Data}^{(4)} \quad \text{QT} := \text{Data}^{(2)} \quad \text{OD} := \text{Data}^{(5)}$$

$$R_{ID} := \text{regress}(\text{Axl}, \text{ID}, 3)$$

$$R_{QT} := \text{regress}(\text{Axl}, \text{QT}, 3)$$

$$R_{OD} := \text{regress}(\text{Axl}, \text{OD}, 3)$$

$$R_{MD} := \text{regress}(\text{Axl}, \text{MD}, 3)$$

$$R_{TQ} := \text{regress}(\text{Axl}, \text{TQ}, 3)$$


$$\text{FL}_{\text{Cntr}} := \begin{cases} \text{Ref}_{\text{Point}} - c_0 & \text{if Val} = 1 \\ \text{Ref}_{\text{Point}} & \text{if Val} = 2 \\ \text{Ref}_{\text{Point}} + c_0 & \text{otherwise} \end{cases}$$

Flaw center Location Location above Nozzle Bottom

$$U_{\text{Tip}} := \text{FL}_{\text{Cntr}} + c_0$$

$$\text{IncStrs.avg} := \frac{\text{UL}_{\text{Strs.Dist}} - U_{\text{Tip}}}{n}$$

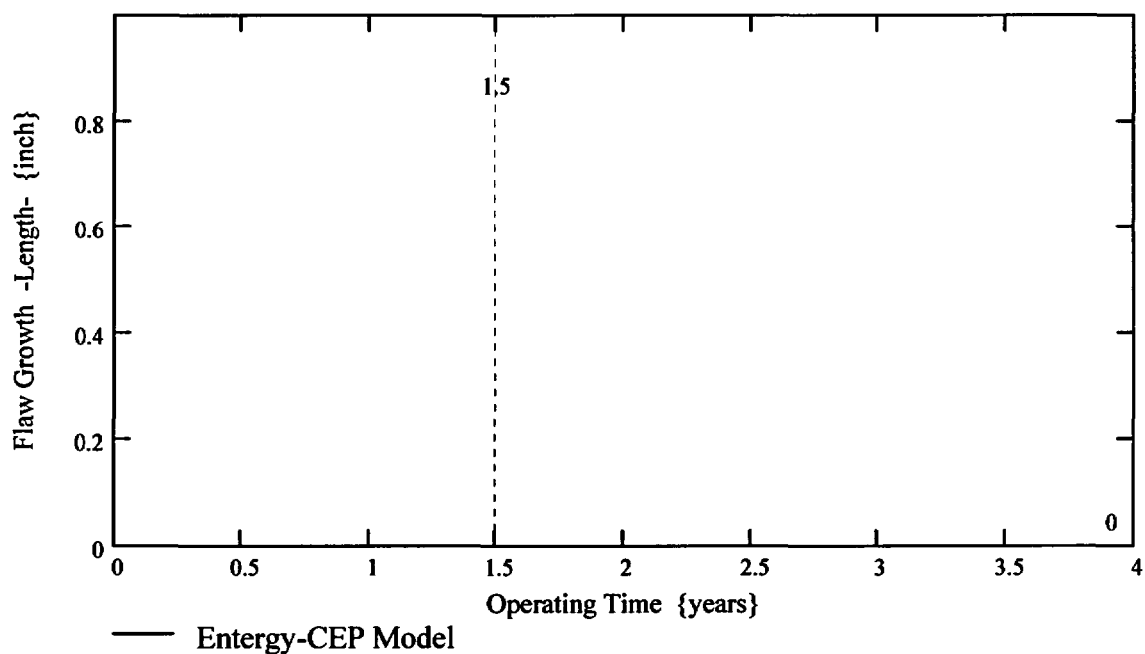
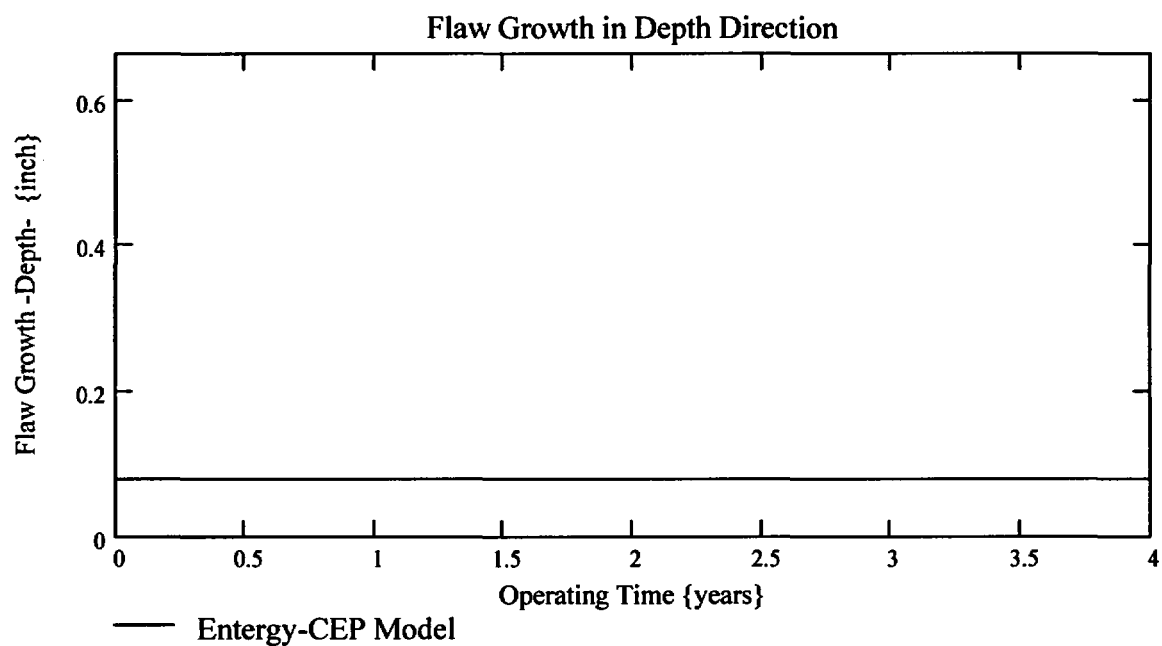
No User Input is required beyond this Point

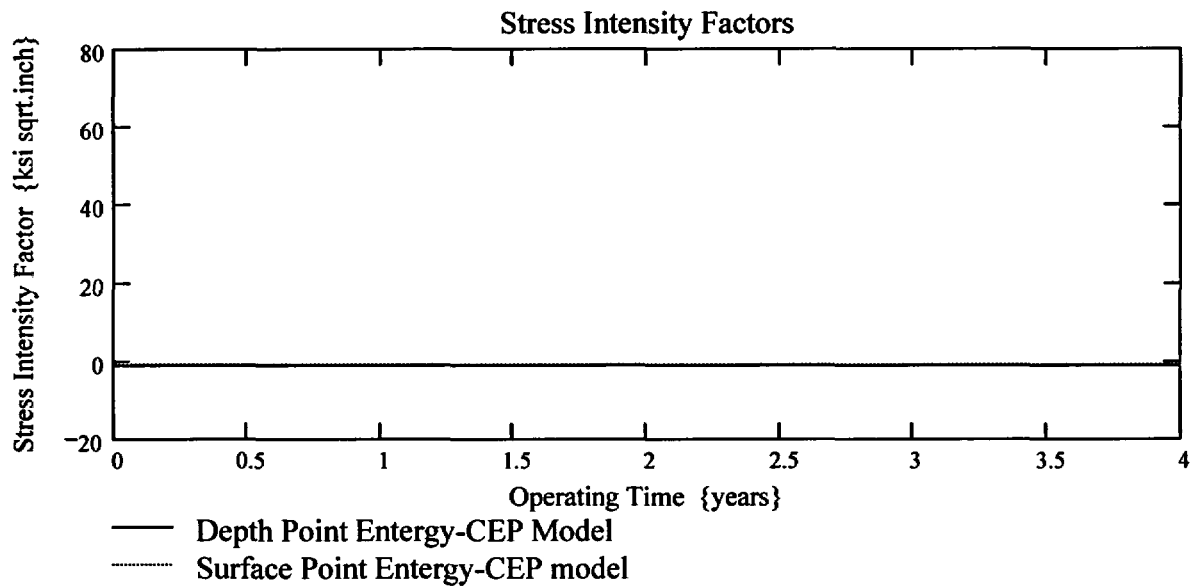
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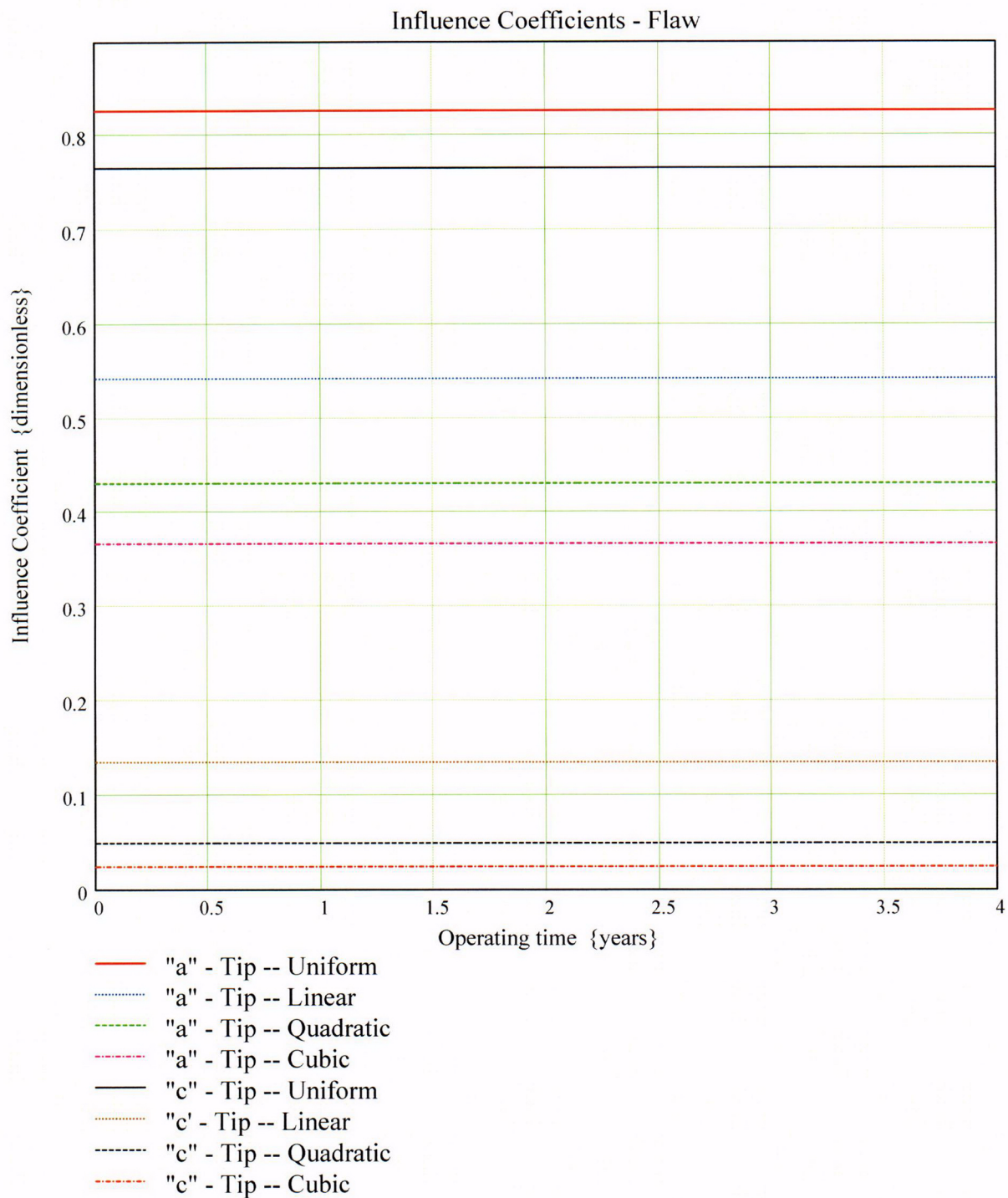
Developed by:
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$$\text{PropLength} = 4.924$$







$$CGR_{sambi(k,8)} =$$

0.827
0.827
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$$CGR_{sambi(k,6)} =$$

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-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776
-0.776

$$CGR_{sambi(k,5)} =$$

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-1.013
-1.013
-1.013
-1.013
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-1.013
-1.013
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